

Engineering Mechanics

# STATICS

Fourteenth Edition

R. C. Hibbeler



# Fundamental Equations of Statics

## Cartesian Vector

$$\mathbf{A} = A_x \mathbf{i} + A_y \mathbf{j} + A_z \mathbf{k}$$

Magnitude

$$A = \sqrt{A_x^2 + A_y^2 + A_z^2}$$

Directions

$$\begin{aligned} \mathbf{u}_A &= \frac{\mathbf{A}}{A} = \frac{A_x}{A} \mathbf{i} + \frac{A_y}{A} \mathbf{j} + \frac{A_z}{A} \mathbf{k} \\ &= \cos \alpha \mathbf{i} + \cos \beta \mathbf{j} + \cos \gamma \mathbf{k} \\ \cos^2 \alpha + \cos^2 \beta + \cos^2 \gamma &= 1 \end{aligned}$$

## Dot Product

$$\begin{aligned} \mathbf{A} \cdot \mathbf{B} &= AB \cos \theta \\ &= A_x B_x + A_y B_y + A_z B_z \end{aligned}$$

## Cross Product

$$\mathbf{C} = \mathbf{A} \times \mathbf{B} = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ A_x & A_y & A_z \\ B_x & B_y & B_z \end{vmatrix}$$

## Cartesian Position Vector

$$\mathbf{r} = (x_2 - x_1) \mathbf{i} + (y_2 - y_1) \mathbf{j} + (z_2 - z_1) \mathbf{k}$$

## Cartesian Force Vector

$$\mathbf{F} = F \mathbf{u} = F \left( \frac{\mathbf{r}}{r} \right)$$

## Moment of a Force

$$\begin{aligned} M_o &= Fd \\ \mathbf{M}_o &= \mathbf{r} \times \mathbf{F} = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ r_x & r_y & r_z \\ F_x & F_y & F_z \end{vmatrix} \end{aligned}$$

## Moment of a Force About a Specified Axis

$$M_a = \mathbf{u} \cdot \mathbf{r} \times \mathbf{F} = \begin{vmatrix} u_x & u_y & u_z \\ r_x & r_y & r_z \\ F_x & F_y & F_z \end{vmatrix}$$

## Simplification of a Force and Couple System

$$\begin{aligned} \mathbf{F}_R &= \Sigma \mathbf{F} \\ (\mathbf{M}_R)_O &= \Sigma \mathbf{M} + \Sigma \mathbf{M}_O \end{aligned}$$

## Equilibrium

Particle

$$\Sigma F_x = 0, \Sigma F_y = 0, \Sigma F_z = 0$$

Rigid Body-Two Dimensions

$$\Sigma F_x = 0, \Sigma F_y = 0, \Sigma M_O = 0$$

Rigid Body-Three Dimensions

$$\begin{aligned} \Sigma F_x &= 0, \Sigma F_y = 0, \Sigma F_z = 0 \\ \Sigma M_{x'} &= 0, \Sigma M_{y'} = 0, \Sigma M_{z'} = 0 \end{aligned}$$

## Friction

Static (maximum)  $F_s = \mu_s N$

Kinetic  $F_k = \mu_k N$

## Center of Gravity

Particles or Discrete Parts

$$\bar{r} = \frac{\Sigma \tilde{r} W}{\Sigma W}$$

Body

$$\bar{r} = \frac{\int \tilde{r} dW}{\int dW}$$

## Area and Mass Moments of Inertia

$$I = \int r^2 dA \quad I = \int r^2 dm$$

Parallel-Axis Theorem

$$I = \bar{I} + Ad^2 \quad I = \bar{I} + md^2$$

Radius of Gyration

$$k = \sqrt{\frac{I}{A}} \quad k = \sqrt{\frac{I}{m}}$$

## Virtual Work

$$\delta U = 0$$



## SI Prefixes

<i>Multiple</i>	<i>Exponential Form</i>	<i>Prefix</i>	<i>SI Symbol</i>
1 000 000 000	$10^9$	giga	G
1 000 000	$10^6$	mega	M
1 000	$10^3$	kilo	k
<hr/>			
<i>Submultiple</i>			
0.001	$10^{-3}$	milli	m
0.000 001	$10^{-6}$	micro	$\mu$
0.000 000 001	$10^{-9}$	nano	n

## Conversion Factors (FPS) to (SI)

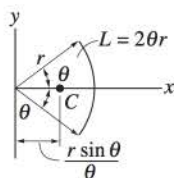
<i>Quantity</i>	<i>Unit of Measurement (FPS)</i>	<i>Equals</i>	<i>Unit of Measurement (SI)</i>
Force	lb		4.448 N
Mass	slug		14.59 kg
Length	ft		0.3048 m

## Conversion Factors (FPS)

1 ft = 12 in. (inches)  
 1 mi. (mile) = 5280 ft  
 1 kip (kilopound) = 1000 lb  
 1 ton = 2000 lb

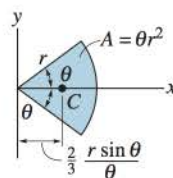
# Geometric Properties of Line and Area Elements

## Centroid Location



Circular arc segment

## Centroid Location

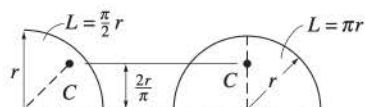


Circular sector area

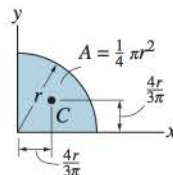
## Area Moment of Inertia

$$I_x = \frac{1}{4} r^4 (\theta - \frac{1}{2} \sin 2\theta)$$

$$I_y = \frac{1}{4} r^4 (\theta + \frac{1}{2} \sin 2\theta)$$



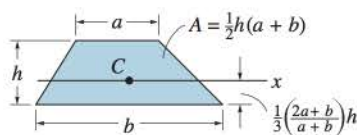
Quarter and semicircle arcs



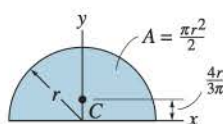
Quarter circle area

$$I_x = \frac{1}{16} \pi r^4$$

$$I_y = \frac{1}{16} \pi r^4$$



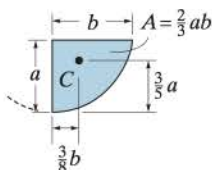
Trapezoidal area



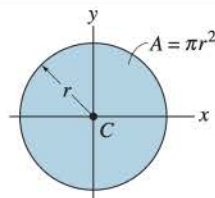
Semicircular area

$$I_x = \frac{1}{8} \pi r^4$$

$$I_y = \frac{1}{8} \pi r^4$$



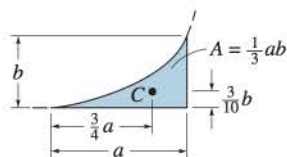
Semiparabolic area



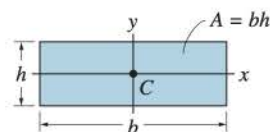
Circular area

$$I_x = \frac{1}{4} \pi r^4$$

$$I_y = \frac{1}{4} \pi r^4$$



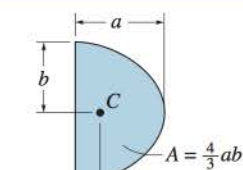
Exparabolic area



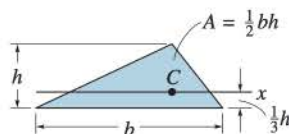
Rectangular area

$$I_x = \frac{1}{12} b h^3$$

$$I_y = \frac{1}{12} h b^3$$



Parabolic area

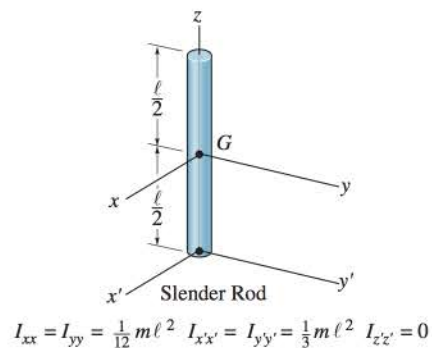
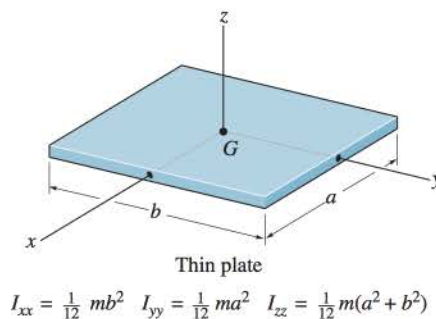
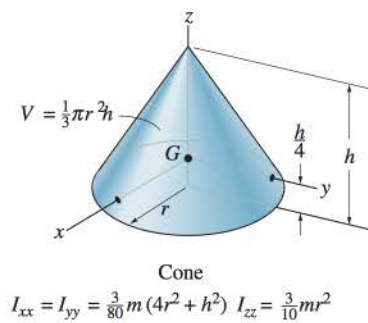
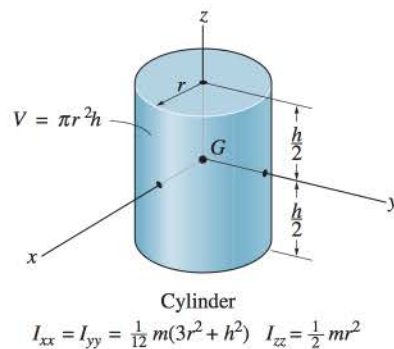
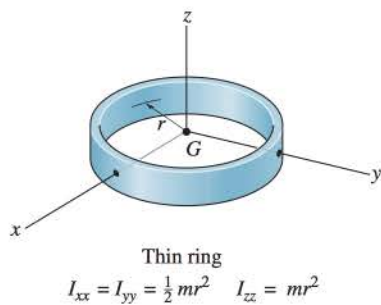
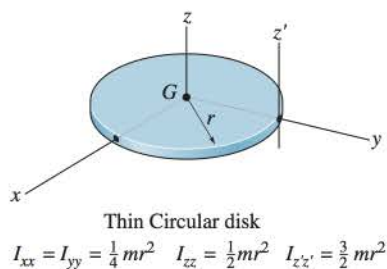
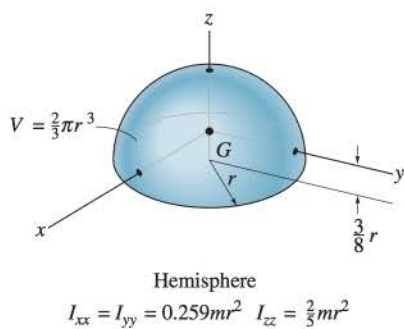
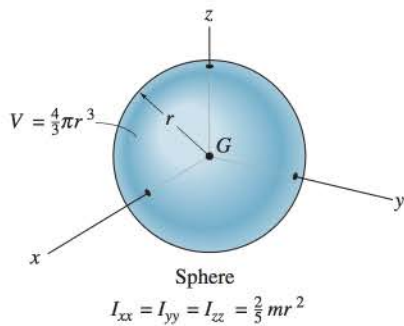


Triangular area

$$I_x = \frac{1}{36} b h^3$$



# Center of Gravity and Mass Moment of Inertia of Homogeneous Solids



ENGINEERING MECHANICS

# STATICS

FOURTEENTH EDITION



*This page intentionally left blank*

ENGINEERING MECHANICS

# STATICS

FOURTEENTH EDITION

R. C. HIBBELER

PEARSON

Hoboken Boston Columbus San Francisco New York  
Indianapolis London Toronto Sydney Singapore Tokyo Montreal Dubai  
Madrid Hong Kong Mexico City Munich Paris Amsterdam Cape Town



## Library of Congress Cataloging-in-Publication Data on File

*Vice President and Editorial Director, ECS:* Marcia Horton  
*Senior Editor:* Norrin Dias  
*Editorial Assistant:* Michelle Bayman  
*Program and Project Management Team Lead:* Scott Disanno  
*Program Manager:* Sandra L. Rodriguez  
*Project Manager:* Rose Kernan  
*Cover Designer:* Black Horse Designs  
*Art Editor:* Gregory Dulles  
*Sr. Digital Producer:* Felipe Gonzalez  
*Operations Specialist:* Maura Zaldivar-Garcia  
*Product Marketing Manager:* Bram Van Kempen  
*Field Marketing Manager:* Demetrius Hall  
*Marketing Assistant:* Jon Bryant

*Cover Image:* Cubm/Shutterstock

---

© 2016 by R.C. Hibbeler  
Published by Pearson Prentice Hall  
Pearson Education, Inc.  
Hoboken, New Jersey 07030

All rights reserved. No part of this book may be reproduced or transmitted in any form or by any means, without permission in writing from the publisher.

Pearson Prentice Hall™ is a trademark of Pearson Education, Inc.

The author and publisher of this book have used their best efforts in preparing this book. These efforts include the development, research, and testing of the theories and programs to determine their effectiveness. The author and publisher shall not be liable in any event for incidental or consequential damages with, or arising out of, the furnishing, performance, or use of these programs.

Pearson Education Ltd., *London*  
Pearson Education Australia Pty. Ltd., *Sydney*  
Pearson Education Singapore, Pte. Ltd.  
Pearson Education North Asia Ltd., *Hong Kong*  
Pearson Education Canada, Inc., *Toronto*  
Pearson Educación de Mexico, S.A. de C.V.  
Pearson Education — Japan, *Tokyo*  
Pearson Education Malaysia, Pte. Ltd.  
Pearson Education, Inc., *Hoboken, New Jersey*

Printed in the United States of America

10 9 8 7 6 5 4 3 2 1

ISBN-10: 0133918920

ISBN-13: 9780133918922

**PEARSON**

## To the Student

With the hope that this work will stimulate  
an interest in Engineering Mechanics  
and provide an acceptable guide to its understanding.



*This page intentionally left blank*

The main purpose of this book is to provide the student with a clear and thorough presentation of the theory and application of engineering mechanics. To achieve this objective, this work has been shaped by the comments and suggestions of hundreds of reviewers in the teaching profession, as well as many of the author's students.

---

## New to this Edition

**Preliminary Problems.** This new feature can be found throughout the text, and is given just before the Fundamental Problems. The intent here is to test the student's conceptual understanding of the theory. Normally the solutions require little or no calculation, and as such, these problems provide a basic understanding of the concepts before they are applied numerically. All the solutions are given in the back of the text.

**Expanded Important Points Sections.** Summaries have been added which reinforces the reading material and highlights the important definitions and concepts of the sections.

**Re-writing of Text Material.** Further clarification of concepts has been included in this edition, and important definitions are now in boldface throughout the text to highlight their importance.

**End-of-the-Chapter Review Problems.** All the review problems now have solutions given in the back, so that students can check their work when studying for exams, and reviewing their skills when the chapter is finished.

**New Photos.** The relevance of knowing the subject matter is reflected by the real-world applications depicted in the over 30 new or updated photos placed throughout the book. These photos generally are used to explain how the relevant principles apply to real-world situations and how materials behave under load.

**New Problems.** There are approximately 30% new problems that have been added to this edition, which involve applications to many different fields of engineering.

---

## Hallmark Features

Besides the new features mentioned above, other outstanding features that define the contents of the text include the following.

**Organization and Approach.** Each chapter is organized into well-defined sections that contain an explanation of specific topics, illustrative example problems, and a set of homework problems. The topics within each section are placed into subgroups defined by boldface titles. The purpose of this is to present a structured method for introducing each new definition or concept and to make the book convenient for later reference and review.

**Chapter Contents.** Each chapter begins with an illustration demonstrating a broad-range application of the material within the chapter. A bulleted list of the chapter contents is provided to give a general overview of the material that will be covered.

**Emphasis on Free-Body Diagrams.** Drawing a free-body diagram is particularly important when solving problems, and for this reason this step is strongly emphasized throughout the book. In particular, special sections and examples are devoted to show how to draw free-body diagrams. Specific homework problems have also been added to develop this practice.

**Procedures for Analysis.** A general procedure for analyzing any mechanical problem is presented at the end of the first chapter. Then this procedure is customized to relate to specific types of problems that are covered throughout the book. This unique feature provides the student with a logical and orderly method to follow when applying the theory. The example problems are solved using this outlined method in order to clarify its numerical application. Realize, however, that once the relevant principles have been mastered and enough confidence and judgment have been obtained, the student can then develop his or her own procedures for solving problems.

**Important Points.** This feature provides a review or summary of the most important concepts in a section and highlights the most significant points that should be realized when applying the theory to solve problems.

**Fundamental Problems.** These problem sets are selectively located just after most of the example problems. They provide students with simple applications of the concepts, and therefore, the chance to develop their problem-solving skills before attempting to solve any of the standard problems that follow. In addition, they can be used for preparing for exams, and they can be used at a later time when preparing for the Fundamentals in Engineering Exam.

**Conceptual Understanding.** Through the use of photographs placed throughout the book, theory is applied in a simplified way in order to illustrate some of its more important conceptual features and instill the physical meaning of many

of the terms used in the equations. These simplified applications increase interest in the subject matter and better prepare the student to understand the examples and solve problems.

**Homework Problems.** Apart from the Fundamental and Conceptual type problems mentioned previously, other types of problems contained in the book include the following:

- **Free-Body Diagram Problems.** Some sections of the book contain introductory problems that only require drawing the free-body diagram for the specific problems within a problem set. These assignments will impress upon the student the importance of mastering this skill as a requirement for a complete solution of any equilibrium problem.
- **General Analysis and Design Problems.** The majority of problems in the book depict realistic situations encountered in engineering practice. Some of these problems come from actual products used in industry. It is hoped that this realism will both stimulate the student's interest in engineering mechanics and provide a means for developing the skill to reduce any such problem from its physical description to a model or symbolic representation to which the principles of mechanics may be applied.

Throughout the book, there is an approximate balance of problems using either SI or FPS units. Furthermore, in any set, an attempt has been made to arrange the problems in order of increasing difficulty except for the end of chapter review problems, which are presented in random order.

- **Computer Problems.** An effort has been made to include some problems that may be solved using a numerical procedure executed on either a desktop computer or a programmable pocket calculator. The intent here is to broaden the student's capacity for using other forms of mathematical analysis without sacrificing the time needed to focus on the application of the principles of mechanics. Problems of this type, which either can or must be solved using numerical procedures, are identified by a "square" symbol (■) preceding the problem number.

The many homework problems in this edition, have been placed into two different categories. Problems that are simply indicated by a problem number have an answer and in some cases an additional numerical result given in the back of the book. An asterisk (\*) before every fourth problem number indicates a problem without an answer.

**Accuracy.** As with the previous editions, apart from the author, the accuracy of the text and problem solutions has been thoroughly checked by four other parties: Scott Hendricks, Virginia Polytechnic Institute and State University; Karim Nohra, University of South Florida; Kurt Norlin, Bittner Development Group; and finally Kai Beng, a practicing engineer, who in addition to accuracy review provided suggestions for problem development.

---

## Contents

The book is divided into 11 chapters, in which the principles are first applied to simple, then to more complicated situations. In a general sense, each principle is applied first to a particle, then a rigid body subjected to a coplanar system of forces, and finally to three-dimensional force systems acting on a rigid body.

Chapter 1 begins with an introduction to mechanics and a discussion of units. The vector properties of a concurrent force system are introduced in Chapter 2. This theory is then applied to the equilibrium of a particle in Chapter 3. Chapter 4 contains a general discussion of both concentrated and distributed force systems and the methods used to simplify them. The principles of rigid-body equilibrium are developed in Chapter 5 and then applied to specific problems involving the equilibrium of trusses, frames, and machines in Chapter 6, and to the analysis of internal forces in beams and cables in Chapter 7. Applications to problems involving frictional forces are discussed in Chapter 8, and topics related to the center of gravity and centroid are treated in Chapter 9. If time permits, sections involving more advanced topics, indicated by stars (★), may be covered. Most of these topics are included in Chapter 10 (area and mass moments of inertia) and Chapter 11 (virtual work and potential energy). Note that this material also provides a suitable reference for basic principles when it is discussed in more advanced courses. Finally, Appendix A provides a review and list of mathematical formulas needed to solve the problems in the book.

**Alternative Coverage.** At the discretion of the instructor, some of the material may be presented in a different sequence with no loss of continuity. For example, it is possible to introduce the concept of a force and all the necessary methods of vector analysis by first covering Chapter 2 and Section 4.2 (the cross product). Then after covering the rest of Chapter 4 (force and moment systems), the equilibrium methods of Chapters 3 and 5 can be discussed.

---

## Acknowledgments

The author has endeavored to write this book so that it will appeal to both the student and instructor. Through the years, many people have helped in its development, and I will always be grateful for their valued suggestions and comments. Specifically, I wish to thank all the individuals who have contributed their comments relative to preparing the fourteenth edition of this work, and in particular, O. Barton, Jr. of the U.S. Naval Academy, K. Cook-Chennault at Rutgers, the State University of New Jersey, A. Itani of the University of Nevada, Y. Laio of Arizona State University, T. Miller of Oregon State University, I. Orabi of the University of New Haven, M. Reynolds of the University of Arkansas, N. Schulz of the University of Portland, C. Sulzbach of the Colorado School of Mines, R. Viesca of Tufts University, P. Ziehl of the University of South Carolina.



There are a few other people that I also feel deserve particular recognition. These include comments sent to me by J. Dix, H. Kuhlman, S. Larwood, D. Pollock, H. Wenzel. A long-time friend and associate, Kai Beng Yap, was of great help to me in preparing and checking problem solutions. A special note of thanks also goes to Kurt Norlin of Bittner Development Group in this regard. During the production process I am thankful for the assistance of Martha McMaster, my proofreader and Rose Kernan, my production editor. Also, to my wife, Conny, who helped in the preparation of the manuscript for publication.

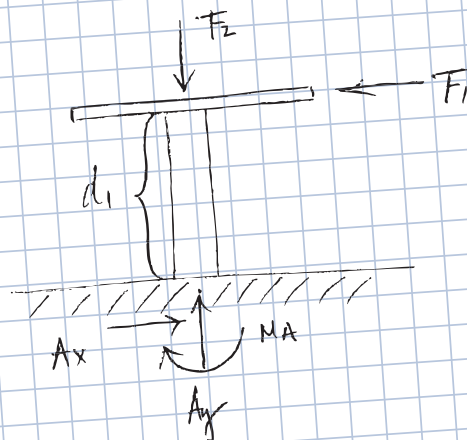
Lastly, many thanks are extended to all my students and to members of the teaching profession who have freely taken the time to e-mail me their suggestions and comments. Since this list is too long to mention, it is hoped that those who have given help in this manner will accept this anonymous recognition.

I would greatly appreciate hearing from you if at any time you have any comments, suggestions, or problems related to any matters regarding this edition.

*Russell Charles Hibbeler*  
*hibbeler@bellsouth.net*

your work...

PART A



given:  $F_1 = 55.0 \text{ lb}$ ,  $F_2 = 170 \text{ lb}$ ,  $M_A = 160 \text{ lb}\cdot\text{ft}$

$$\sum M_A = 0 \Rightarrow (F_1 + F_2) \times d_1 - M_A = 0$$

$$(F_1 + F_2) \times d_1 = M_A$$

$$d_1 = \frac{M_A}{F_1 + F_2}$$

$$= \frac{160 \text{ lb}\cdot\text{ft}}{55.0 \text{ lb} + 170 \text{ lb}}$$

$$d_1 = 0.711 \text{ ft}$$

# your answer **specific feedback**

Express your answer numerically in feet to three significant figures.

$\sqrt{\square}$

$\Delta \Sigma \phi$

$\updownarrow$

vec

$\curvearrowright$

$\curvearrowleft$

$\circlearrowright$

?

$d_1 =$  $\text{ft}$

**Submit**

[Hints](#)

[My Answers](#)

[Give Up](#)

[Review Part](#)

**Incorrect; Try Again; 5 attempts remaining**

The sum of the two forces do not contribute to the moment about point A. The magnitude of the moment about A is equal to the force multiplied by the perpendicular distance between point A and the line of action of the force. What is the perpendicular distance between each force's line of action and point A?

---

## Resources for Instructors

- **MasteringEngineering.** This online Tutorial Homework program allows you to integrate dynamic homework with automatic grading and adaptive tutoring. MasteringEngineering allows you to easily track the performance of your entire class on an assignment-by-assignment basis, or the detailed work of an individual student.
- **Instructor's Solutions Manual.** This supplement provides complete solutions supported by problem statements and problem figures. The fourteenth edition manual was revised to improve readability and was triple accuracy checked. The Instructor's Solutions Manual is available on Pearson Higher Education website: [www.pearsonhighered.com](http://www.pearsonhighered.com).
- **Instructor's Resource.** Visual resources to accompany the text are located on the Pearson Higher Education website: [www.pearsonhighered.com](http://www.pearsonhighered.com). If you are in need of a login and password for this site, please contact your local Pearson representative. Visual resources include all art from the text, available in PowerPoint slide and JPEG format.
- **Video Solutions.** Developed by Professor Edward Berger, University of Virginia, video solutions are located in the study area of MasteringEngineering and offer step-by-step solution walkthroughs of representative homework problems from each section of the text. Make efficient use of class time and office hours by showing students the complete and concise problem-solving approaches that they can access any time and view at their own pace. The videos are designed to be a flexible resource to be used however each instructor and student prefers. A valuable tutorial resource, the videos are also helpful for student self-evaluation as students can pause the videos to check their understanding and work alongside the video.

---

## Resources for Students

- **MasteringEngineering.** Tutorial homework problems emulate the instructor's office-hour environment, guiding students through engineering concepts with self-paced individualized coaching. These in-depth tutorial homework problems are designed to coach students with feedback specific to their errors and optional hints that break problems down into simpler steps.
- **Statics Study Pack.** This supplement contains chapter-by-chapter study materials and a Free-Body Diagram Workbook.
- **Video Solutions.** Complete, step-by-step solution walkthroughs of representative homework problems from each section. Videos offer fully worked solutions that show every step of representative homework problems—this helps students make vital connections between concepts. Videos are available in the Study Area of MasteringEngineering.
- **Statics Practice Problems Workbook.** This workbook contains additional worked problems. The problems are partially solved and are designed to help guide students through difficult topics.

---

## Ordering Options

The *Statics Study Pack* is available as a stand-alone item for student purchase and also available packaged with the texts. The ISBN for the valuepack is as follows:

- *Engineering Mechanics: Statics* with Study Pack: ISBN: 0134136683
- *Engineering Mechanics: Statics* Plus MasteringEngineering with Pearson eText—Access Card Package: ISBN 0134160681

---

## Custom Solutions

Please contact your local Pearson Sales Representative for more details about custom options or visit

[www.pearsonlearningsolutions.com](http://www.pearsonlearningsolutions.com), keyword: Hibbeler.



*This page intentionally left blank*

# CONTENTS



## 1

### General Principles 3

Chapter Objectives 3

- 1.1 Mechanics 3
- 1.2 Fundamental Concepts 4
- 1.3 Units of Measurement 7
- 1.4 The International System of Units 9
- 1.5 Numerical Calculations 10
- 1.6 General Procedure for Analysis 12



## 2

### Force Vectors 17

Chapter Objectives 17

- 2.1 Scalars and Vectors 17
- 2.2 Vector Operations 18
- 2.3 Vector Addition of Forces 20
- 2.4 Addition of a System of Coplanar Forces 33
- 2.5 Cartesian Vectors 44
- 2.6 Addition of Cartesian Vectors 47
- 2.7 Position Vectors 56
- 2.8 Force Vector Directed Along a Line 59
- 2.9 Dot Product 69



### 3 Equilibrium of a Particle 87

Chapter Objectives 87

- 3.1 Condition for the Equilibrium of a Particle 87
- 3.2 The Free-Body Diagram 88
- 3.3 Coplanar Force Systems 91
- 3.4 Three-Dimensional Force Systems 106



### 4 Force System Resultants 121

Chapter Objectives 121

- 4.1 Moment of a Force—Scalar Formulation 121
- 4.2 Cross Product 125
- 4.3 Moment of a Force—Vector Formulation 128
- 4.4 Principle of Moments 132
- 4.5 Moment of a Force about a Specified Axis 145
- 4.6 Moment of a Couple 154
- 4.7 Simplification of a Force and Couple System 166
- 4.8 Further Simplification of a Force and Couple System 177
- 4.9 Reduction of a Simple Distributed Loading 190



## 5 Equilibrium of a Rigid Body 207

Chapter Objectives 207

- 5.1 Conditions for Rigid-Body Equilibrium 207
- 5.2 Free-Body Diagrams 209
- 5.3 Equations of Equilibrium 220
- 5.4 Two- and Three-Force Members 230
- 5.5 Free-Body Diagrams 245
- 5.6 Equations of Equilibrium 250
- 5.7 Constraints and Statical Determinacy 251



## 6 Structural Analysis 273

Chapter Objectives 273

- 6.1 Simple Trusses 273
- 6.2 The Method of Joints 276
- 6.3 Zero-Force Members 282
- 6.4 The Method of Sections 291
- 6.5 Space Trusses 301
- 6.6 Frames and Machines 305



## 7

### Internal Forces 343

Chapter Objectives 343

- 7.1 Internal Loadings Developed in Structural Members 343
- 7.2 Shear and Moment Equations and Diagrams 361
- 7.3 Relations between Distributed Load, Shear, and Moment 370
- 7.4 Cables 381

## 8

### Friction 401

Chapter Objectives 401

- 8.1 Characteristics of Dry Friction 401
- 8.2 Problems Involving Dry Friction 406
- 8.3 Wedges 430
- 8.4 Frictional Forces on Screws 432
- 8.5 Frictional Forces on Flat Belts 439
- 8.6 Frictional Forces on Collar Bearings, Pivot Bearings, and Disks 447
- 8.7 Frictional Forces on Journal Bearings 450
- 8.8 Rolling Resistance 452



## 9 Center of Gravity and Centroid 465

Chapter Objectives 465

- 9.1 Center of Gravity, Center of Mass, and the Centroid of a Body 465
- 9.2 Composite Bodies 488
- 9.3 Theorems of Pappus and Guldinus 502
- 9.4 Resultant of a General Distributed Loading 511
- 9.5 Fluid Pressure 512



## 10 Moments of Inertia 529

Chapter Objectives 529

- 10.1 Definition of Moments of Inertia for Areas 529
- 10.2 Parallel-Axis Theorem for an Area 530
- 10.3 Radius of Gyration of an Area 531
- 10.4 Moments of Inertia for Composite Areas 540
- 10.5 Product of Inertia for an Area 548
- 10.6 Moments of Inertia for an Area about Inclined Axes 552
- 10.7 Mohr's Circle for Moments of Inertia 555
- 10.8 Mass Moment of Inertia 563





# 11

## Virtual Work 581

Chapter Objectives 581

11.1 Definition of Work 581

11.2 Principle of Virtual Work 583

11.3 Principle of Virtual Work for a System of Connected Rigid Bodies 585

11.4 Conservative Forces 597

11.5 Potential Energy 598

11.6 Potential-Energy Criterion for Equilibrium 600

11.7 Stability of Equilibrium Configuration 601

## Appendix

A. Mathematical Review and Expressions 616

**Fundamental Problems  
Partial Solutions And  
Answers 620**

**Preliminary Problems  
Statics Solutions 638**

**Review Problem  
Solutions 648**

**Answers to Selected  
Problems 658**

**Index 671**



# CREDITS

Chapter opening images are credited as follows:

**Chapter 1**, Andrew Peacock/Lonely Planet Images/Getty Images

**Chapter 2**, Vasiliy Koval/Fotolia

**Chapter 3**, Igor Tumarkin/ITPS/Shutterstock

**Chapter 4**, Rolf Adlercreutz/Alamy

**Chapter 5**, YuryZap/Shutterstock

**Chapter 6**, Tim Scrivener/Alamy

**Chapter 7**, Tony Freeman/Science Source

**Chapter 8**, Pavel Polkovnikov/Shutterstock

**Chapter 9**, Heather Reeder/Shutterstock

**Chapter 10**, Michael N. Paras/AGE Fotostock./Alamy

**Chapter 11**, John Kershaw/Alamy

*This page intentionally left blank*

ENGINEERING MECHANICS

# STATICS

FOURTEENTH EDITION

# Chapter 1



(© Andrew Peacock/Lonely Planet Images/Getty Images)

Large cranes such as this one are required to lift extremely large loads. Their design is based on the basic principles of statics and dynamics, which form the subject matter of engineering mechanics.

# General Principles

## CHAPTER OBJECTIVES

- To provide an introduction to the basic quantities and idealizations of mechanics.
- To give a statement of Newton's Laws of Motion and Gravitation.
- To review the principles for applying the SI system of units.
- To examine the standard procedures for performing numerical calculations.
- To present a general guide for solving problems.

---

## 1.1 Mechanics

**Mechanics** is a branch of the physical sciences that is concerned with the state of rest or motion of bodies that are subjected to the action of forces. In general, this subject can be subdivided into three branches: *rigid-body mechanics*, *deformable-body mechanics*, and *fluid mechanics*. In this book we will study rigid-body mechanics since it is a basic requirement for the study of the mechanics of deformable bodies and the mechanics of fluids. Furthermore, rigid-body mechanics is essential for the design and analysis of many types of structural members, mechanical components, or electrical devices encountered in engineering.

Rigid-body mechanics is divided into two areas: statics and dynamics. **Statics** deals with the equilibrium of bodies, that is, those that are either at rest or move with a constant velocity; whereas **dynamics** is concerned with the accelerated motion of bodies. We can consider statics as a special case of dynamics, in which the acceleration is zero; however, statics deserves separate treatment in engineering education since many objects are designed with the intention that they remain in equilibrium.

**Historical Development.** The subject of statics developed very early in history because its principles can be formulated simply from measurements of geometry and force. For example, the writings of Archimedes (287–212 B.C.) deal with the principle of the lever. Studies of the pulley, inclined plane, and wrench are also recorded in ancient writings—at times when the requirements for engineering were limited primarily to building construction.

Since the principles of dynamics depend on an accurate measurement of time, this subject developed much later. Galileo Galilei (1564–1642) was one of the first major contributors to this field. His work consisted of experiments using pendulums and falling bodies. The most significant contributions in dynamics, however, were made by Isaac Newton (1642–1727), who is noted for his formulation of the three fundamental laws of motion and the law of universal gravitational attraction. Shortly after these laws were postulated, important techniques for their application were developed by other scientists and engineers, some of whom will be mentioned throughout the text.

---

## 1.2 Fundamental Concepts

Before we begin our study of engineering mechanics, it is important to understand the meaning of certain fundamental concepts and principles.

**Basic Quantities.** The following four quantities are used throughout mechanics.

**Length.** *Length* is used to locate the position of a point in space and thereby describe the size of a physical system. Once a standard unit of length is defined, one can then use it to define distances and geometric properties of a body as multiples of this unit.

**Time.** *Time* is conceived as a succession of events. Although the principles of statics are time independent, this quantity plays an important role in the study of dynamics.

**Mass.** *Mass* is a measure of a quantity of matter that is used to compare the action of one body with that of another. This property manifests itself as a gravitational attraction between two bodies and provides a measure of the resistance of matter to a change in velocity.

**Force.** In general, *force* is considered as a “push” or “pull” exerted by one body on another. This interaction can occur when there is direct contact between the bodies, such as a person pushing on a wall, or it can occur through a distance when the bodies are physically separated. Examples of the latter type include gravitational, electrical, and magnetic forces. In any case, a force is completely characterized by its magnitude, direction, and point of application.

**Idealizations.** Models or idealizations are used in mechanics in order to simplify application of the theory. Here we will consider three important idealizations.

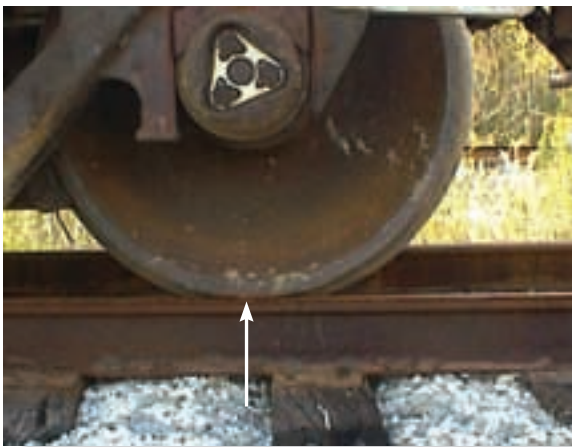
**Particle.** A *particle* has a mass, but a size that can be neglected. For example, the size of the earth is insignificant compared to the size of its orbit, and therefore the earth can be modeled as a particle when studying its orbital motion. When a body is idealized as a particle, the principles of mechanics reduce to a rather simplified form since the geometry of the body *will not be involved* in the analysis of the problem.

**Rigid Body.** A *rigid body* can be considered as a combination of a large number of particles in which all the particles remain at a fixed distance from one another, both before and after applying a load. This model is important because the body's shape does not change when a load is applied, and so we do not have to consider the type of material from which the body is made. In most cases the actual deformations occurring in structures, machines, mechanisms, and the like are relatively small, and the rigid-body assumption is suitable for analysis.

**Concentrated Force.** A *concentrated force* represents the effect of a loading which is assumed to act at a point on a body. We can represent a load by a concentrated force, provided the area over which the load is applied is very small compared to the overall size of the body. An example would be the contact force between a wheel and the ground.



Three forces act on the ring. Since these forces all meet at a point, then for any force analysis, we can assume the ring to be represented as a particle. (© Russell C. Hibbeler)

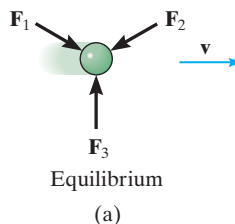


Steel is a common engineering material that does not deform very much under load. Therefore, we can consider this railroad wheel to be a rigid body acted upon by the concentrated force of the rail. (© Russell C. Hibbeler)



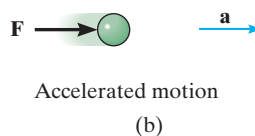
**Newton's Three Laws of Motion.** Engineering mechanics is formulated on the basis of Newton's three laws of motion, the validity of which is based on experimental observation. These laws apply to the motion of a particle as measured from a *nonaccelerating* reference frame. They may be briefly stated as follows.

**First Law.** A particle originally at rest, or moving in a straight line with constant velocity, tends to remain in this state provided the particle is *not* subjected to an unbalanced force, Fig. 1-1a.

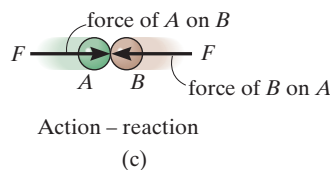


**Second Law.** A particle acted upon by an *unbalanced force*  $\mathbf{F}$  experiences an acceleration  $\mathbf{a}$  that has the same direction as the force and a magnitude that is directly proportional to the force, Fig. 1-1b.\* If  $\mathbf{F}$  is applied to a particle of mass  $m$ , this law may be expressed mathematically as

$$\mathbf{F} = m\mathbf{a} \quad (1-1)$$



**Third Law.** The mutual forces of action and reaction between two particles are equal, opposite, and collinear, Fig. 1-1c.



**Fig. 1-1**

\*Stated another way, the unbalanced force acting on the particle is proportional to the time rate of change of the particle's linear momentum.

**Newton's Law of Gravitational Attraction.** Shortly after formulating his three laws of motion, Newton postulated a law governing the gravitational attraction between any two particles. Stated mathematically,

$$F = G \frac{m_1 m_2}{r^2} \quad (1-2)$$

where

$F$  = force of gravitation between the two particles

$G$  = universal constant of gravitation; according to experimental evidence,  $G = 66.73(10^{-12}) \text{ m}^3/(\text{kg} \cdot \text{s}^2)$

$m_1, m_2$  = mass of each of the two particles

$r$  = distance between the two particles

**Weight.** According to Eq. 1-2, any two particles or bodies have a mutual attractive (gravitational) force acting between them. In the case of a particle located at or near the surface of the earth, however, the only gravitational force having any sizable magnitude is that between the earth and the particle. Consequently, this force, termed the **weight**, will be the only gravitational force considered in our study of mechanics.

From Eq. 1-2, we can develop an approximate expression for finding the weight  $W$  of a particle having a mass  $m_1 = m$ . If we assume the earth to be a nonrotating sphere of constant density and having a mass  $m_2 = M_e$ , then if  $r$  is the distance between the earth's center and the particle, we have

$$W = G \frac{m M_e}{r^2}$$

Letting  $g = GM_e/r^2$  yields

$$W = mg \quad (1-3)$$

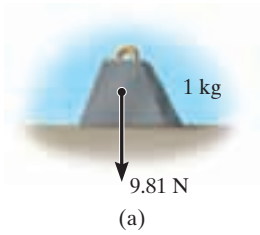
By comparison with  $\mathbf{F} = m\mathbf{a}$ , we can see that  $g$  is the acceleration due to gravity. Since it depends on  $r$ , then the weight of a body is *not* an absolute quantity. Instead, its magnitude is determined from where the measurement was made. For most engineering calculations, however,  $g$  is determined at sea level and at a latitude of  $45^\circ$ , which is considered the “standard location.”



The astronaut's weight is diminished since she is far removed from the gravitational field of the earth. (© NikoNomad/Shutterstock)

## 1.3 Units of Measurement

The four basic quantities—length, time, mass, and force—are not all independent from one another; in fact, they are *related* by Newton's second law of motion,  $\mathbf{F} = m\mathbf{a}$ . Because of this, the *units* used to measure these quantities cannot *all* be selected arbitrarily. The equality  $\mathbf{F} = m\mathbf{a}$  is maintained only if three of the four units, called **base units**, are *defined* and the fourth unit is then *derived* from the equation.

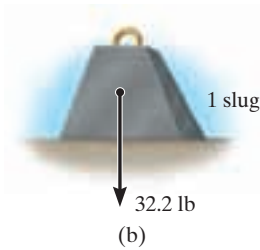


**SI Units.** The International System of units, abbreviated SI after the French “Système International d’Unités,” is a modern version of the metric system which has received worldwide recognition. As shown in Table 1–1, the SI system defines length in meters (m), time in seconds (s), and mass in kilograms (kg). The unit of force, called a **newton** (N), is *derived* from  $\mathbf{F} = m\mathbf{a}$ . Thus, 1 newton is equal to a force required to give 1 kilogram of mass an acceleration of  $1 \text{ m/s}^2$  ( $\text{N} = \text{kg} \cdot \text{m/s}^2$ ).

If the weight of a body located at the “standard location” is to be determined in newtons, then Eq. 1–3 must be applied. Here measurements give  $g = 9.806 \text{ 65 m/s}^2$ ; however, for calculations, the value  $g = 9.81 \text{ m/s}^2$  will be used. Thus,

$$W = mg \quad (g = 9.81 \text{ m/s}^2) \tag{1-4}$$

Therefore, a body of mass 1 kg has a weight of 9.81 N, a 2-kg body weighs 19.62 N, and so on, Fig. 1–2a.



**U.S. Customary.** In the U.S. Customary system of units (FPS) length is measured in feet (ft), time in seconds (s), and force in pounds (lb), Table 1–1. The unit of mass, called a **slug**, is *derived* from  $\mathbf{F} = m\mathbf{a}$ . Hence, 1 slug is equal to the amount of matter accelerated at  $1 \text{ ft/s}^2$  when acted upon by a force of 1 lb ( $\text{slug} = \text{lb} \cdot \text{s}^2/\text{ft}$ ).

Therefore, if the measurements are made at the “standard location,” where  $g = 32.2 \text{ ft/s}^2$ , then from Eq. 1–3,

$$m = \frac{W}{g} \quad (g = 32.2 \text{ ft/s}^2) \tag{1-5}$$

And so a body weighing 32.2 lb has a mass of 1 slug, a 64.4-lb body has a mass of 2 slugs, and so on, Fig. 1–2b.

Fig. 1–2

TABLE 1–1 Systems of Units				
Name	Length	Time	Mass	Force
International System of Units SI	meter	second	kilogram	newton*
	m	s	kg	$\frac{\text{N}}{\left(\frac{\text{kg} \cdot \text{m}}{\text{s}^2}\right)}$
U.S. Customary FPS	foot	second	slug*	pound
	ft	s	$\left(\frac{\text{lb} \cdot \text{s}^2}{\text{ft}}\right)$	lb
*Derived unit.				

**Conversion of Units.** Table 1–2 provides a set of direct conversion factors between FPS and SI units for the basic quantities. Also, in the FPS system, recall that 1 ft = 12 in. (inches), 5280 ft = 1 mi (mile), 1000 lb = 1 kip (kilo-pound), and 2000 lb = 1 ton.

**TABLE 1–2 Conversion Factors**

Quantity	Unit of Measurement (FPS)	Equals	Unit of Measurement (SI)
Force	lb		4.448 N
Mass	slug		14.59 kg
Length	ft		0.3048 m

## 1.4 The International System of Units

The SI system of units is used extensively in this book since it is intended to become the worldwide standard for measurement. Therefore, we will now present some of the rules for its use and some of its terminology relevant to engineering mechanics.

**Prefixes.** When a numerical quantity is either very large or very small, the units used to define its size may be modified by using a prefix. Some of the prefixes used in the SI system are shown in Table 1–3. Each represents a multiple or submultiple of a unit which, if applied successively, moves the decimal point of a numerical quantity to every third place.\* For example, 4 000 000 N = 4 000 kN (kilo-newton) = 4 MN (mega-newton), or 0.005 m = 5 mm (milli-meter). Notice that the SI system does not include the multiple deca (10) or the submultiple centi (0.01), which form part of the metric system. Except for some volume and area measurements, the use of these prefixes is to be avoided in science and engineering.

**TABLE 1–3 Prefixes**

	Exponential Form	Prefix	SI Symbol
<i>Multiple</i>			
1 000 000 000	$10^9$	giga	G
1 000 000	$10^6$	mega	M
1 000	$10^3$	kilo	k
<i>Submultiple</i>			
0.001	$10^{-3}$	milli	m
0.000 001	$10^{-6}$	micro	$\mu$
0.000 000 001	$10^{-9}$	nano	n

\*The kilogram is the only base unit that is defined with a prefix.

**Rules for Use.** Here are a few of the important rules that describe the proper use of the various SI symbols:

- Quantities defined by several units which are multiples of one another are separated by a *dot* to avoid confusion with prefix notation, as indicated by  $N = \text{kg} \cdot \text{m}/\text{s}^2 = \text{kg} \cdot \text{m} \cdot \text{s}^{-2}$ . Also,  $\text{m} \cdot \text{s}$  (meter-second), whereas  $\text{ms}$  (milli-second).
- The exponential power on a unit having a prefix refers to both the unit *and* its prefix. For example,  $\mu\text{N}^2 = (\mu\text{N})^2 = \mu\text{N} \cdot \mu\text{N}$ . Likewise,  $\text{mm}^2$  represents  $(\text{mm})^2 = \text{mm} \cdot \text{mm}$ .
- With the exception of the base unit the kilogram, in general avoid the use of a prefix in the denominator of composite units. For example, do not write  $\text{N}/\text{mm}$ , but rather  $\text{kN}/\text{m}$ ; also,  $\text{m}/\text{mg}$  should be written as  $\text{Mm}/\text{kg}$ .
- When performing calculations, represent the numbers in terms of their *base or derived units* by converting all prefixes to powers of 10. The final result should then be expressed using a *single prefix*. Also, after calculation, it is best to keep numerical values between 0.1 and 1000; otherwise, a suitable prefix should be chosen. For example,

$$\begin{aligned}(50 \text{ kN})(60 \text{ nm}) &= [50(10^3) \text{ N}][60(10^{-9}) \text{ m}] \\ &= 3000(10^{-6}) \text{ N} \cdot \text{m} = 3(10^{-3}) \text{ N} \cdot \text{m} = 3 \text{ mN} \cdot \text{m}\end{aligned}$$

## 1.5 Numerical Calculations



Computers are often used in engineering for advanced design and analysis. (© Blaise Pascall/Alamy)

Numerical work in engineering practice is most often performed by using handheld calculators and computers. It is important, however, that the answers to any problem be reported with justifiable accuracy using appropriate significant figures. In this section we will discuss these topics together with some other important aspects involved in all engineering calculations.

**Dimensional Homogeneity.** The terms of any equation used to describe a physical process must be **dimensionally homogeneous**; that is, each term must be expressed in the same units. Provided this is the case, all the terms of an equation can then be combined if numerical values are substituted for the variables. Consider, for example, the equation  $s = vt + \frac{1}{2}at^2$ , where, in SI units,  $s$  is the position in meters,  $m$ ,  $t$  is time in seconds,  $s$ ,  $v$  is velocity in  $\text{m}/\text{s}$  and  $a$  is acceleration in  $\text{m}/\text{s}^2$ . Regardless of how this equation is evaluated, it maintains its dimensional homogeneity. In the form stated, each of the three terms is expressed in meters  $[m, (\text{m}/\text{s})\text{s}, (\text{m}/\text{s}^2)\text{s}^2]$  or solving for  $a$ ,  $a = 2s/t^2 - 2v/t$ , the terms are each expressed in units of  $\text{m}/\text{s}^2$   $[\text{m}/\text{s}^2, \text{m}/\text{s}^2, (\text{m}/\text{s})/\text{s}]$ .

Keep in mind that problems in mechanics always involve the solution of dimensionally homogeneous equations, and so this fact can then be used as a partial check for algebraic manipulations of an equation.

**Significant Figures.** The number of significant figures contained in any number determines the accuracy of the number. For instance, the number 4981 contains four significant figures. However, if zeros occur at the end of a whole number, it may be unclear as to how many significant figures the number represents. For example, 23 400 might have three (234), four (2340), or five (23 400) significant figures. To avoid these ambiguities, we will use *engineering notation* to report a result. This requires that numbers be rounded off to the appropriate number of significant digits and then expressed in multiples of  $(10^3)$ , such as  $(10^3)$ ,  $(10^6)$ , or  $(10^{-9})$ . For instance, if 23 400 has five significant figures, it is written as  $23.400(10^3)$ , but if it has only three significant figures, it is written as  $23.4(10^3)$ .

If zeros occur at the beginning of a number that is less than one, then the zeros are not significant. For example, 0.008 21 has three significant figures. Using engineering notation, this number is expressed as  $8.21(10^{-3})$ . Likewise, 0.000 582 can be expressed as  $0.582(10^{-3})$  or  $582(10^{-6})$ .

**Rounding Off Numbers.** Rounding off a number is necessary so that the accuracy of the result will be the same as that of the problem data. As a general rule, any numerical figure ending in a number greater than five is rounded up and a number less than five is not rounded up. The rules for rounding off numbers are best illustrated by examples. Suppose the number 3.5587 is to be rounded off to *three* significant figures. Because the fourth digit (8) is *greater than 5*, the third number is rounded up to 3.56. Likewise 0.5896 becomes 0.590 and 9.3866 becomes 9.39. If we round off 1.341 to three significant figures, because the fourth digit (1) is *less than 5*, then we get 1.34. Likewise 0.3762 becomes 0.376 and 9.871 becomes 9.87. There is a special case for any number that ends in a 5. As a general rule, if the digit preceding the 5 is an *even number*, then this digit is *not* rounded up. If the digit preceding the 5 is an *odd number*, then it is rounded up. For example, 75.25 rounded off to three significant digits becomes 75.2, 0.1275 becomes 0.128, and 0.2555 becomes 0.256.

**Calculations.** When a sequence of calculations is performed, it is best to store the intermediate results in the calculator. In other words, do not round off calculations until expressing the final result. This procedure maintains precision throughout the series of steps to the final solution. In this text we will generally round off the answers to three significant figures since most of the data in engineering mechanics, such as geometry and loads, may be reliably measured to this accuracy.



When solving problems, do the work as neatly as possible. Being neat will stimulate clear and orderly thinking, and vice versa. (© Russell C. Hibbeler)

## 1.6 General Procedure for Analysis

Attending a lecture, reading this book, and studying the example problems helps, but **the most effective way of learning the principles of engineering mechanics is to solve problems**. To be successful at this, it is important to always present the work in a *logical* and *orderly manner*, as suggested by the following sequence of steps:

- Read the problem carefully and try to correlate the actual physical situation with the theory studied.
- Tabulate the problem data and *draw to a large scale* any necessary diagrams.
- Apply the relevant principles, generally in mathematical form. When writing any equations, be sure they are dimensionally homogeneous.
- Solve the necessary equations, and report the answer with no more than three significant figures.
- Study the answer with technical judgment and common sense to determine whether or not it seems reasonable.

### Important Points

- Statics is the study of bodies that are at rest or move with constant velocity.
- A particle has a mass but a size that can be neglected, and a rigid body does not deform under load.
- A force is considered as a “push” or “pull” of one body on another.
- Concentrated forces are assumed to act at a point on a body.
- Newton’s three laws of motion should be memorized.
- Mass is measure of a quantity of matter that does not change from one location to another. Weight refers to the gravitational attraction of the earth on a body or quantity of mass. Its magnitude depends upon the elevation at which the mass is located.
- In the SI system the unit of force, the newton, is a derived unit. The meter, second, and kilogram are base units.
- Prefixes G, M, k, m,  $\mu$ , and n are used to represent large and small numerical quantities. Their exponential size should be known, along with the rules for using the SI units.
- Perform numerical calculations with several significant figures, and then report the final answer to three significant figures.
- Algebraic manipulations of an equation can be checked in part by verifying that the equation remains dimensionally homogeneous.
- Know the rules for rounding off numbers.

**EXAMPLE 1.1**

Convert 2 km/h to m/s How many ft/s is this?

**SOLUTION**

Since 1 km = 1000 m and 1 h = 3600 s, the factors of conversion are arranged in the following order, so that a cancellation of the units can be applied:

$$\begin{aligned} 2 \text{ km/h} &= \frac{2 \cancel{\text{km}}}{\cancel{\text{h}}} \left( \frac{1000 \text{ m}}{\cancel{\text{km}}} \right) \left( \frac{1 \cancel{\text{h}}}{3600 \text{ s}} \right) \\ &= \frac{2000 \text{ m}}{3600 \text{ s}} = 0.556 \text{ m/s} \end{aligned} \quad \text{Ans.}$$

From Table 1-2, 1 ft = 0.3048 m. Thus,

$$\begin{aligned} 0.556 \text{ m/s} &= \left( \frac{0.556 \cancel{\text{m}}}{\text{s}} \right) \left( \frac{1 \text{ ft}}{0.3048 \cancel{\text{m}}} \right) \\ &= 1.82 \text{ ft/s} \end{aligned} \quad \text{Ans.}$$

**NOTE:** Remember to round off the final answer to three significant figures.

**EXAMPLE 1.2**

Convert the quantities 300 lb · s and 52 slug/ft<sup>3</sup> to appropriate SI units.

**SOLUTION**

Using Table 1-2, 1 lb = 4.448 N.

$$\begin{aligned} 300 \text{ lb} \cdot \text{s} &= 300 \cancel{\text{lb}} \cdot \text{s} \left( \frac{4.448 \text{ N}}{1 \cancel{\text{lb}}} \right) \\ &= 1334.5 \text{ N} \cdot \text{s} = 1.33 \text{ kN} \cdot \text{s} \end{aligned} \quad \text{Ans.}$$

Since 1 slug = 14.59 kg and 1 ft = 0.3048 m, then

$$\begin{aligned} 52 \text{ slug/ft}^3 &= \frac{52 \cancel{\text{slug}}}{\cancel{\text{ft}}^3} \left( \frac{14.59 \text{ kg}}{1 \cancel{\text{slug}}} \right) \left( \frac{1 \cancel{\text{ft}}}{0.3048 \text{ m}} \right)^3 \\ &= 26.8(10^3) \text{ kg/m}^3 \\ &= 26.8 \text{ Mg/m}^3 \end{aligned} \quad \text{Ans.}$$



**EXAMPLE 1.3**

Evaluate each of the following and express with SI units having an appropriate prefix: (a)  $(50 \text{ mN})(6 \text{ GN})$ , (b)  $(400 \text{ mm})(0.6 \text{ MN})^2$ , (c)  $45 \text{ MN}^3/900 \text{ Gg}$ .

**SOLUTION**

First convert each number to base units, perform the indicated operations, then choose an appropriate prefix.

**Part (a)**

$$\begin{aligned}
 (50 \text{ mN})(6 \text{ GN}) &= [50(10^{-3}) \text{ N}][6(10^9) \text{ N}] \\
 &= 300(10^6) \text{ N}^2 \\
 &= 300(10^6) \text{ N}^2 \left( \frac{1 \text{ kN}}{10^3 \text{ N}} \right) \left( \frac{1 \text{ kN}}{10^3 \text{ N}} \right) \\
 &= 300 \text{ kN}^2
 \end{aligned}$$

*Ans.*

**NOTE:** Keep in mind the convention  $\text{kN}^2 = (\text{kN})^2 = 10^6 \text{ N}^2$ .

**Part (b)**

$$\begin{aligned}
 (400 \text{ mm})(0.6 \text{ MN})^2 &= [400(10^{-3}) \text{ m}][0.6(10^6) \text{ N}]^2 \\
 &= [400(10^{-3}) \text{ m}][0.36(10^{12}) \text{ N}^2] \\
 &= 144(10^9) \text{ m} \cdot \text{N}^2 \\
 &= 144 \text{ Gm} \cdot \text{N}^2
 \end{aligned}$$

*Ans.*

We can also write

$$\begin{aligned}
 144(10^9) \text{ m} \cdot \text{N}^2 &= 144(10^9) \text{ m} \cdot \text{N}^2 \left( \frac{1 \text{ MN}}{10^6 \text{ N}} \right) \left( \frac{1 \text{ MN}}{10^6 \text{ N}} \right) \\
 &= 0.144 \text{ m} \cdot \text{MN}^2
 \end{aligned}$$

*Ans.*

**Part (c)**

$$\begin{aligned}
 \frac{45 \text{ MN}^3}{900 \text{ Gg}} &= \frac{45(10^6 \text{ N})^3}{900(10^6) \text{ kg}} \\
 &= 50(10^9) \text{ N}^3/\text{kg} \\
 &= 50(10^9) \text{ N}^3 \left( \frac{1 \text{ kN}}{10^3 \text{ N}} \right)^3 \frac{1}{\text{kg}} \\
 &= 50 \text{ kN}^3/\text{kg}
 \end{aligned}$$

*Ans.*

# PROBLEMS

1

*The answers to all but every fourth problem (asterisk) are given in the back of the book.*

**1-1.** What is the weight in newtons of an object that has a mass of (a) 8 kg, (b) 0.04 kg, and (c) 760 Mg?

**1-2.** Represent each of the following combinations of units in the correct SI form: (a) kN/ $\mu$ s, (b) Mg/mN, and (c) MN/(kg  $\cdot$  ms).

**1-3.** Represent each of the following combinations of units in the correct SI form: (a) Mg/ms, (b) N/mm, (c) mN/(kg  $\cdot$   $\mu$ s).

**\*1-4.** Convert: (a) 200 lb  $\cdot$  ft to N  $\cdot$  m, (b) 350 lb/ft<sup>3</sup> to kN/m<sup>3</sup>, (c) 8 ft/h to mm/s. Express the result to three significant figures. Use an appropriate prefix.

**1-5.** Represent each of the following as a number between 0.1 and 1000 using an appropriate prefix: (a) 45 320 kN, (b) 568(10<sup>5</sup>) mm, and (c) 0.00563 mg.

**1-6.** Round off the following numbers to three significant figures: (a) 58 342 m, (b) 68.534 s, (c) 2553 N, and (d) 7555 kg.

**1-7.** Represent each of the following quantities in the correct SI form using an appropriate prefix: (a) 0.000 431 kg, (b) 35.3(10<sup>3</sup>) N, (c) 0.005 32 km.

**\*1-8.** Represent each of the following combinations of units in the correct SI form using an appropriate prefix: (a) Mg/mm, (b) mN/ $\mu$ s, (c)  $\mu$ m  $\cdot$  Mg.

**1-9.** Represent each of the following combinations of units in the correct SI form using an appropriate prefix: (a) m/ms, (b)  $\mu$ km, (c) ks/mg, and (d) km  $\cdot$   $\mu$ N.

**1-10.** Represent each of the following combinations of units in the correct SI form: (a) GN  $\cdot$   $\mu$ m, (b) kg/ $\mu$ m, (c) N/ks<sup>2</sup>, and (d) kN/ $\mu$ s.

**1-11.** Represent each of the following with SI units having an appropriate prefix: (a) 8653 ms, (b) 8368 N, (c) 0.893 kg.

**\*1-12.** Evaluate each of the following to three significant figures and express each answer in SI units using an appropriate prefix: (a) (684  $\mu$ m)/(43 ms), (b) (28 ms)(0.0458 Mm)/(348 mg), (c) (2.68 mm)(426 Mg).

**1-13.** The density (mass/volume) of aluminum is 5.26 slug/ft<sup>3</sup>. Determine its density in SI units. Use an appropriate prefix.

**1-14.** Evaluate each of the following to three significant figures and express each answer in SI units using an appropriate prefix: (a) (212 mN)<sup>2</sup>, (b) (52 800 ms)<sup>2</sup>, and (c) [548(10<sup>6</sup>)]<sup>1/2</sup> ms.

**1-15.** Using the SI system of units, show that Eq. 1-2 is a dimensionally homogeneous equation which gives  $F$  in newtons. Determine to three significant figures the gravitational force acting between two spheres that are touching each other. The mass of each sphere is 200 kg and the radius is 300 mm.

**\*1-16.** The *pascal* (Pa) is actually a very small unit of pressure. To show this, convert 1 Pa = 1 N/m<sup>2</sup> to lb/ft<sup>2</sup>. Atmosphere pressure at sea level is 14.7 lb/in<sup>2</sup>. How many pascals is this?

**1-17.** Water has a density of 1.94 slug/ft<sup>3</sup>. What is the density expressed in SI units? Express the answer to three significant figures.

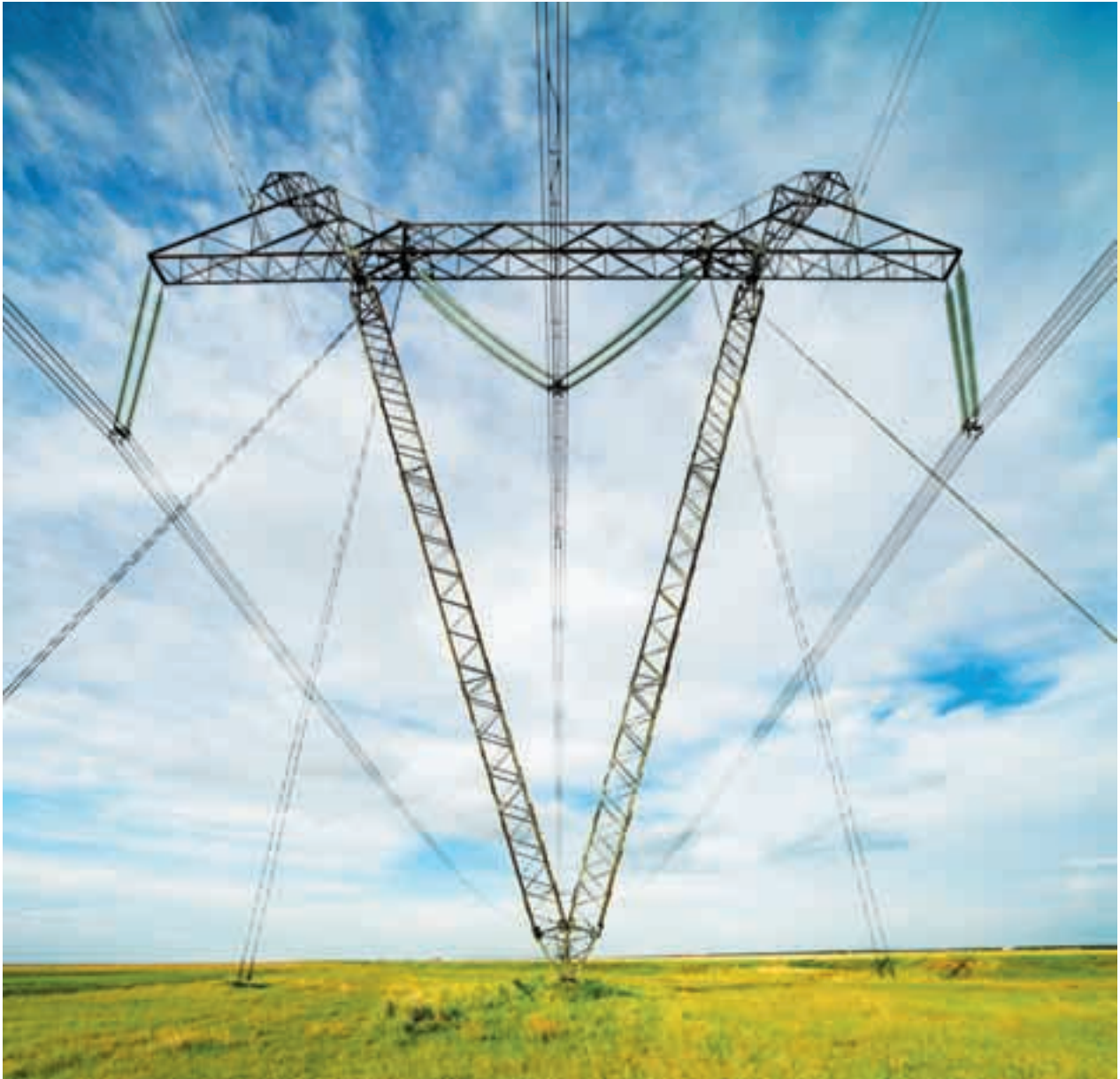
**1-18.** Evaluate each of the following to three significant figures and express each answer in SI units using an appropriate prefix: (a) 354 mg(45 km)/(0.0356 kN), (b) (0.004 53 Mg)(201 ms), (c) 435 MN/23.2 mm.

**1-19.** A concrete column has a diameter of 350 mm and a length of 2 m. If the density (mass/volume) of concrete is 2.45 Mg/m<sup>3</sup>, determine the weight of the column in pounds.

**\*1-20.** If a man weighs 155 lb on earth, specify (a) his mass in slugs, (b) his mass in kilograms, and (c) his weight in newtons. If the man is on the moon, where the acceleration due to gravity is  $g_m = 5.30 \text{ ft/s}^2$ , determine (d) his weight in pounds, and (e) his mass in kilograms.

**1-21.** Two particles have a mass of 8 kg and 12 kg, respectively. If they are 800 mm apart, determine the force of gravity acting between them. Compare this result with the weight of each particle.

# Chapter 2



(© Vasiliy Koval/Fotolia)

This electric transmission tower is stabilized by cables that exert forces on the tower at their points of connection. In this chapter we will show how to express these forces as Cartesian vectors, and then determine their resultant.

# Force Vectors

## CHAPTER OBJECTIVES

- To show how to add forces and resolve them into components using the Parallelogram Law.
- To express force and position in Cartesian vector form and explain how to determine the vector's magnitude and direction.
- To introduce the dot product in order to use it to find the angle between two vectors or the projection of one vector onto another.

## 2.1 Scalars and Vectors

Many physical quantities in engineering mechanics are measured using either scalars or vectors.

**Scalar.** A *scalar* is any positive or negative physical quantity that can be completely specified by its *magnitude*. Examples of scalar quantities include length, mass, and time.

**Vector.** A *vector* is any physical quantity that requires both a *magnitude* and a *direction* for its complete description. Examples of vectors encountered in statics are force, position, and moment. A vector is shown graphically by an arrow. The length of the arrow represents the *magnitude* of the vector, and the angle  $\theta$  between the vector and a fixed axis defines the *direction of its line of action*. The head or tip of the arrow indicates the *sense of direction* of the vector, Fig. 2–1.

In print, vector quantities are represented by boldface letters such as **A**, and the magnitude of a vector is italicized, *A*. For handwritten work, it is often convenient to denote a vector quantity by simply drawing an arrow above it,  $\vec{A}$ .

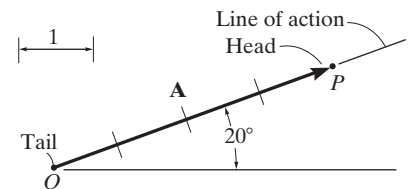
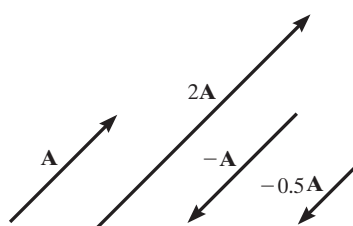


Fig. 2–1

## 2.2 Vector Operations

**Multiplication and Division of a Vector by a Scalar.** If a vector is multiplied by a positive scalar, its magnitude is increased by that amount. Multiplying by a negative scalar will also change the directional sense of the vector. Graphic examples of these operations are shown in Fig. 2-2.



Scalar multiplication and division

Fig. 2-2

**Vector Addition.** When adding two vectors together it is important to account for both their magnitudes and their directions. To do this we must use the *parallelogram law of addition*. To illustrate, the two *component vectors*  $\mathbf{A}$  and  $\mathbf{B}$  in Fig. 2-3a are added to form a *resultant vector*  $\mathbf{R} = \mathbf{A} + \mathbf{B}$  using the following procedure:

- First join the tails of the components at a point to make them concurrent, Fig. 2-3b.
- From the head of  $\mathbf{B}$ , draw a line parallel to  $\mathbf{A}$ . Draw another line from the head of  $\mathbf{A}$  that is parallel to  $\mathbf{B}$ . These two lines intersect at point  $P$  to form the adjacent sides of a parallelogram.
- The diagonal of this parallelogram that extends to  $P$  forms  $\mathbf{R}$ , which then represents the resultant vector  $\mathbf{R} = \mathbf{A} + \mathbf{B}$ , Fig. 2-3c.

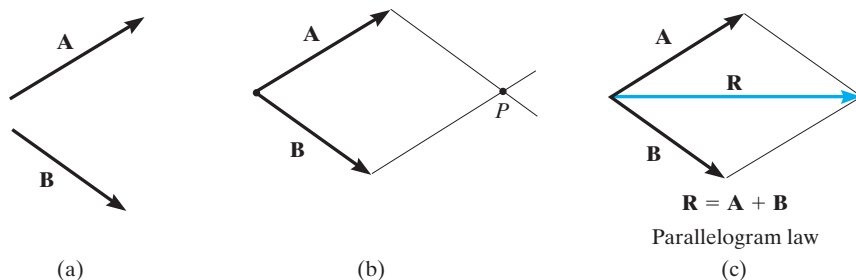


Fig. 2-3

We can also add  $\mathbf{B}$  to  $\mathbf{A}$ , Fig. 2-4a, using the *triangle rule*, which is a special case of the parallelogram law, whereby vector  $\mathbf{B}$  is added to vector  $\mathbf{A}$  in a “head-to-tail” fashion, i.e., by connecting the head of  $\mathbf{A}$  to the tail of  $\mathbf{B}$ , Fig. 2-4b. The resultant  $\mathbf{R}$  extends from the tail of  $\mathbf{A}$  to the head of  $\mathbf{B}$ . In a similar manner,  $\mathbf{R}$  can also be obtained by adding  $\mathbf{A}$  to  $\mathbf{B}$ , Fig. 2-4c. By comparison, it is seen that vector addition is commutative; in other words, the vectors can be added in either order, i.e.,  $\mathbf{R} = \mathbf{A} + \mathbf{B} = \mathbf{B} + \mathbf{A}$ .

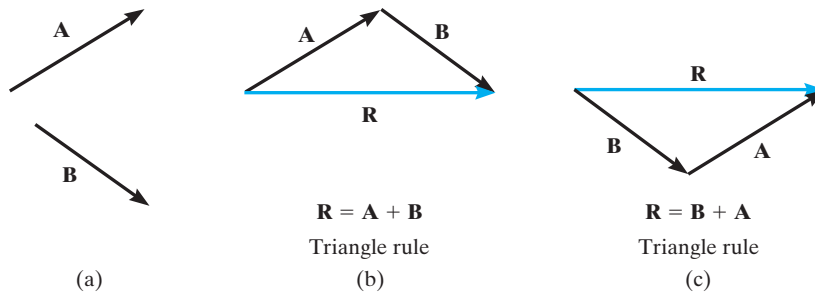
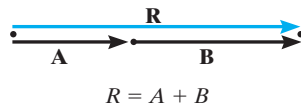


Fig. 2-4

As a special case, if the two vectors  $\mathbf{A}$  and  $\mathbf{B}$  are *collinear*, i.e., both have the same line of action, the parallelogram law reduces to an *algebraic* or *scalar addition*  $R = A + B$ , as shown in Fig. 2-5.



Addition of collinear vectors

Fig. 2-5

**Vector Subtraction.** The resultant of the *difference* between two vectors  $\mathbf{A}$  and  $\mathbf{B}$  of the same type may be expressed as

$$\mathbf{R}' = \mathbf{A} - \mathbf{B} = \mathbf{A} + (-\mathbf{B})$$

This vector sum is shown graphically in Fig. 2-6. Subtraction is therefore defined as a special case of addition, so the rules of vector addition also apply to vector subtraction.

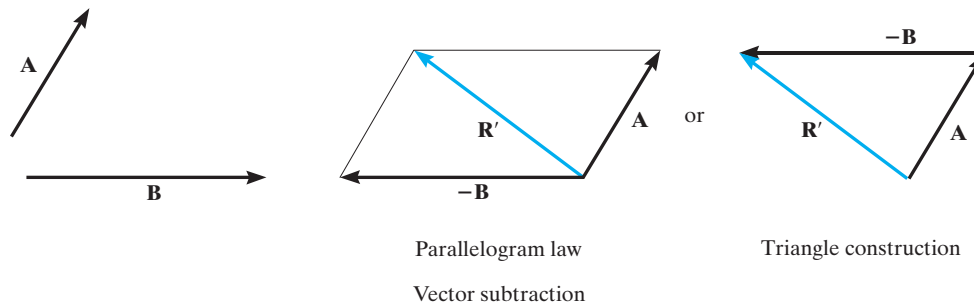
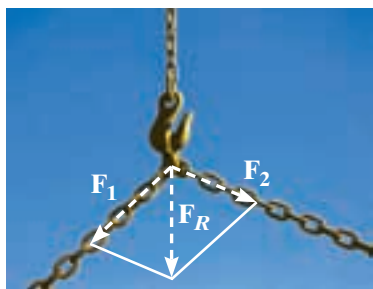


Fig. 2-6

## 2.3 Vector Addition of Forces



The parallelogram law must be used to determine the resultant of the two forces acting on the hook. (© Russell C. Hibbeler)

Experimental evidence has shown that a force is a vector quantity since it has a specified magnitude, direction, and sense and it adds according to the parallelogram law. Two common problems in statics involve either finding the resultant force, knowing its components, or resolving a known force into two components. We will now describe how each of these problems is solved using the parallelogram law.

**Finding a Resultant Force.** The two component forces  $\mathbf{F}_1$  and  $\mathbf{F}_2$  acting on the pin in Fig. 2-7a can be added together to form the resultant force  $\mathbf{F}_R = \mathbf{F}_1 + \mathbf{F}_2$ , as shown in Fig. 2-7b. From this construction, or using the triangle rule, Fig. 2-7c, we can apply the law of cosines or the law of sines to the triangle in order to obtain the magnitude of the resultant force and its direction.

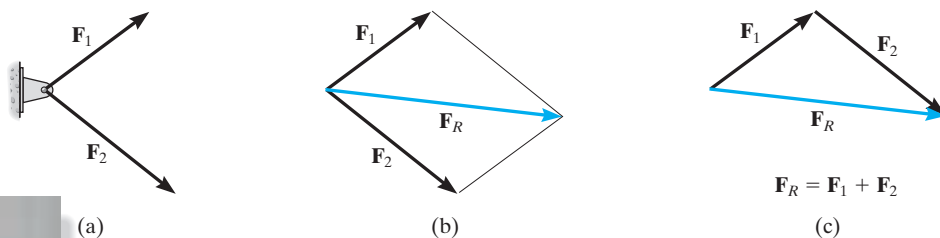


Fig. 2-7



Using the parallelogram law the supporting force  $\mathbf{F}$  can be resolved into components acting along the  $u$  and  $v$  axes. (© Russell C. Hibbeler)

**Finding the Components of a Force.** Sometimes it is necessary to resolve a force into two *components* in order to study its pulling or pushing effect in two specific directions. For example, in Fig. 2-8a,  $\mathbf{F}$  is to be resolved into two components along the two members, defined by the  $u$  and  $v$  axes. In order to determine the magnitude of each component, a parallelogram is constructed first, by drawing lines starting from the tip of  $\mathbf{F}$ , one line parallel to  $u$ , and the other line parallel to  $v$ . These lines then intersect with the  $v$  and  $u$  axes, forming a parallelogram. The force components  $\mathbf{F}_u$  and  $\mathbf{F}_v$  are then established by simply joining the tail of  $\mathbf{F}$  to the intersection points on the  $u$  and  $v$  axes, Fig. 2-8b. This parallelogram can then be reduced to a triangle, which represents the triangle rule, Fig. 2-8c. From this, the law of sines can then be applied to determine the unknown magnitudes of the components.



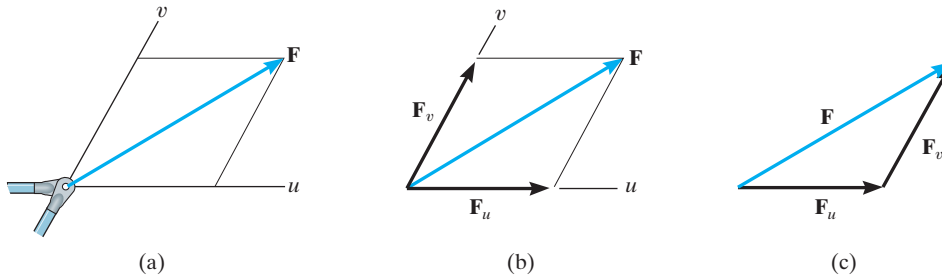


Fig. 2-8

**Addition of Several Forces.** If more than two forces are to be added, successive applications of the parallelogram law can be carried out in order to obtain the resultant force. For example, if three forces  $\mathbf{F}_1$ ,  $\mathbf{F}_2$ ,  $\mathbf{F}_3$  act at a point  $O$ , Fig. 2-9, the resultant of any two of the forces is found, say,  $\mathbf{F}_1 + \mathbf{F}_2$ —and then this resultant is added to the third force, yielding the resultant of all three forces; i.e.,  $\mathbf{F}_R = (\mathbf{F}_1 + \mathbf{F}_2) + \mathbf{F}_3$ . Using the parallelogram law to add more than two forces, as shown here, often requires extensive geometric and trigonometric calculation to determine the numerical values for the magnitude and direction of the resultant. Instead, problems of this type are easily solved by using the “rectangular-component method,” which is explained in Sec. 2.4.

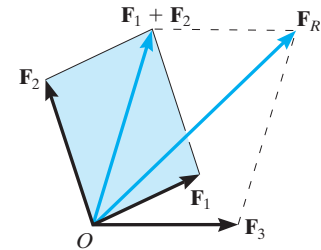
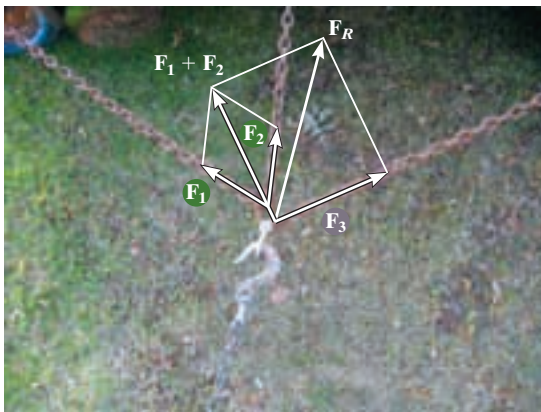


Fig. 2-9

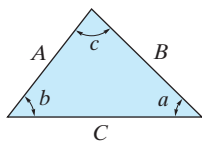
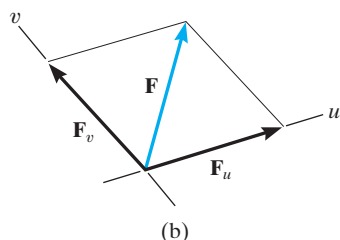
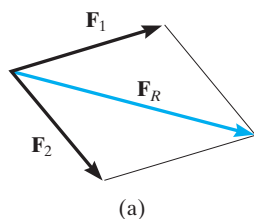


The resultant force  $\mathbf{F}_R$  on the hook requires the addition of  $\mathbf{F}_1 + \mathbf{F}_2$ , then this resultant is added to  $\mathbf{F}_3$ . (© Russell C. Hibbeler)



## Important Points

- A scalar is a positive or negative number.
- A vector is a quantity that has a magnitude, direction, and sense.
- Multiplication or division of a vector by a scalar will change the magnitude of the vector. The sense of the vector will change if the scalar is negative.
- As a special case, if the vectors are collinear, the resultant is formed by an algebraic or scalar addition.



Cosine law:

$$C = \sqrt{A^2 + B^2 - 2AB \cos c}$$

Sine law:

$$\frac{A}{\sin a} = \frac{B}{\sin b} = \frac{C}{\sin c}$$

**Fig. 2–10**

## Procedure for Analysis

Problems that involve the addition of two forces can be solved as follows:

### Parallelogram Law.

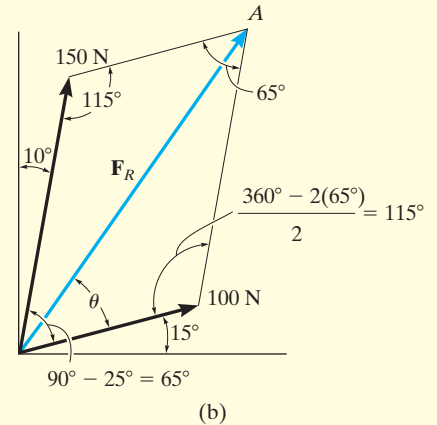
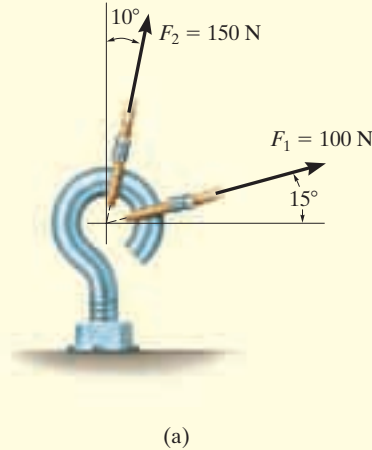
- Two “component” forces  $\mathbf{F}_1$  and  $\mathbf{F}_2$  in Fig. 2–10a add according to the parallelogram law, yielding a *resultant* force  $\mathbf{F}_R$  that forms the diagonal of the parallelogram.
- If a force  $\mathbf{F}$  is to be resolved into *components* along two axes  $u$  and  $v$ , Fig. 2–10b, then start at the head of force  $\mathbf{F}$  and construct lines parallel to the axes, thereby forming the parallelogram. The sides of the parallelogram represent the components,  $\mathbf{F}_u$  and  $\mathbf{F}_v$ .
- Label all the known and unknown force magnitudes and the angles on the sketch and identify the two unknowns as the magnitude and direction of  $\mathbf{F}_R$ , or the magnitudes of its components.

### Trigonometry.

- Redraw a half portion of the parallelogram to illustrate the triangular head-to-tail addition of the components.
- From this triangle, the magnitude of the resultant force can be determined using the law of cosines, and its direction is determined from the law of sines. The magnitudes of two force components are determined from the law of sines. The formulas are given in Fig. 2–10c.

**EXAMPLE 2.1**

The screw eye in Fig. 2–11a is subjected to two forces,  $\mathbf{F}_1$  and  $\mathbf{F}_2$ . Determine the magnitude and direction of the resultant force.

**SOLUTION**

**Parallelogram Law.** The parallelogram is formed by drawing a line from the head of  $\mathbf{F}_1$  that is parallel to  $\mathbf{F}_2$ , and another line from the head of  $\mathbf{F}_2$  that is parallel to  $\mathbf{F}_1$ . The resultant force  $\mathbf{F}_R$  extends to where these lines intersect at point A, Fig. 2–11b. The two unknowns are the magnitude of  $\mathbf{F}_R$  and the angle  $\theta$  (theta).

**Trigonometry.** From the parallelogram, the vector triangle is constructed, Fig. 2–11c. Using the law of cosines

$$\begin{aligned} F_R &= \sqrt{(100 \text{ N})^2 + (150 \text{ N})^2 - 2(100 \text{ N})(150 \text{ N}) \cos 115^\circ} \\ &= \sqrt{10\,000 + 22\,500 - 30\,000(-0.4226)} = 212.6 \text{ N} \\ &= 213 \text{ N} \end{aligned} \quad \text{Ans.}$$

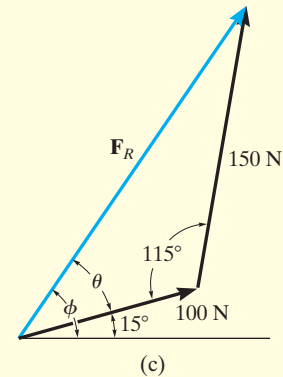
Applying the law of sines to determine  $\theta$ ,

$$\begin{aligned} \frac{150 \text{ N}}{\sin \theta} &= \frac{212.6 \text{ N}}{\sin 115^\circ} & \sin \theta &= \frac{150 \text{ N}}{212.6 \text{ N}} (\sin 115^\circ) \\ & & \theta &= 39.8^\circ \end{aligned}$$

Thus, the direction  $\phi$  (phi) of  $\mathbf{F}_R$ , measured from the horizontal, is

$$\phi = 39.8^\circ + 15.0^\circ = 54.8^\circ \quad \text{Ans.}$$

**NOTE:** The results seem reasonable, since Fig. 2–11b shows  $\mathbf{F}_R$  to have a magnitude larger than its components and a direction that is between them.



**Fig. 2–11**

## EXAMPLE 2.2

Resolve the horizontal 600-lb force in Fig. 2–12a into components acting along the  $u$  and  $v$  axes and determine the magnitudes of these components.

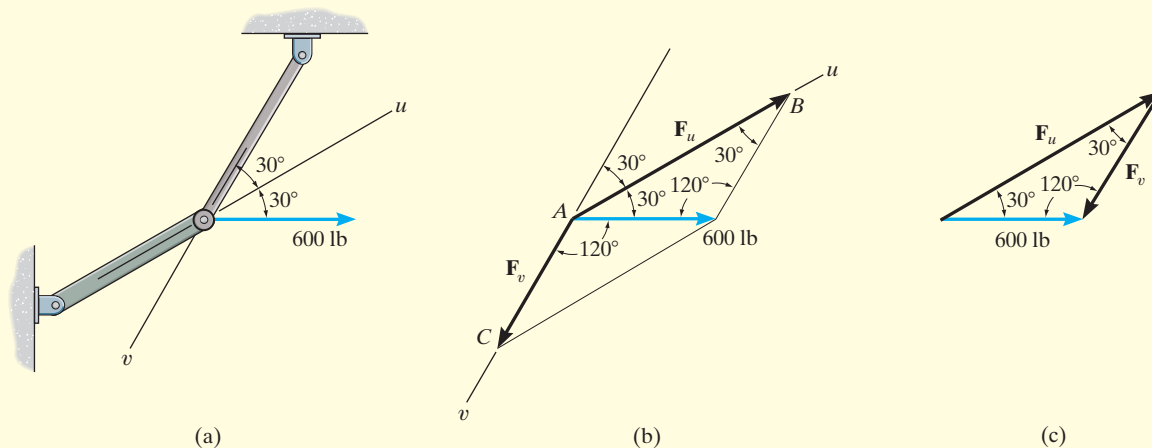


Fig. 2–12

## SOLUTION

The parallelogram is constructed by extending a line from the *head* of the 600-lb force parallel to the  $v$  axis until it intersects the  $u$  axis at point  $B$ , Fig. 2–12b. The arrow from  $A$  to  $B$  represents  $F_u$ . Similarly, the line extended from the head of the 600-lb force drawn parallel to the  $u$  axis intersects the  $v$  axis at point  $C$ , which gives  $F_v$ .

The vector addition using the triangle rule is shown in Fig. 2–12c. The two unknowns are the magnitudes of  $F_u$  and  $F_v$ . Applying the law of sines,

$$\frac{F_u}{\sin 120^\circ} = \frac{600 \text{ lb}}{\sin 30^\circ}$$

$$F_u = 1039 \text{ lb} \quad \text{Ans.}$$

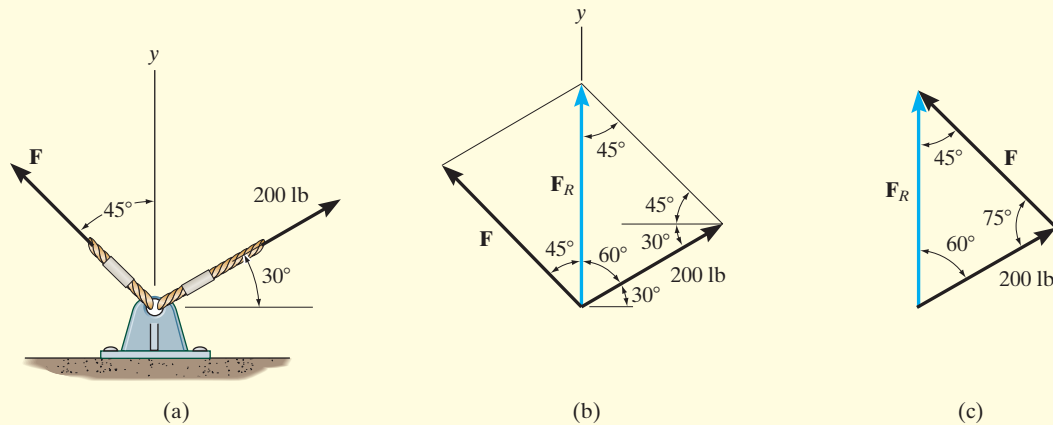
$$\frac{F_v}{\sin 30^\circ} = \frac{600 \text{ lb}}{\sin 30^\circ}$$

$$F_v = 600 \text{ lb} \quad \text{Ans.}$$

**NOTE:** The result for  $F_u$  shows that sometimes a component can have a greater magnitude than the resultant.

**EXAMPLE 2.3**

Determine the magnitude of the component force  $\mathbf{F}$  in Fig. 2-13a and the magnitude of the resultant force  $\mathbf{F}_R$  if  $\mathbf{F}_R$  is directed along the positive y axis.

**Fig. 2-13****SOLUTION**

The parallelogram law of addition is shown in Fig. 2-13b, and the triangle rule is shown in Fig. 2-13c. The magnitudes of  $\mathbf{F}_R$  and  $\mathbf{F}$  are the two unknowns. They can be determined by applying the law of sines.

$$\frac{F}{\sin 60^\circ} = \frac{200 \text{ lb}}{\sin 45^\circ}$$

$$F = 245 \text{ lb} \quad \text{Ans.}$$

$$\frac{F_R}{\sin 75^\circ} = \frac{200 \text{ lb}}{\sin 45^\circ}$$

$$F_R = 273 \text{ lb} \quad \text{Ans.}$$

## EXAMPLE 2.4

It is required that the resultant force acting on the eyebolt in Fig. 2–14a be directed along the positive  $x$  axis and that  $\mathbf{F}_2$  have a *minimum* magnitude. Determine this magnitude, the angle  $\theta$ , and the corresponding resultant force.

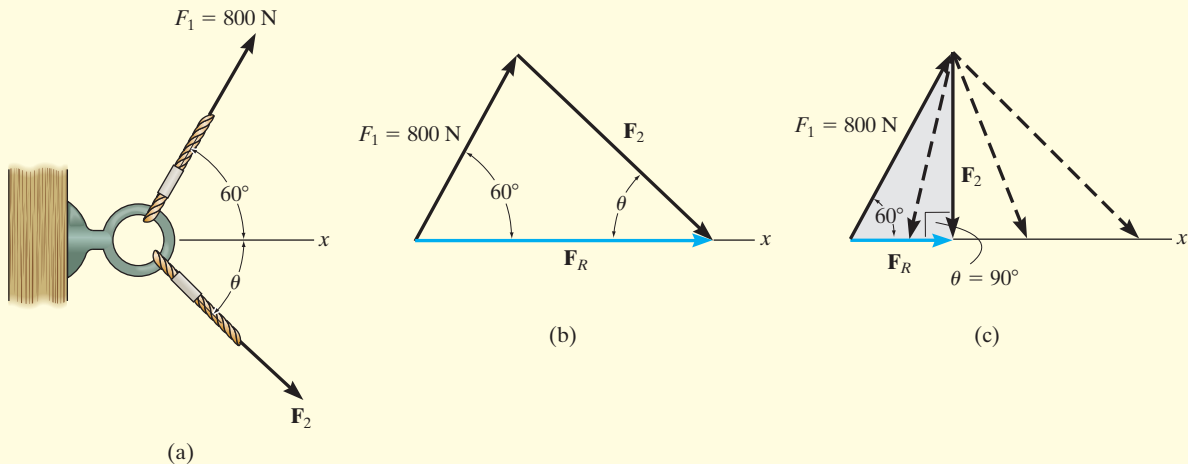


Fig. 2–14

## SOLUTION

The triangle rule for  $\mathbf{F}_R = \mathbf{F}_1 + \mathbf{F}_2$  is shown in Fig. 2–14b. Since the magnitudes (lengths) of  $\mathbf{F}_R$  and  $\mathbf{F}_2$  are not specified, then  $\mathbf{F}_2$  can actually be any vector that has its head touching the line of action of  $\mathbf{F}_R$ , Fig. 2–14c. However, as shown, the magnitude of  $\mathbf{F}_2$  is a *minimum* or the shortest length when its line of action is *perpendicular* to the line of action of  $\mathbf{F}_R$ , that is, when

$$\theta = 90^\circ \quad \text{Ans.}$$

Since the vector addition now forms the shaded right triangle, the two unknown magnitudes can be obtained by trigonometry.

$$F_R = (800\text{ N})\cos 60^\circ = 400\text{ N} \quad \text{Ans.}$$

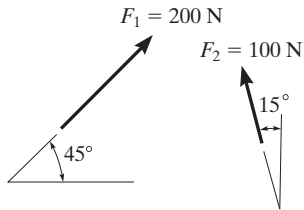
$$F_2 = (800\text{ N})\sin 60^\circ = 693\text{ N} \quad \text{Ans.}$$

**It is strongly suggested that you test yourself on the solutions to these examples, by covering them over and then trying to draw the parallelogram law, and thinking about how the sine and cosine laws are used to determine the unknowns. Then before solving any of the problems, try to solve the Preliminary Problems and some of the Fundamental Problems given on the next pages. The solutions and answers to these are given in the back of the book. Doing this throughout the book will help immensely in developing your problem-solving skills.**

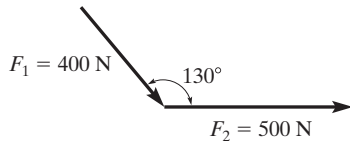
## PRELIMINARY PROBLEMS

*Partial solutions and answers to all Preliminary Problems are given in the back of the book.*

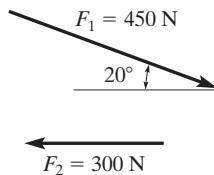
**P2-1.** In each case, construct the parallelogram law to show  $\mathbf{F}_R = \mathbf{F}_1 + \mathbf{F}_2$ . Then establish the triangle rule, where  $\mathbf{F}_R = \mathbf{F}_1 + \mathbf{F}_2$ . Label all known and unknown sides and interior angles.



(a)



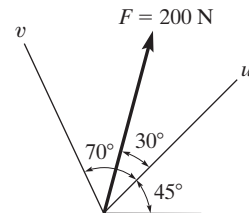
(b)



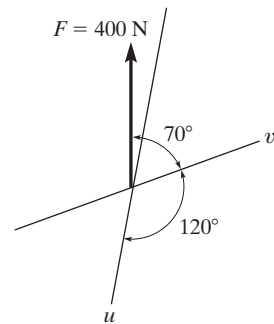
(c)

**Prob. P2-1**

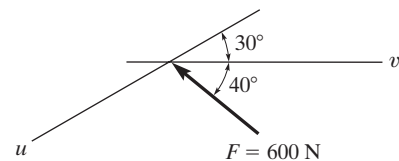
**P2-2.** In each case, show how to resolve the force  $\mathbf{F}$  into components acting along the  $u$  and  $v$  axes using the parallelogram law. Then establish the triangle rule to show  $\mathbf{F}_R = \mathbf{F}_u + \mathbf{F}_v$ . Label all known and unknown sides and interior angles.



(a)



(b)



(c)

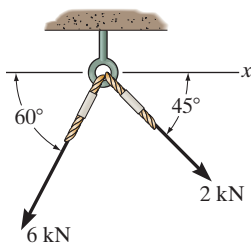
**Prob. P2-2**

## FUNDAMENTAL PROBLEMS

*Partial solutions and answers to all Fundamental Problems are given in the back of the book.*

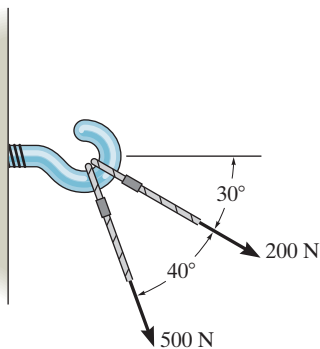
2

**F2-1.** Determine the magnitude of the resultant force acting on the screw eye and its direction measured clockwise from the  $x$  axis.



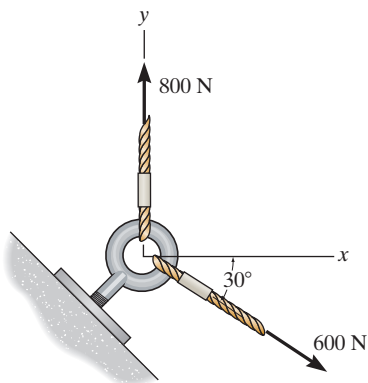
**Prob. F2-1**

**F2-2.** Two forces act on the hook. Determine the magnitude of the resultant force.



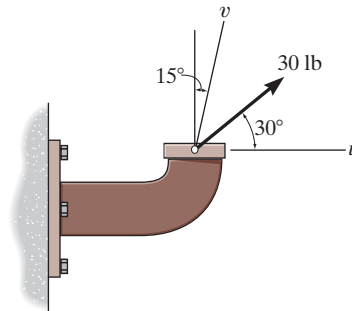
**Prob. F2-2**

**F2-3.** Determine the magnitude of the resultant force and its direction measured counterclockwise from the positive  $x$  axis.



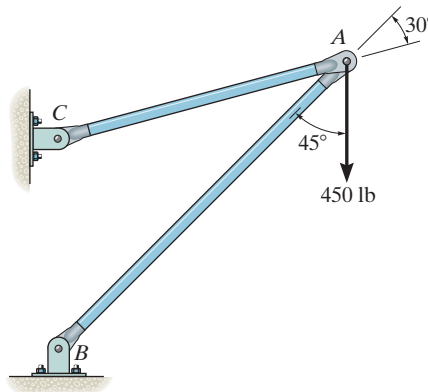
**Prob. F2-3**

**F2-4.** Resolve the 30-lb force into components along the  $u$  and  $v$  axes, and determine the magnitude of each of these components.



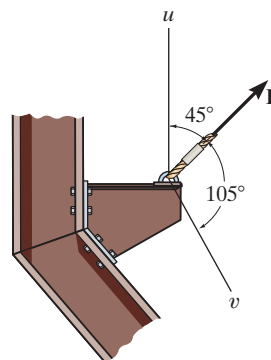
**Prob. F2-4**

**F2-5.** The force  $F = 450$  lb acts on the frame. Resolve this force into components acting along members  $AB$  and  $AC$ , and determine the magnitude of each component.



**Prob. F2-5**

**F2-6.** If force  $\mathbf{F}$  is to have a component along the  $u$  axis of  $F_u = 6$  kN, determine the magnitude of  $\mathbf{F}$  and the magnitude of its component  $\mathbf{F}_v$  along the  $v$  axis.

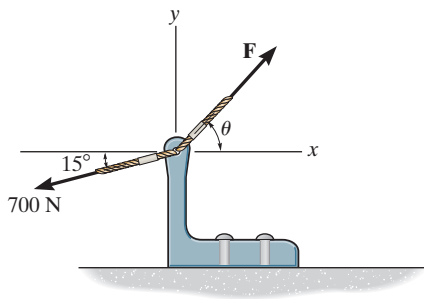


**Prob. F2-6**

## PROBLEMS

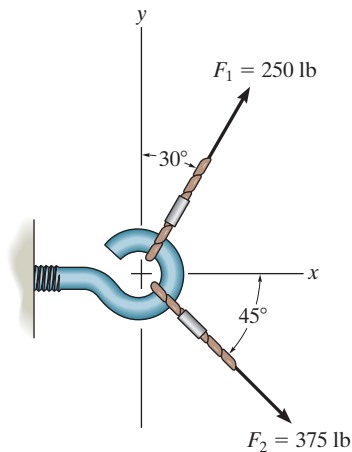
**2-1.** If  $\theta = 60^\circ$  and  $F = 450$  N, determine the magnitude of the resultant force and its direction, measured counterclockwise from the positive  $x$  axis.

**2-2.** If the magnitude of the resultant force is to be 500 N, directed along the positive  $y$  axis, determine the magnitude of force  $\mathbf{F}$  and its direction  $\theta$ .



**Probs. 2-1/2**

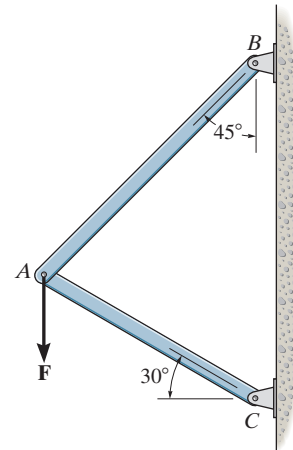
**2-3.** Determine the magnitude of the resultant force  $\mathbf{F}_R = \mathbf{F}_1 + \mathbf{F}_2$  and its direction, measured counterclockwise from the positive  $x$  axis.



**Prob. 2-3**

**\*2-4.** The vertical force  $\mathbf{F}$  acts downward at  $A$  on the two-membered frame. Determine the magnitudes of the two components of  $\mathbf{F}$  directed along the axes of  $AB$  and  $AC$ . Set  $F = 500$  N.

**2-5.** Solve Prob. 2-4 with  $F = 350$  lb.

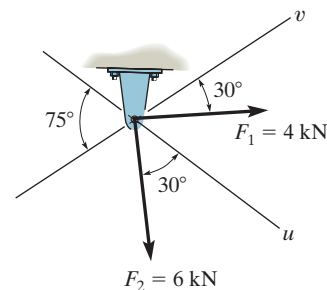


**Probs. 2-4/5**

**2-6.** Determine the magnitude of the resultant force  $\mathbf{F}_R = \mathbf{F}_1 + \mathbf{F}_2$  and its direction, measured clockwise from the positive  $u$  axis.

**2-7.** Resolve the force  $\mathbf{F}_1$  into components acting along the  $u$  and  $v$  axes and determine the magnitudes of the components.

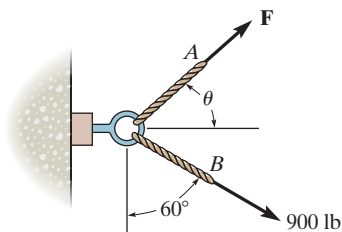
**\*2-8.** Resolve the force  $\mathbf{F}_2$  into components acting along the  $u$  and  $v$  axes and determine the magnitudes of the components.



**Probs. 2-6/7/8**

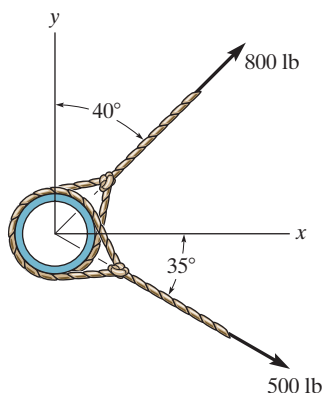


**2-9.** If the resultant force acting on the support is to be 1200 lb, directed horizontally to the right, determine the force  $\mathbf{F}$  in rope  $A$  and the corresponding angle  $\theta$ .



**Prob. 2-9**

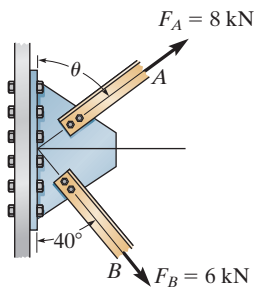
**2-10.** Determine the magnitude of the resultant force and its direction, measured counterclockwise from the positive  $x$  axis.



**Prob. 2-10**

**2-11.** The plate is subjected to the two forces at  $A$  and  $B$  as shown. If  $\theta = 60^\circ$ , determine the magnitude of the resultant of these two forces and its direction measured clockwise from the horizontal.

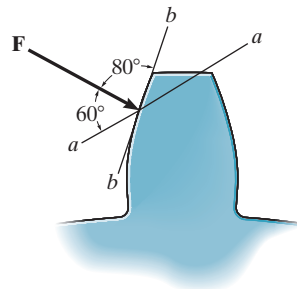
**\*2-12.** Determine the angle  $\theta$  for connecting member  $A$  to the plate so that the resultant force of  $\mathbf{F}_A$  and  $\mathbf{F}_B$  is directed horizontally to the right. Also, what is the magnitude of the resultant force?



**Probs. 2-11/12**

**2-13.** The force acting on the gear tooth is  $F = 20$  lb. Resolve this force into two components acting along the lines  $aa$  and  $bb$ .

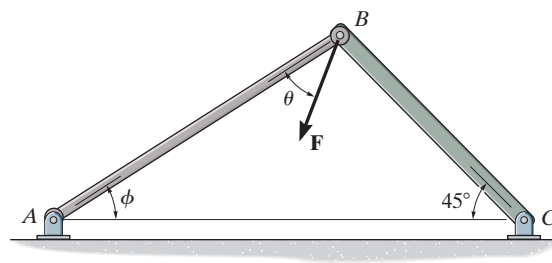
**2-14.** The component of force  $\mathbf{F}$  acting along line  $aa$  is required to be 30 lb. Determine the magnitude of  $\mathbf{F}$  and its component along line  $bb$ .



**Probs. 2-13/14**

**2-15.** Force  $\mathbf{F}$  acts on the frame such that its component acting along member  $AB$  is 650 lb, directed from  $B$  towards  $A$ , and the component acting along member  $BC$  is 500 lb, directed from  $B$  towards  $C$ . Determine the magnitude of  $\mathbf{F}$  and its direction  $\theta$ . Set  $\phi = 60^\circ$ .

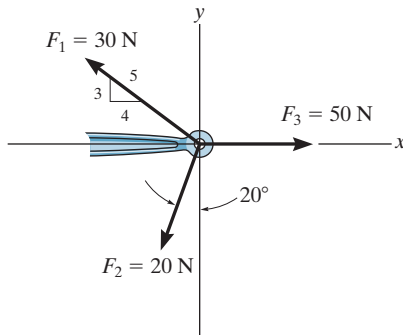
**\*2-16.** Force  $\mathbf{F}$  acts on the frame such that its component acting along member  $AB$  is 650 lb, directed from  $B$  towards  $A$ . Determine the required angle  $\phi$  ( $0^\circ \leq \phi \leq 45^\circ$ ) and the component acting along member  $BC$ . Set  $F = 850$  lb and  $\theta = 30^\circ$ .



**Probs. 2-15/16**

**2-17.** Determine the magnitude and direction of the resultant  $\mathbf{F}_R = \mathbf{F}_1 + \mathbf{F}_2 + \mathbf{F}_3$  of the three forces by first finding the resultant  $\mathbf{F}' = \mathbf{F}_1 + \mathbf{F}_2$  and then forming  $\mathbf{F}_R = \mathbf{F}' + \mathbf{F}_3$ .

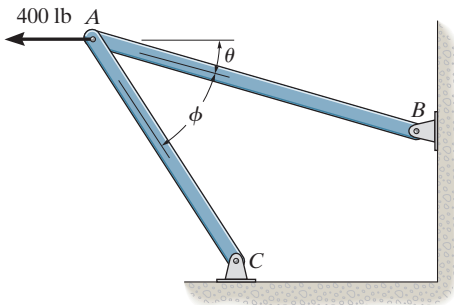
**2-18.** Determine the magnitude and direction of the resultant  $\mathbf{F}_R = \mathbf{F}_1 + \mathbf{F}_2 + \mathbf{F}_3$  of the three forces by first finding the resultant  $\mathbf{F}' = \mathbf{F}_2 + \mathbf{F}_3$  and then forming  $\mathbf{F}_R = \mathbf{F}' + \mathbf{F}_1$ .



**Probs. 2-17/18**

**2-19.** Determine the design angle  $\theta$  ( $0^\circ \leq \theta \leq 90^\circ$ ) for strut  $AB$  so that the 400-lb horizontal force has a component of 500 lb directed from  $A$  towards  $C$ . What is the component of force acting along member  $AB$ ? Take  $\phi = 40^\circ$ .

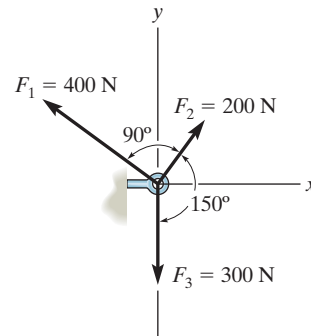
**\*2-20.** Determine the design angle  $\phi$  ( $0^\circ \leq \phi \leq 90^\circ$ ) between struts  $AB$  and  $AC$  so that the 400-lb horizontal force has a component of 600 lb which acts up to the left, in the same direction as from  $B$  towards  $A$ . Take  $\theta = 30^\circ$ .



**Probs. 2-19/20**

**2-21.** Determine the magnitude and direction of the resultant force,  $\mathbf{F}_R$  measured counterclockwise from the positive  $x$  axis. Solve the problem by first finding the resultant  $\mathbf{F}' = \mathbf{F}_1 + \mathbf{F}_2$  and then forming  $\mathbf{F}_R = \mathbf{F}' + \mathbf{F}_3$ .

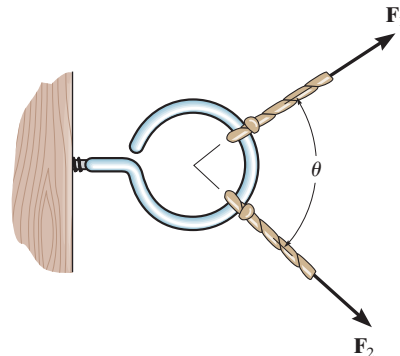
**2-22.** Determine the magnitude and direction of the resultant force, measured counterclockwise from the positive  $x$  axis. Solve  $\mathbf{F}' = \mathbf{F}_2 + \mathbf{F}_3$  and then forming  $\mathbf{F}_R = \mathbf{F}' + \mathbf{F}_1$ .



**Probs. 2-21/22**

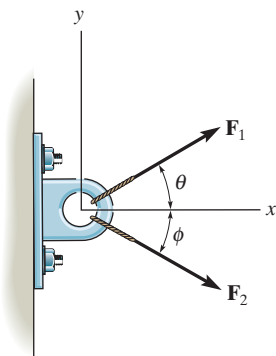
**2-23.** Two forces act on the screw eye. If  $F_1 = 400$  N and  $F_2 = 600$  N, determine the angle  $\theta$  ( $0^\circ \leq \theta \leq 180^\circ$ ) between them, so that the resultant force has a magnitude of  $F_R = 800$  N.

**\*2-24.** Two forces  $\mathbf{F}_1$  and  $\mathbf{F}_2$  act on the screw eye. If their lines of action are at an angle  $\theta$  apart and the magnitude of each force is  $F_1 = F_2 = F$ , determine the magnitude of the resultant force  $\mathbf{F}_R$  and the angle between  $\mathbf{F}_R$  and  $\mathbf{F}_1$ .



**Probs. 2-23/24**

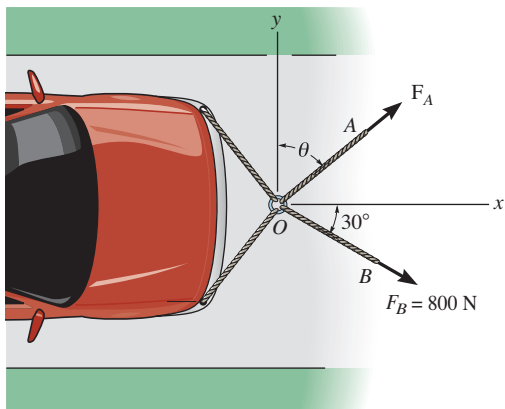
**2-25.** If  $F_1 = 30$  lb and  $F_2 = 40$  lb, determine the angles  $\theta$  and  $\phi$  so that the resultant force is directed along the positive  $x$  axis and has a magnitude of  $F_R = 60$  lb.



**Prob. 2-25**

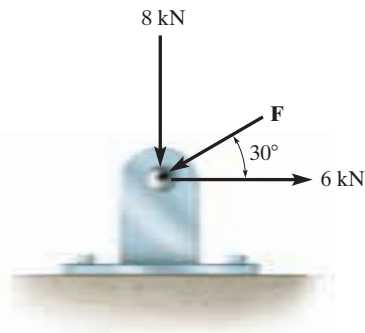
**2-26.** Determine the magnitude and direction  $\theta$  of  $\mathbf{F}_A$  so that the resultant force is directed along the positive  $x$  axis and has a magnitude of 1250 N.

**2-27.** Determine the magnitude and direction, measured counterclockwise from the positive  $x$  axis, of the resultant force acting on the ring at  $O$ , if  $F_A = 750$  N and  $\theta = 45^\circ$ .



**Probs. 2-26/27**

**\*2-28.** Determine the magnitude of force  $\mathbf{F}$  so that the resultant  $\mathbf{F}_R$  of the three forces is as small as possible. What is the minimum magnitude of  $\mathbf{F}_R$ ?

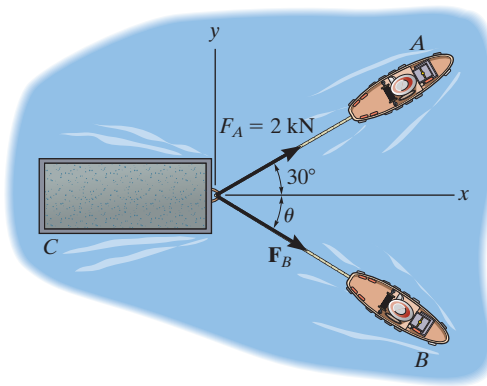


**Prob. 2-28**

**2-29.** If the resultant force of the two tugboats is 3 kN, directed along the positive  $x$  axis, determine the required magnitude of force  $\mathbf{F}_B$  and its direction  $\theta$ .

**2-30.** If  $\mathbf{F}_B = 3$  kN and  $\theta = 45^\circ$ , determine the magnitude of the resultant force of the two tugboats and its direction measured clockwise from the positive  $x$  axis.

**2-31.** If the resultant force of the two tugboats is required to be directed towards the positive  $x$  axis, and  $\mathbf{F}_B$  is to be a minimum, determine the magnitude of  $\mathbf{F}_R$  and  $\mathbf{F}_B$  and the angle  $\theta$ .



**Probs. 2-29/30/31**

## 2.4 Addition of a System of Coplanar Forces

When a force is resolved into two components along the  $x$  and  $y$  axes, the components are then called **rectangular components**. For analytical work we can represent these components in one of two ways, using either scalar or Cartesian vector notation.

**Scalar Notation.** The rectangular components of force  $\mathbf{F}$  shown in Fig. 2-15a are found using the parallelogram law, so that  $\mathbf{F} = \mathbf{F}_x + \mathbf{F}_y$ . Because these components form a right triangle, they can be determined from

$$F_x = F \cos \theta \quad \text{and} \quad F_y = F \sin \theta$$

Instead of using the angle  $\theta$ , however, the direction of  $\mathbf{F}$  can also be defined using a small “slope” triangle, as in the example shown in Fig. 2-15b. Since this triangle and the larger shaded triangle are similar, the proportional length of the sides gives

$$\frac{F_x}{F} = \frac{a}{c}$$

or

$$F_x = F \left( \frac{a}{c} \right)$$

and

$$\frac{F_y}{F} = \frac{b}{c}$$

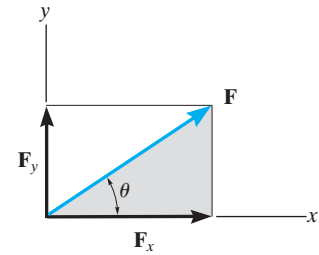
or

$$F_y = -F \left( \frac{b}{c} \right)$$

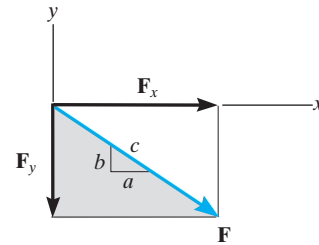
Here the  $y$  component is a *negative scalar* since  $\mathbf{F}_y$  is directed along the negative  $y$  axis.

It is important to keep in mind that this positive and negative scalar notation is to be used only for computational purposes, not for graphical representations in figures. Throughout the book, the *head of a vector arrow in any figure* indicates the sense of the vector *graphically*; algebraic signs are not used for this purpose. Thus, the vectors in Figs. 2-15a and 2-15b are designated by using boldface (vector) notation.\* Whenever italic symbols are written near vector arrows in figures, they indicate the *magnitude* of the vector, which is *always a positive* quantity.

\*Negative signs are used only in figures with boldface notation when showing equal but opposite pairs of vectors, as in Fig. 2-2.



(a)



(b)

**Fig. 2-15**

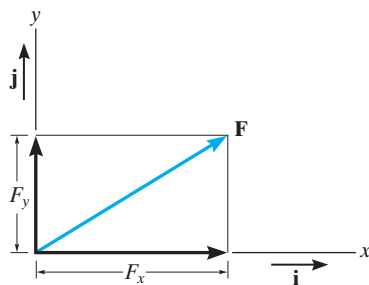


Fig. 2-16

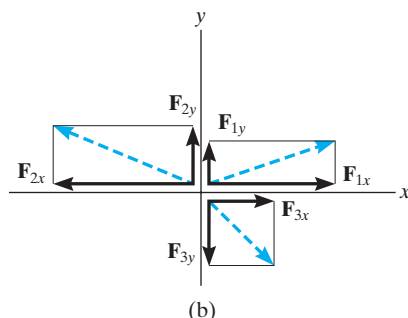
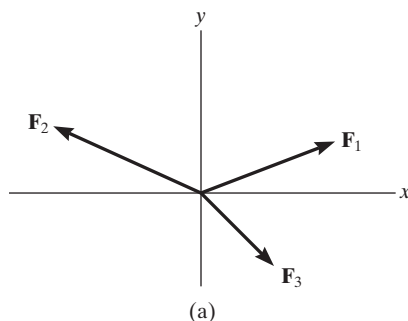
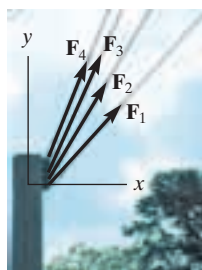


Fig. 2-17



The resultant force of the four cable forces acting on the post can be determined by adding algebraically the separate  $x$  and  $y$  components of each cable force. This resultant  $\mathbf{F}_R$  produces the *same pulling effect* on the post as all four cables. (© Russell C. Hibbeler)

**Cartesian Vector Notation.** It is also possible to represent the  $x$  and  $y$  components of a force in terms of Cartesian unit vectors  $\mathbf{i}$  and  $\mathbf{j}$ . They are called unit vectors because they have a dimensionless magnitude of 1, and so they can be used to designate the *directions* of the  $x$  and  $y$  axes, respectively, Fig. 2-16.\*

Since the *magnitude* of each component of  $\mathbf{F}$  is *always a positive quantity*, which is represented by the (positive) scalars  $F_x$  and  $F_y$ , then we can express  $\mathbf{F}$  as a **Cartesian vector**,

$$\mathbf{F} = F_x \mathbf{i} + F_y \mathbf{j}$$

**Coplanar Force Resultants.** We can use either of the two methods just described to determine the resultant of several **coplanar forces**, i.e., forces that all lie in the same plane. To do this, each force is first resolved into its  $x$  and  $y$  components, and then the respective components are added using *scalar algebra* since they are collinear. The resultant force is then formed by adding the resultant components using the parallelogram law. For example, consider the three concurrent forces in Fig. 2-17a, which have  $x$  and  $y$  components shown in Fig. 2-17b. Using Cartesian vector notation, each force is first represented as a Cartesian vector, i.e.,

$$\mathbf{F}_1 = F_{1x} \mathbf{i} + F_{1y} \mathbf{j}$$

$$\mathbf{F}_2 = -F_{2x} \mathbf{i} + F_{2y} \mathbf{j}$$

$$\mathbf{F}_3 = F_{3x} \mathbf{i} - F_{3y} \mathbf{j}$$

The vector resultant is therefore

$$\begin{aligned} \mathbf{F}_R &= \mathbf{F}_1 + \mathbf{F}_2 + \mathbf{F}_3 \\ &= F_{1x} \mathbf{i} + F_{1y} \mathbf{j} - F_{2x} \mathbf{i} + F_{2y} \mathbf{j} + F_{3x} \mathbf{i} - F_{3y} \mathbf{j} \\ &= (F_{1x} - F_{2x} + F_{3x}) \mathbf{i} + (F_{1y} + F_{2y} - F_{3y}) \mathbf{j} \\ &= (F_{Rx}) \mathbf{i} + (F_{Ry}) \mathbf{j} \end{aligned}$$

If *scalar notation* is used, then indicating the positive directions of components along the  $x$  and  $y$  axes with symbolic arrows, we have

$$\rightarrow \quad (F_R)_x = F_{1x} - F_{2x} + F_{3x}$$

$$+ \uparrow \quad (F_R)_y = F_{1y} + F_{2y} - F_{3y}$$

These are the *same* results as the  $\mathbf{i}$  and  $\mathbf{j}$  components of  $\mathbf{F}_R$  determined above.

\*For handwritten work, unit vectors are usually indicated using a circumflex, e.g.,  $\hat{i}$  and  $\hat{j}$ . Also, realize that  $F_x$  and  $F_y$  in Fig. 2-16 represent the *magnitudes* of the components, which are *always positive scalars*. The directions are defined by  $\mathbf{i}$  and  $\mathbf{j}$ . If instead we used scalar notation, then  $F_x$  and  $F_y$  could be positive or negative scalars, since they would account for *both* the magnitude and direction of the components.

We can represent the components of the resultant force of any number of coplanar forces symbolically by the algebraic sum of the  $x$  and  $y$  components of all the forces, i.e.,

$$\begin{aligned}(F_R)_x &= \Sigma F_x \\ (F_R)_y &= \Sigma F_y\end{aligned}\quad (2-1)$$

Once these components are determined, they may be sketched along the  $x$  and  $y$  axes with their proper sense of direction, and the resultant force can be determined from vector addition, as shown in Fig. 2-17c. From this sketch, the magnitude of  $\mathbf{F}_R$  is then found from the Pythagorean theorem; that is,

$$F_R = \sqrt{(F_R)_x^2 + (F_R)_y^2}$$

Also, the angle  $\theta$ , which specifies the direction of the resultant force, is determined from trigonometry:

$$\theta = \tan^{-1} \left| \frac{(F_R)_y}{(F_R)_x} \right|$$

The above concepts are illustrated numerically in the examples which follow.

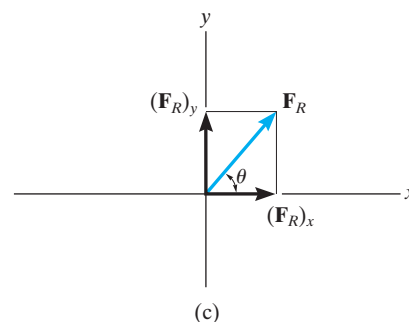


Fig. 2-17 (cont.)

### Important Points

- The resultant of several coplanar forces can easily be determined if an  $x, y$  coordinate system is established and the forces are resolved along the axes.
- The direction of each force is specified by the angle its line of action makes with one of the axes, or by a slope triangle.
- The orientation of the  $x$  and  $y$  axes is arbitrary, and their positive direction can be specified by the Cartesian unit vectors  $\mathbf{i}$  and  $\mathbf{j}$ .
- The  $x$  and  $y$  components of the *resultant force* are simply the algebraic addition of the components of all the coplanar forces.
- The magnitude of the resultant force is determined from the Pythagorean theorem, and when the resultant components are sketched on the  $x$  and  $y$  axes, Fig. 2-17c, the direction  $\theta$  can be determined from trigonometry.

## EXAMPLE 2.5

Determine the  $x$  and  $y$  components of  $\mathbf{F}_1$  and  $\mathbf{F}_2$  acting on the boom shown in Fig. 2–18a. Express each force as a Cartesian vector.

## SOLUTION

**Scalar Notation.** By the parallelogram law,  $\mathbf{F}_1$  is resolved into  $x$  and  $y$  components, Fig. 2–18b. Since  $\mathbf{F}_{1x}$  acts in the  $-x$  direction, and  $\mathbf{F}_{1y}$  acts in the  $+y$  direction, we have

$$F_{1x} = -200 \sin 30^\circ \text{ N} = -100 \text{ N} = 100 \text{ N} \leftarrow \quad \text{Ans.}$$

$$F_{1y} = 200 \cos 30^\circ \text{ N} = 173 \text{ N} = 173 \text{ N} \uparrow \quad \text{Ans.}$$

The force  $\mathbf{F}_2$  is resolved into its  $x$  and  $y$  components, as shown in Fig. 2–18c. Here the *slope* of the line of action for the force is indicated. From this “slope triangle” we could obtain the angle  $\theta$ , e.g.,  $\theta = \tan^{-1}\left(\frac{5}{12}\right)$ , and then proceed to determine the magnitudes of the components in the same manner as for  $\mathbf{F}_1$ . The easier method, however, consists of using proportional parts of similar triangles, i.e.,

$$\frac{F_{2x}}{260 \text{ N}} = \frac{12}{13} \quad F_{2x} = 260 \text{ N} \left( \frac{12}{13} \right) = 240 \text{ N}$$

Similarly,

$$F_{2y} = 260 \text{ N} \left( \frac{5}{13} \right) = 100 \text{ N}$$

Notice how the magnitude of the *horizontal component*,  $\mathbf{F}_{2x}$ , was obtained by multiplying the force magnitude by the ratio of the *horizontal leg* of the slope triangle divided by the hypotenuse; whereas the magnitude of the *vertical component*,  $\mathbf{F}_{2y}$ , was obtained by multiplying the force magnitude by the ratio of the *vertical leg* divided by the hypotenuse. Hence, using scalar notation to represent these components, we have

$$F_{2x} = 240 \text{ N} = 240 \text{ N} \rightarrow \quad \text{Ans.}$$

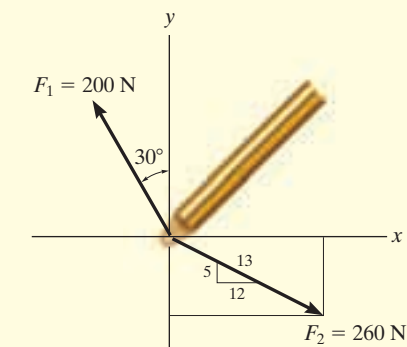
$$F_{2y} = -100 \text{ N} = 100 \text{ N} \downarrow \quad \text{Ans.}$$

**Cartesian Vector Notation.** Having determined the magnitudes and directions of the components of each force, we can express each force as a Cartesian vector.

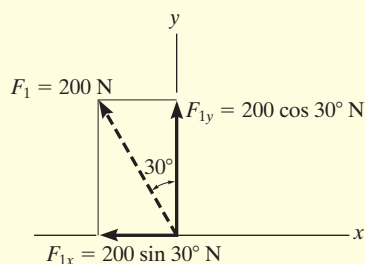
$$\mathbf{F}_1 = \{-100\mathbf{i} + 173\mathbf{j}\} \text{ N} \quad \text{Ans.}$$

$$\mathbf{F}_2 = \{240\mathbf{i} - 100\mathbf{j}\} \text{ N} \quad \text{Ans.}$$

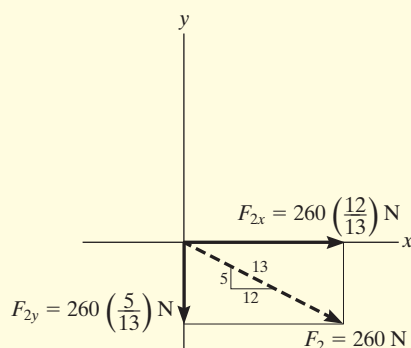
2



(a)



(b)



(c)

Fig. 2–18

**EXAMPLE 2.6**

The link in Fig. 2–19a is subjected to two forces  $\mathbf{F}_1$  and  $\mathbf{F}_2$ . Determine the magnitude and direction of the resultant force.

**SOLUTION I**

**Scalar Notation.** First we resolve each force into its  $x$  and  $y$  components, Fig. 2–19b, then we sum these components algebraically.

$$\begin{aligned} \rightarrow (F_R)_x &= \Sigma F_x; & (F_R)_x &= 600 \cos 30^\circ \text{ N} - 400 \sin 45^\circ \text{ N} \\ & & &= 236.8 \text{ N} \rightarrow \end{aligned}$$

$$\begin{aligned} +\uparrow (F_R)_y &= \Sigma F_y; & (F_R)_y &= 600 \sin 30^\circ \text{ N} + 400 \cos 45^\circ \text{ N} \\ & & &= 582.8 \text{ N} \uparrow \end{aligned}$$

The resultant force, shown in Fig. 2–19c, has a *magnitude* of

$$\begin{aligned} F_R &= \sqrt{(236.8 \text{ N})^2 + (582.8 \text{ N})^2} \\ &= 629 \text{ N} \end{aligned}$$

*Ans.*

From the vector addition,

$$\theta = \tan^{-1} \left( \frac{582.8 \text{ N}}{236.8 \text{ N}} \right) = 67.9^\circ$$

*Ans.***SOLUTION II**

**Cartesian Vector Notation.** From Fig. 2–19b, each force is first expressed as a Cartesian vector.

$$\mathbf{F}_1 = \{600 \cos 30^\circ \mathbf{i} + 600 \sin 30^\circ \mathbf{j}\} \text{ N}$$

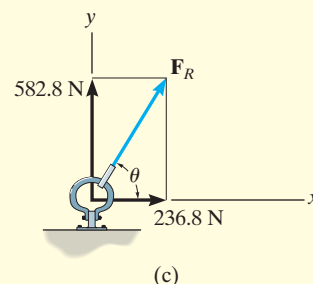
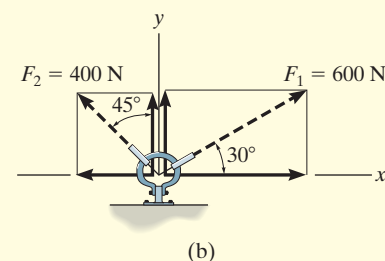
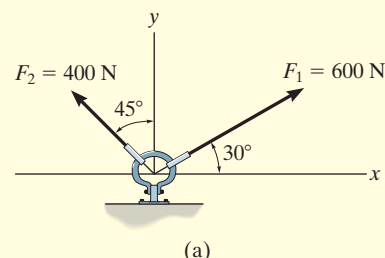
$$\mathbf{F}_2 = \{-400 \sin 45^\circ \mathbf{i} + 400 \cos 45^\circ \mathbf{j}\} \text{ N}$$

Then,

$$\begin{aligned} \mathbf{F}_R &= \mathbf{F}_1 + \mathbf{F}_2 = (600 \cos 30^\circ \text{ N} - 400 \sin 45^\circ \text{ N})\mathbf{i} \\ &\quad + (600 \sin 30^\circ \text{ N} + 400 \cos 45^\circ \text{ N})\mathbf{j} \\ &= \{236.8\mathbf{i} + 582.8\mathbf{j}\} \text{ N} \end{aligned}$$

The magnitude and direction of  $\mathbf{F}_R$  are determined in the same manner as before.

**NOTE:** Comparing the two methods of solution, notice that the use of scalar notation is more efficient since the components can be found *directly*, without first having to express each force as a Cartesian vector before adding the components. Later, however, we will show that Cartesian vector analysis is very beneficial for solving three-dimensional problems.



**Fig. 2–19**



## EXAMPLE 2.7

The end of the boom  $O$  in Fig. 2–20a is subjected to three concurrent and coplanar forces. Determine the magnitude and direction of the resultant force.

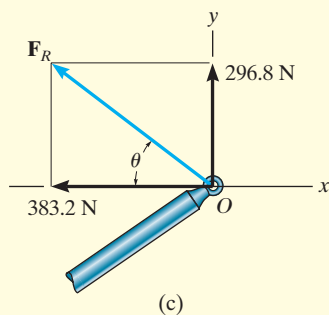
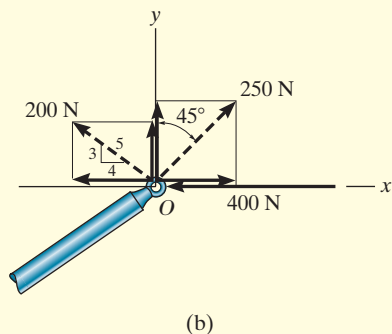
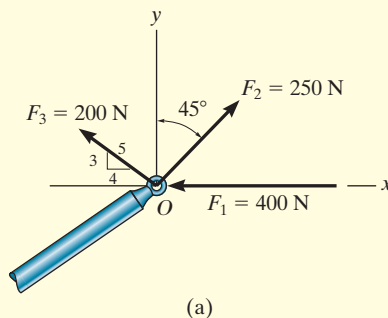


Fig. 2–20

## SOLUTION

Each force is resolved into its  $x$  and  $y$  components, Fig. 2–20b. Summing the  $x$  components, we have

$$\begin{aligned} \rightarrow (F_R)_x &= \Sigma F_x; & (F_R)_x &= -400 \text{ N} + 250 \sin 45^\circ \text{ N} - 200\left(\frac{4}{5}\right) \text{ N} \\ & & &= -383.2 \text{ N} = 383.2 \text{ N} \leftarrow \end{aligned}$$

The negative sign indicates that  $F_{Rx}$  acts to the left, i.e., in the negative  $x$  direction, as noted by the small arrow. Obviously, this occurs because  $F_1$  and  $F_3$  in Fig. 2–20b contribute a greater pull to the left than  $F_2$  which pulls to the right. Summing the  $y$  components yields

$$\begin{aligned} + \uparrow (F_R)_y &= \Sigma F_y; & (F_R)_y &= 250 \cos 45^\circ \text{ N} + 200\left(\frac{3}{5}\right) \text{ N} \\ & & &= 296.8 \text{ N} \uparrow \end{aligned}$$

The resultant force, shown in Fig. 2–20c, has a *magnitude* of

$$\begin{aligned} F_R &= \sqrt{(-383.2 \text{ N})^2 + (296.8 \text{ N})^2} \\ &= 485 \end{aligned} \quad \text{Ans.}$$

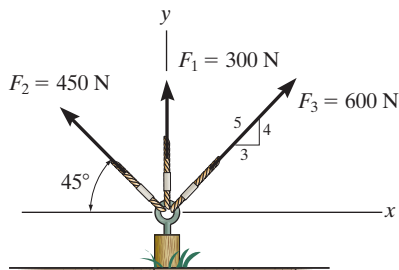
From the vector addition in Fig. 2–20c, the direction angle  $\theta$  is

$$\theta = \tan^{-1}\left(\frac{296.8}{383.2}\right) = 37.8^\circ \quad \text{Ans.}$$

**NOTE:** Application of this method is more convenient, compared to using two applications of the parallelogram law, first to add  $\mathbf{F}_1$  and  $\mathbf{F}_2$  then adding  $\mathbf{F}_3$  to this resultant.

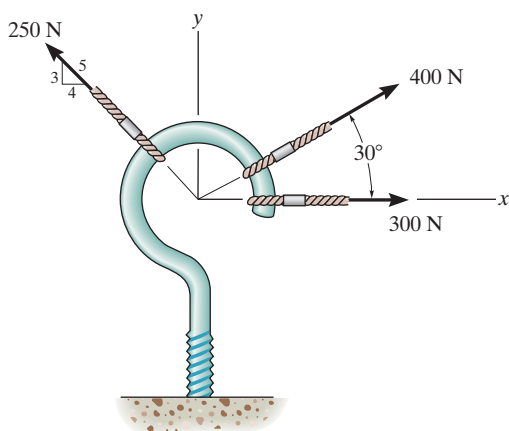
## FUNDAMENTAL PROBLEMS

**F2-7.** Resolve each force acting on the post into its  $x$  and  $y$  components.



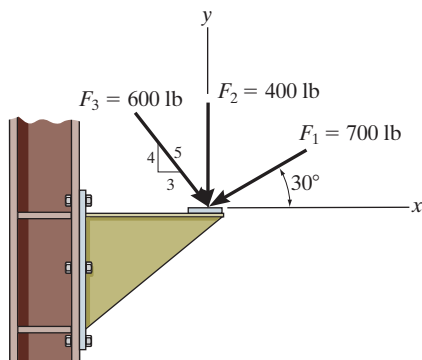
**Prob. F2-7**

**F2-8.** Determine the magnitude and direction of the resultant force.



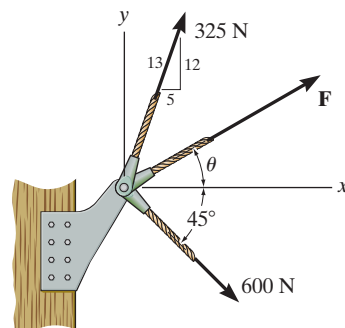
**Prob. F2-8**

**F2-9.** Determine the magnitude of the resultant force acting on the corbel and its direction  $\theta$  measured counterclockwise from the  $x$  axis.



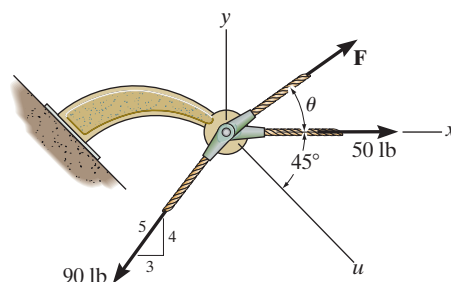
**Prob. F2-9**

**F2-10.** If the resultant force acting on the bracket is to be 750 N directed along the positive  $x$  axis, determine the magnitude of  $\mathbf{F}$  and its direction  $\theta$ .



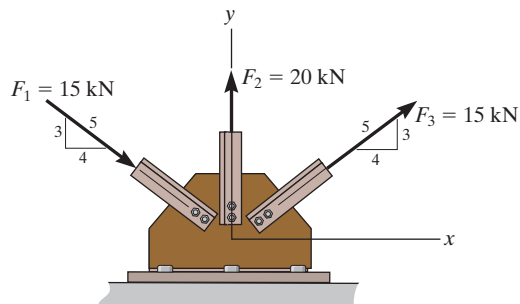
**Prob. F2-10**

**F2-11.** If the magnitude of the resultant force acting on the bracket is to be 80 lb directed along the  $u$  axis, determine the magnitude of  $\mathbf{F}$  and its direction  $\theta$ .



**Prob. F2-11**

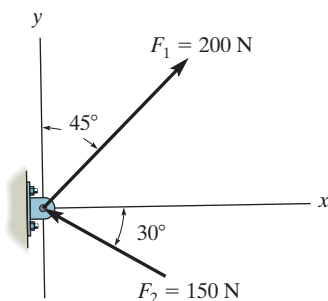
**F2-12.** Determine the magnitude of the resultant force and its direction  $\theta$  measured counterclockwise from the positive  $x$  axis.



**Prob. F2-12**

## PROBLEMS

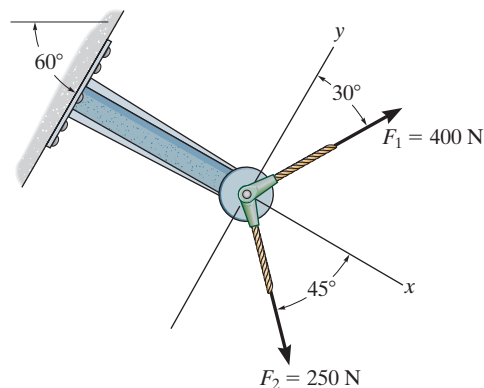
**\*2–32.** Determine the magnitude of the resultant force and its direction, measured counterclockwise from the positive  $x$  axis.



**Prob. 2–32**

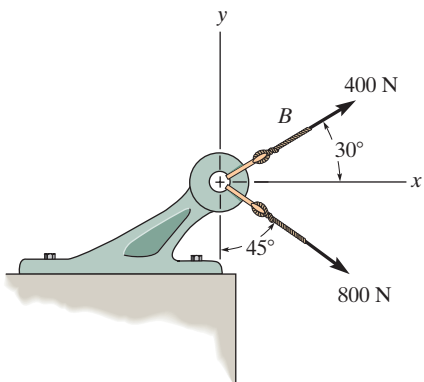
**2–34.** Resolve  $\mathbf{F}_1$  and  $\mathbf{F}_2$  into their  $x$  and  $y$  components.

**2–35.** Determine the magnitude of the resultant force and its direction measured counterclockwise from the positive  $x$  axis.



**Probs. 2–34/35**

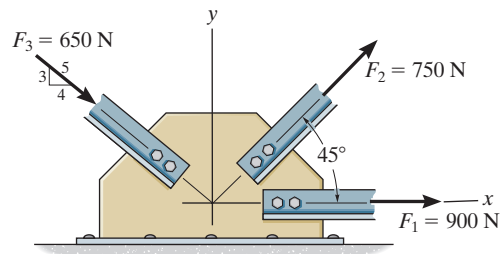
**2–33.** Determine the magnitude of the resultant force and its direction, measured clockwise from the positive  $x$  axis.



**Prob. 2–33**

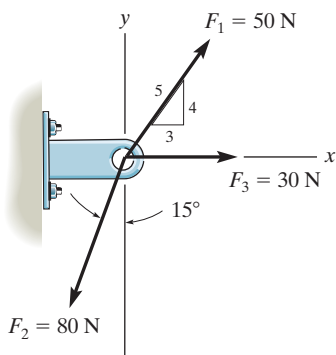
**\*2–36.** Resolve each force acting on the *gusset plate* into its  $x$  and  $y$  components, and express each force as a Cartesian vector.

**2–37.** Determine the magnitude of the resultant force acting on the plate and its direction, measured counterclockwise from the positive  $x$  axis.



**Probs. 2–36/37**

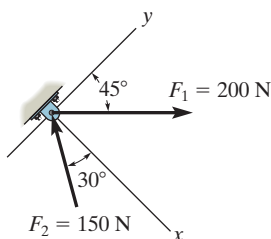
**2-38.** Express each of the three forces acting on the support in Cartesian vector form and determine the magnitude of the resultant force and its direction, measured clockwise from positive  $x$  axis.



**Prob. 2-38**

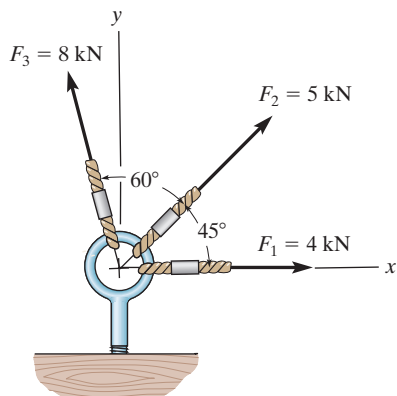
**2-39.** Determine the  $x$  and  $y$  components of  $\mathbf{F}_1$  and  $\mathbf{F}_2$ .

**\*2-40.** Determine the magnitude of the resultant force and its direction, measured counterclockwise from the positive  $x$  axis.



**Probs. 2-39/40**

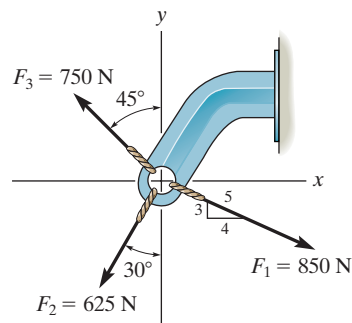
**2-41.** Determine the magnitude of the resultant force and its direction, measured counterclockwise from the positive  $x$  axis.



**Prob. 2-41**

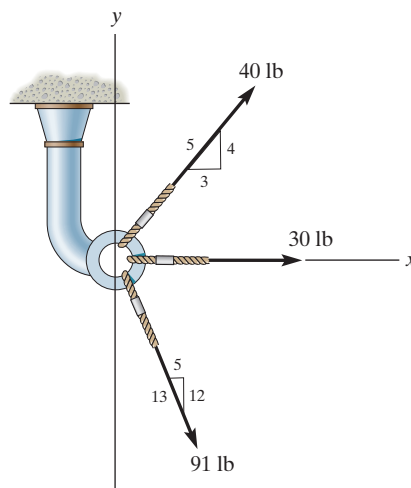
**2-42.** Express  $\mathbf{F}_1$ ,  $\mathbf{F}_2$ , and  $\mathbf{F}_3$  as Cartesian vectors.

**2-43.** Determine the magnitude of the resultant force and its direction, measured counterclockwise from the positive  $x$  axis.



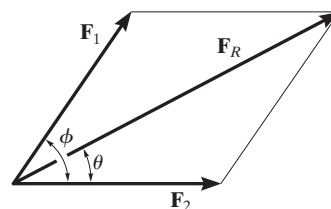
**Probs. 2-42/43**

**\*2-44.** Determine the magnitude of the resultant force and its direction, measured clockwise from the positive  $x$  axis.



**Prob. 2-44**

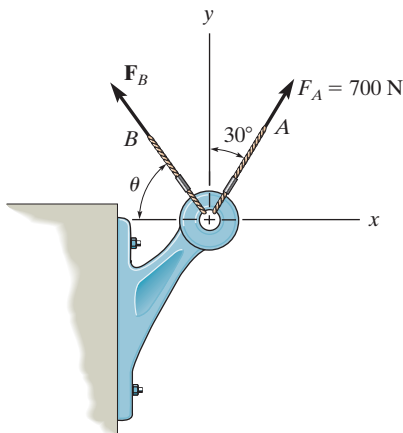
**2-45.** Determine the magnitude and direction  $\theta$  of the resultant force  $\mathbf{F}_R$ . Express the result in terms of the magnitudes of the components  $\mathbf{F}_1$  and  $\mathbf{F}_2$  and the angle  $\phi$ .



**Prob. 2-45**

**2-46.** Determine the magnitude and orientation  $\theta$  of  $\mathbf{F}_B$  so that the resultant force is directed along the positive  $y$  axis and has a magnitude of 1500 N.

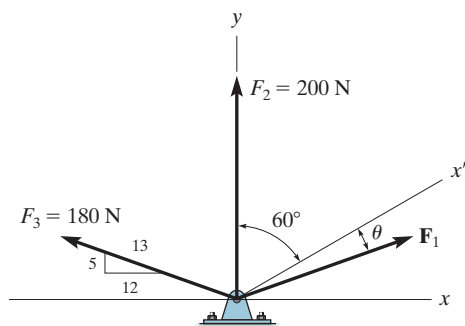
**2-47.** Determine the magnitude and orientation, measured counterclockwise from the positive  $y$  axis, of the resultant force acting on the bracket, if  $F_B = 600$  N and  $\theta = 20^\circ$ .



**Probs. 2-46/47**

**\*2-48.** Three forces act on the bracket. Determine the magnitude and direction  $\theta$  of  $\mathbf{F}_1$  so that the resultant force is directed along the positive  $x'$  axis and has a magnitude of 800 N.

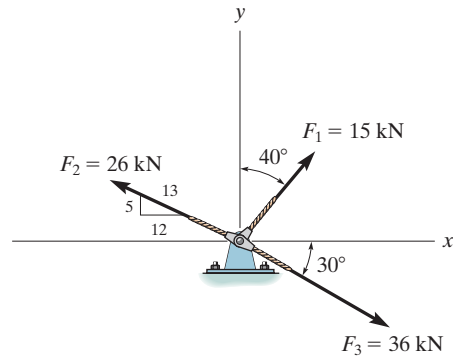
**2-49.** If  $F_1 = 300$  N and  $\theta = 10^\circ$ , determine the magnitude and direction, measured counterclockwise from the positive  $x'$  axis, of the resultant force acting on the bracket.



**Probs. 2-48/49**

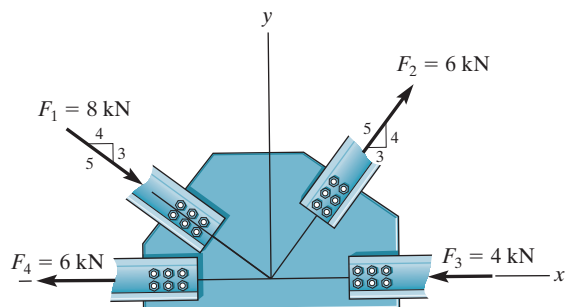
**2-50.** Express  $\mathbf{F}_1$ ,  $\mathbf{F}_2$ , and  $\mathbf{F}_3$  as Cartesian vectors.

**2-51.** Determine the magnitude of the resultant force and its direction, measured counterclockwise from the positive  $x$  axis.



**Probs. 2-50/51**

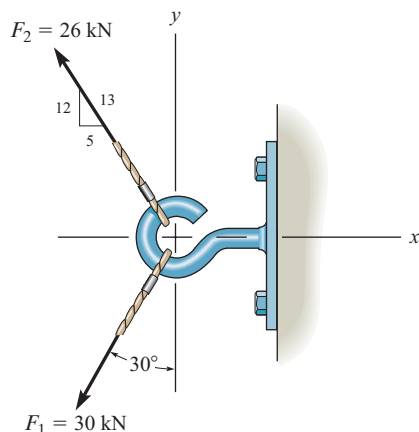
**\*2-52.** Determine the  $x$  and  $y$  components of each force acting on the *gusset plate* of a bridge truss. Show that the resultant force is zero.



**Prob. 2-52**

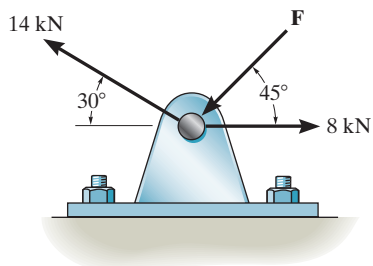
**2-53.** Express  $\mathbf{F}_1$  and  $\mathbf{F}_2$  as Cartesian vectors.

**2-54.** Determine the magnitude of the resultant force and its direction measured counterclockwise from the positive  $x$  axis.



**Probs. 2-53/54**

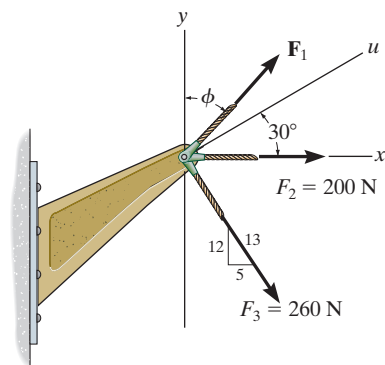
**2-55.** Determine the magnitude of force  $\mathbf{F}$  so that the resultant force of the three forces is as small as possible. What is the magnitude of the resultant force?



**Prob. 2-55**

**\*2-56.** If the magnitude of the resultant force acting on the bracket is to be 450 N directed along the positive  $u$  axis, determine the magnitude of  $\mathbf{F}_1$  and its direction  $\phi$ .

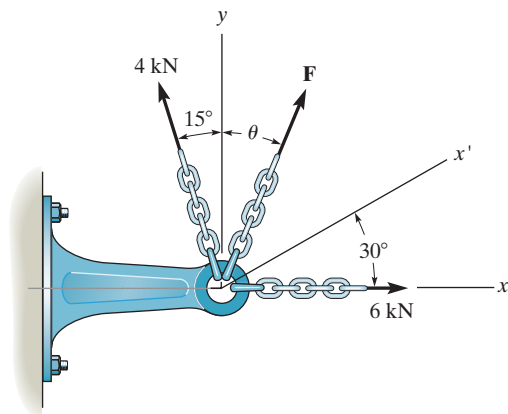
**2-57.** If the resultant force acting on the bracket is required to be a minimum, determine the magnitudes of  $\mathbf{F}_1$  and the resultant force. Set  $\phi = 30^\circ$ .



**Probs. 2-56/57**

**2-58.** Three forces act on the bracket. Determine the magnitude and direction  $\theta$  of  $\mathbf{F}$  so that the resultant force is directed along the positive  $x'$  axis and has a magnitude of 8 kN.

**2-59.** If  $F = 5$  kN and  $\theta = 30^\circ$ , determine the magnitude of the resultant force and its direction, measured counterclockwise from the positive  $x$  axis.



**Probs. 2-58/59**

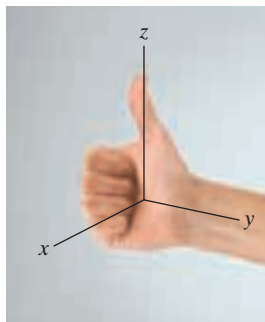


Fig. 2-21 (© Russell C. Hibbeler)

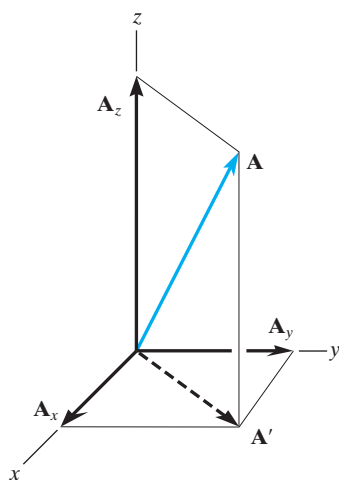


Fig. 2-22

## 2.5 Cartesian Vectors

The operations of vector algebra, when applied to solving problems in *three dimensions*, are greatly simplified if the vectors are first represented in Cartesian vector form. In this section we will present a general method for doing this; then in the next section we will use this method for finding the resultant force of a system of concurrent forces.

**Right-Handed Coordinate System.** We will use a right-handed coordinate system to develop the theory of vector algebra that follows. A rectangular coordinate system is said to be **right-handed** if the thumb of the right hand points in the direction of the positive  $z$  axis when the right-hand fingers are curled about this axis and directed from the positive  $x$  towards the positive  $y$  axis, Fig. 2-21.

**Rectangular Components of a Vector.** A vector  $\mathbf{A}$  may have one, two, or three rectangular components along the  $x$ ,  $y$ ,  $z$  coordinate axes, depending on how the vector is oriented relative to the axes. In general, though, when  $\mathbf{A}$  is directed within an octant of the  $x$ ,  $y$ ,  $z$  frame, Fig. 2-22, then by two successive applications of the parallelogram law, we may resolve the vector into components as  $\mathbf{A} = \mathbf{A}' + \mathbf{A}_z$  and then  $\mathbf{A}' = \mathbf{A}_x + \mathbf{A}_y$ . Combining these equations, to eliminate  $\mathbf{A}'$ ,  $\mathbf{A}$  is represented by the vector sum of its *three* rectangular components,

$$\mathbf{A} = \mathbf{A}_x + \mathbf{A}_y + \mathbf{A}_z \quad (2-2)$$

**Cartesian Unit Vectors.** In three dimensions, the set of Cartesian unit vectors,  $\mathbf{i}$ ,  $\mathbf{j}$ ,  $\mathbf{k}$ , is used to designate the directions of the  $x$ ,  $y$ ,  $z$  axes, respectively. As stated in Sec. 2-4, the *sense* (or arrowhead) of these vectors will be represented analytically by a plus or minus sign, depending on whether they are directed along the positive or negative  $x$ ,  $y$ , or  $z$  axes. The positive Cartesian unit vectors are shown in Fig. 2-23.

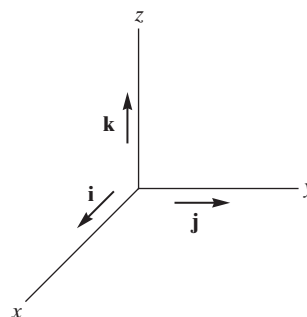


Fig. 2-23

**Cartesian Vector Representation.** Since the three components of  $\mathbf{A}$  in Eq. 2-2 act in the positive  $\mathbf{i}$ ,  $\mathbf{j}$ , and  $\mathbf{k}$  directions, Fig. 2-24, we can write  $\mathbf{A}$  in Cartesian vector form as

$$\mathbf{A} = A_x \mathbf{i} + A_y \mathbf{j} + A_z \mathbf{k} \quad (2-3)$$

There is a distinct advantage to writing vectors in this manner. Separating the *magnitude* and *direction* of each *component vector* will simplify the operations of vector algebra, particularly in three dimensions.

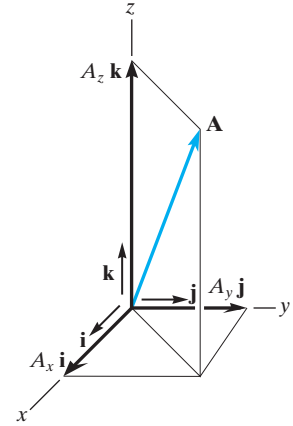


Fig. 2-24

**Magnitude of a Cartesian Vector.** It is always possible to obtain the magnitude of  $\mathbf{A}$  provided it is expressed in Cartesian vector form. As shown in Fig. 2-25, from the blue right triangle,  $A = \sqrt{A'^2 + A_z^2}$ , and from the gray right triangle,  $A' = \sqrt{A_x^2 + A_y^2}$ . Combining these equations to eliminate  $A'$  yields

$$A = \sqrt{A_x^2 + A_y^2 + A_z^2} \quad (2-4)$$

Hence, the magnitude of  $\mathbf{A}$  is equal to the positive square root of the sum of the squares of its components.

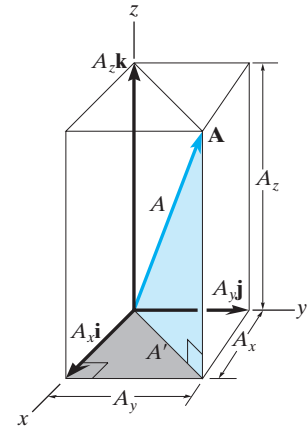


Fig. 2-25

**Coordinate Direction Angles.** We will define the *direction* of  $\mathbf{A}$  by the **coordinate direction angles**  $\alpha$  (alpha),  $\beta$  (beta), and  $\gamma$  (gamma), measured between the *tail* of  $\mathbf{A}$  and the *positive*  $x$ ,  $y$ ,  $z$  axes provided they are located at the tail of  $\mathbf{A}$ , Fig. 2-26. Note that regardless of where  $\mathbf{A}$  is directed, each of these angles will be between  $0^\circ$  and  $180^\circ$ .

To determine  $\alpha$ ,  $\beta$ , and  $\gamma$ , consider the projection of  $\mathbf{A}$  onto the  $x$ ,  $y$ ,  $z$  axes, Fig. 2-27. Referring to the colored right triangles shown in the figure, we have

$$\cos \alpha = \frac{A_x}{A} \quad \cos \beta = \frac{A_y}{A} \quad \cos \gamma = \frac{A_z}{A} \quad (2-5)$$

These numbers are known as the **direction cosines** of  $\mathbf{A}$ . Once they have been obtained, the coordinate direction angles  $\alpha$ ,  $\beta$ ,  $\gamma$  can then be determined from the inverse cosines.

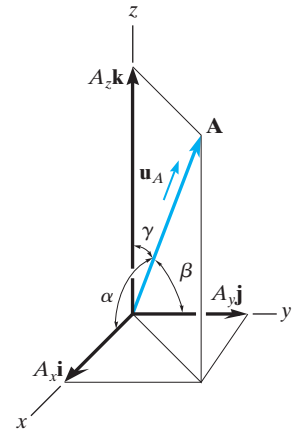


Fig. 2-26



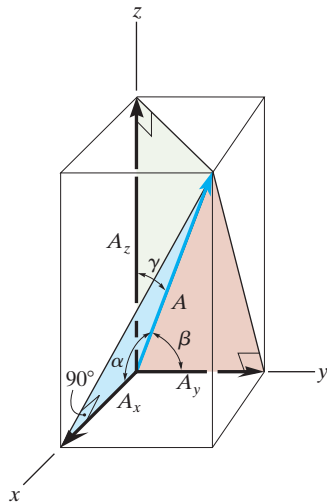


Fig. 2-27

An easy way of obtaining these direction cosines is to form a unit vector  $\mathbf{u}_A$  in the direction of  $\mathbf{A}$ , Fig. 2-26. If  $\mathbf{A}$  is expressed in Cartesian vector form,  $\mathbf{A} = A_x\mathbf{i} + A_y\mathbf{j} + A_z\mathbf{k}$ , then  $\mathbf{u}_A$  will have a magnitude of one and be dimensionless provided  $\mathbf{A}$  is divided by its magnitude, i.e.,

$$\mathbf{u}_A = \frac{\mathbf{A}}{A} = \frac{A_x}{A}\mathbf{i} + \frac{A_y}{A}\mathbf{j} + \frac{A_z}{A}\mathbf{k} \quad (2-6)$$

where  $A = \sqrt{A_x^2 + A_y^2 + A_z^2}$ . By comparison with Eqs. 2-5, it is seen that the  $\mathbf{i}, \mathbf{j}, \mathbf{k}$  components of  $\mathbf{u}_A$  represent the direction cosines of  $\mathbf{A}$ , i.e.,

$$\mathbf{u}_A = \cos \alpha \mathbf{i} + \cos \beta \mathbf{j} + \cos \gamma \mathbf{k} \quad (2-7)$$

Since the magnitude of a vector is equal to the positive square root of the sum of the squares of the magnitudes of its components, and  $\mathbf{u}_A$  has a magnitude of one, then from the above equation an important relation among the direction cosines can be formulated as

$$\cos^2 \alpha + \cos^2 \beta + \cos^2 \gamma = 1 \quad (2-8)$$

Here we can see that if only *two* of the coordinate angles are known, the third angle can be found using this equation.

Finally, if the magnitude and coordinate direction angles of  $\mathbf{A}$  are known, then  $\mathbf{A}$  may be expressed in Cartesian vector form as

$$\begin{aligned} \mathbf{A} &= A \mathbf{u}_A \\ &= A \cos \alpha \mathbf{i} + A \cos \beta \mathbf{j} + A \cos \gamma \mathbf{k} \\ &= A_x \mathbf{i} + A_y \mathbf{j} + A_z \mathbf{k} \end{aligned} \quad (2-9)$$

**Transverse and Azimuth Angles.** Sometimes, the direction of  $\mathbf{A}$  can be specified using two angles, namely, a **transverse angle**  $\theta$  and an **azimuth angle**  $\phi$  (phi), such as shown in Fig. 2-28. The components of  $\mathbf{A}$  can then be determined by applying trigonometry first to the light blue right triangle, which yields

$$A_z = A \cos \phi$$

and

$$A' = A \sin \phi$$

Now applying trigonometry to the dark blue right triangle,

$$A_x = A' \cos \theta = A \sin \phi \cos \theta$$

$$A_y = A' \sin \theta = A \sin \phi \sin \theta$$

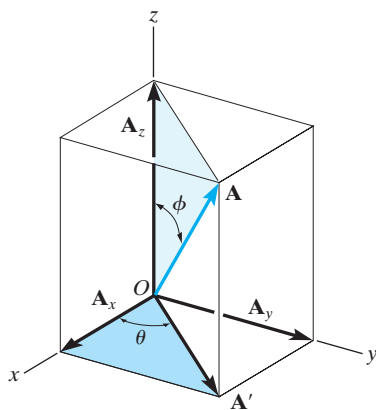


Fig. 2-28

Therefore  $\mathbf{A}$  written in Cartesian vector form becomes

$$\mathbf{A} = A \sin \phi \cos \theta \mathbf{i} + A \sin \phi \sin \theta \mathbf{j} + A \cos \phi \mathbf{k}$$

You should not memorize this equation, rather it is important to understand how the components were determined using trigonometry.

2

## 2.6 Addition of Cartesian Vectors

The addition (or subtraction) of two or more vectors is greatly simplified if the vectors are expressed in terms of their Cartesian components. For example, if  $\mathbf{A} = A_x \mathbf{i} + A_y \mathbf{j} + A_z \mathbf{k}$  and  $\mathbf{B} = B_x \mathbf{i} + B_y \mathbf{j} + B_z \mathbf{k}$ , Fig. 2–29, then the resultant vector,  $\mathbf{R}$ , has components which are the scalar sums of the  $\mathbf{i}, \mathbf{j}, \mathbf{k}$  components of  $\mathbf{A}$  and  $\mathbf{B}$ , i.e.,

$$\mathbf{R} = \mathbf{A} + \mathbf{B} = (A_x + B_x)\mathbf{i} + (A_y + B_y)\mathbf{j} + (A_z + B_z)\mathbf{k}$$

If this is generalized and applied to a system of several concurrent forces, then the force resultant is the vector sum of all the forces in the system and can be written as

$$\mathbf{F}_R = \Sigma \mathbf{F} = \Sigma F_x \mathbf{i} + \Sigma F_y \mathbf{j} + \Sigma F_z \mathbf{k} \quad (2-10)$$

Here  $\Sigma F_x$ ,  $\Sigma F_y$ , and  $\Sigma F_z$  represent the algebraic sums of the respective  $x, y, z$  or  $\mathbf{i}, \mathbf{j}, \mathbf{k}$  components of each force in the system.

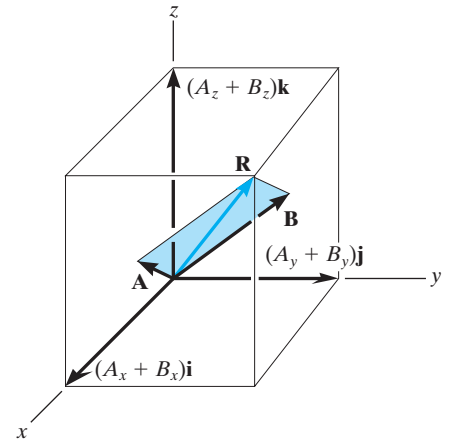


Fig. 2–29



Cartesian vector analysis provides a convenient method for finding both the resultant force and its components in three dimensions. (© Russell C. Hibbeler)

### Important Points

- A Cartesian vector  $\mathbf{A}$  has  $\mathbf{i}, \mathbf{j}, \mathbf{k}$  components along the  $x, y, z$  axes. If  $\mathbf{A}$  is known, its magnitude is defined by  $A = \sqrt{A_x^2 + A_y^2 + A_z^2}$ .
- The direction of a Cartesian vector can be defined by the three angles  $\alpha, \beta, \gamma$ , measured from the *positive*  $x, y, z$  axes to the *tail* of the vector. To find these angles formulate a unit vector in the direction of  $\mathbf{A}$ , i.e.,  $\mathbf{u}_A = \mathbf{A}/A$ , and determine the inverse cosines of its components. Only two of these angles are independent of one another; the third angle is found from  $\cos^2 \alpha + \cos^2 \beta + \cos^2 \gamma = 1$ .
- The direction of a Cartesian vector can also be specified using a transverse angle  $\theta$  and azimuth angle  $\phi$ .

## EXAMPLE 2.8

Express the force  $\mathbf{F}$  shown in Fig. 2–30a as a Cartesian vector.

## SOLUTION

The angles of  $60^\circ$  and  $45^\circ$  defining the direction of  $\mathbf{F}$  are *not* coordinate direction angles. Two successive applications of the parallelogram law are needed to resolve  $\mathbf{F}$  into its  $x$ ,  $y$ ,  $z$  components. First  $\mathbf{F} = \mathbf{F}' + \mathbf{F}_z$ , then  $\mathbf{F}' = \mathbf{F}_x + \mathbf{F}_y$ , Fig. 2–30b. By trigonometry, the magnitudes of the components are

$$F_z = 100 \sin 60^\circ \text{ lb} = 86.6 \text{ lb}$$

$$F' = 100 \cos 60^\circ \text{ lb} = 50 \text{ lb}$$

$$F_x = F' \cos 45^\circ = 50 \cos 45^\circ \text{ lb} = 35.4 \text{ lb}$$

$$F_y = F' \sin 45^\circ = 50 \sin 45^\circ \text{ lb} = 35.4 \text{ lb}$$

Realizing that  $\mathbf{F}_y$  has a direction defined by  $-\mathbf{j}$ , we have

$$\mathbf{F} = \{35.4\mathbf{i} - 35.4\mathbf{j} + 86.6\mathbf{k}\} \text{ lb}$$

*Ans.*

To show that the magnitude of this vector is indeed 100 lb, apply Eq. 2–4,

$$\begin{aligned} F &= \sqrt{F_x^2 + F_y^2 + F_z^2} \\ &= \sqrt{(35.4)^2 + (35.4)^2 + (86.6)^2} = 100 \text{ lb} \end{aligned}$$

If needed, the coordinate direction angles of  $\mathbf{F}$  can be determined from the components of the unit vector acting in the direction of  $\mathbf{F}$ . Hence,

$$\begin{aligned} \mathbf{u} &= \frac{\mathbf{F}}{F} = \frac{F_x}{F}\mathbf{i} + \frac{F_y}{F}\mathbf{j} + \frac{F_z}{F}\mathbf{k} \\ &= \frac{35.4}{100}\mathbf{i} - \frac{35.4}{100}\mathbf{j} + \frac{86.6}{100}\mathbf{k} \\ &= 0.354\mathbf{i} - 0.354\mathbf{j} + 0.866\mathbf{k} \end{aligned}$$

so that

$$\alpha = \cos^{-1}(0.354) = 69.3^\circ$$

$$\beta = \cos^{-1}(-0.354) = 111^\circ$$

$$\gamma = \cos^{-1}(0.866) = 30.0^\circ$$

These results are shown in Fig. 2–30c.

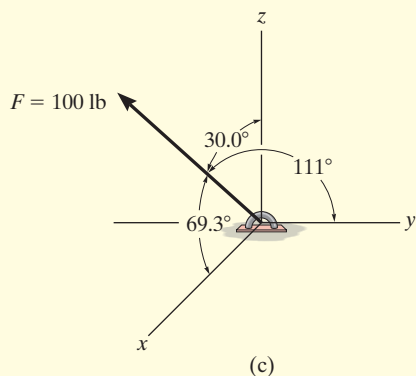
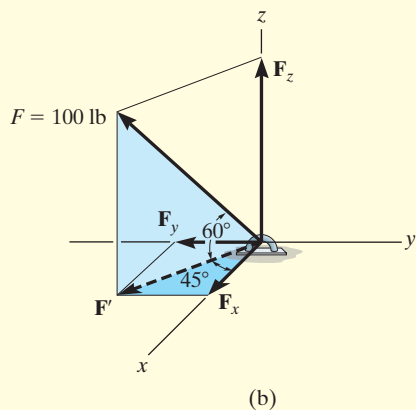
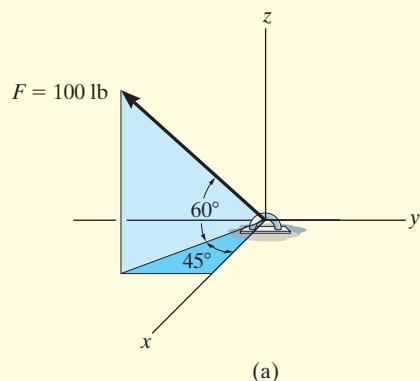


Fig. 2–30

**EXAMPLE 2.9**

Two forces act on the hook shown in Fig. 2–31*a*. Specify the magnitude of  $\mathbf{F}_2$  and its coordinate direction angles so that the resultant force  $\mathbf{F}_R$  acts along the positive  $y$  axis and has a magnitude of 800 N.

**SOLUTION**

To solve this problem, the resultant force  $\mathbf{F}_R$  and its two components,  $\mathbf{F}_1$  and  $\mathbf{F}_2$ , will each be expressed in Cartesian vector form. Then, as shown in Fig. 2–31*b*, it is necessary that  $\mathbf{F}_R = \mathbf{F}_1 + \mathbf{F}_2$ .

Applying Eq. 2–9,

$$\begin{aligned}\mathbf{F}_1 &= F_1 \cos \alpha_1 \mathbf{i} + F_1 \cos \beta_1 \mathbf{j} + F_1 \cos \gamma_1 \mathbf{k} \\ &= 300 \cos 45^\circ \mathbf{i} + 300 \cos 60^\circ \mathbf{j} + 300 \cos 120^\circ \mathbf{k} \\ &= \{212.1\mathbf{i} + 150\mathbf{j} - 150\mathbf{k}\} \text{ N} \\ \mathbf{F}_2 &= F_{2x}\mathbf{i} + F_{2y}\mathbf{j} + F_{2z}\mathbf{k}\end{aligned}$$

Since  $\mathbf{F}_R$  has a magnitude of 800 N and acts in the  $+\mathbf{j}$  direction,

$$\mathbf{F}_R = (800 \text{ N})(+\mathbf{j}) = \{800\mathbf{j}\} \text{ N}$$

We require

$$\begin{aligned}\mathbf{F}_R &= \mathbf{F}_1 + \mathbf{F}_2 \\ 800\mathbf{j} &= 212.1\mathbf{i} + 150\mathbf{j} - 150\mathbf{k} + F_{2x}\mathbf{i} + F_{2y}\mathbf{j} + F_{2z}\mathbf{k} \\ 800\mathbf{j} &= (212.1 + F_{2x})\mathbf{i} + (150 + F_{2y})\mathbf{j} + (-150 + F_{2z})\mathbf{k}\end{aligned}$$

To satisfy this equation the  $\mathbf{i}$ ,  $\mathbf{j}$ ,  $\mathbf{k}$  components of  $\mathbf{F}_R$  must be equal to the corresponding  $\mathbf{i}$ ,  $\mathbf{j}$ ,  $\mathbf{k}$  components of  $(\mathbf{F}_1 + \mathbf{F}_2)$ . Hence,

$$\begin{aligned}0 &= 212.1 + F_{2x} & F_{2x} &= -212.1 \text{ N} \\ 800 &= 150 + F_{2y} & F_{2y} &= 650 \text{ N} \\ 0 &= -150 + F_{2z} & F_{2z} &= 150 \text{ N}\end{aligned}$$

The magnitude of  $\mathbf{F}_2$  is thus

$$\begin{aligned}F_2 &= \sqrt{(-212.1 \text{ N})^2 + (650 \text{ N})^2 + (150 \text{ N})^2} \\ &= 700 \text{ N}\end{aligned}$$

*Ans.*

We can use Eq. 2–9 to determine  $\alpha_2, \beta_2, \gamma_2$ .

$$\cos \alpha_2 = \frac{-212.1}{700}; \quad \alpha_2 = 108^\circ$$

*Ans.*

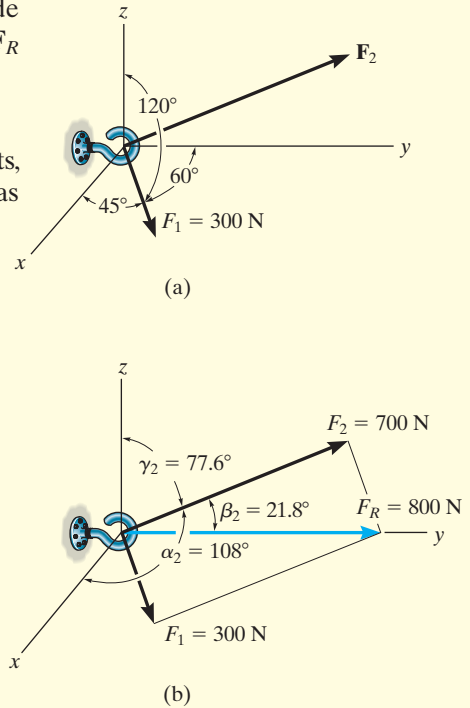
$$\cos \beta_2 = \frac{650}{700}; \quad \beta_2 = 21.8^\circ$$

*Ans.*

$$\cos \gamma_2 = \frac{150}{700}; \quad \gamma_2 = 77.6^\circ$$

*Ans.*

These results are shown in Fig. 2–31*b*.



**Fig. 2–31**

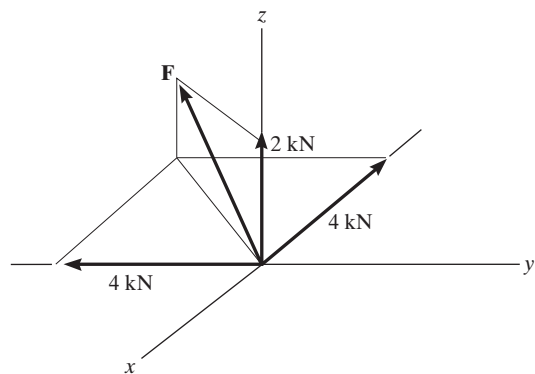
## PRELIMINARY PROBLEMS

**P2-3.** Sketch the following forces on the  $x, y, z$  coordinate axes. Show  $\alpha, \beta, \gamma$ .

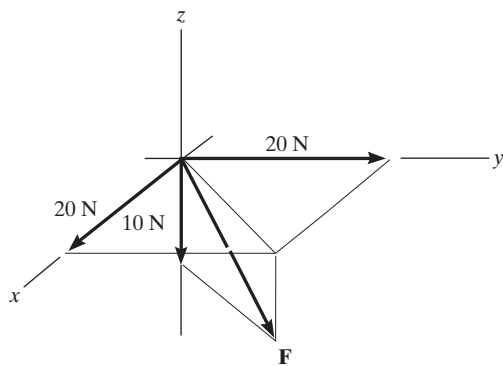
a)  $\mathbf{F} = \{50\mathbf{i} + 60\mathbf{j} - 10\mathbf{k}\}$  kN

b)  $\mathbf{F} = \{-40\mathbf{i} - 80\mathbf{j} + 60\mathbf{k}\}$  kN

**P2-4.** In each case, establish  $\mathbf{F}$  as a Cartesian vector, and find the magnitude of  $\mathbf{F}$  and the direction cosine of  $\beta$ .



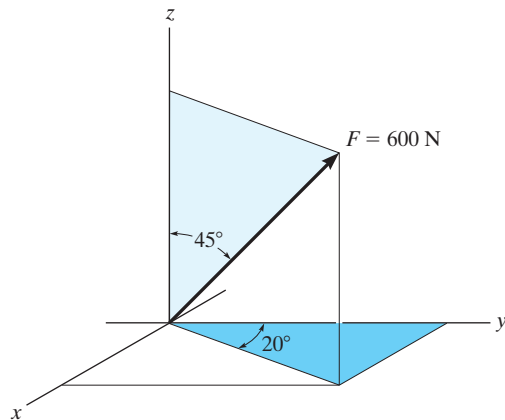
(a)



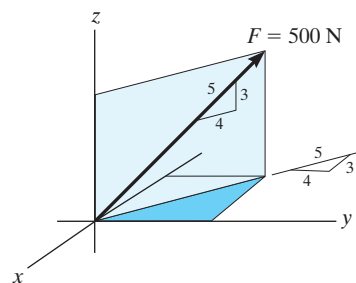
(b)

**Prob. P2-4**

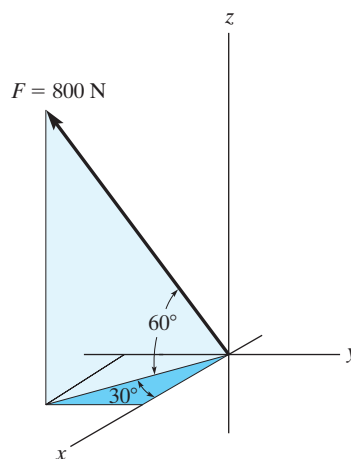
**P2-5.** Show how to resolve each force into its  $x, y, z$  components. Set up the calculation used to find the magnitude of each component.



(a)



(b)

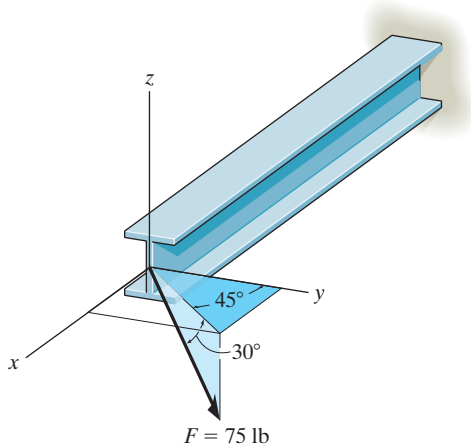


(c)

**Prob. P2-5**

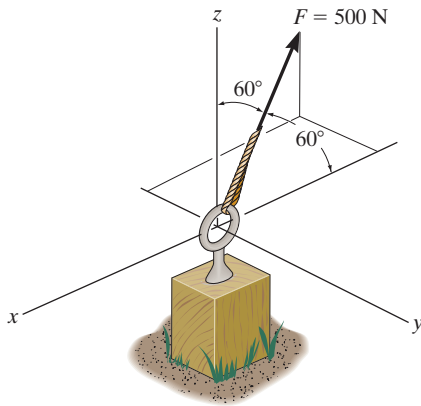
## FUNDAMENTAL PROBLEMS

**F2-13.** Determine the coordinate direction angles of the force.



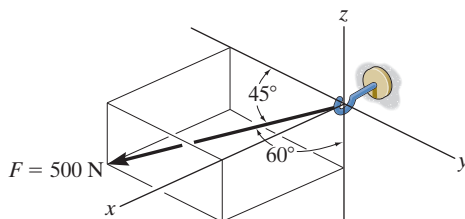
**Prob. F2-13**

**F2-14.** Express the force as a Cartesian vector.



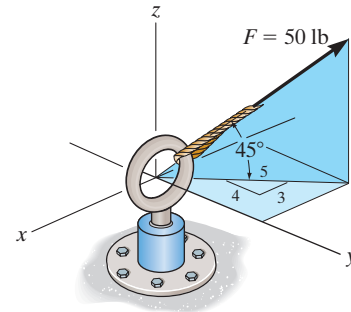
**Prob. F2-14**

**F2-15.** Express the force as a Cartesian vector.



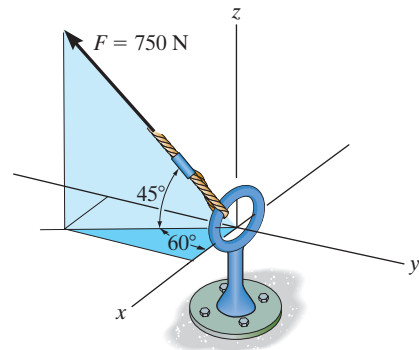
**Prob. F2-15**

**F2-16.** Express the force as a Cartesian vector.



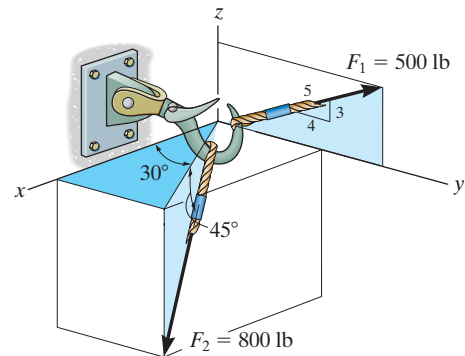
**Prob. F2-16**

**F2-17.** Express the force as a Cartesian vector.



**Prob. F2-17**

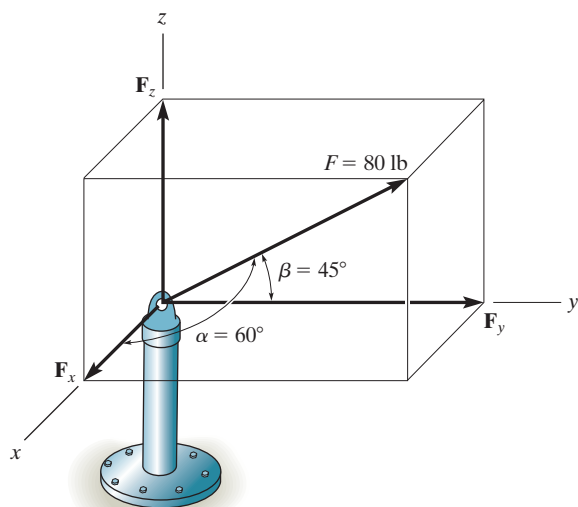
**F2-18.** Determine the resultant force acting on the hook.



**Prob. F2-18**

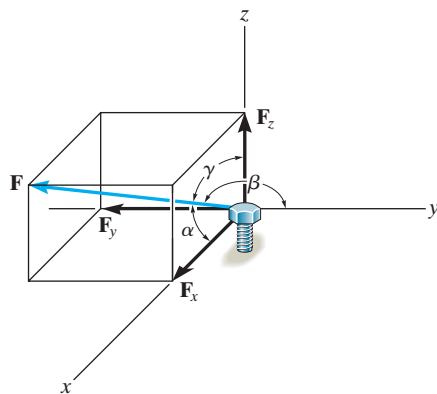
## PROBLEMS

**\*2-60.** The force  $\mathbf{F}$  has a magnitude of 80 lb and acts within the octant shown. Determine the magnitudes of the  $x$ ,  $y$ ,  $z$  components of  $\mathbf{F}$ .



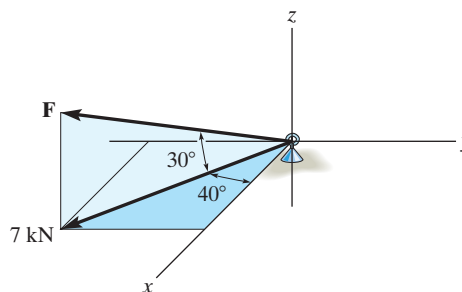
**Prob. 2-60**

**2-61.** The bolt is subjected to the force  $\mathbf{F}$ , which has components acting along the  $x$ ,  $y$ ,  $z$  axes as shown. If the magnitude of  $\mathbf{F}$  is 80 N, and  $\alpha = 60^\circ$  and  $\gamma = 45^\circ$ , determine the magnitudes of its components.



**Prob. 2-61**

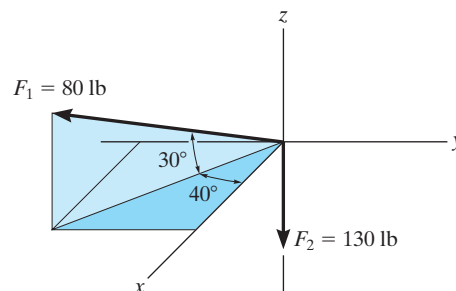
**2-62.** Determine the magnitude and coordinate direction angles of the force  $\mathbf{F}$  acting on the support. The component of  $\mathbf{F}$  in the  $x$ - $y$  plane is 7 kN.



**Prob. 2-62**

**2-63.** Determine the magnitude and coordinate direction angles of the resultant force and sketch this vector on the coordinate system.

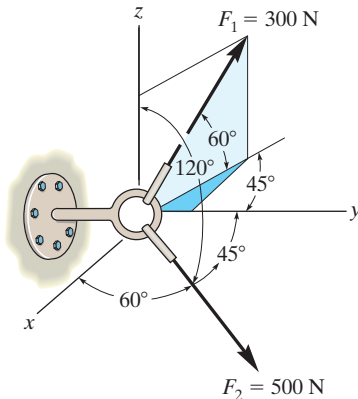
**\*2-64.** Specify the coordinate direction angles of  $\mathbf{F}_1$  and  $\mathbf{F}_2$  and express each force as a Cartesian vector.



**Probs. 2-63/64**

**2–65.** The screw eye is subjected to the two forces shown. Express each force in Cartesian vector form and then determine the resultant force. Find the magnitude and coordinate direction angles of the resultant force.

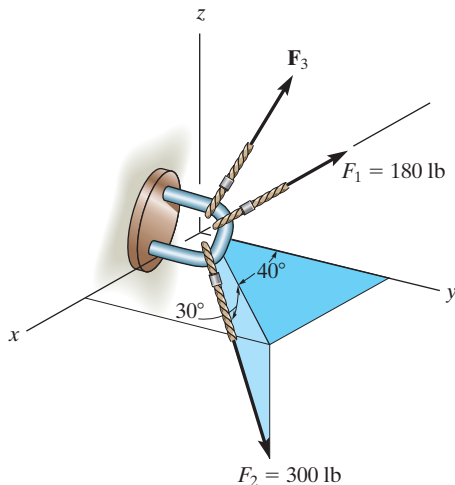
**2–66.** Determine the coordinate direction angles of  $\mathbf{F}_1$ .



**Probs. 2–65/66**

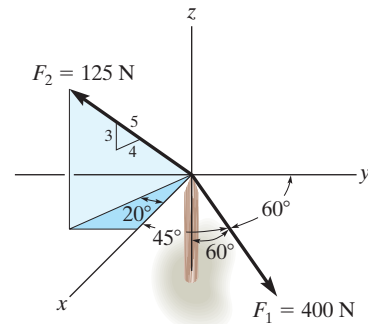
**2–67.** Determine the magnitude and coordinate direction angles of  $\mathbf{F}_3$  so that the resultant of the three forces acts along the positive  $y$  axis and has a magnitude of 600 lb.

**\*2–68.** Determine the magnitude and coordinate direction angles of  $\mathbf{F}_3$  so that the resultant of the three forces is zero.



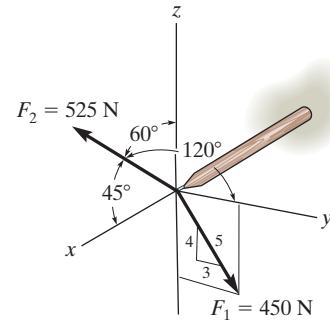
**Probs. 2–67/68**

**2–69.** Determine the magnitude and coordinate direction angles of the resultant force, and sketch this vector on the coordinate system.



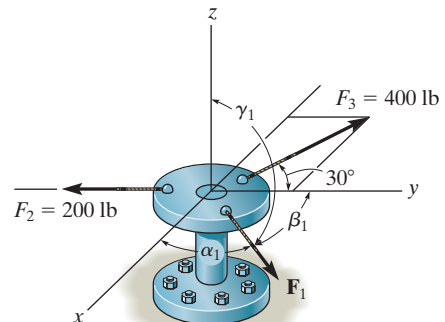
**Prob. 2–69**

**2–70.** Determine the magnitude and coordinate direction angles of the resultant force, and sketch this vector on the coordinate system.



**Prob. 2–70**

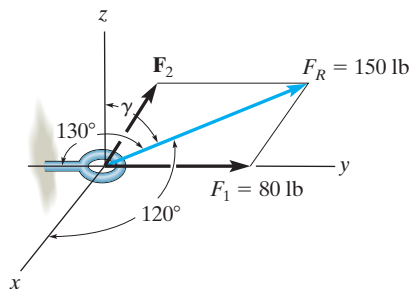
**2–71.** Specify the magnitude and coordinate direction angles  $\alpha_1$ ,  $\beta_1$ ,  $\gamma_1$  of  $\mathbf{F}_1$  so that the resultant of the three forces acting on the bracket is  $\mathbf{F}_R = \{-350\mathbf{k}\}$  lb. Note that  $\mathbf{F}_3$  lies in the  $x$ - $y$  plane.



**Prob. 2–71**



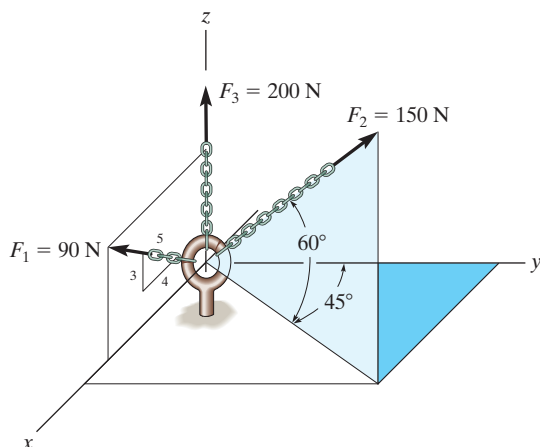
**\*2-72.** Two forces  $\mathbf{F}_1$  and  $\mathbf{F}_2$  act on the screw eye. If the resultant force  $\mathbf{F}_R$  has a magnitude of 150 lb and the coordinate direction angles shown, determine the magnitude of  $\mathbf{F}_2$  and its coordinate direction angles.



**Prob. 2-72**

**2-73.** Express each force in Cartesian vector form.

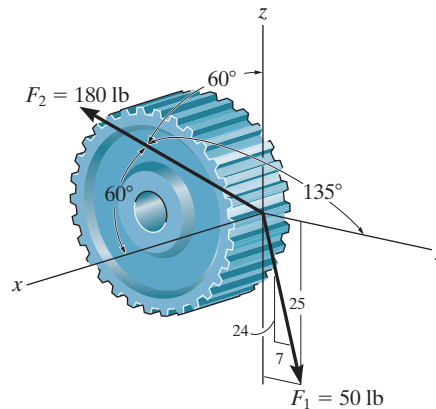
**2-74.** Determine the magnitude and coordinate direction angles of the resultant force, and sketch this vector on the coordinate system.



**Probs. 2-73/74**

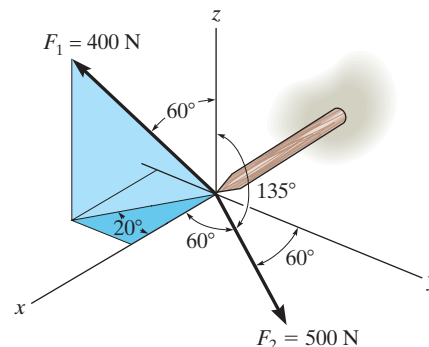
**2-75.** The spur gear is subjected to the two forces caused by contact with other gears. Express each force as a Cartesian vector.

**\*2-76.** The spur gear is subjected to the two forces caused by contact with other gears. Determine the resultant of the two forces and express the result as a Cartesian vector.



**Probs. 2-75/76**

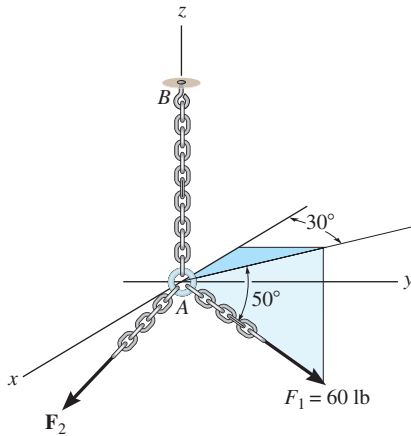
**2-77.** Determine the magnitude and coordinate direction angles of the resultant force, and sketch this vector on the coordinate system.



**Prob. 2-77**

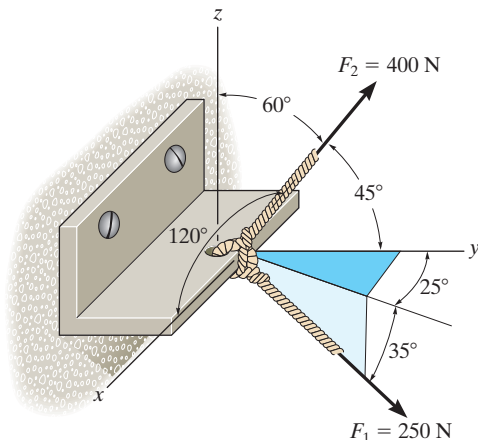
**2-78.** The two forces  $\mathbf{F}_1$  and  $\mathbf{F}_2$  acting at  $A$  have a resultant force of  $\mathbf{F}_R = \{-100\mathbf{k}\}$  lb. Determine the magnitude and coordinate direction angles of  $\mathbf{F}_2$ .

**2-79.** Determine the coordinate direction angles of the force  $\mathbf{F}_1$  and indicate them on the figure.



**Probs. 2-78/79**

**\*2-80.** The bracket is subjected to the two forces shown. Express each force in Cartesian vector form and then determine the resultant force  $\mathbf{F}_R$ . Find the magnitude and coordinate direction angles of the resultant force.

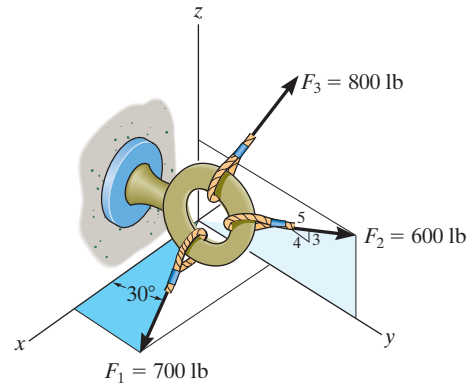


**Prob. 2-80**

**2-81.** If the coordinate direction angles for  $\mathbf{F}_3$  are  $\alpha_3 = 120^\circ$ ,  $\beta_3 = 60^\circ$  and  $\gamma_3 = 45^\circ$ , determine the magnitude and coordinate direction angles of the resultant force acting on the eyebolt.

**2-82.** If the coordinate direction angles for  $\mathbf{F}_3$  are  $\alpha_3 = 120^\circ$ ,  $\beta_3 = 45^\circ$ , and  $\gamma_3 = 60^\circ$ , determine the magnitude and coordinate direction angles of the resultant force acting on the eyebolt.

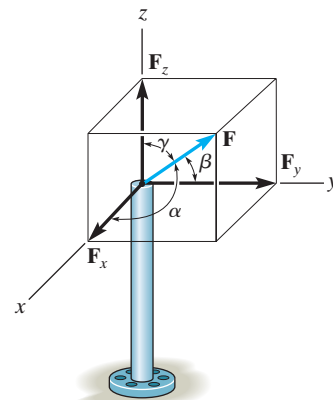
**2-83.** If the direction of the resultant force acting on the eyebolt is defined by the unit vector  $\mathbf{u}_{F_R} = \cos 30^\circ \mathbf{j} + \sin 30^\circ \mathbf{k}$ , determine the coordinate direction angles of  $\mathbf{F}_3$  and the magnitude of  $\mathbf{F}_R$ .



**Probs. 2-81/82/83**

**\*2-84.** The pole is subjected to the force  $\mathbf{F}$ , which has components acting along the  $x$ ,  $y$ ,  $z$  axes as shown. If the magnitude of  $\mathbf{F}$  is 3 kN,  $\beta = 30^\circ$ , and  $\gamma = 75^\circ$ , determine the magnitudes of its three components.

**2-85.** The pole is subjected to the force  $\mathbf{F}$  which has components  $F_x = 1.5$  kN and  $F_z = 1.25$  kN. If  $\beta = 75^\circ$ , determine the magnitudes of  $\mathbf{F}$  and  $\mathbf{F}_y$ .



**Probs. 2-84/85**

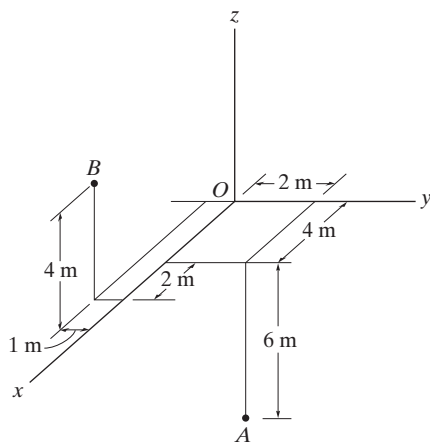


Fig. 2-32

## 2.7 Position Vectors

In this section we will introduce the concept of a position vector. It will be shown that this vector is of importance in formulating a Cartesian force vector directed between two points in space.

**x, y, z Coordinates.** Throughout the book we will use a *right-handed* coordinate system to reference the location of points in space. We will also use the convention followed in many technical books, which requires the positive  $z$  axis to be directed *upward* (the zenith direction) so that it measures the height of an object or the altitude of a point. The  $x, y$  axes then lie in the horizontal plane, Fig. 2-32. Points in space are located relative to the origin of coordinates,  $O$ , by successive measurements along the  $x, y, z$  axes. For example, the coordinates of point  $A$  are obtained by starting at  $O$  and measuring  $x_A = +4$  m along the  $x$  axis, then  $y_A = +2$  m along the  $y$  axis, and finally  $z_A = -6$  m along the  $z$  axis, so that  $A(4 \text{ m}, 2 \text{ m}, -6 \text{ m})$ . In a similar manner, measurements along the  $x, y, z$  axes from  $O$  to  $B$  yield the coordinates of  $B$ , that is,  $B(6 \text{ m}, -1 \text{ m}, 4 \text{ m})$ .

**Position Vector.** A *position vector*  $\mathbf{r}$  is defined as a fixed vector which locates a point in space relative to another point. For example, if  $\mathbf{r}$  extends from the origin of coordinates,  $O$ , to point  $P(x, y, z)$ , Fig. 2-33a, then  $\mathbf{r}$  can be expressed in Cartesian vector form as

$$\mathbf{r} = x\mathbf{i} + y\mathbf{j} + z\mathbf{k}$$

Note how the head-to-tail vector addition of the three components yields vector  $\mathbf{r}$ , Fig. 2-33b. Starting at the origin  $O$ , one “travels”  $x$  in the  $+\mathbf{i}$  direction, then  $y$  in the  $+\mathbf{j}$  direction, and finally  $z$  in the  $+\mathbf{k}$  direction to arrive at point  $P(x, y, z)$ .

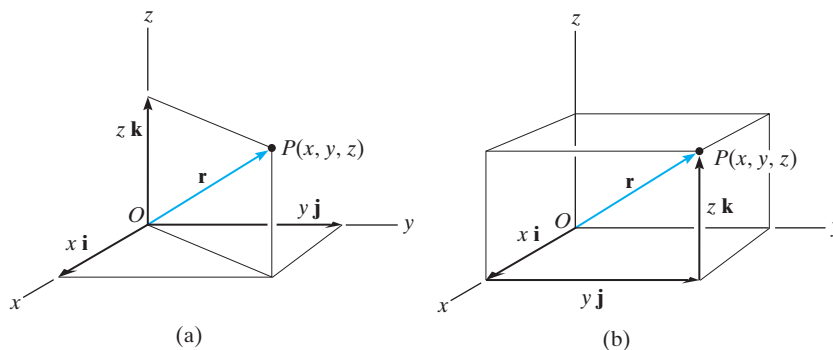


Fig. 2-33

In the more general case, the position vector may be directed from point  $A$  to point  $B$  in space, Fig. 2–34a. This vector is also designated by the symbol  $\mathbf{r}$ . As a matter of convention, we will *sometimes* refer to this vector with *two subscripts* to indicate from and to the point where it is directed. Thus,  $\mathbf{r}$  can also be designated as  $\mathbf{r}_{AB}$ . Also, note that  $\mathbf{r}_A$  and  $\mathbf{r}_B$  in Fig. 2–34a are referenced with only one subscript since they extend from the origin of coordinates.

From Fig. 2–34a, by the head-to-tail vector addition, using the triangle rule, we require

$$\mathbf{r}_A + \mathbf{r} = \mathbf{r}_B$$

Solving for  $\mathbf{r}$  and expressing  $\mathbf{r}_A$  and  $\mathbf{r}_B$  in Cartesian vector form yields

$$\mathbf{r} = \mathbf{r}_B - \mathbf{r}_A = (x_B\mathbf{i} + y_B\mathbf{j} + z_B\mathbf{k}) - (x_A\mathbf{i} + y_A\mathbf{j} + z_A\mathbf{k})$$

or

$$\mathbf{r} = (x_B - x_A)\mathbf{i} + (y_B - y_A)\mathbf{j} + (z_B - z_A)\mathbf{k} \quad (2-11)$$

Thus, the  $\mathbf{i}$ ,  $\mathbf{j}$ ,  $\mathbf{k}$  components of the position vector  $\mathbf{r}$  may be formed by taking the coordinates of the tail of the vector  $A(x_A, y_A, z_A)$  and subtracting them from the corresponding coordinates of the head  $B(x_B, y_B, z_B)$ . We can also form these components *directly*, Fig. 2–34b, by starting at  $A$  and moving through a distance of  $(x_B - x_A)$  along the positive  $x$  axis ( $+\mathbf{i}$ ), then  $(y_B - y_A)$  along the positive  $y$  axis ( $+\mathbf{j}$ ), and finally  $(z_B - z_A)$  along the positive  $z$  axis ( $+\mathbf{k}$ ) to get to  $B$ .

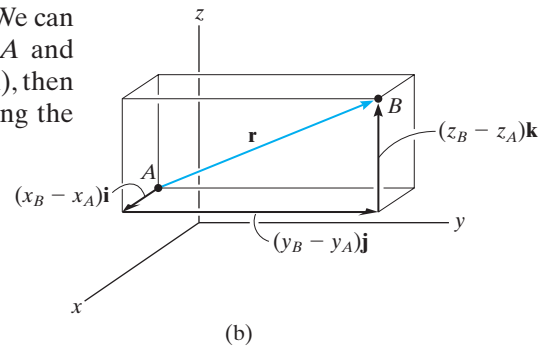
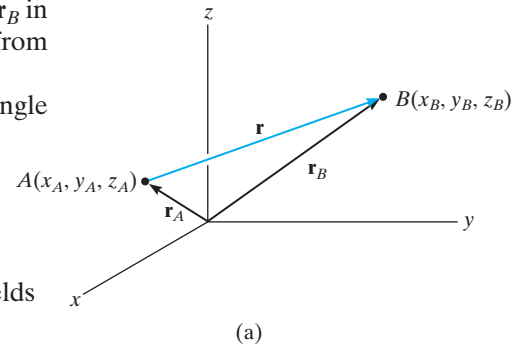
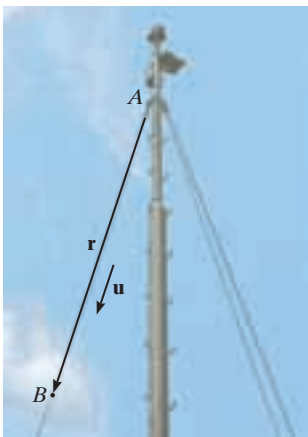
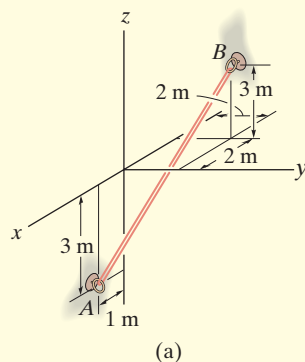


Fig. 2–34

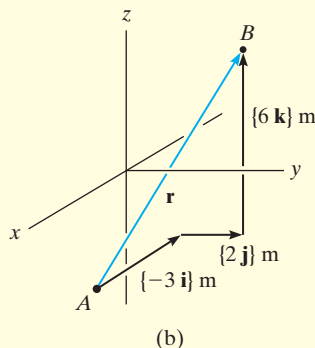


If an  $x, y, z$  coordinate system is established, then the coordinates of two points  $A$  and  $B$  on the cable can be determined. From this the position vector  $\mathbf{r}$  acting along the cable can be formulated. Its magnitude represents the distance from  $A$  to  $B$ , and its unit vector,  $\mathbf{u} = \mathbf{r}/r$ , gives the direction defined by  $\alpha, \beta, \gamma$ .  
(© Russell C. Hibbeler)

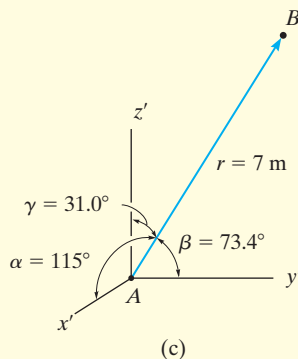
## EXAMPLE 2.10



(a)



(b)



(c)

Fig. 2-35

An elastic rubber band is attached to points  $A$  and  $B$  as shown in Fig. 2-35a. Determine its length and its direction measured from  $A$  toward  $B$ .

## SOLUTION

We first establish a position vector from  $A$  to  $B$ , Fig. 2-35b. In accordance with Eq. 2-11, the coordinates of the tail  $A(1 \text{ m}, 0, -3 \text{ m})$  are subtracted from the coordinates of the head  $B(-2 \text{ m}, 2 \text{ m}, 3 \text{ m})$ , which yields

$$\begin{aligned}\mathbf{r} &= [-2 \text{ m} - 1 \text{ m}]\mathbf{i} + [2 \text{ m} - 0]\mathbf{j} + [3 \text{ m} - (-3 \text{ m})]\mathbf{k} \\ &= \{-3\mathbf{i} + 2\mathbf{j} + 6\mathbf{k}\} \text{ m}\end{aligned}$$

These components of  $\mathbf{r}$  can also be determined *directly* by realizing that they represent the direction and distance one must travel along each axis in order to move from  $A$  to  $B$ , i.e., along the  $x$  axis  $\{-3\mathbf{i}\} \text{ m}$ , along the  $y$  axis  $\{2\mathbf{j}\} \text{ m}$ , and finally along the  $z$  axis  $\{6\mathbf{k}\} \text{ m}$ .

The length of the rubber band is therefore

$$r = \sqrt{(-3 \text{ m})^2 + (2 \text{ m})^2 + (6 \text{ m})^2} = 7 \text{ m} \quad \text{Ans.}$$

Formulating a unit vector in the direction of  $\mathbf{r}$ , we have

$$\mathbf{u} = \frac{\mathbf{r}}{r} = -\frac{3}{7}\mathbf{i} + \frac{2}{7}\mathbf{j} + \frac{6}{7}\mathbf{k}$$

The components of this unit vector give the coordinate direction angles

$$\alpha = \cos^{-1}\left(-\frac{3}{7}\right) = 115^\circ \quad \text{Ans.}$$

$$\beta = \cos^{-1}\left(\frac{2}{7}\right) = 73.4^\circ \quad \text{Ans.}$$

$$\gamma = \cos^{-1}\left(\frac{6}{7}\right) = 31.0^\circ \quad \text{Ans.}$$

**NOTE:** These angles are measured from the *positive axes* of a localized coordinate system placed at the tail of  $\mathbf{r}$ , as shown in Fig. 2-35c.

## 2.8 Force Vector Directed Along a Line

Quite often in three-dimensional statics problems, the direction of a force is specified by two points through which its line of action passes. Such a situation is shown in Fig. 2–36, where the force  $\mathbf{F}$  is directed along the cord  $AB$ . We can formulate  $\mathbf{F}$  as a Cartesian vector by realizing that it has the *same direction and sense* as the position vector  $\mathbf{r}$  directed from point  $A$  to point  $B$  on the cord. This common direction is specified by the **unit vector**  $\mathbf{u} = \mathbf{r}/r$ . Hence,

$$\mathbf{F} = F\mathbf{u} = F\left(\frac{\mathbf{r}}{r}\right) = F\left(\frac{(x_B - x_A)\mathbf{i} + (y_B - y_A)\mathbf{j} + (z_B - z_A)\mathbf{k}}{\sqrt{(x_B - x_A)^2 + (y_B - y_A)^2 + (z_B - z_A)^2}}\right)$$

Although we have represented  $\mathbf{F}$  symbolically in Fig. 2–36, note that it has *units of force*, unlike  $\mathbf{r}$ , which has units of length.

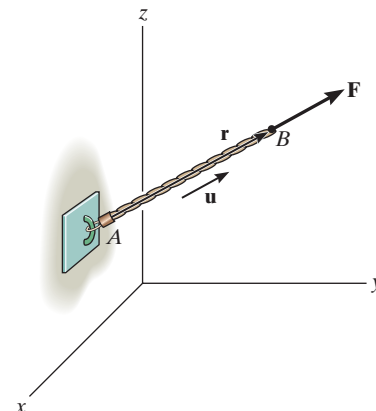
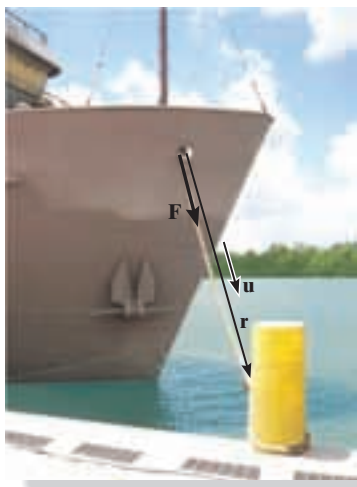


Fig. 2–36



The force  $\mathbf{F}$  acting along the rope can be represented as a Cartesian vector by establishing  $x, y, z$  axes and first forming a position vector  $\mathbf{r}$  along the length of the rope. Then the corresponding unit vector  $\mathbf{u} = \mathbf{r}/r$  that defines the direction of both the rope and the force can be determined. Finally, the magnitude of the force is combined with its direction,  $\mathbf{F} = F\mathbf{u}$ . (© Russell C. Hibbeler)

### Important Points

- A position vector locates one point in space relative to another point.
- The easiest way to formulate the components of a position vector is to determine the distance and direction that must be traveled along the  $x, y, z$  directions—going from the tail to the head of the vector.
- A force  $\mathbf{F}$  acting in the direction of a position vector  $\mathbf{r}$  can be represented in Cartesian form if the unit vector  $\mathbf{u}$  of the position vector is determined and it is multiplied by the magnitude of the force, i.e.,  $\mathbf{F} = F\mathbf{u} = F(\mathbf{r}/r)$ .

## EXAMPLE 2.11

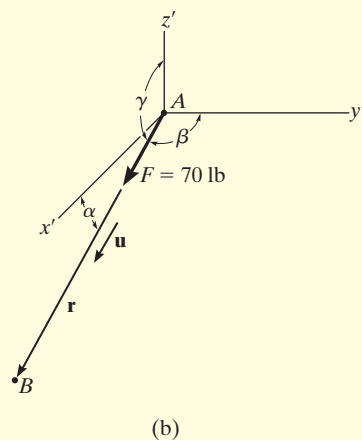
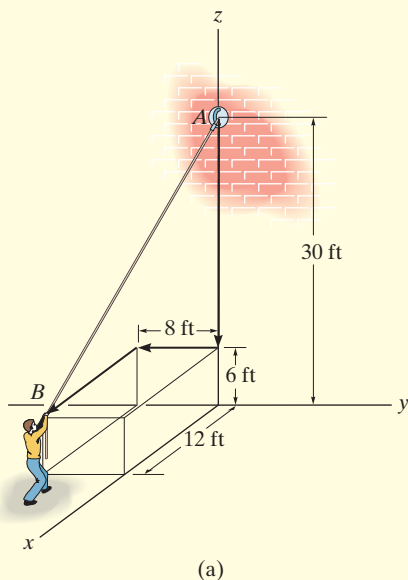


Fig. 2-37

The man shown in Fig. 2-37a pulls on the cord with a force of 70 lb. Represent this force acting on the support  $A$  as a Cartesian vector and determine its direction.

## SOLUTION

Force  $\mathbf{F}$  is shown in Fig. 2-37b. The *direction* of this vector,  $\mathbf{u}$ , is determined from the position vector  $\mathbf{r}$ , which extends from  $A$  to  $B$ . Rather than using the coordinates of the end points of the cord,  $\mathbf{r}$  can be determined *directly* by noting in Fig. 2-37a that one must travel from  $A$   $\{-24\mathbf{k}\}$  ft, then  $\{-8\mathbf{j}\}$  ft, and finally  $\{12\mathbf{i}\}$  ft to get to  $B$ . Thus,

$$\mathbf{r} = \{12\mathbf{i} - 8\mathbf{j} - 24\mathbf{k}\} \text{ ft}$$

The magnitude of  $\mathbf{r}$ , which represents the *length* of cord  $AB$ , is

$$r = \sqrt{(12 \text{ ft})^2 + (-8 \text{ ft})^2 + (-24 \text{ ft})^2} = 28 \text{ ft}$$

Forming the unit vector that defines the direction and sense of both  $\mathbf{r}$  and  $\mathbf{F}$ , we have

$$\mathbf{u} = \frac{\mathbf{r}}{r} = \frac{12}{28}\mathbf{i} - \frac{8}{28}\mathbf{j} - \frac{24}{28}\mathbf{k}$$

Since  $\mathbf{F}$  has a *magnitude* of 70 lb and a *direction* specified by  $\mathbf{u}$ , then

$$\begin{aligned} \mathbf{F} &= F\mathbf{u} = 70 \text{ lb} \left( \frac{12}{28}\mathbf{i} - \frac{8}{28}\mathbf{j} - \frac{24}{28}\mathbf{k} \right) \\ &= \{30\mathbf{i} - 20\mathbf{j} - 60\mathbf{k}\} \text{ lb} \end{aligned} \quad \text{Ans.}$$

The coordinate direction angles are measured between  $\mathbf{r}$  (or  $\mathbf{F}$ ) and the *positive axes* of a localized coordinate system with origin placed at  $A$ , Fig. 2-37b. From the components of the unit vector:

$$\alpha = \cos^{-1}\left(\frac{12}{28}\right) = 64.6^\circ \quad \text{Ans.}$$

$$\beta = \cos^{-1}\left(\frac{-8}{28}\right) = 107^\circ \quad \text{Ans.}$$

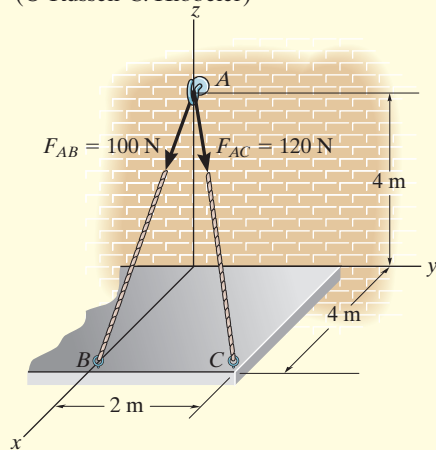
$$\gamma = \cos^{-1}\left(\frac{-24}{28}\right) = 149^\circ \quad \text{Ans.}$$

**NOTE:** These results make sense when compared with the angles identified in Fig. 2-37b.

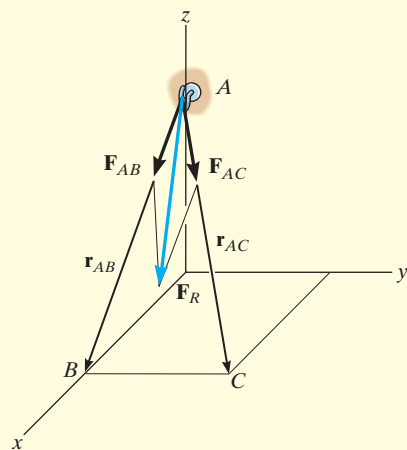
## EXAMPLE 2.12



(© Russell C. Hibbeler)



(a)



(b)

Fig. 2-38

The roof is supported by cables as shown in the photo. If the cables exert forces  $F_{AB} = 100 \text{ N}$  and  $F_{AC} = 120 \text{ N}$  on the wall hook at  $A$  as shown in Fig. 2-38a, determine the resultant force acting at  $A$ . Express the result as a Cartesian vector.

## SOLUTION

The resultant force  $\mathbf{F}_R$  is shown graphically in Fig. 2-38b. We can express this force as a Cartesian vector by first formulating  $\mathbf{F}_{AB}$  and  $\mathbf{F}_{AC}$  as Cartesian vectors and then adding their components. The directions of  $\mathbf{F}_{AB}$  and  $\mathbf{F}_{AC}$  are specified by forming unit vectors  $\mathbf{u}_{AB}$  and  $\mathbf{u}_{AC}$  along the cables. These unit vectors are obtained from the associated position vectors  $\mathbf{r}_{AB}$  and  $\mathbf{r}_{AC}$ . With reference to Fig. 2-38a, to go from  $A$  to  $B$ , we must travel  $\{-4\mathbf{k}\}$  m, and then  $\{4\mathbf{i}\}$  m. Thus,

$$\mathbf{r}_{AB} = \{4\mathbf{i} - 4\mathbf{k}\} \text{ m}$$

$$r_{AB} = \sqrt{(4 \text{ m})^2 + (-4 \text{ m})^2} = 5.66 \text{ m}$$

$$\mathbf{F}_{AB} = F_{AB} \left( \frac{\mathbf{r}_{AB}}{r_{AB}} \right) = (100 \text{ N}) \left( \frac{4}{5.66}\mathbf{i} - \frac{4}{5.66}\mathbf{k} \right)$$

$$\mathbf{F}_{AB} = \{70.7\mathbf{i} - 70.7\mathbf{k}\} \text{ N}$$

To go from  $A$  to  $C$ , we must travel  $\{-4\mathbf{k}\}$  m, then  $\{2\mathbf{j}\}$  m, and finally  $\{4\mathbf{i}\}$ . Thus,

$$\mathbf{r}_{AC} = \{4\mathbf{i} + 2\mathbf{j} - 4\mathbf{k}\} \text{ m}$$

$$r_{AC} = \sqrt{(4 \text{ m})^2 + (2 \text{ m})^2 + (-4 \text{ m})^2} = 6 \text{ m}$$

$$\mathbf{F}_{AC} = F_{AC} \left( \frac{\mathbf{r}_{AC}}{r_{AC}} \right) = (120 \text{ N}) \left( \frac{4}{6}\mathbf{i} + \frac{2}{6}\mathbf{j} - \frac{4}{6}\mathbf{k} \right)$$

$$= \{80\mathbf{i} + 40\mathbf{j} - 80\mathbf{k}\} \text{ N}$$

The resultant force is therefore

$$\mathbf{F}_R = \mathbf{F}_{AB} + \mathbf{F}_{AC} = \{70.7\mathbf{i} - 70.7\mathbf{k}\} \text{ N} + \{80\mathbf{i} + 40\mathbf{j} - 80\mathbf{k}\} \text{ N}$$

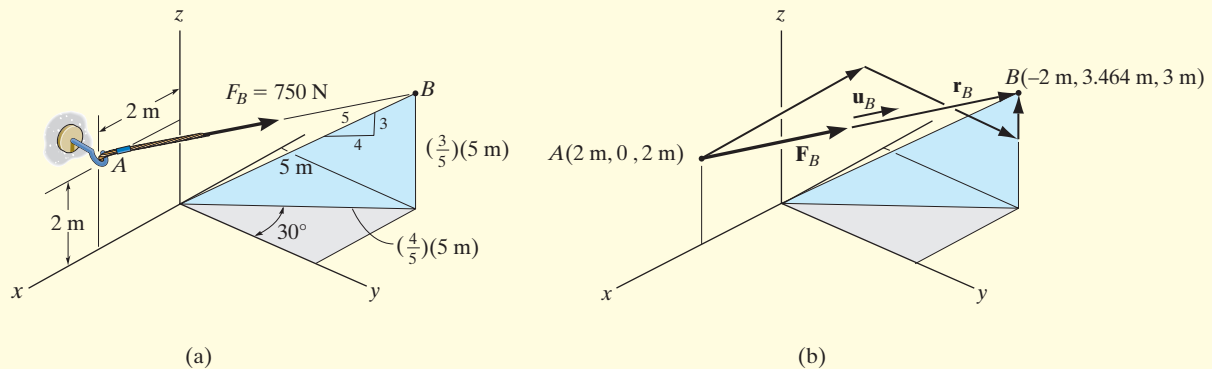
$$= \{151\mathbf{i} + 40\mathbf{j} - 151\mathbf{k}\} \text{ N}$$

Ans.



**EXAMPLE 2.13**

The force in Fig. 2–39*a* acts on the hook. Express it as a Cartesian vector.



**Fig. 2–39**

**SOLUTION**

As shown in Fig. 2–39*b*, the coordinates for points *A* and *B* are

$$A(2 \text{ m}, 0, 2 \text{ m})$$

and

$$B\left[-\left(\frac{4}{5}\right)5 \sin 30^\circ \text{ m}, \left(\frac{4}{5}\right)5 \cos 30^\circ \text{ m}, \left(\frac{3}{5}\right)5 \text{ m}\right]$$

or

$$B(-2 \text{ m}, 3.464 \text{ m}, 3 \text{ m})$$

Therefore, to go from *A* to *B*, one must travel  $\{-4\mathbf{i}\}$  m, then  $\{3.464\mathbf{j}\}$  m, and finally  $\{1\mathbf{k}\}$  m. Thus,

$$\begin{aligned} \mathbf{u}_B &= \left( \frac{\mathbf{r}_B}{r_B} \right) = \frac{\{-4\mathbf{i} + 3.464\mathbf{j} + 1\mathbf{k}\} \text{ m}}{\sqrt{(-4 \text{ m})^2 + (3.464 \text{ m})^2 + (1 \text{ m})^2}} \\ &= -0.7428\mathbf{i} + 0.6433\mathbf{j} + 0.1857\mathbf{k} \end{aligned}$$

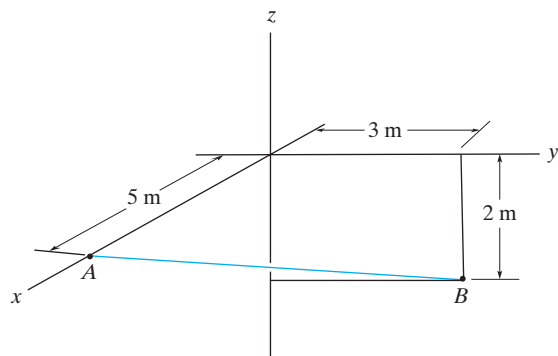
Force  $\mathbf{F}_B$  expressed as a Cartesian vector becomes

$$\begin{aligned} \mathbf{F}_B &= F_B \mathbf{u}_B = (750 \text{ N})(-0.7428\mathbf{i} + 0.6433\mathbf{j} + 0.1857\mathbf{k}) \\ &= \{-557\mathbf{i} + 482\mathbf{j} + 139\mathbf{k}\} \text{ N} \end{aligned}$$

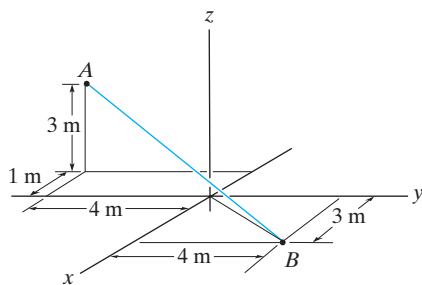
*Ans.*

## PRELIMINARY PROBLEMS

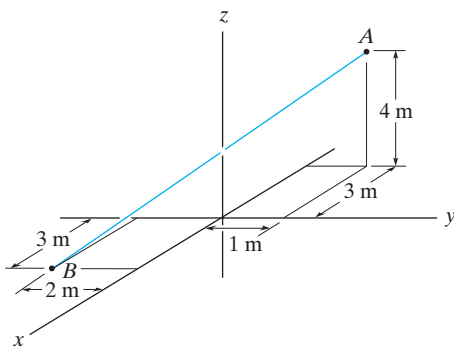
**P2-6.** In each case, establish a position vector from point  $A$  to point  $B$ .



(a)



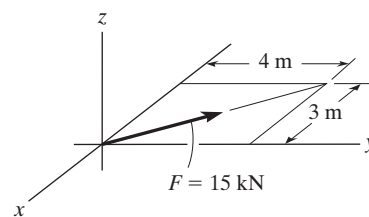
(b)



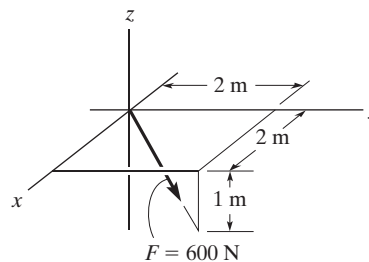
(c)

**Prob. P2-6**

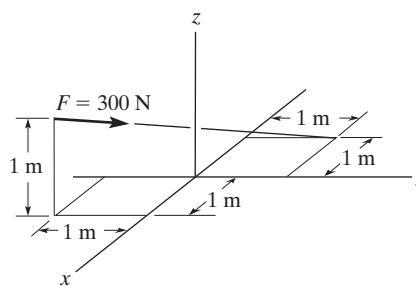
**P2-7.** In each case, express  $\mathbf{F}$  as a Cartesian vector.



(a)



(b)

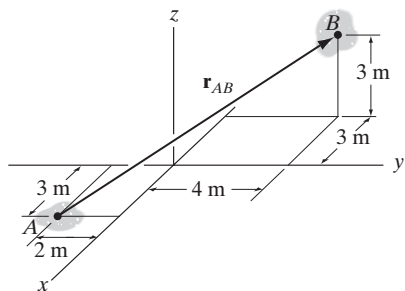


(c)

**Prob. P2-7**

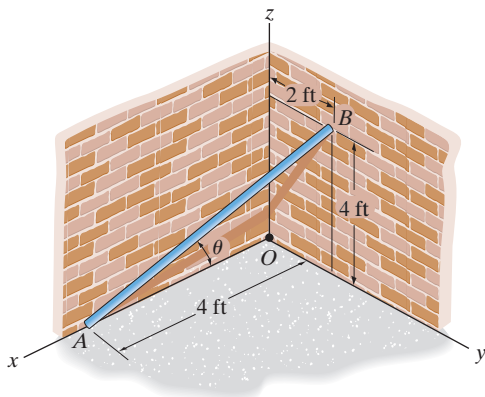
## FUNDAMENTAL PROBLEMS

**F2-19.** Express the position vector  $\mathbf{r}_{AB}$  in Cartesian vector form, then determine its magnitude and coordinate direction angles.



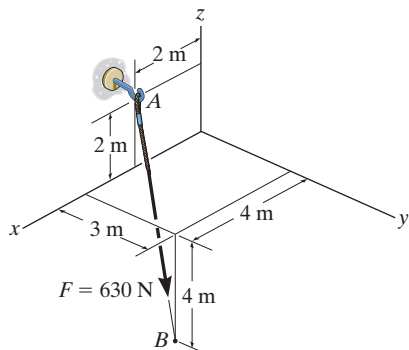
**Prob. F2-19**

**F2-20.** Determine the length of the rod and the position vector directed from A to B. What is the angle  $\theta$ ?



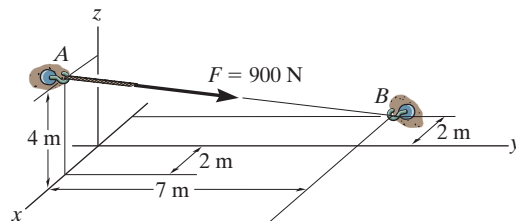
**Prob. F2-20**

**F2-21.** Express the force as a Cartesian vector.



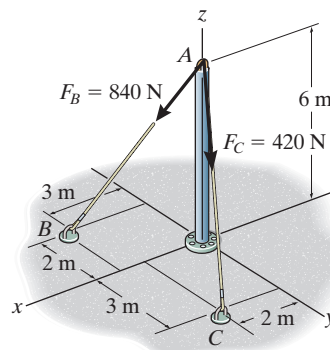
**Prob. F2-21**

**F2-22.** Express the force as a Cartesian vector.



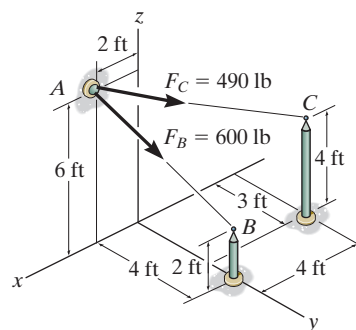
**Prob. F2-22**

**F2-23.** Determine the magnitude of the resultant force at A.



**Prob. F2-23**

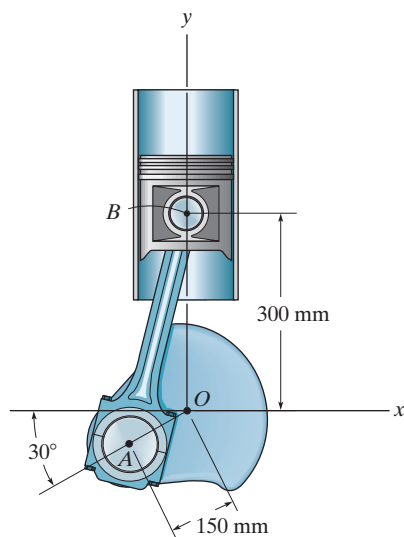
**F2-24.** Determine the resultant force at A.



**Prob. F2-24**

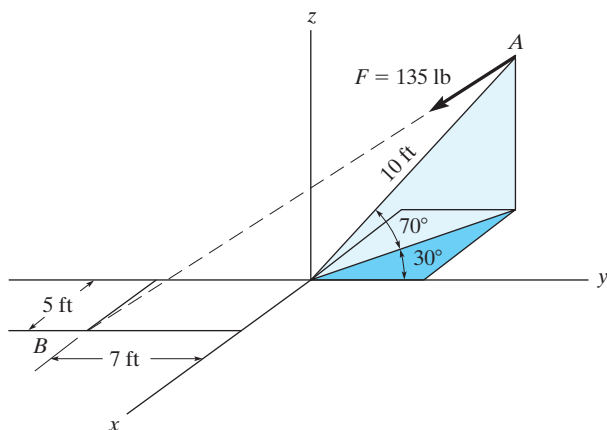
## PROBLEMS

**2-86.** Determine the length of the connecting rod  $AB$  by first formulating a Cartesian position vector from  $A$  to  $B$  and then determining its magnitude.



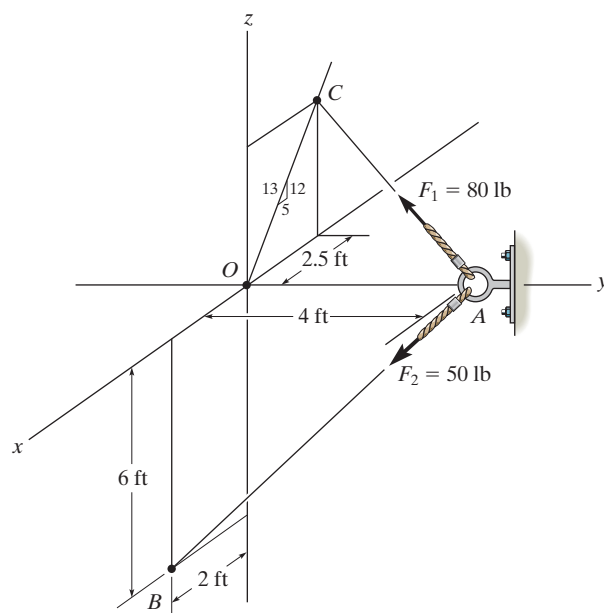
Prob. 2-86

**2-87.** Express force  $\mathbf{F}$  as a Cartesian vector; then determine its coordinate direction angles.



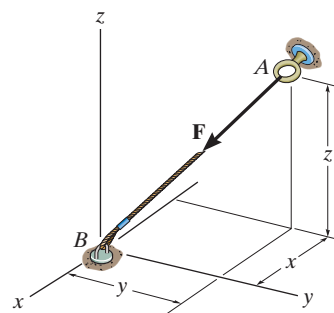
Prob. 2-87

**\*2-88.** Express each of the forces in Cartesian vector form and determine the magnitude and coordinate direction angles of the resultant force.



Prob. 2-88

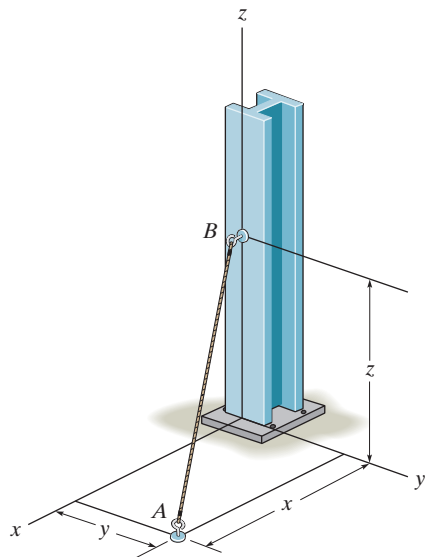
**2-89.** If  $\mathbf{F} = \{350\mathbf{i} - 250\mathbf{j} - 450\mathbf{k}\}$  N and cable  $AB$  is 9 m long, determine the  $x$ ,  $y$ ,  $z$  coordinates of point  $A$ .



Prob. 2-89

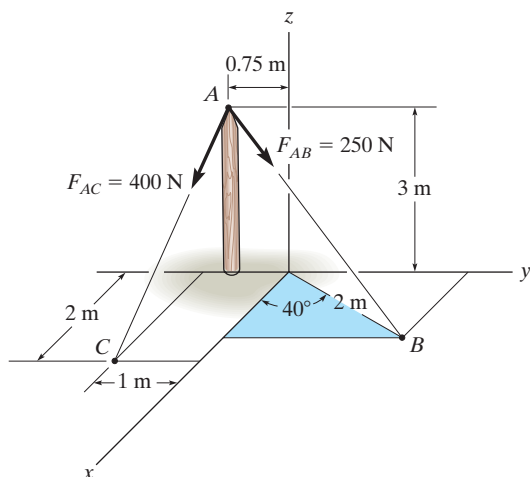
**2-90.** The 8-m-long cable is anchored to the ground at  $A$ . If  $x = 4$  m and  $y = 2$  m, determine the coordinate  $z$  to the highest point of attachment along the column.

**2-91.** The 8-m-long cable is anchored to the ground at  $A$ . If  $z = 5$  m, determine the location  $+x, +y$  of point  $A$ . Choose a value such that  $x = y$ .



**Probs. 2-90/91**

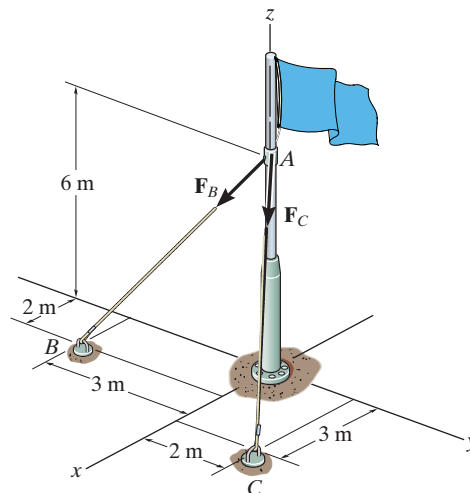
**\*2-92.** Express each of the forces in Cartesian vector form and determine the magnitude and coordinate direction angles of the resultant force.



**Prob. 2-92**

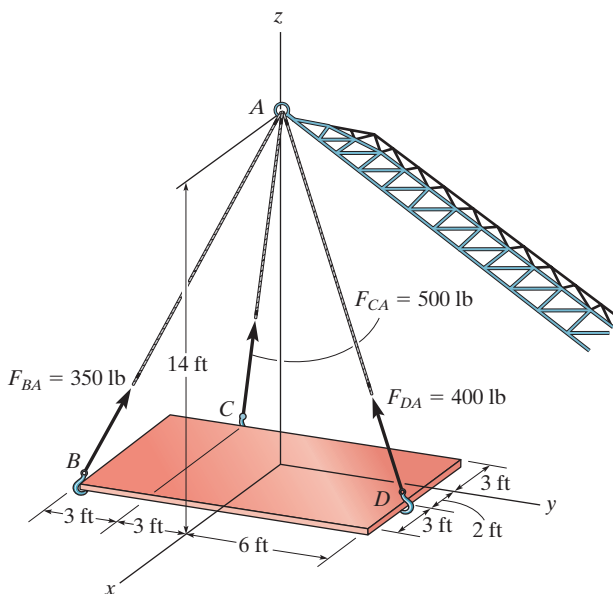
**2-93.** If  $F_B = 560$  N and  $F_C = 700$  N, determine the magnitude and coordinate direction angles of the resultant force acting on the flag pole.

**2-94.** If  $F_B = 700$  N, and  $F_C = 560$  N, determine the magnitude and coordinate direction angles of the resultant force acting on the flag pole.



**Probs. 2-93/94**

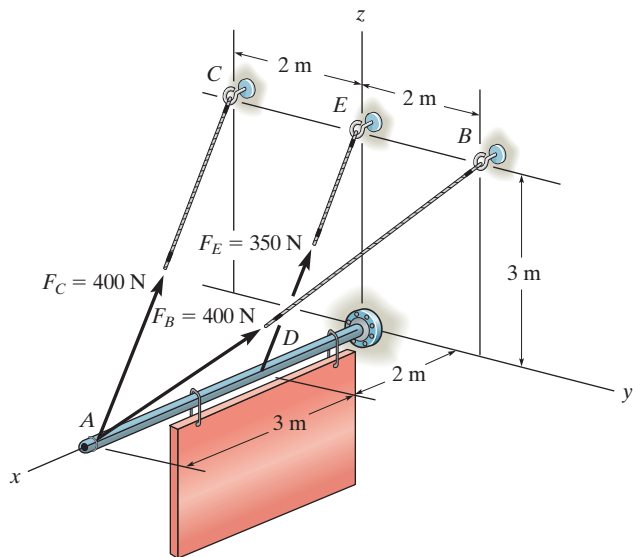
**2-95.** The plate is suspended using the three cables which exert the forces shown. Express each force as a Cartesian vector.



**Prob. 2-95**

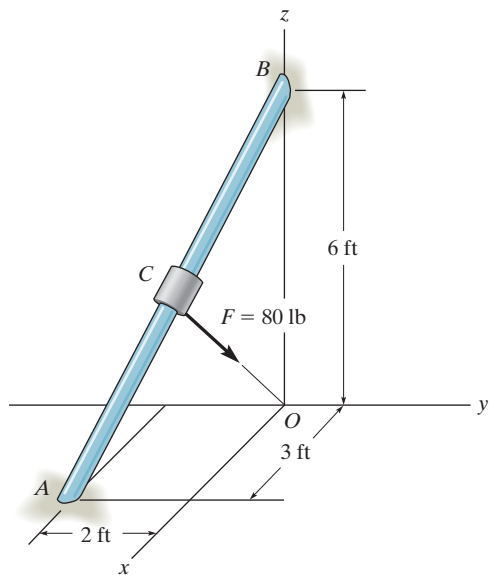
**\*2-96.** The three supporting cables exert the forces shown on the sign. Represent each force as a Cartesian vector.

**2-97.** Determine the magnitude and coordinate direction angles of the resultant force of the two forces acting on the sign at point  $A$ .



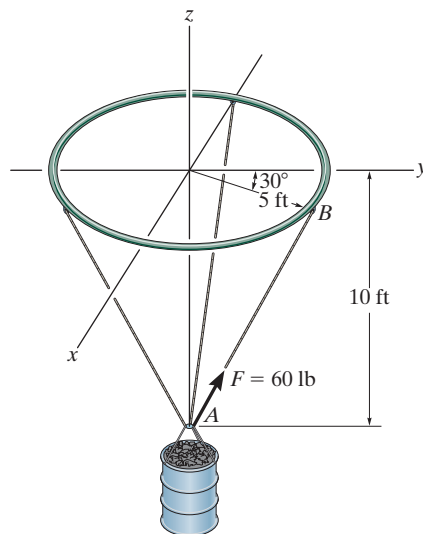
**Probs. 2-96/97**

**2-98.** The force  $\mathbf{F}$  has a magnitude of 80 lb and acts at the midpoint  $C$  of the thin rod. Express the force as a Cartesian vector.



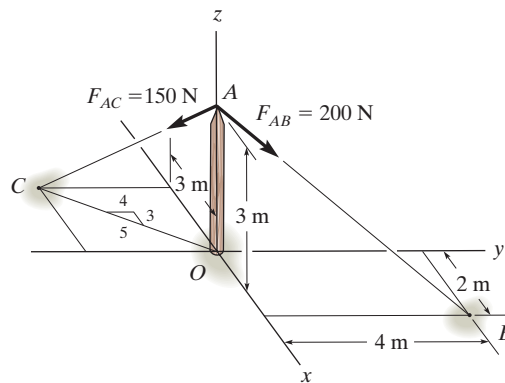
**Prob. 2-98**

**2-99.** The load at  $A$  creates a force of 60 lb in wire  $AB$ . Express this force as a Cartesian vector acting on  $A$  and directed toward  $B$  as shown.



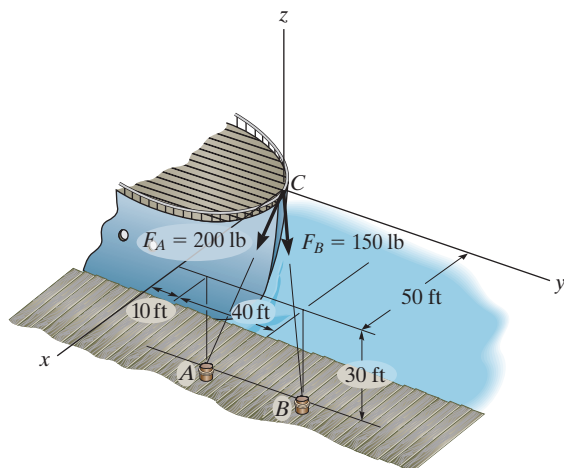
**Prob. 2-99**

**\*2-100.** Determine the magnitude and coordinate direction angles of the resultant force acting at point  $A$  on the post.



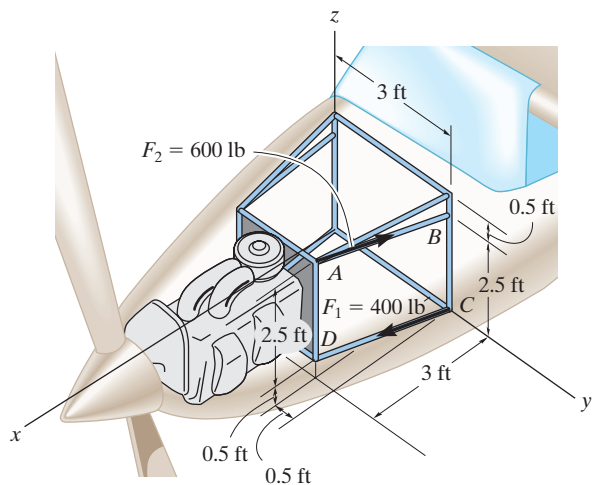
**Prob. 2-100**

**2-101.** The two mooring cables exert forces on the stern of a ship as shown. Represent each force as a Cartesian vector and determine the magnitude and coordinate direction angles of the resultant.



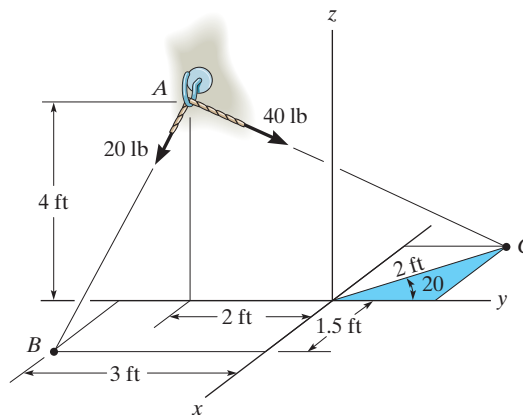
**Prob. 2-101**

**2-102.** The engine of the lightweight plane is supported by struts that are connected to the space truss that makes up the structure of the plane. The anticipated loading in two of the struts is shown. Express each of those forces as Cartesian vector.



**Prob. 2-102**

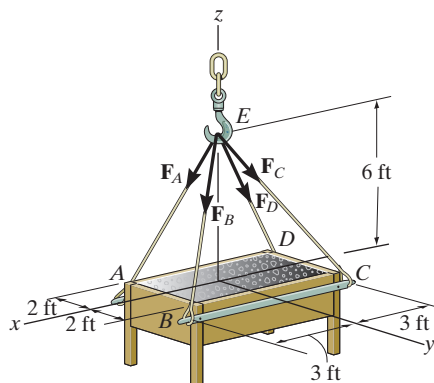
**2-103.** Determine the magnitude and coordinate direction angles of the resultant force.



**Prob. 2-103**

**\*2-104.** If the force in each cable tied to the bin is 70 lb, determine the magnitude and coordinate direction angles of the resultant force.

**2-105.** If the resultant of the four forces is  $\mathbf{F}_R = \{-360\mathbf{k}\}$  lb, determine the tension developed in each cable. Due to symmetry, the tension in the four cables is the same.



**Probs. 2-104/105**

## 2.9 Dot Product

Occasionally in statics one has to find the angle between two lines or the components of a force parallel and perpendicular to a line. In two dimensions, these problems can readily be solved by trigonometry since the geometry is easy to visualize. In three dimensions, however, this is often difficult, and consequently vector methods should be employed for the solution. The dot product, which defines a particular method for “multiplying” two vectors, can be used to solve the above-mentioned problems.

The **dot product** of vectors **A** and **B**, written  $\mathbf{A} \cdot \mathbf{B}$  and read “**A** dot **B**,” is defined as the product of the magnitudes of **A** and **B** and the cosine of the angle  $\theta$  between their tails, Fig. 2–40. Expressed in equation form,

$$\mathbf{A} \cdot \mathbf{B} = AB \cos \theta \quad (2-12)$$

where  $0^\circ \leq \theta \leq 180^\circ$ . The dot product is often referred to as the *scalar product* of vectors since the result is a *scalar* and not a vector.

### Laws of Operation.

1. Commutative law:  $\mathbf{A} \cdot \mathbf{B} = \mathbf{B} \cdot \mathbf{A}$
2. Multiplication by a scalar:  $a(\mathbf{A} \cdot \mathbf{B}) = (a\mathbf{A}) \cdot \mathbf{B} = \mathbf{A} \cdot (a\mathbf{B})$
3. Distributive law:  $\mathbf{A} \cdot (\mathbf{B} + \mathbf{D}) = (\mathbf{A} \cdot \mathbf{B}) + (\mathbf{A} \cdot \mathbf{D})$

It is easy to prove the first and second laws by using Eq. 2–12. The proof of the distributive law is left as an exercise (see Prob. 2–112).

**Cartesian Vector Formulation.** Equation 2–12 must be used to find the dot product for any two Cartesian unit vectors. For example,  $\mathbf{i} \cdot \mathbf{i} = (1)(1) \cos 0^\circ = 1$  and  $\mathbf{i} \cdot \mathbf{j} = (1)(1) \cos 90^\circ = 0$ . If we want to find the dot product of two general vectors **A** and **B** that are expressed in Cartesian vector form, then we have

$$\begin{aligned} \mathbf{A} \cdot \mathbf{B} &= (A_x \mathbf{i} + A_y \mathbf{j} + A_z \mathbf{k}) \cdot (B_x \mathbf{i} + B_y \mathbf{j} + B_z \mathbf{k}) \\ &= A_x B_x (\mathbf{i} \cdot \mathbf{i}) + A_x B_y (\mathbf{i} \cdot \mathbf{j}) + A_x B_z (\mathbf{i} \cdot \mathbf{k}) \\ &\quad + A_y B_x (\mathbf{j} \cdot \mathbf{i}) + A_y B_y (\mathbf{j} \cdot \mathbf{j}) + A_y B_z (\mathbf{j} \cdot \mathbf{k}) \\ &\quad + A_z B_x (\mathbf{k} \cdot \mathbf{i}) + A_z B_y (\mathbf{k} \cdot \mathbf{j}) + A_z B_z (\mathbf{k} \cdot \mathbf{k}) \end{aligned}$$

Carrying out the dot-product operations, the final result becomes

$$\mathbf{A} \cdot \mathbf{B} = A_x B_x + A_y B_y + A_z B_z \quad (2-13)$$

Thus, to determine the dot product of two Cartesian vectors, multiply their corresponding  $x$ ,  $y$ ,  $z$  components and sum these products algebraically. Note that the result will be either a positive or negative *scalar*, or it could be zero.

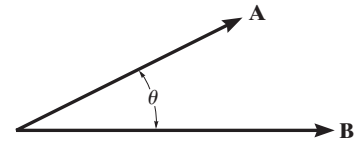


Fig. 2–40



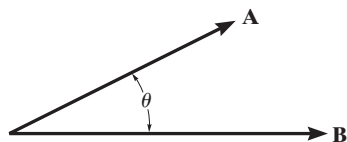
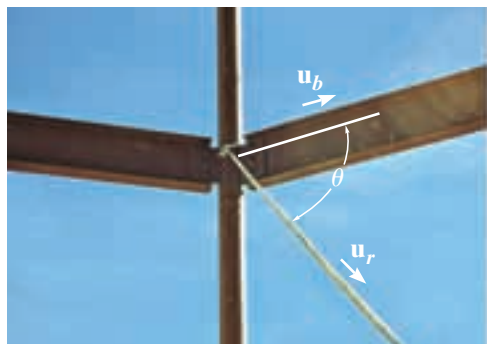
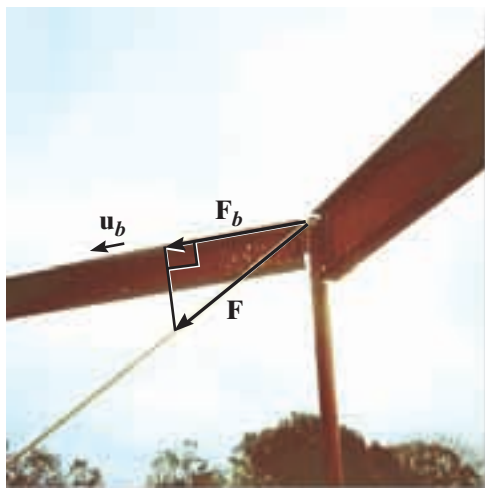


Fig. 2-40 (Repeated)



The angle  $\theta$  between the rope and the beam can be determined by formulating unit vectors along the beam and rope and then using the dot product  $\mathbf{u}_b \cdot \mathbf{u}_r = (1)(1) \cos \theta$ . (© Russell C. Hibbeler)



The projection of the cable force  $\mathbf{F}$  along the beam can be determined by first finding the unit vector  $\mathbf{u}_b$  that defines this direction. Then apply the dot product,  $F_b = \mathbf{F} \cdot \mathbf{u}_b$ . (© Russell C. Hibbeler)

**Applications.** The dot product has two important applications in mechanics.

- **The angle formed between two vectors or intersecting lines.** The angle  $\theta$  between the tails of vectors  $\mathbf{A}$  and  $\mathbf{B}$  in Fig. 2-40 can be determined from Eq. 2-12 and written as

$$\theta = \cos^{-1} \left( \frac{\mathbf{A} \cdot \mathbf{B}}{AB} \right) \quad 0^\circ \leq \theta \leq 180^\circ$$

Here  $\mathbf{A} \cdot \mathbf{B}$  is found from Eq. 2-13. In particular, notice that if  $\mathbf{A} \cdot \mathbf{B} = 0$ ,  $\theta = \cos^{-1} 0 = 90^\circ$  so that  $\mathbf{A}$  will be *perpendicular* to  $\mathbf{B}$ .

- **The components of a vector parallel and perpendicular to a line.** The component of vector  $\mathbf{A}$  parallel to or collinear with the line  $aa$  in Fig. 2-40 is defined by  $A_a$  where  $A_a = A \cos \theta$ . This component is sometimes referred to as the **projection** of  $\mathbf{A}$  onto the line, since a *right angle* is formed in the construction. If the *direction* of the line is specified by the unit vector  $\mathbf{u}_a$ , then since  $u_a = 1$ , we can determine the magnitude of  $A_a$  directly from the dot product (Eq. 2-12); i.e.,

$$A_a = A \cos \theta = \mathbf{A} \cdot \mathbf{u}_a$$

Hence, the scalar projection of  $\mathbf{A}$  along a line is determined from the dot product of  $\mathbf{A}$  and the unit vector  $\mathbf{u}_a$  which defines the direction of the line. Notice that if this result is positive, then  $\mathbf{A}_a$  has a directional sense which is the same as  $\mathbf{u}_a$ , whereas if  $A_a$  is a negative scalar, then  $\mathbf{A}_a$  has the opposite sense of direction to  $\mathbf{u}_a$ .

The component  $\mathbf{A}_a$  represented as a *vector* is therefore

$$\mathbf{A}_a = A_a \mathbf{u}_a$$

The component of  $\mathbf{A}$  that is *perpendicular* to line  $aa$  can also be obtained, Fig. 2-41. Since  $\mathbf{A} = \mathbf{A}_a + \mathbf{A}_\perp$ , then  $\mathbf{A}_\perp = \mathbf{A} - \mathbf{A}_a$ . There are two possible ways of obtaining  $A_\perp$ . One way would be to determine  $\theta$  from the dot product,  $\theta = \cos^{-1}(\mathbf{A} \cdot \mathbf{u}_a / A)$ , then  $A_\perp = A \sin \theta$ . Alternatively, if  $A_a$  is known, then by Pythagorean's theorem we can also write  $A_\perp = \sqrt{A^2 - A_a^2}$ .

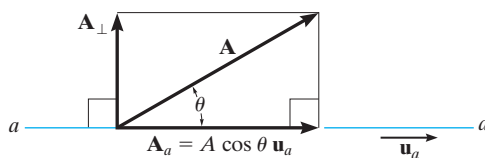


Fig. 2-41

## Important Points

- The dot product is used to determine the angle between two vectors or the projection of a vector in a specified direction.
- If vectors  $\mathbf{A}$  and  $\mathbf{B}$  are expressed in Cartesian vector form, the dot product is determined by multiplying the respective  $x$ ,  $y$ ,  $z$  scalar components and algebraically adding the results, i.e.,  $\mathbf{A} \cdot \mathbf{B} = A_x B_x + A_y B_y + A_z B_z$ .
- From the definition of the dot product, the angle formed between the tails of vectors  $\mathbf{A}$  and  $\mathbf{B}$  is  $\theta = \cos^{-1}(\mathbf{A} \cdot \mathbf{B} / AB)$ .
- The magnitude of the projection of vector  $\mathbf{A}$  along a line  $aa$  whose direction is specified by  $\mathbf{u}_a$  is determined from the dot product  $A_a = \mathbf{A} \cdot \mathbf{u}_a$ .

### EXAMPLE 2.14

Determine the magnitudes of the projection of the force  $\mathbf{F}$  in Fig. 2–42 onto the  $u$  and  $v$  axes.

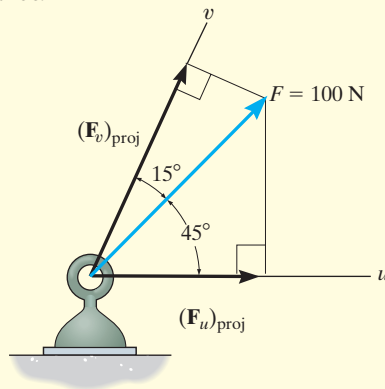


Fig. 2–42

### SOLUTION

**Projections of Force.** The graphical representation of the *projections* is shown in Fig. 2–42. From this figure, the magnitudes of the projections of  $\mathbf{F}$  onto the  $u$  and  $v$  axes can be obtained by trigonometry:

$$(F_u)_{\text{proj}} = (100 \text{ N}) \cos 45^\circ = 70.7 \text{ N} \quad \text{Ans.}$$

$$(F_v)_{\text{proj}} = (100 \text{ N}) \cos 15^\circ = 96.6 \text{ N} \quad \text{Ans.}$$

**NOTE:** These projections are not equal to the magnitudes of the components of force  $\mathbf{F}$  along the  $u$  and  $v$  axes found from the parallelogram law. They will only be equal if the  $u$  and  $v$  axes are *perpendicular* to one another.

## EXAMPLE 2.15

The frame shown in Fig. 2-43a is subjected to a horizontal force  $\mathbf{F} = \{300\mathbf{j}\}$  N. Determine the magnitudes of the components of this force parallel and perpendicular to member  $AB$ .

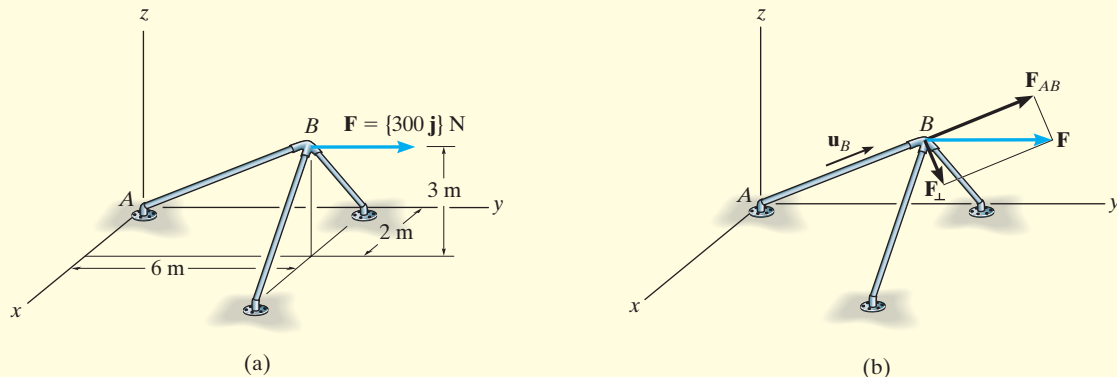


Fig. 2-43

## SOLUTION

The magnitude of the component of  $\mathbf{F}$  along  $AB$  is equal to the dot product of  $\mathbf{F}$  and the unit vector  $\mathbf{u}_B$ , which defines the direction of  $AB$ , Fig. 2-43b. Since

$$\mathbf{u}_B = \frac{\mathbf{r}_B}{r_B} = \frac{2\mathbf{i} + 6\mathbf{j} + 3\mathbf{k}}{\sqrt{(2)^2 + (6)^2 + (3)^2}} = 0.286\mathbf{i} + 0.857\mathbf{j} + 0.429\mathbf{k}$$

then

$$\begin{aligned} F_{AB} &= F \cos \theta = \mathbf{F} \cdot \mathbf{u}_B = (300\mathbf{j}) \cdot (0.286\mathbf{i} + 0.857\mathbf{j} + 0.429\mathbf{k}) \\ &= (0)(0.286) + (300)(0.857) + (0)(0.429) \\ &= 257.1 \text{ N} \end{aligned}$$

Ans.

Since the result is a positive scalar,  $\mathbf{F}_{AB}$  has the same sense of direction as  $\mathbf{u}_B$ , Fig. 2-43b.

Expressing  $\mathbf{F}_{AB}$  in Cartesian vector form, we have

$$\begin{aligned} \mathbf{F}_{AB} &= F_{AB}\mathbf{u}_B = (257.1 \text{ N})(0.286\mathbf{i} + 0.857\mathbf{j} + 0.429\mathbf{k}) \\ &= \{73.5\mathbf{i} + 220\mathbf{j} + 110\mathbf{k}\} \text{ N} \end{aligned}$$

Ans.

The perpendicular component, Fig. 2-43b, is therefore

$$\begin{aligned} \mathbf{F}_\perp &= \mathbf{F} - \mathbf{F}_{AB} = 300\mathbf{j} - (73.5\mathbf{i} + 220\mathbf{j} + 110\mathbf{k}) \\ &= \{-73.5\mathbf{i} + 79.6\mathbf{j} - 110\mathbf{k}\} \text{ N} \end{aligned}$$

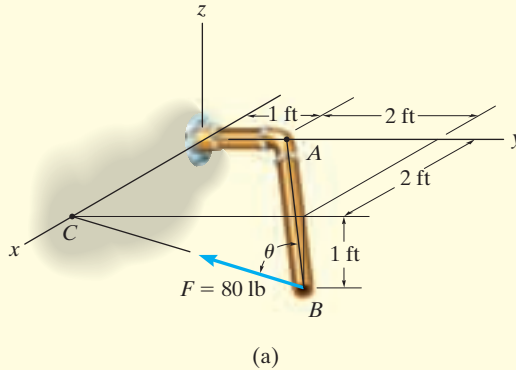
Its magnitude can be determined either from this vector or by using the Pythagorean theorem, Fig. 2-43b:

$$\begin{aligned} F_\perp &= \sqrt{F^2 - F_{AB}^2} = \sqrt{(300 \text{ N})^2 - (257.1 \text{ N})^2} \\ &= 155 \text{ N} \end{aligned}$$

Ans.

**EXAMPLE 2.16**

The pipe in Fig. 2–44a is subjected to the force of  $F = 80$  lb. Determine the angle  $\theta$  between  $\mathbf{F}$  and the pipe segment  $BA$  and the projection of  $\mathbf{F}$  along this segment.

**SOLUTION**

**Angle  $\theta$ .** First we will establish position vectors from  $B$  to  $A$  and  $B$  to  $C$ ; Fig. 2–44b. Then we will determine the angle  $\theta$  between the tails of these two vectors.

$$\mathbf{r}_{BA} = \{-2\mathbf{i} - 2\mathbf{j} + 1\mathbf{k}\} \text{ ft}, \quad r_{BA} = 3 \text{ ft}$$

$$\mathbf{r}_{BC} = \{-3\mathbf{j} + 1\mathbf{k}\} \text{ ft}, \quad r_{BC} = \sqrt{10} \text{ ft}$$

Thus,

$$\cos \theta = \frac{\mathbf{r}_{BA} \cdot \mathbf{r}_{BC}}{r_{BA} r_{BC}} = \frac{(-2)(0) + (-2)(-3) + (1)(1)}{3\sqrt{10}} = 0.7379$$

$$\theta = 42.5^\circ$$

*Ans.*

**Components of  $\mathbf{F}$ .** The component of  $\mathbf{F}$  along  $BA$  is shown in Fig. 2–44c. We must first formulate the unit vector along  $BA$  and force  $\mathbf{F}$  as Cartesian vectors.

$$\mathbf{u}_{BA} = \frac{\mathbf{r}_{BA}}{r_{BA}} = \frac{(-2\mathbf{i} - 2\mathbf{j} + 1\mathbf{k})}{3} = -\frac{2}{3}\mathbf{i} - \frac{2}{3}\mathbf{j} + \frac{1}{3}\mathbf{k}$$

$$\mathbf{F} = 80 \text{ lb} \left( \frac{\mathbf{r}_{BC}}{r_{BC}} \right) = 80 \left( \frac{-3\mathbf{j} + 1\mathbf{k}}{\sqrt{10}} \right) = -75.89\mathbf{j} + 25.30\mathbf{k}$$

Thus,

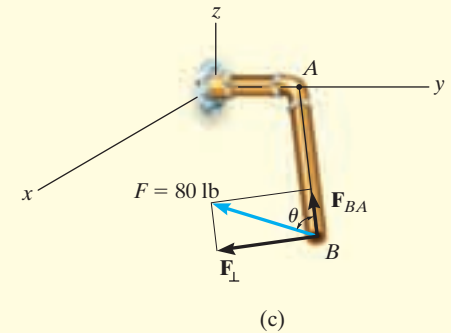
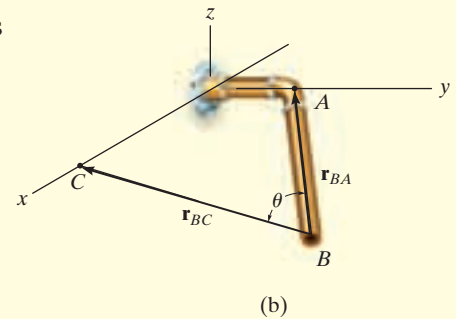
$$F_{BA} = \mathbf{F} \cdot \mathbf{u}_{BA} = (-75.89\mathbf{j} + 25.30\mathbf{k}) \cdot \left( -\frac{2}{3}\mathbf{i} - \frac{2}{3}\mathbf{j} + \frac{1}{3}\mathbf{k} \right)$$

$$= 0 \left( -\frac{2}{3} \right) + (-75.89) \left( -\frac{2}{3} \right) + (25.30) \left( \frac{1}{3} \right)$$

$$= 59.0 \text{ lb}$$

*Ans.*

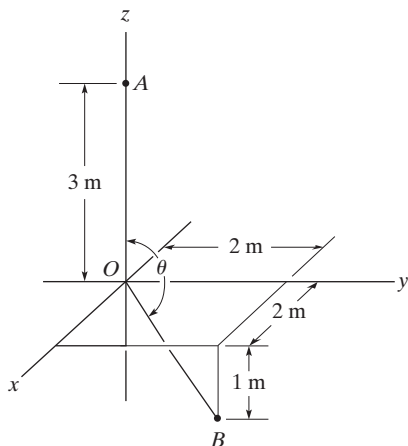
**NOTE:** Since  $\theta$  has been calculated, then also,  $F_{BA} = F \cos \theta = 80 \text{ lb} \cos 42.5^\circ = 59.0 \text{ lb}$ .



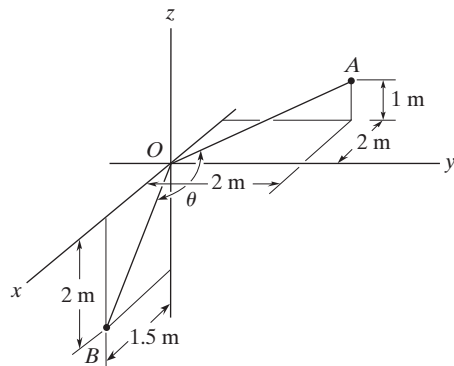
**Fig. 2–44**

## PRELIMINARY PROBLEMS

**P2-8.** In each case, set up the dot product to find the angle  $\theta$ . Do not calculate the result.



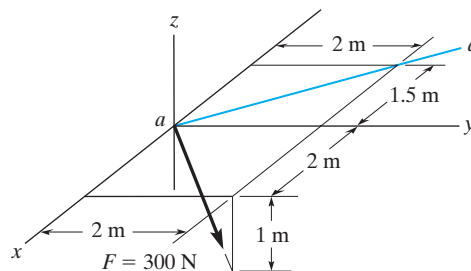
(a)



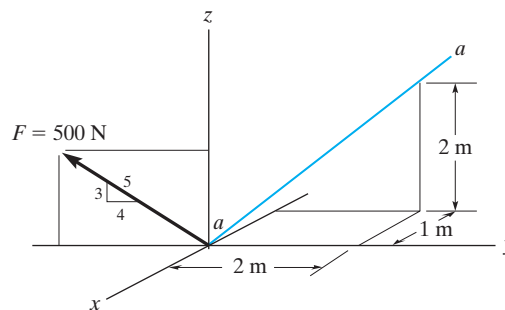
(b)

**Prob. P2-8**

**P2-9.** In each case, set up the dot product to find the magnitude of the projection of the force  $\mathbf{F}$  along  $a$ - $a$  axes. Do not calculate the result.



(a)

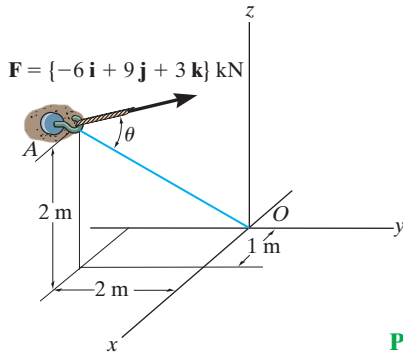


(b)

**Prob. P2-9**

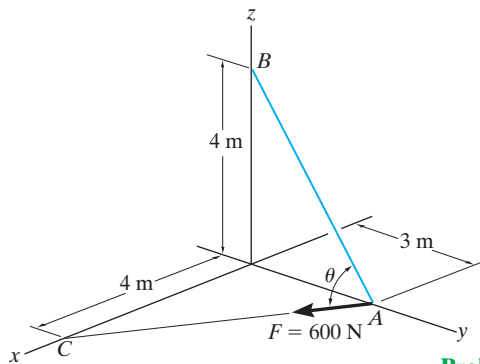
## FUNDAMENTAL PROBLEMS

**F2-25.** Determine the angle  $\theta$  between the force and the line  $AO$ .



**Prob. F2-25**

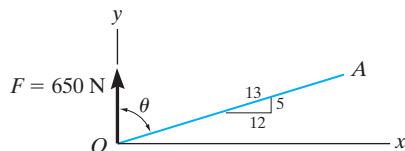
**F2-26.** Determine the angle  $\theta$  between the force and the line  $AB$ .



**Prob. F2-26**

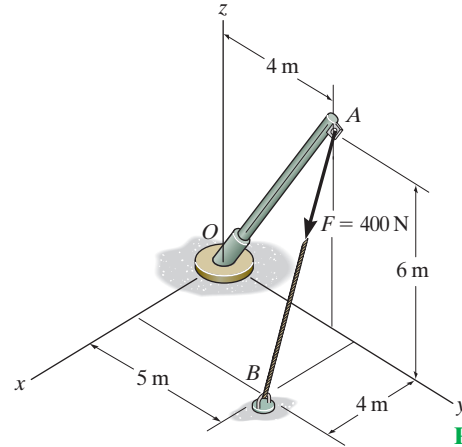
**F2-27.** Determine the angle  $\theta$  between the force and the line  $OA$ .

**F2-28.** Determine the projected component of the force along the line  $OA$ .



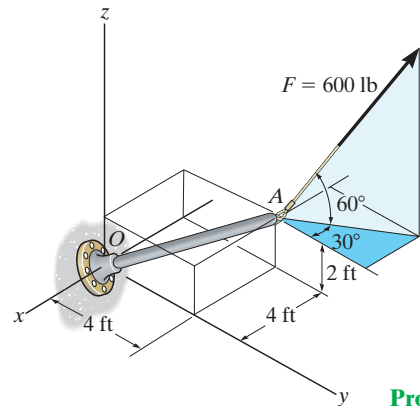
**Probs. F2-27/28**

**F2-29.** Find the magnitude of the projected component of the force along the pipe  $AO$ .



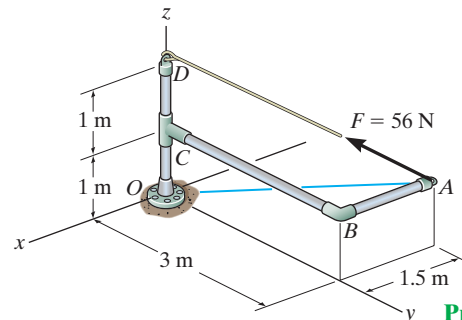
**Prob. F2-29**

**F2-30.** Determine the components of the force acting parallel and perpendicular to the axis of the pole.



**Prob. F2-30**

**F2-31.** Determine the magnitudes of the components of the force  $F = 56$  N acting along and perpendicular to line  $AO$ .

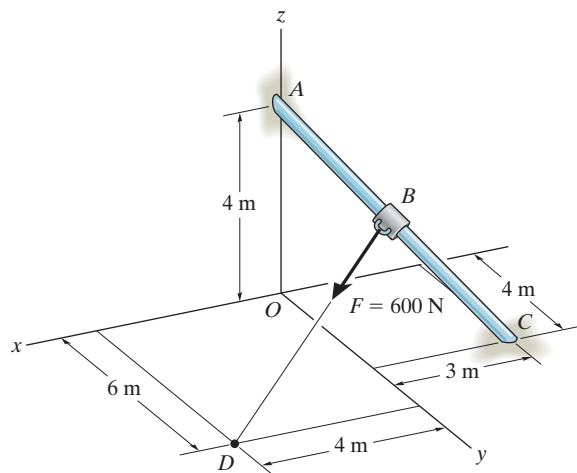


**Prob. F2-31**

## PROBLEMS

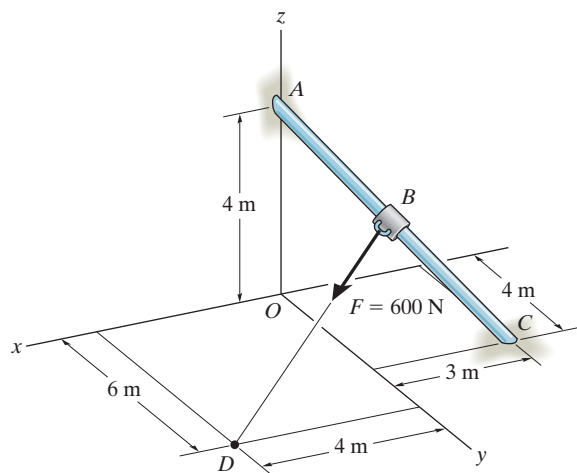
**2-106.** Express the force  $\mathbf{F}$  in Cartesian vector form if it acts at the midpoint  $B$  of the rod.

2



Prob. 2-106

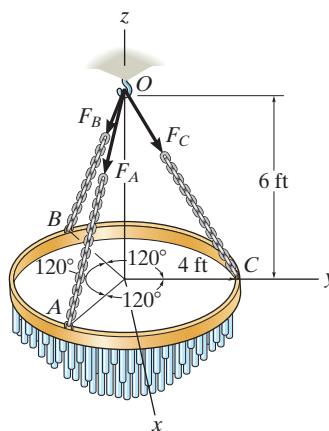
**2-107.** Express force  $\mathbf{F}$  in Cartesian vector form if point  $B$  is located 3 m along the rod from end  $C$ .



Prob. 2-107

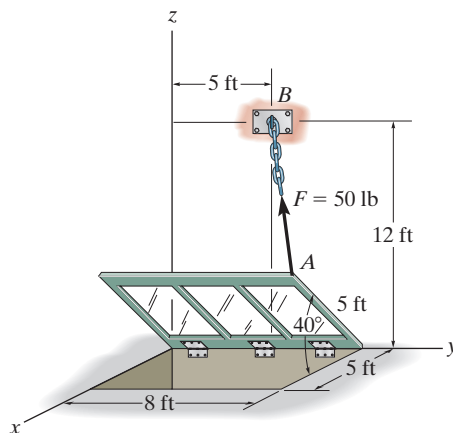
**\*2-108.** The chandelier is supported by three chains which are concurrent at point  $O$ . If the force in each chain has a magnitude of 60 lb, express each force as a Cartesian vector and determine the magnitude and coordinate direction angles of the resultant force.

**2-109.** The chandelier is supported by three chains which are concurrent at point  $O$ . If the resultant force at  $O$  has a magnitude of 130 lb and is directed along the negative  $z$  axis, determine the force in each chain.



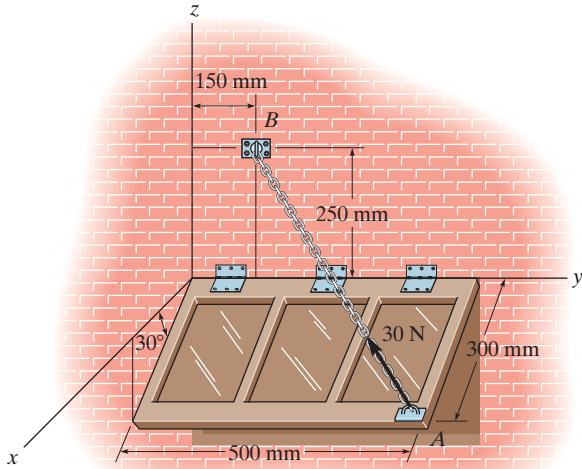
Probs. 2-108/109

**2-110.** The window is held open by chain  $AB$ . Determine the length of the chain, and express the 50-lb force acting at  $A$  along the chain as a Cartesian vector and determine its coordinate direction angles.



Prob. 2-110

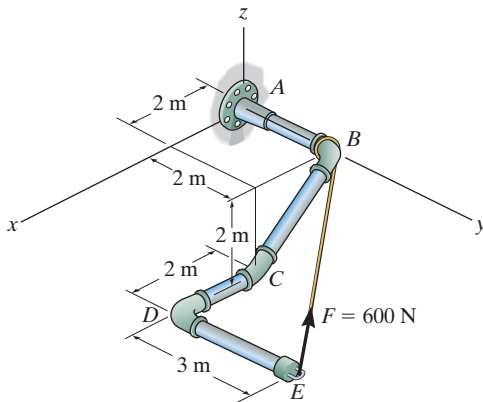
**2-111.** The window is held open by cable  $AB$ . Determine the length of the cable and express the 30-N force acting at  $A$  along the cable as a Cartesian vector.



**Prob. 2-111**

**\*2-112.** Given the three vectors  $\mathbf{A}$ ,  $\mathbf{B}$ , and  $\mathbf{D}$ , show that  $\mathbf{A} \cdot (\mathbf{B} + \mathbf{D}) = (\mathbf{A} \cdot \mathbf{B}) + (\mathbf{A} \cdot \mathbf{D})$ .

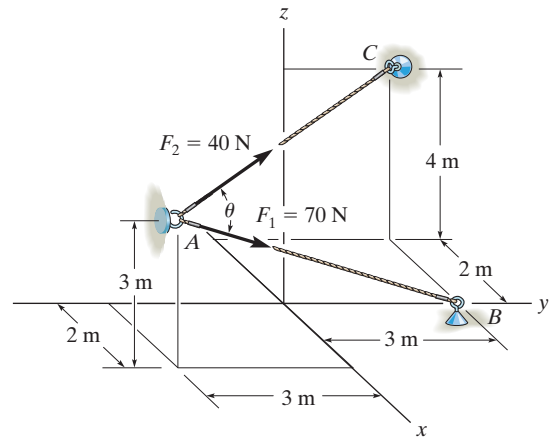
**2-113.** Determine the magnitudes of the components of  $F = 600$  N acting along and perpendicular to segment  $DE$  of the pipe assembly.



**Probs. 2-112/113**

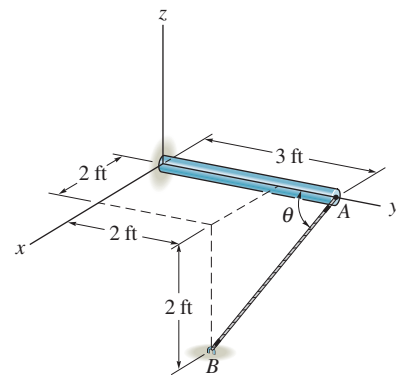
**2-114.** Determine the angle  $\theta$  between the two cables.

**2-115.** Determine the magnitude of the projection of the force  $\mathbf{F}_1$  along cable  $AC$ .



**Probs. 2-114/115**

**\*2-116.** Determine the angle  $\theta$  between the  $y$  axis of the pole and the wire  $AB$ .

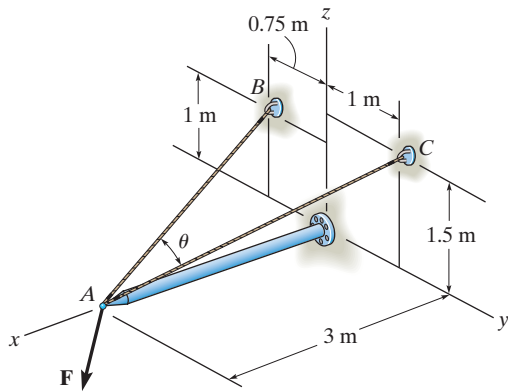


**Prob. 2-116**



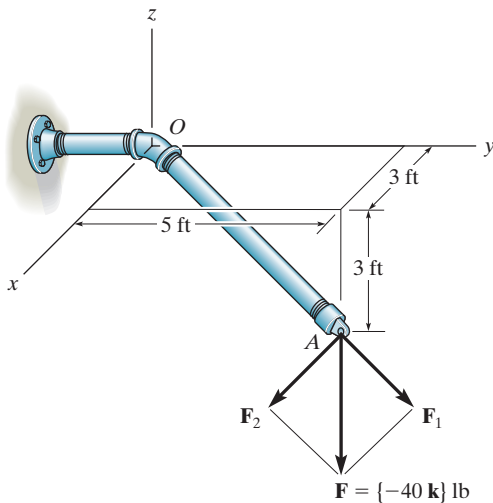
**2-117.** Determine the magnitudes of the projected components of the force  $\mathbf{F} = [60\mathbf{i} + 12\mathbf{j} - 40\mathbf{k}]$  N along the cables  $AB$  and  $AC$ .

**2-118.** Determine the angle  $\theta$  between cables  $AB$  and  $AC$ .



**Probs. 2-117/118**

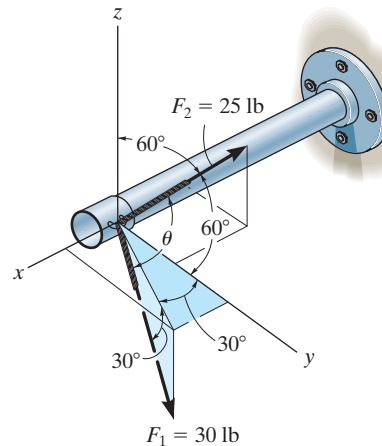
**2-119.** A force of  $\mathbf{F} = \{-40\mathbf{k}\}$  lb acts at the end of the pipe. Determine the magnitudes of the components  $\mathbf{F}_1$  and  $\mathbf{F}_2$  which are directed along the pipe's axis and perpendicular to it.



**Prob. 2-119**

**\*2-120.** Two cables exert forces on the pipe. Determine the magnitude of the projected component of  $\mathbf{F}_1$  along the line of action of  $\mathbf{F}_2$ .

**2-121.** Determine the angle  $\theta$  between the two cables attached to the pipe.

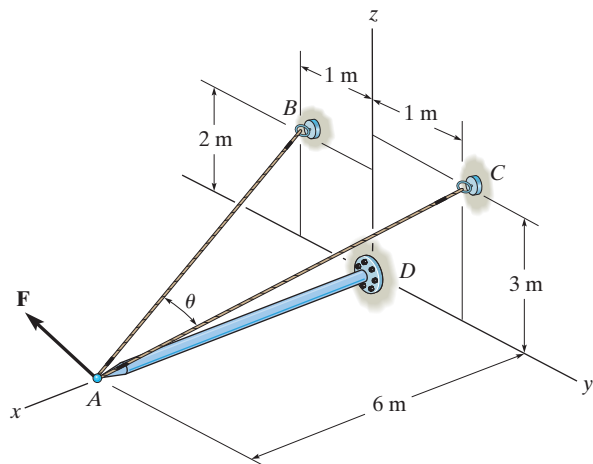


**Probs. 2-120/121**

**2-122.** Determine the angle  $\theta$  between the cables  $AB$  and  $AC$ .

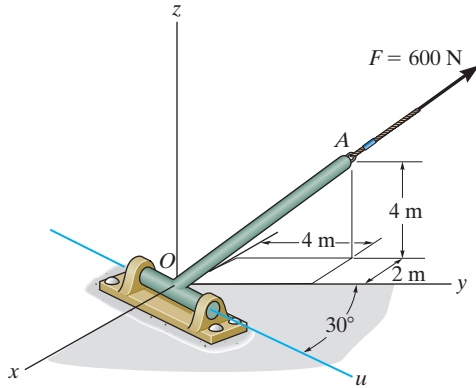
**2-123.** Determine the magnitude of the projected component of the force  $\mathbf{F} = \{400\mathbf{i} - 200\mathbf{j} + 500\mathbf{k}\}$  N acting along the cable  $BA$ .

**\*2-124.** Determine the magnitude of the projected component of the force  $\mathbf{F} = \{400\mathbf{i} - 200\mathbf{j} + 500\mathbf{k}\}$  N acting along the cable  $CA$ .



**Probs. 2-122/123/124**

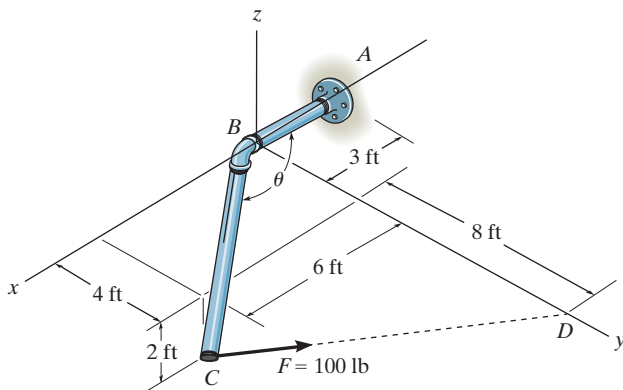
**2-125.** Determine the magnitude of the projection of force  $F = 600$  N along the  $u$  axis.



**Prob. 2-125**

**2-126.** Determine the magnitude of the projected component of the 100-lb force acting along the axis  $BC$  of the pipe.

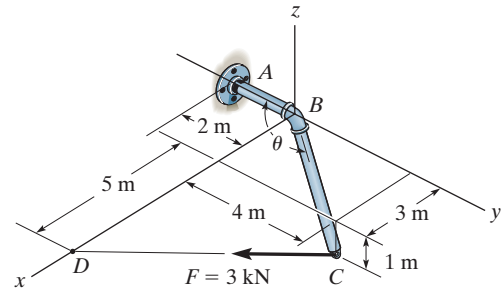
**2-127.** Determine the angle  $\theta$  between pipe segments  $BA$  and  $BC$ .



**Probs. 2-126/127**

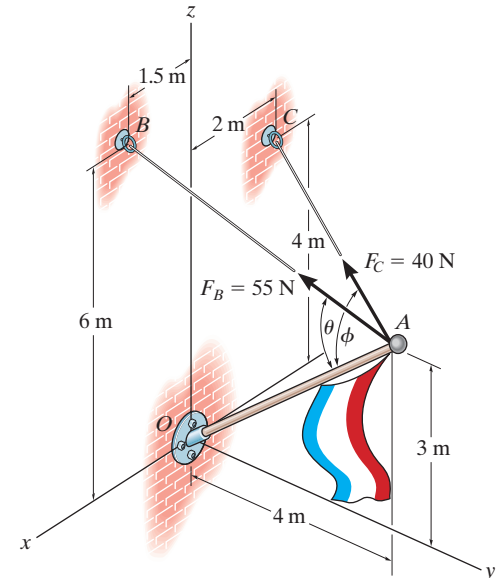
**\*2-128.** Determine the angle  $\theta$  between  $BA$  and  $BC$ .

**2-129.** Determine the magnitude of the projected component of the 3 kN force acting along the axis  $BC$  of the pipe.



**Probs. 2-128/129**

**2-130.** Determine the angles  $\theta$  and  $\phi$  made between the axes  $OA$  of the flag pole and  $AB$  and  $AC$ , respectively, of each cable.

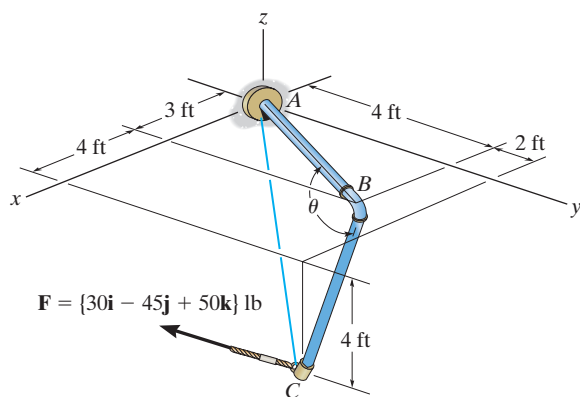


**Prob. 2-130**

**2-131.** Determine the magnitudes of the components of  $\mathbf{F}$  acting along and perpendicular to segment  $BC$  of the pipe assembly.

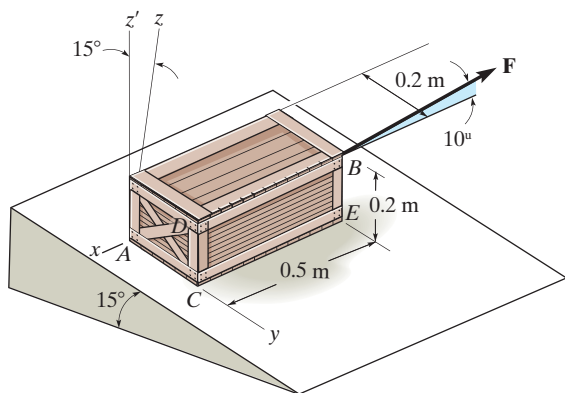
**\*2-132.** Determine the magnitude of the projected component of  $\mathbf{F}$  along  $AC$ . Express this component as a Cartesian vector.

**2-133.** Determine the angle  $\theta$  between the pipe segments  $BA$  and  $BC$ .



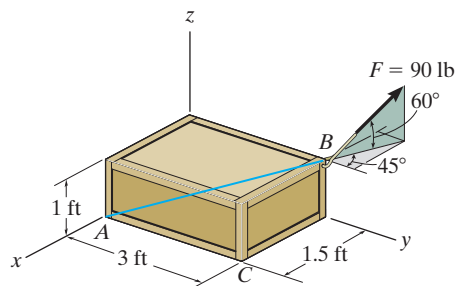
**Probs. 2-131/132/133**

**2-134.** If the force  $F = 100$  N lies in the plane  $DBEC$ , which is parallel to the  $x$ - $z$  plane, and makes an angle of  $10^\circ$  with the extended line  $DB$  as shown, determine the angle that  $\mathbf{F}$  makes with the diagonal  $AB$  of the crate.



**Prob. 2-134**

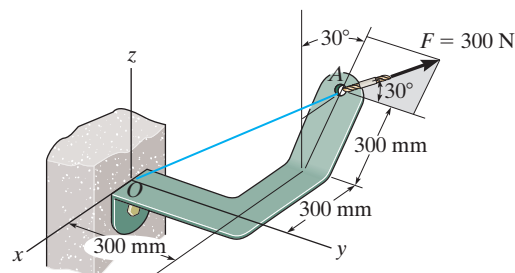
**2-135.** Determine the magnitudes of the components of the force  $F = 90$  lb acting parallel and perpendicular to diagonal  $AB$  of the crate.



**Prob. 2-135**

**\*2-136.** Determine the magnitudes of the projected components of the force  $F = 300$  N acting along the  $x$  and  $y$  axes.

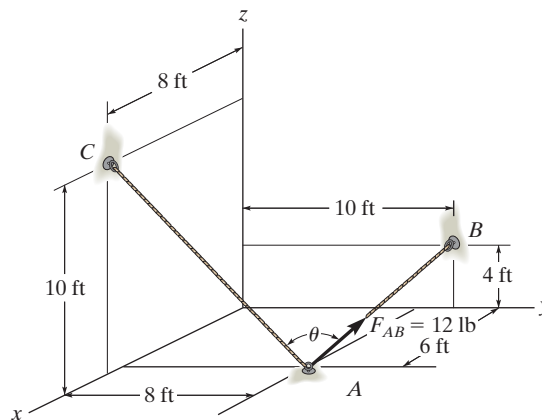
**2-137.** Determine the magnitude of the projected component of the force  $F = 300$  N acting along line  $OA$ .



**Probs. 2-136/137**

**2-138.** Determine the angle  $\theta$  between the two cables.

**2-139.** Determine the projected component of the force  $F = 12$  lb acting in the direction of cable  $AC$ . Express the result as a Cartesian vector.



**Probs. 2-138/139**

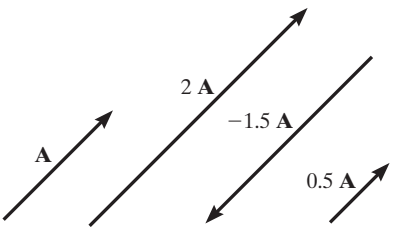
CHAPTER REVIEW

A scalar is a positive or negative number; e.g., mass and temperature.

A vector has a magnitude and direction, where the arrowhead represents the sense of the vector.

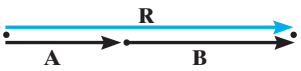


Multiplication or division of a vector by a scalar will change only the magnitude of the vector. If the scalar is negative, the sense of the vector will change so that it acts in the opposite sense.



If vectors are collinear, the resultant is simply the algebraic or scalar addition.

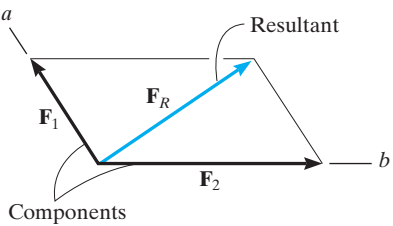
$$R = A + B$$



Parallelogram Law

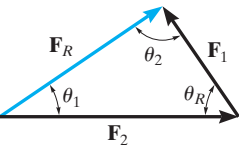
Two forces add according to the parallelogram law. The *components* form the sides of the parallelogram and the *resultant* is the diagonal.

To find the components of a force along any two axes, extend lines from the head of the force, parallel to the axes, to form the components.



To obtain the components of the resultant, show how the forces add by tip-to-tail using the triangle rule, and then use the law of cosines and the law of sines to calculate their values.

$$F_R = \sqrt{F_1^2 + F_2^2 - 2 F_1 F_2 \cos \theta_R}$$
$$\frac{F_1}{\sin \theta_1} = \frac{F_2}{\sin \theta_2} = \frac{F_R}{\sin \theta_R}$$

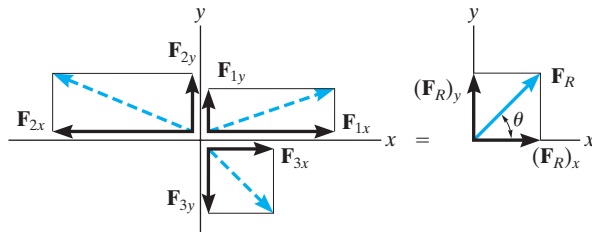
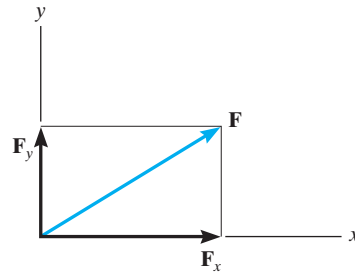


**Rectangular Components: Two Dimensions**

Vectors  $\mathbf{F}_x$  and  $\mathbf{F}_y$  are rectangular components of  $\mathbf{F}$ .

The resultant force is determined from the algebraic sum of its components.

$$\begin{aligned}(F_R)_x &= \Sigma F_x \\ (F_R)_y &= \Sigma F_y \\ F_R &= \sqrt{(F_R)_x^2 + (F_R)_y^2} \\ \theta &= \tan^{-1} \left| \frac{(F_R)_y}{(F_R)_x} \right|\end{aligned}$$

**Cartesian Vectors**

The unit vector  $\mathbf{u}$  has a length of 1, no units, and it points in the direction of the vector  $\mathbf{F}$ .

A force can be resolved into its Cartesian components along the  $x$ ,  $y$ ,  $z$  axes so that  $\mathbf{F} = F_x \mathbf{i} + F_y \mathbf{j} + F_z \mathbf{k}$ .

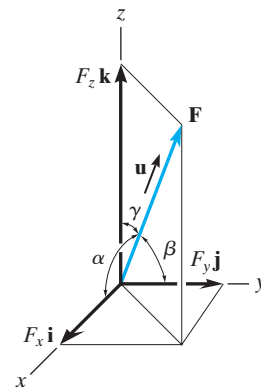
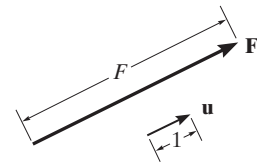
The magnitude of  $\mathbf{F}$  is determined from the positive square root of the sum of the squares of its components.

The coordinate direction angles  $\alpha, \beta, \gamma$  are determined by formulating a unit vector in the direction of  $\mathbf{F}$ . The  $x$ ,  $y$ ,  $z$  components of  $\mathbf{u}$  represent  $\cos \alpha, \cos \beta, \cos \gamma$ .

$$\mathbf{u} = \frac{\mathbf{F}}{F}$$

$$F = \sqrt{F_x^2 + F_y^2 + F_z^2}$$

$$\begin{aligned}\mathbf{u} &= \frac{\mathbf{F}}{F} = \frac{F_x}{F} \mathbf{i} + \frac{F_y}{F} \mathbf{j} + \frac{F_z}{F} \mathbf{k} \\ \mathbf{u} &= \cos \alpha \mathbf{i} + \cos \beta \mathbf{j} + \cos \gamma \mathbf{k}\end{aligned}$$



The coordinate direction angles are related so that only two of the three angles are independent of one another.

To find the resultant of a concurrent force system, express each force as a Cartesian vector and add the  $\mathbf{i}$ ,  $\mathbf{j}$ ,  $\mathbf{k}$  components of all the forces in the system.

$$\cos^2 \alpha + \cos^2 \beta + \cos^2 \gamma = 1$$

$$\mathbf{F}_R = \Sigma \mathbf{F} = \Sigma F_x \mathbf{i} + \Sigma F_y \mathbf{j} + \Sigma F_z \mathbf{k}$$

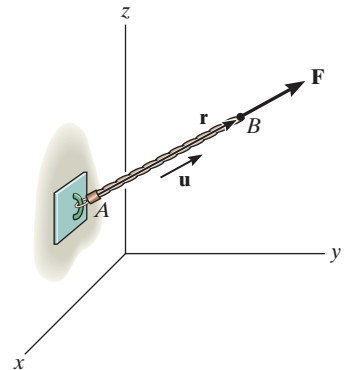
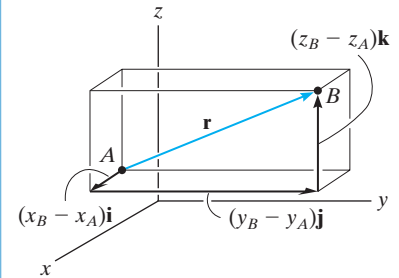
### Position and Force Vectors

A position vector locates one point in space relative to another. The easiest way to formulate the components of a position vector is to determine the distance and direction that one must travel along the  $x$ ,  $y$ , and  $z$  directions—going from the tail to the head of the vector.

If the line of action of a force passes through points  $A$  and  $B$ , then the force acts in the same direction as the position vector  $\mathbf{r}$ , which is defined by the unit vector  $\mathbf{u}$ . The force can then be expressed as a Cartesian vector.

$$\begin{aligned} \mathbf{r} &= (x_B - x_A)\mathbf{i} \\ &+ (y_B - y_A)\mathbf{j} \\ &+ (z_B - z_A)\mathbf{k} \end{aligned}$$

$$\mathbf{F} = F\mathbf{u} = F\left(\frac{\mathbf{r}}{r}\right)$$



### Dot Product

The dot product between two vectors  $\mathbf{A}$  and  $\mathbf{B}$  yields a scalar. If  $\mathbf{A}$  and  $\mathbf{B}$  are expressed in Cartesian vector form, then the dot product is the sum of the products of their  $x$ ,  $y$ , and  $z$  components.

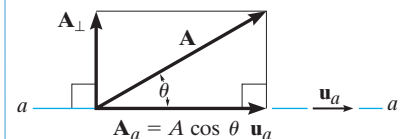
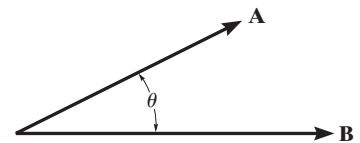
The dot product can be used to determine the angle between  $\mathbf{A}$  and  $\mathbf{B}$ .

The dot product is also used to determine the projected component of a vector  $\mathbf{A}$  onto an axis  $aa$  defined by its unit vector  $\mathbf{u}_a$ .

$$\begin{aligned} \mathbf{A} \cdot \mathbf{B} &= AB \cos \theta \\ &= A_x B_x + A_y B_y + A_z B_z \end{aligned}$$

$$\theta = \cos^{-1}\left(\frac{\mathbf{A} \cdot \mathbf{B}}{AB}\right)$$

$$A_a = A \cos \theta \mathbf{u}_a = (\mathbf{A} \cdot \mathbf{u}_a)\mathbf{u}_a$$

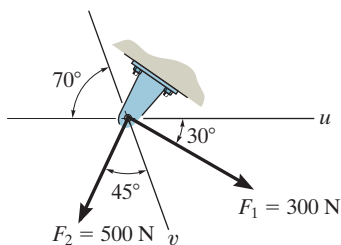


## REVIEW PROBLEMS

*Partial solutions and answers to all Review Problems are given in the back of the book.*

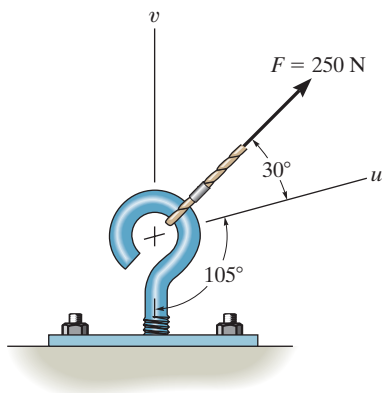
2

**R2-1.** Determine the magnitude of the resultant force  $\mathbf{F}_R$  and its direction, measured clockwise from the positive  $u$  axis.



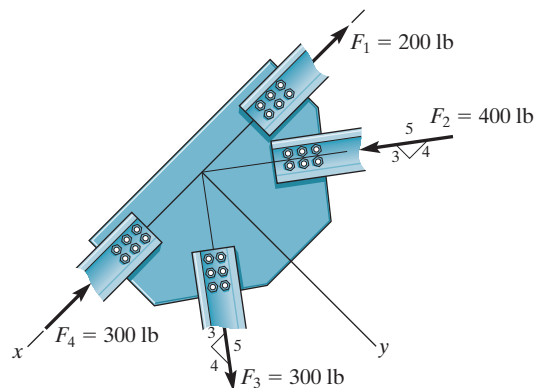
**Prob. R2-1**

**R2-2.** Resolve  $\mathbf{F}$  into components along the  $u$  and  $v$  axes and determine the magnitudes of these components.



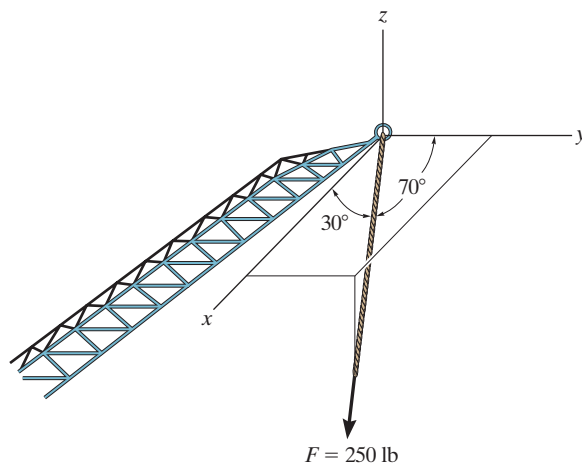
**Prob. R2-2**

**R2-3.** Determine the magnitude of the resultant force acting on the gusset plate of the bridge truss.



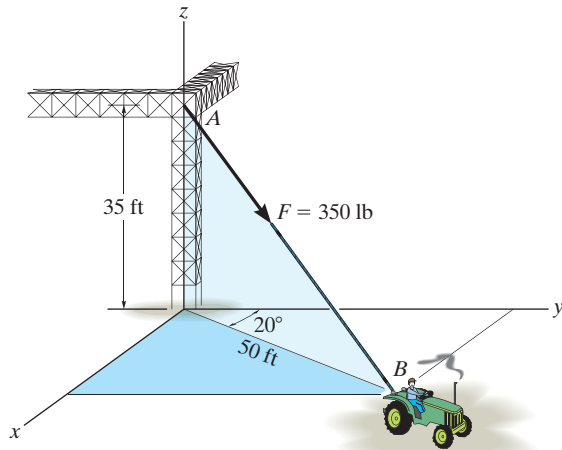
**Prob. R2-3**

**R2-4.** The cable at the end of the crane boom exerts a force of 250 lb on the boom as shown. Express  $\mathbf{F}$  as a Cartesian vector.



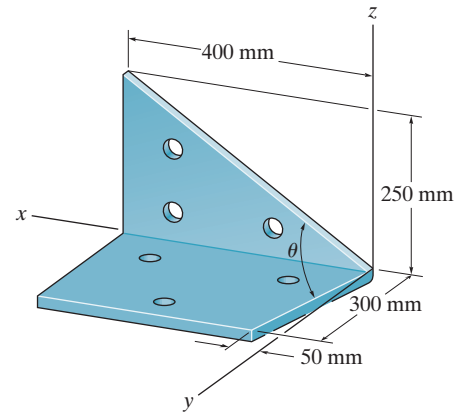
**Prob. R2-4**

**R2-5.** The cable attached to the tractor at  $B$  exerts a force of 350 lb on the framework. Express this force as a Cartesian vector.



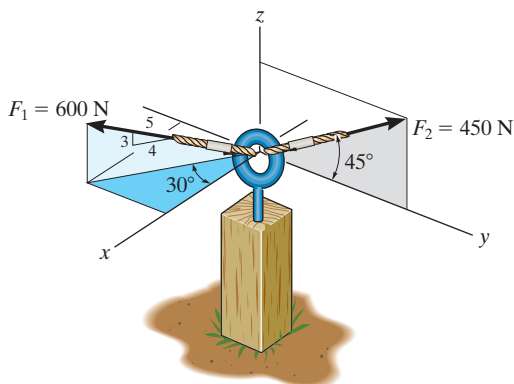
**Prob. R2-5**

**R2-7.** Determine the angle  $\theta$  between the edges of the sheet-metal bracket.



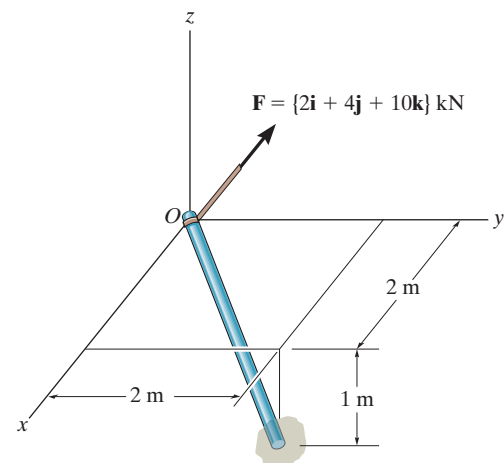
**Prob. R2-7**

**R2-6.** Express  $\mathbf{F}_1$  and  $\mathbf{F}_2$  as Cartesian vectors.



**Prob. R2-6**

**R2-8.** Determine the projection of the force  $\mathbf{F}$  along the pole.



**Prob. R2-8**



# Chapter 3



(© Igor Tumarkin/ITPS/Shutterstock)

When this load is lifted at constant velocity, or is just suspended, then it is in a state of equilibrium. In this chapter we will study equilibrium for a particle and show how these ideas can be used to calculate the forces in cables used to hold suspended loads.

# Equilibrium of a Particle

## CHAPTER OBJECTIVES

- To introduce the concept of the free-body diagram for a particle.
- To show how to solve particle equilibrium problems using the equations of equilibrium.

## 3.1 Condition for the Equilibrium of a Particle

A particle is said to be in **equilibrium** if it remains at rest if originally at rest, or has a constant velocity if originally in motion. Most often, however, the term “equilibrium” or, more specifically, “static equilibrium” is used to describe an object at rest. To maintain equilibrium, it is *necessary* to satisfy Newton’s first law of motion, which requires the *resultant force* acting on a particle to be equal to *zero*. This condition is stated by the *equation of equilibrium*,

$$\Sigma \mathbf{F} = \mathbf{0} \quad (3-1)$$

where  $\Sigma \mathbf{F}$  is the vector *sum of all the forces* acting on the particle.

Not only is Eq. 3-1 a necessary condition for equilibrium, it is also a *sufficient* condition. This follows from Newton’s second law of motion, which can be written as  $\Sigma \mathbf{F} = m\mathbf{a}$ . Since the force system satisfies Eq. 3-1, then  $m\mathbf{a} = \mathbf{0}$ , and therefore the particle’s acceleration  $\mathbf{a} = \mathbf{0}$ . Consequently, the particle indeed moves with constant velocity or remains at rest.

## 3.2 The Free-Body Diagram

To apply the equation of equilibrium, we must account for *all* the known and unknown forces ( $\Sigma \mathbf{F}$ ) which act *on* the particle. The best way to do this is to think of the particle as isolated and “free” from its surroundings. A drawing that shows the particle with *all* the forces that act on it is called a **free-body diagram (FBD)**.

Before presenting a formal procedure as to how to draw a free-body diagram, we will first consider three types of supports often encountered in particle equilibrium problems.

**Springs.** If a **linearly elastic spring** (or cord) of undeformed length  $l_0$  is used to support a particle, the length of the spring will change in direct proportion to the force  $\mathbf{F}$  acting on it, Fig. 3-1a. A characteristic that defines the “elasticity” of a spring is the **spring constant** or **stiffness**  $k$ .

The magnitude of force exerted on a linearly elastic spring which has a stiffness  $k$  and is deformed (elongated or compressed) a distance  $s = l - l_0$ , measured from its *unloaded* position, is

$$F = ks \quad (3-2)$$

If  $s$  is positive, causing an elongation, then  $\mathbf{F}$  must pull on the spring; whereas if  $s$  is negative, causing a shortening, then  $\mathbf{F}$  must push on it. For example, if the spring in Fig. 3-1a has an unstretched length of 0.8 m and a stiffness  $k = 500 \text{ N/m}$  and it is stretched to a length of 1 m, so that  $s = l - l_0 = 1 \text{ m} - 0.8 \text{ m} = 0.2 \text{ m}$ , then a force  $F = ks = 500 \text{ N/m}(0.2 \text{ m}) = 100 \text{ N}$  is needed.

**Cables and Pulleys.** Unless otherwise stated throughout this book, except in Sec. 7.4, all cables (or cords) will be assumed to have negligible weight and they cannot stretch. Also, a cable can support *only* a tension or “pulling” force, and this force always acts in the direction of the cable. In Chapter 5, it will be shown that the tension force developed in a continuous cable which passes over a frictionless pulley must have a *constant* magnitude to keep the cable in equilibrium. Hence, for any angle  $\theta$ , shown in Fig. 3-1b, the cable is subjected to a constant tension  $T$  throughout its length.

**Smooth Contact.** If an object rests on a *smooth surface*, then the surface will exert a force on the object that is normal to the surface at the point of contact. An example of this is shown in Fig. 3-2a. In addition to this normal force  $\mathbf{N}$ , the cylinder is also subjected to its weight  $\mathbf{W}$  and the force  $\mathbf{T}$  of the cord. Since these three forces are concurrent at the center of the cylinder, Fig. 3-2b, we can apply the equation of equilibrium to this “particle,” which is the same as applying it to the cylinder.

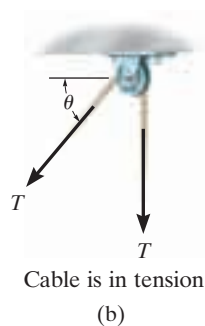
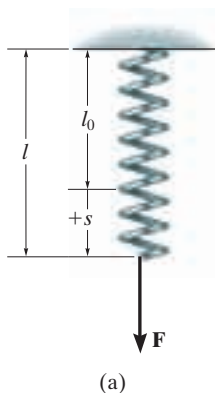


Fig. 3-1

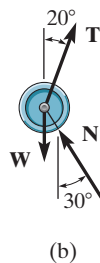
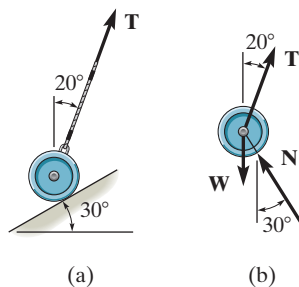


Fig. 3-2

## Procedure for Drawing a Free-Body Diagram

Since we must account for *all the forces acting on the particle* when applying the equations of equilibrium, the importance of first drawing a free-body diagram cannot be overemphasized. To construct a free-body diagram, the following three steps are necessary.

### Draw Outlined Shape.

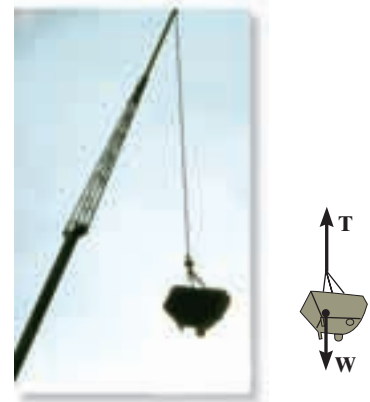
Imagine the particle to be *isolated* or cut “free” from its surroundings. This requires *removing* all the supports and drawing the particle’s outlined shape.

### Show All Forces.

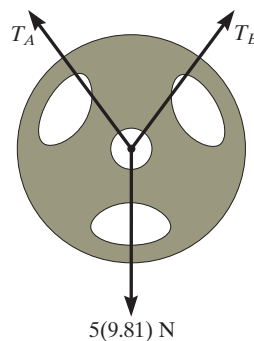
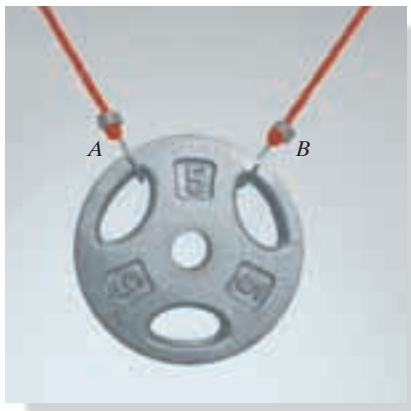
Indicate on this sketch *all* the forces that act *on the particle*. These forces can be *active forces*, which tend to set the particle in motion, or they can be *reactive forces* which are the result of the constraints or supports that tend to prevent motion. To account for all these forces, it may be helpful to trace around the particle’s boundary, carefully noting each force acting on it.

### Identify Each Force.

The forces that are *known* should be labeled with their proper magnitudes and directions. Letters are used to represent the magnitudes and directions of forces that are unknown.



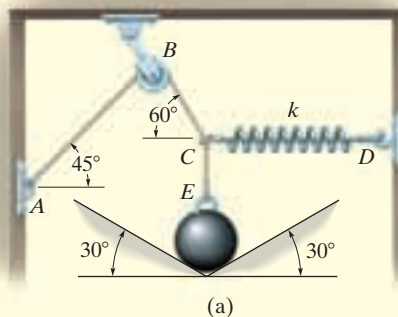
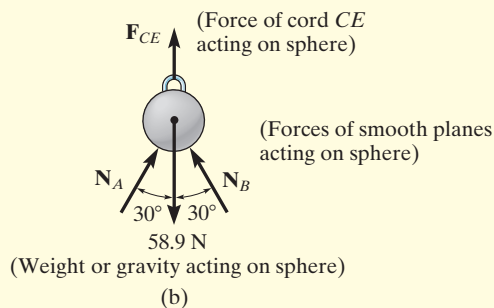
The bucket is held in equilibrium by the cable, and instinctively we know that the force in the cable must equal the weight of the bucket. By drawing a free-body diagram of the bucket we can understand why this is so. This diagram shows that there are only two forces *acting on the bucket*, namely, its weight  $\mathbf{W}$  and the force  $\mathbf{T}$  of the cable. For equilibrium, the resultant of these forces must be equal to zero, and so  $T = W$ . (© Russell C. Hibbeler)



The 5-kg plate is suspended by two straps  $A$  and  $B$ . To find the force in each strap we should consider the free-body diagram of the plate. As noted, the three forces acting on it are concurrent at the center. (© Russell C. Hibbeler)

**EXAMPLE 3.1**

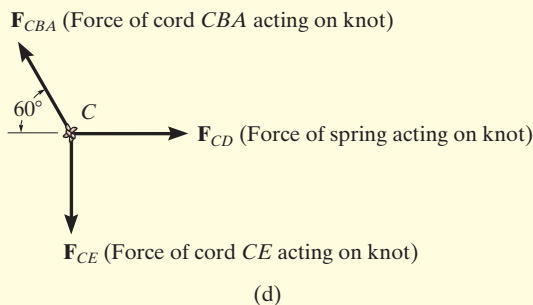
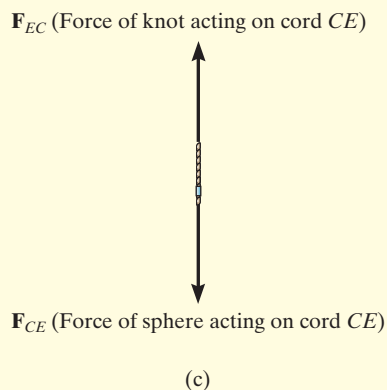
The sphere in Fig. 3–3a has a mass of 6 kg and is supported as shown. Draw a free-body diagram of the sphere, the cord  $CE$ , and the knot at  $C$ .

**SOLUTION**

**Sphere.** Once the supports are *removed*, we can see that there are four forces acting on the sphere, namely, its weight,  $6\text{ kg } (9.81\text{ m/s}^2) = 58.9\text{ N}$ , the force of cord  $CE$ , and the two normal forces caused by the smooth inclined planes. The free-body diagram is shown in Fig. 3–3b.

**Cord  $CE$ .** When the cord  $CE$  is isolated from its surroundings, its free-body diagram shows only two forces acting on it, namely, the force of the sphere and the force of the knot, Fig. 3–3c. Notice that  $F_{CE}$  shown here is equal but opposite to that shown in Fig. 3–3b, a consequence of Newton's third law of action–reaction. Also,  $F_{CE}$  and  $F_{EC}$  pull on the cord and keep it in tension so that it doesn't collapse. For equilibrium,  $F_{CE} = F_{EC}$ .

**Knot.** The knot at  $C$  is subjected to three forces, Fig. 3–3d. They are caused by the cords  $CBA$  and  $CE$  and the spring  $CD$ . As required, the free-body diagram shows all these forces labeled with their magnitudes and directions. It is important to recognize that the weight of the sphere does not directly act on the knot. Instead, the cord  $CE$  subjects the knot to this force.

**Fig. 3–3**

### 3.3 Coplanar Force Systems

If a particle is subjected to a system of coplanar forces that lie in the  $x$ - $y$  plane, as in Fig. 3-4, then each force can be resolved into its  $\mathbf{i}$  and  $\mathbf{j}$  components. For equilibrium, these forces must sum to produce a zero force resultant, i.e.,

$$\begin{aligned}\Sigma \mathbf{F} &= \mathbf{0} \\ \Sigma F_x \mathbf{i} + \Sigma F_y \mathbf{j} &= \mathbf{0}\end{aligned}$$

For this vector equation to be satisfied, the resultant force's  $x$  and  $y$  components must both be equal to zero. Hence,

$$\begin{aligned}\Sigma F_x &= 0 \\ \Sigma F_y &= 0\end{aligned}\quad (3-3)$$

These two equations can be solved for at most two unknowns, generally represented as angles and magnitudes of forces shown on the particle's free-body diagram.

When applying each of the two equations of equilibrium, we must account for the sense of direction of any component by using an *algebraic sign* which corresponds to the arrowhead direction of the component along the  $x$  or  $y$  axis. It is important to note that if a force has an *unknown magnitude*, then the arrowhead sense of the force on the free-body diagram can be *assumed*. Then if the *solution* yields a *negative scalar*, this indicates that the sense of the force is opposite to that which was assumed.

For example, consider the free-body diagram of the particle subjected to the two forces shown in Fig. 3-5. Here it is *assumed* that the *unknown force*  $\mathbf{F}$  acts to the right, that is, in the positive  $x$  direction, to maintain equilibrium. Applying the equation of equilibrium along the  $x$  axis, we have

$$\rightarrow \Sigma F_x = 0; \quad +F + 10 \text{ N} = 0$$

Both terms are “positive” since both forces act in the positive  $x$  direction. When this equation is solved,  $F = -10 \text{ N}$ . Here the *negative sign* indicates that  $\mathbf{F}$  must act to the left to hold the particle in equilibrium, Fig. 3-5. Notice that if the  $+x$  axis in Fig. 3-5 were directed to the left, both terms in the above equation would be negative, but again, after solving,  $F = -10 \text{ N}$ , indicating that  $\mathbf{F}$  would have to be directed to the left.

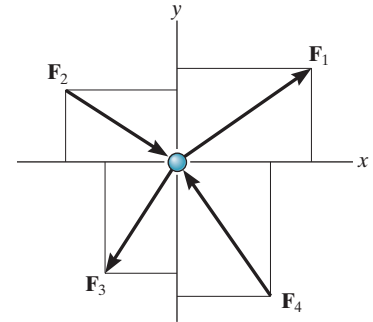


Fig. 3-4

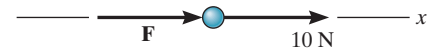
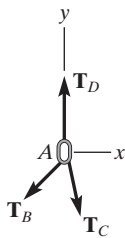
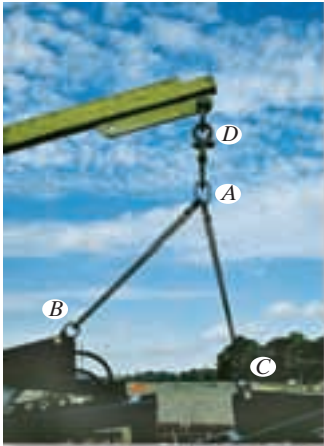


Fig. 3-5

## Important Points

The first step in solving any equilibrium problem is to draw the particle's free-body diagram. This requires *removing all the supports* and isolating or freeing the particle from its surroundings and then showing all the forces that act on it.

Equilibrium means the particle is at rest or moving at constant velocity. In two dimensions, the necessary and sufficient conditions for equilibrium require  $\Sigma F_x = 0$  and  $\Sigma F_y = 0$ .



The chains exert three forces on the ring at  $A$ , as shown on its free-body diagram. The ring will not move, or will move with constant velocity, provided the summation of these forces along the  $x$  and along the  $y$  axis equals zero. If one of the three forces is known, the magnitudes of the other two forces can be obtained from the two equations of equilibrium. (© Russell C. Hibbeler)

## Procedure for Analysis

Coplanar force equilibrium problems for a particle can be solved using the following procedure.

### Free-Body Diagram.

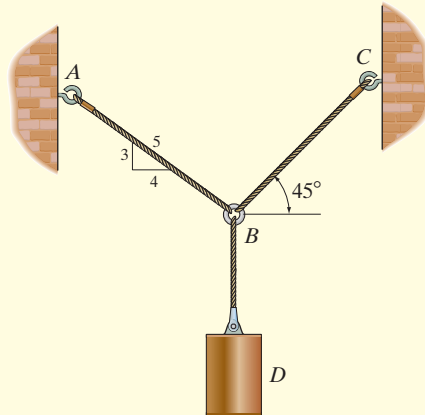
- Establish the  $x, y$  axes in any suitable orientation.
- Label all the known and unknown force magnitudes and directions on the diagram.
- The sense of a force having an unknown magnitude can be assumed.

### Equations of Equilibrium.

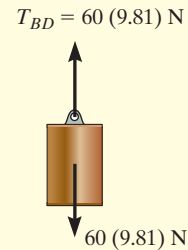
- Apply the equations of equilibrium,  $\Sigma F_x = 0$  and  $\Sigma F_y = 0$ . For convenience, arrows can be written alongside each equation to define the positive directions.
- Components are positive if they are directed along a positive axis, and negative if they are directed along a negative axis.
- If more than two unknowns exist and the problem involves a spring, apply  $F = ks$  to relate the spring force to the deformation  $s$  of the spring.
- Since the magnitude of a force is always a positive quantity, then if the solution for a force yields a negative result, this indicates that its sense is the reverse of that shown on the free-body diagram.

**EXAMPLE 3.2**

Determine the tension in cables  $BA$  and  $BC$  necessary to support the 60-kg cylinder in Fig. 3–6a.



(a)



(b)

**SOLUTION**

**Free-Body Diagram.** Due to equilibrium, the weight of the cylinder causes the tension in cable  $BD$  to be  $T_{BD} = 60(9.81)$  N, Fig. 3–6b. The forces in cables  $BA$  and  $BC$  can be determined by investigating the equilibrium of ring  $B$ . Its free-body diagram is shown in Fig. 3–6c. The magnitudes of  $T_A$  and  $T_C$  are unknown, but their directions are known.

**Equations of Equilibrium.** Applying the equations of equilibrium along the  $x$  and  $y$  axes, we have

$$\rightarrow \Sigma F_x = 0; \quad T_C \cos 45^\circ - \left(\frac{4}{5}\right)T_A = 0 \quad (1)$$

$$+ \uparrow \Sigma F_y = 0; \quad T_C \sin 45^\circ + \left(\frac{3}{5}\right)T_A - 60(9.81) \text{ N} = 0 \quad (2)$$

Equation (1) can be written as  $T_A = 0.8839T_C$ . Substituting this into Eq. (2) yields

$$T_C \sin 45^\circ + \left(\frac{3}{5}\right)(0.8839T_C) - 60(9.81) \text{ N} = 0$$

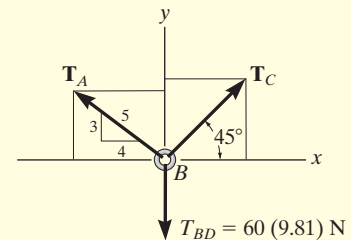
so that

$$T_C = 475.66 \text{ N} = 476 \text{ N} \quad \text{Ans.}$$

Substituting this result into either Eq. (1) or Eq. (2), we get

$$T_A = 420 \text{ N} \quad \text{Ans.}$$

**NOTE:** The accuracy of these results, of course, depends on the accuracy of the data, i.e., measurements of geometry and loads. For most engineering work involving a problem such as this, the data as measured to three significant figures would be sufficient.



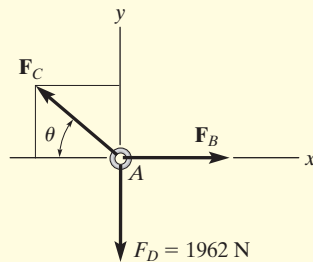
(c)

**Fig. 3–6**



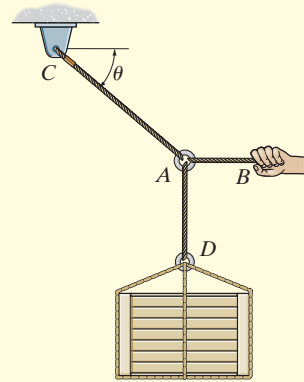
## EXAMPLE 3.3

The 200-kg crate in Fig. 3–7a is suspended using the ropes  $AB$  and  $AC$ . Each rope can withstand a maximum force of 10 kN before it breaks. If  $AB$  always remains horizontal, determine the smallest angle  $\theta$  to which the crate can be suspended before one of the ropes breaks.



(b)

Fig. 3–7



(a)

## SOLUTION

**Free-Body Diagram.** We will study the equilibrium of ring  $A$ . There are three forces acting on it, Fig. 3–7b. The magnitude of  $\mathbf{F}_D$  is equal to the weight of the crate, i.e.,  $F_D = 200(9.81) \text{ N} = 1962 \text{ N} < 10 \text{ kN}$ .

**Equations of Equilibrium.** Applying the equations of equilibrium along the  $x$  and  $y$  axes,

$$\rightarrow \Sigma F_x = 0; \quad -F_C \cos \theta + F_B = 0; \quad F_C = \frac{F_B}{\cos \theta} \quad (1)$$

$$+\uparrow \Sigma F_y = 0; \quad F_C \sin \theta - 1962 \text{ N} = 0 \quad (2)$$

From Eq. (1),  $F_C$  is always greater than  $F_B$  since  $\cos \theta \leq 1$ . Therefore, rope  $AC$  will reach the maximum tensile force of 10 kN *before* rope  $AB$ . Substituting  $F_C = 10 \text{ kN}$  into Eq. (2), we get

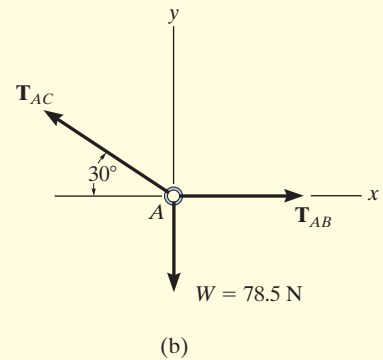
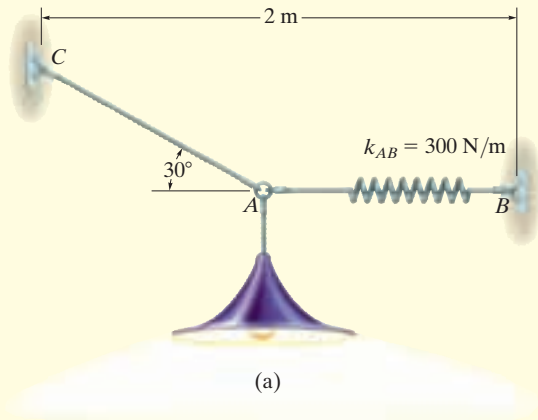
$$\begin{aligned} [10(10^3) \text{ N}] \sin \theta - 1962 \text{ N} &= 0 \\ \theta &= \sin^{-1}(0.1962) = 11.31^\circ = 11.3^\circ \end{aligned} \quad \text{Ans.}$$

The force developed in rope  $AB$  can be obtained by substituting the values for  $\theta$  and  $F_C$  into Eq. (1).

$$\begin{aligned} 10(10^3) \text{ N} &= \frac{F_B}{\cos 11.31^\circ} \\ F_B &= 9.81 \text{ kN} \end{aligned}$$

**EXAMPLE 3.4**

Determine the required length of cord  $AC$  in Fig. 3–8a so that the 8-kg lamp can be suspended in the position shown. The *undeformed* length of spring  $AB$  is  $l'_{AB} = 0.4$  m, and the spring has a stiffness of  $k_{AB} = 300$  N/m.

**Fig. 3–8****SOLUTION**

If the force in spring  $AB$  is known, the stretch of the spring can be found using  $F = ks$ . From the problem geometry, it is then possible to calculate the required length of  $AC$ .

**Free-Body Diagram.** The lamp has a weight  $W = 8(9.81) = 78.5$  N and so the free-body diagram of the ring at  $A$  is shown in Fig. 3–8b.

**Equations of Equilibrium.** Using the  $x, y$  axes,

$$\begin{aligned} \rightarrow \Sigma F_x &= 0; & T_{AB} - T_{AC} \cos 30^\circ &= 0 \\ + \uparrow \Sigma F_y &= 0; & T_{AC} \sin 30^\circ - 78.5 \text{ N} &= 0 \end{aligned}$$

Solving, we obtain

$$T_{AC} = 157.0 \text{ N}$$

$$T_{AB} = 135.9 \text{ N}$$

The stretch of spring  $AB$  is therefore

$$\begin{aligned} T_{AB} &= k_{AB}s_{AB}; & 135.9 \text{ N} &= 300 \text{ N/m}(s_{AB}) \\ s_{AB} &= 0.453 \text{ m} \end{aligned}$$

so the stretched length is

$$\begin{aligned} l_{AB} &= l'_{AB} + s_{AB} \\ l_{AB} &= 0.4 \text{ m} + 0.453 \text{ m} = 0.853 \text{ m} \end{aligned}$$

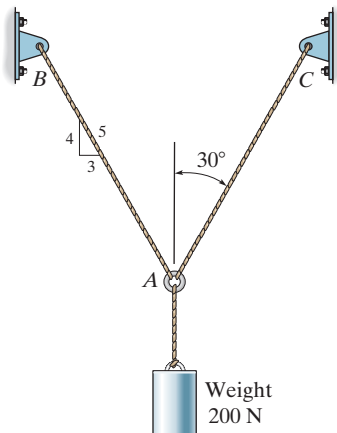
The horizontal distance from  $C$  to  $B$ , Fig. 3–8a, requires

$$\begin{aligned} 2 \text{ m} &= l_{AC} \cos 30^\circ + 0.853 \text{ m} \\ l_{AC} &= 1.32 \text{ m} \end{aligned}$$

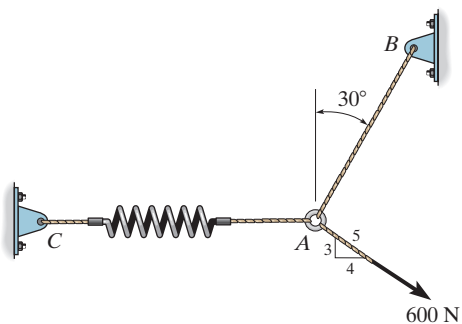
*Ans.*

## PRELIMINARY PROBLEMS

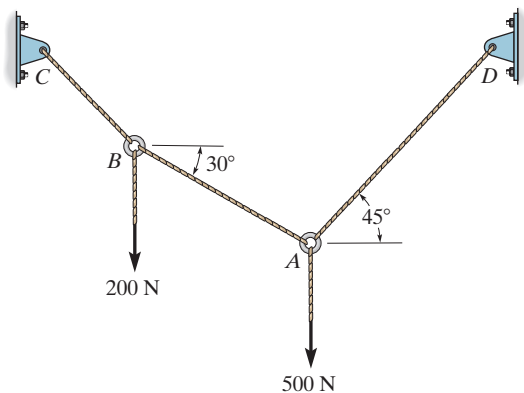
**P3-1.** In each case, draw a free-body diagram of the ring at  $A$  and identify each force.



(a)



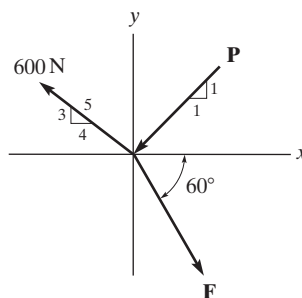
(b)



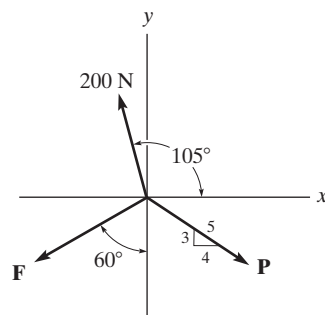
(c)

**Prob. P3-1**

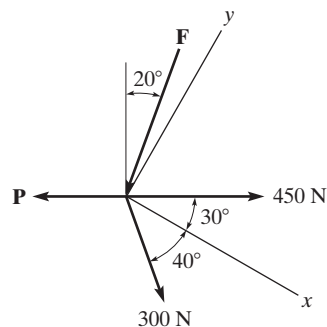
**P3-2.** Write the two equations of equilibrium,  $\Sigma F_x = 0$  and  $\Sigma F_y = 0$ . Do not solve.



(a)



(b)



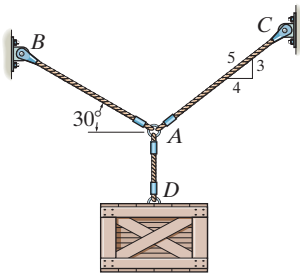
(c)

**Prob. P3-2**

## FUNDAMENTAL PROBLEMS

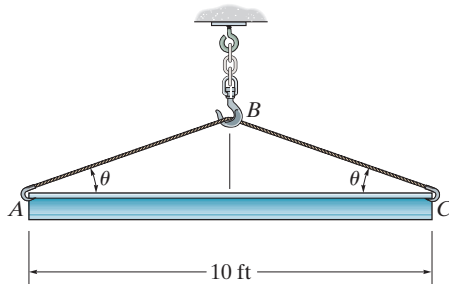
*All problem solutions must include an FBD.*

**F3-1.** The crate has a weight of 550 lb. Determine the force in each supporting cable.



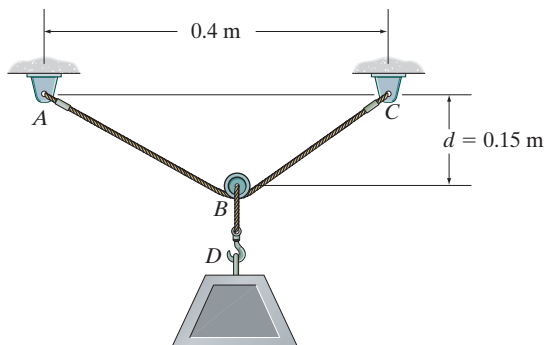
**Prob. F3-1**

**F3-2.** The beam has a weight of 700 lb. Determine the shortest cable ABC that can be used to lift it if the maximum force the cable can sustain is 1500 lb.



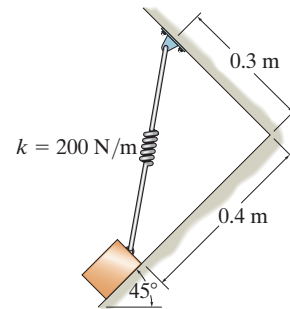
**Prob. F3-2**

**F3-3.** If the 5-kg block is suspended from the pulley B and the sag of the cord is  $d = 0.15$  m, determine the force in cord ABC. Neglect the size of the pulley.



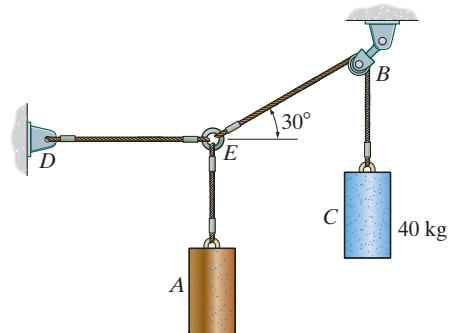
**Prob. F3-3**

**F3-4.** The block has a mass of 5 kg and rests on the smooth plane. Determine the unstretched length of the spring.



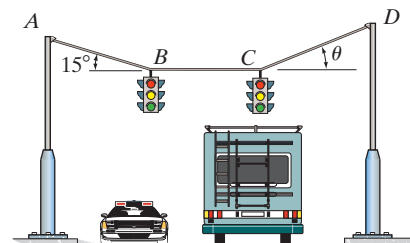
**Prob. F3-4**

**F3-5.** If the mass of cylinder C is 40 kg, determine the mass of cylinder A in order to hold the assembly in the position shown.



**Prob. F3-5**

**F3-6.** Determine the tension in cables AB, BC, and CD, necessary to support the 10-kg and 15-kg traffic lights at B and C, respectively. Also, find the angle  $\theta$ .



**Prob. F3-6**

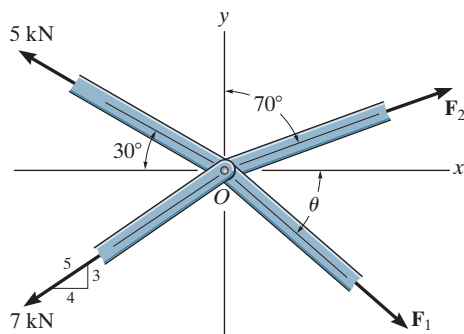
## PROBLEMS

*All problem solutions must include an FBD.*

**3-1.** The members of a truss are pin connected at joint  $O$ . Determine the magnitudes of  $F_1$  and  $F_2$  for equilibrium. Set  $\theta = 60^\circ$ .

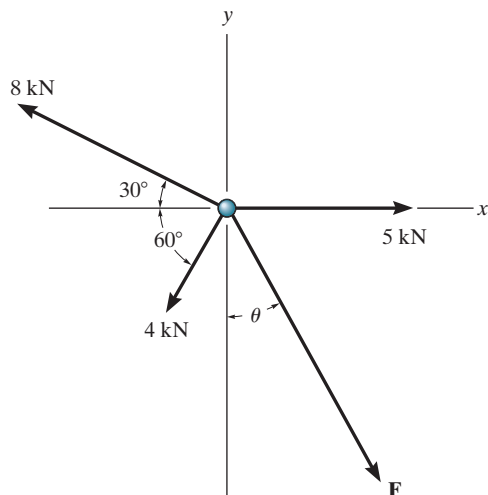
3

**3-2.** The members of a truss are pin connected at joint  $O$ . Determine the magnitude of  $F_1$  and its angle  $\theta$  for equilibrium. Set  $F_2 = 6 \text{ kN}$ .



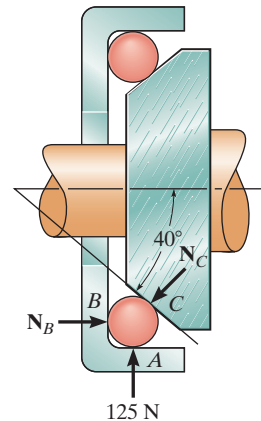
**Probs. 3-1/2**

**3-3.** Determine the magnitude and direction  $\theta$  of  $\mathbf{F}$  so that the particle is in equilibrium.



**Prob. 3-3**

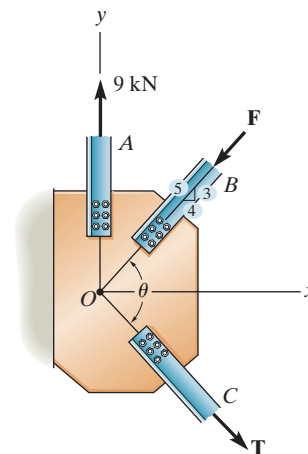
**\*3-4.** The bearing consists of rollers, symmetrically confined within the housing. The bottom one is subjected to a 125-N force at its contact  $A$  due to the load on the shaft. Determine the normal reactions  $N_B$  and  $N_C$  on the bearing at its contact points  $B$  and  $C$  for equilibrium.



**Prob. 3-4**

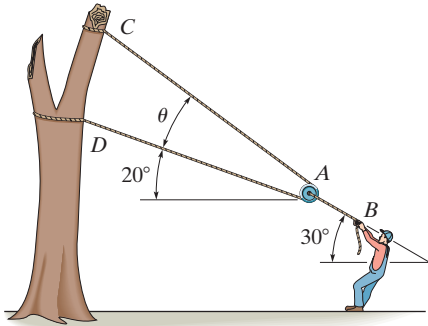
**3-5.** The members of a truss are connected to the gusset plate. If the forces are concurrent at point  $O$ , determine the magnitudes of  $\mathbf{F}$  and  $\mathbf{T}$  for equilibrium. Take  $\theta = 90^\circ$ .

**3-6.** The gusset plate is subjected to the forces of three members. Determine the tension force in member  $C$  and its angle  $\theta$  for equilibrium. The forces are concurrent at point  $O$ . Take  $F = 8 \text{ kN}$ .



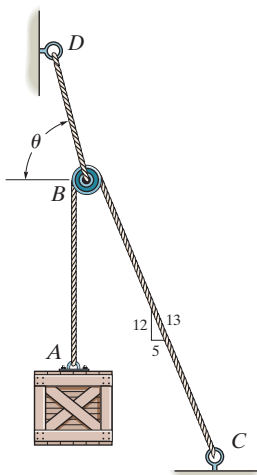
**Probs. 3-5/6**

**3-7.** The man attempts to pull down the tree using the cable and *small* pulley arrangement shown. If the tension in *AB* is 60 lb, determine the tension in cable *CAD* and the angle  $\theta$  which the cable makes at the pulley.



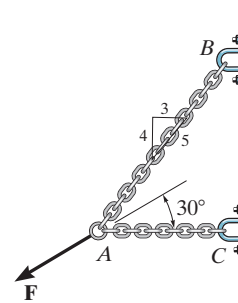
**Prob. 3-7**

**\*3-8.** The cords *ABC* and *BD* can each support a maximum load of 100 lb. Determine the maximum weight of the crate, and the angle  $\theta$  for equilibrium.



**Prob. 3-8**

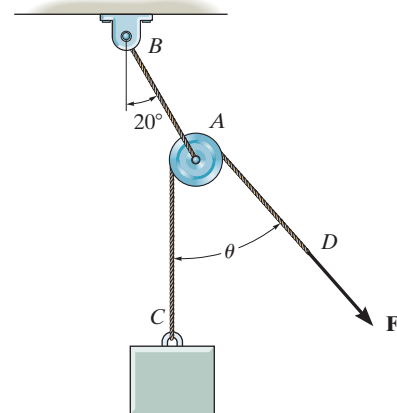
**3-9.** Determine the maximum force  $\mathbf{F}$  that can be supported in the position shown if each chain can support a maximum tension of 600 lb before it fails.



**Prob. 3-9**

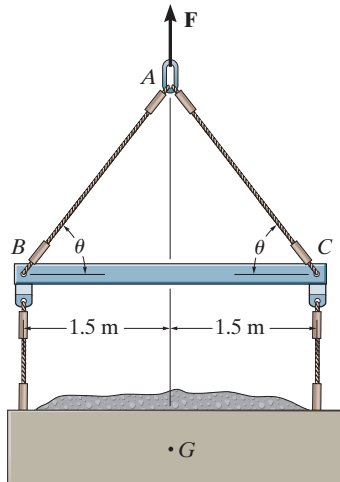
**3-10.** The block has a weight of 20 lb and is being hoisted at uniform velocity. Determine the angle  $\theta$  for equilibrium and the force in cord *AB*.

**3-11.** Determine the maximum weight  $W$  of the block that can be suspended in the position shown if cords *AB* and *CAD* can each support a maximum tension of 80 lb. Also, what is the angle  $\theta$  for equilibrium?



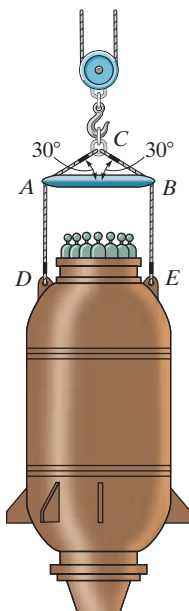
**Probs. 3-10/11**

**3-12.** The lift sling is used to hoist a container having a mass of 500 kg. Determine the force in each of the cables  $AB$  and  $AC$  as a function of  $\theta$ . If the maximum tension allowed in each cable is 5 kN, determine the shortest length of cables  $AB$  and  $AC$  that can be used for the lift. The center of gravity of the container is located at  $G$ .



**Prob. 3-12**

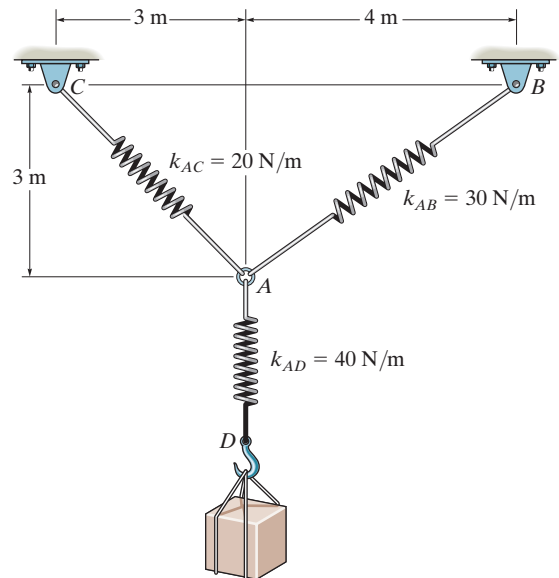
**3-13.** A nuclear-reactor vessel has a weight of  $500(10^3)$  lb. Determine the horizontal compressive force that the spreader bar  $AB$  exerts on point  $A$  and the force that each cable segment  $CA$  and  $AD$  exert on this point while the vessel is hoisted upward at constant velocity.



**Prob. 3-13**

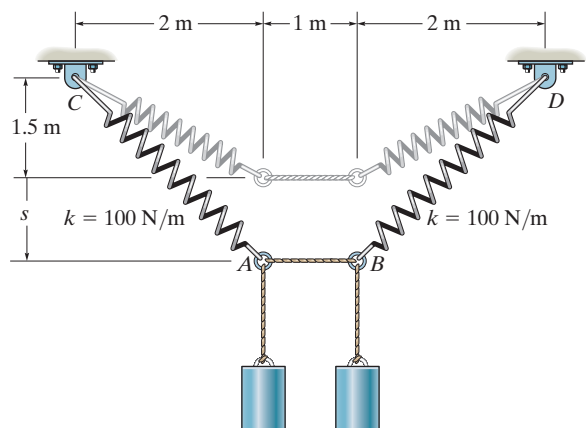
**3-14.** Determine the stretch in each spring for equilibrium of the 2-kg block. The springs are shown in the equilibrium position.

**3-15.** The unstretched length of spring  $AB$  is 3 m. If the block is held in the equilibrium position shown, determine the mass of the block at  $D$ .



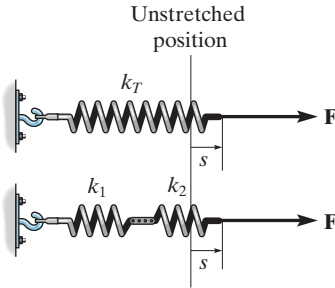
**Probs. 3-14/15**

**\*3-16.** Determine the mass of each of the two cylinders if they cause a sag of  $s = 0.5$  m when suspended from the rings at  $A$  and  $B$ . Note that  $s = 0$  when the cylinders are removed.



**Prob. 3-16**

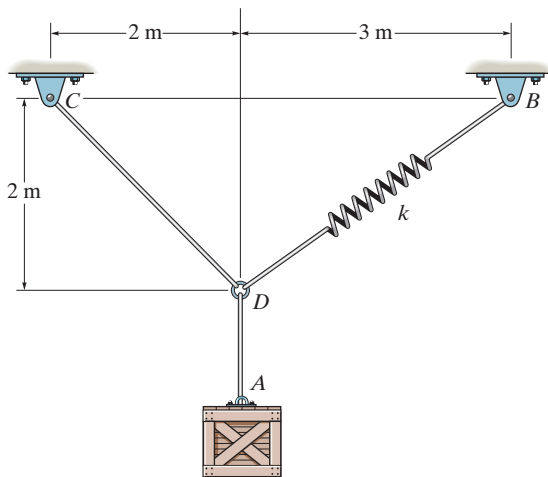
**3-17.** Determine the stiffness  $k_T$  of the single spring such that the force  $\mathbf{F}$  will stretch it by the same amount  $s$  as the force  $\mathbf{F}$  stretches the two springs. Express  $k_T$  in terms of stiffness  $k_1$  and  $k_2$  of the two springs.



**Prob. 3-17**

**3-18.** If the spring  $DB$  has an unstretched length of 2 m, determine the stiffness of the spring to hold the 40-kg crate in the position shown.

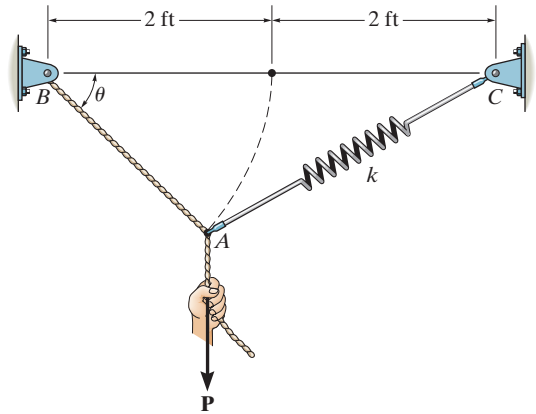
**3-19.** Determine the unstretched length of  $DB$  to hold the 40-kg crate in the position shown. Take  $k = 180 \text{ N/m}$ .



**Probs. 3-18/19**

**\*3-20.** A vertical force  $P = 10 \text{ lb}$  is applied to the ends of the 2-ft cord  $AB$  and spring  $AC$ . If the spring has an unstretched length of 2 ft, determine the angle  $\theta$  for equilibrium. Take  $k = 15 \text{ lb/ft}$ .

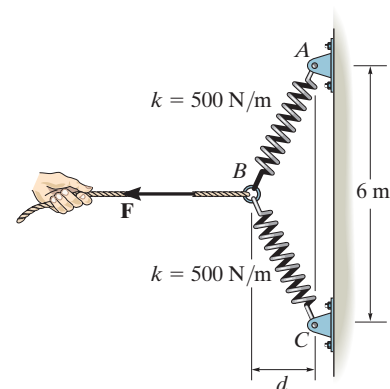
**3-21.** Determine the unstretched length of spring  $AC$  if a force  $P = 80 \text{ lb}$  causes the angle  $\theta = 60^\circ$  for equilibrium. Cord  $AB$  is 2 ft long. Take  $k = 50 \text{ lb/ft}$ .



**Probs. 3-20/21**

**3-22.** The springs  $BA$  and  $BC$  each have a stiffness of  $500 \text{ N/m}$  and an unstretched length of 3 m. Determine the horizontal force  $\mathbf{F}$  applied to the cord which is attached to the small ring  $B$  so that the displacement of  $AB$  from the wall is  $d = 1.5 \text{ m}$ .

**3-23.** The springs  $BA$  and  $BC$  each have a stiffness of  $500 \text{ N/m}$  and an unstretched length of 3 m. Determine the displacement  $d$  of the cord from the wall when a force  $F = 175 \text{ N}$  is applied to the cord.

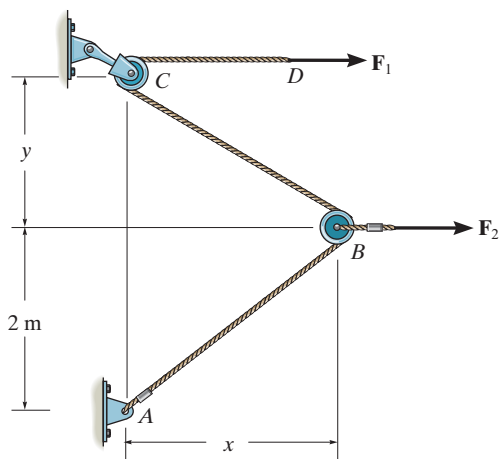


**Probs. 3-22/23**



**\*3-24.** Determine the distances  $x$  and  $y$  for equilibrium if  $F_1 = 800$  N and  $F_2 = 1000$  N.

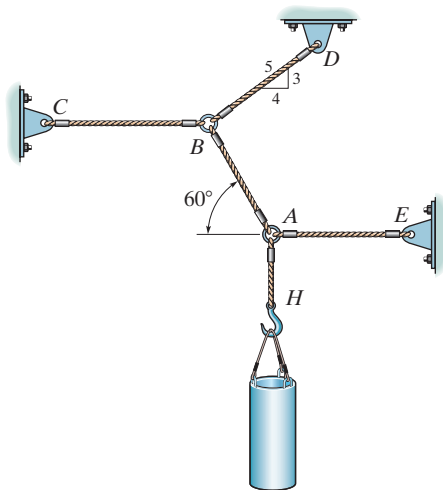
**3-25.** Determine the magnitude of  $F_1$  and the distance  $y$  if  $x = 1.5$  m and  $F_2 = 1000$  N.



Probs. 3-24/25

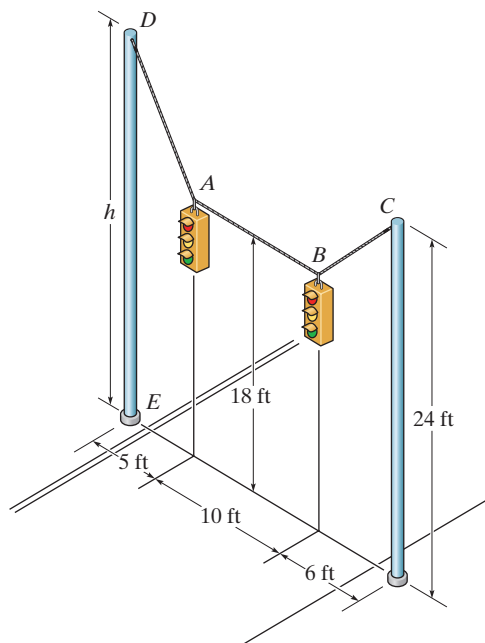
**3-26.** The 30-kg pipe is supported at  $A$  by a system of five cords. Determine the force in each cord for equilibrium.

**3-27.** Each cord can sustain a maximum tension of 500 N. Determine the largest mass of pipe that can be supported.



Probs. 3-26/27

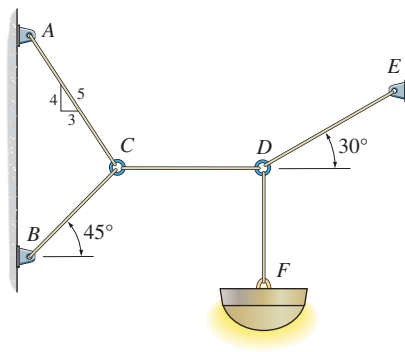
**\*3-28.** The street-lights at  $A$  and  $B$  are suspended from the two poles as shown. If each light has a weight of 50 lb, determine the tension in each of the three supporting cables and the required height  $h$  of the pole  $DE$  so that cable  $AB$  is horizontal.



Prob. 3-28

**3-29.** Determine the tension developed in each cord required for equilibrium of the 20-kg lamp.

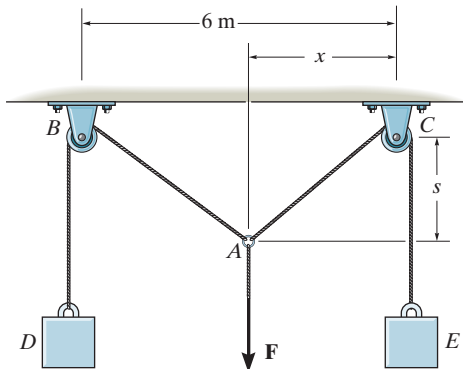
**3-30.** Determine the maximum mass of the lamp that the cord system can support so that no single cord develops a tension exceeding 400 N.



Probs. 3-29/30

**3–31.** Blocks  $D$  and  $E$  have a mass of 4 kg and 6 kg, respectively. If  $x = 2$  m determine the force  $\mathbf{F}$  and the sag  $s$  for equilibrium.

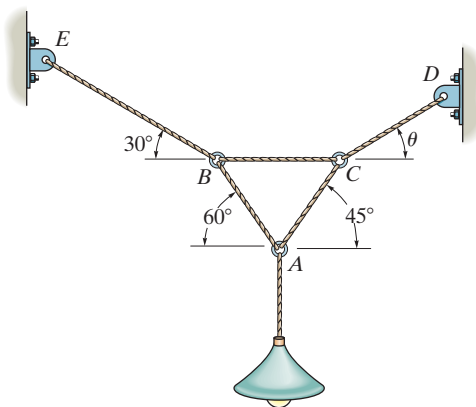
**\*3–32.** Blocks  $D$  and  $E$  have a mass of 4 kg and 6 kg, respectively. If  $F = 80$  N, determine the sag  $s$  and distance  $x$  for equilibrium.



**Probs. 3–31/32**

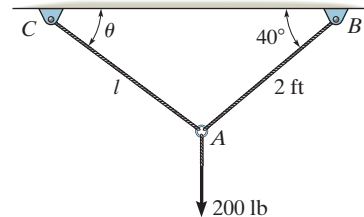
**3–33.** The lamp has a weight of 15 lb and is supported by the six cords connected together as shown. Determine the tension in each cord and the angle  $\theta$  for equilibrium. Cord  $BC$  is horizontal.

**3–34.** Each cord can sustain a maximum tension of 20 lb. Determine the largest weight of the lamp that can be supported. Also, determine  $\theta$  of cord  $DC$  for equilibrium.



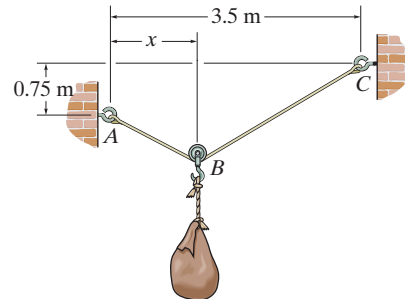
**Probs. 3–33/34**

**3–35.** The ring of negligible size is subjected to a vertical force of 200 lb. Determine the required length  $l$  of cord  $AC$  such that the tension acting in  $AC$  is 160 lb. Also, what is the force in cord  $AB$ ? *Hint:* Use the equilibrium condition to determine the required angle  $\theta$  for attachment, then determine  $l$  using trigonometry applied to triangle  $ABC$ .



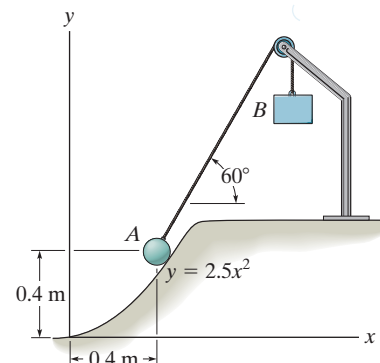
**Prob. 3–35**

**\*3–36.** Cable  $ABC$  has a length of 5 m. Determine the position  $x$  and the tension developed in  $ABC$  required for equilibrium of the 100-kg sack. Neglect the size of the pulley at  $B$ .



**Prob. 3–36**

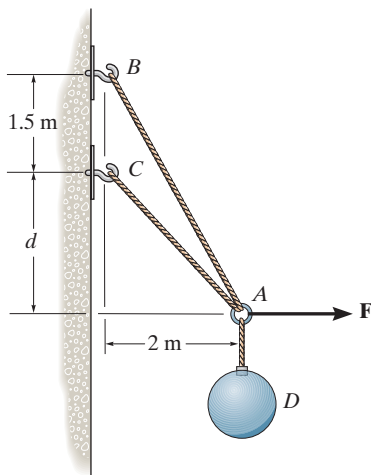
**3–37.** A 4-kg sphere rests on the smooth parabolic surface. Determine the normal force it exerts on the surface and the mass  $m_B$  of block  $B$  needed to hold it in the equilibrium position shown.



**Prob. 3–37**

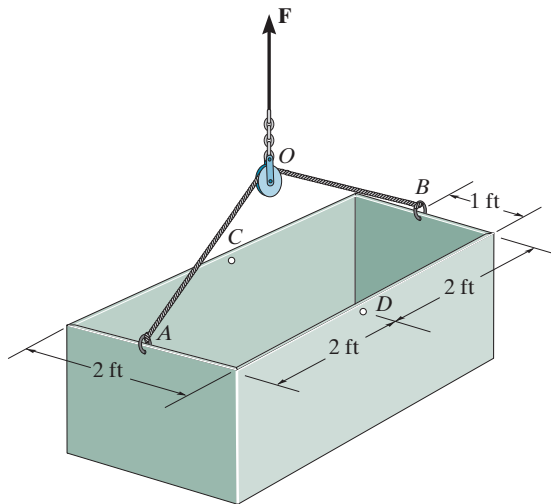
**3–38.** Determine the forces in cables  $AC$  and  $AB$  needed to hold the 20-kg ball  $D$  in equilibrium. Take  $F = 300$  N and  $d = 1$  m.

**3–39.** The ball  $D$  has a mass of 20 kg. If a force of  $F = 100$  N is applied horizontally to the ring at  $A$ , determine the dimension  $d$  so that the force in cable  $AC$  is zero.



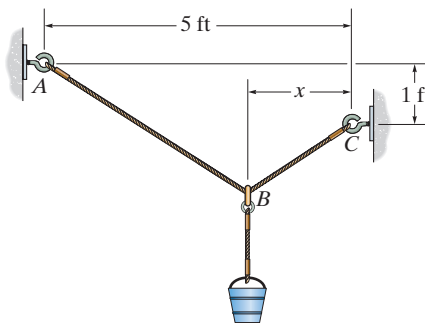
**Probs. 3–38/39**

**\*3–40.** The 200-lb uniform container is suspended by means of a 6-ft-long cable, which is attached to the sides of the tank and passes over the small pulley located at  $O$ . If the cable can be attached at either points  $A$  and  $B$ , or  $C$  and  $D$ , determine which attachment produces the least amount of tension in the cable. What is this tension?



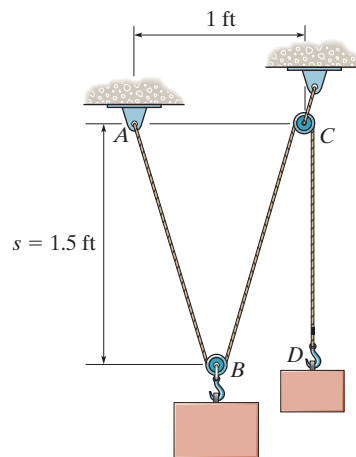
**Prob. 3–40**

**3–41.** The single elastic cord  $ABC$  is used to support the 40-lb load. Determine the position  $x$  and the tension in the cord that is required for equilibrium. The cord passes through the smooth ring at  $B$  and has an unstretched length of 6 ft and stiffness of  $k = 50$  lb/ft.



**Prob. 3–41**

**3–42.** A “scale” is constructed with a 4-ft-long cord and the 10-lb block  $D$ . The cord is fixed to a pin at  $A$  and passes over two small pulleys. Determine the weight of the suspended block  $B$  if the system is in equilibrium when  $s = 1.5$  ft.



**Prob. 3–42**

## CONCEPTUAL PROBLEMS

**C3-1.** The concrete wall panel is hoisted into position using the two cables  $AB$  and  $AC$  of equal length. Establish appropriate dimensions and use an equilibrium analysis to show that the longer the cables the less the force in each cable.



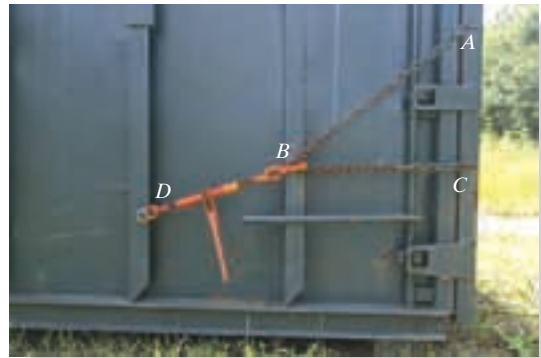
**Prob. C3-1** (© Russell C. Hibbeler)

**C3-2.** The hoisting cables  $BA$  and  $BC$  each have a length of 20 ft. If the maximum tension that can be supported by each cable is 900 lb, determine the maximum distance  $AC$  between them in order to lift the uniform 1200-lb truss with constant velocity.



**Prob. C3-2** (© Russell C. Hibbeler)

**C3-3.** The device  $DB$  is used to pull on the chain  $ABC$  to hold a door closed on the bin. If the angle between  $AB$  and  $BC$  is  $30^\circ$ , determine the angle between  $DB$  and  $BC$  for equilibrium.



**Prob. C3-3** (© Russell C. Hibbeler)

**C3-4.** Chain  $AB$  is 1 m long and chain  $AC$  is 1.2 m long. If the distance  $BC$  is 1.5 m, and  $AB$  can support a maximum force of 2 kN, whereas  $AC$  can support a maximum force of 0.8 kN, determine the largest vertical force  $F$  that can be applied to the link at  $A$ .



**Prob. C3-4** (© Russell C. Hibbeler)

### 3.4 Three-Dimensional Force Systems

In Section 3.1 we stated that the necessary and sufficient condition for particle equilibrium is

$$\Sigma \mathbf{F} = \mathbf{0} \quad (3-4)$$

In the case of a three-dimensional force system, as in Fig. 3–9, we can resolve the forces into their respective  $\mathbf{i}$ ,  $\mathbf{j}$ ,  $\mathbf{k}$  components, so that  $\Sigma F_x \mathbf{i} + \Sigma F_y \mathbf{j} + \Sigma F_z \mathbf{k} = \mathbf{0}$ . To satisfy this equation we require

$$\begin{aligned} \Sigma F_x &= 0 \\ \Sigma F_y &= 0 \\ \Sigma F_z &= 0 \end{aligned} \quad (3-5)$$

These three equations state that the *algebraic sum* of the components of all the forces acting on the particle along each of the coordinate axes must be zero. Using them we can solve for at most three unknowns, generally represented as coordinate direction angles or magnitudes of forces shown on the particle's free-body diagram.

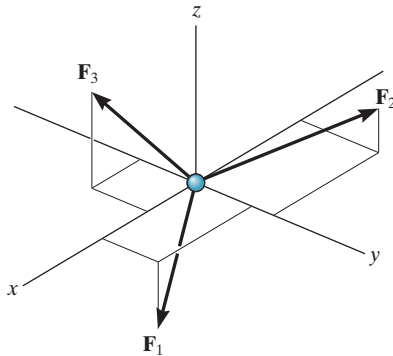


Fig. 3–9

#### Procedure for Analysis

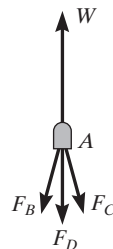
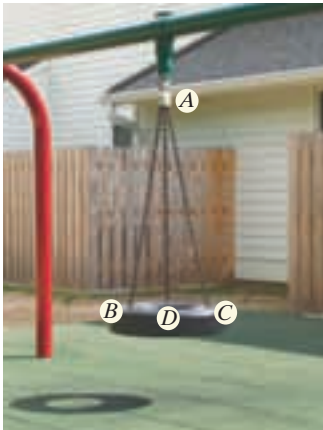
Three-dimensional force equilibrium problems for a particle can be solved using the following procedure.

##### Free-Body Diagram.

- Establish the  $x$ ,  $y$ ,  $z$  axes in any suitable orientation.
- Label all the known and unknown force magnitudes and directions on the diagram.
- The sense of a force having an unknown magnitude can be assumed.

##### Equations of Equilibrium.

- Use the scalar equations of equilibrium,  $\Sigma F_x = 0$ ,  $\Sigma F_y = 0$ ,  $\Sigma F_z = 0$ , in cases where it is easy to resolve each force into its  $x$ ,  $y$ ,  $z$  components.
- If the three-dimensional geometry appears difficult, then first express each force on the free-body diagram as a Cartesian vector, substitute these vectors into  $\Sigma \mathbf{F} = \mathbf{0}$ , and then set the  $\mathbf{i}$ ,  $\mathbf{j}$ ,  $\mathbf{k}$  components equal to zero.
- If the solution for a force yields a negative result, this indicates that its sense is the reverse of that shown on the free-body diagram.



The joint at  $A$  is subjected to the force from the support as well as forces from each of the three chains. If the tire and any load on it have a weight  $W$ , then the force at the support will be  $\mathbf{W}$ , and the three scalar equations of equilibrium can be applied to the free-body diagram of the joint in order to determine the chain forces,  $\mathbf{F}_B$ ,  $\mathbf{F}_C$ , and  $\mathbf{F}_D$ . (© Russell C. Hibbeler)

**EXAMPLE 3.5**

A 90-lb load is suspended from the hook shown in Fig. 3–10a. If the load is supported by two cables and a spring having a stiffness  $k = 500 \text{ lb/ft}$ , determine the force in the cables and the stretch of the spring for equilibrium. Cable  $AD$  lies in the  $x$ – $y$  plane and cable  $AC$  lies in the  $x$ – $z$  plane.

**SOLUTION**

The stretch of the spring can be determined once the force in the spring is determined.

**Free-Body Diagram.** The connection at  $A$  is chosen for the equilibrium analysis since the cable forces are concurrent at this point. The free-body diagram is shown in Fig. 3–10b.

**Equations of Equilibrium.** By inspection, each force can easily be resolved into its  $x$ ,  $y$ ,  $z$  components, and therefore the three scalar equations of equilibrium can be used. Considering components directed along each positive axis as “positive,” we have

$$\Sigma F_x = 0; \quad F_D \sin 30^\circ - \left(\frac{4}{5}\right) F_C = 0 \quad (1)$$

$$\Sigma F_y = 0; \quad -F_D \cos 30^\circ + F_B = 0 \quad (2)$$

$$\Sigma F_z = 0; \quad \left(\frac{3}{5}\right) F_C - 90 \text{ lb} = 0 \quad (3)$$

Solving Eq. (3) for  $F_C$ , then Eq. (1) for  $F_D$ , and finally Eq. (2) for  $F_B$ , yields

$$F_C = 150 \text{ lb} \quad \text{Ans.}$$

$$F_D = 240 \text{ lb} \quad \text{Ans.}$$

$$F_B = 207.8 \text{ lb} = 208 \text{ lb} \quad \text{Ans.}$$

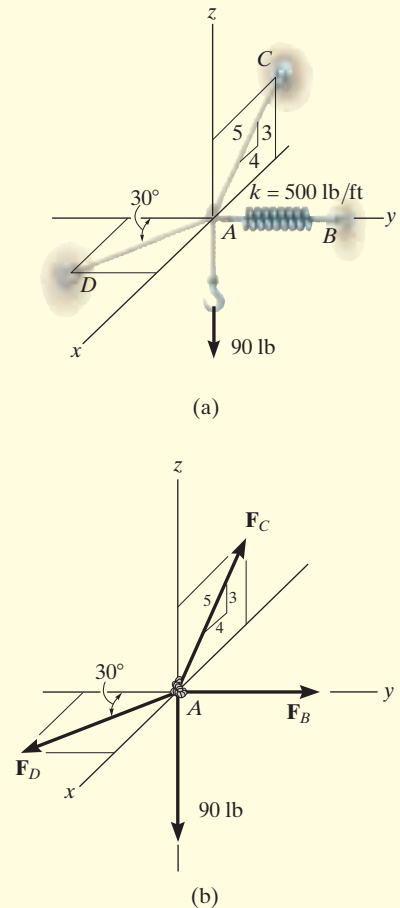
The stretch of the spring is therefore

$$F_B = k s_{AB}$$

$$207.8 \text{ lb} = (500 \text{ lb/ft})(s_{AB})$$

$$s_{AB} = 0.416 \text{ ft} \quad \text{Ans.}$$

**NOTE:** Since the results for all the cable forces are positive, each cable is in tension; that is, it pulls on point  $A$  as expected, Fig. 3–10b.



**Fig. 3–10**

## EXAMPLE 3.6

The 10-kg lamp in Fig. 3–11a is suspended from the three equal-length cords. Determine its smallest vertical distance  $s$  from the ceiling if the force developed in any cord is not allowed to exceed 50 N.

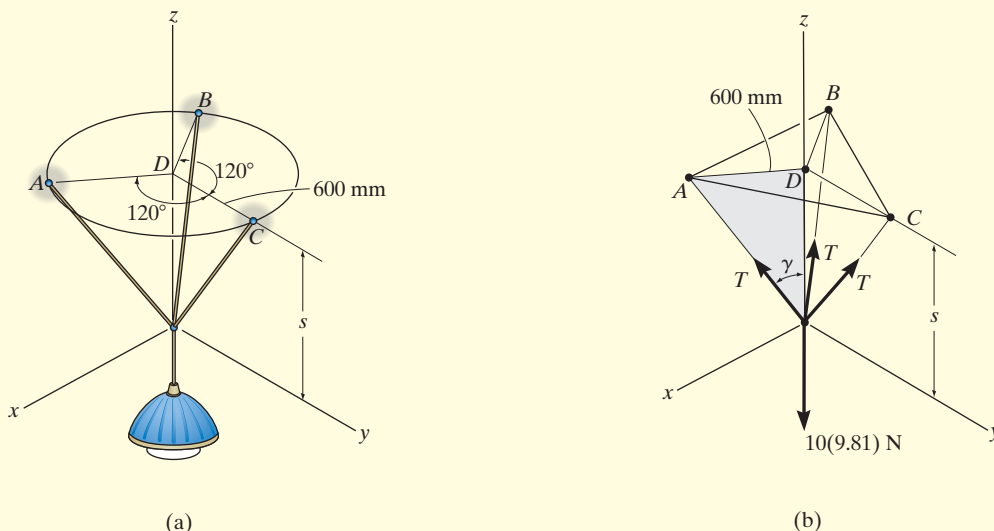


Fig. 3–11

## SOLUTION

**Free-Body Diagram.** Due to symmetry, Fig. 3–11b, the distance  $DA = DB = DC = 600$  mm. It follows that from  $\sum F_x = 0$  and  $\sum F_y = 0$ , the tension  $T$  in each cord will be the same. Also, the angle between each cord and the  $z$  axis is  $\gamma$ .

**Equation of Equilibrium.** Applying the equilibrium equation along the  $z$  axis, with  $T = 50$  N, we have

$$\sum F_z = 0; \quad 3[(50 \text{ N}) \cos \gamma] - 10(9.81) \text{ N} = 0$$

$$\gamma = \cos^{-1} \frac{98.1}{150} = 49.16^\circ$$

From the shaded triangle shown in Fig. 3–11b,

$$\tan 49.16^\circ = \frac{600 \text{ mm}}{s}$$

$$s = 519 \text{ mm}$$

*Ans.*

**EXAMPLE 3.7**

Determine the force in each cable used to support the 40-lb crate shown in Fig. 3–12a.

**SOLUTION**

**Free-Body Diagram.** As shown in Fig. 3–12b, the free-body diagram of point A is considered in order to “expose” the three unknown forces in the cables.

**Equations of Equilibrium.** First we will express each force in Cartesian vector form. Since the coordinates of points B and C are  $B(-3 \text{ ft}, -4 \text{ ft}, 8 \text{ ft})$  and  $C(-3 \text{ ft}, 4 \text{ ft}, 8 \text{ ft})$ , we have

$$\begin{aligned}\mathbf{F}_B &= F_B \left[ \frac{-3\mathbf{i} - 4\mathbf{j} + 8\mathbf{k}}{\sqrt{(-3)^2 + (-4)^2 + (8)^2}} \right] \\ &= -0.318F_B\mathbf{i} - 0.424F_B\mathbf{j} + 0.848F_B\mathbf{k}\end{aligned}$$

$$\begin{aligned}\mathbf{F}_C &= F_C \left[ \frac{-3\mathbf{i} + 4\mathbf{j} + 8\mathbf{k}}{\sqrt{(-3)^2 + (4)^2 + (8)^2}} \right] \\ &= -0.318F_C\mathbf{i} + 0.424F_C\mathbf{j} + 0.848F_C\mathbf{k}\end{aligned}$$

$$\mathbf{F}_D = F_D\mathbf{i}$$

$$\mathbf{W} = \{-40\mathbf{k}\} \text{ lb}$$

Equilibrium requires

$$\begin{aligned}\Sigma \mathbf{F} &= \mathbf{0}; & \mathbf{F}_B + \mathbf{F}_C + \mathbf{F}_D + \mathbf{W} &= \mathbf{0} \\ & & -0.318F_B\mathbf{i} - 0.424F_B\mathbf{j} + 0.848F_B\mathbf{k} \\ & & -0.318F_C\mathbf{i} + 0.424F_C\mathbf{j} + 0.848F_C\mathbf{k} + F_D\mathbf{i} - 40\mathbf{k} &= \mathbf{0}\end{aligned}$$

Equating the respective  $\mathbf{i}, \mathbf{j}, \mathbf{k}$  components to zero yields

$$\Sigma F_x = 0; \quad -0.318F_B - 0.318F_C + F_D = 0 \quad (1)$$

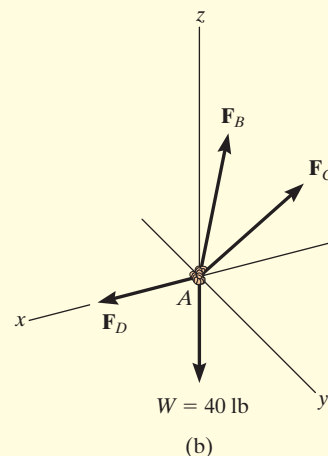
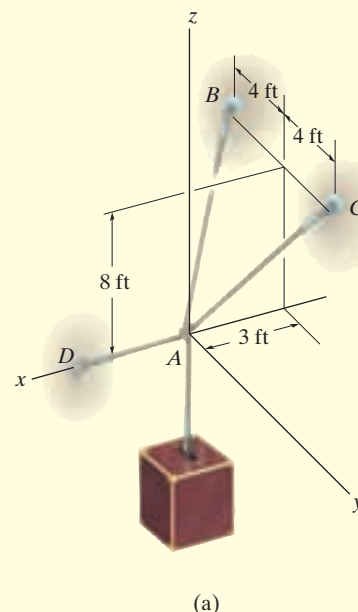
$$\Sigma F_y = 0; \quad -0.424F_B + 0.424F_C = 0 \quad (2)$$

$$\Sigma F_z = 0; \quad 0.848F_B + 0.848F_C - 40 = 0 \quad (3)$$

Equation (2) states that  $F_B = F_C$ . Thus, solving Eq. (3) for  $F_B$  and  $F_C$  and substituting the result into Eq. (1) to obtain  $F_D$ , we have

$$F_B = F_C = 23.6 \text{ lb} \quad \text{Ans.}$$

$$F_D = 15.0 \text{ lb} \quad \text{Ans.}$$



**Fig. 3–12**



## EXAMPLE 3.8

Determine the tension in each cord used to support the 100-kg crate shown in Fig. 3–13a.

## SOLUTION

**Free-Body Diagram.** The force in each of the cords can be determined by investigating the equilibrium of point A. The free-body diagram is shown in Fig. 3–13b. The weight of the crate is  $W = 100(9.81) = 981 \text{ N}$ .

**Equations of Equilibrium.** Each force on the free-body diagram is first expressed in Cartesian vector form. Using Eq. 2–9 for  $\mathbf{F}_C$  and noting point  $D(-1 \text{ m}, 2 \text{ m}, 2 \text{ m})$  for  $\mathbf{F}_D$ , we have

$$\mathbf{F}_B = F_B \mathbf{i}$$

$$\begin{aligned} \mathbf{F}_C &= F_C \cos 120^\circ \mathbf{i} + F_C \cos 135^\circ \mathbf{j} + F_C \cos 60^\circ \mathbf{k} \\ &= -0.5F_C \mathbf{i} - 0.707F_C \mathbf{j} + 0.5F_C \mathbf{k} \end{aligned}$$

$$\begin{aligned} \mathbf{F}_D &= F_D \left[ \frac{-1\mathbf{i} + 2\mathbf{j} + 2\mathbf{k}}{\sqrt{(-1)^2 + (2)^2 + (2)^2}} \right] \\ &= -0.333F_D \mathbf{i} + 0.667F_D \mathbf{j} + 0.667F_D \mathbf{k} \end{aligned}$$

$$\mathbf{W} = \{-981\mathbf{k}\} \text{ N}$$

Equilibrium requires

$$\Sigma \mathbf{F} = \mathbf{0};$$

$$\mathbf{F}_B + \mathbf{F}_C + \mathbf{F}_D + \mathbf{W} = \mathbf{0}$$

$$\begin{aligned} F_B \mathbf{i} - 0.5F_C \mathbf{i} - 0.707F_C \mathbf{j} + 0.5F_C \mathbf{k} \\ - 0.333F_D \mathbf{i} + 0.667F_D \mathbf{j} + 0.667F_D \mathbf{k} - 981\mathbf{k} = \mathbf{0} \end{aligned}$$

Equating the respective  $\mathbf{i}, \mathbf{j}, \mathbf{k}$  components to zero,

$$\Sigma F_x = 0; \quad F_B - 0.5F_C - 0.333F_D = 0 \quad (1)$$

$$\Sigma F_y = 0; \quad -0.707F_C + 0.667F_D = 0 \quad (2)$$

$$\Sigma F_z = 0; \quad 0.5F_C + 0.667F_D - 981 = 0 \quad (3)$$

Solving Eq. (2) for  $F_D$  in terms of  $F_C$  and substituting this into Eq. (3) yields  $F_C$ .  $F_D$  is then determined from Eq. (2). Finally, substituting the results into Eq. (1) gives  $F_B$ . Hence,

$$F_C = 813 \text{ N} \quad \text{Ans.}$$

$$F_D = 862 \text{ N} \quad \text{Ans.}$$

$$F_B = 694 \text{ N} \quad \text{Ans.}$$

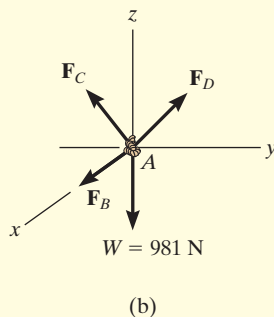
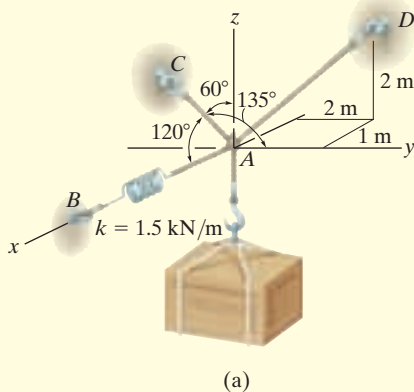
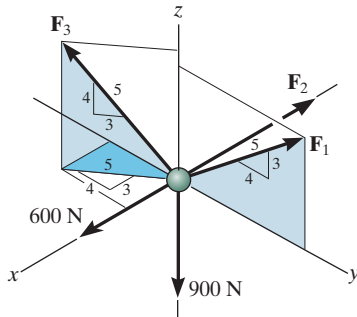


Fig. 3–13

## FUNDAMENTAL PROBLEMS

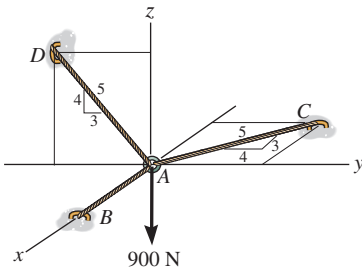
*All problem solutions must include an FBD.*

**F3-7.** Determine the magnitude of forces  $F_1$ ,  $F_2$ ,  $F_3$ , so that the particle is held in equilibrium.



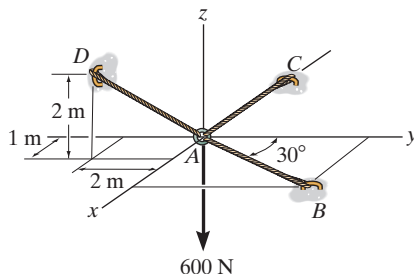
**Prob. F3-7**

**F3-8.** Determine the tension developed in cables  $AB$ ,  $AC$ , and  $AD$ .



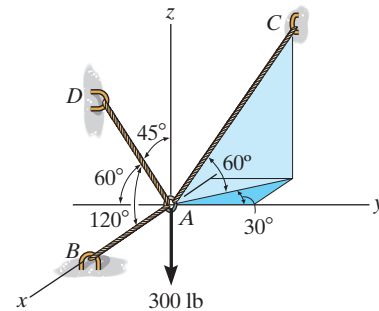
**Prob. F3-8**

**F3-9.** Determine the tension developed in cables  $AB$ ,  $AC$ , and  $AD$ .



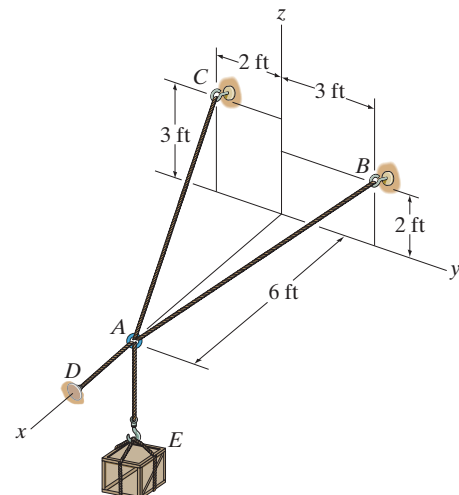
**Prob. F3-9**

**F3-10.** Determine the tension developed in cables  $AB$ ,  $AC$ , and  $AD$ .



**Prob. F3-10**

**F3-11.** The 150-lb crate is supported by cables  $AB$ ,  $AC$ , and  $AD$ . Determine the tension in these wires.

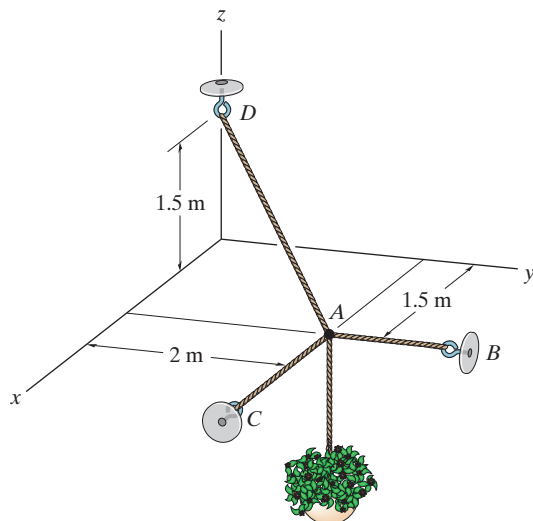


**Prob. F3-11**

## PROBLEMS

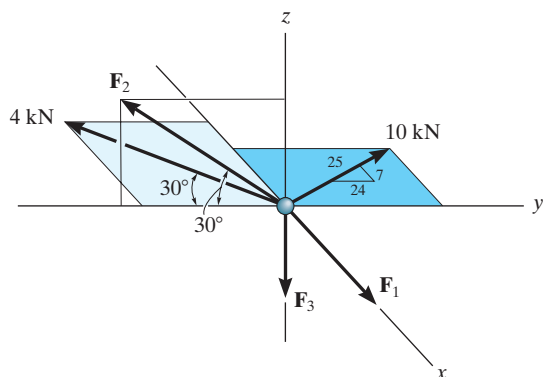
*All problem solutions must include an FBD.*

**3-43.** The three cables are used to support the 40-kg flowerpot. Determine the force developed in each cable for equilibrium.



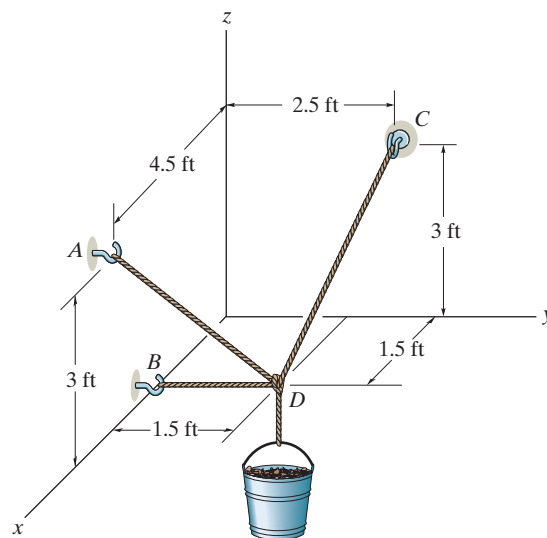
**Prob. 3-43**

**\*3-44.** Determine the magnitudes of  $F_1$ ,  $F_2$ , and  $F_3$  for equilibrium of the particle.



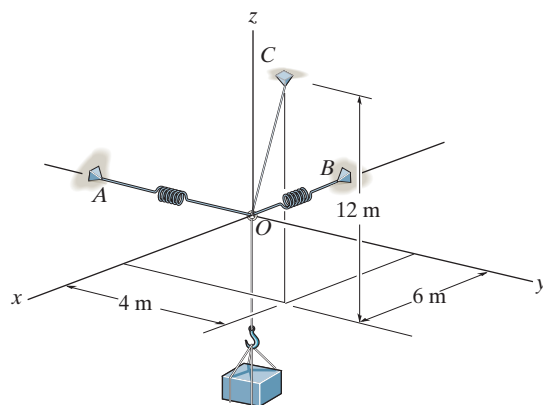
**Prob. 3-44**

**3-45.** If the bucket and its contents have a total weight of 20 lb, determine the force in the supporting cables  $DA$ ,  $DB$ , and  $DC$ .



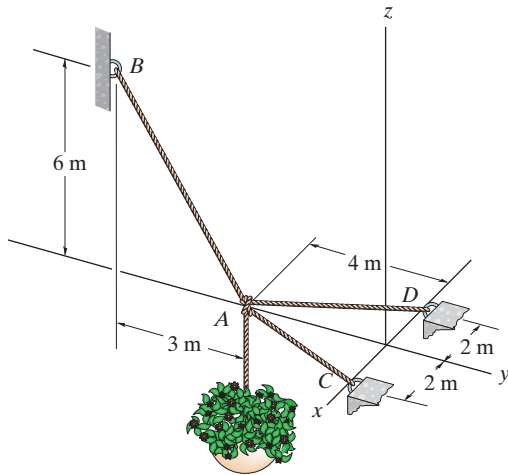
**Prob. 3-45**

**3-46.** Determine the stretch in each of the two springs required to hold the 20-kg crate in the equilibrium position shown. Each spring has an unstretched length of 2 m and a stiffness of  $k = 300$  N/m.



**Prob. 3-46**

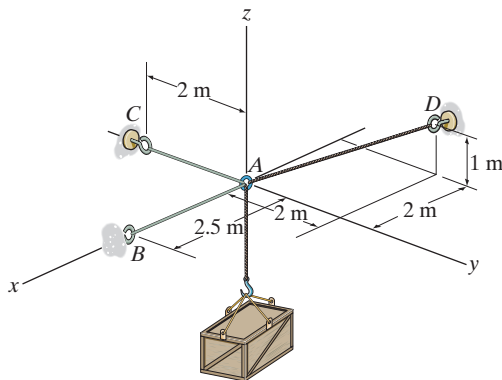
**3-47.** Determine the force in each cable needed to support the 20-kg flowerpot.



**Prob. 3-47**

**\*3-48.** Determine the tension in the cables in order to support the 100-kg crate in the equilibrium position shown.

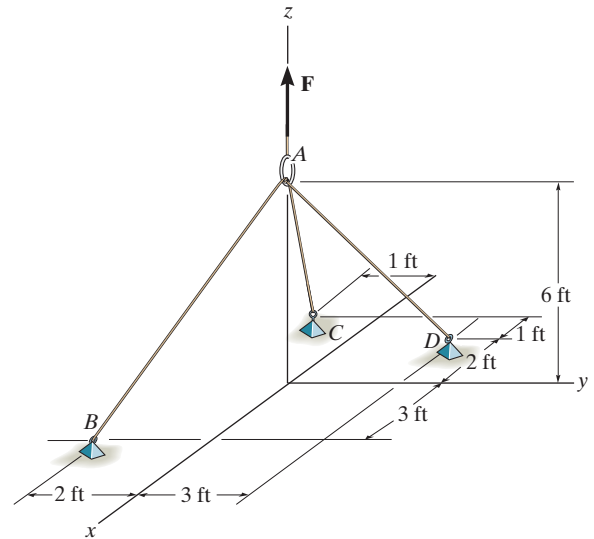
**3-49.** Determine the maximum mass of the crate so that the tension developed in any cable does not exceed 3 kN.



**Probs. 3-48/49**

**3-50.** Determine the force in each cable if  $F = 500$  lb.

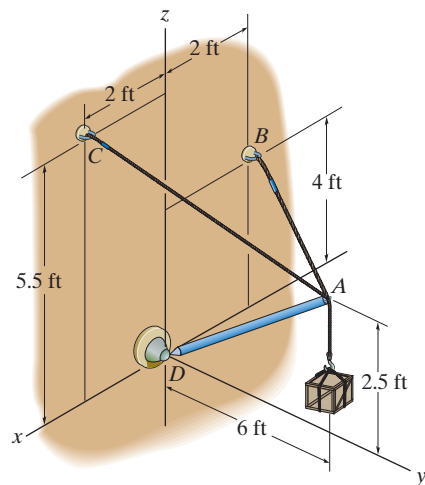
**3-51.** Determine the greatest force  $\mathbf{F}$  that can be applied to the ring if each cable can support a maximum force of 800 lb.



**Probs. 3-50/51**

**\*3-52.** Determine the tension developed in cables  $AB$  and  $AC$  and the force developed along strut  $AD$  for equilibrium of the 400-lb crate.

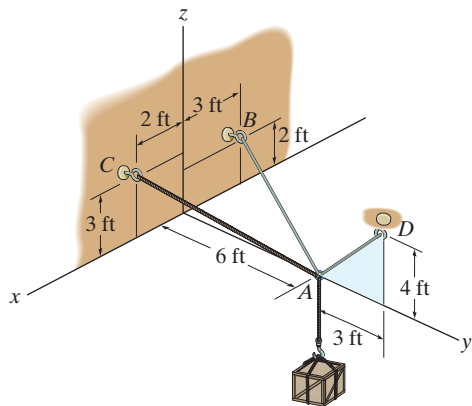
**3-53.** If the tension developed in each cable cannot exceed 300 lb, determine the largest weight of the crate that can be supported. Also, what is the force developed along strut  $AD$ ?



**Probs. 3-52/53**

**3-54.** Determine the tension developed in each cable for equilibrium of the 300-lb crate.

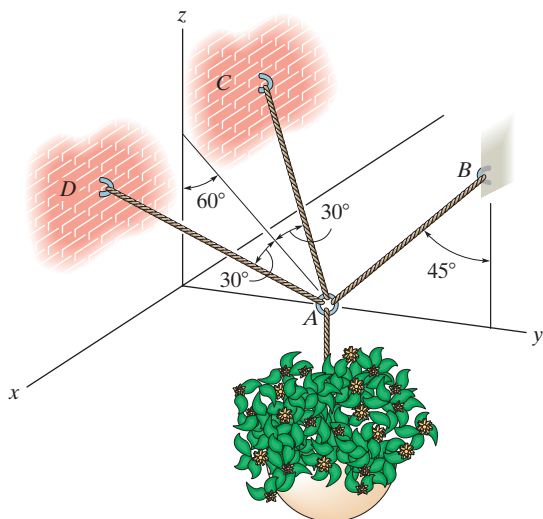
**3-55.** Determine the maximum weight of the crate that can be suspended from cables  $AB$ ,  $AC$ , and  $AD$  so that the tension developed in any one of the cables does not exceed 250 lb.



**Probs. 3-54/55**

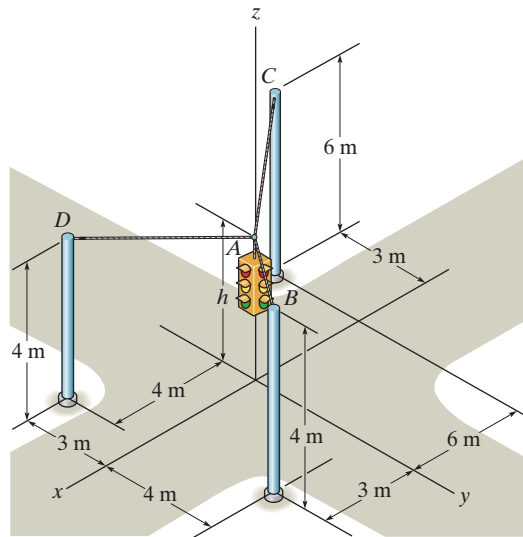
**\*3-56.** The 25-kg flowerpot is supported at  $A$  by the three cords. Determine the force acting in each cord for equilibrium.

**3-57.** If each cord can sustain a maximum tension of 50 N before it fails, determine the greatest weight of the flowerpot the cords can support.



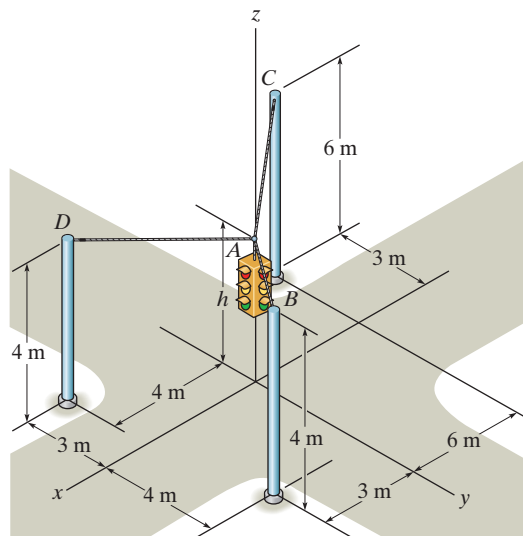
**Probs. 3-56/57**

**3-58.** Determine the tension developed in the three cables required to support the traffic light, which has a mass of 15 kg. Take  $h = 4$  m.



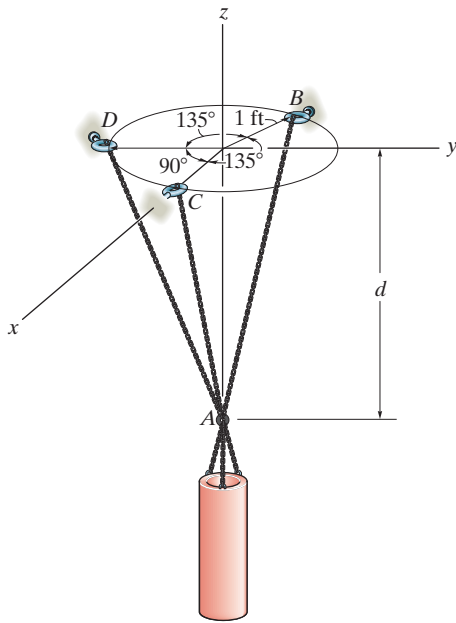
**Prob. 3-58**

**3-59.** Determine the tension developed in the three cables required to support the traffic light, which has a mass of 20 kg. Take  $h = 3.5$  m.



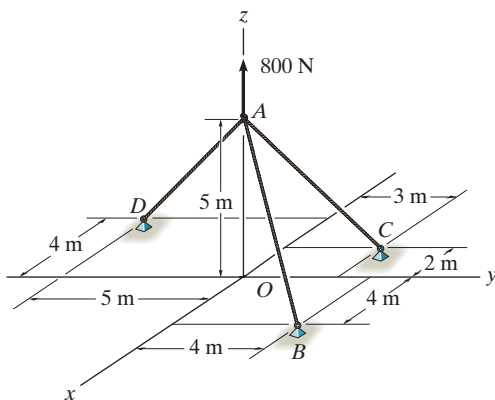
**Prob. 3-59**

**\*3–60.** The 800-lb cylinder is supported by three chains as shown. Determine the force in each chain for equilibrium. Take  $d = 1$  ft.



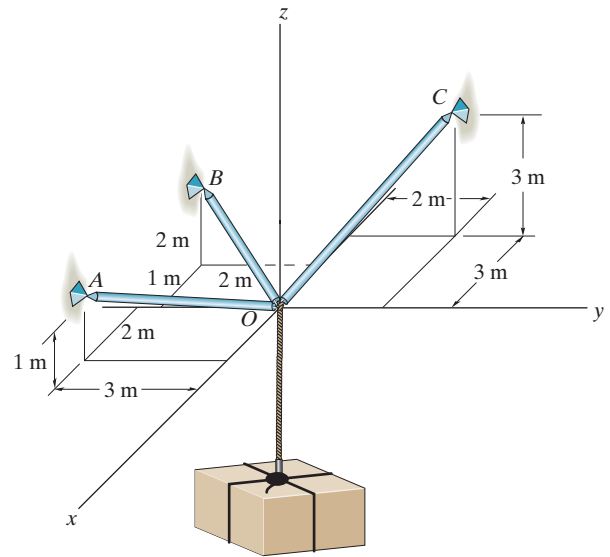
**Prob. 3–60**

**3–61.** Determine the tension in each cable for equilibrium.



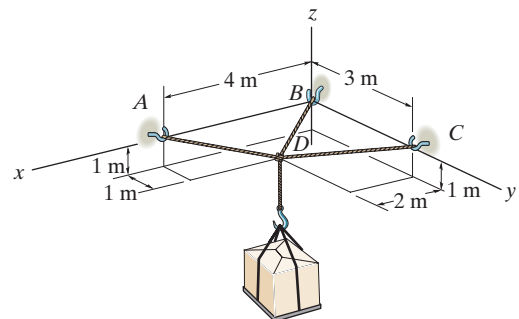
**Prob. 3–61**

**3–62.** If the maximum force in each rod can not exceed 1500 N, determine the greatest mass of the crate that can be supported.



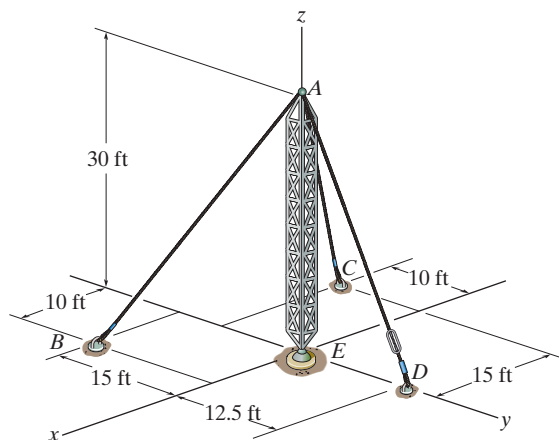
**Prob. 3–62**

**3–63.** The crate has a mass of 130 kg. Determine the tension developed in each cable for equilibrium.



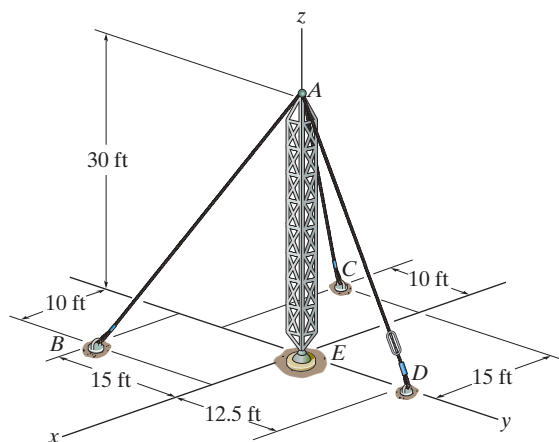
**Prob. 3–63**

**\*3–64.** If cable  $AD$  is tightened by a turnbuckle and develops a tension of 1300 lb, determine the tension developed in cables  $AB$  and  $AC$  and the force developed along the antenna tower  $AE$  at point  $A$ .



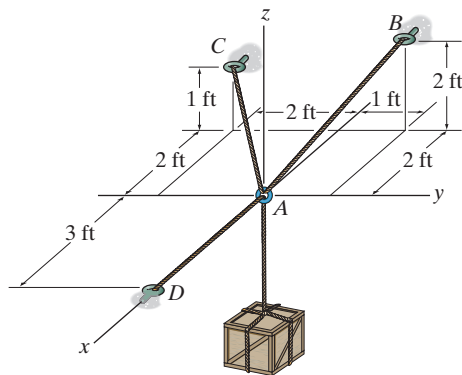
**Prob. 3–64**

**3–65.** If the tension developed in either cable  $AB$  or  $AC$  can not exceed 1000 lb, determine the maximum tension that can be developed in cable  $AD$  when it is tightened by the turnbuckle. Also, what is the force developed along the antenna tower at point  $A$ ?



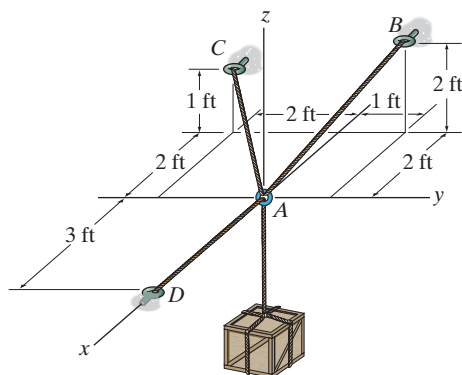
**Prob. 3–65**

**3–66.** Determine the tension developed in cables  $AB$ ,  $AC$ , and  $AD$  required for equilibrium of the 300-lb crate.



**Prob. 3–66**

**3–67.** Determine the maximum weight of the crate so that the tension developed in any cable does not exceed 450 lb.



**Prob. 3–67**

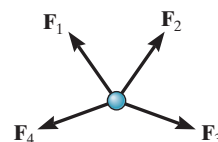
## CHAPTER REVIEW

### Particle Equilibrium

When a particle is at rest or moves with constant velocity, it is in equilibrium. This requires that all the forces acting on the particle form a zero resultant force.

In order to account for all the forces that act on a particle, it is necessary to draw its free-body diagram. This diagram is an outlined shape of the particle that shows all the forces listed with their known or unknown magnitudes and directions.

$$F_R = \Sigma F = 0$$



### Two Dimensions

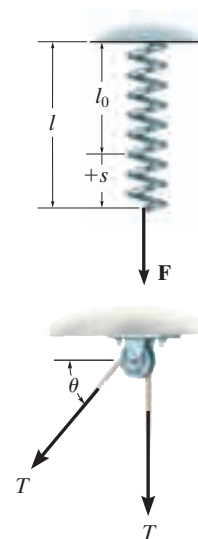
If the problem involves a linearly elastic spring, then the stretch or compression  $s$  of the spring can be related to the force applied to it.

The tensile force developed in a *continuous cable* that passes over a frictionless pulley must have a *constant* magnitude throughout the cable to keep the cable in equilibrium.

The two scalar equations of force equilibrium can be applied with reference to an established  $x, y$  coordinate system.

$$F = ks$$

$$\begin{aligned}\Sigma F_x &= 0 \\ \Sigma F_y &= 0\end{aligned}$$

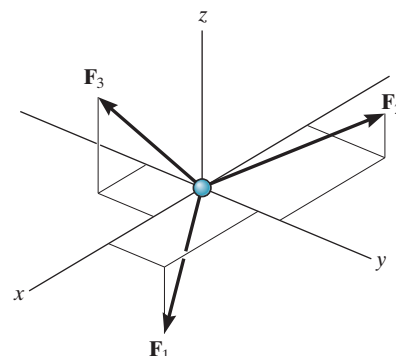


### Three Dimensions

If the three-dimensional geometry is difficult to visualize, then the equilibrium equation should be applied using a Cartesian vector analysis. This requires first expressing each force on the free-body diagram as a Cartesian vector. When the forces are summed and set equal to zero, then the  $\mathbf{i}$ ,  $\mathbf{j}$ , and  $\mathbf{k}$  components are also zero.

$$\Sigma \mathbf{F} = \mathbf{0}$$

$$\begin{aligned}\Sigma F_x &= 0 \\ \Sigma F_y &= 0 \\ \Sigma F_z &= 0\end{aligned}$$

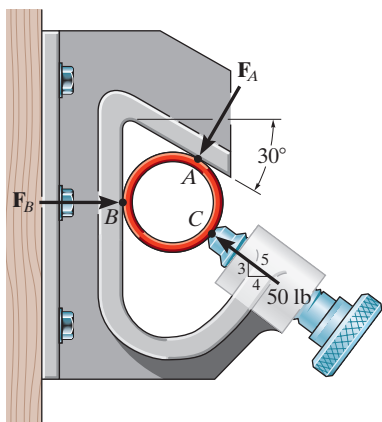




## REVIEW PROBLEMS

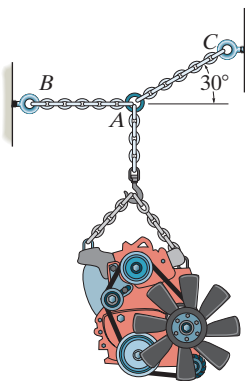
*All problem solutions must include an FBD.*

**R3-1.** The pipe is held in place by the vise. If the bolt exerts a force of 50 lb on the pipe in the direction shown, determine the forces  $F_A$  and  $F_B$  that the smooth contacts at  $A$  and  $B$  exert on the pipe.



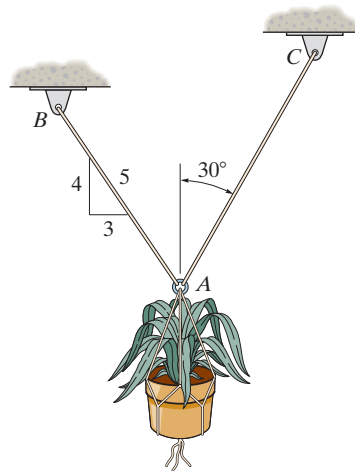
**Prob. R3-1**

**R3-2.** Determine the maximum weight of the engine that can be supported without exceeding a tension of 450 lb in chain  $AB$  and 480 lb in chain  $AC$ .



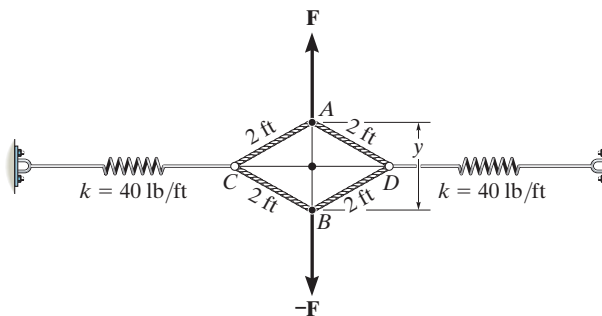
**Prob. R3-2**

**R3-3.** Determine the maximum weight of the flowerpot that can be supported without exceeding a cable tension of 50 lb in either cable  $AB$  or  $AC$ .



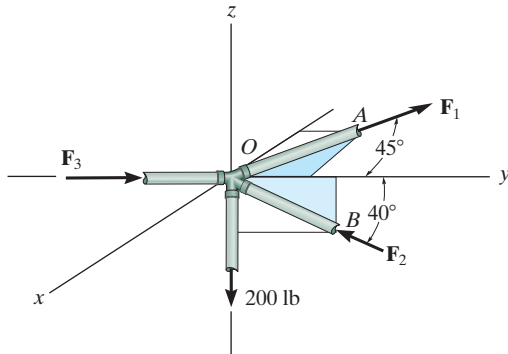
**Prob. R3-3**

**R3-4.** When  $y$  is zero, the springs sustain a force of 60 lb. Determine the magnitude of the applied vertical forces  $\mathbf{F}$  and  $-\mathbf{F}$  required to pull point  $A$  away from point  $B$  a distance of  $y = 2$  ft. The ends of cords  $CAD$  and  $CBD$  are attached to rings at  $C$  and  $D$ .



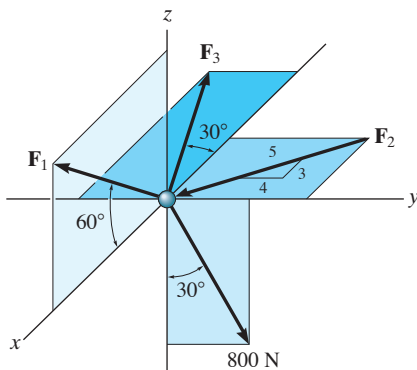
**Prob. R3-4**

**R3-5.** The joint of a space frame is subjected to four member forces. Member  $OA$  lies in the  $x$ - $y$  plane and member  $OB$  lies in the  $y$ - $z$  plane. Determine the force acting in each of the members required for equilibrium of the joint.



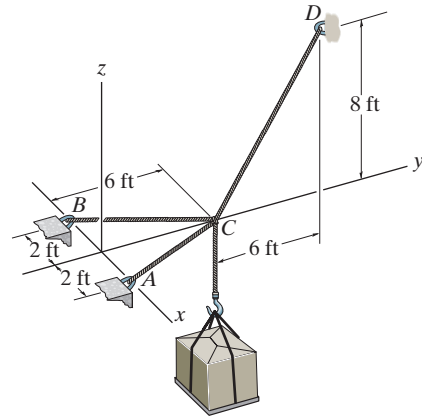
**Prob. R3-5**

**R3-6.** Determine the magnitudes of  $\mathbf{F}_1$ ,  $\mathbf{F}_2$ , and  $\mathbf{F}_3$  for equilibrium of the particle.



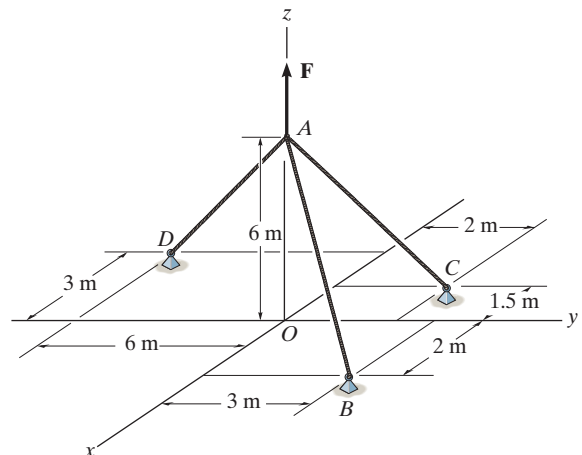
**Prob. R3-6**

**R3-7.** Determine the force in each cable needed to support the 500-lb load.



**Prob. R3-7**

**R3-8.** If cable  $AB$  is subjected to a tension of 700 N, determine the tension in cables  $AC$  and  $AD$  and the magnitude of the vertical force  $\mathbf{F}$ .



**Prob. R3-8**

# Chapter 4



(© Rolf Adlercreutz/Alamy)

The force applied to this wrench will produce rotation or a tendency for rotation. This effect is called a moment, and in this chapter we will study how to determine the moment of a system of forces and calculate their resultants.

# Force System Resultants

## CHAPTER OBJECTIVES

- To discuss the concept of the moment of a force and show how to calculate it in two and three dimensions.
- To provide a method for finding the moment of a force about a specified axis.
- To define the moment of a couple.
- To show how to find the resultant effect of a nonconcurrent force system.
- To indicate how to reduce a simple distributed loading to a resultant force acting at a specified location.

## 4.1 Moment of a Force—Scalar Formulation

When a force is applied to a body it will produce a tendency for the body to rotate about a point that is not on the line of action of the force. This tendency to rotate is sometimes called a **torque**, but most often it is called the moment of a force or simply the **moment**. For example, consider a wrench used to unscrew the bolt in Fig. 4–1a. If a force is applied to the handle of the wrench it will tend to turn the bolt about point  $O$  (or the  $z$  axis). The magnitude of the moment is directly proportional to the magnitude of  $\mathbf{F}$  and the perpendicular distance or **moment arm**  $d$ . The larger the force or the longer the moment arm, the greater the moment or turning effect. Note that if the force  $\mathbf{F}$  is applied at an angle  $\theta \neq 90^\circ$ , Fig. 4–1b, then it will be more difficult to turn the bolt since the moment arm  $d' = d \sin \theta$  will be smaller than  $d$ . If  $\mathbf{F}$  is applied along the wrench, Fig. 4–1c, its moment arm will be zero since the line of action of  $\mathbf{F}$  will intersect point  $O$  (the  $z$  axis). As a result, the moment of  $\mathbf{F}$  about  $O$  is also zero and no turning can occur.

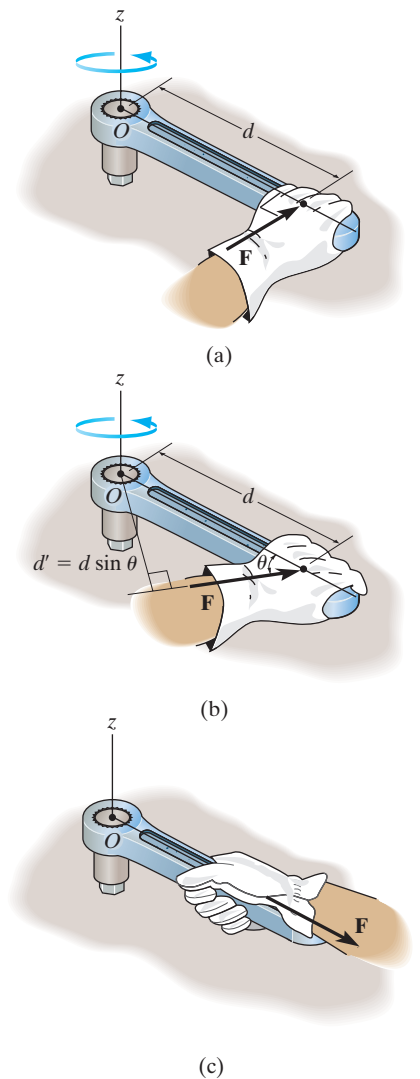


Fig. 4–1

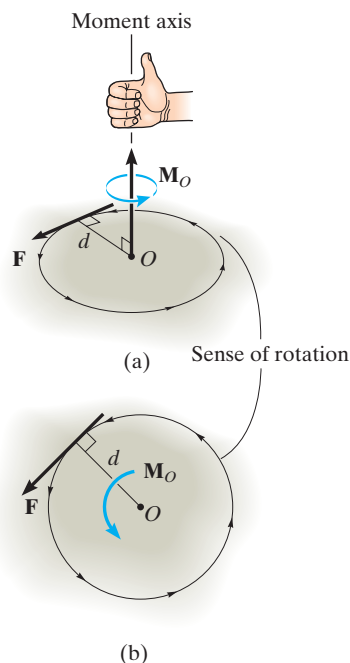


Fig. 4-2

We can generalize the above discussion and consider the force  $\mathbf{F}$  and point  $O$  which lie in the shaded plane as shown in Fig. 4-2a. The moment  $\mathbf{M}_O$  about point  $O$ , or about an axis passing through  $O$  and perpendicular to the plane, is a *vector quantity* since it has a specified magnitude and direction.

**Magnitude.** The magnitude of  $\mathbf{M}_O$  is

$$M_O = Fd \quad (4-1)$$

where  $d$  is the **moment arm** or *perpendicular distance* from the axis at point  $O$  to the line of action of the force. Units of moment magnitude consist of force times distance, e.g.,  $\text{N} \cdot \text{m}$  or  $\text{lb} \cdot \text{ft}$ .

**Direction.** The direction of  $\mathbf{M}_O$  is defined by its **moment axis**, which is perpendicular to the plane that contains the force  $\mathbf{F}$  and its moment arm  $d$ . The right-hand rule is used to establish the sense of direction of  $\mathbf{M}_O$ . According to this rule, the natural curl of the fingers of the right hand, as they are drawn towards the palm, represent the rotation, or if no movement is possible, there is a tendency for rotation caused by the moment. As this action is performed, the thumb of the right hand will give the directional sense of  $\mathbf{M}_O$ , Fig. 4-2a. Notice that the moment vector is represented three-dimensionally by a curl around an arrow. In two dimensions this vector is represented only by the curl as in Fig. 4-2b. Since in this case the moment will tend to cause a counterclockwise rotation, the moment vector is actually directed out of the page.

**Resultant Moment.** For two-dimensional problems, where all the forces lie within the  $x$ - $y$  plane, Fig. 4-3, the resultant moment  $(\mathbf{M}_R)_O$  about point  $O$  (the  $z$  axis) can be determined by *finding the algebraic sum* of the moments caused by all the forces in the system. As a convention, we will generally consider *positive moments* as *counterclockwise* since they are directed along the positive  $z$  axis (out of the page). *Clockwise moments* will be *negative*. Doing this, the directional sense of each moment can be represented by a *plus* or *minus* sign. Using this sign convention, with a symbolic curl to define the positive direction, the resultant moment in Fig. 4-3 is therefore

$$\curvearrowleft + (M_R)_O = \Sigma Fd; \quad (M_R)_O = F_1d_1 - F_2d_2 + F_3d_3$$

If the numerical result of this sum is a positive scalar,  $(\mathbf{M}_R)_O$  will be a counterclockwise moment (out of the page); and if the result is negative,  $(\mathbf{M}_R)_O$  will be a clockwise moment (into the page).

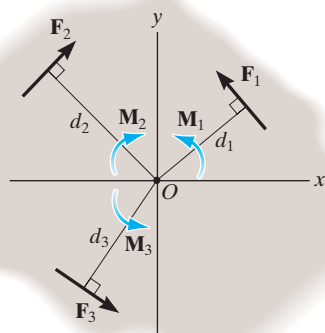


Fig. 4-3

**EXAMPLE 4.1**

For each case illustrated in Fig. 4–4, determine the moment of the force about point  $O$ .

**SOLUTION (SCALAR ANALYSIS)**

The line of action of each force is extended as a dashed line in order to establish the moment arm  $d$ . Also illustrated is the tendency of rotation of the member as caused by the force. Furthermore, the orbit of the force about  $O$  is shown as a colored curl. Thus,

Fig. 4–4a  $M_O = (100 \text{ N})(2 \text{ m}) = 200 \text{ N} \cdot \text{m} \curvearrowright$

*Ans.*

Fig. 4–4b  $M_O = (50 \text{ N})(0.75 \text{ m}) = 37.5 \text{ N} \cdot \text{m} \curvearrowright$

*Ans.*

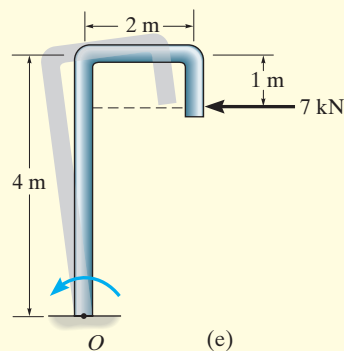
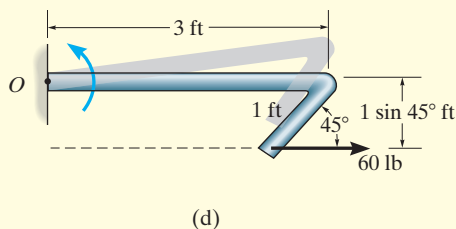
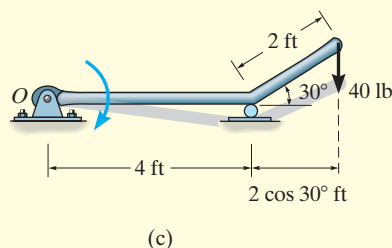
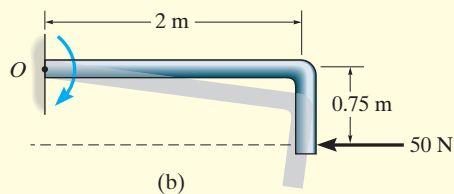
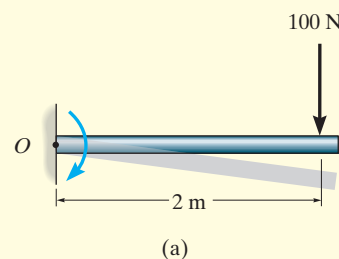
Fig. 4–4c  $M_O = (40 \text{ lb})(4 \text{ ft} + 2 \cos 30^\circ \text{ ft}) = 229 \text{ lb} \cdot \text{ft} \curvearrowright$

*Ans.*

Fig. 4–4d  $M_O = (60 \text{ lb})(1 \sin 45^\circ \text{ ft}) = 42.4 \text{ lb} \cdot \text{ft} \curvearrowright$

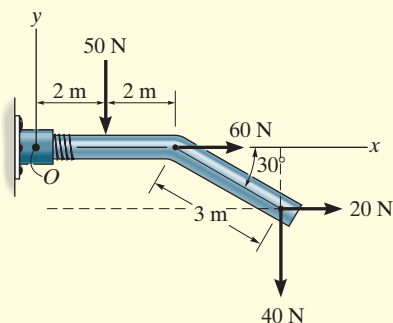
*Ans.*

Fig. 4–4e  $M_O = (7 \text{ kN})(4 \text{ m} - 1 \text{ m}) = 21.0 \text{ kN} \cdot \text{m} \curvearrowright$

*Ans.***Fig. 4–4**

**EXAMPLE 4.2**

Determine the resultant moment of the four forces acting on the rod shown in Fig. 4-5 about point  $O$ .



**Fig. 4-5**

**SOLUTION**

Assuming that positive moments act in the  $+\mathbf{k}$  direction, i.e., counterclockwise, we have

$$\zeta + (M_R)_O = \Sigma Fd;$$

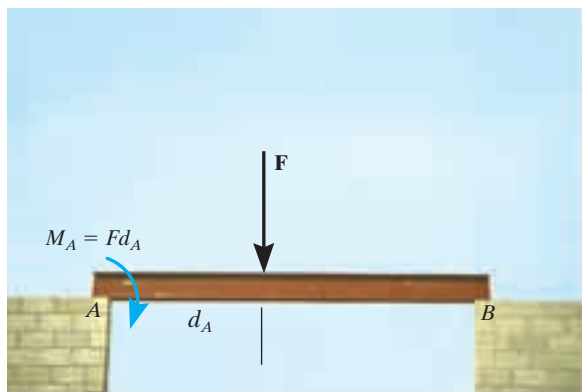
$$(M_R)_O = -50 \text{ N}(2 \text{ m}) + 60 \text{ N}(0) + 20 \text{ N}(3 \sin 30^\circ \text{ m})$$

$$-40 \text{ N}(4 \text{ m} + 3 \cos 30^\circ \text{ m})$$

$$(M_R)_O = -334 \text{ N} \cdot \text{m} = 334 \text{ N} \cdot \text{m} \curvearrowright$$

*Ans.*

For this calculation, note how the moment-arm distances for the 20-N and 40-N forces are established from the extended (dashed) lines of action of each of these forces.



As illustrated by the example problems, the moment of a force does not always cause a rotation. For example, the force  $\mathbf{F}$  tends to rotate the beam clockwise about its support at  $A$  with a moment  $M_A = Fd_A$ . The actual rotation would occur if the support at  $B$  were removed. (© Russell C. Hibbeler)



The ability to remove the nail will require the moment of  $\mathbf{F}_H$  about point  $O$  to be larger than the moment of the force  $\mathbf{F}_N$  about  $O$  that is needed to pull the nail out. (© Russell C. Hibbeler)

## 4.2 Cross Product

The moment of a force will be formulated using Cartesian vectors in the next section. Before doing this, however, it is first necessary to expand our knowledge of vector algebra and introduce the cross-product method of vector multiplication, first used by Willard Gibbs in lectures given in the late 19th century.

The **cross product** of two vectors **A** and **B** yields the vector **C**, which is written

$$\mathbf{C} = \mathbf{A} \times \mathbf{B} \quad (4-2)$$

and is read “**C** equals **A** cross **B**.”

**Magnitude.** The *magnitude* of **C** is defined as the product of the magnitudes of **A** and **B** and the sine of the angle  $\theta$  between their tails ( $0^\circ \leq \theta \leq 180^\circ$ ). Thus,  $C = AB \sin \theta$ .

**Direction.** Vector **C** has a *direction* that is perpendicular to the plane containing **A** and **B** such that **C** is specified by the right-hand rule; i.e., curling the fingers of the right hand from vector **A** (cross) to vector **B**, the thumb points in the direction of **C**, as shown in Fig. 4-6.

Knowing both the magnitude and direction of **C**, we can write

$$\mathbf{C} = \mathbf{A} \times \mathbf{B} = (AB \sin \theta) \mathbf{u}_C \quad (4-3)$$

where the scalar  $AB \sin \theta$  defines the *magnitude* of **C** and the unit vector  $\mathbf{u}_C$  defines the *direction* of **C**. The terms of Eq. 4-3 are illustrated graphically in Fig. 4-6.

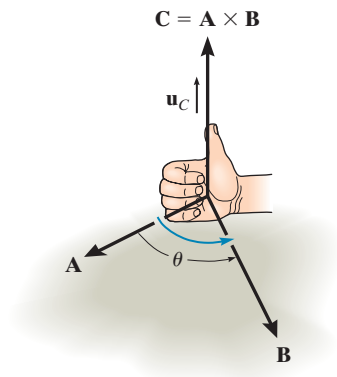


Fig. 4-6



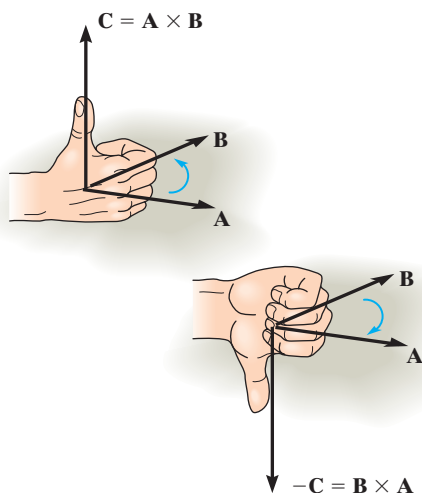


Fig. 4-7

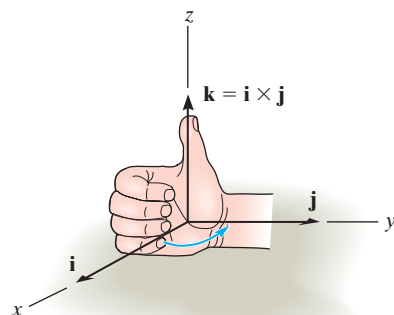


Fig. 4-8

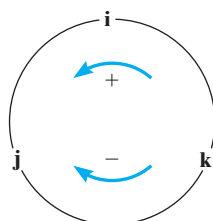


Fig. 4-9

## Laws of Operation.

- The commutative law is *not* valid; i.e.,  $\mathbf{A} \times \mathbf{B} \neq \mathbf{B} \times \mathbf{A}$ . Rather,

$$\mathbf{A} \times \mathbf{B} = -\mathbf{B} \times \mathbf{A}$$

This is shown in Fig. 4-7 by using the right-hand rule. The cross product  $\mathbf{B} \times \mathbf{A}$  yields a vector that has the same magnitude but acts in the opposite direction to  $\mathbf{C}$ ; i.e.,  $\mathbf{B} \times \mathbf{A} = -\mathbf{C}$ .

- If the cross product is multiplied by a scalar  $a$ , it obeys the associative law;

$$a(\mathbf{A} \times \mathbf{B}) = (a\mathbf{A}) \times \mathbf{B} = \mathbf{A} \times (a\mathbf{B}) = (\mathbf{A} \times \mathbf{B})a$$

This property is easily shown since the magnitude of the resultant vector ( $|a|AB \sin \theta$ ) and its direction are the same in each case.

- The vector cross product also obeys the distributive law of addition,

$$\mathbf{A} \times (\mathbf{B} + \mathbf{D}) = (\mathbf{A} \times \mathbf{B}) + (\mathbf{A} \times \mathbf{D})$$

- The proof of this identity is left as an exercise (see Prob. 4-1). It is important to note that *proper order* of the cross products must be maintained, since they are not commutative.

**Cartesian Vector Formulation.** Equation 4-3 may be used to find the cross product of any pair of Cartesian unit vectors. For example, to find  $\mathbf{i} \times \mathbf{j}$ , the magnitude of the resultant vector is  $(i)(j)(\sin 90^\circ) = (1)(1)(1) = 1$ , and its direction is determined using the right-hand rule. As shown in Fig. 4-8, the resultant vector points in the  $+\mathbf{k}$  direction. Thus,  $\mathbf{i} \times \mathbf{j} = (1)\mathbf{k}$ . In a similar manner,

$$\mathbf{i} \times \mathbf{j} = \mathbf{k} \quad \mathbf{i} \times \mathbf{k} = -\mathbf{j} \quad \mathbf{i} \times \mathbf{i} = \mathbf{0}$$

$$\mathbf{j} \times \mathbf{k} = \mathbf{i} \quad \mathbf{j} \times \mathbf{i} = -\mathbf{k} \quad \mathbf{j} \times \mathbf{j} = \mathbf{0}$$

$$\mathbf{k} \times \mathbf{i} = \mathbf{j} \quad \mathbf{k} \times \mathbf{j} = -\mathbf{i} \quad \mathbf{k} \times \mathbf{k} = \mathbf{0}$$

These results should *not* be memorized; rather, it should be clearly understood how each is obtained by using the right-hand rule and the definition of the cross product. A simple scheme shown in Fig. 4-9 is helpful for obtaining the same results when the need arises. If the circle is constructed as shown, then “crossing” two unit vectors in a *counterclockwise* fashion around the circle yields the *positive* third unit vector; e.g.,  $\mathbf{k} \times \mathbf{i} = \mathbf{j}$ . “Crossing” *clockwise*, a *negative* unit vector is obtained; e.g.,  $\mathbf{i} \times \mathbf{k} = -\mathbf{j}$ .

Let us now consider the cross product of two general vectors  $\mathbf{A}$  and  $\mathbf{B}$  which are expressed in Cartesian vector form. We have

$$\begin{aligned}\mathbf{A} \times \mathbf{B} &= (A_x \mathbf{i} + A_y \mathbf{j} + A_z \mathbf{k}) \times (B_x \mathbf{i} + B_y \mathbf{j} + B_z \mathbf{k}) \\ &= A_x B_x (\mathbf{i} \times \mathbf{i}) + A_x B_y (\mathbf{i} \times \mathbf{j}) + A_x B_z (\mathbf{i} \times \mathbf{k}) \\ &\quad + A_y B_x (\mathbf{j} \times \mathbf{i}) + A_y B_y (\mathbf{j} \times \mathbf{j}) + A_y B_z (\mathbf{j} \times \mathbf{k}) \\ &\quad + A_z B_x (\mathbf{k} \times \mathbf{i}) + A_z B_y (\mathbf{k} \times \mathbf{j}) + A_z B_z (\mathbf{k} \times \mathbf{k})\end{aligned}$$

Carrying out the cross-product operations and combining terms yields

$$\mathbf{A} \times \mathbf{B} = (A_y B_z - A_z B_y) \mathbf{i} - (A_x B_z - A_z B_x) \mathbf{j} + (A_x B_y - A_y B_x) \mathbf{k} \quad (4-4)$$

This equation may also be written in a more compact determinant form as

$$\mathbf{A} \times \mathbf{B} = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ A_x & A_y & A_z \\ B_x & B_y & B_z \end{vmatrix} \quad (4-5)$$

Thus, to find the cross product of any two Cartesian vectors  $\mathbf{A}$  and  $\mathbf{B}$ , it is necessary to expand a determinant whose first row of elements consists of the unit vectors  $\mathbf{i}$ ,  $\mathbf{j}$ , and  $\mathbf{k}$  and whose second and third rows represent the  $x$ ,  $y$ ,  $z$  components of the two vectors  $\mathbf{A}$  and  $\mathbf{B}$ , respectively.\*

\*A determinant having three rows and three columns can be expanded using three minors, each of which is multiplied by one of the three terms in the first row. There are four elements in each minor, for example,

By *definition*, this determinant notation represents the terms  $(A_{11}A_{22} - A_{12}A_{21})$ , which is simply the product of the two elements intersected by the arrow slanting downward to the right ( $A_{11}A_{22}$ ) *minus* the product of the two elements intersected by the arrow slanting downward to the left ( $A_{12}A_{21}$ ). For a  $3 \times 3$  determinant, such as Eq. 4-5, the three minors can be generated in accordance with the following scheme:

For element  $\mathbf{i}$ :  $\begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ A_x & A_y & A_z \\ B_x & B_y & B_z \end{vmatrix} = \mathbf{i}(A_y B_z - A_z B_y)$

For element  $\mathbf{j}$ :  $\begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ A_x & A_y & A_z \\ B_x & B_y & B_z \end{vmatrix} = -\mathbf{j}(A_x B_z - A_z B_x)$

For element  $\mathbf{k}$ :  $\begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ A_x & A_y & A_z \\ B_x & B_y & B_z \end{vmatrix} = \mathbf{k}(A_x B_y - A_y B_x)$

Remember the negative sign

Adding the results and noting that the  $\mathbf{j}$  element *must include the minus sign* yields the expanded form of  $\mathbf{A} \times \mathbf{B}$  given by Eq. 4-4.

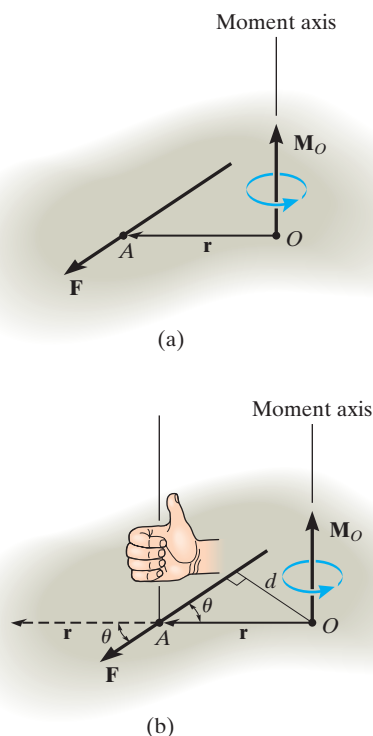


Fig. 4-10

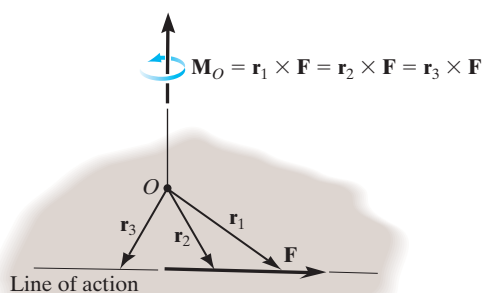


Fig. 4-11

## 4.3 Moment of a Force—Vector Formulation

The moment of a force  $\mathbf{F}$  about point  $O$ , or actually about the moment axis passing through  $O$  and perpendicular to the plane containing  $O$  and  $\mathbf{F}$ , Fig. 4-10a, can be expressed using the vector cross product, namely,

$$\mathbf{M}_O = \mathbf{r} \times \mathbf{F} \quad (4-6)$$

Here  $\mathbf{r}$  represents a position vector directed *from*  $O$  to *any point* on the line of action of  $\mathbf{F}$ . We will now show that indeed the moment  $\mathbf{M}_O$ , when determined by this cross product, has the proper magnitude and direction.

**Magnitude.** The magnitude of the cross product is defined from Eq. 4-3 as  $M_O = rF \sin \theta$ , where the angle  $\theta$  is measured between the *tails* of  $\mathbf{r}$  and  $\mathbf{F}$ . To establish this angle,  $\mathbf{r}$  must be treated as a sliding vector so that  $\theta$  can be constructed properly, Fig. 4-10b. Since the moment arm  $d = r \sin \theta$ , then

$$M_O = rF \sin \theta = F(r \sin \theta) = Fd$$

which agrees with Eq. 4-1.

**Direction.** The direction and sense of  $\mathbf{M}_O$  in Eq. 4-6 are determined by the right-hand rule as it applies to the cross product. Thus, sliding  $\mathbf{r}$  to the dashed position and curling the right-hand fingers from  $\mathbf{r}$  toward  $\mathbf{F}$ , “ $\mathbf{r}$  cross  $\mathbf{F}$ ,” the thumb is directed upward or perpendicular to the plane containing  $\mathbf{r}$  and  $\mathbf{F}$  and this is in the *same direction* as  $\mathbf{M}_O$ , the moment of the force about point  $O$ , Fig. 4-10b. Note that the “curl” of the fingers, like the curl around the moment vector, indicates the sense of rotation caused by the force. Since the cross product does not obey the commutative law, the order of  $\mathbf{r} \times \mathbf{F}$  must be maintained to produce the correct sense of direction for  $\mathbf{M}_O$ .

**Principle of Transmissibility.** The cross product operation is often used in three dimensions since the perpendicular distance or moment arm from point  $O$  to the line of action of the force is not needed. In other words, we can use any position vector  $\mathbf{r}$  measured from point  $O$  to any point on the line of action of the force  $\mathbf{F}$ , Fig. 4-11. Thus,

$$\mathbf{M}_O = \mathbf{r}_1 \times \mathbf{F} = \mathbf{r}_2 \times \mathbf{F} = \mathbf{r}_3 \times \mathbf{F}$$

Since  $\mathbf{F}$  can be applied at any point along its line of action and still create this *same moment* about point  $O$ , then  $\mathbf{F}$  can be considered a *sliding vector*. This property is called the *principle of transmissibility* of a force.

**Cartesian Vector Formulation.** If we establish  $x, y, z$  coordinate axes, then the position vector  $\mathbf{r}$  and force  $\mathbf{F}$  can be expressed as Cartesian vectors, Fig. 4-12a. Applying Eq. 4-5 we have

$$\mathbf{M}_O = \mathbf{r} \times \mathbf{F} = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ r_x & r_y & r_z \\ F_x & F_y & F_z \end{vmatrix} \quad (4-7)$$

where

$r_x, r_y, r_z$  represent the  $x, y, z$  components of the position vector drawn from point  $O$  to *any point* on the line of action of the force

$F_x, F_y, F_z$  represent the  $x, y, z$  components of the force vector

If the determinant is expanded, then like Eq. 4-4 we have

$$\mathbf{M}_O = (r_y F_z - r_z F_y)\mathbf{i} - (r_x F_z - r_z F_x)\mathbf{j} + (r_x F_y - r_y F_x)\mathbf{k} \quad (4-8)$$

The physical meaning of these three moment components becomes evident by studying Fig. 4-12b. For example, the  $\mathbf{i}$  component of  $\mathbf{M}_O$  can be determined from the moments of  $\mathbf{F}_x$ ,  $\mathbf{F}_y$ , and  $\mathbf{F}_z$  about the  $x$  axis. The component  $\mathbf{F}_x$  does *not* create a moment or tendency to cause turning about the  $x$  axis since this force is *parallel* to the  $x$  axis. The line of action of  $\mathbf{F}_y$  passes through point  $B$ , and so the magnitude of the moment of  $\mathbf{F}_y$  about point  $A$  on the  $x$  axis is  $r_z F_y$ . By the right-hand rule this component acts in the *negative i* direction. Likewise,  $\mathbf{F}_z$  passes through point  $C$  and so it contributes a moment component of  $r_y F_z \mathbf{i}$  about the  $x$  axis. Thus,  $(M_O)_x = (r_y F_z - r_z F_y)$  as shown in Eq. 4-8. As an exercise, establish the  $\mathbf{j}$  and  $\mathbf{k}$  components of  $\mathbf{M}_O$  in this manner and show that indeed the expanded form of the determinant, Eq. 4-8, represents the moment of  $\mathbf{F}$  about point  $O$ . Once  $\mathbf{M}_O$  is determined, realize that it will always be *perpendicular* to the shaded plane containing vectors  $\mathbf{r}$  and  $\mathbf{F}$ , Fig. 4-12a.

**Resultant Moment of a System of Forces.** If a body is acted upon by a system of forces, Fig. 4-13, the resultant moment of the forces about point  $O$  can be determined by vector addition of the moment of each force. This resultant can be written symbolically as

$$(\mathbf{M}_R)_O = \Sigma(\mathbf{r} \times \mathbf{F}) \quad (4-9)$$

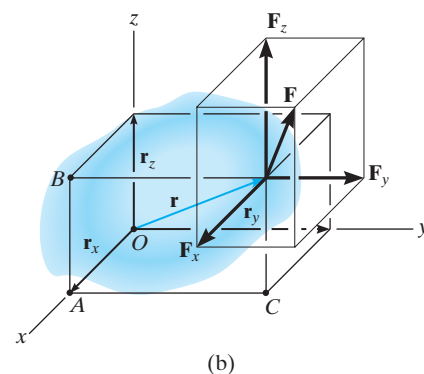
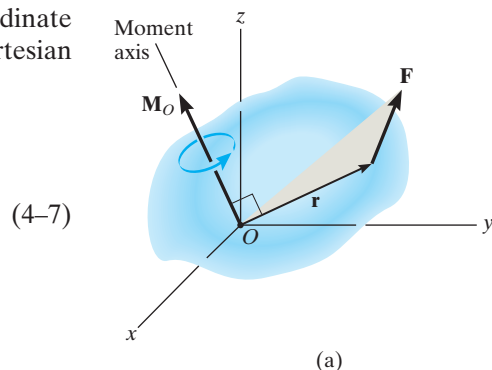


Fig. 4-12

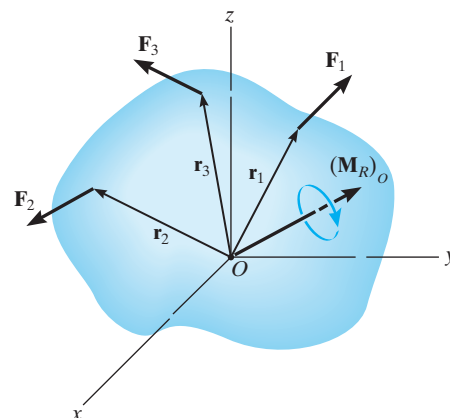


Fig. 4-13

## EXAMPLE 4.3

Determine the moment produced by the force  $\mathbf{F}$  in Fig. 4-14a about point  $O$ . Express the result as a Cartesian vector.

## SOLUTION

As shown in Fig. 4-14b, either  $\mathbf{r}_A$  or  $\mathbf{r}_B$  can be used to determine the moment about point  $O$ . These position vectors are

$$\mathbf{r}_A = \{12\mathbf{k}\} \text{ m} \text{ and } \mathbf{r}_B = \{4\mathbf{i} + 12\mathbf{j}\} \text{ m}$$

Force  $\mathbf{F}$  expressed as a Cartesian vector is

$$\begin{aligned} \mathbf{F} &= F\mathbf{u}_{AB} = 2 \text{ kN} \left[ \frac{\{4\mathbf{i} + 12\mathbf{j} - 12\mathbf{k}\} \text{ m}}{\sqrt{(4 \text{ m})^2 + (12 \text{ m})^2 + (-12 \text{ m})^2}} \right] \\ &= \{0.4588\mathbf{i} + 1.376\mathbf{j} - 1.376\mathbf{k}\} \text{ kN} \end{aligned}$$

Thus

$$\begin{aligned} \mathbf{M}_O &= \mathbf{r}_A \times \mathbf{F} = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 0 & 0 & 12 \\ 0.4588 & 1.376 & -1.376 \end{vmatrix} \\ &= [0(-1.376) - 12(1.376)]\mathbf{i} - [0(-1.376) - 12(0.4588)]\mathbf{j} \\ &\quad + [0(1.376) - 0(0.4588)]\mathbf{k} \\ &= \{-16.5\mathbf{i} + 5.51\mathbf{j}\} \text{ kN} \cdot \text{m} \end{aligned}$$

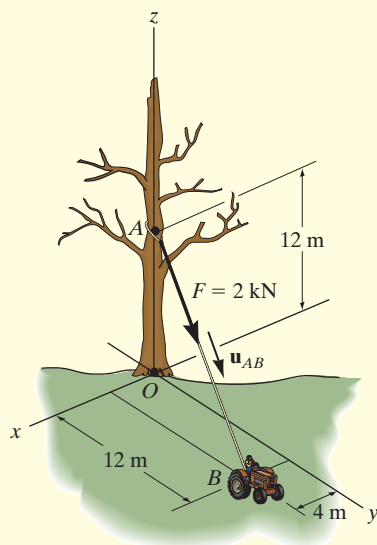
*Ans.*

or

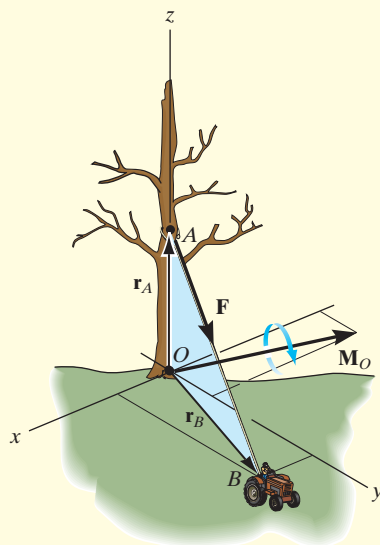
$$\begin{aligned} \mathbf{M}_O &= \mathbf{r}_B \times \mathbf{F} = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 4 & 12 & 0 \\ 0.4588 & 1.376 & -1.376 \end{vmatrix} \\ &= [12(-1.376) - 0(1.376)]\mathbf{i} - [4(-1.376) - 0(0.4588)]\mathbf{j} \\ &\quad + [4(1.376) - 12(0.4588)]\mathbf{k} \\ &= \{-16.5\mathbf{i} + 5.51\mathbf{j}\} \text{ kN} \cdot \text{m} \end{aligned}$$

*Ans.*

**NOTE:** As shown in Fig. 4-14b,  $\mathbf{M}_O$  acts perpendicular to the plane that contains  $\mathbf{F}$ ,  $\mathbf{r}_A$ , and  $\mathbf{r}_B$ . Had this problem been worked using  $M_O = Fd$ , notice the difficulty that would arise in obtaining the moment arm  $d$ .



(a)

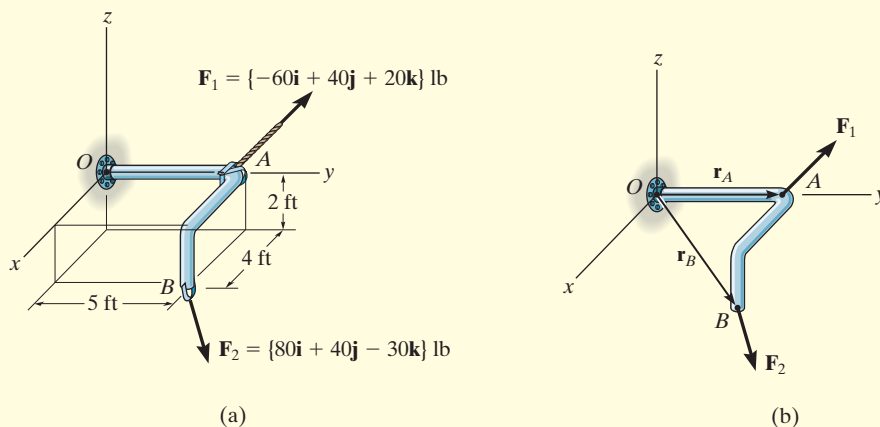


(b)

Fig. 4-14

**EXAMPLE 4.4**

Two forces act on the rod shown in Fig. 4–15a. Determine the resultant moment they create about the flange at  $O$ . Express the result as a Cartesian vector.

**SOLUTION**

Position vectors are directed from point  $O$  to each force as shown in Fig. 4–15b. These vectors are

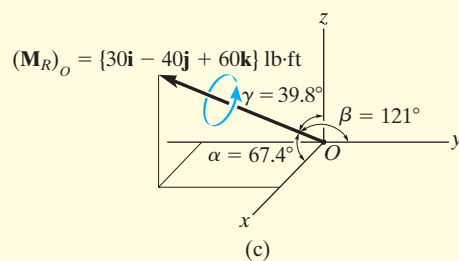
$$\mathbf{r}_A = \{5\mathbf{j}\} \text{ ft}$$

$$\mathbf{r}_B = \{4\mathbf{i} + 5\mathbf{j} - 2\mathbf{k}\} \text{ ft}$$

The resultant moment about  $O$  is therefore

$$\begin{aligned} (\mathbf{M}_R)_O &= \Sigma(\mathbf{r} \times \mathbf{F}) \\ &= \mathbf{r}_A \times \mathbf{F}_1 + \mathbf{r}_B \times \mathbf{F}_2 \\ &= \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 0 & 5 & 0 \\ -60 & 40 & 20 \end{vmatrix} + \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 4 & 5 & -2 \\ 80 & 40 & -30 \end{vmatrix} \\ &= [5(20) - 0(40)]\mathbf{i} - [0]\mathbf{j} + [0(40) - (5)(-60)]\mathbf{k} \\ &\quad + [5(-30) - (-2)(40)]\mathbf{i} - [4(-30) - (-2)(80)]\mathbf{j} + [4(40) - 5(80)]\mathbf{k} \\ &= \{30\mathbf{i} - 40\mathbf{j} + 60\mathbf{k}\} \text{ lb} \cdot \text{ft} \end{aligned}$$

*Ans.*



**Fig. 4–15**

**NOTE:** This result is shown in Fig. 4–15c. The coordinate direction angles were determined from the unit vector for  $(\mathbf{M}_R)_O$ . Realize that the two forces tend to cause the rod to rotate about the moment axis in the manner shown by the curl indicated on the moment vector.

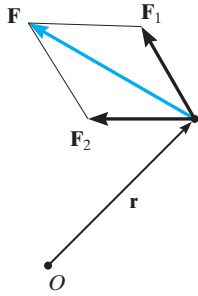


Fig. 4-16

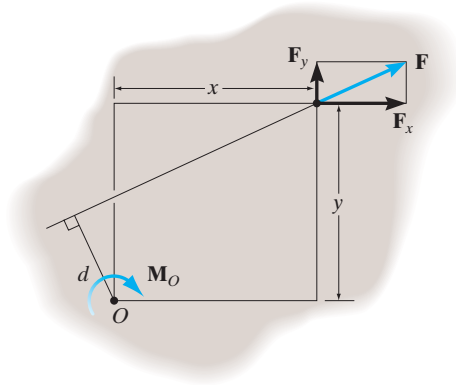
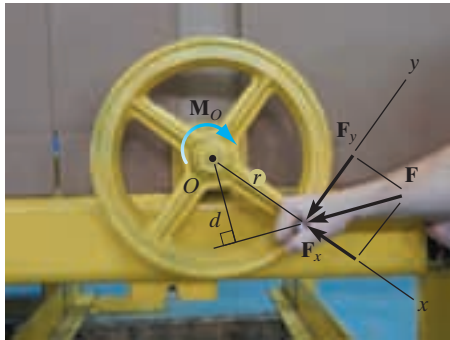


Fig. 4-17



The moment of the force about point  $O$  is  $M_O = Fd$ . But it is easier to find this moment using  $M_O = F_x(0) + F_y r = F_y r$ . (© Russell C. Hibbeler)

## 4.4 Principle of Moments

A concept often used in mechanics is the *principle of moments*, which is sometimes referred to as **Varignon's theorem** since it was originally developed by the French mathematician Pierre Varignon (1654–1722). It states that *the moment of a force about a point is equal to the sum of the moments of the components of the force about the point*. This theorem can be proven easily using the vector cross product since the cross product obeys the *distributive law*. For example, consider the moments of the force  $\mathbf{F}$  and two of its components about point  $O$ , Fig. 4-16. Since  $\mathbf{F} = \mathbf{F}_1 + \mathbf{F}_2$  we have

$$\mathbf{M}_O = \mathbf{r} \times \mathbf{F} = \mathbf{r} \times (\mathbf{F}_1 + \mathbf{F}_2) = \mathbf{r} \times \mathbf{F}_1 + \mathbf{r} \times \mathbf{F}_2$$

For two-dimensional problems, Fig. 4-17, we can use the principle of moments by resolving the force into its rectangular components and then determine the moment using a scalar analysis. Thus,

$$M_O = F_x y - F_y x$$

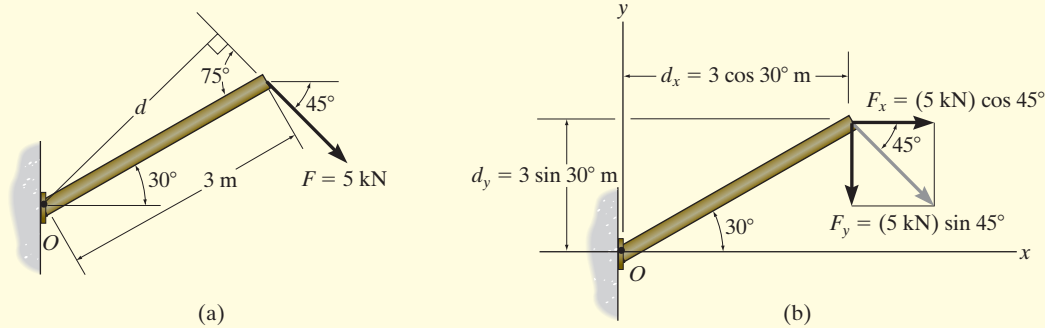
This method is generally easier than finding the same moment using  $M_O = Fd$ .

### Important Points

- The moment of a force creates the tendency of a body to turn about an axis passing through a specific point  $O$ .
- Using the right-hand rule, the sense of rotation is indicated by the curl of the fingers, and the thumb is directed along the moment axis, or line of action of the moment.
- The magnitude of the moment is determined from  $M_O = Fd$ , where  $d$  is called the moment arm, which represents the perpendicular or shortest distance from point  $O$  to the line of action of the force.
- In three dimensions the vector cross product is used to determine the moment, i.e.,  $\mathbf{M}_O = \mathbf{r} \times \mathbf{F}$ . Remember that  $\mathbf{r}$  is directed *from* point  $O$  *to any point* on the line of action of  $\mathbf{F}$ .

**EXAMPLE 4.5**

Determine the moment of the force in Fig. 4–18a about point  $O$ .

**SOLUTION I**

The moment arm  $d$  in Fig. 4–18a can be found from trigonometry.

$$d = (3 \text{ m}) \sin 75^\circ = 2.898 \text{ m}$$

Thus,

$$M_O = Fd = (5 \text{ kN})(2.898 \text{ m}) = 14.5 \text{ kN} \cdot \text{m} \quad \text{Ans.}$$

Since the force tends to rotate or orbit clockwise about point  $O$ , the moment is directed into the page.

**SOLUTION II**

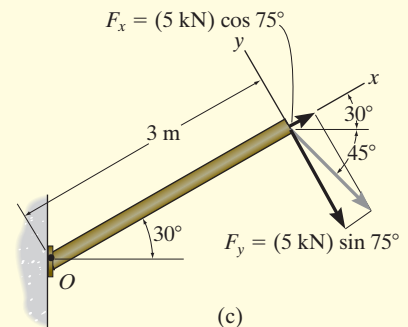
The  $x$  and  $y$  components of the force are indicated in Fig. 4–18b. Considering counterclockwise moments as positive, and applying the principle of moments, we have

$$\begin{aligned} \zeta + M_O &= -F_x d_y - F_y d_x \\ &= -(5 \cos 45^\circ \text{ kN})(3 \sin 30^\circ \text{ m}) - (5 \sin 45^\circ \text{ kN})(3 \cos 30^\circ \text{ m}) \\ &= -14.5 \text{ kN} \cdot \text{m} = 14.5 \text{ kN} \cdot \text{m} \quad \text{Ans.} \end{aligned}$$

**SOLUTION III**

The  $x$  and  $y$  axes can be set parallel and perpendicular to the rod's axis as shown in Fig. 4–18c. Here  $F_x$  produces no moment about point  $O$  since its line of action passes through this point. Therefore,

$$\begin{aligned} \zeta + M_O &= -F_y d_x \\ &= -(5 \sin 75^\circ \text{ kN})(3 \text{ m}) \\ &= -14.5 \text{ kN} \cdot \text{m} = 14.5 \text{ kN} \cdot \text{m} \quad \text{Ans.} \end{aligned}$$

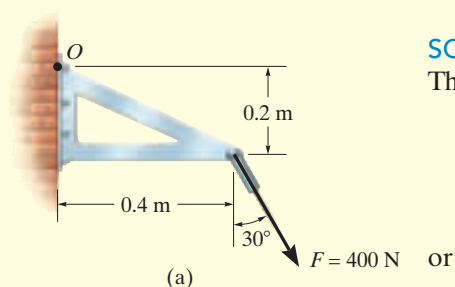


**Fig. 4–18**



**EXAMPLE 4.6**

Force  $\mathbf{F}$  acts at the end of the angle bracket in Fig. 4–19a. Determine the moment of the force about point  $O$ .

**SOLUTION I (SCALAR ANALYSIS)**

The force is resolved into its  $x$  and  $y$  components, Fig. 4–19b, then

$$\begin{aligned}\zeta + M_O &= 400 \sin 30^\circ \text{ N}(0.2 \text{ m}) - 400 \cos 30^\circ \text{ N}(0.4 \text{ m}) \\ &= -98.6 \text{ N} \cdot \text{m} = 98.6 \text{ N} \cdot \text{m} \curvearrowright\end{aligned}$$

$$\mathbf{M}_O = \{-98.6\mathbf{k}\} \text{ N} \cdot \text{m}$$

*Ans.***SOLUTION II (VECTOR ANALYSIS)**

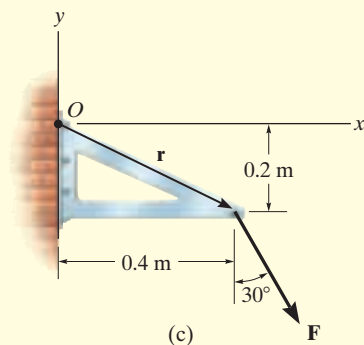
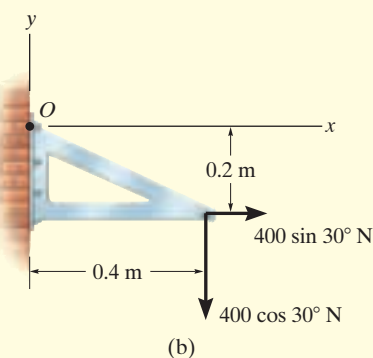
Using a Cartesian vector approach, the force and position vectors, Fig. 4–19c, are

$$\mathbf{r} = \{0.4\mathbf{i} - 0.2\mathbf{j}\} \text{ m}$$

$$\begin{aligned}\mathbf{F} &= \{400 \sin 30^\circ \mathbf{i} - 400 \cos 30^\circ \mathbf{j}\} \text{ N} \\ &= \{200.0\mathbf{i} - 346.4\mathbf{j}\} \text{ N}\end{aligned}$$

The moment is therefore

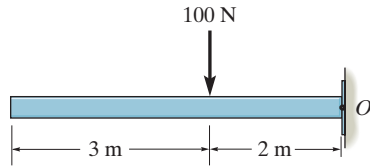
$$\begin{aligned}\mathbf{M}_O &= \mathbf{r} \times \mathbf{F} = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 0.4 & -0.2 & 0 \\ 200.0 & -346.4 & 0 \end{vmatrix} \\ &= 0\mathbf{i} - 0\mathbf{j} + [0.4(-346.4) - (-0.2)(200.0)]\mathbf{k} \\ &= \{-98.6\mathbf{k}\} \text{ N} \cdot \text{m}\end{aligned}$$

*Ans.***Fig. 4–19**

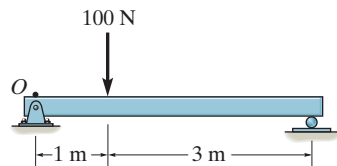
**NOTE:** It is seen that the scalar analysis (Solution I) provides a more *convenient method* for analysis than Solution II since the direction of the moment and the moment arm for each component force are easy to establish. Hence, this method is generally recommended for solving problems displayed in two dimensions, whereas a Cartesian vector analysis is generally recommended only for solving three-dimensional problems.

## PRELIMINARY PROBLEMS

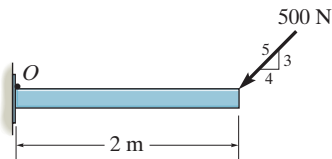
**P4-1.** In each case, determine the moment of the force about point  $O$ .



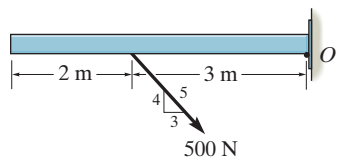
(a)



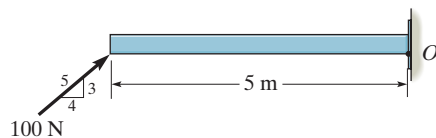
(b)



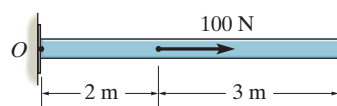
(c)



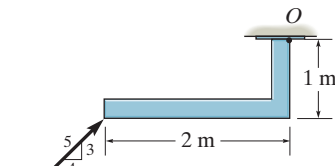
(d)



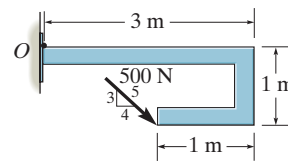
(e)



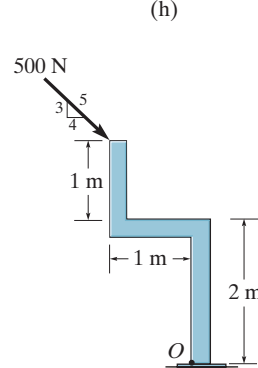
(f)



(g)

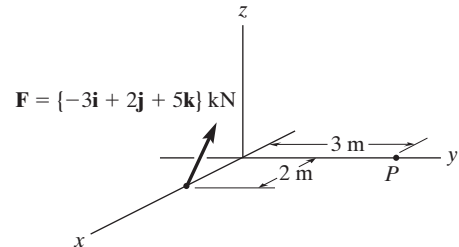


(h)

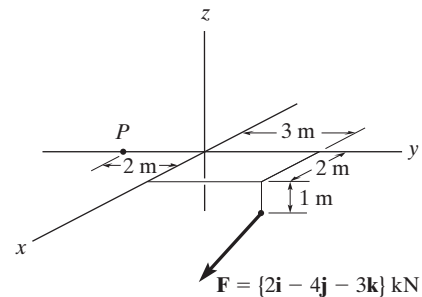


(i)

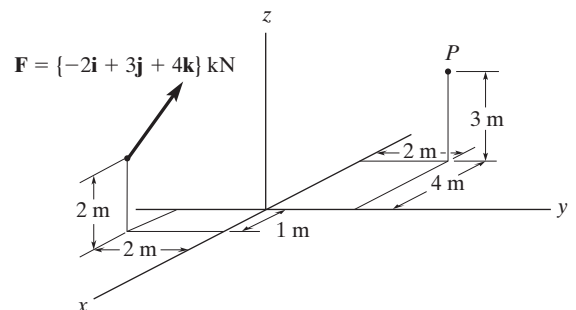
**P4-2.** In each case, set up the determinant to find the moment of the force about point  $P$ .



(a)



(b)



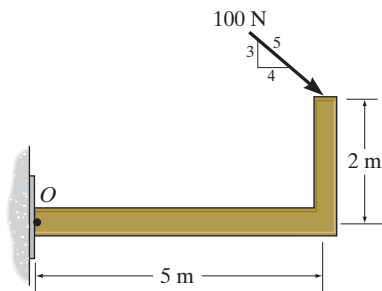
(c)

**Prob. P4-1**

**Prob. P4-2**

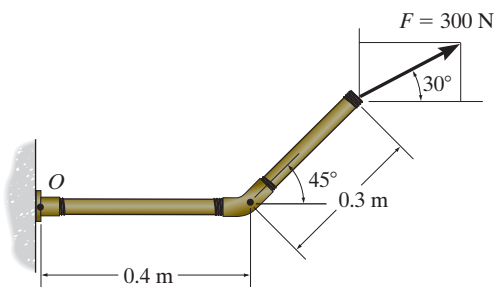
## FUNDAMENTAL PROBLEMS

**F4-1.** Determine the moment of the force about point  $O$ .



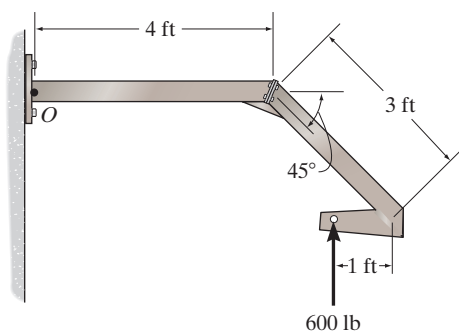
**Prob. F4-1**

**F4-2.** Determine the moment of the force about point  $O$ .



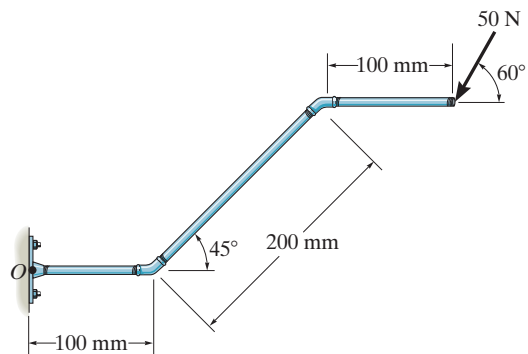
**Prob. F4-2**

**F4-3.** Determine the moment of the force about point  $O$ .



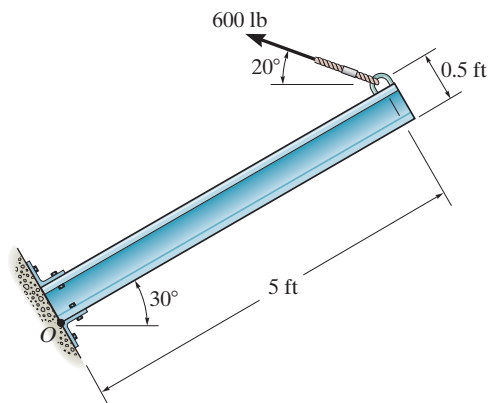
**Prob. F4-3**

**F4-4.** Determine the moment of the force about point  $O$ . Neglect the thickness of the member.



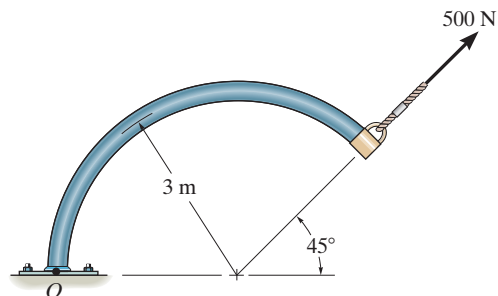
**Prob. F4-4**

**F4-5.** Determine the moment of the force about point  $O$ .



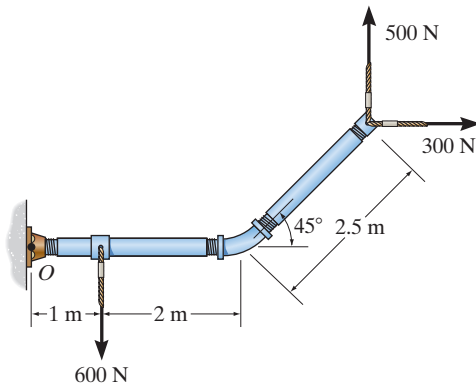
**Prob. F4-5**

**F4-6.** Determine the moment of the force about point  $O$ .



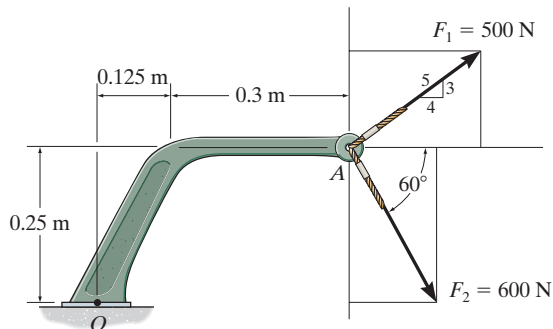
**Prob. F4-6**

**F4-7.** Determine the resultant moment produced by the forces about point  $O$ .



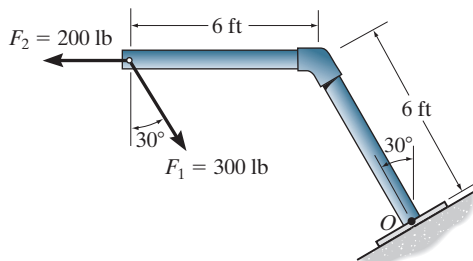
**Prob. F4-7**

**F4-8.** Determine the resultant moment produced by the forces about point  $O$ .



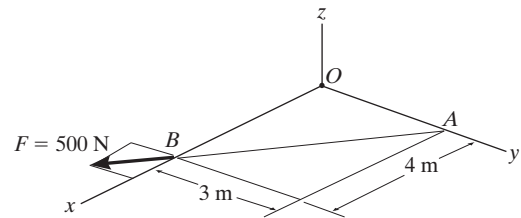
**Prob. F4-8**

**F4-9.** Determine the resultant moment produced by the forces about point  $O$ .



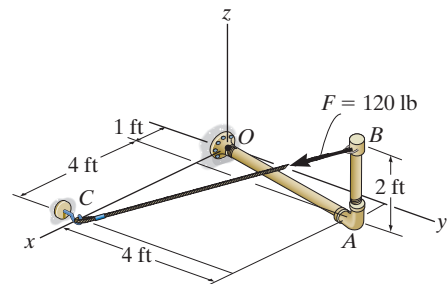
**Prob. F4-9**

**F4-10.** Determine the moment of force  $\mathbf{F}$  about point  $O$ . Express the result as a Cartesian vector.



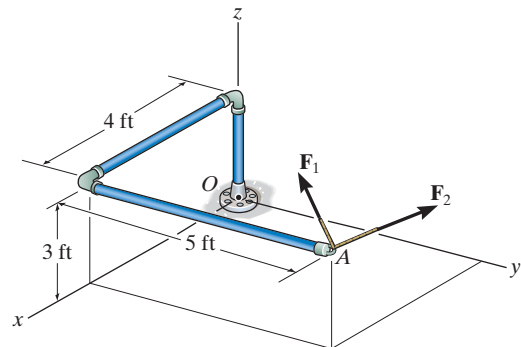
**Prob. F4-10**

**F4-11.** Determine the moment of force  $\mathbf{F}$  about point  $O$ . Express the result as a Cartesian vector.



**Prob. F4-11**

**F4-12.** If the two forces  $\mathbf{F}_1 = \{100\mathbf{i} - 120\mathbf{j} + 75\mathbf{k}\}$  lb and  $\mathbf{F}_2 = \{-200\mathbf{i} + 250\mathbf{j} + 100\mathbf{k}\}$  lb act at  $A$ , determine the resultant moment produced by these forces about point  $O$ . Express the result as a Cartesian vector.



**Prob. F4-12**

## PROBLEMS

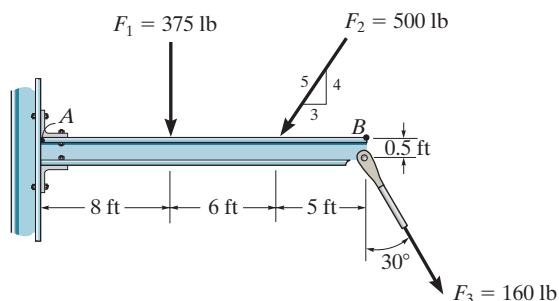
**4-1.** If  $\mathbf{A}$ ,  $\mathbf{B}$ , and  $\mathbf{D}$  are given vectors, prove the distributive law for the vector cross product, i.e.,  $\mathbf{A} \times (\mathbf{B} + \mathbf{D}) = (\mathbf{A} \times \mathbf{B}) + (\mathbf{A} \times \mathbf{D})$ .

**4-2.** Prove the triple scalar product identity  $\mathbf{A} \cdot (\mathbf{B} \times \mathbf{C}) = (\mathbf{A} \times \mathbf{B}) \cdot \mathbf{C}$ .

**4-3.** Given the three nonzero vectors  $\mathbf{A}$ ,  $\mathbf{B}$ , and  $\mathbf{C}$ , show that if  $\mathbf{A} \cdot (\mathbf{B} \times \mathbf{C}) = 0$ , the three vectors *must* lie in the same plane.

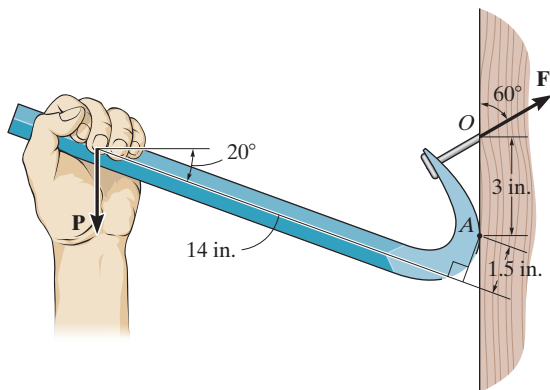
**\*4-4.** Determine the moment about point  $A$  of each of the three forces acting on the beam.

**4-5.** Determine the moment about point  $B$  of each of the three forces acting on the beam.



**Probs. 4-4/5**

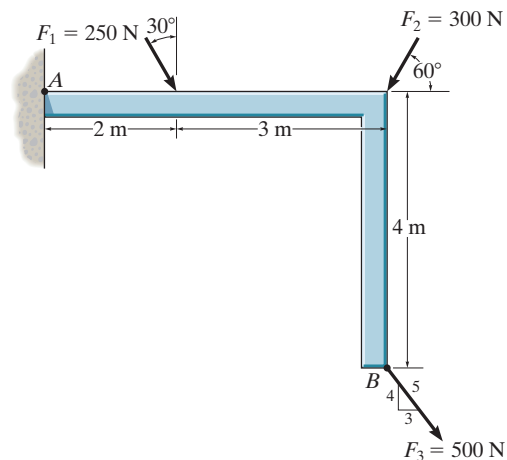
**4-6.** The crowbar is subjected to a vertical force of  $P = 25$  lb at the grip, whereas it takes a force of  $F = 155$  lb at the claw to pull the nail out. Find the moment of each force about point  $A$  and determine if  $\mathbf{P}$  is sufficient to pull out the nail. The crowbar contacts the board at point  $A$ .



**Prob. 4-6**

**4-7.** Determine the moment of each of the three forces about point  $A$ .

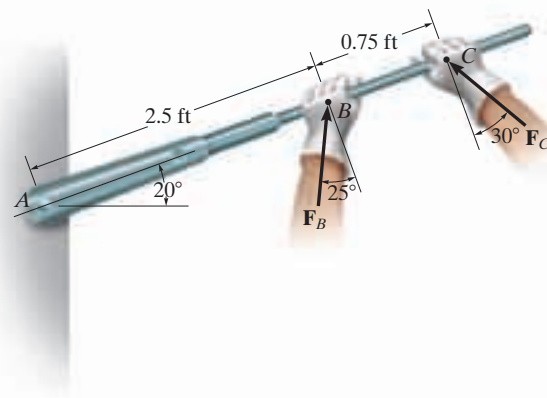
**\*4-8.** Determine the moment of each of the three forces about point  $B$ .



**Probs. 4-7/8**

**4-9.** Determine the moment of each force about the bolt located at  $A$ . Take  $F_B = 40$  lb,  $F_C = 50$  lb.

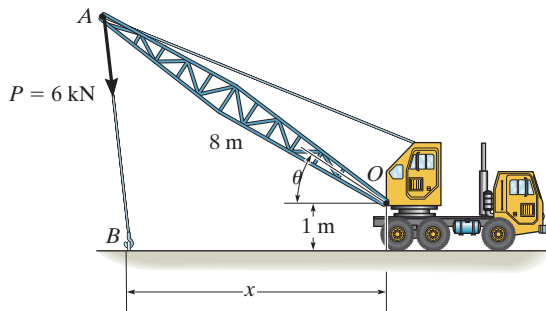
**4-10.** If  $F_B = 30$  lb and  $F_C = 45$  lb, determine the resultant moment about the bolt located at  $A$ .



**Probs. 4-9/10**

**4-11.** The towline exerts a force of  $P = 6 \text{ kN}$  at the end of the 8-m-long crane boom. If  $\theta = 30^\circ$ , determine the placement  $x$  of the hook at  $B$  so that this force creates a maximum moment about point  $O$ . What is this moment?

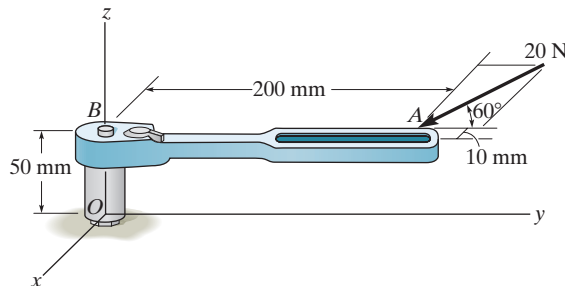
**\*4-12.** The towline exerts a force of  $P = 6 \text{ kN}$  at the end of the 8-m-long crane boom. If  $x = 10 \text{ m}$ , determine the position  $\theta$  of the boom so that this force creates a maximum moment about point  $O$ . What is this moment?



**Probs. 4-11/12**

**4-13.** The 20-N horizontal force acts on the handle of the socket wrench. What is the moment of this force about point  $B$ . Specify the coordinate direction angles  $\alpha$ ,  $\beta$ ,  $\gamma$  of the moment axis.

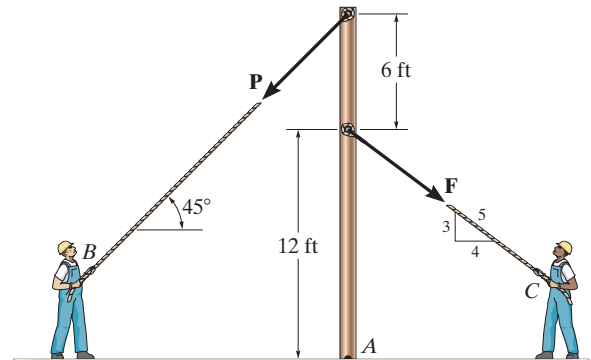
**4-14.** The 20-N horizontal force acts on the handle of the socket wrench. Determine the moment of this force about point  $O$ . Specify the coordinate direction angles  $\alpha$ ,  $\beta$ ,  $\gamma$  of the moment axis.



**Probs. 4-13/14**

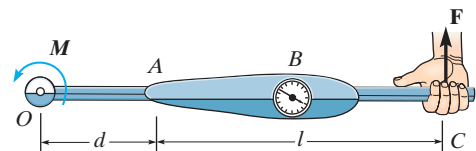
**4-15.** Two men exert forces of  $F = 80 \text{ lb}$  and  $P = 50 \text{ lb}$  on the ropes. Determine the moment of each force about  $A$ . Which way will the pole rotate, clockwise or counterclockwise?

**\*4-16.** If the man at  $B$  exerts a force of  $P = 30 \text{ lb}$  on his rope, determine the magnitude of the force  $\mathbf{F}$  the man at  $C$  must exert to prevent the pole from rotating, i.e., so the resultant moment about  $A$  of both forces is zero.



**Probs. 4-15/16**

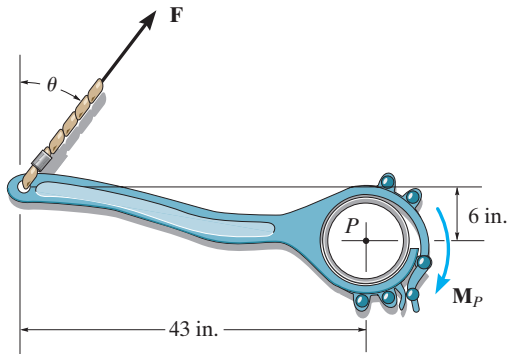
**4-17.** The torque wrench  $ABC$  is used to measure the moment or torque applied to a bolt when the bolt is located at  $A$  and a force is applied to the handle at  $C$ . The mechanic reads the torque on the scale at  $B$ . If an extension  $AO$  of length  $d$  is used on the wrench, determine the required scale reading if the desired torque on the bolt at  $O$  is to be  $M$ .



**Prob. 4-17**

**4-18.** The tongs are used to grip the ends of the drilling pipe  $P$ . Determine the torque (moment)  $M_P$  that the applied force  $F = 150$  lb exerts on the pipe about point  $P$  as a function of  $\theta$ . Plot this moment  $M_P$  versus  $\theta$  for  $0 \leq \theta \leq 90^\circ$ .

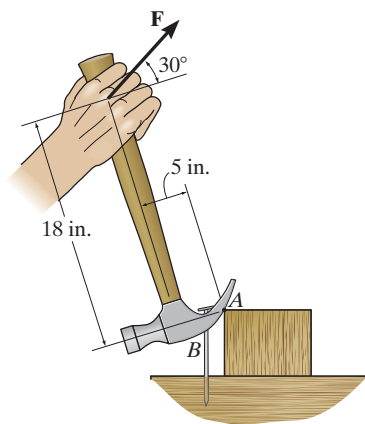
**4-19.** The tongs are used to grip the ends of the drilling pipe  $P$ . If a torque (moment) of  $M_P = 800$  lb·ft is needed at  $P$  to turn the pipe, determine the cable force  $F$  that must be applied to the tongs. Set  $\theta = 30^\circ$ .



**Probs. 4-18/19**

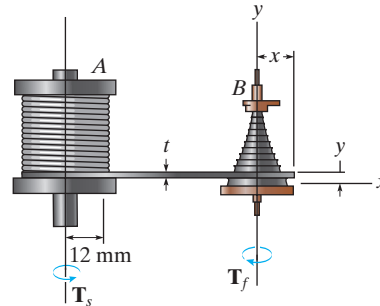
**\*4-20.** The handle of the hammer is subjected to the force of  $F = 20$  lb. Determine the moment of this force about the point  $A$ .

**4-21.** In order to pull out the nail at  $B$ , the force  $F$  exerted on the handle of the hammer must produce a clockwise moment of  $500$  lb·in. about point  $A$ . Determine the required magnitude of force  $F$ .



**Probs. 4-20/21**

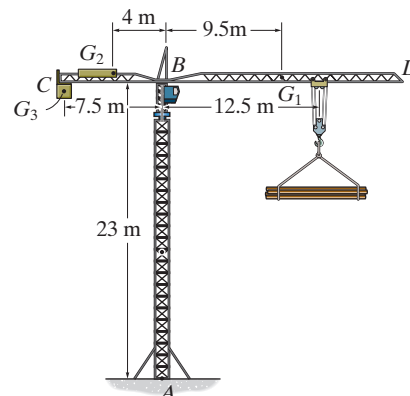
**4-22.** Old clocks were constructed using a *fusee*  $B$  to drive the gears and watch hands. The purpose of the fusee is to increase the leverage developed by the mainspring  $A$  as it uncoils and thereby loses some of its tension. The mainspring can develop a torque (moment)  $T_s = k\theta$ , where  $k = 0.015$  N·m/rad is the torsional stiffness and  $\theta$  is the angle of twist of the spring in radians. If the torque  $T_f$  developed by the fusee is to remain constant as the mainspring winds down, and  $x = 10$  mm when  $\theta = 4$  rad, determine the required radius of the fusee when  $\theta = 3$  rad.



**Prob. 4-22**

**4-23.** The tower crane is used to hoist the 2-Mg load upward at constant velocity. The 1.5-Mg jib  $BD$ , 0.5-Mg jib  $BC$ , and 6-Mg counterweight  $C$  have centers of mass at  $G_1$ ,  $G_2$ , and  $G_3$ , respectively. Determine the resultant moment produced by the load and the weights of the tower crane jibs about point  $A$  and about point  $B$ .

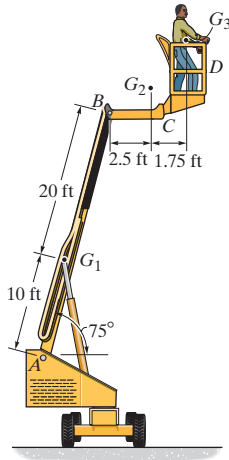
**\*4-24.** The tower crane is used to hoist a 2-Mg load upward at constant velocity. The 1.5-Mg jib  $BD$  and 0.5-Mg jib  $BC$  have centers of mass at  $G_1$  and  $G_2$ , respectively. Determine the required mass of the counterweight  $C$  so that the resultant moment produced by the load and the weight of the tower crane jibs about point  $A$  is zero. The center of mass for the counterweight is located at  $G_3$ .



**Probs. 4-23/24**

**4-25.** If the 1500-lb boom  $AB$ , the 200-lb cage  $BCD$ , and the 175-lb man have centers of gravity located at points  $G_1$ ,  $G_2$ , and  $G_3$ , respectively, determine the resultant moment produced by each weight about point  $A$ .

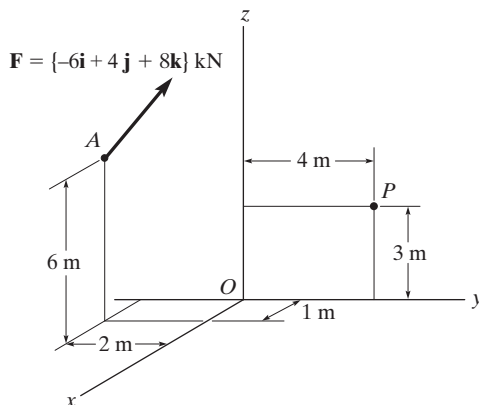
**4-26.** If the 1500-lb boom  $AB$ , the 200-lb cage  $BCD$ , and the 175-lb man have centers of gravity located at points  $G_1$ ,  $G_2$ , and  $G_3$ , respectively, determine the resultant moment produced by all the weights about point  $A$ .



**Probs. 4-25/26**

**4-27.** Determine the moment of the force  $\mathbf{F}$  about point  $O$ . Express the result as a Cartesian vector.

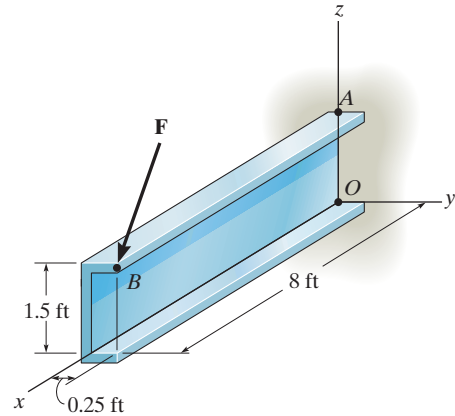
**\*4-28.** Determine the moment of the force  $\mathbf{F}$  about point  $P$ . Express the result as a Cartesian vector.



**Probs. 4-27/28**

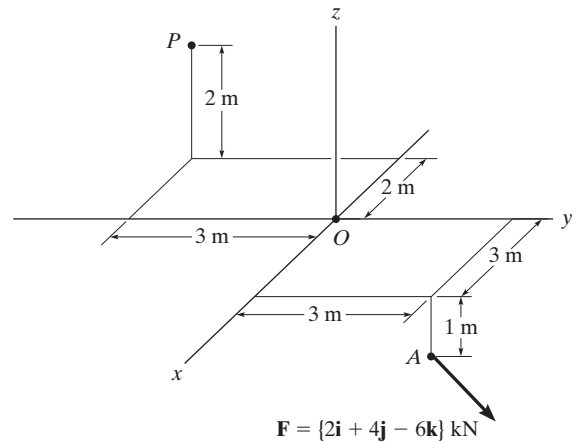
**4-29.** The force  $\mathbf{F} = \{400\mathbf{i} - 100\mathbf{j} - 700\mathbf{k}\}$  lb acts at the end of the beam. Determine the moment of this force about point  $O$ .

**4-30.** The force  $\mathbf{F} = \{400\mathbf{i} - 100\mathbf{j} - 700\mathbf{k}\}$  lb acts at the end of the beam. Determine the moment of this force about point  $A$ .



**Probs. 4-29/30**

**4-31.** Determine the moment of the force  $\mathbf{F}$  about point  $P$ . Express the result as a Cartesian vector.

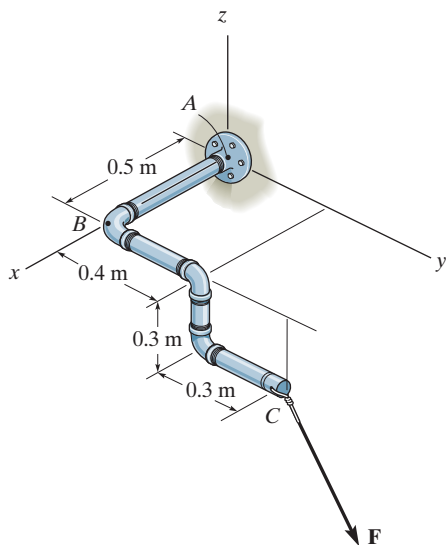


**Prob. 4-31**



**\*4-32.** The pipe assembly is subjected to the force of  $\mathbf{F} = \{600\mathbf{i} + 800\mathbf{j} - 500\mathbf{k}\}$  N. Determine the moment of this force about point A.

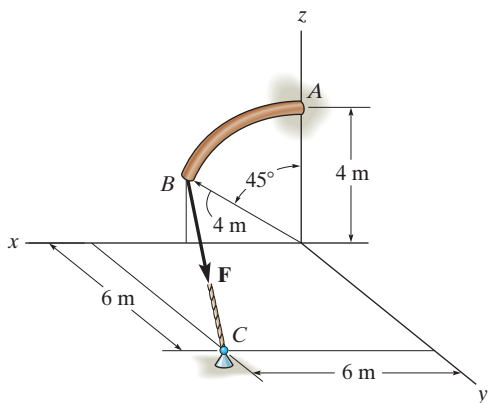
**4-33.** The pipe assembly is subjected to the force of  $\mathbf{F} = \{600\mathbf{i} + 800\mathbf{j} - 500\mathbf{k}\}$  N. Determine the moment of this force about point B.



**Probs. 4-32/33**

**4-34.** Determine the moment of the force of  $F = 600$  N about point A.

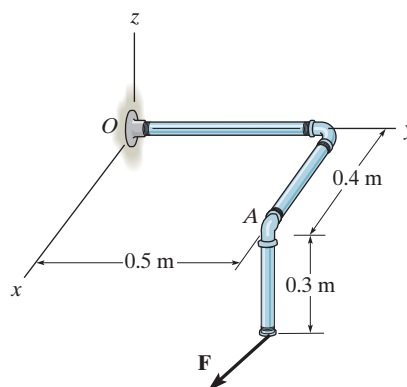
**4-35.** Determine the smallest force  $F$  that must be applied along the rope in order to cause the curved rod, which has a radius of 4 m, to fail at the support A. This requires a moment of  $M = 1500$  N·m to be developed at A.



**Probs. 4-34/35**

**\*4-36.** Determine the coordinate direction angles  $\alpha$ ,  $\beta$ ,  $\gamma$  of force  $\mathbf{F}$ , so that the moment of  $\mathbf{F}$  about O is zero.

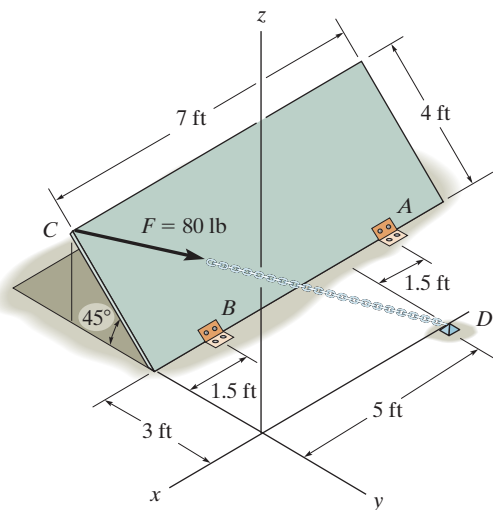
**4-37.** Determine the moment of force  $\mathbf{F}$  about point O. The force has a magnitude of 800 N and coordinate direction angles of  $\alpha = 60^\circ$ ,  $\beta = 120^\circ$ ,  $\gamma = 45^\circ$ . Express the result as a Cartesian vector.



**Probs. 4-36/37**

**4-38.** Determine the moment of the force  $\mathbf{F}$  about the door hinge at A. Express the result as a Cartesian vector.

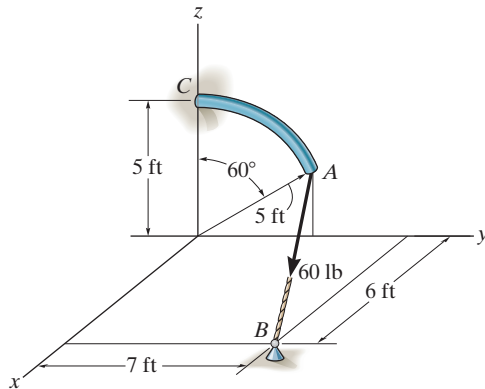
**4-39.** Determine the moment of the force  $\mathbf{F}$  about the door hinge at B. Express the result as a Cartesian vector.



**Probs. 4-38/39**

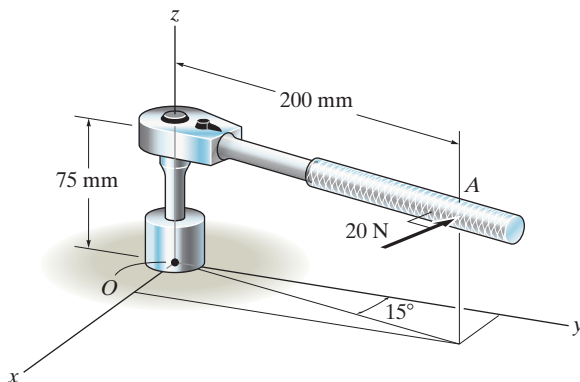
**\*4-40.** The curved rod has a radius of 5 ft. If a force of 60 lb acts at its end as shown, determine the moment of this force about point  $C$ .

**4-41.** Determine the smallest force  $F$  that must be applied along the rope in order to cause the curved rod, which has a radius of 5 ft, to fail at the support  $C$ . This requires a moment of  $M = 80 \text{ lb} \cdot \text{ft}$  to be developed at  $C$ .



**Probs. 4-40/41**

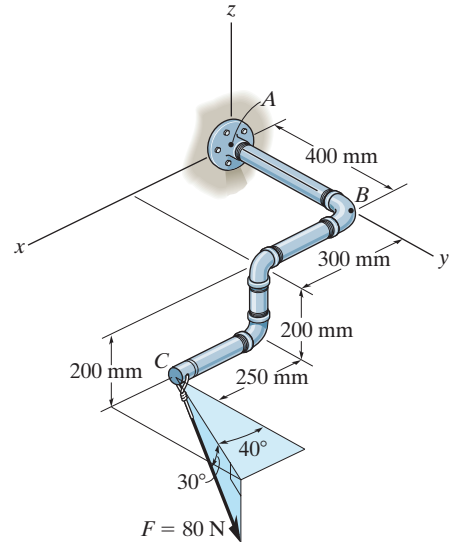
**4-42.** A 20-N horizontal force is applied perpendicular to the handle of the socket wrench. Determine the magnitude and the coordinate direction angles of the moment created by this force about point  $O$ .



**Prob. 4-42**

**4-43.** The pipe assembly is subjected to the 80-N force. Determine the moment of this force about point  $A$ .

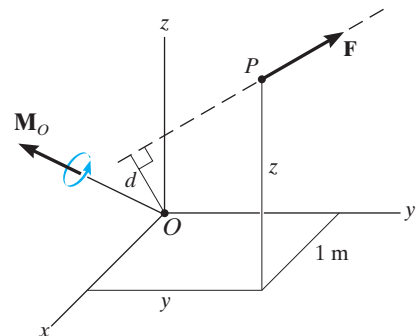
**\*4-44.** The pipe assembly is subjected to the 80-N force. Determine the moment of this force about point  $B$ .



**Probs. 4-43/44**

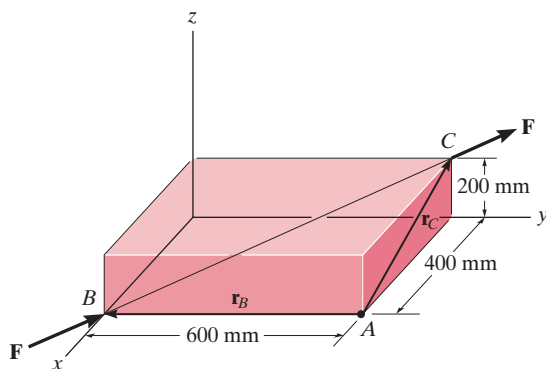
**4-45.** A force of  $\mathbf{F} = \{6\mathbf{i} - 2\mathbf{j} + 1\mathbf{k}\} \text{ kN}$  produces a moment of  $\mathbf{M}_O = \{4\mathbf{i} + 5\mathbf{j} - 14\mathbf{k}\} \text{ kN} \cdot \text{m}$  about the origin, point  $O$ . If the force acts at a point having an  $x$  coordinate of  $x = 1 \text{ m}$ , determine the  $y$  and  $z$  coordinates. *Note:* The figure shows  $\mathbf{F}$  and  $\mathbf{M}_O$  in an arbitrary position.

**4-46.** The force  $\mathbf{F} = \{6\mathbf{i} + 8\mathbf{j} + 10\mathbf{k}\} \text{ N}$  creates a moment about point  $O$  of  $\mathbf{M}_O = \{-14\mathbf{i} + 8\mathbf{j} + 2\mathbf{k}\} \text{ N} \cdot \text{m}$ . If the force passes through a point having an  $x$  coordinate of  $1 \text{ m}$ , determine the  $y$  and  $z$  coordinates of the point. Also, realizing that  $M_O = Fd$ , determine the perpendicular distance  $d$  from point  $O$  to the line of action of  $\mathbf{F}$ . *Note:* The figure shows  $\mathbf{F}$  and  $\mathbf{M}_O$  in an arbitrary position.



**Probs. 4-45/46**

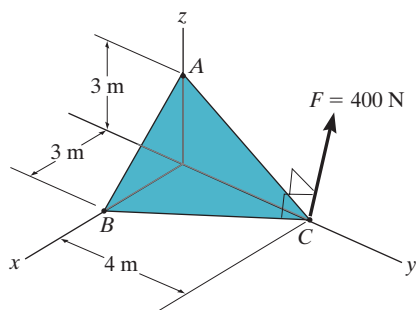
**4-47.** A force  $\mathbf{F}$  having a magnitude of  $F = 100\text{ N}$  acts along the diagonal of the parallelepiped. Determine the moment of  $\mathbf{F}$  about the point  $A$ , using  $\mathbf{M}_A = \mathbf{r}_B \times \mathbf{F}$  and  $\mathbf{M}_A = \mathbf{r}_C \times \mathbf{F}$ .



**Prob. 4-47**

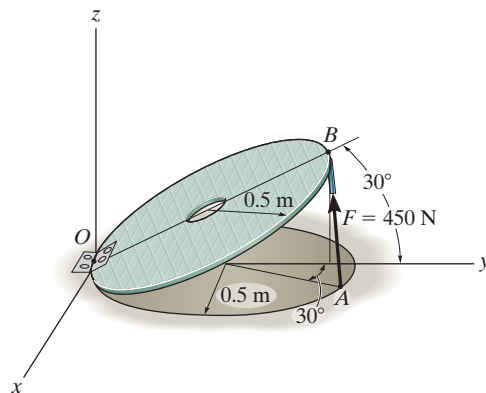
**\*4-48.** Force  $\mathbf{F}$  acts perpendicular to the inclined plane. Determine the moment produced by  $\mathbf{F}$  about point  $A$ . Express the result as a Cartesian vector.

**4-49.** Force  $\mathbf{F}$  acts perpendicular to the inclined plane. Determine the moment produced by  $\mathbf{F}$  about point  $B$ . Express the result as a Cartesian vector.



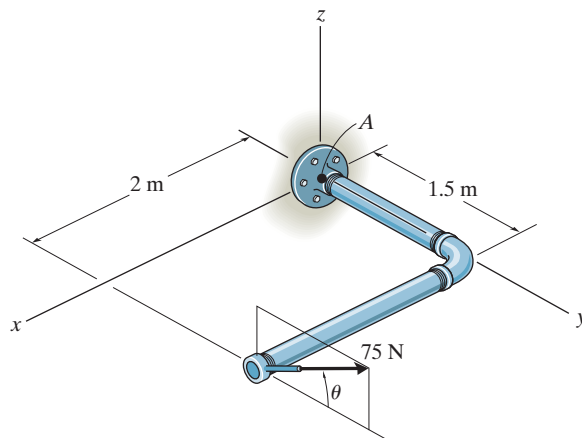
**Probs. 4-48/49**

**4-50.** Strut  $AB$  of the 1-m-diameter hatch door exerts a force of 450 N on point  $B$ . Determine the moment of this force about point  $O$ .



**Prob. 4-50**

**4-51.** Using a ring collar, the 75-N force can act in the vertical plane at various angles  $\theta$ . Determine the magnitude of the moment it produces about point  $A$ , plot the result of  $M$  (ordinate) versus  $\theta$  (abscissa) for  $0^\circ \leq \theta \leq 180^\circ$ , and specify the angles that give the maximum and minimum moment.



**Prob. 4-51**

## 4.5 Moment of a Force about a Specified Axis

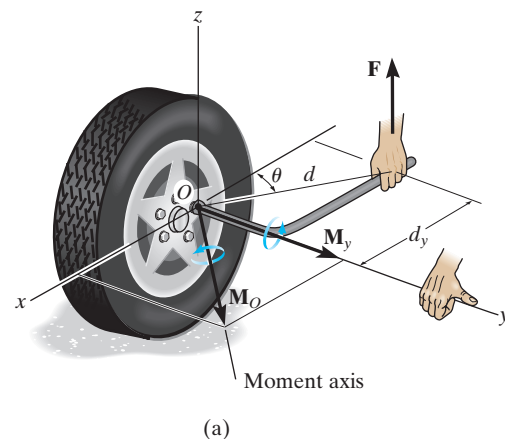
Sometimes, the moment produced by a force about a *specified axis* must be determined. For example, suppose the lug nut at  $O$  on the car tire in Fig. 4–20*a* needs to be loosened. The force applied to the wrench will create a tendency for the wrench and the nut to rotate about the *moment axis* passing through  $O$ ; however, the nut can only rotate about the  $y$  axis. Therefore, to determine the turning effect, only the  $y$  component of the moment is needed, and the total moment produced is not important. To determine this component, we can use either a scalar or vector analysis.

**Scalar Analysis.** To use a scalar analysis in the case of the lug nut in Fig. 4–20*a*, the moment arm, or perpendicular distance from the axis to the line of action of the force, is  $d_y = d \cos \theta$ . Thus, the moment of  $\mathbf{F}$  about the  $y$  axis is  $M_y = F d_y = F(d \cos \theta)$ . According to the right-hand rule,  $\mathbf{M}_y$  is directed along the positive  $y$  axis as shown in the figure. In general, for any axis  $a$ , the moment is

$$M_a = F d_a \quad (4-10)$$



(© Russell C. Hibbeler)



**Fig. 4–20**

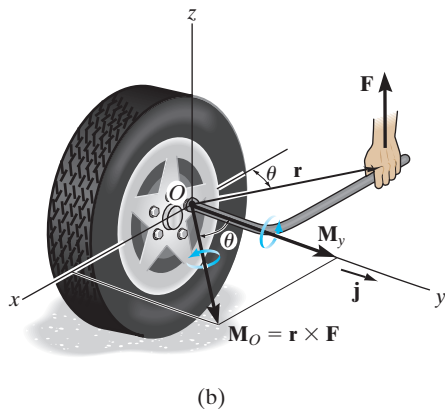


Fig. 4-20 (cont.)

**Vector Analysis.** To find the moment of force  $\mathbf{F}$  in Fig. 4-20b about the  $y$  axis using a vector analysis, we must first determine the moment of the force about *any point*  $O$  on the  $y$  axis by applying Eq. 4-7,  $\mathbf{M}_O = \mathbf{r} \times \mathbf{F}$ . The component  $M_y$  along the  $y$  axis is the *projection* of  $\mathbf{M}_O$  onto the  $y$  axis. It can be found using the *dot product* discussed in Chapter 2, so that  $M_y = \mathbf{j} \cdot \mathbf{M}_O = \mathbf{j} \cdot (\mathbf{r} \times \mathbf{F})$ , where  $\mathbf{j}$  is the unit vector for the  $y$  axis.

We can generalize this approach by letting  $\mathbf{u}_a$  be the unit vector that specifies the direction of the  $a$  axis shown in Fig. 4-21. Then the moment of  $\mathbf{F}$  about a point  $O$  on the axis is  $\mathbf{M}_O = \mathbf{r} \times \mathbf{F}$ , and the projection of this moment onto the  $a$  axis is  $M_a = \mathbf{u}_a \cdot (\mathbf{r} \times \mathbf{F})$ . This combination is referred to as the *scalar triple product*. If the vectors are written in Cartesian form, we have

$$\begin{aligned} M_a &= [u_{ax}\mathbf{i} + u_{ay}\mathbf{j} + u_{az}\mathbf{k}] \cdot \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ r_x & r_y & r_z \\ F_x & F_y & F_z \end{vmatrix} \\ &= u_{ax}(r_y F_z - r_z F_y) - u_{ay}(r_x F_z - r_z F_x) + u_{az}(r_x F_y - r_y F_x) \end{aligned}$$

This result can also be written in the form of a determinant, making it easier to memorize.\*

$$M_a = \mathbf{u}_a \cdot (\mathbf{r} \times \mathbf{F}) = \begin{vmatrix} u_{ax} & u_{ay} & u_{az} \\ r_x & r_y & r_z \\ F_x & F_y & F_z \end{vmatrix} \quad (4-11)$$

where

$u_{ax}, u_{ay}, u_{az}$  represent the  $x, y, z$  components of the unit vector defining the direction of the  $a$  axis

$r_x, r_y, r_z$  represent the  $x, y, z$  components of the position vector extended from *any point*  $O$  on the  $a$  axis to *any point*  $A$  on the line of action of the force

$F_x, F_y, F_z$  represent the  $x, y, z$  components of the force vector.

When  $M_a$  is evaluated from Eq. 4-11, it will yield a positive or negative scalar. The sign of this scalar indicates the sense of direction of  $\mathbf{M}_a$  along the  $a$  axis. If it is positive, then  $\mathbf{M}_a$  will have the same sense as  $\mathbf{u}_a$ , whereas if it is negative, then  $\mathbf{M}_a$  will act opposite to  $\mathbf{u}_a$ . Once the  $a$  axis is established, point your right-hand thumb in the direction of  $\mathbf{M}_a$ , and the curl of your fingers will indicate the sense of twist about the axis, Fig. 4-21.

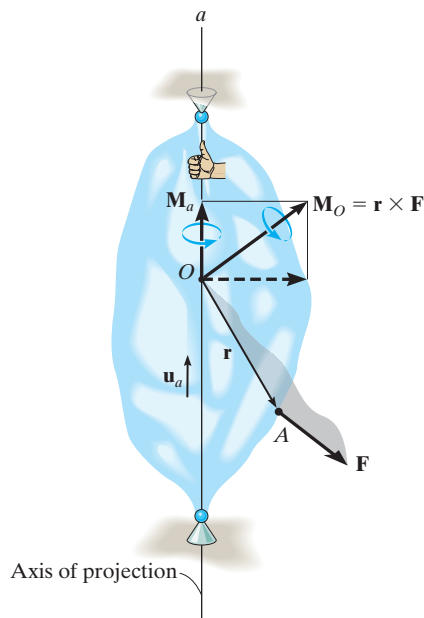


Fig. 4-21

\*Take a minute to expand this determinant, to show that it will yield the above result.

Provided  $M_a$  is determined, we can then express  $\mathbf{M}_a$  as a Cartesian vector, namely,

$$\mathbf{M}_a = M_a \mathbf{u}_a \quad (4-12)$$

The examples which follow illustrate numerical applications of the above concepts.

### Important Points

- The moment of a force about a specified axis can be determined provided the perpendicular distance  $d_a$  from the force line of action to the axis can be determined.  $M_a = Fd_a$ .
- If vector analysis is used,  $M_a = \mathbf{u}_a \cdot (\mathbf{r} \times \mathbf{F})$ , where  $\mathbf{u}_a$  defines the direction of the axis and  $\mathbf{r}$  is extended from *any point* on the axis to *any point* on the line of action of the force.
- If  $M_a$  is calculated as a negative scalar, then the sense of direction of  $\mathbf{M}_a$  is opposite to  $\mathbf{u}_a$ .
- The moment  $\mathbf{M}_a$  expressed as a Cartesian vector is determined from  $\mathbf{M}_a = M_a \mathbf{u}_a$ .

### EXAMPLE 4.7

Determine the resultant moment of the three forces in Fig. 4–22 about the  $x$  axis, the  $y$  axis, and the  $z$  axis.

#### SOLUTION

A force that is *parallel* to a coordinate axis or has a line of action that passes through the axis does *not* produce any moment or tendency for turning about that axis. Therefore, defining the positive direction of the moment of a force according to the right-hand rule, as shown in the figure, we have

$$M_x = (60 \text{ lb})(2 \text{ ft}) + (50 \text{ lb})(2 \text{ ft}) + 0 = 220 \text{ lb} \cdot \text{ft} \quad \text{Ans.}$$

$$M_y = 0 - (50 \text{ lb})(3 \text{ ft}) - (40 \text{ lb})(2 \text{ ft}) = -230 \text{ lb} \cdot \text{ft} \quad \text{Ans.}$$

$$M_z = 0 + 0 - (40 \text{ lb})(2 \text{ ft}) = -80 \text{ lb} \cdot \text{ft} \quad \text{Ans.}$$

The negative signs indicate that  $\mathbf{M}_y$  and  $\mathbf{M}_z$  act in the  $-y$  and  $-z$  directions, respectively.

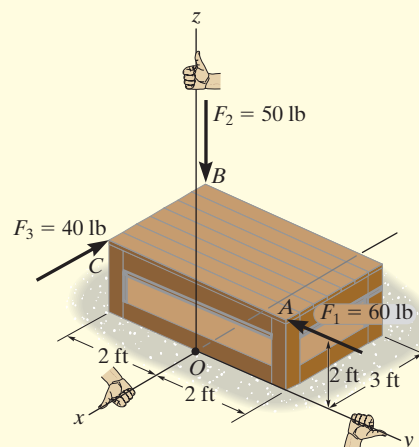


Fig. 4–22

## EXAMPLE 4.8

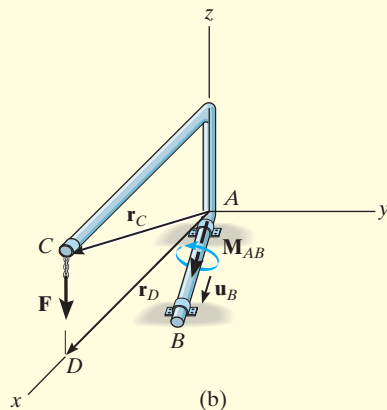
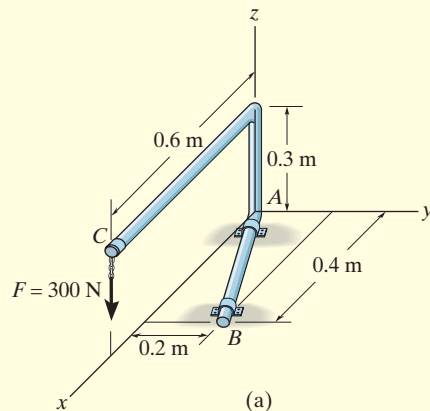


Fig. 4-23

Determine the moment  $\mathbf{M}_{AB}$  produced by the force  $\mathbf{F}$  in Fig. 4-23a, which tends to rotate the rod about the  $AB$  axis.

## SOLUTION

A vector analysis using  $M_{AB} = \mathbf{u}_B \cdot (\mathbf{r} \times \mathbf{F})$  will be considered for the solution rather than trying to find the moment arm or perpendicular distance from the line of action of  $\mathbf{F}$  to the  $AB$  axis. Each of the terms in the equation will now be identified.

Unit vector  $\mathbf{u}_B$  defines the direction of the  $AB$  axis of the rod, Fig. 4-23b, where

$$\mathbf{u}_B = \frac{\mathbf{r}_B}{r_B} = \frac{\{0.4\mathbf{i} + 0.2\mathbf{j}\} \text{ m}}{\sqrt{(0.4 \text{ m})^2 + (0.2 \text{ m})^2}} = 0.8944\mathbf{i} + 0.4472\mathbf{j}$$

Vector  $\mathbf{r}$  is directed from *any point* on the  $AB$  axis to *any point* on the line of action of the force. For example, position vectors  $\mathbf{r}_C$  and  $\mathbf{r}_D$  are suitable, Fig. 4-23b. (Although not shown,  $\mathbf{r}_{BC}$  or  $\mathbf{r}_{BD}$  can also be used.) For simplicity, we choose  $\mathbf{r}_D$ , where

$$\mathbf{r}_D = \{0.6\mathbf{i}\} \text{ m}$$

The force is

$$\mathbf{F} = \{-300\mathbf{k}\} \text{ N}$$

Substituting these vectors into the determinant form and expanding, we have

$$\begin{aligned} M_{AB} &= \mathbf{u}_B \cdot (\mathbf{r}_D \times \mathbf{F}) = \begin{vmatrix} 0.8944 & 0.4472 & 0 \\ 0.6 & 0 & 0 \\ 0 & 0 & -300 \end{vmatrix} \\ &= 0.8944[0(-300) - 0(0)] - 0.4472[0.6(-300) - 0(0)] \\ &\quad + 0[0.6(0) - 0(0)] \\ &= 80.50 \text{ N} \cdot \text{m} \end{aligned}$$

This positive result indicates that the sense of  $\mathbf{M}_{AB}$  is in the same direction as  $\mathbf{u}_B$ .

Expressing  $\mathbf{M}_{AB}$  in Fig. 4-23b as a Cartesian vector yields

$$\begin{aligned} \mathbf{M}_{AB} &= M_{AB}\mathbf{u}_B = (80.50 \text{ N} \cdot \text{m})(0.8944\mathbf{i} + 0.4472\mathbf{j}) \\ &= \{72.0\mathbf{i} + 36.0\mathbf{j}\} \text{ N} \cdot \text{m} \end{aligned}$$

*Ans.*

**NOTE:** If axis  $AB$  is defined using a unit vector directed from  $B$  toward  $A$ , then in the above formulation  $-\mathbf{u}_B$  would have to be used. This would lead to  $M_{AB} = -80.50 \text{ N} \cdot \text{m}$ . Consequently,  $\mathbf{M}_{AB} = M_{AB}(-\mathbf{u}_B)$ , and the same result would be obtained.

**EXAMPLE 4.9**

Determine the magnitude of the moment of force  $\mathbf{F}$  about segment  $OA$  of the pipe assembly in Fig. 4-24a.

**SOLUTION**

The moment of  $\mathbf{F}$  about the  $OA$  axis is determined from  $M_{OA} = \mathbf{u}_{OA} \cdot (\mathbf{r} \times \mathbf{F})$ , where  $\mathbf{r}$  is a position vector extending from any point on the  $OA$  axis to any point on the line of action of  $\mathbf{F}$ . As indicated in Fig. 4-24b, either  $\mathbf{r}_{OD}$ ,  $\mathbf{r}_{OC}$ ,  $\mathbf{r}_{AD}$ , or  $\mathbf{r}_{AC}$  can be used; however,  $\mathbf{r}_{OD}$  will be considered since it will simplify the calculation.

The unit vector  $\mathbf{u}_{OA}$ , which specifies the direction of the  $OA$  axis, is

$$\mathbf{u}_{OA} = \frac{\mathbf{r}_{OA}}{r_{OA}} = \frac{\{0.3\mathbf{i} + 0.4\mathbf{j}\} \text{ m}}{\sqrt{(0.3 \text{ m})^2 + (0.4 \text{ m})^2}} = 0.6\mathbf{i} + 0.8\mathbf{j}$$

and the position vector  $\mathbf{r}_{OD}$  is

$$\mathbf{r}_{OD} = \{0.5\mathbf{i} + 0.5\mathbf{k}\} \text{ m}$$

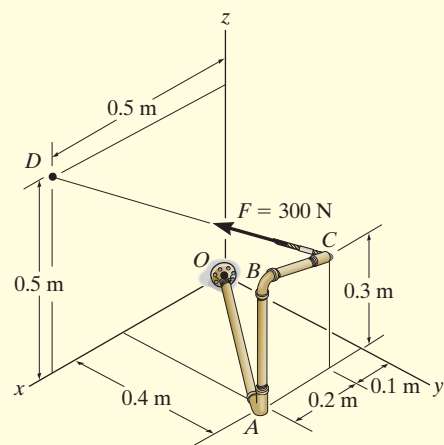
The force  $\mathbf{F}$  expressed as a Cartesian vector is

$$\begin{aligned} \mathbf{F} &= F \left( \frac{\mathbf{r}_{CD}}{r_{CD}} \right) \\ &= (300 \text{ N}) \left[ \frac{\{0.4\mathbf{i} - 0.4\mathbf{j} + 0.2\mathbf{k}\} \text{ m}}{\sqrt{(0.4 \text{ m})^2 + (-0.4 \text{ m})^2 + (0.2 \text{ m})^2}} \right] \\ &= \{200\mathbf{i} - 200\mathbf{j} + 100\mathbf{k}\} \text{ N} \end{aligned}$$

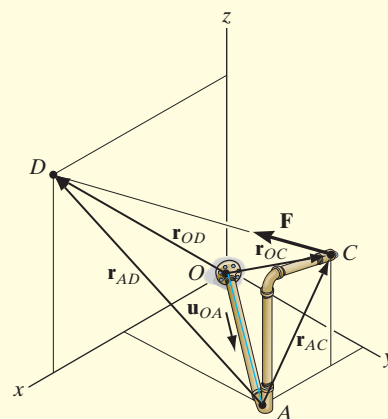
Therefore,

$$\begin{aligned} M_{OA} &= \mathbf{u}_{OA} \cdot (\mathbf{r}_{OD} \times \mathbf{F}) \\ &= \begin{vmatrix} 0.6 & 0.8 & 0 \\ 0.5 & 0 & 0.5 \\ 200 & -200 & 100 \end{vmatrix} \\ &= 0.6[0(100) - (0.5)(-200)] - 0.8[0.5(100) - (0.5)(200)] + 0 \\ &= 100 \text{ N} \cdot \text{m} \end{aligned}$$

*Ans.*



(a)



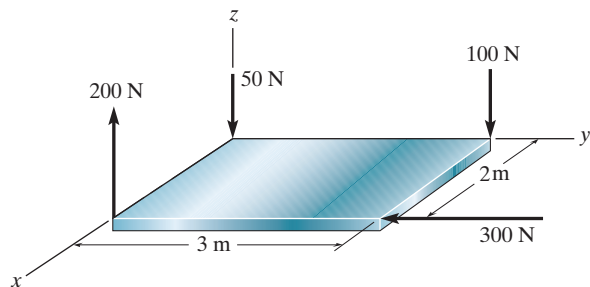
(b)

**Fig. 4-24**

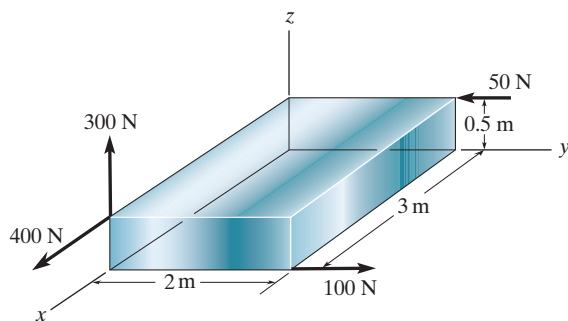


## PRELIMINARY PROBLEMS

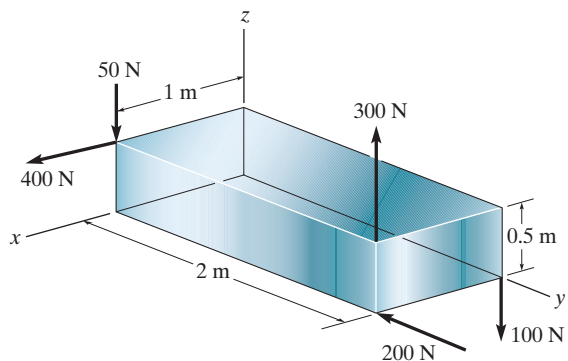
**P4-3.** In each case, determine the resultant moment of the forces acting about the  $x$ ,  $y$ , and  $z$  axes.



(a)



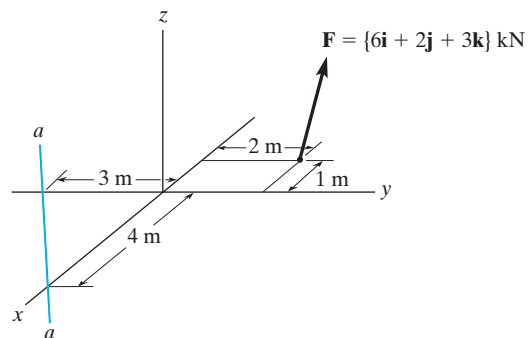
(b)



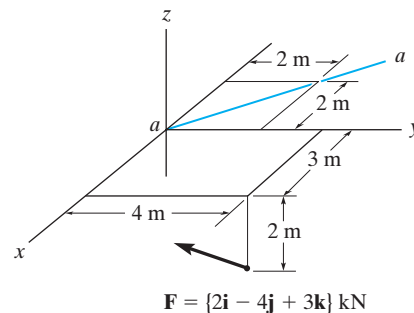
(c)

**Prob. P4-3**

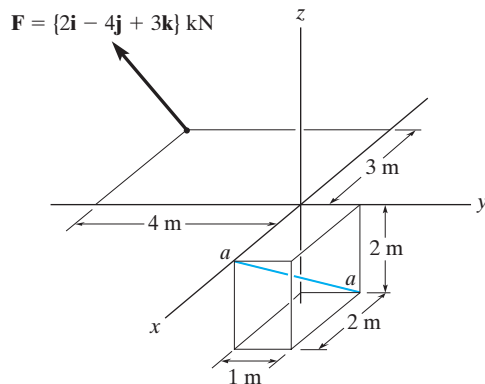
**P4-4.** In each case, set up the determinant needed to find the moment of the force about the  $a$ - $a$  axes.



(a)



(b)



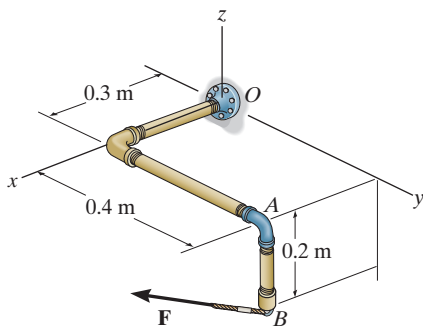
(c)

**Prob. P4-4**

## FUNDAMENTAL PROBLEMS

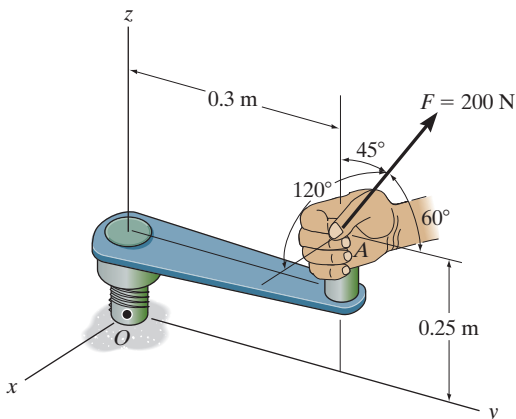
**F4-13.** Determine the magnitude of the moment of the force  $\mathbf{F} = \{300\mathbf{i} - 200\mathbf{j} + 150\mathbf{k}\}$  N about the  $x$  axis.

**F4-14.** Determine the magnitude of the moment of the force  $\mathbf{F} = \{300\mathbf{i} - 200\mathbf{j} + 150\mathbf{k}\}$  N about the  $OA$  axis.



**Probs. F4-13/14**

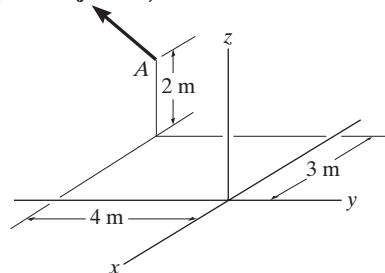
**F4-15.** Determine the magnitude of the moment of the 200-N force about the  $x$  axis. Solve the problem using both a scalar and a vector analysis.



**Prob. F4-15**

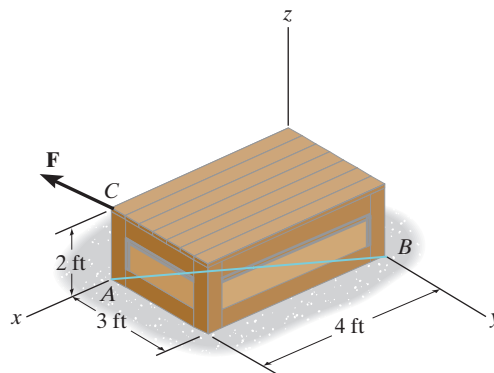
**F4-16.** Determine the magnitude of the moment of the force about the  $y$  axis.

$$\mathbf{F} = \{30\mathbf{i} - 20\mathbf{j} + 50\mathbf{k}\} \text{ N}$$



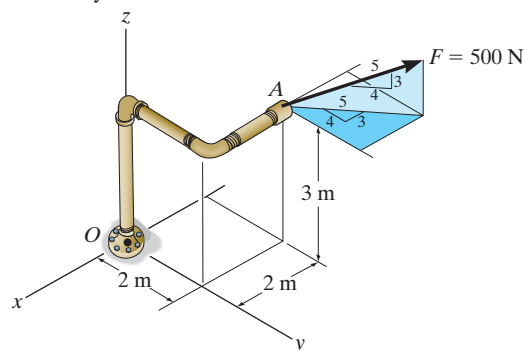
**Prob. F4-16**

**F4-17.** Determine the moment of the force  $\mathbf{F} = \{50\mathbf{i} - 40\mathbf{j} + 20\mathbf{k}\}$  lb about the  $AB$  axis. Express the result as a Cartesian vector.



**Prob. F4-17**

**F4-18.** Determine the moment of force  $\mathbf{F}$  about the  $x$ , the  $y$ , and the  $z$  axes. Solve the problem using both a scalar and a vector analysis.

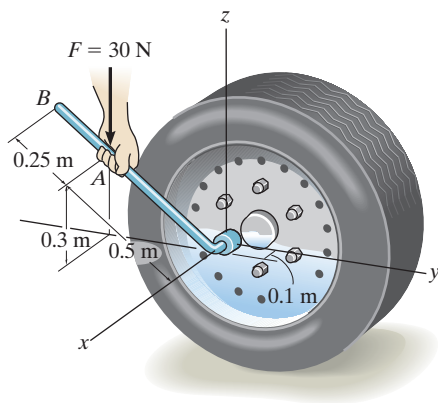


**Prob. F4-18**

## PROBLEMS

**\*4-52.** The lug nut on the wheel of the automobile is to be removed using the wrench and applying the vertical force of  $F = 30 \text{ N}$  at  $A$ . Determine if this force is adequate, provided  $14 \text{ N} \cdot \text{m}$  of torque about the  $x$  axis is initially required to turn the nut. If the  $30\text{-N}$  force can be applied at  $A$  in any other direction, will it be possible to turn the nut?

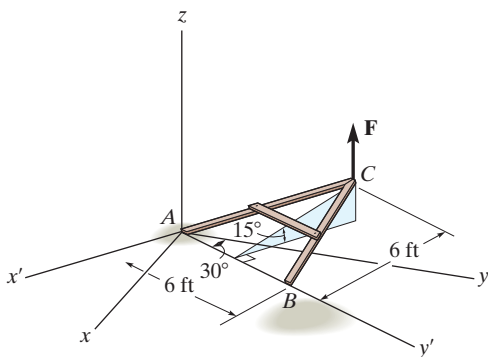
**4-53.** Solve Prob. 4-52 if the cheater pipe  $AB$  is slipped over the handle of the wrench and the  $30\text{-N}$  force can be applied at any point and in any direction on the assembly.



Probs. 4-52/53

**4-54.** The A-frame is being hoisted into an upright position by the vertical force of  $F = 80 \text{ lb}$ . Determine the moment of this force about the  $y'$  axis passing through points  $A$  and  $B$  when the frame is in the position shown.

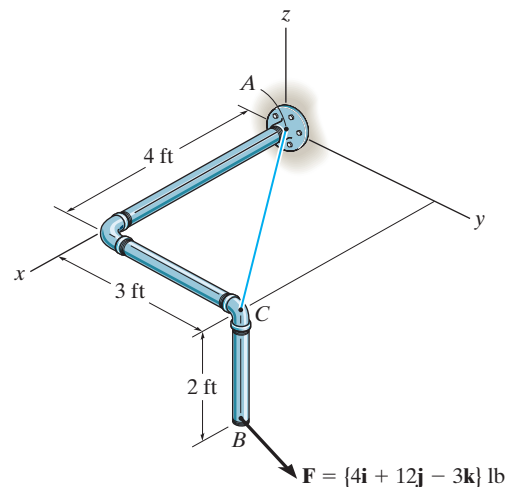
**4-55.** The A-frame is being hoisted into an upright position by the vertical force of  $F = 80 \text{ lb}$ . Determine the moment of this force about the  $x$  axis when the frame is in the position shown.



Probs. 4-54/55

**\*4-56.** Determine the magnitude of the moments of the force  $\mathbf{F}$  about the  $x$ ,  $y$ , and  $z$  axes. Solve the problem (a) using a Cartesian vector approach and (b) using a scalar approach.

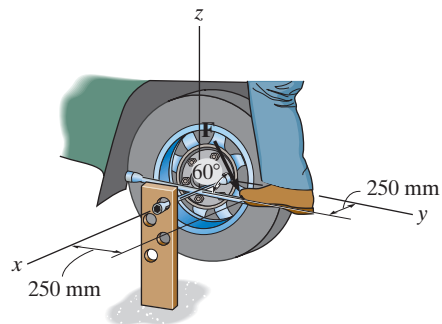
**4-57.** Determine the moment of this force  $\mathbf{F}$  about an axis extending between  $A$  and  $C$ . Express the result as a Cartesian vector.



Probs. 4-56/57

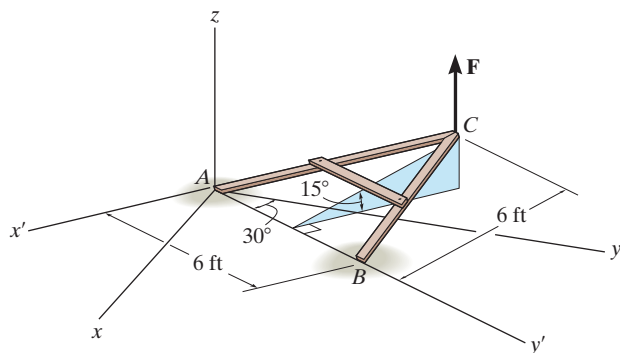
**4-58.** The board is used to hold the end of a four-way lug wrench in the position shown when the man applies a force of  $F = 100 \text{ N}$ . Determine the magnitude of the moment produced by this force about the  $x$  axis. Force  $\mathbf{F}$  lies in a vertical plane.

**4-59.** The board is used to hold the end of a four-way lug wrench in position. If a torque of  $30 \text{ N} \cdot \text{m}$  about the  $x$  axis is required to tighten the nut, determine the required magnitude of the force  $\mathbf{F}$  that the man's foot must apply on the end of the wrench in order to turn it. Force  $\mathbf{F}$  lies in a vertical plane.



Probs. 4-58/59

**\*4-60.** The A-frame is being hoisted into an upright position by the vertical force of  $F = 80$  lb. Determine the moment of this force about the  $y$  axis when the frame is in the position shown.

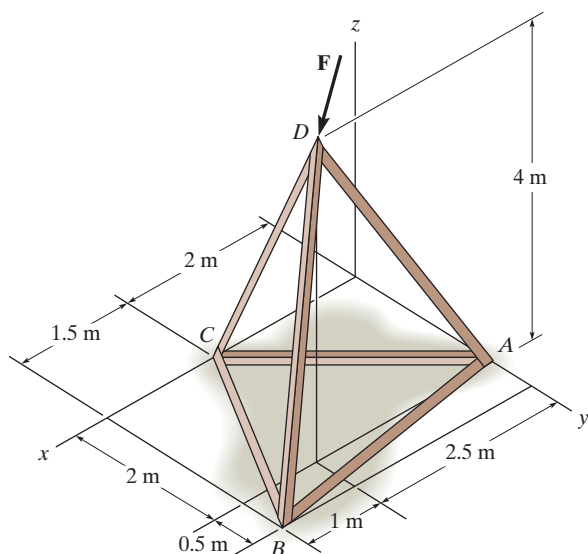


**Prob. 4-60**

**4-61.** Determine the magnitude of the moment of the force  $\mathbf{F} = \{50\mathbf{i} - 20\mathbf{j} + 80\mathbf{k}\}$  N about the base line  $AB$  of the tripod.

**4-62.** Determine the magnitude of the moment of the force  $\mathbf{F} = \{50\mathbf{i} - 20\mathbf{j} + 80\mathbf{k}\}$  N about the base line  $BC$  of the tripod.

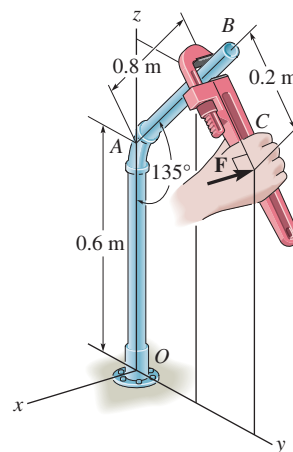
**4-63.** Determine the magnitude of the moment of the force  $\mathbf{F} = \{50\mathbf{i} - 20\mathbf{j} + 80\mathbf{k}\}$  N about the base line  $CA$  of the tripod.



**Probs. 4-61/62/63**

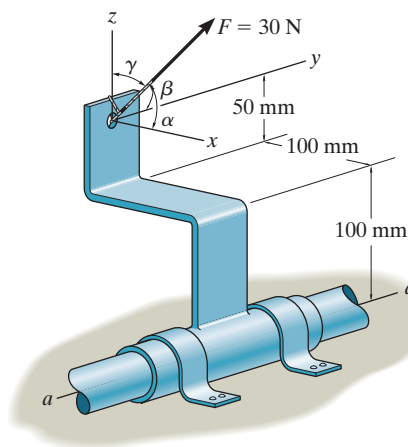
**\*4-64.** A horizontal force of  $\mathbf{F} = \{-50\mathbf{i}\}$  N is applied perpendicular to the handle of the pipe wrench. Determine the moment that this force exerts along the axis  $OA$  ( $z$  axis) of the pipe assembly. Both the wrench and pipe assembly,  $OABC$ , lie in the  $y$ - $z$  plane. *Suggestion:* Use a scalar analysis.

**4-65.** Determine the magnitude of the horizontal force  $\mathbf{F} = -F\mathbf{i}$  acting on the handle of the wrench so that this force produces a component of the moment along the  $OA$  axis ( $z$  axis) of the pipe assembly of  $\mathbf{M}_z = \{4\mathbf{k}\}$  N·m. Both the wrench and the pipe assembly,  $OABC$ , lie in the  $y$ - $z$  plane. *Suggestion:* Use a scalar analysis.



**Probs. 4-64/65**

**4-66.** The force of  $F = 30$  N acts on the bracket as shown. Determine the moment of the force about the  $a$ - $a$  axis of the pipe if  $\alpha = 60^\circ$ ,  $\beta = 60^\circ$ , and  $\gamma = 45^\circ$ . Also, determine the coordinate direction angles of  $F$  in order to produce the maximum moment about the  $a$ - $a$  axis. What is this moment?



**Prob. 4-66**

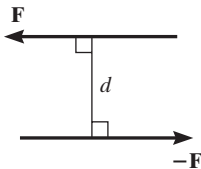


Fig. 4-25

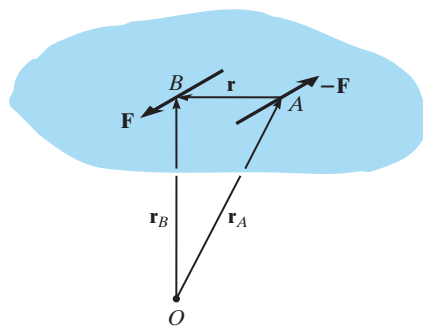


Fig. 4-26

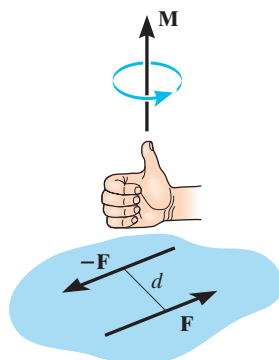


Fig. 4-27

## 4.6 Moment of a Couple

A **couple** is defined as two parallel forces that have the same magnitude, but opposite directions, and are separated by a perpendicular distance  $d$ , Fig. 4-25. Since the resultant force is zero, the only effect of a couple is to produce an actual rotation, or if no movement is possible, there is a tendency of rotation in a specified direction. For example, imagine that you are driving a car with both hands on the steering wheel and you are making a turn. One hand will push up on the wheel while the other hand pulls down, which causes the steering wheel to rotate.

The moment produced by a couple is called a **couple moment**. We can determine its value by finding the sum of the moments of both couple forces about *any* arbitrary point. For example, in Fig. 4-26, position vectors  $r_A$  and  $r_B$  are directed from point  $O$  to points  $A$  and  $B$  lying on the line of action of  $-F$  and  $F$ . The couple moment determined about  $O$  is therefore

$$\mathbf{M} = \mathbf{r}_B \times \mathbf{F} + \mathbf{r}_A \times -\mathbf{F} = (\mathbf{r}_B - \mathbf{r}_A) \times \mathbf{F}$$

However  $\mathbf{r}_B = \mathbf{r}_A + \mathbf{r}$  or  $\mathbf{r} = \mathbf{r}_B - \mathbf{r}_A$ , so that

$$\mathbf{M} = \mathbf{r} \times \mathbf{F} \quad (4-13)$$

This result indicates that a couple moment is a **free vector**, i.e., it can act at *any point* since  $\mathbf{M}$  depends *only* upon the position vector  $\mathbf{r}$  directed *between* the forces and *not* the position vectors  $\mathbf{r}_A$  and  $\mathbf{r}_B$ , directed from the arbitrary point  $O$  to the forces. This concept is unlike the moment of a force, which requires a definite point (or axis) about which moments are determined.

**Scalar Formulation.** The moment of a couple,  $\mathbf{M}$ , Fig. 4-27, is defined as having a *magnitude* of

$$M = Fd \quad (4-14)$$

where  $F$  is the magnitude of one of the forces and  $d$  is the perpendicular distance or moment arm between the forces. The *direction* and *sense* of the couple moment are determined by the right-hand rule, where the thumb indicates this direction when the fingers are curled with the sense of rotation caused by the couple forces. In all cases,  $\mathbf{M}$  will act perpendicular to the plane containing these forces.

**Vector Formulation.** The moment of a couple can also be expressed by the vector cross product using Eq. 4-13, i.e.,

$$\mathbf{M} = \mathbf{r} \times \mathbf{F} \quad (4-15)$$

Application of this equation is easily remembered if one thinks of taking the moments of both forces about a point lying on the line of action of one of the forces. For example, if moments are taken about point  $A$  in Fig. 4-26, the moment of  $-F$  is *zero* about this point, and the moment of  $F$  is defined from Eq. 4-15. Therefore, in the formulation  $\mathbf{r}$  is crossed with the force  $\mathbf{F}$  to which it is directed.

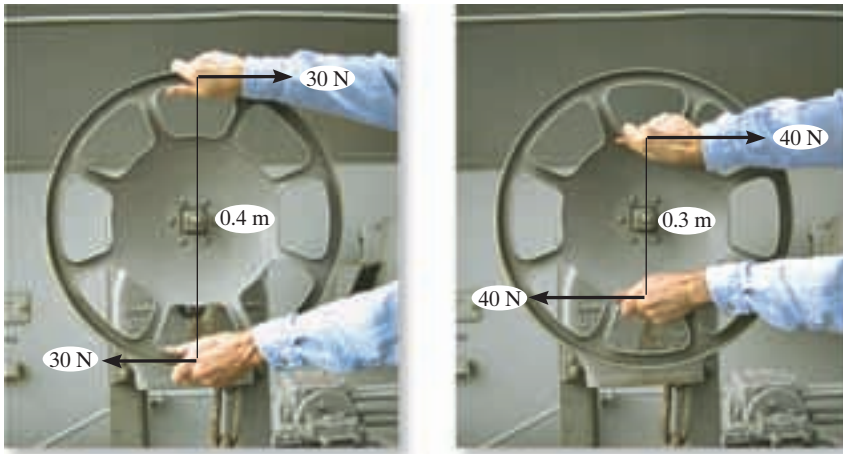


Fig. 4-28 (© Russell C. Hibbeler)

**Equivalent Couples.** If two couples produce a moment with the *same magnitude and direction*, then these two couples are *equivalent*. For example, the two couples shown in Fig. 4-28 are *equivalent* because each couple moment has a magnitude of  $M = 30 \text{ N}(0.4 \text{ m}) = 40 \text{ N}(0.3 \text{ m}) = 12 \text{ N} \cdot \text{m}$ , and each is directed into the plane of the page. Notice that larger forces are required in the second case to create the same turning effect because the hands are placed closer together. Also, if the wheel was connected to the shaft at a point other than at its center, then the wheel would still turn when each couple is applied since the  $12 \text{ N} \cdot \text{m}$  couple is a free vector.

**Resultant Couple Moment.** Since couple moments are vectors, their resultant can be determined by vector addition. For example, consider the couple moments  $\mathbf{M}_1$  and  $\mathbf{M}_2$  acting on the pipe in Fig. 4-29a. Since each couple moment is a free vector, we can join their tails at any arbitrary point and find the resultant couple moment,  $\mathbf{M}_R = \mathbf{M}_1 + \mathbf{M}_2$  as shown in Fig. 4-29b.

If more than two couple moments act on the body, we may generalize this concept and write the vector resultant as

$$\mathbf{M}_R = \Sigma(\mathbf{r} \times \mathbf{F}) \quad (4-16)$$

These concepts are illustrated numerically in the examples that follow. In general, problems projected in two dimensions should be solved using a scalar analysis since the moment arms and force components are easy to determine.

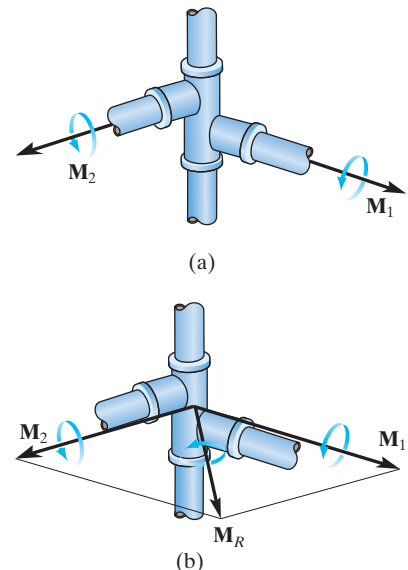


Fig. 4-29



Steering wheels on vehicles have been made smaller than on older vehicles because power steering does not require the driver to apply a large couple moment to the rim of the wheel. (© Russell C. Hibbeler)

## Important Points

- A couple moment is produced by two noncollinear forces that are equal in magnitude but opposite in direction. Its effect is to produce pure rotation, or tendency for rotation in a specified direction.
- A couple moment is a free vector, and as a result it causes the same rotational effect on a body regardless of where the couple moment is applied to the body.
- The moment of the two couple forces can be determined about *any point*. For convenience, this point is often chosen on the line of action of one of the forces in order to eliminate the moment of this force about the point.
- In three dimensions the couple moment is often determined using the vector formulation,  $\mathbf{M} = \mathbf{r} \times \mathbf{F}$ , where  $\mathbf{r}$  is directed from *any point* on the line of action of one of the forces to *any point* on the line of action of the other force  $\mathbf{F}$ .
- A resultant couple moment is simply the vector sum of all the couple moments of the system.

## EXAMPLE 4.10

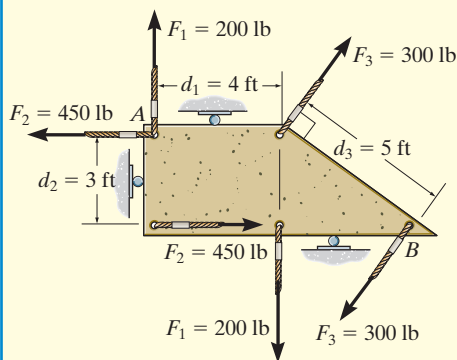


Fig. 4-30

Determine the resultant couple moment of the three couples acting on the plate in Fig. 4-30.

### SOLUTION

As shown the perpendicular distances between each pair of couple forces are  $d_1 = 4$  ft,  $d_2 = 3$  ft, and  $d_3 = 5$  ft. Considering counterclockwise couple moments as positive, we have

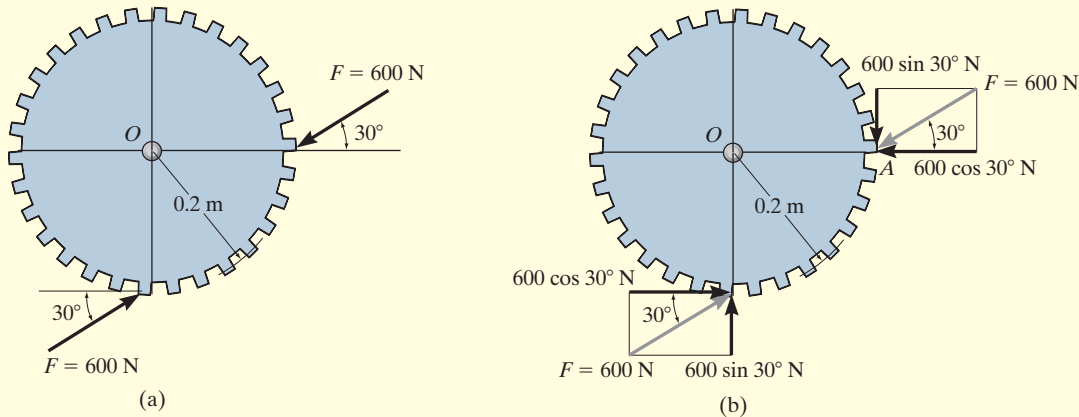
$$\begin{aligned}\zeta + M_R &= \Sigma M; \quad M_R = -F_1 d_1 + F_2 d_2 - F_3 d_3 \\ &= -(200 \text{ lb})(4 \text{ ft}) + (450 \text{ lb})(3 \text{ ft}) - (300 \text{ lb})(5 \text{ ft}) \\ &= -950 \text{ lb} \cdot \text{ft} = 950 \text{ lb} \cdot \text{ft} \curvearrowright\end{aligned}$$

Ans.

The negative sign indicates that  $\mathbf{M}_R$  has a clockwise rotational sense.

**EXAMPLE 4.11**

Determine the magnitude and direction of the couple moment acting on the gear in Fig. 4–31*a*.

**SOLUTION**

The easiest solution requires resolving each force into its components as shown in Fig. 4–31*b*. The couple moment can be determined by summing the moments of these force components about any point, for example, the center  $O$  of the gear or point  $A$ . If we consider counterclockwise moments as positive, we have

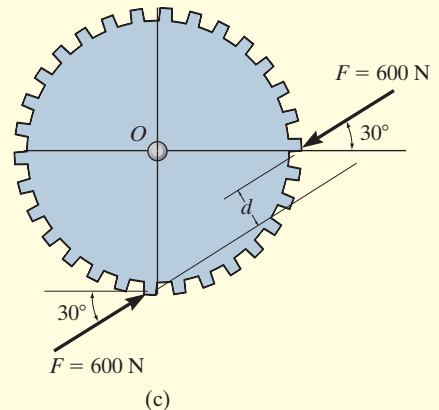
$$\zeta + M = \Sigma M_O; \quad M = (600 \cos 30^\circ \text{ N})(0.2 \text{ m}) - (600 \sin 30^\circ \text{ N})(0.2 \text{ m}) \\ = 43.9 \text{ N} \cdot \text{m} \curvearrowright \quad \text{Ans.}$$

or

$$\zeta + M = \Sigma M_A; \quad M = (600 \cos 30^\circ \text{ N})(0.2 \text{ m}) - (600 \sin 30^\circ \text{ N})(0.2 \text{ m}) \\ = 43.9 \text{ N} \cdot \text{m} \curvearrowright \quad \text{Ans.}$$

This positive result indicates that **M** has a counterclockwise rotational sense, so it is directed outward, perpendicular to the page.

**NOTE:** The same result can also be obtained using  $M = Fd$ , where  $d$  is the perpendicular distance between the lines of action of the couple forces, Fig. 4–31*c*. However, the computation for  $d$  is more involved. Realize that the couple moment is a free vector and can act at any point on the gear and produce the same turning effect about point  $O$ .

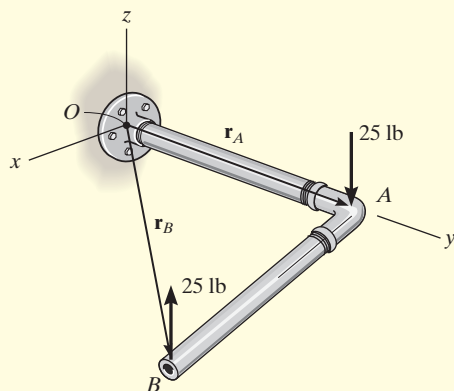


**Fig. 4-31**

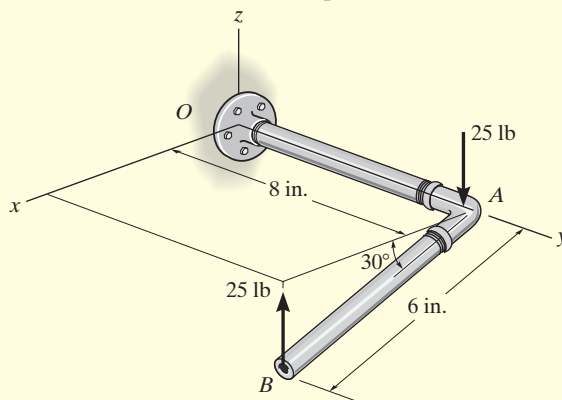


## EXAMPLE 4.12

Determine the couple moment acting on the pipe shown in Fig. 4–32a. Segment  $AB$  is directed  $30^\circ$  below the  $x$ – $y$  plane.



(b)



(a)

## SOLUTION I (VECTOR ANALYSIS)

The moment of the two couple forces can be found about *any* point. If point  $O$  is considered, Fig. 4–32b, we have

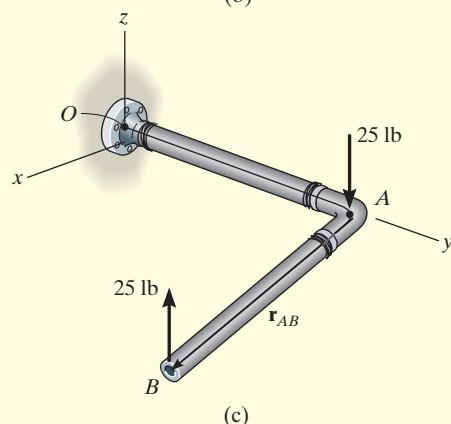
$$\begin{aligned}\mathbf{M} &= \mathbf{r}_A \times (-25\mathbf{k}) + \mathbf{r}_B \times (25\mathbf{k}) \\ &= (8\mathbf{j}) \times (-25\mathbf{k}) + (6 \cos 30^\circ \mathbf{i} + 8\mathbf{j} - 6 \sin 30^\circ \mathbf{k}) \times (25\mathbf{k}) \\ &= -200\mathbf{i} - 129.9\mathbf{j} + 200\mathbf{i} \\ &= \{-130\mathbf{j}\} \text{ lb} \cdot \text{in.}\end{aligned}$$

Ans.

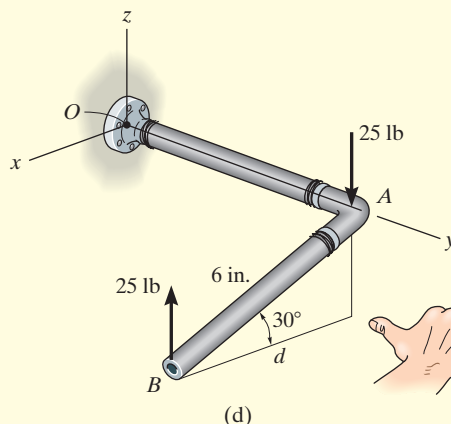
It is *easier* to take moments of the couple forces about a point lying on the line of action of one of the forces, e.g., point  $A$ , Fig. 4–32c. In this case the moment of the force at  $A$  is zero, so that

$$\begin{aligned}\mathbf{M} &= \mathbf{r}_{AB} \times (25\mathbf{k}) \\ &= (6 \cos 30^\circ \mathbf{i} - 6 \sin 30^\circ \mathbf{k}) \times (25\mathbf{k}) \\ &= \{-130\mathbf{j}\} \text{ lb} \cdot \text{in.}\end{aligned}$$

Ans.



(c)



(d)

## SOLUTION II (SCALAR ANALYSIS)

Although this problem is shown in three dimensions, the geometry is simple enough to use the scalar equation  $M = Fd$ . The perpendicular distance between the lines of action of the couple forces is  $d = 6 \cos 30^\circ = 5.196$  in., Fig. 4–32d. Hence, taking moments of the forces about either point  $A$  or point  $B$  yields

$$M = Fd = 25 \text{ lb} (5.196 \text{ in.}) = 129.9 \text{ lb} \cdot \text{in.}$$

Applying the right-hand rule,  $\mathbf{M}$  acts in the  $-\mathbf{j}$  direction. Thus,

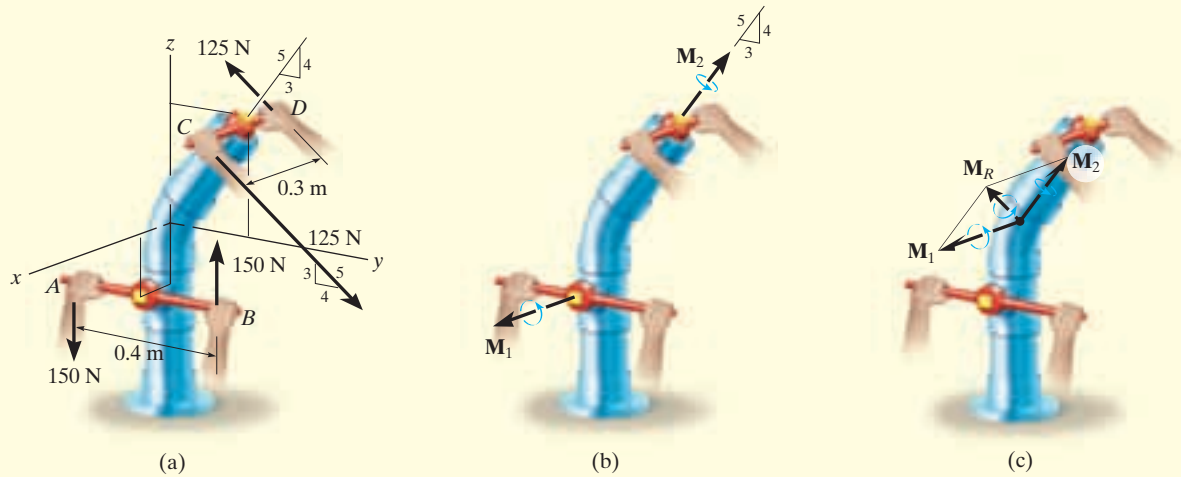
$$\mathbf{M} = \{-130\mathbf{j}\} \text{ lb} \cdot \text{in.}$$

Ans.

Fig. 4–32

**EXAMPLE 4.13**

Replace the two couples acting on the pipe column in Fig. 4–33a by a resultant couple moment.

**Fig. 4–33****SOLUTION (VECTOR ANALYSIS)**

The couple moment  $\mathbf{M}_1$ , developed by the forces at  $A$  and  $B$ , can easily be determined from a scalar formulation.

$$M_1 = Fd = 150 \text{ N}(0.4 \text{ m}) = 60 \text{ N} \cdot \text{m}$$

By the right-hand rule,  $\mathbf{M}_1$  acts in the  $+\mathbf{i}$  direction, Fig. 4–33b. Hence,

$$\mathbf{M}_1 = \{60\mathbf{i}\} \text{ N} \cdot \text{m}$$

Vector analysis will be used to determine  $\mathbf{M}_2$ , caused by forces at  $C$  and  $D$ . If moments are calculated about point  $D$ , Fig. 4–33a,  $\mathbf{M}_2 = \mathbf{r}_{DC} \times \mathbf{F}_C$ , then

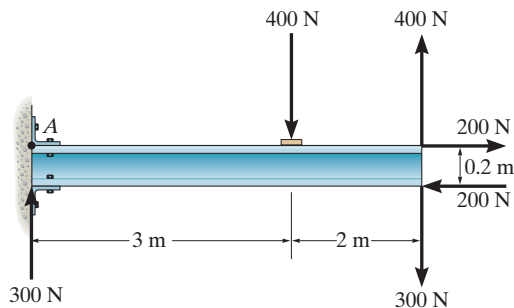
$$\begin{aligned} \mathbf{M}_2 &= \mathbf{r}_{DC} \times \mathbf{F}_C = (0.3\mathbf{i}) \times \left[ 125\left(\frac{4}{5}\right)\mathbf{j} - 125\left(\frac{3}{5}\right)\mathbf{k} \right] \\ &= (0.3\mathbf{i}) \times [100\mathbf{j} - 75\mathbf{k}] = 30(\mathbf{i} \times \mathbf{j}) - 22.5(\mathbf{i} \times \mathbf{k}) \\ &= \{22.5\mathbf{j} + 30\mathbf{k}\} \text{ N} \cdot \text{m} \end{aligned}$$

Since  $\mathbf{M}_1$  and  $\mathbf{M}_2$  are free vectors, they may be moved to some arbitrary point and added vectorially, Fig. 4–33c. The resultant couple moment becomes

$$\mathbf{M}_R = \mathbf{M}_1 + \mathbf{M}_2 = \{60\mathbf{i} + 22.5\mathbf{j} + 30\mathbf{k}\} \text{ N} \cdot \text{m} \quad \text{Ans.}$$

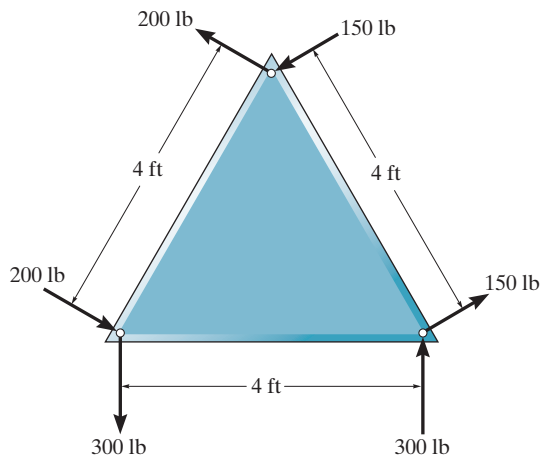
## FUNDAMENTAL PROBLEMS

**F4-19.** Determine the resultant couple moment acting on the beam.



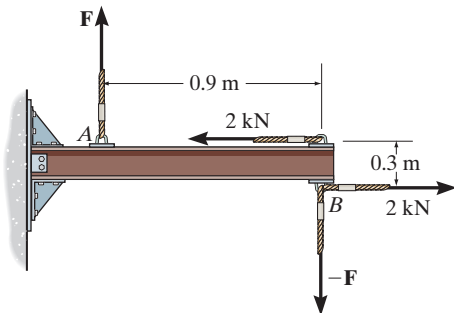
**Prob. F4-19**

**F4-20.** Determine the resultant couple moment acting on the triangular plate.



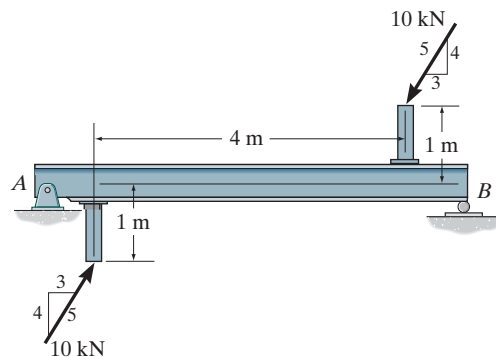
**Prob. F4-20**

**F4-21.** Determine the magnitude of  $F$  so that the resultant couple moment acting on the beam is  $1.5 \text{ kN} \cdot \text{m}$  clockwise.



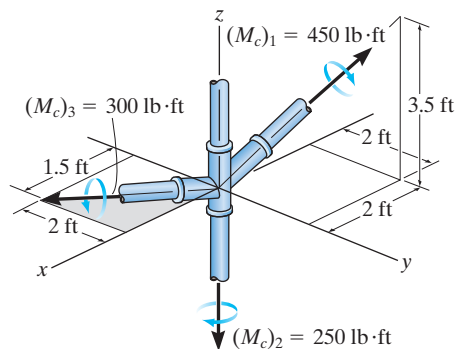
**Prob. F4-21**

**F4-22.** Determine the couple moment acting on the beam.



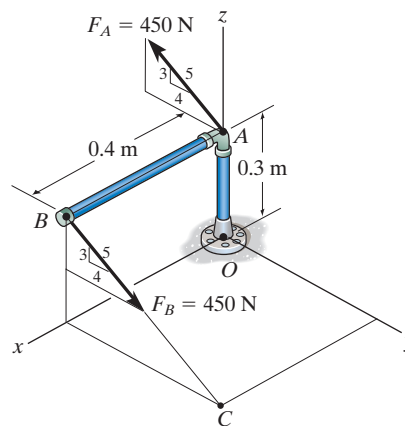
**Prob. F4-22**

**F4-23.** Determine the resultant couple moment acting on the pipe assembly.



**Prob. F4-23**

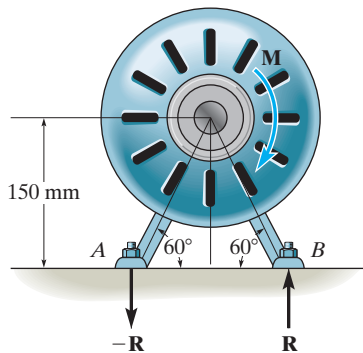
**F4-24.** Determine the couple moment acting on the pipe assembly and express the result as a Cartesian vector.



**Prob. F4-24**

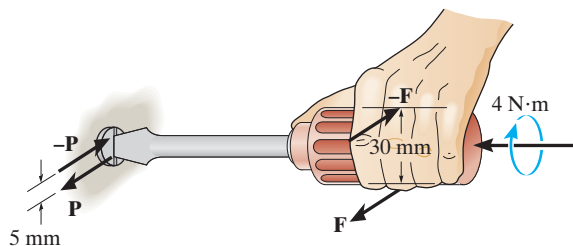
## PROBLEMS

**4-67.** A clockwise couple  $M = 5 \text{ N} \cdot \text{m}$  is resisted by the shaft of the electric motor. Determine the magnitude of the reactive forces  $-\mathbf{R}$  and  $\mathbf{R}$  which act at supports  $A$  and  $B$  so that the resultant of the two couples is zero.



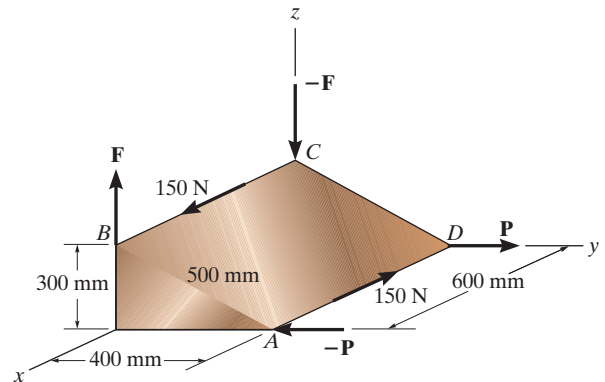
Prob. 4-67

**\*4-68.** A twist of  $4 \text{ N} \cdot \text{m}$  is applied to the handle of the screwdriver. Resolve this couple moment into a pair of couple forces  $\mathbf{F}$  exerted on the handle and  $\mathbf{P}$  exerted on the blade.



Prob. 4-68

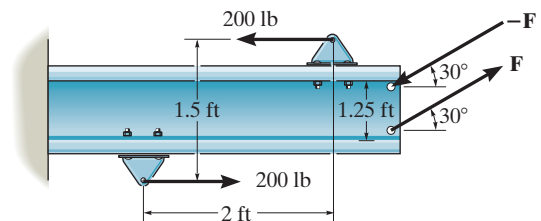
**4-69.** If the resultant couple of the three couples acting on the triangular block is to be zero, determine the magnitude of forces  $\mathbf{F}$  and  $\mathbf{P}$ .



Prob. 4-69

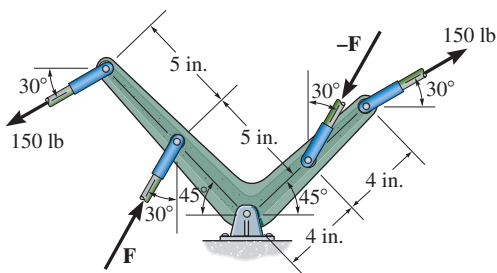
**4-70.** Two couples act on the beam. If  $F = 125 \text{ lb}$ , determine the resultant couple moment.

**4-71.** Two couples act on the beam. Determine the magnitude of  $\mathbf{F}$  so that the resultant couple moment is  $450 \text{ lb} \cdot \text{ft}$ , counterclockwise. Where on the beam does the resultant couple moment act?



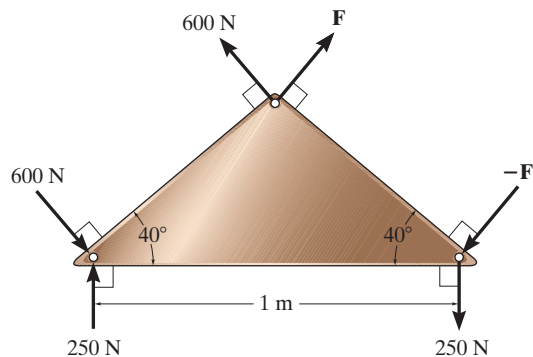
Probs. 4-70/71

**\*4-72.** Determine the magnitude of the couple forces  $\mathbf{F}$  so that the resultant couple moment on the crank is zero.



**Prob. 4-72**

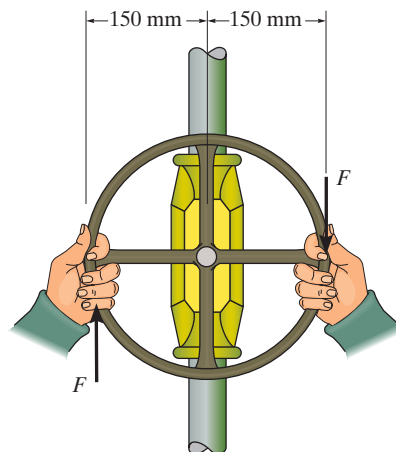
**4-73.** The ends of the triangular plate are subjected to three couples. Determine the magnitude of the force  $\mathbf{F}$  so that the resultant couple moment is  $400 \text{ N} \cdot \text{m}$  clockwise.



**Prob. 4-73**

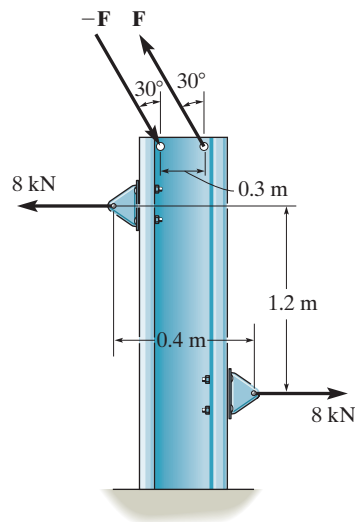
**4-74.** The man tries to open the valve by applying the couple forces of  $F = 75 \text{ N}$  to the wheel. Determine the couple moment produced.

**4-75.** If the valve can be opened with a couple moment of  $25 \text{ N} \cdot \text{m}$ , determine the required magnitude of each couple force which must be applied to the wheel.



**Probs. 4-74/75**

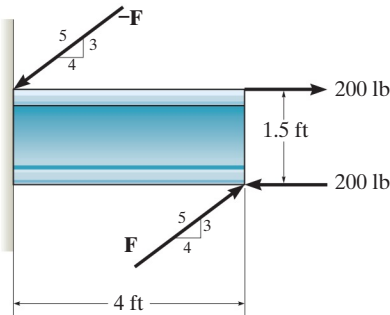
**\*4-76.** Determine the magnitude of  $\mathbf{F}$  so that the resultant couple moment is  $12 \text{ kN} \cdot \text{m}$ , counterclockwise. Where on the beam does the resultant couple moment act?



**Prob. 4-76**

**4-77.** Two couples act on the beam as shown. If  $F = 150$  lb, determine the resultant couple moment.

**4-78.** Two couples act on the beam as shown. Determine the magnitude of  $F$  so that the resultant couple moment is  $300 \text{ lb} \cdot \text{ft}$  counterclockwise. Where on the beam does the resultant couple act?

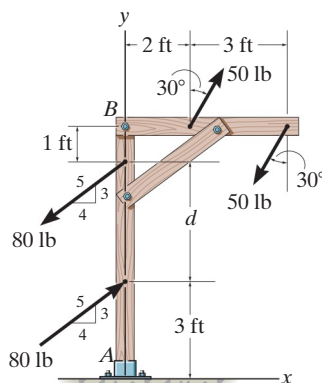


**Probs. 4-77/78**

**4-79.** Two couples act on the frame. If the resultant couple moment is to be zero, determine the distance  $d$  between the 80-lb couple forces.

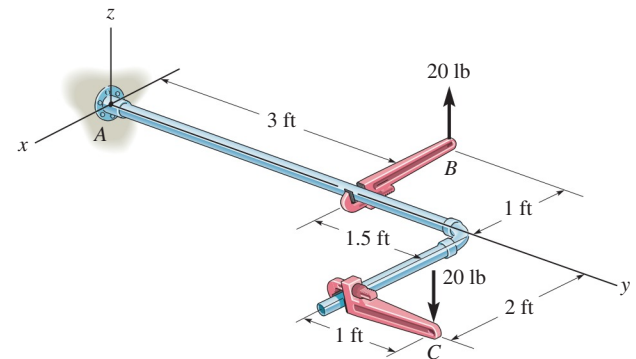
**\*4-80.** Two couples act on the frame. If  $d = 4$  ft, determine the resultant couple moment. Compute the result by resolving each force into  $x$  and  $y$  components and (a) finding the moment of each couple (Eq. 4-13) and (b) summing the moments of all the force components about point  $A$ .

**4-81.** Two couples act on the frame. If  $d = 4$  ft, determine the resultant couple moment. Compute the result by resolving each force into  $x$  and  $y$  components and (a) finding the moment of each couple (Eq. 4-13) and (b) summing the moments of all the force components about point  $B$ .



**Probs. 4-79/80/81**

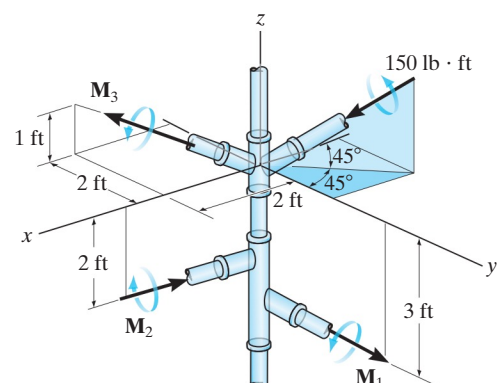
**4-82.** Express the moment of the couple acting on the pipe assembly in Cartesian vector form. What is the magnitude of the couple moment?



**Prob. 4-82**

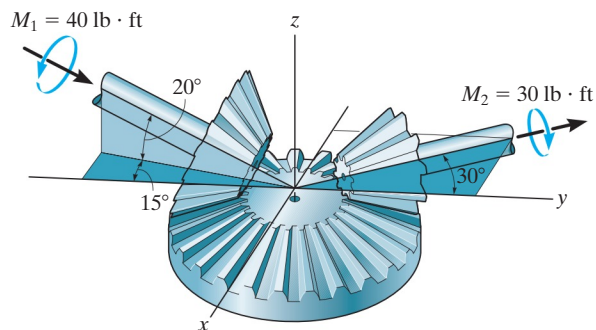
**4-83.** If  $M_1 = 180 \text{ lb} \cdot \text{ft}$ ,  $M_2 = 90 \text{ lb} \cdot \text{ft}$ , and  $M_3 = 120 \text{ lb} \cdot \text{ft}$ , determine the magnitude and coordinate direction angles of the resultant couple moment.

**\*4-84.** Determine the magnitudes of couple moments  $M_1$ ,  $M_2$ , and  $M_3$  so that the resultant couple moment is zero.



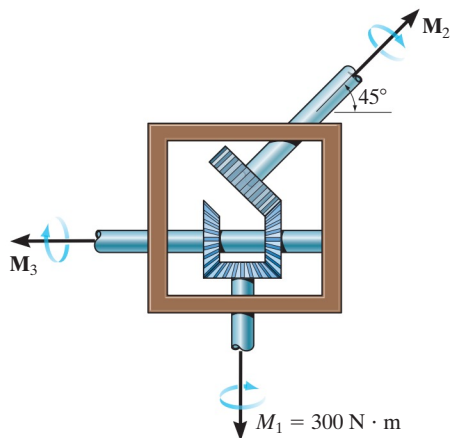
**Probs. 4-83/84**

**4-85.** The gears are subjected to the couple moments shown. Determine the magnitude and coordinate direction angles of the resultant couple moment.



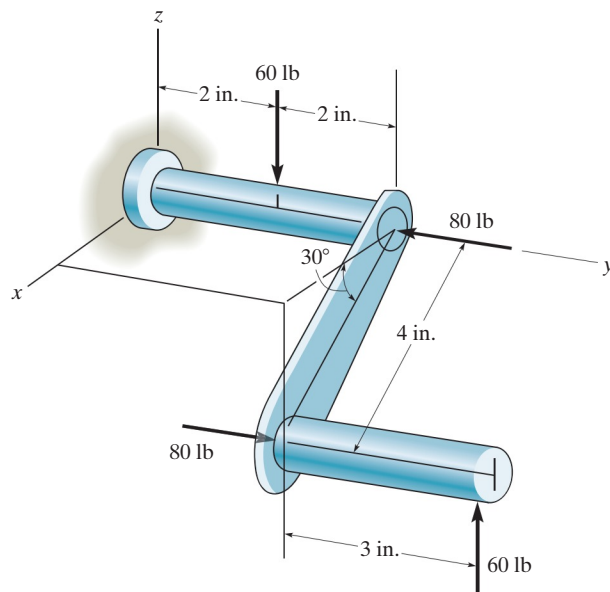
**Prob. 4-85**

**4-86.** Determine the required magnitude of the couple moments  $M_2$  and  $M_3$  so that the resultant couple moment is zero.



**Prob. 4-86**

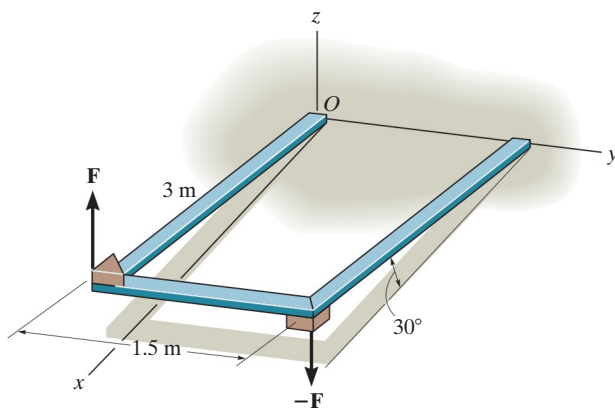
**4-87.** Determine the resultant couple moment of the two couples that act on the assembly. Specify its magnitude and coordinate direction angles.



**Prob. 4-87**

**\*4-88.** Express the moment of the couple acting on the frame in Cartesian vector form. The forces are applied perpendicular to the frame. What is the magnitude of the couple moment? Take  $F = 50$  N.

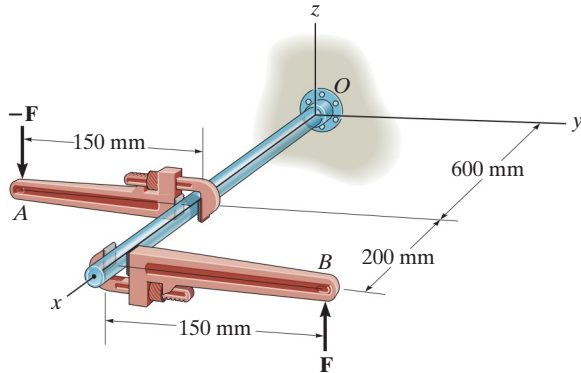
**4-89.** In order to turn over the frame, a couple moment is applied as shown. If the component of this couple moment along the  $x$  axis is  $M_x = \{-20\mathbf{i}\}$  N·m, determine the magnitude  $F$  of the couple forces.



**Probs. 4-88/89**

**4-90.** Express the moment of the couple acting on the pipe in Cartesian vector form. What is the magnitude of the couple moment? Take  $F = 125$  N.

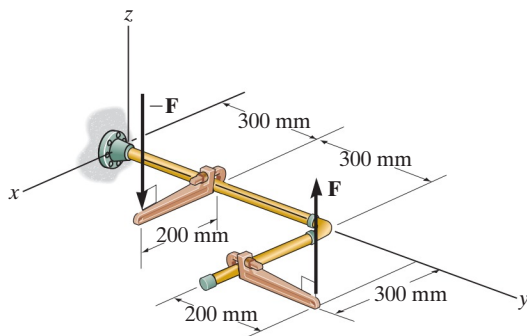
**4-91.** If the couple moment acting on the pipe has a magnitude of  $300 \text{ N} \cdot \text{m}$ , determine the magnitude  $F$  of the forces applied to the wrenches.



**Probs. 4-90/91**

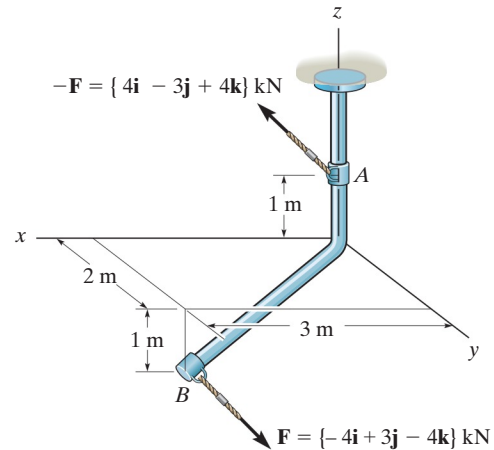
**\*4-92.** If  $F = 80$  N, determine the magnitude and coordinate direction angles of the couple moment. The pipe assembly lies in the  $x$ - $y$  plane.

**4-93.** If the magnitude of the couple moment acting on the pipe assembly is  $50 \text{ N} \cdot \text{m}$ , determine the magnitude of the couple forces applied to each wrench. The pipe assembly lies in the  $x$ - $y$  plane.



**Probs. 4-92/93**

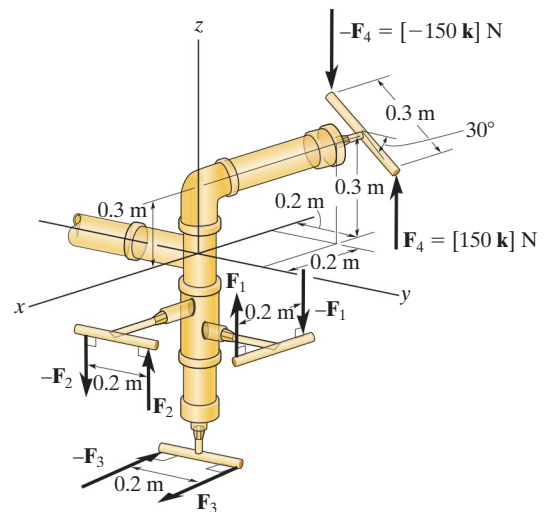
**4-94.** Express the moment of the couple acting on the rod in Cartesian vector form. What is the magnitude of the couple moment?



**Prob. 4-94**

**4-95.** If  $F_1 = 100$  N,  $F_2 = 120$  N, and  $F_3 = 80$  N, determine the magnitude and coordinate direction angles of the resultant couple moment.

**\*4-96.** Determine the required magnitude of  $F_1$ ,  $F_2$ , and  $F_3$  so that the resultant couple moment is  $(M_c)_R = [50\mathbf{i} - 45\mathbf{j} - 20\mathbf{k}] \text{ N} \cdot \text{m}$ .



**Probs. 4-95/96**



## 4.7 Simplification of a Force and Couple System

Sometimes it is convenient to reduce a system of forces and couple moments acting on a body to a simpler form by replacing it with an **equivalent system**, consisting of a single resultant force acting at a specific point and a resultant couple moment. A system is equivalent if the **external effects** it produces on a body are the same as those caused by the original force and couple moment system. In this context, the external effects of a system refer to the *translating and rotating motion* of the body if the body is free to move, or it refers to the *reactive forces* at the supports if the body is held fixed.

For example, consider holding the stick in Fig. 4–34*a*, which is subjected to the force  $\mathbf{F}$  at point  $A$ . If we attach a pair of equal but opposite forces  $\mathbf{F}$  and  $-\mathbf{F}$  at point  $B$ , which is *on the line of action* of  $\mathbf{F}$ , Fig. 4–34*b*, we observe that  $-\mathbf{F}$  at  $B$  and  $\mathbf{F}$  at  $A$  will cancel each other, leaving only  $\mathbf{F}$  at  $B$ , Fig. 4–34*c*. Force  $\mathbf{F}$  has now been moved from  $A$  to  $B$  without modifying its *external effects* on the stick; i.e., the reaction at the grip remains the same. This demonstrates the **principle of transmissibility**, which states that a force acting on a body (stick) is a **sliding vector** since it can be applied at any point along its line of action.

We can also use the above procedure to move a force to a point that is *not* on the line of action of the force. If  $\mathbf{F}$  is applied perpendicular to the stick, as in Fig. 4–35*a*, then we can attach a pair of equal but opposite forces  $\mathbf{F}$  and  $-\mathbf{F}$  to  $B$ , Fig. 4–35*b*. Force  $\mathbf{F}$  is now applied at  $B$ , and the other two forces,  $\mathbf{F}$  at  $A$  and  $-\mathbf{F}$  at  $B$ , form a couple that produces the couple moment  $M = Fd$ , Fig. 4–35*c*. Therefore, the force  $\mathbf{F}$  can be moved from  $A$  to  $B$  provided a couple moment  $\mathbf{M}$  is added to maintain an equivalent system. This couple moment is determined by taking the moment of  $\mathbf{F}$  about  $B$ . Since  $\mathbf{M}$  is actually a *free vector*, it can act at any point on the stick. In both cases the systems are equivalent, which causes a downward force  $\mathbf{F}$  and clockwise couple moment  $M = Fd$  to be felt at the grip.

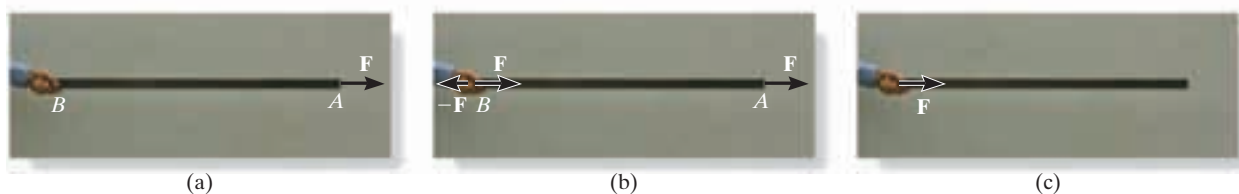


Fig. 4–34 (© Russell C. Hibbeler)

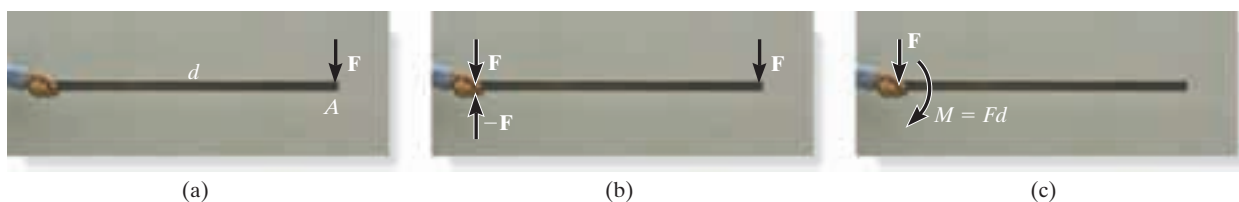


Fig. 4–35 (© Russell C. Hibbeler)

**System of Forces and Couple Moments.** Using the above method, a system of several forces and couple moments acting on a body can be reduced to an equivalent single resultant force acting at a point  $O$  and a resultant couple moment. For example, in Fig. 4-36a,  $O$  is not on the line of action of  $\mathbf{F}_1$ , and so this force can be moved to point  $O$  provided a couple moment  $(\mathbf{M}_O)_1 = \mathbf{r}_1 \times \mathbf{F}_1$  is added to the body. Similarly, the couple moment  $(\mathbf{M}_O)_2 = \mathbf{r}_2 \times \mathbf{F}_2$  should be added to the body when we move  $\mathbf{F}_2$  to point  $O$ . Finally, since the couple moment  $\mathbf{M}$  is a free vector, it can just be moved to point  $O$ . By doing this, we obtain the equivalent system shown in Fig. 4-36b, which produces the same external effects (support reactions) on the body as that of the force and couple system shown in Fig. 4-36a. If we sum the forces and couple moments, we obtain the resultant force  $\mathbf{F}_R = \mathbf{F}_1 + \mathbf{F}_2$  and the resultant couple moment  $(\mathbf{M}_R)_O = \mathbf{M} + (\mathbf{M}_O)_1 + (\mathbf{M}_O)_2$ , Fig. 4-36c.

Notice that  $\mathbf{F}_R$  is independent of the location of point  $O$  since it is simply a summation of the forces. However,  $(\mathbf{M}_R)_O$  depends upon this location since the moments  $\mathbf{M}_1$  and  $\mathbf{M}_2$  are determined using the position vectors  $\mathbf{r}_1$  and  $\mathbf{r}_2$ , which extend from  $O$  to each force. Also note that  $(\mathbf{M}_R)_O$  is a free vector and can act at *any point* on the body, although point  $O$  is generally chosen as its point of application.

We can generalize the above method of reducing a force and couple system to an equivalent resultant force  $\mathbf{F}_R$  acting at point  $O$  and a resultant couple moment  $(\mathbf{M}_R)_O$  by using the following two equations.

$$\begin{aligned} \mathbf{F}_R &= \Sigma \mathbf{F} \\ (\mathbf{M}_R)_O &= \Sigma \mathbf{M}_O + \Sigma \mathbf{M} \end{aligned} \quad (4-17)$$

The first equation states that the resultant force of the system is equivalent to the sum of all the forces; and the second equation states that the resultant couple moment of the system is equivalent to the sum of all the couple moments  $\Sigma \mathbf{M}$  plus the moments of all the forces  $\Sigma \mathbf{M}_O$  about point  $O$ . If the force system lies in the  $x$ - $y$  plane and any couple moments are perpendicular to this plane, then the above equations reduce to the following three scalar equations.

$$\begin{aligned} (F_R)_x &= \Sigma F_x \\ (F_R)_y &= \Sigma F_y \\ (M_R)_O &= \Sigma M_O + \Sigma M \end{aligned} \quad (4-18)$$

Here the resultant force is determined from the vector sum of its two components  $(F_R)_x$  and  $(F_R)_y$ .

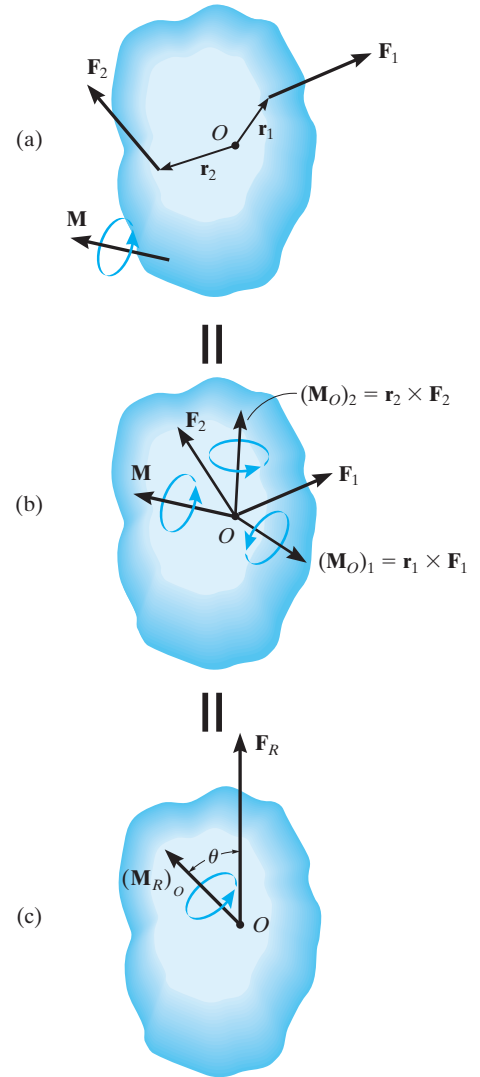
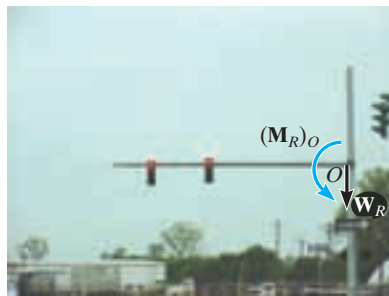
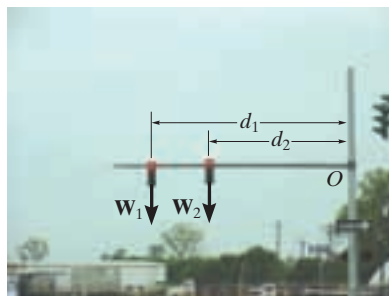


Fig. 4-36



The weights of these traffic lights can be replaced by their equivalent resultant force  $W_R = W_1 + W_2$  and a couple moment  $(M_R)_O = W_1 d_1 + W_2 d_2$  at the support,  $O$ . In both cases the support must provide the same resistance to translation and rotation in order to keep the member in the horizontal position. (© Russell C. Hibbeler)

## Important Points

- Force is a sliding vector, since it will create the same external effects on a body when it is applied at any point  $P$  along its line of action. This is called the principle of transmissibility.
- A couple moment is a free vector since it will create the same external effects on a body when it is applied at any point  $P$  on the body.
- When a force is moved to another point  $P$  that is not on its line of action, it will create the same external effects on the body if a couple moment is also applied to the body. The couple moment is determined by taking the moment of the force about point  $P$ .

## Procedure for Analysis

The following points should be kept in mind when simplifying a force and couple moment system to an equivalent resultant force and couple system.

- Establish the coordinate axes with the origin located at point  $O$  and the axes having a selected orientation.

### Force Summation.

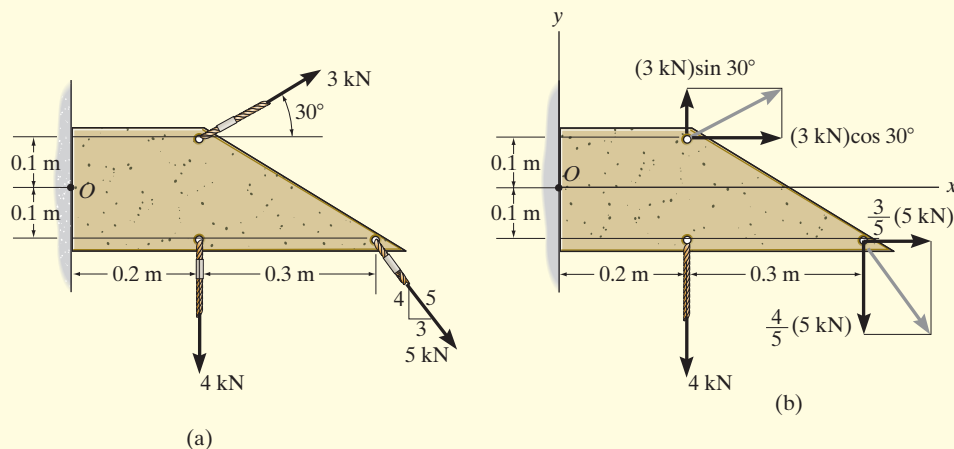
- If the force system is *coplanar*, resolve each force into its  $x$  and  $y$  components. If a component is directed along the positive  $x$  or  $y$  axis, it represents a positive scalar; whereas if it is directed along the negative  $x$  or  $y$  axis, it is a negative scalar.
- In three dimensions, represent each force as a Cartesian vector before summing the forces.

### Moment Summation.

- When determining the moments of a *coplanar* force system about point  $O$ , it is generally advantageous to use the principle of moments, i.e., determine the moments of the components of each force, rather than the moment of the force itself.
- In three dimensions use the vector cross product to determine the moment of each force about point  $O$ . Here the position vectors extend from  $O$  to any point on the line of action of each force.

**EXAMPLE 4.14**

Replace the force and couple system shown in Fig. 4–37a by an equivalent resultant force and couple moment acting at point  $O$ .

**SOLUTION**

**Force Summation.** The 3 kN and 5 kN forces are resolved into their  $x$  and  $y$  components as shown in Fig. 4–37b. We have

$$\rightarrow (F_R)_x = \Sigma F_x; \quad (F_R)_x = (3 \text{ kN}) \cos 30^\circ + \left(\frac{3}{5}\right)(5 \text{ kN}) = 5.598 \text{ kN} \rightarrow$$

$$+\uparrow (F_R)_y = \Sigma F_y; \quad (F_R)_y = (3 \text{ kN}) \sin 30^\circ - \left(\frac{4}{5}\right)(5 \text{ kN}) - 4 \text{ kN} = -6.50 \text{ kN} = 6.50 \text{ kN} \downarrow$$

Using the Pythagorean theorem, Fig. 4–37c, the magnitude of  $\mathbf{F}_R$  is

$$F_R = \sqrt{(F_R)_x^2 + (F_R)_y^2} = \sqrt{(5.598 \text{ kN})^2 + (6.50 \text{ kN})^2} = 8.58 \text{ kN} \quad \text{Ans.}$$

Its direction  $\theta$  is

$$\theta = \tan^{-1} \left( \frac{(F_R)_y}{(F_R)_x} \right) = \tan^{-1} \left( \frac{6.50 \text{ kN}}{5.598 \text{ kN}} \right) = 49.3^\circ \quad \text{Ans.}$$

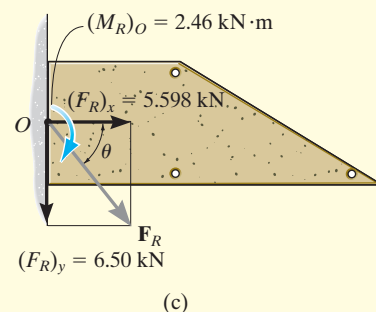
**Moment Summation.** The moments of 3 kN and 5 kN about point  $O$  will be determined using their  $x$  and  $y$  components. Referring to Fig. 4–37b, we have

$$\zeta + (M_R)_O = \Sigma M_O;$$

$$\begin{aligned} (M_R)_O &= (3 \text{ kN}) \sin 30^\circ (0.2 \text{ m}) - (3 \text{ kN}) \cos 30^\circ (0.1 \text{ m}) + \left(\frac{3}{5}\right)(5 \text{ kN}) (0.1 \text{ m}) \\ &\quad - \left(\frac{4}{5}\right)(5 \text{ kN}) (0.5 \text{ m}) - (4 \text{ kN}) (0.2 \text{ m}) \\ &= -2.46 \text{ kN} \cdot \text{m} = 2.46 \text{ kN} \cdot \text{m} \zeta \quad \text{Ans.} \end{aligned}$$

This clockwise moment is shown in Fig. 4–37c.

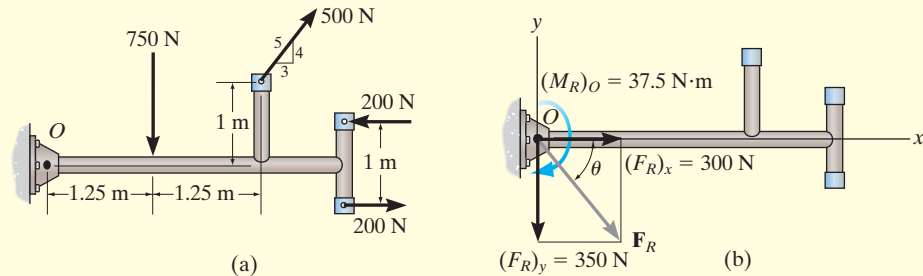
**NOTE:** Realize that the resultant force and couple moment in Fig. 4–37c will produce the same external effects or reactions at the supports as those produced by the force system, Fig. 4–37a.



**Fig. 4–37**

**EXAMPLE 4.15**

Replace the force and couple system acting on the member in Fig. 4-38a by an equivalent resultant force and couple moment acting at point  $O$ .

**Fig. 4-38****SOLUTION**

**Force Summation.** Since the couple forces of 200 N are equal but opposite, they produce a zero resultant force, and so it is not necessary to consider them in the force summation. The 500-N force is resolved into its  $x$  and  $y$  components, thus,

$$+\rightarrow (F_R)_x = \Sigma F_x; (F_R)_x = \left(\frac{3}{5}\right)(500 \text{ N}) = 300 \text{ N} \rightarrow$$

$$+\uparrow (F_R)_y = \Sigma F_y; (F_R)_y = (500 \text{ N})\left(\frac{4}{5}\right) - 750 \text{ N} = -350 \text{ N} = 350 \text{ N} \downarrow$$

From Fig. 4-15b, the magnitude of  $\mathbf{F}_R$  is

$$\begin{aligned} F_R &= \sqrt{(F_R)_x^2 + (F_R)_y^2} \\ &= \sqrt{(300 \text{ N})^2 + (350 \text{ N})^2} = 461 \text{ N} \end{aligned} \quad \text{Ans.}$$

And the angle  $\theta$  is

$$\theta = \tan^{-1}\left(\frac{(F_R)_y}{(F_R)_x}\right) = \tan^{-1}\left(\frac{350 \text{ N}}{300 \text{ N}}\right) = 49.4^\circ \quad \text{Ans.}$$

**Moment Summation.** Since the couple moment is a free vector, it can act at any point on the member. Referring to Fig. 4-38a, we have

$$\begin{aligned} \zeta + (M_R)_O &= \Sigma M_O + \Sigma M \\ (M_R)_O &= (500 \text{ N})\left(\frac{4}{5}\right)(2.5 \text{ m}) - (500 \text{ N})\left(\frac{3}{5}\right)(1 \text{ m}) \\ &\quad - (750 \text{ N})(1.25 \text{ m}) + 200 \text{ N}\cdot\text{m} \\ &= -37.5 \text{ N}\cdot\text{m} = 37.5 \text{ N}\cdot\text{m} \curvearrowright \end{aligned} \quad \text{Ans.}$$

This clockwise moment is shown in Fig. 4-38b.

**EXAMPLE 4.16**

The structural member is subjected to a couple moment  $\mathbf{M}$  and forces  $\mathbf{F}_1$  and  $\mathbf{F}_2$  in Fig. 4-39a. Replace this system by an equivalent resultant force and couple moment acting at its base, point  $O$ .

**SOLUTION (VECTOR ANALYSIS)**

The three-dimensional aspects of the problem can be simplified by using a Cartesian vector analysis. Expressing the forces and couple moment as Cartesian vectors, we have

$$\mathbf{F}_1 = \{-800\mathbf{k}\} \text{ N}$$

$$\mathbf{F}_2 = (300 \text{ N})\mathbf{u}_{CB}$$

$$= (300 \text{ N})\left(\frac{\mathbf{r}_{CB}}{r_{CB}}\right)$$

$$= 300 \text{ N} \left[ \frac{\{-0.15\mathbf{i} + 0.1\mathbf{j}\} \text{ m}}{\sqrt{(-0.15 \text{ m})^2 + (0.1 \text{ m})^2}} \right] = \{-249.6\mathbf{i} + 166.4\mathbf{j}\} \text{ N}$$

$$\mathbf{M} = -500\left(\frac{4}{5}\right)\mathbf{j} + 500\left(\frac{3}{5}\right)\mathbf{k} = \{-400\mathbf{j} + 300\mathbf{k}\} \text{ N} \cdot \text{m}$$

**Force Summation.**

$$\begin{aligned} \mathbf{F}_R &= \Sigma \mathbf{F}; & \mathbf{F}_R &= \mathbf{F}_1 + \mathbf{F}_2 = -800\mathbf{k} - 249.6\mathbf{i} + 166.4\mathbf{j} \\ & & &= \{-250\mathbf{i} + 166\mathbf{j} - 800\mathbf{k}\} \text{ N} \end{aligned}$$

**Moment Summation.**

$$(\mathbf{M}_R)_O = \Sigma \mathbf{M} + \Sigma \mathbf{M}_O$$

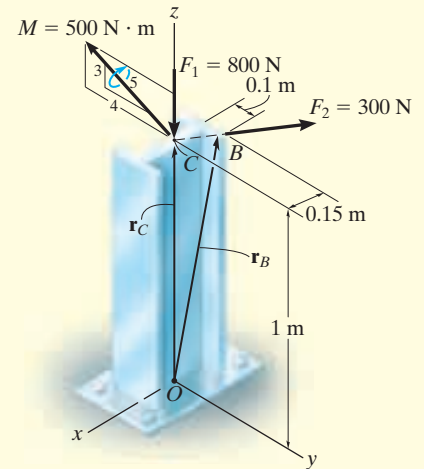
$$(\mathbf{M}_R)_O = \mathbf{M} + \mathbf{r}_C \times \mathbf{F}_1 + \mathbf{r}_B \times \mathbf{F}_2$$

$$(\mathbf{M}_R)_O = (-400\mathbf{j} + 300\mathbf{k}) + (1\mathbf{k}) \times (-800\mathbf{k}) + \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ -0.15 & 0.1 & 1 \\ -249.6 & 166.4 & 0 \end{vmatrix}$$

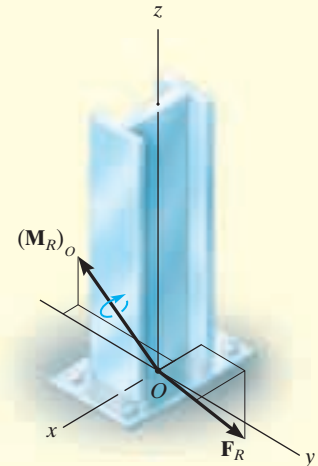
$$= (-400\mathbf{j} + 300\mathbf{k}) + (0) + (-166.4\mathbf{i} - 249.6\mathbf{j})$$

$$= \{-166\mathbf{i} - 650\mathbf{j} + 300\mathbf{k}\} \text{ N} \cdot \text{m}$$

The results are shown in Fig. 4-39b.



(a)

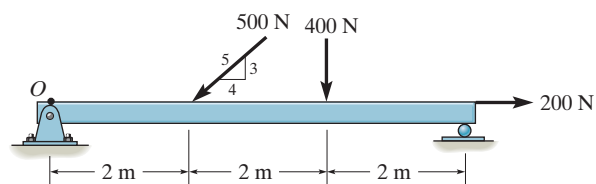


(b)

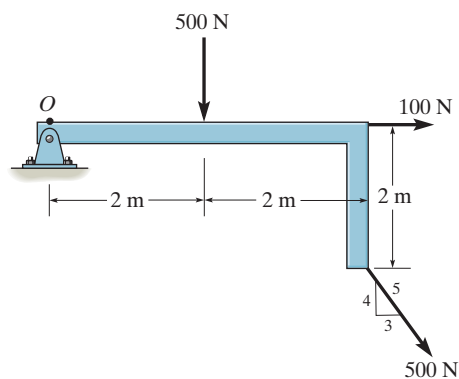
**Fig. 4-39***Ans.**Ans.*

## PRELIMINARY PROBLEM

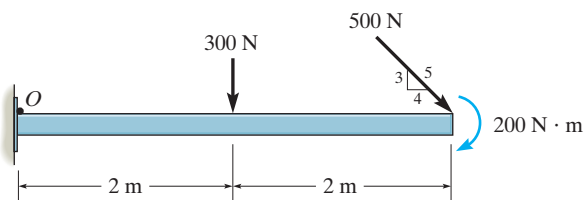
**P4-5.** In each case, determine the  $x$  and  $y$  components of the resultant force and the resultant couple moment at point  $O$ .



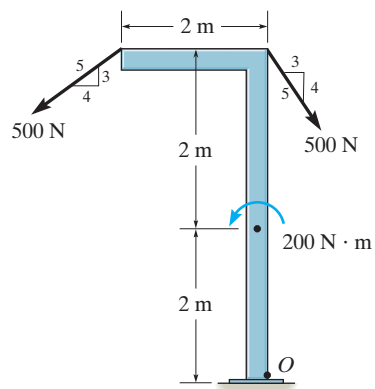
(a)



(c)



(b)

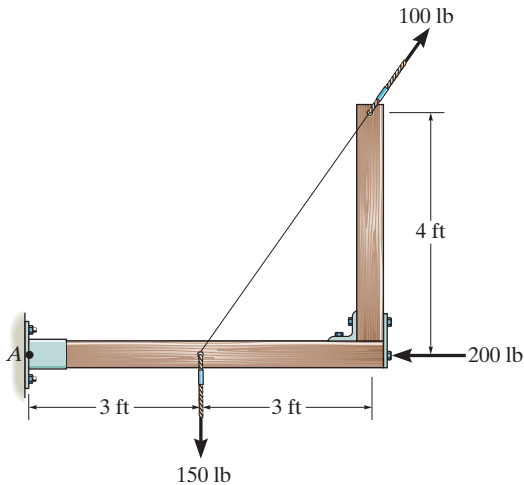


(d)

**Prob. P4-5**

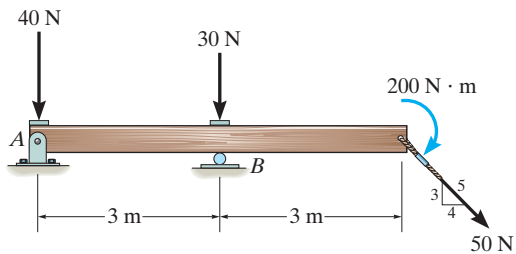
## FUNDAMENTAL PROBLEMS

**F4-25.** Replace the loading system by an equivalent resultant force and couple moment acting at point A.



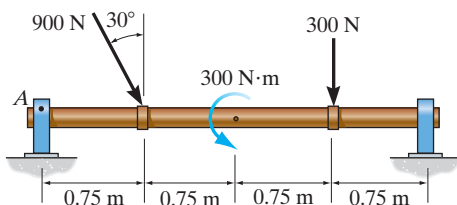
**Prob. F4-25**

**F4-26.** Replace the loading system by an equivalent resultant force and couple moment acting at point A.



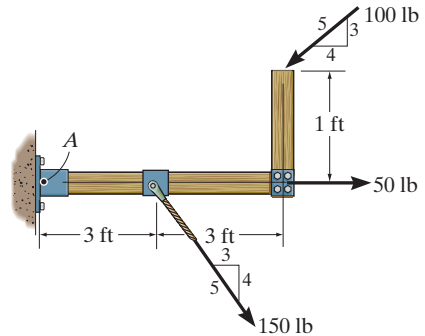
**Prob. F4-26**

**F4-27.** Replace the loading system by an equivalent resultant force and couple moment acting at point A.



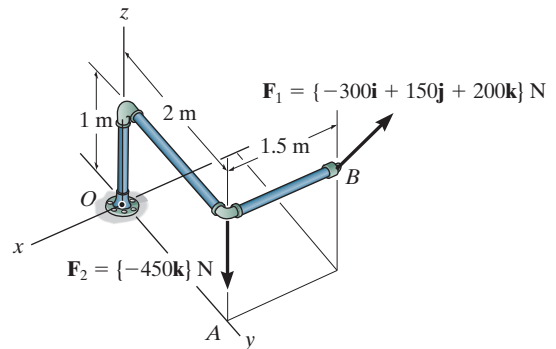
**Prob. F4-27**

**F4-28.** Replace the loading system by an equivalent resultant force and couple moment acting at point A.



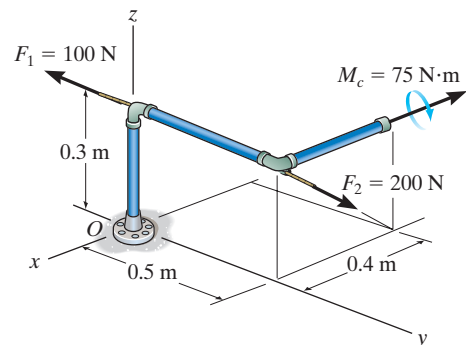
**Prob. F4-28**

**F4-29.** Replace the loading system by an equivalent resultant force and couple moment acting at point O.



**Prob. F4-29**

**F4-30.** Replace the loading system by an equivalent resultant force and couple moment acting at point O.



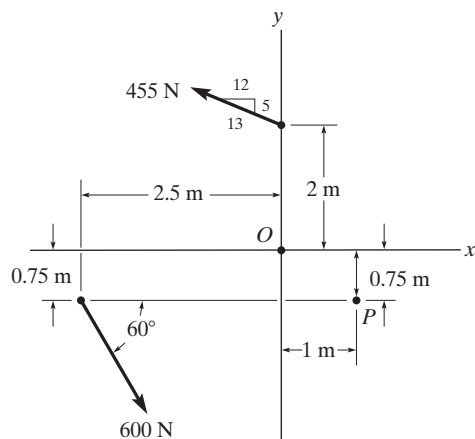
**Prob. F4-30**



## PROBLEMS

**4-97.** Replace the force system by an equivalent resultant force and couple moment at point  $O$ .

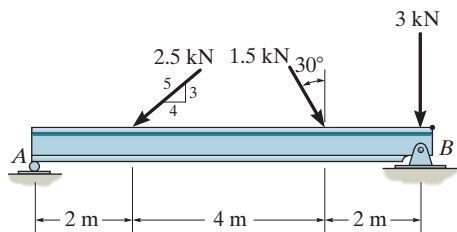
**4-98.** Replace the force system by an equivalent resultant force and couple moment at point  $P$ .



**Probs. 4-97/98**

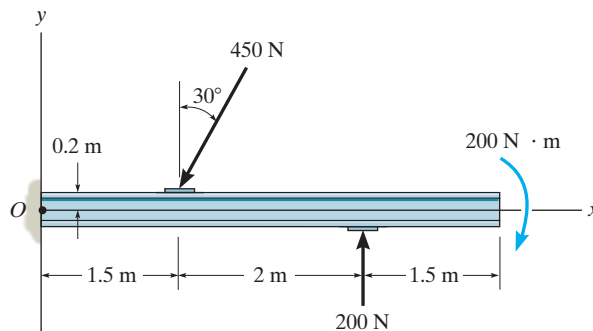
**4-99.** Replace the force system acting on the beam by an equivalent force and couple moment at point  $A$ .

**\*4-100.** Replace the force system acting on the beam by an equivalent force and couple moment at point  $B$ .



**Probs. 4-99/100**

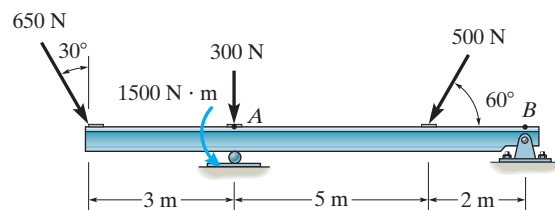
**4-101.** Replace the loading system acting on the beam by an equivalent resultant force and couple moment at point  $O$ .



**Prob. 4-101**

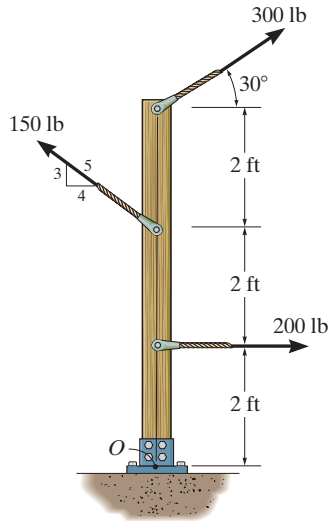
**4-102.** Replace the loading system acting on the post by an equivalent resultant force and couple moment at point  $A$ .

**4-103.** Replace the loading system acting on the post by an equivalent resultant force and couple moment at point  $B$ .



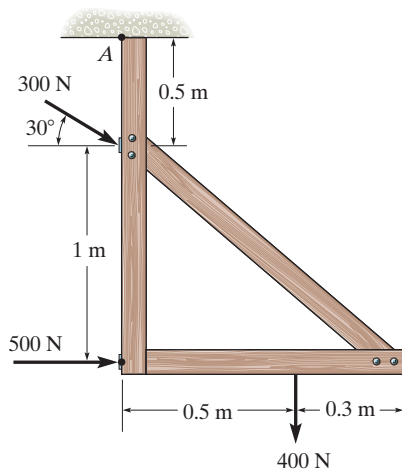
**Probs. 4-102/103**

**\*4-104.** Replace the force system acting on the post by a resultant force and couple moment at point  $O$ .



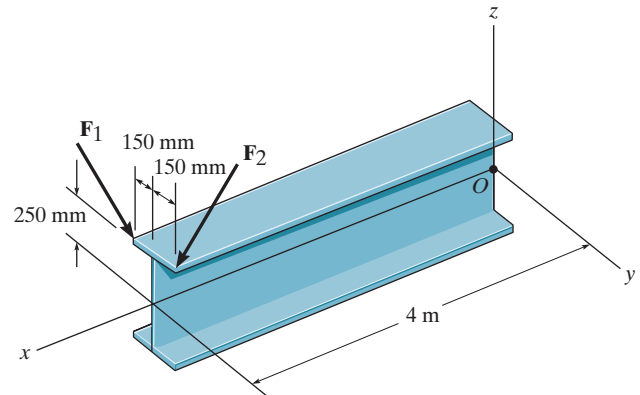
**Prob. 4-104**

**4-105.** Replace the force system acting on the frame by an equivalent resultant force and couple moment acting at point  $A$ .



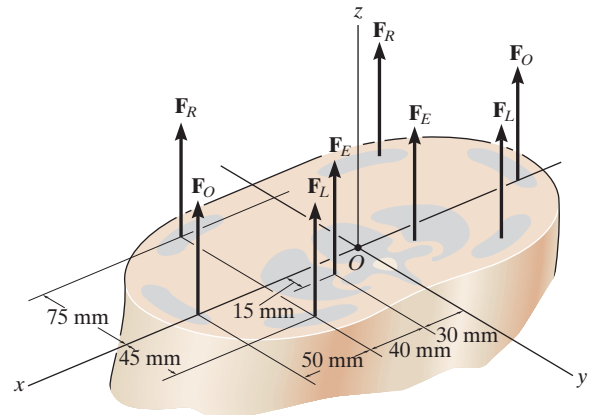
**Prob. 4-105**

**4-106.** The forces  $\mathbf{F}_1 = \{-4\mathbf{i} + 2\mathbf{j} - 3\mathbf{k}\}$  kN and  $\mathbf{F}_2 = \{3\mathbf{i} - 4\mathbf{j} - 2\mathbf{k}\}$  kN act on the end of the beam. Replace these forces by an equivalent force and couple moment acting at point  $O$ .



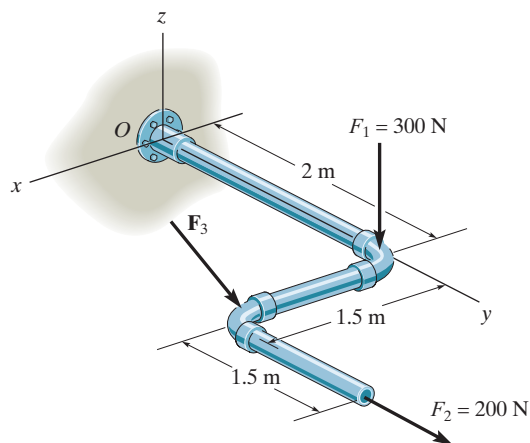
**Prob. 4-106**

**4-107.** A biomechanical model of the lumbar region of the human trunk is shown. The forces acting in the four muscle groups consist of  $F_R = 35$  N for the rectus,  $F_O = 45$  N for the oblique,  $F_L = 23$  N for the lumbar latissimus dorsi, and  $F_E = 32$  N for the erector spinae. These loadings are symmetric with respect to the  $y$ - $z$  plane. Replace this system of parallel forces by an equivalent force and couple moment acting at the spine, point  $O$ . Express the results in Cartesian vector form.



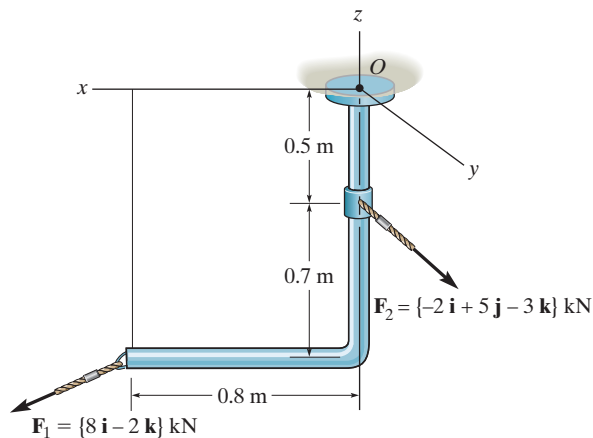
**Prob. 4-107**

**\*4-108.** Replace the force system by an equivalent resultant force and couple moment at point  $O$ . Take  $\mathbf{F}_3 = \{-200\mathbf{i} + 500\mathbf{j} - 300\mathbf{k}\}$  N.



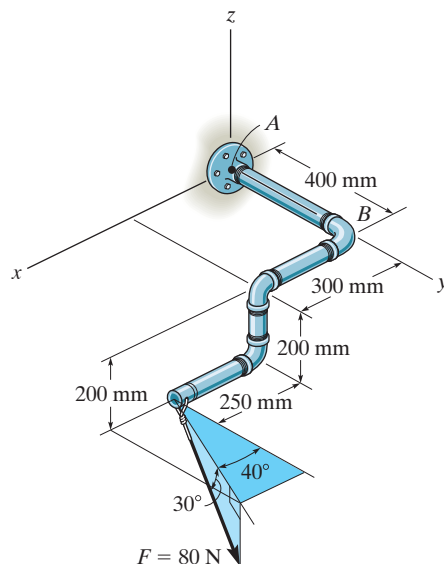
**Prob. 4-108**

**4-109.** Replace the loading by an equivalent resultant force and couple moment at point  $O$ .



**Prob. 4-109**

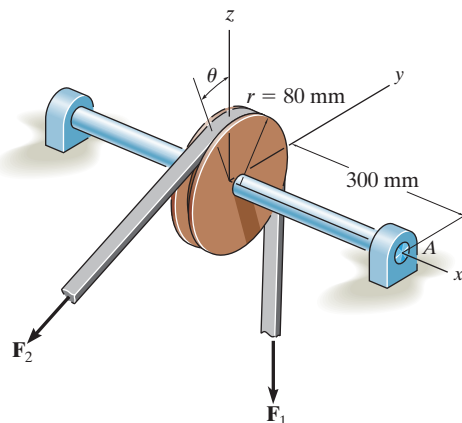
**4-110.** Replace the force of  $F = 80$  N acting on the pipe assembly by an equivalent resultant force and couple moment at point  $A$ .



**Prob. 4-110**

**4-111.** The belt passing over the pulley is subjected to forces  $\mathbf{F}_1$  and  $\mathbf{F}_2$ , each having a magnitude of 40 N.  $\mathbf{F}_1$  acts in the  $-\mathbf{k}$  direction. Replace these forces by an equivalent force and couple moment at point  $A$ . Express the result in Cartesian vector form. Set  $\theta = 0^\circ$  so that  $\mathbf{F}_2$  acts in the  $-\mathbf{j}$  direction.

**\*4-112.** The belt passing over the pulley is subjected to two forces  $\mathbf{F}_1$  and  $\mathbf{F}_2$ , each having a magnitude of 40 N.  $\mathbf{F}_1$  acts in the  $-\mathbf{k}$  direction. Replace these forces by an equivalent force and couple moment at point  $A$ . Express the result in Cartesian vector form. Take  $\theta = 45^\circ$ .



**Probs. 4-111/112**

## 4.8 Further Simplification of a Force and Couple System

In the preceding section, we developed a way to reduce a force and couple moment system acting on a rigid body into an equivalent resultant force  $\mathbf{F}_R$  acting at a specific point  $O$  and a resultant couple moment  $(\mathbf{M}_R)_O$ . The force system can be further reduced to an equivalent single resultant force provided the lines of action of  $\mathbf{F}_R$  and  $(\mathbf{M}_R)_O$  are *perpendicular* to each other. Because of this condition, concurrent, coplanar, and parallel force systems can be further simplified.

**Concurrent Force System.** Since a *concurrent force system* is one in which the lines of action of all the forces intersect at a common point  $O$ , Fig. 4-40a, then the force system produces no moment about this point. As a result, the equivalent system can be represented by a single resultant force  $\mathbf{F}_R = \Sigma \mathbf{F}$  acting at  $O$ , Fig. 4-40b.

**Coplanar Force System.** In the case of a *coplanar force system*, the lines of action of all the forces lie in the same plane, Fig. 4-41a, and so the resultant force  $\mathbf{F}_R = \Sigma \mathbf{F}$  of this system also lies in this plane. Furthermore, the moment of each of the forces about any point  $O$  is directed perpendicular to this plane. Thus, the resultant moment  $(\mathbf{M}_R)_O$  and resultant force  $\mathbf{F}_R$  will be *mutually perpendicular*, Fig. 4-41b. The resultant moment can be replaced by moving the resultant force  $\mathbf{F}_R$  a perpendicular or moment arm distance  $d$  away from point  $O$  such that  $\mathbf{F}_R$  produces the *same moment*  $(\mathbf{M}_R)_O$  about point  $O$ , Fig. 4-41c. This distance  $d$  can be determined from the scalar equation  $(M_R)_O = F_R d = \Sigma M_O$  or  $d = (M_R)_O / F_R$ .

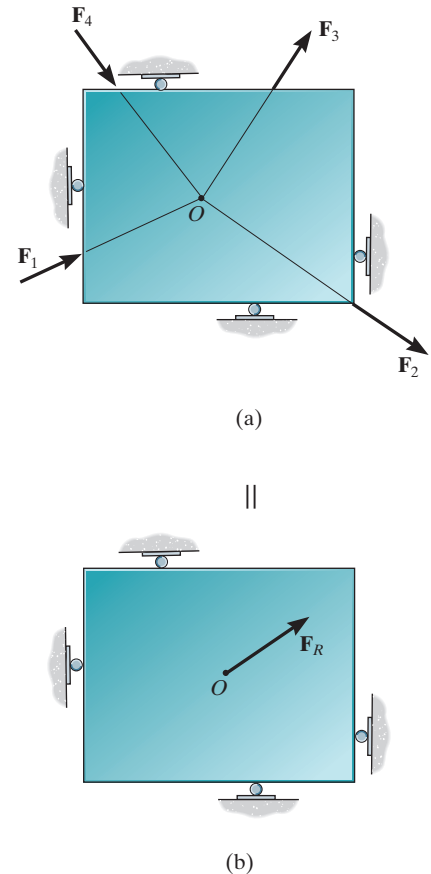


Fig. 4-40

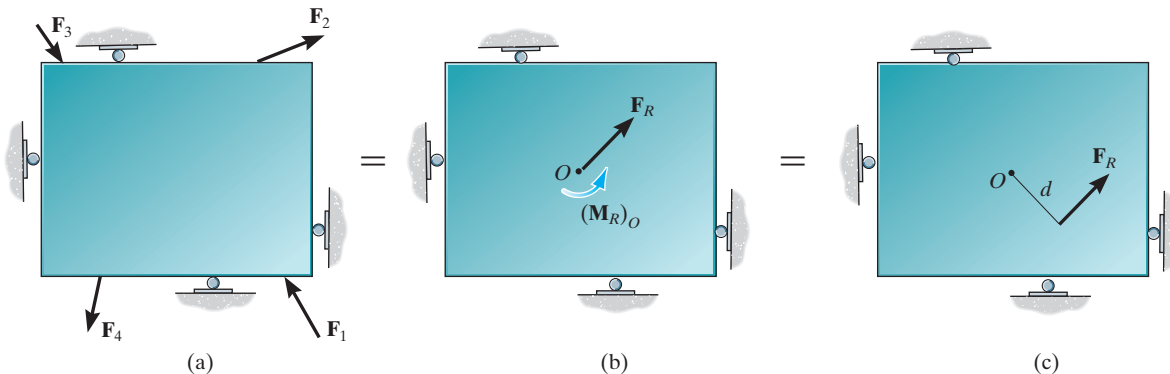


Fig. 4-41

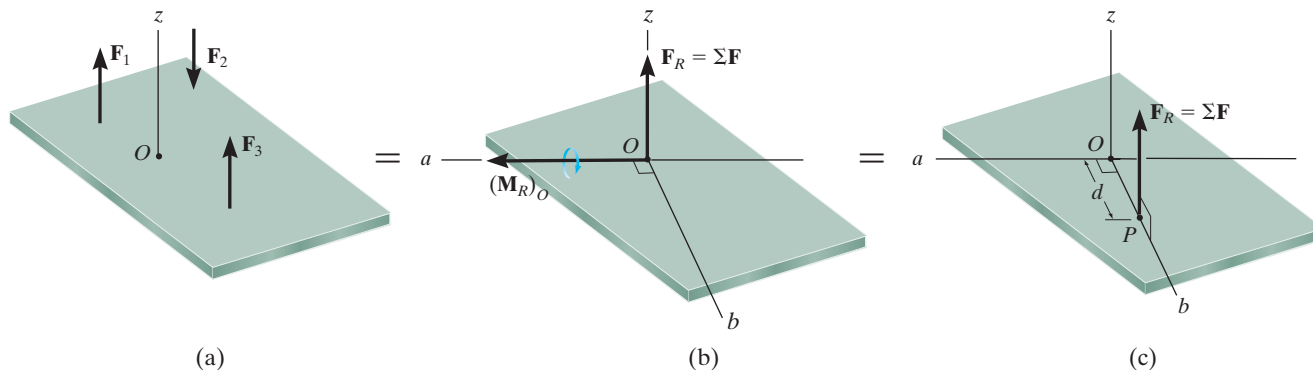


Fig. 4-42

**Parallel Force System.** The *parallel force system* shown in Fig. 4-42a consists of forces that are all parallel to the  $z$  axis. Thus, the resultant force  $\mathbf{F}_R = \Sigma \mathbf{F}$  at point  $O$  must also be parallel to this axis, Fig. 4-42b. The moment produced by each force lies in the plane of the plate, and so the resultant couple moment,  $(\mathbf{M}_R)_O$ , will also lie in this plane, along the moment axis  $a$  since  $\mathbf{F}_R$  and  $(\mathbf{M}_R)_O$  are mutually perpendicular. As a result, the force system can be further reduced to an equivalent single resultant force  $\mathbf{F}_R$ , acting through point  $P$  located on the perpendicular  $b$  axis, Fig. 4-42c. The distance  $d$  along this axis from point  $O$  requires  $(M_R)_O = F_R d = \Sigma M_O$  or  $d = \Sigma M_O / F_R$ .

## Procedure for Analysis

The technique used to reduce a coplanar or parallel force system to a single resultant force follows a similar procedure outlined in the previous section.

- Establish the  $x, y, z$ , axes and locate the resultant force  $\mathbf{F}_R$  an arbitrary distance away from the origin of the coordinates.

### Force Summation.

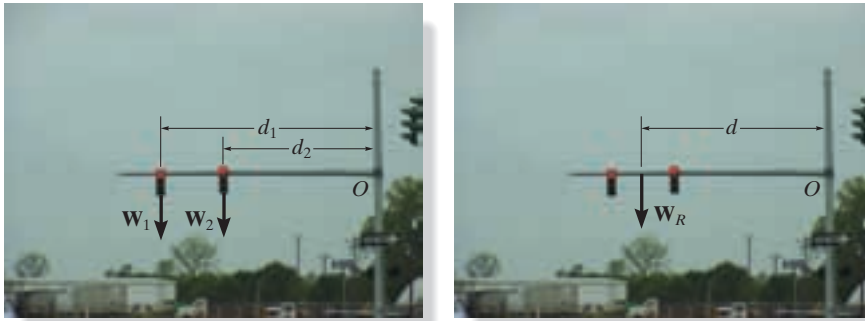
- The resultant force is equal to the sum of all the forces in the system.
- For a coplanar force system, resolve each force into its  $x$  and  $y$  components. Positive components are directed along the positive  $x$  and  $y$  axes, and negative components are directed along the negative  $x$  and  $y$  axes.

### Moment Summation.

- The moment of the resultant force about point  $O$  is equal to the sum of all the couple moments in the system plus the moments of all the forces in the system about  $O$ .
- This moment condition is used to find the location of the resultant force from point  $O$ .



The four cable forces are all concurrent at point  $O$  on this bridge tower. Consequently they produce no resultant moment there, only a resultant force  $\mathbf{F}_R$ . Note that the designers have positioned the cables so that  $\mathbf{F}_R$  is directed *along* the bridge tower directly to the support, so that it does not cause any bending of the tower. (© Russell C. Hibbeler)



Here the weights of the traffic lights are replaced by their resultant force  $W_R = W_1 + W_2$  which acts at a distance  $d = (W_1 d_1 + W_2 d_2) / W_R$  from  $O$ . Both systems are equivalent. (© Russell C. Hibbeler)

**Reduction to a Wrench.** In general, a three-dimensional force and couple moment system will have an equivalent resultant force  $\mathbf{F}_R$  acting at point  $O$  and a resultant couple moment  $(\mathbf{M}_R)_O$  that are *not perpendicular* to one another, as shown in Fig. 4-43a. Although a force system such as this cannot be further reduced to an equivalent single resultant force, the resultant couple moment  $(\mathbf{M}_R)_O$  can be resolved into components parallel and perpendicular to the line of action of  $\mathbf{F}_R$ , Fig. 4-43a. If this appears difficult to do in three dimensions, use the dot product to get  $\mathbf{M}_{\parallel} = (\mathbf{M}_R)_O \cdot \mathbf{u}_{F_R}$  and then  $\mathbf{M}_{\perp} = \mathbf{M}_R - \mathbf{M}_{\parallel}$ . The perpendicular component  $\mathbf{M}_{\perp}$  can be replaced if we move  $\mathbf{F}_R$  to point  $P$ , a distance  $d$  from point  $O$  along the  $b$  axis, Fig. 4-43b. As shown, this axis is perpendicular to both the  $a$  axis and the line of action of  $\mathbf{F}_R$ . The location of  $P$  can be determined from  $d = M_{\perp} / F_R$ . Finally, because  $\mathbf{M}_{\parallel}$  is a free vector, it can be moved to point  $P$ , Fig. 4-43c. This combination of a resultant force  $\mathbf{F}_R$  and collinear couple moment  $\mathbf{M}_{\parallel}$  will tend to translate and rotate the body about its axis and is referred to as a **wrench** or **screw**. A wrench is the simplest system that can represent any general force and couple moment system acting on a body.

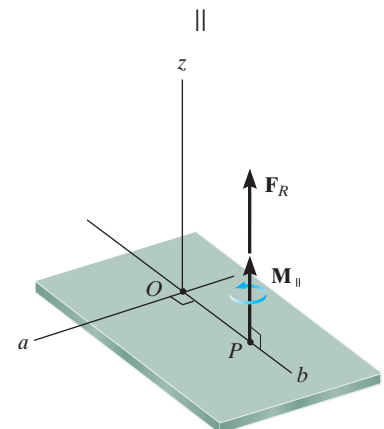
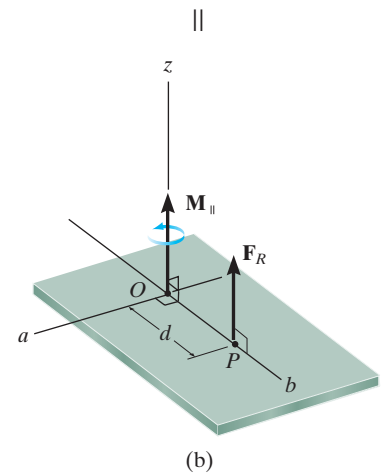
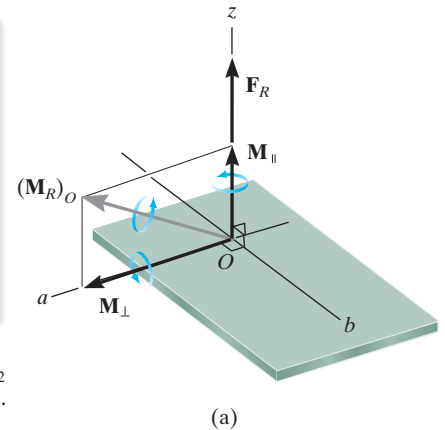


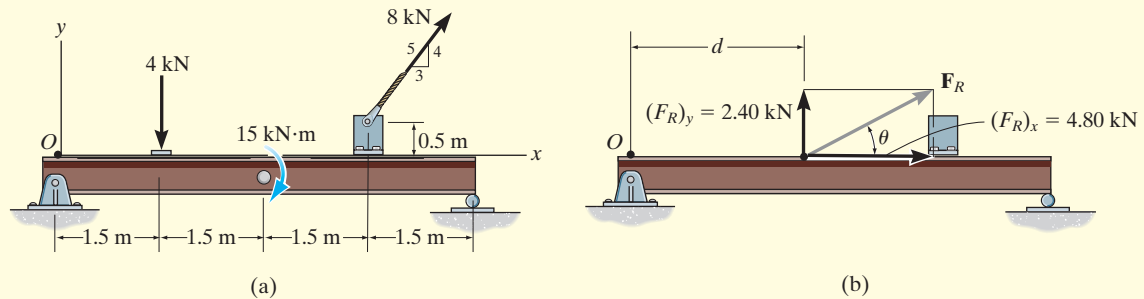
Fig. 4-43

### Important Point

- A concurrent, coplanar, or parallel force system can always be reduced to a single resultant force acting at a specific point  $P$ . For any other type of force system, the simplest reduction is a wrench, which consists of resultant force and collinear couple moment acting at a specific point  $P$ .

**EXAMPLE 4.17**

Replace the force and couple moment system acting on the beam in Fig. 4-44a by an equivalent resultant force, and find where its line of action intersects the beam, measured from point  $O$ .

**Fig. 4-44****SOLUTION**

**Force Summation.** Summing the force components,

$$\begin{aligned} \rightarrow (F_R)_x &= \Sigma F_x; & (F_R)_x &= 8 \text{ kN} \left( \frac{3}{5} \right) = 4.80 \text{ kN} \rightarrow \\ + \uparrow (F_R)_y &= \Sigma F_y; & (F_R)_y &= -4 \text{ kN} + 8 \text{ kN} \left( \frac{4}{5} \right) = 2.40 \text{ kN} \uparrow \end{aligned}$$

From Fig. 4-44b, the magnitude of  $\mathbf{F}_R$  is

$$F_R = \sqrt{(4.80 \text{ kN})^2 + (2.40 \text{ kN})^2} = 5.37 \text{ kN} \quad \text{Ans.}$$

The angle  $\theta$  is

$$\theta = \tan^{-1} \left( \frac{2.40 \text{ kN}}{4.80 \text{ kN}} \right) = 26.6^\circ \quad \text{Ans.}$$

**Moment Summation.** We must equate the moment of  $\mathbf{F}_R$  about point  $O$  in Fig. 4-44b to the sum of the moments of the force and couple moment system about point  $O$  in Fig. 4-44a. Since the line of action of  $(\mathbf{F}_R)_x$  acts through point  $O$ , only  $(\mathbf{F}_R)_y$  produces a moment about this point. Thus,

$$\begin{aligned} \zeta + (M_R)_O &= \Sigma M_O; & 2.40 \text{ kN}(d) &= -(4 \text{ kN})(1.5 \text{ m}) - 15 \text{ kN} \cdot \text{m} \\ & & &- \left[ 8 \text{ kN} \left( \frac{3}{5} \right) \right] (0.5 \text{ m}) + \left[ 8 \text{ kN} \left( \frac{4}{5} \right) \right] (4.5 \text{ m}) \end{aligned}$$

$$d = 2.25 \text{ m} \quad \text{Ans.}$$

**EXAMPLE 4.18**

The jib crane shown in Fig. 4-45a is subjected to three coplanar forces. Replace this loading by an equivalent resultant force and specify where the resultant's line of action intersects the column  $AB$  and boom  $BC$ .

**SOLUTION**

**Force Summation.** Resolving the 250-lb force into  $x$  and  $y$  components and summing the force components yields

$$+\rightarrow (F_R)_x = \Sigma F_x; \quad (F_R)_x = -250 \text{ lb} \left( \frac{3}{5} \right) - 175 \text{ lb} = -325 \text{ lb} = 325 \text{ lb} \leftarrow$$

$$+\uparrow (F_R)_y = \Sigma F_y; \quad (F_R)_y = -250 \text{ lb} \left( \frac{4}{5} \right) - 60 \text{ lb} = -260 \text{ lb} = 260 \text{ lb} \downarrow$$

As shown by the vector addition in Fig. 4-45b,

$$F_R = \sqrt{(325 \text{ lb})^2 + (260 \text{ lb})^2} = 416 \text{ lb} \quad \text{Ans.}$$

$$\theta = \tan^{-1} \left( \frac{260 \text{ lb}}{325 \text{ lb}} \right) = 38.7^\circ \nearrow \quad \text{Ans.}$$

**Moment Summation.** Moments will be summed about point  $A$ . Assuming the line of action of  $\mathbf{F}_R$  intersects  $AB$  at a distance  $y$  from  $A$ , Fig. 4-45b, we have

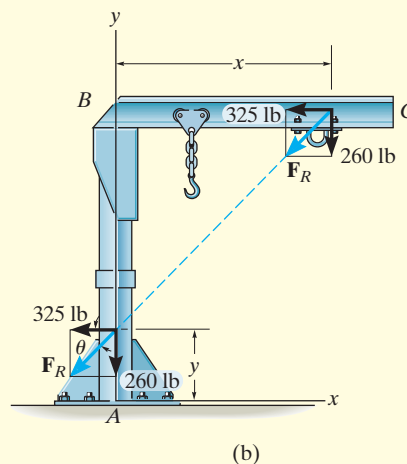
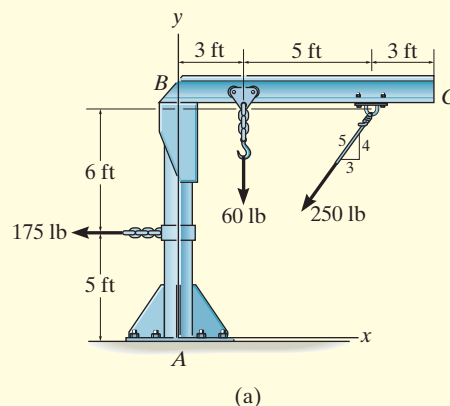
$$\begin{aligned} \zeta + (M_R)_A &= \Sigma M_A; & 325 \text{ lb} (y) + 260 \text{ lb} (0) \\ &= 175 \text{ lb} (5 \text{ ft}) - 60 \text{ lb} (3 \text{ ft}) + 250 \text{ lb} \left( \frac{3}{5} \right) (11 \text{ ft}) - 250 \text{ lb} \left( \frac{4}{5} \right) (8 \text{ ft}) \end{aligned}$$

$$y = 2.29 \text{ ft} \quad \text{Ans.}$$

By the principle of transmissibility,  $\mathbf{F}_R$  can be placed at a distance  $x$  where it intersects  $BC$ , Fig. 4-45b. In this case we have

$$\begin{aligned} \zeta + (M_R)_A &= \Sigma M_A; & 325 \text{ lb} (11 \text{ ft}) - 260 \text{ lb} (x) \\ &= 175 \text{ lb} (5 \text{ ft}) - 60 \text{ lb} (3 \text{ ft}) + 250 \text{ lb} \left( \frac{3}{5} \right) (11 \text{ ft}) - 250 \text{ lb} \left( \frac{4}{5} \right) (8 \text{ ft}) \end{aligned}$$

$$x = 10.9 \text{ ft} \quad \text{Ans.}$$

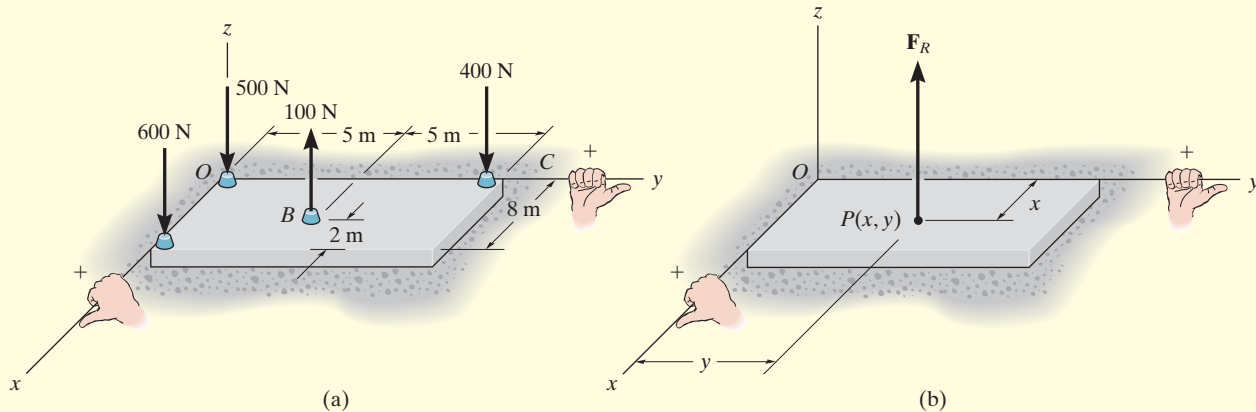


**Fig. 4-45**



**EXAMPLE 4.19**

The slab in Fig. 4-46a is subjected to four parallel forces. Determine the magnitude and direction of a resultant force equivalent to the given force system, and locate its point of application on the slab.

**Fig. 4-46****SOLUTION (SCALAR ANALYSIS)**

**Force Summation.** From Fig. 4-46a, the resultant force is

$$\begin{aligned}
 +\uparrow F_R &= \Sigma F; & F_R &= -600 \text{ N} + 100 \text{ N} - 400 \text{ N} - 500 \text{ N} \\
 & & &= -1400 \text{ N} = 1400 \text{ N} \downarrow
 \end{aligned}$$

*Ans.*

**Moment Summation.** We require the moment about the  $x$  axis of the resultant force, Fig. 4-46b, to be equal to the sum of the moments about the  $x$  axis of all the forces in the system, Fig. 4-46a. The moment arms are determined from the  $y$  coordinates, since these coordinates represent the *perpendicular distances* from the  $x$  axis to the lines of action of the forces. Using the right-hand rule, we have

$$\begin{aligned}
 (M_R)_x &= \Sigma M_x; \\
 -(1400 \text{ N})y &= 600 \text{ N}(0) + 100 \text{ N}(5 \text{ m}) - 400 \text{ N}(10 \text{ m}) + 500 \text{ N}(0) \\
 -1400y &= -3500 & y &= 2.50 \text{ m}
 \end{aligned}$$

*Ans.*

In a similar manner, a moment equation can be written about the  $y$  axis using moment arms defined by the  $x$  coordinates of each force.

$$\begin{aligned}
 (M_R)_y &= \Sigma M_y; \\
 (1400 \text{ N})x &= 600 \text{ N}(8 \text{ m}) - 100 \text{ N}(6 \text{ m}) + 400 \text{ N}(0) + 500 \text{ N}(0) \\
 1400x &= 4200 \\
 x &= 3 \text{ m}
 \end{aligned}$$

*Ans.*

**NOTE:** A force of  $F_R = 1400 \text{ N}$  placed at point  $P(3.00 \text{ m}, 2.50 \text{ m})$  on the slab, Fig. 4-46b, is therefore equivalent to the parallel force system acting on the slab in Fig. 4-46a.

**EXAMPLE 4.20**

Replace the force system in Fig. 4-47a by an equivalent resultant force and specify its point of application on the pedestal.

**SOLUTION**

**Force Summation.** Here we will demonstrate a vector analysis. Summing forces,

$$\begin{aligned}\mathbf{F}_R &= \Sigma \mathbf{F}; \mathbf{F}_R = \mathbf{F}_A + \mathbf{F}_B + \mathbf{F}_C \\ &= \{-300\mathbf{k}\} \text{ lb} + \{-500\mathbf{k}\} \text{ lb} + \{100\mathbf{k}\} \text{ lb} \\ &= \{-700\mathbf{k}\} \text{ lb}\end{aligned}$$

*Ans.*

**Location.** Moments will be summed about point  $O$ . The resultant force  $\mathbf{F}_R$  is assumed to act through point  $P(x, y, 0)$ , Fig. 4-47b. Thus

$$(\mathbf{M}_R)_O = \Sigma \mathbf{M}_O;$$

$$\mathbf{r}_P \times \mathbf{F}_R = (\mathbf{r}_A \times \mathbf{F}_A) + (\mathbf{r}_B \times \mathbf{F}_B) + (\mathbf{r}_C \times \mathbf{F}_C)$$

$$(x\mathbf{i} + y\mathbf{j}) \times (-700\mathbf{k}) = [(4\mathbf{i}) \times (-300\mathbf{k})]$$

$$+ [(-4\mathbf{i} + 2\mathbf{j}) \times (-500\mathbf{k})] + [(-4\mathbf{j}) \times (100\mathbf{k})]$$

$$-700x(\mathbf{i} \times \mathbf{k}) - 700y(\mathbf{j} \times \mathbf{k}) = -1200(\mathbf{i} \times \mathbf{k}) + 2000(\mathbf{i} \times \mathbf{k})$$

$$- 1000(\mathbf{j} \times \mathbf{k}) - 400(\mathbf{j} \times \mathbf{k})$$

$$700x\mathbf{j} - 700y\mathbf{i} = 1200\mathbf{j} - 2000\mathbf{j} - 1000\mathbf{i} - 400\mathbf{i}$$

Equating the  $\mathbf{i}$  and  $\mathbf{j}$  components,

$$-700y = -1400 \quad (1)$$

$$y = 2 \text{ in.} \quad \text{Ans.}$$

$$700x = -800 \quad (2)$$

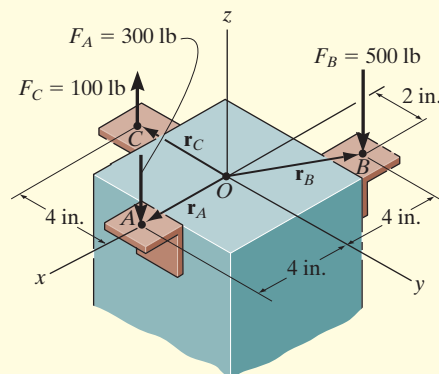
$$x = -1.14 \text{ in.} \quad \text{Ans.}$$

The negative sign indicates that the  $x$  coordinate of point  $P$  is negative.

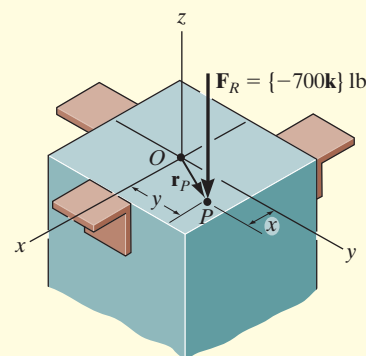
**NOTE:** It is also possible to establish Eq. 1 and 2 directly by summing moments about the  $x$  and  $y$  axes. Using the right-hand rule, we have

$$(\mathbf{M}_R)_x = \Sigma M_x; \quad -700y = -100 \text{ lb}(4 \text{ in.}) - 500 \text{ lb}(2 \text{ in.})$$

$$(\mathbf{M}_R)_y = \Sigma M_y; \quad 700x = 300 \text{ lb}(4 \text{ in.}) - 500 \text{ lb}(4 \text{ in.})$$



(a)

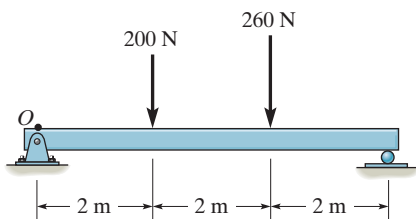


(b)

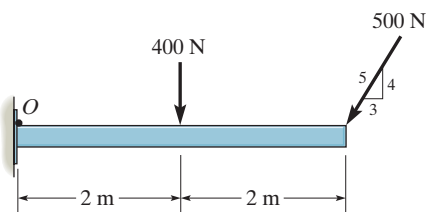
**Fig. 4-47**

## PRELIMINARY PROBLEMS

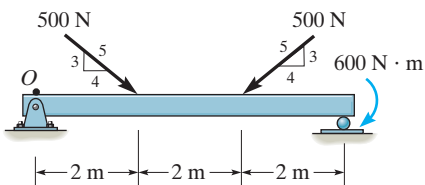
**P4-6.** In each case, determine the  $x$  and  $y$  components of the resultant force and specify the distance where this force acts from point  $O$ .



(a)



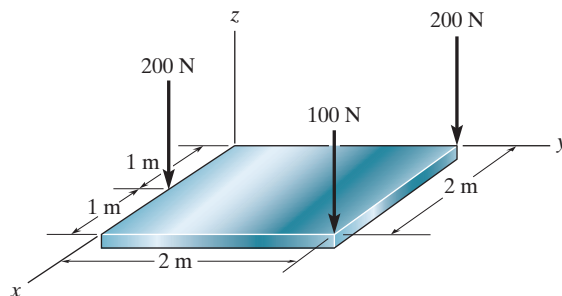
(b)



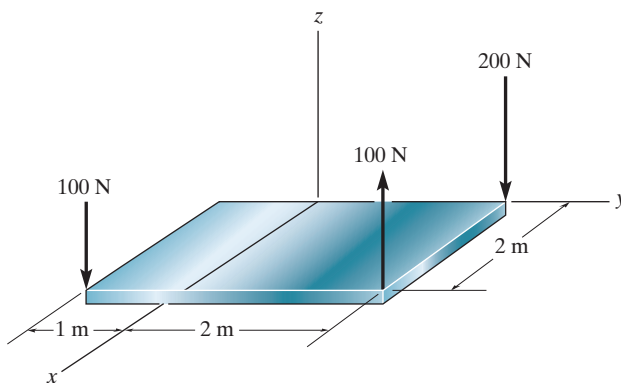
(c)

**Prob. P4-6**

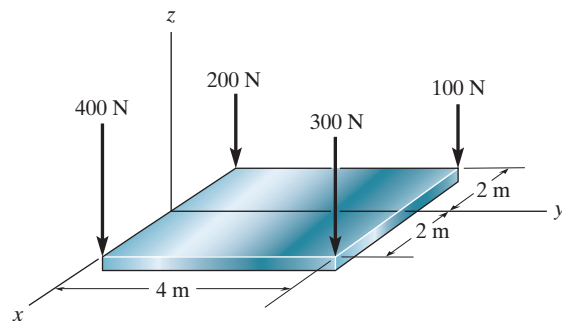
**P4-7.** In each case, determine the resultant force and specify its coordinates  $x$  and  $y$  where it acts on the  $x$ - $y$  plane.



(a)



(b)

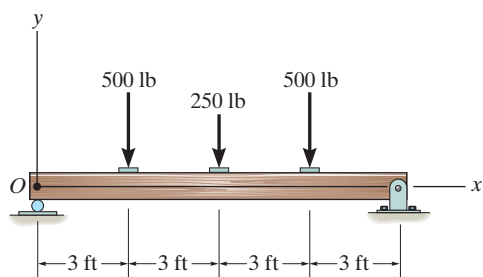


(c)

**Prob. P4-7**

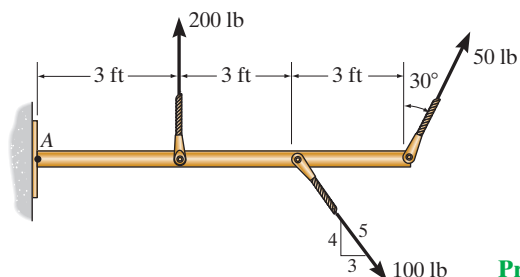
## FUNDAMENTAL PROBLEMS

**F4-31.** Replace the loading system by an equivalent resultant force and specify where the resultant's line of action intersects the beam measured from  $O$ .



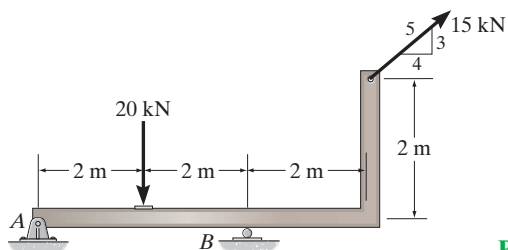
**Prob. F4-31**

**F4-32.** Replace the loading system by an equivalent resultant force and specify where the resultant's line of action intersects the member measured from  $A$ .



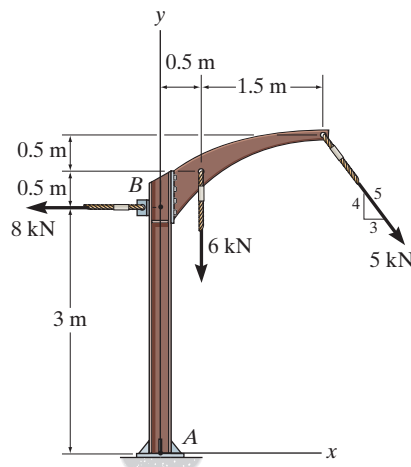
**Prob. F4-32**

**F4-33.** Replace the loading system by an equivalent resultant force and specify where the resultant's line of action intersects the horizontal segment of the member measured from  $A$ .



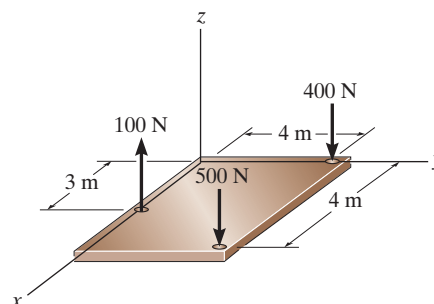
**Prob. F4-33**

**F4-34.** Replace the loading system by an equivalent resultant force and specify where the resultant's line of action intersects the member  $AB$  measured from  $A$ .



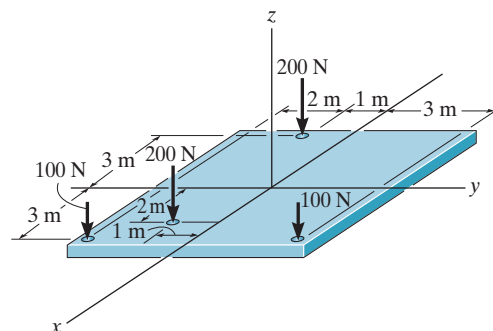
**Prob. F4-34**

**F4-35.** Replace the loading shown by an equivalent single resultant force and specify the  $x$  and  $y$  coordinates of its line of action.



**Prob. F4-35**

**F4-36.** Replace the loading shown by an equivalent single resultant force and specify the  $x$  and  $y$  coordinates of its line of action.

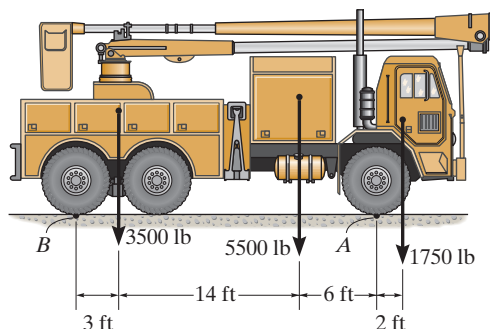


**Prob. F4-36**

## PROBLEMS

**4-113.** The weights of the various components of the truck are shown. Replace this system of forces by an equivalent resultant force and specify its location measured from  $B$ .

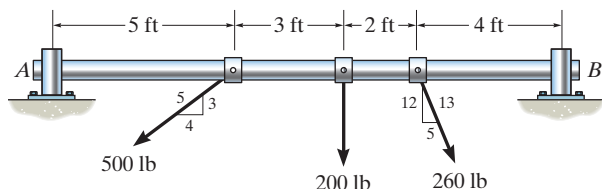
**4-114.** The weights of the various components of the truck are shown. Replace this system of forces by an equivalent resultant force and specify its location measured from point  $A$ .



**Probs. 4-113/114**

**4-115.** Replace the three forces acting on the shaft by a single resultant force. Specify where the force acts, measured from end  $A$ .

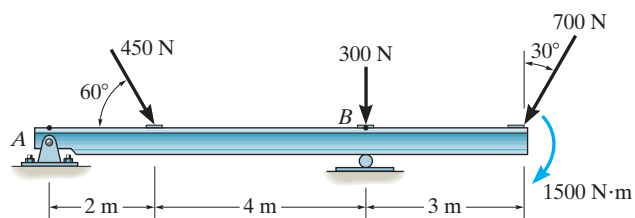
**\*4-116.** Replace the three forces acting on the shaft by a single resultant force. Specify where the force acts, measured from end  $B$ .



**Probs. 4-115/116**

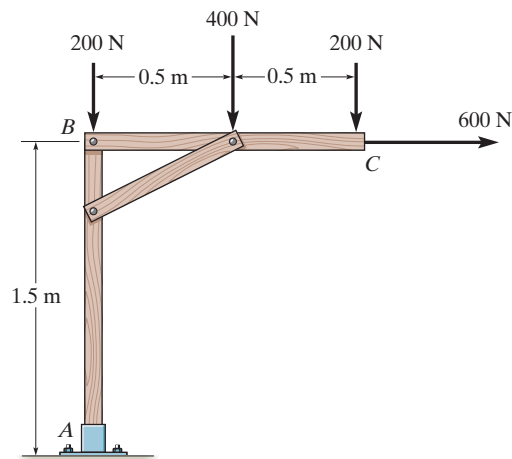
**4-117.** Replace the loading acting on the beam by a single resultant force. Specify where the force acts, measured from end  $A$ .

**4-118.** Replace the loading acting on the beam by a single resultant force. Specify where the force acts, measured from  $B$ .



**Probs. 4-117/118**

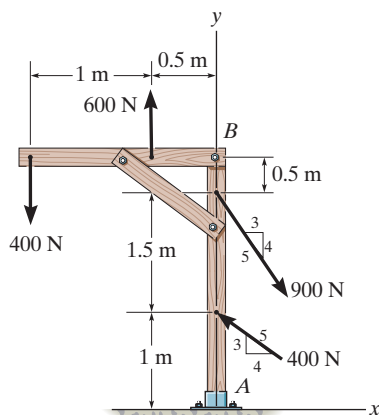
**4-119.** Replace the loading on the frame by a single resultant force. Specify where its line of action intersects a vertical line along member  $AB$ , measured from  $A$ .



**Prob. 4-119**

**\*4-120.** Replace the loading on the frame by a single resultant force. Specify where its line of action intersects a vertical line along member  $AB$ , measured from  $A$ .

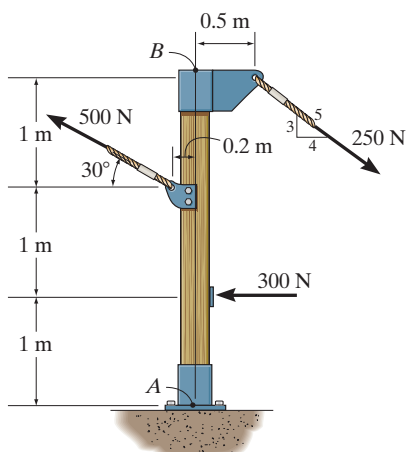
**4-121.** Replace the loading on the frame by a single resultant force. Specify where its line of action intersects a horizontal line along member  $CB$ , measured from end  $C$ .



**Probs. 4-120/121**

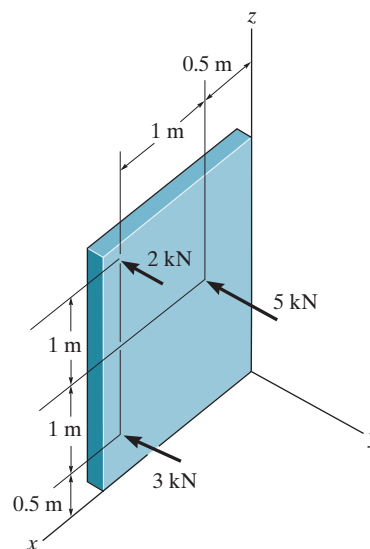
**4-122.** Replace the force system acting on the post by a resultant force, and specify where its line of action intersects the post  $AB$  measured from point  $A$ .

**4-123.** Replace the force system acting on the post by a resultant force, and specify where its line of action intersects the post  $AB$  measured from point  $B$ .



**Probs. 4-122/123**

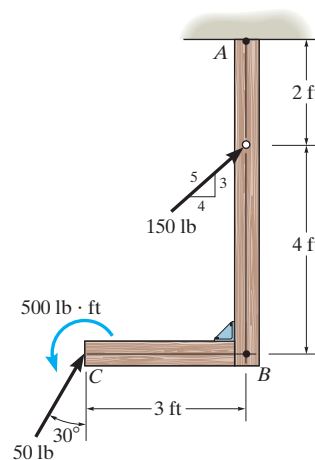
**\*4-124.** Replace the parallel force system acting on the plate by a resultant force and specify its location on the  $x$ - $z$  plane.



**Prob. 4-124**

**4-125.** Replace the force and couple system acting on the frame by an equivalent resultant force and specify where the resultant's line of action intersects member  $AB$ , measured from  $A$ .

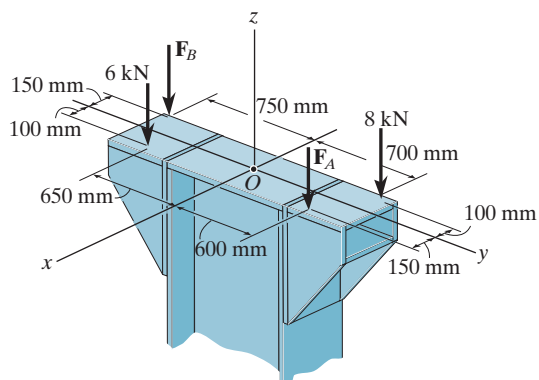
**4-126.** Replace the force and couple system acting on the frame by an equivalent resultant force and specify where the resultant's line of action intersects member  $BC$ , measured from  $B$ .



**Probs. 4-125/126**

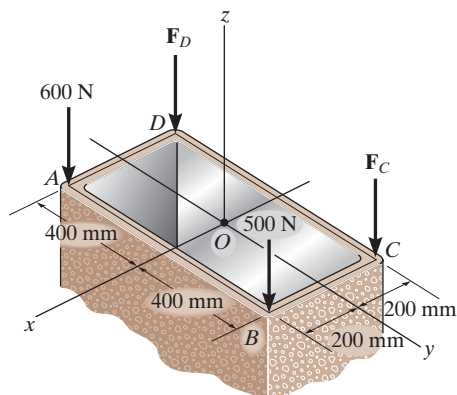
**4-127.** If  $F_A = 7$  kN and  $F_B = 5$  kN, represent the force system acting on the corbels by a resultant force, and specify its location on the  $x$ - $y$  plane.

**\*4-128.** Determine the magnitudes of  $F_A$  and  $F_B$  so that the resultant force passes through point  $O$  of the column.



**Probs. 4-127/128**

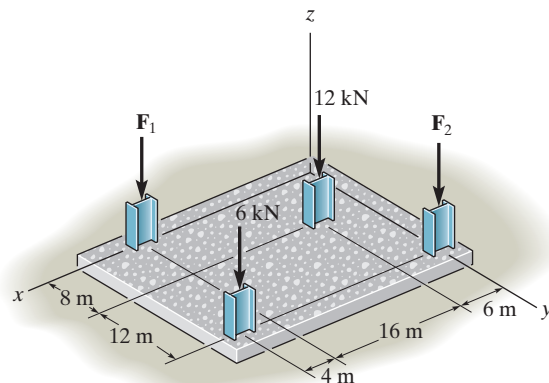
**4-129.** The tube supports the four parallel forces. Determine the magnitudes of forces  $F_C$  and  $F_D$  acting at  $C$  and  $D$  so that the equivalent resultant force of the force system acts through the midpoint  $O$  of the tube.



**Prob. 4-129**

**4-130.** The building slab is subjected to four parallel column loadings. Determine the equivalent resultant force and specify its location ( $x$ ,  $y$ ) on the slab. Take  $F_1 = 8$  kN and  $F_2 = 9$  kN.

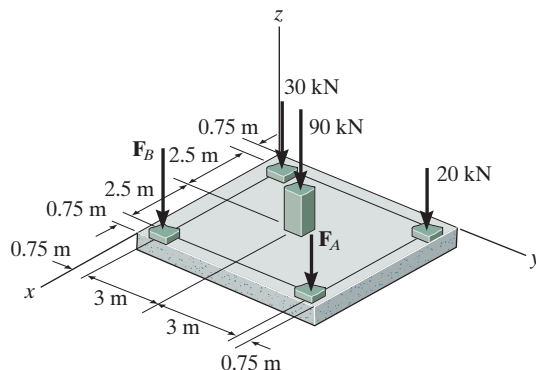
**4-131.** The building slab is subjected to four parallel column loadings. Determine  $F_1$  and  $F_2$  if the resultant force acts through point (12 m, 10 m).



**Probs. 4-130/131**

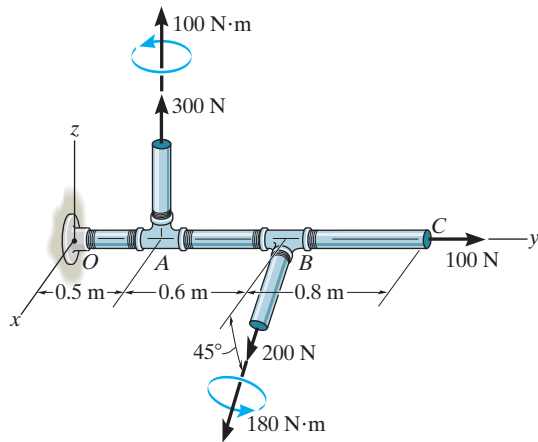
**\*4-132.** If  $F_A = 40$  kN and  $F_B = 35$  kN, determine the magnitude of the resultant force and specify the location of its point of application ( $x$ ,  $y$ ) on the slab.

**4-133.** If the resultant force is required to act at the center of the slab, determine the magnitude of the column loadings  $F_A$  and  $F_B$  and the magnitude of the resultant force.



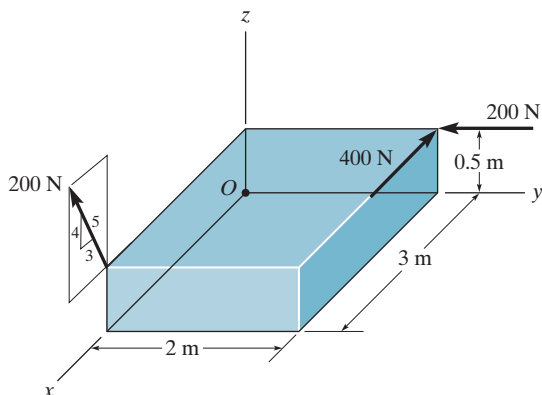
**Probs. 4-132/133**

**4-134.** Replace the two wrenches and the force, acting on the pipe assembly, by an equivalent resultant force and couple moment at point  $O$ .



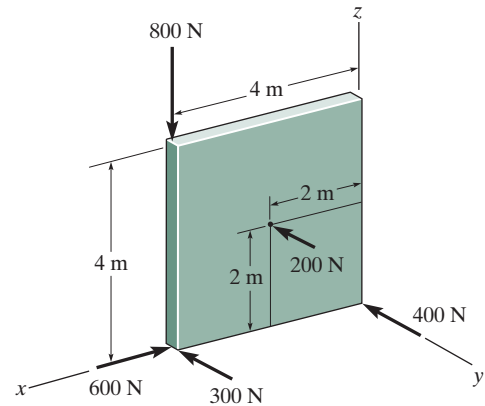
**Prob. 4-134**

**4-135.** Replace the force system by a wrench and specify the magnitude of the force and couple moment of the wrench and the point where the wrench intersects the  $x$ - $z$  plane.



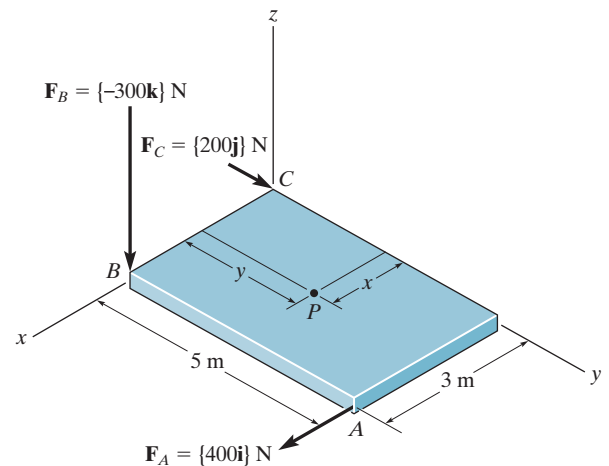
**Prob. 4-135**

**\*4-136.** Replace the five forces acting on the plate by a wrench. Specify the magnitude of the force and couple moment for the wrench and the point  $P(x, z)$  where the wrench intersects the  $x$ - $z$  plane.



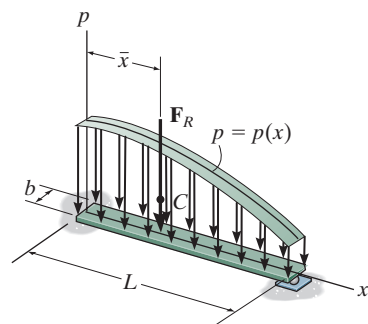
**Prob. 4-136**

**4-137.** Replace the three forces acting on the plate by a wrench. Specify the magnitude of the force and couple moment for the wrench and the point  $P(x, y)$  where the wrench intersects the plate.

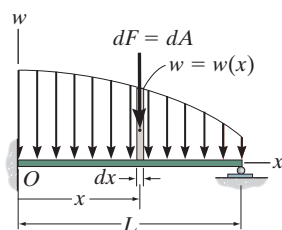


**Prob. 4-137**

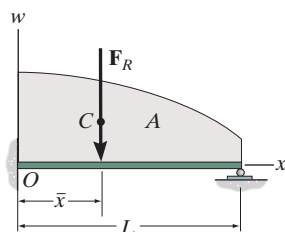




(a)



(b)



(c)

Fig. 4-48

## 4.9 Reduction of a Simple Distributed Loading

Sometimes, a body may be subjected to a loading that is distributed over its surface. For example, the pressure of the wind on the face of a sign, the pressure of water within a tank, or the weight of sand on the floor of a storage container, are all **distributed loadings**. The pressure exerted at each point on the surface indicates the intensity of the loading. It is measured using pascals Pa (or  $\text{N/m}^2$ ) in SI units or  $\text{lb/ft}^2$  in the U.S. Customary system.

**Loading Along a Single Axis.** The most common type of distributed loading encountered in engineering practice can be represented along a single axis.\* For example, consider the beam (or plate) in Fig. 4-48a that has a constant width and is subjected to a pressure loading that varies only along the  $x$  axis. This loading can be described by the function  $p = p(x) \text{ N/m}^2$ . It contains only one variable  $x$ , and for this reason, we can also represent it as a *coplanar distributed load*. To do so, we multiply the loading function by the width  $b$  m of the beam, so that  $w(x) = p(x)b \text{ N/m}$ , Fig. 4-48b. Using the methods of Sec. 4.8, we can replace this coplanar parallel force system with a single equivalent resultant force  $\mathbf{F}_R$  acting at a specific location on the beam, Fig. 4-48c.

**Magnitude of Resultant Force.** From Eq. 4-17 ( $F_R = \Sigma F$ ), the magnitude of  $\mathbf{F}_R$  is equivalent to the sum of all the forces in the system. In this case integration must be used since there is an infinite number of parallel forces  $d\mathbf{F}$  acting on the beam, Fig. 4-48b. Since  $d\mathbf{F}$  is acting on an element of length  $dx$ , and  $w(x)$  is a force per unit length, then  $dF = w(x) dx = dA$ . In other words, the magnitude of  $d\mathbf{F}$  is determined from the colored differential area  $dA$  under the loading curve. For the entire length  $L$ ,

$$+\downarrow F_R = \Sigma F;$$

$$F_R = \int_L w(x) dx = \int_A dA = A \quad (4-19)$$

Therefore, the magnitude of the resultant force is equal to the area  $A$  under the loading diagram, Fig. 4-48c.

\*The more general case of a surface loading acting on a body is considered in Sec. 9.5.

**Location of Resultant Force.** Applying Eq. 4-17 ( $M_{R_O} = \Sigma M_O$ ), the location  $\bar{x}$  of the line of action of  $\mathbf{F}_R$  can be determined by equating the moments of the force resultant and the parallel force distribution about point  $O$  (the  $y$  axis). Since  $d\mathbf{F}$  produces a moment of  $x dF = xw(x) dx$  about  $O$ , Fig. 4-48b, then for the entire length, Fig. 4-48c,

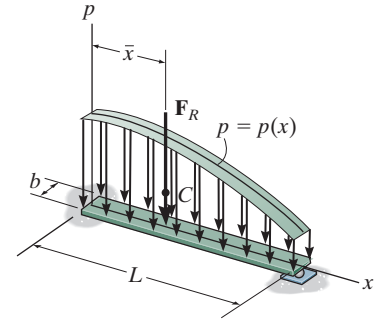
$$\zeta + (M_R)_O = \Sigma M_O; \quad -\bar{x}F_R = -\int_L xw(x) dx$$

Solving for  $\bar{x}$ , using Eq. 4-19, we have

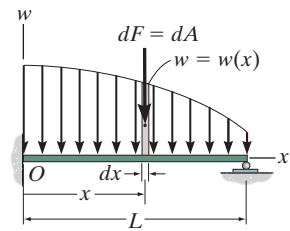
$$\bar{x} = \frac{\int_L xw(x) dx}{\int_L w(x) dx} = \frac{\int_A x dA}{\int_A dA} \quad (4-20)$$

This coordinate  $\bar{x}$ , locates the geometric center or **centroid** of the *area* under the distributed loading. In other words, the resultant force has a line of action which passes through the centroid  $C$  (geometric center) of the area under the loading diagram, Fig. 4-48c. Detailed treatment of the integration techniques for finding the location of the centroid for areas is given in Chapter 9. In many cases, however, the distributed-loading diagram is in the shape of a rectangle, triangle, or some other simple geometric form. The centroid location for such common shapes does not have to be determined from the above equation but can be obtained directly from the tabulation given on the inside back cover.

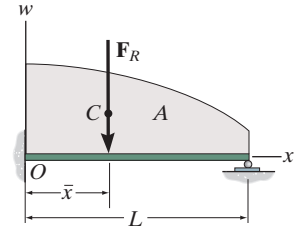
Once  $\bar{x}$  is determined,  $\mathbf{F}_R$  by symmetry passes through point  $(\bar{x}, 0)$  on the surface of the beam, Fig. 4-48a. Therefore, in this case the resultant force has a magnitude equal to the volume under the loading curve  $p = p(x)$  and a line of action which passes through the centroid (geometric center) of this volume.



(a)



(b)

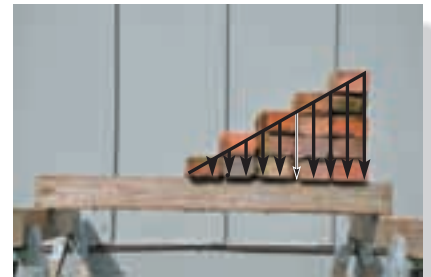


(c)

Fig. 4-48 (Repeated)

## Important Points

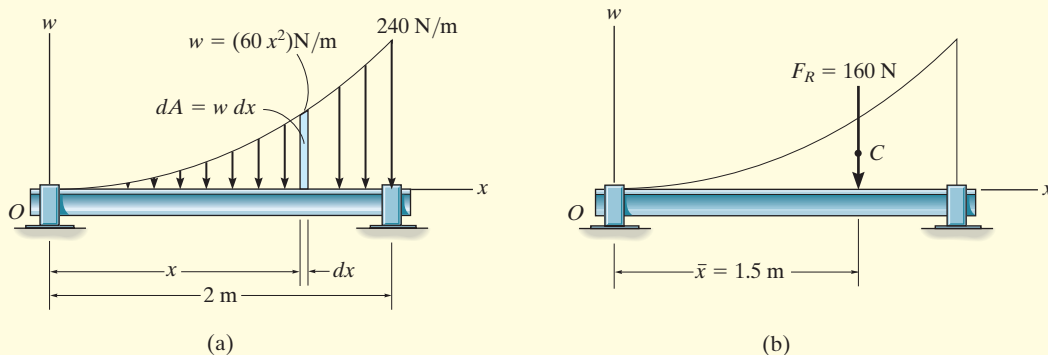
- Coplanar distributed loadings are defined by using a loading function  $w = w(x)$  that indicates the intensity of the loading along the length of a member. This intensity is measured in N/m or lb/ft.
- The external effects caused by a coplanar distributed load acting on a body can be represented by a single resultant force.
- This resultant force is equivalent to the *area* under the loading diagram, and has a line of action that passes through the *centroid* or geometric center of this area.



The pile of brick creates an approximate triangular distributed loading on the board.  
(© Russell C. Hibbeler)

**EXAMPLE 4.21**

Determine the magnitude and location of the equivalent resultant force acting on the shaft in Fig. 4–49a.

**Fig. 4–49****SOLUTION**

Since  $w = w(x)$  is given, this problem will be solved by integration.

The differential element has an area  $dA = w dx = 60x^2 dx$ . Applying Eq. 4–19,

$$+\downarrow F_R = \Sigma F;$$

$$\begin{aligned} F_R &= \int_A dA = \int_0^{2 \text{ m}} 60x^2 dx = 60 \left( \frac{x^3}{3} \right) \bigg|_0^{2 \text{ m}} = 60 \left( \frac{2^3}{3} - \frac{0^3}{3} \right) \\ &= 160 \text{ N} \end{aligned}$$

*Ans.*

The location  $\bar{x}$  of  $F_R$  measured from  $O$ , Fig. 4–49b, is determined from Eq. 4–20.

$$\begin{aligned} \bar{x} &= \frac{\int_A x dA}{\int_A dA} = \frac{\int_0^{2 \text{ m}} x(60x^2) dx}{160 \text{ N}} = \frac{60 \left( \frac{x^4}{4} \right) \bigg|_0^{2 \text{ m}}}{160 \text{ N}} = \frac{60 \left( \frac{2^4}{4} - \frac{0^4}{4} \right)}{160 \text{ N}} \\ &= 1.5 \text{ m} \end{aligned}$$

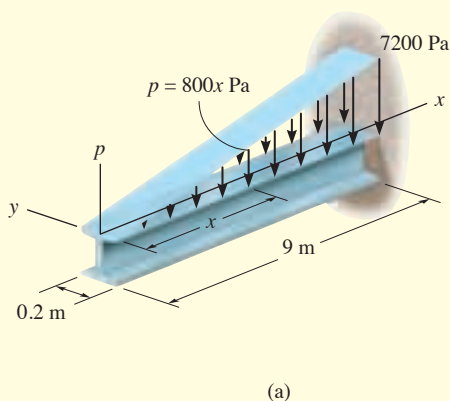
*Ans.*

**NOTE:** These results can be checked by using the table on the inside back cover, where it is shown that the formula for an exparabolic area of length  $a$ , height  $b$ , and shape shown in Fig. 4–49a, is

$$A = \frac{ab}{3} = \frac{2 \text{ m}(240 \text{ N/m})}{3} = 160 \text{ N} \text{ and } \bar{x} = \frac{3}{4}a = \frac{3}{4}(2 \text{ m}) = 1.5 \text{ m}$$

**EXAMPLE 4.22**

A distributed loading of  $p = (800x)$  Pa acts over the top surface of the beam shown in Fig. 4-50a. Determine the magnitude and location of the equivalent resultant force.

**SOLUTION**

Since the loading intensity is uniform along the width of the beam (the  $y$  axis), the loading can be viewed in two dimensions as shown in Fig. 4-50b. Here

$$\begin{aligned} w &= (800x \text{ N/m}^2)(0.2 \text{ m}) \\ &= (160x) \text{ N/m} \end{aligned}$$

At  $x = 9 \text{ m}$ , note that  $w = 1440 \text{ N/m}$ . Although we may again apply Eqs. 4-19 and 4-20 as in the previous example, it is simpler to use the table on the inside back cover.

The magnitude of the resultant force is equivalent to the area of the triangle.

$$F_R = \frac{1}{2}(9 \text{ m})(1440 \text{ N/m}) = 6480 \text{ N} = 6.48 \text{ kN} \quad \text{Ans.}$$

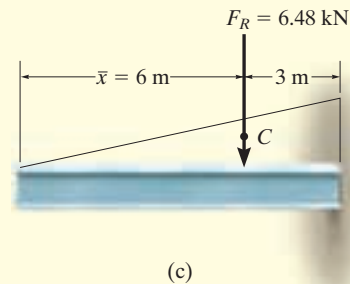
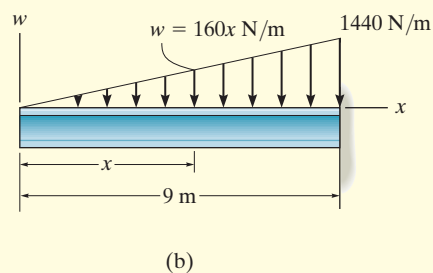
The line of action of  $F_R$  passes through the *centroid*  $C$  of this triangle. Hence,

$$\bar{x} = 9 \text{ m} - \frac{1}{3}(9 \text{ m}) = 6 \text{ m} \quad \text{Ans.}$$

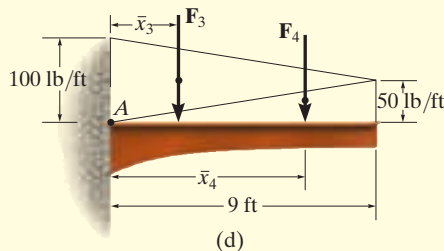
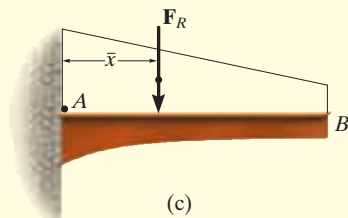
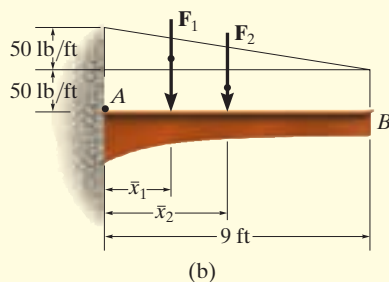
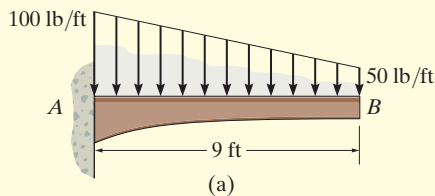
The results are shown in Fig. 4-50c.

**NOTE:** We may also view the resultant  $F_R$  as *acting* through the *centroid* of the *volume* of the loading diagram  $p = p(x)$  in Fig. 4-50a. Hence  $F_R$  intersects the  $x$ - $y$  plane at the point (6 m, 0). Furthermore, the magnitude of  $F_R$  is equal to the volume under the loading diagram; i.e.,

$$F_R = V = \frac{1}{2}(7200 \text{ N/m}^2)(9 \text{ m})(0.2 \text{ m}) = 6.48 \text{ kN} \quad \text{Ans.}$$



**Fig. 4-50**

**EXAMPLE 4.23****Fig. 4-51**

The granular material exerts the distributed loading on the beam as shown in Fig. 4-51a. Determine the magnitude and location of the equivalent resultant of this load.

**SOLUTION**

The area of the loading diagram is a *trapezoid*, and therefore the solution can be obtained directly from the area and centroid formulas for a trapezoid listed on the inside back cover. Since these formulas are not easily remembered, instead we will solve this problem by using “composite” areas. Here we will divide the trapezoidal loading into a rectangular and triangular loading as shown in Fig. 4-51b. The magnitude of the force represented by each of these loadings is equal to its associated *area*,

$$F_1 = \frac{1}{2}(9 \text{ ft})(50 \text{ lb/ft}) = 225 \text{ lb}$$

$$F_2 = (9 \text{ ft})(50 \text{ lb/ft}) = 450 \text{ lb}$$

The lines of action of these parallel forces act through the respective *centroids* of their associated areas and therefore intersect the beam at

$$\bar{x}_1 = \frac{1}{3}(9 \text{ ft}) = 3 \text{ ft}$$

$$\bar{x}_2 = \frac{1}{2}(9 \text{ ft}) = 4.5 \text{ ft}$$

The two parallel forces  $F_1$  and  $F_2$  can be reduced to a single resultant  $F_R$ . The magnitude of  $F_R$  is

$$+\downarrow F_R = \Sigma F; \quad F_R = 225 + 450 = 675 \text{ lb} \quad \text{Ans.}$$

We can find the location of  $F_R$  with reference to point A, Figs. 4-51b and 4-51c. We require

$$\zeta + (M_R)_A = \Sigma M_A; \quad \bar{x}(675) = 3(225) + 4.5(450) \quad \text{Ans.}$$

$$\bar{x} = 4 \text{ ft}$$

**NOTE:** The trapezoidal area in Fig. 4-51a can also be divided into two triangular areas as shown in Fig. 4-51d. In this case

$$F_3 = \frac{1}{2}(9 \text{ ft})(100 \text{ lb/ft}) = 450 \text{ lb}$$

$$F_4 = \frac{1}{2}(9 \text{ ft})(50 \text{ lb/ft}) = 225 \text{ lb}$$

and

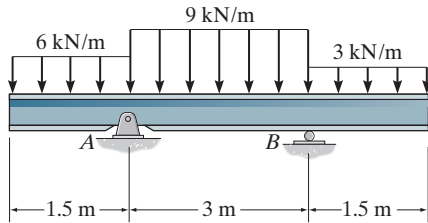
$$\bar{x}_3 = \frac{1}{3}(9 \text{ ft}) = 3 \text{ ft}$$

$$\bar{x}_4 = 9 \text{ ft} - \frac{1}{3}(9 \text{ ft}) = 6 \text{ ft}$$

Using these results, show that again  $F_R = 675 \text{ lb}$  and  $\bar{x} = 4 \text{ ft}$ .

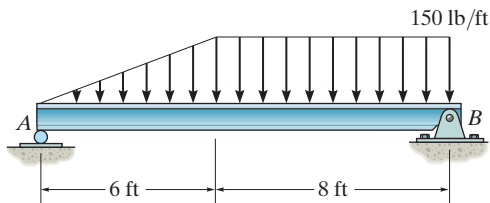
## FUNDAMENTAL PROBLEMS

**F4-37.** Determine the resultant force and specify where it acts on the beam measured from  $A$ .



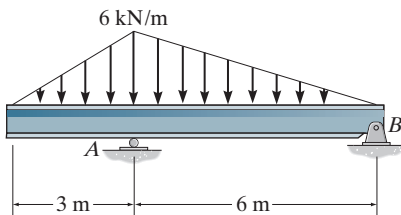
**Prob. F4-37**

**F4-38.** Determine the resultant force and specify where it acts on the beam measured from  $A$ .



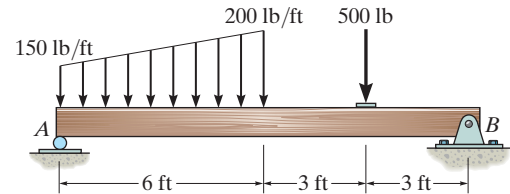
**Prob. F4-38**

**F4-39.** Determine the resultant force and specify where it acts on the beam measured from  $A$ .



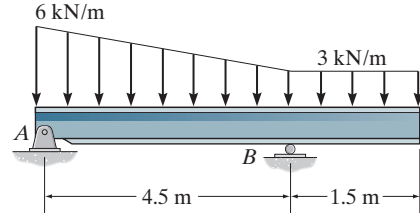
**Prob. F4-39**

**F4-40.** Determine the resultant force and specify where it acts on the beam measured from  $A$ .



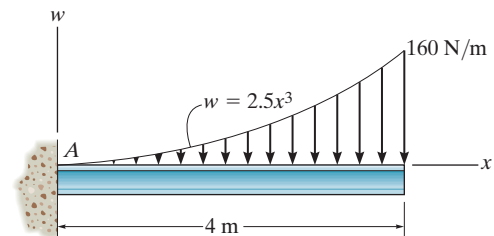
**Prob. F4-40**

**F4-41.** Determine the resultant force and specify where it acts on the beam measured from  $A$ .



**Prob. F4-41**

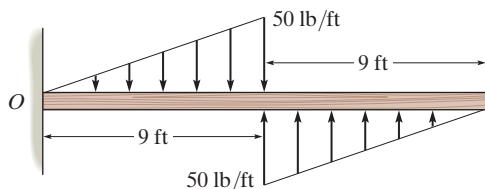
**F4-42.** Determine the resultant force and specify where it acts on the beam measured from  $A$ .



**Prob. F4-42**

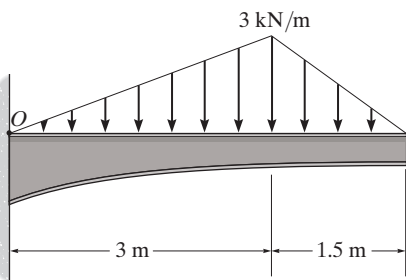
## PROBLEMS

**4-138.** Replace the loading by an equivalent resultant force and couple moment acting at point  $O$ .



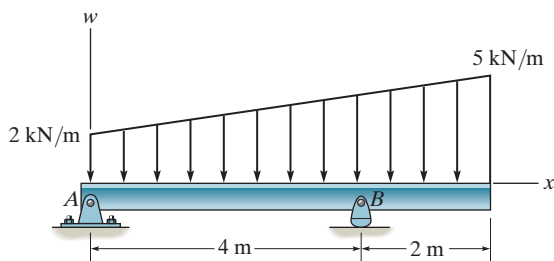
**Prob. 4-138**

**4-139.** Replace the distributed loading with an equivalent resultant force, and specify its location on the beam measured from point  $O$ .



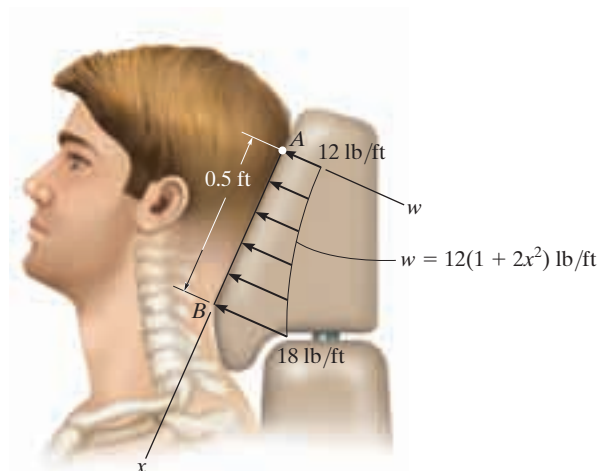
**Prob. 4-139**

**\*4-140.** Replace the loading by an equivalent resultant force and specify its location on the beam, measured from point  $A$ .



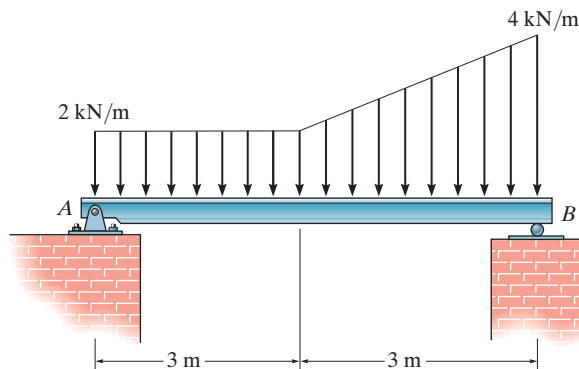
**Prob. 4-140**

**4-141.** Currently eighty-five percent of all neck injuries are caused by rear-end car collisions. To alleviate this problem, an automobile seat restraint has been developed that provides additional pressure contact with the cranium. During dynamic tests the distribution of load on the cranium has been plotted and shown to be parabolic. Determine the equivalent resultant force and its location, measured from point  $A$ .



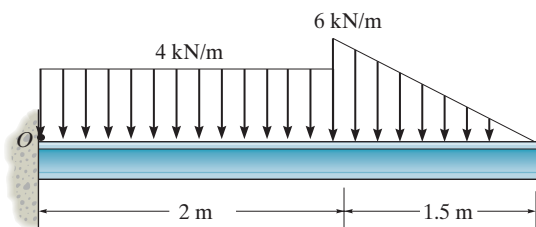
**Prob. 4-141**

**4-142.** Replace the distributed loading by an equivalent resultant force, and specify its location on the beam, measured from the pin at  $A$ .



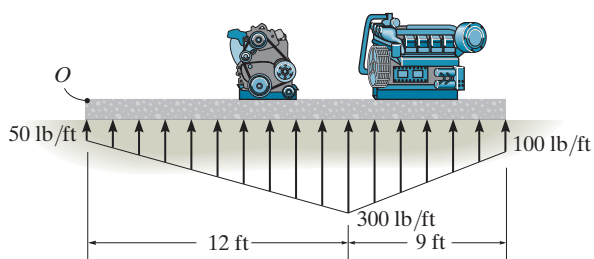
**Prob. 4-142**

**4-143.** Replace this loading by an equivalent resultant force and specify its location, measured from point  $O$ .



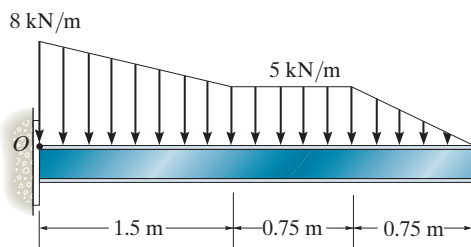
**Prob. 4-143**

**\*4-144.** The distribution of soil loading on the bottom of a building slab is shown. Replace this loading by an equivalent resultant force and specify its location, measured from point  $O$ .



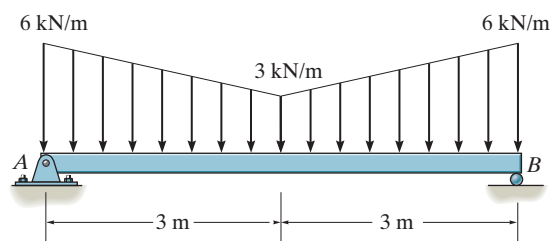
**Prob. 4-144**

**4-145.** Replace the loading by an equivalent resultant force and couple moment acting at point  $O$ .



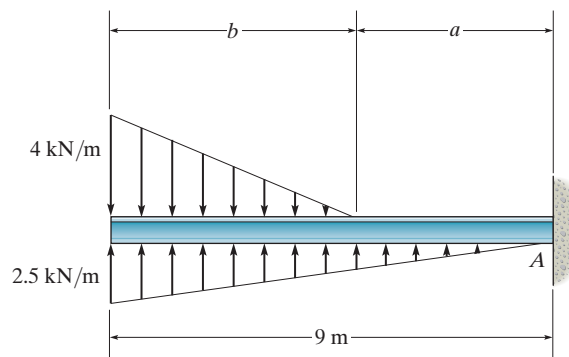
**Prob. 4-145**

**4-146.** Replace the distributed loading by an equivalent resultant force and couple moment acting at point  $A$ .



**Prob. 4-146**

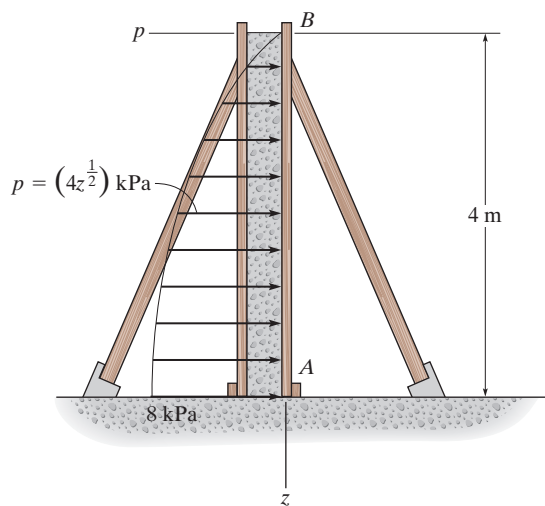
**4-147.** Determine the length  $b$  of the triangular load and its position  $a$  on the beam such that the equivalent resultant force is zero and the resultant couple moment is  $8 \text{ kN} \cdot \text{m}$  clockwise.



**Prob. 4-147**

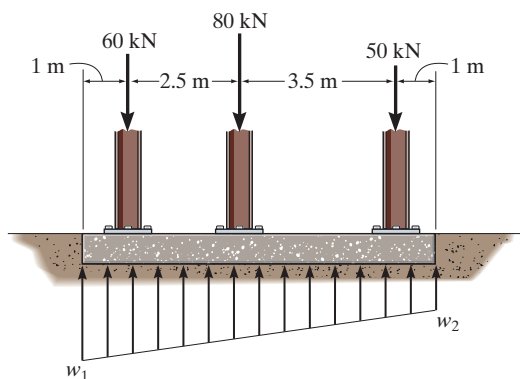


**\*4-148.** The form is used to cast a concrete wall having a width of 5 m. Determine the equivalent resultant force the wet concrete exerts on the form  $AB$  if the pressure distribution due to the concrete can be approximated as shown. Specify the location of the resultant force, measured from point  $B$ .



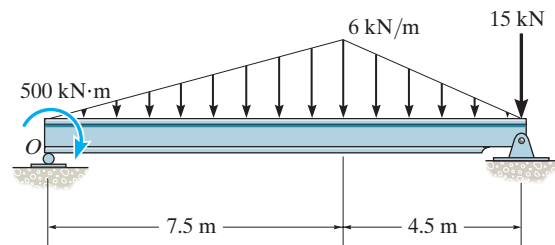
**Prob. 4-148**

**4-149.** If the soil exerts a trapezoidal distribution of load on the bottom of the footing, determine the intensities  $w_1$  and  $w_2$  of this distribution needed to support the column loadings.



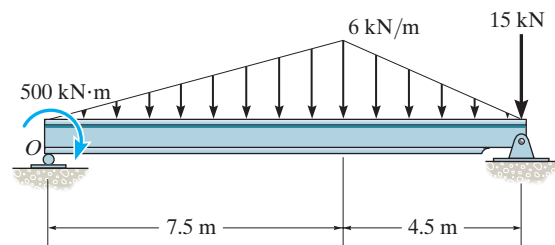
**Prob. 4-149**

**4-150.** Replace the loading by an equivalent force and couple moment acting at point  $O$ .



**Prob. 4-150**

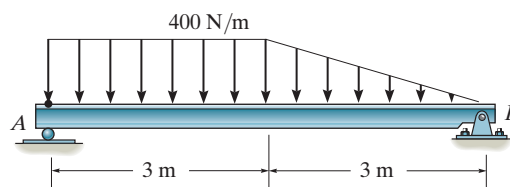
**4-151.** Replace the loading by a single resultant force, and specify the location of the force measured from point  $O$ .



**Prob. 4-151**

**\*4-152.** Replace the loading by an equivalent resultant force and couple moment acting at point  $A$ .

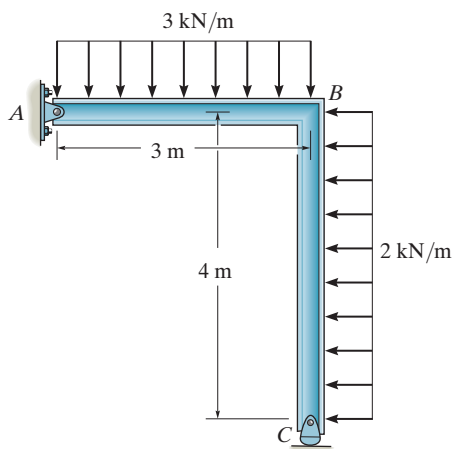
**4-153.** Replace the loading by a single resultant force, and specify its location on the beam measured from point  $A$ .



**Probs. 4-152/153**

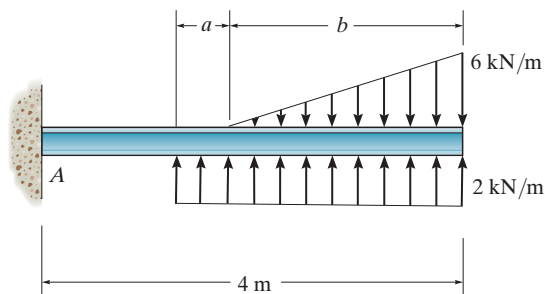
**4-154.** Replace the distributed loading by an equivalent resultant force and specify where its line of action intersects a horizontal line along member  $AB$ , measured from  $A$ .

**4-155.** Replace the distributed loading by an equivalent resultant force and specify where its line of action intersects a vertical line along member  $BC$ , measured from  $C$ .



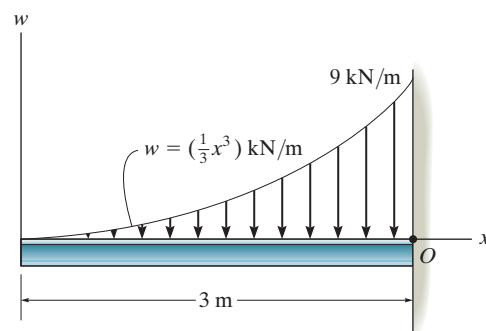
**Probs. 4-154/155**

**\*4-156.** Determine the length  $b$  of the triangular load and its position  $a$  on the beam such that the equivalent resultant force is zero and the resultant couple moment is  $8 \text{ kN} \cdot \text{m}$  clockwise.



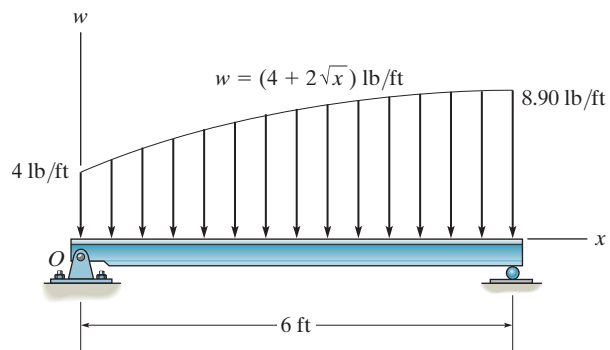
**Prob. 4-156**

**4-157.** Determine the equivalent resultant force and couple moment at point  $O$ .



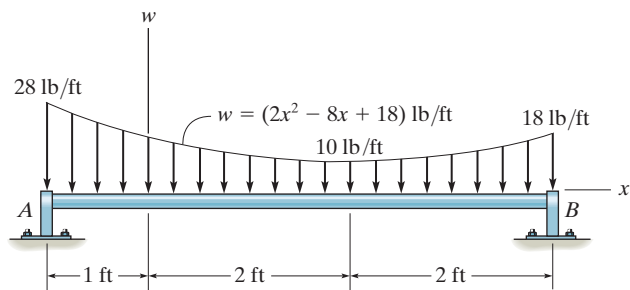
**Prob. 4-157**

**4-158.** Determine the magnitude of the equivalent resultant force and its location, measured from point  $O$ .



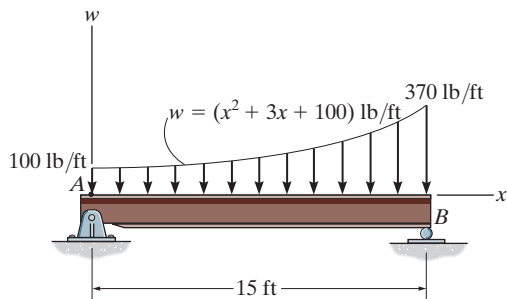
**Prob. 4-158**

**4-159.** The distributed load acts on the shaft as shown. Determine the magnitude of the equivalent resultant force and specify its location, measured from the support,  $A$ .



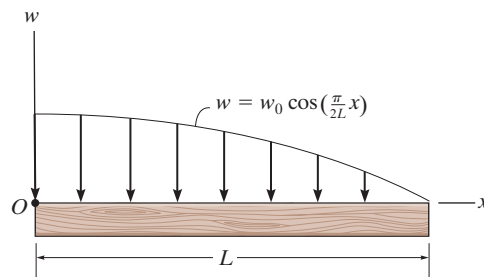
**Prob. 4-159**

**\*4-160.** Replace the distributed loading with an equivalent resultant force, and specify its location on the beam measured from point  $A$ .



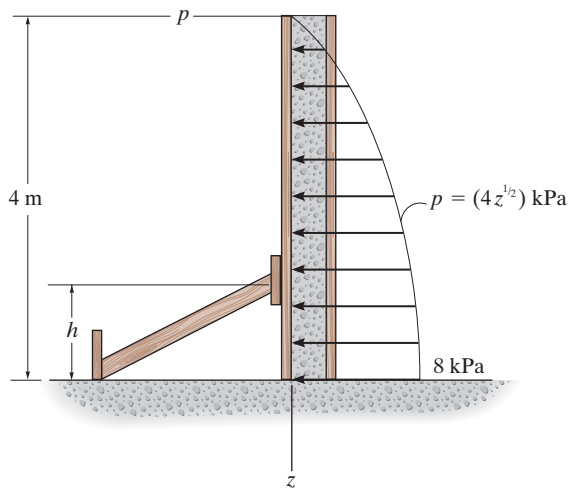
**Prob. 4-160**

**4-161.** Replace the loading by an equivalent resultant force and couple moment acting at point  $O$ .



**Prob. 4-161**

**4-162.** Wet concrete exerts a pressure distribution along the wall of the form. Determine the resultant force of this distribution and specify the height  $h$  where the bracing strut should be placed so that it lies through the line of action of the resultant force. The wall has a width of 5 m.



**Prob. 4-162**

## CHAPTER REVIEW

### Moment of Force—Scalar Definition

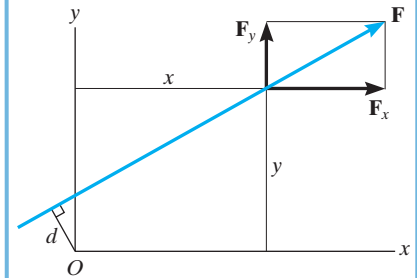
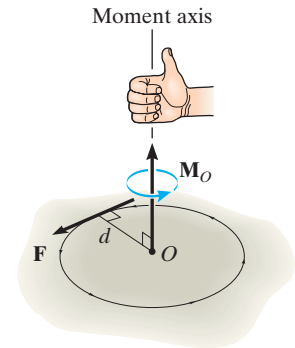
A force produces a turning effect or moment about a point  $O$  that does not lie on its line of action. In scalar form, the moment *magnitude* is the product of the force and the moment arm or perpendicular distance from point  $O$  to the line of action of the force.

The *direction* of the moment is defined using the right-hand rule.  $\mathbf{M}_O$  always acts along an axis perpendicular to the plane containing  $\mathbf{F}$  and  $d$ , and passes through the point  $O$ .

Rather than finding  $d$ , it is normally easier to resolve the force into its  $x$  and  $y$  components, determine the moment of each component about the point, and then sum the results. This is called the principle of moments.

$$M_O = Fd$$

$$M_O = Fd = F_x y - F_y x$$



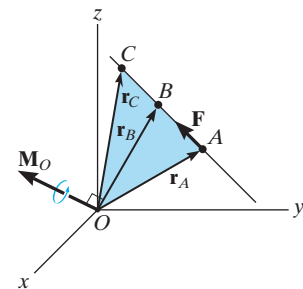
### Moment of a Force—Vector Definition

Since three-dimensional geometry is generally more difficult to visualize, the vector cross product should be used to determine the moment. Here  $\mathbf{M}_O = \mathbf{r} \times \mathbf{F}$ , where  $\mathbf{r}$  is a position vector that extends from point  $O$  to any point  $A$ ,  $B$ , or  $C$  on the line of action of  $\mathbf{F}$ .

If the position vector  $\mathbf{r}$  and force  $\mathbf{F}$  are expressed as Cartesian vectors, then the cross product results from the expansion of a determinant.

$$\mathbf{M}_O = \mathbf{r}_A \times \mathbf{F} = \mathbf{r}_B \times \mathbf{F} = \mathbf{r}_C \times \mathbf{F}$$

$$\mathbf{M}_O = \mathbf{r} \times \mathbf{F} = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ r_x & r_y & r_z \\ F_x & F_y & F_z \end{vmatrix}$$



### Moment about an Axis

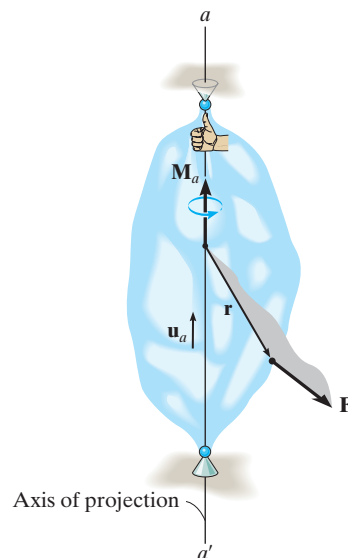
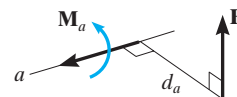
If the moment of a force  $\mathbf{F}$  is to be determined about an arbitrary axis  $a$ , then for a scalar solution the moment arm, or shortest distance  $d_a$  from the line of action of the force to the axis must be used. This distance is perpendicular to both the axis and the force line of action.

Note that when the line of action of  $\mathbf{F}$  intersects the axis, then the moment of  $\mathbf{F}$  about the axis is zero. Also, when the line of action of  $\mathbf{F}$  is parallel to the axis, the moment of  $\mathbf{F}$  about the axis is zero.

In three dimensions, the scalar triple product should be used. Here  $\mathbf{u}_a$  is the unit vector that specifies the direction of the axis, and  $\mathbf{r}$  is a position vector that is directed from any point on the axis to any point on the line of action of the force. If  $M_a$  is calculated as a negative scalar, then the sense of direction of  $\mathbf{M}_a$  is opposite to  $\mathbf{u}_a$ .

$$M_a = Fd_a$$

$$M_a = \mathbf{u}_a \cdot (\mathbf{r} \times \mathbf{F}) = \begin{vmatrix} u_{a_x} & u_{a_y} & u_{a_z} \\ r_x & r_y & r_z \\ F_x & F_y & F_z \end{vmatrix}$$



### Couple Moment

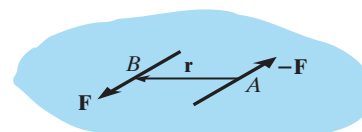
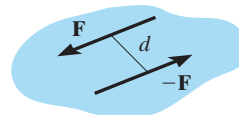
A couple consists of two equal but opposite forces that act a perpendicular distance  $d$  apart. Couples tend to produce a rotation without translation.

The magnitude of the couple moment is  $M = Fd$ , and its direction is established using the right-hand rule.

If the vector cross product is used to determine the moment of a couple, then  $\mathbf{r}$  extends from any point on the line of action of one of the forces to any point on the line of action of the other force  $\mathbf{F}$  that is used in the cross product.

$$M = Fd$$

$$\mathbf{M} = \mathbf{r} \times \mathbf{F}$$

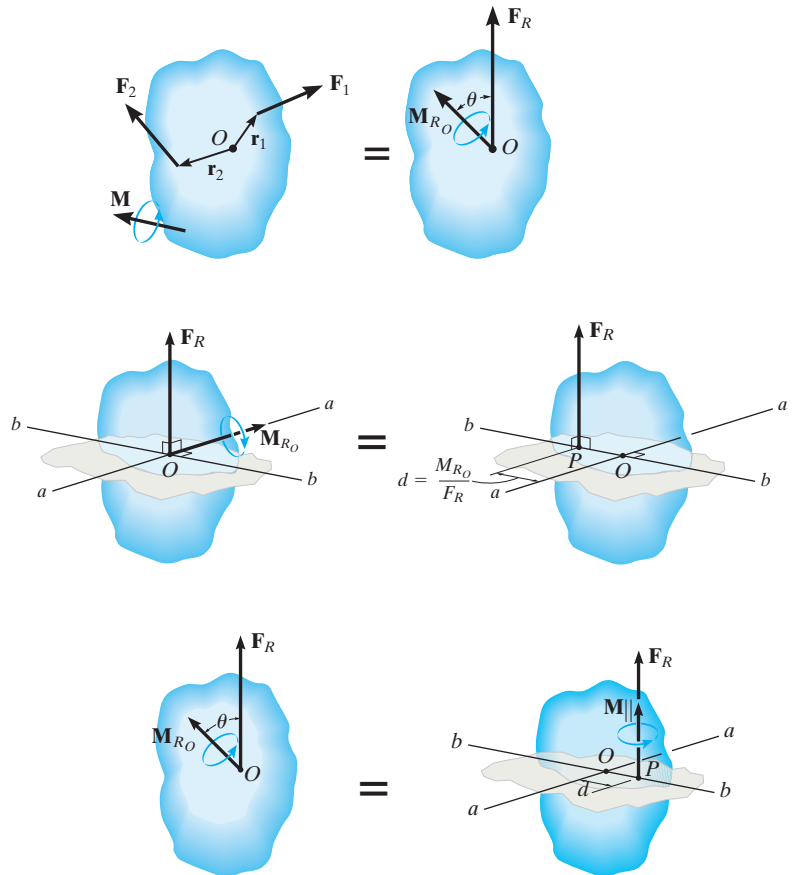


### Simplification of a Force and Couple System

Any system of forces and couples can be reduced to a single resultant force and resultant couple moment acting at a point. The resultant force is the sum of all the forces in the system,  $\mathbf{F}_R = \Sigma \mathbf{F}$ , and the resultant couple moment is equal to the sum of all the moments of the forces about the point and couple moments.  $\mathbf{M}_{R_O} = \Sigma \mathbf{M}_O + \Sigma \mathbf{M}$ .

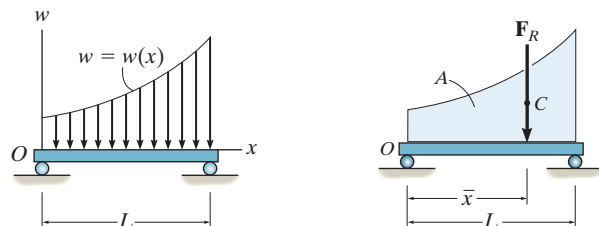
Further simplification to a single resultant force is possible provided the force system is concurrent, coplanar, or parallel. To find the location of the resultant force from a point, it is necessary to equate the moment of the resultant force about the point to the moment of the forces and couples in the system about the same point.

If the resultant force and couple moment at a point are not perpendicular to one another, then this system can be reduced to a wrench, which consists of the resultant force and collinear couple moment.



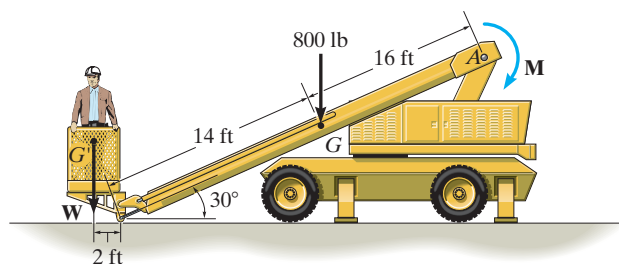
### Coplanar Distributed Loading

A simple distributed loading can be represented by its resultant force, which is equivalent to the *area* under the loading curve. This resultant has a line of action that passes through the *centroid* or geometric center of the area or volume under the loading diagram.



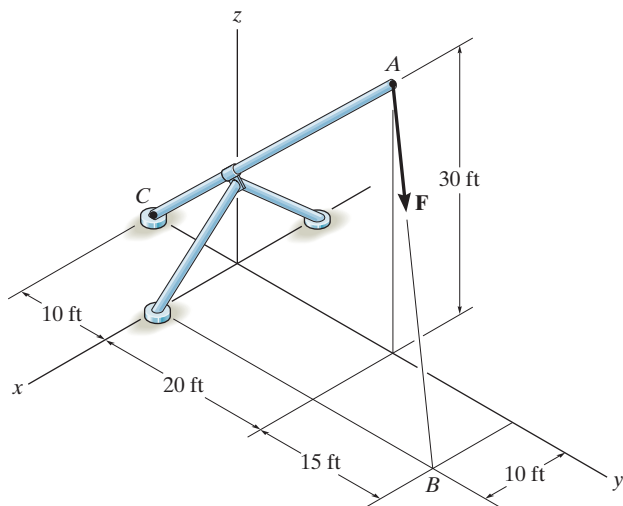
## REVIEW PROBLEMS

**R4-1.** The boom has a length of 30 ft, a weight of 800 lb, and mass center at  $G$ . If the maximum moment that can be developed by a motor at  $A$  is  $M = 20(10^3)$  lb·ft, determine the maximum load  $W$ , having a mass center at  $G'$ , that can be lifted.



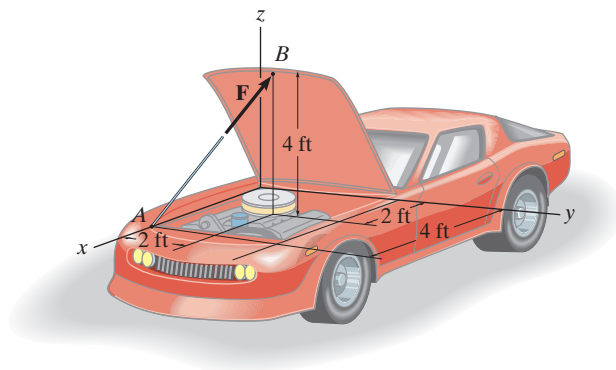
**Prob. R4-1**

**R4-2.** Replace the force  $\mathbf{F}$  having a magnitude of  $F = 50$  lb and acting at point  $A$  by an equivalent force and couple moment at point  $C$ .



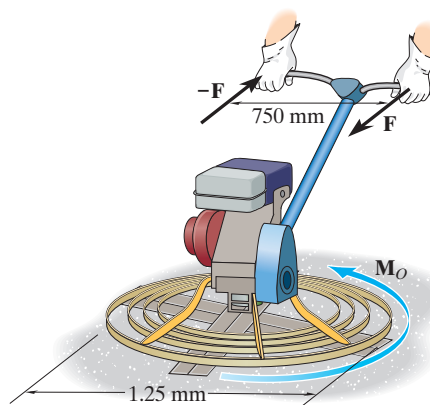
**Prob. R4-2**

**R4-3.** The hood of the automobile is supported by the strut  $AB$ , which exerts a force of  $F = 24$  lb on the hood. Determine the moment of this force about the hinged axis  $y$ .



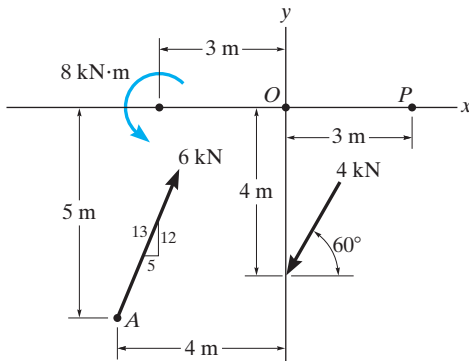
**Prob. R4-3**

**R4-4.** Friction on the concrete surface creates a couple moment of  $M_O = 100$  N·m on the blades of the trowel. Determine the magnitude of the couple forces so that the resultant couple moment on the trowel is zero. The forces lie in the horizontal plane and act perpendicular to the handle of the trowel.



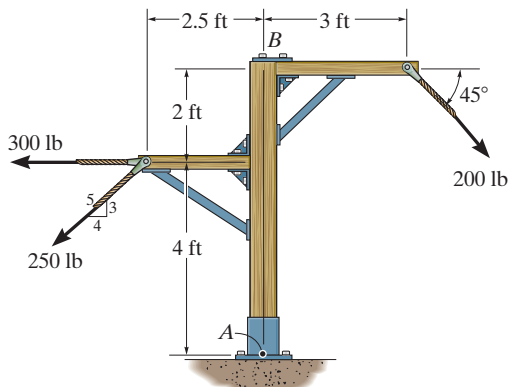
**Prob. R4-4**

**R4-5.** Replace the force and couple system by an equivalent force and couple moment at point  $P$ .



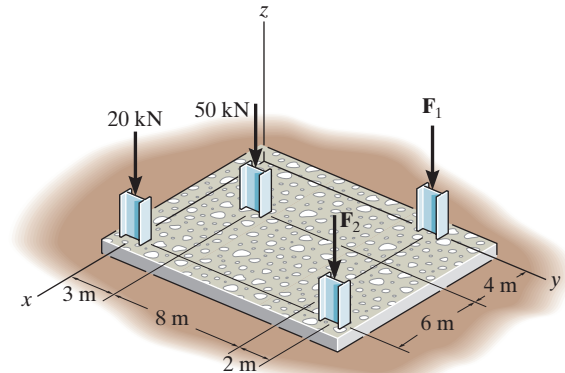
**Prob. R4-5**

**R4-6.** Replace the force system acting on the frame by a resultant force, and specify where its line of action intersects member  $AB$ , measured from point  $A$ .



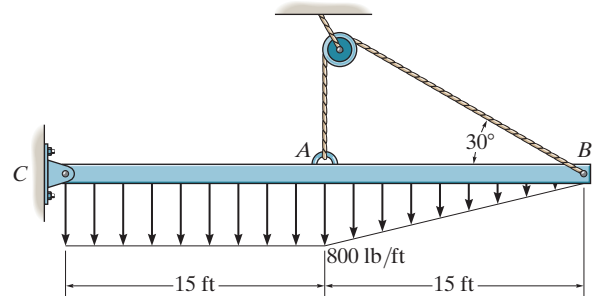
**Prob. R4-6**

**R4-7.** The building slab is subjected to four parallel column loadings. Determine the equivalent resultant force and specify its location  $(x, y)$  on the slab. Take  $F_1 = 30$  kN,  $F_2 = 40$  kN.



**Prob. R4-7**

**R4-8.** Replace the distributed loading by an equivalent resultant force, and specify its location on the beam, measured from the pin at  $C$ .



**Prob. R4-8**



# Chapter 5



(© YuryZap/Shutterstock)

It is important to be able to determine the forces in the cables used to support this boom to ensure that it does not fail. In this chapter we will study how to apply equilibrium methods to determine the forces acting on the supports of a rigid body such as this.

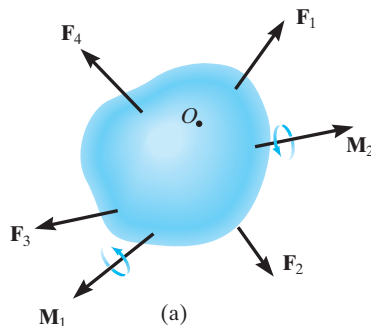
# Equilibrium of a Rigid Body

## CHAPTER OBJECTIVES

- To develop the equations of equilibrium for a rigid body.
- To introduce the concept of the free-body diagram for a rigid body.
- To show how to solve rigid-body equilibrium problems using the equations of equilibrium.

## 5.1 Conditions for Rigid-Body Equilibrium

In this section, we will develop both the necessary and sufficient conditions for the equilibrium of the rigid body in Fig. 5–1*a*. As shown, this body is subjected to an external force and couple moment system that is the result of the effects of gravitational, electrical, magnetic, or contact forces caused by adjacent bodies. The internal forces caused by interactions between particles within the body are not shown in this figure because these forces occur in equal but opposite collinear pairs and hence will cancel out, a consequence of Newton's third law.



**Fig. 5–1**

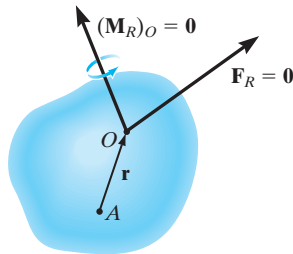
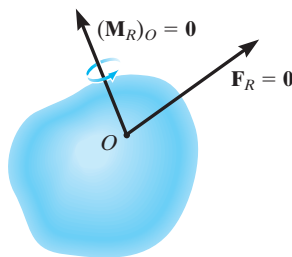
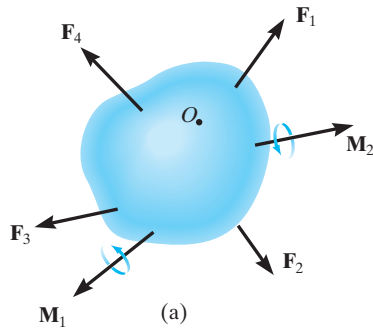


Fig. 5-1 (cont.)

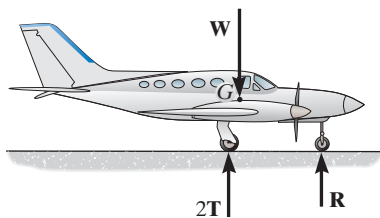


Fig. 5-2

Using the methods of the previous chapter, the force and couple moment system acting on a body can be reduced to an equivalent resultant force and resultant couple moment at any arbitrary point  $O$  on or off the body, Fig. 5-1*b*. If this resultant force and couple moment are both equal to zero, then the body is said to be in **equilibrium**. Mathematically, the equilibrium of a body is expressed as

$$\begin{aligned}\mathbf{F}_R &= \Sigma \mathbf{F} = \mathbf{0} \\ (\mathbf{M}_R)_O &= \Sigma \mathbf{M}_O = \mathbf{0}\end{aligned}\quad (5-1)$$

The first of these equations states that the sum of the forces acting on the body is equal to *zero*. The second equation states that the sum of the moments of all the forces in the system about point  $O$ , added to all the couple moments, is equal to *zero*. These two equations are not only necessary for equilibrium, they are also sufficient. To show this, consider summing moments about some other point, such as point  $A$  in Fig. 5-1*c*. We require

$$\Sigma \mathbf{M}_A = \mathbf{r} \times \mathbf{F}_R + (\mathbf{M}_R)_O = \mathbf{0}$$

Since  $\mathbf{r} \neq \mathbf{0}$ , this equation is satisfied if Eqs. 5-1 are satisfied, namely  $\mathbf{F}_R = \mathbf{0}$  and  $(\mathbf{M}_R)_O = \mathbf{0}$ .

When applying the equations of equilibrium, we will assume that the body remains rigid. In reality, however, all bodies deform when subjected to loads. Although this is the case, most engineering materials such as steel and concrete are very rigid and so their deformation is usually very small. Therefore, when applying the equations of equilibrium, we can generally assume that the body will remain *rigid* and *not deform* under the applied load without introducing any significant error. This way the direction of the applied forces and their moment arms with respect to a fixed reference remain the same both before and after the body is loaded.

## EQUILIBRIUM IN TWO DIMENSIONS

In the first part of the chapter, we will consider the case where the force system acting on a rigid body lies in or may be projected onto a *single* plane and, furthermore, any couple moments acting on the body are directed perpendicular to this plane. This type of force and couple system is often referred to as a two-dimensional or *coplanar* force system. For example, the airplane in Fig. 5-2 has a plane of symmetry through its center axis, and so the loads acting on the airplane are symmetrical with respect to this plane. Thus, each of the two wing tires will support the same load  $\mathbf{T}$ , which is represented on the side (two-dimensional) view of the plane as  $2\mathbf{T}$ .

## 5.2 Free-Body Diagrams

Successful application of the equations of equilibrium requires a complete specification of *all* the known and unknown external forces that act *on* the body. The best way to account for these forces is to draw a **free-body diagram**. This diagram is a sketch of the outlined shape of the body, which represents it as being *isolated* or “free” from its surroundings, i.e., a “free body.” On this sketch it is necessary to show *all* the forces and couple moments that the surroundings exert *on the body* so that these effects can be accounted for when the equations of equilibrium are applied. *A thorough understanding of how to draw a free-body diagram is of primary importance for solving problems in mechanics.*

**Support Reactions.** Before presenting a formal procedure as to how to draw a free-body diagram, we will first consider the various types of reactions that occur at supports and points of contact between bodies subjected to coplanar force systems. As a general rule,

- A support prevents the translation of a body in a given direction by exerting a force on the body in the opposite direction.
- A support prevents the rotation of a body in a given direction by exerting a couple moment on the body in the opposite direction.

For example, let us consider three ways in which a horizontal member, such as a beam, is supported at its end. One method consists of a *roller* or cylinder, Fig. 5–3a. Since this support only prevents the beam from *translating* in the vertical direction, the roller will only exert a *force* on the beam in this direction, Fig. 5–3b.

The beam can be supported in a more restrictive manner by using a *pin*, Fig. 5–3c. The pin passes through a hole in the beam and two leaves which are fixed to the ground. Here the pin can prevent *translation* of the beam in *any direction*  $\phi$ , Fig. 5–3d, and so the pin must exert a *force*  $\mathbf{F}$  on the beam in the opposite direction. For purposes of analysis, it is generally easier to represent this resultant force  $\mathbf{F}$  by its two rectangular components  $\mathbf{F}_x$  and  $\mathbf{F}_y$ , Fig. 5–3e. If  $F_x$  and  $F_y$  are known, then  $F$  and  $\phi$  can be calculated.

The most restrictive way to support the beam would be to use a *fixed support* as shown in Fig. 5–3f. This support will prevent both *translation* and *rotation* of the beam. To do this a *force* and *couple moment* must be developed on the beam at its point of connection, Fig. 5–3g. As in the case of the pin, the force is usually represented by its rectangular components  $\mathbf{F}_x$  and  $\mathbf{F}_y$ .

Table 5–1 lists other common types of supports for bodies subjected to coplanar force systems. (In all cases the angle  $\theta$  is assumed to be known.) Carefully study each of the symbols used to represent these supports and the types of reactions they exert on their contacting members.

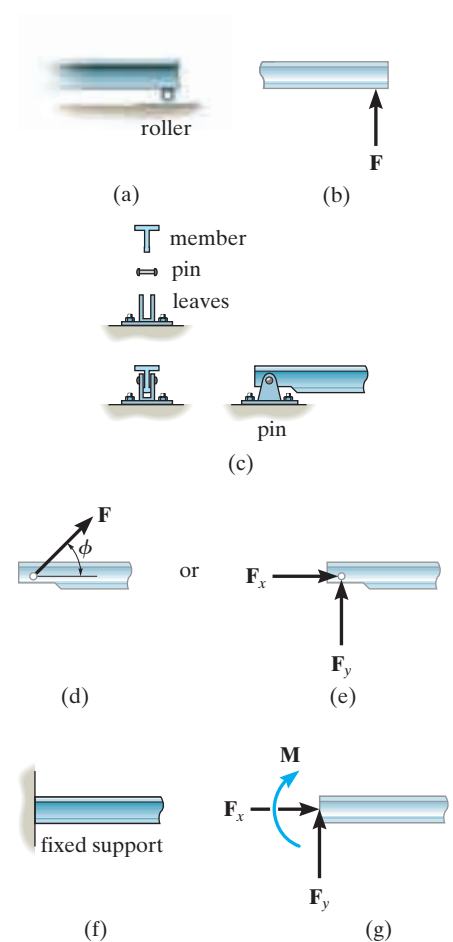
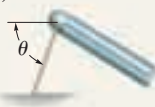
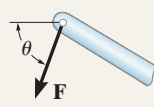
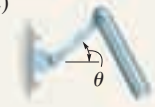
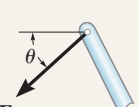
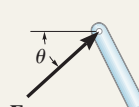



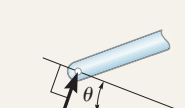

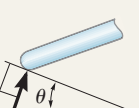
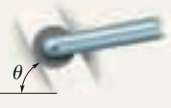
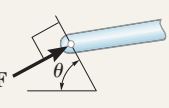
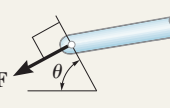
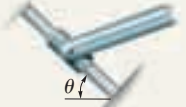
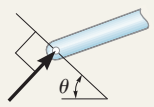
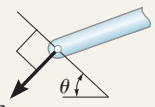
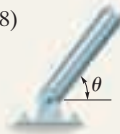
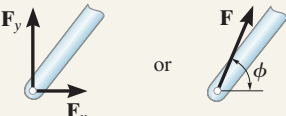

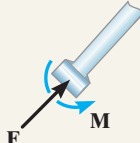

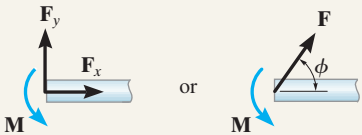


Fig. 5-3

TABLE 5-1 Supports for Rigid Bodies Subjected to Two-Dimensional Force Systems

Types of Connection	Reaction	Number of Unknowns
(1)  cable		One unknown. The reaction is a tension force which acts away from the member in the direction of the cable.
(2)  weightless link	 or 	One unknown. The reaction is a force which acts along the axis of the link.
(3)  roller		One unknown. The reaction is a force which acts perpendicular to the surface at the point of contact.
(4)  rocker		One unknown. The reaction is a force which acts perpendicular to the surface at the point of contact.
(5)  smooth contacting surface		One unknown. The reaction is a force which acts perpendicular to the surface at the point of contact.
(6)  roller or pin in confined smooth slot	 or 	One unknown. The reaction is a force which acts perpendicular to the slot.
(7)  member pin connected to collar on smooth rod	 or 	One unknown. The reaction is a force which acts perpendicular to the rod.

continued

TABLE 5–1 Continued		
Types of Connection	Reaction	Number of Unknowns
(8)  smooth pin or hinge	 or	Two unknowns. The reactions are two components of force, or the magnitude and direction $\phi$ of the resultant force. Note that $\phi$ and $\theta$ are not necessarily equal [usually not, unless the rod shown is a link as in (2)].
(9)  member fixed connected to collar on smooth rod		Two unknowns. The reactions are the couple moment and the force which acts perpendicular to the rod.
(10)  fixed support	 or	Three unknowns. The reactions are the couple moment and the two force components, or the couple moment and the magnitude and direction $\phi$ of the resultant force.

Typical examples of actual supports are shown in the following sequence of photos. The numbers refer to the connection types in Table 5–1.



The cable exerts a force on the bracket in the direction of the cable. (1)



The rocker support for this bridge girder allows horizontal movement so the bridge is free to expand and contract due to a change in temperature. (4)  
(© Russell C. Hibbeler)

This concrete girder rests on the ledge that is assumed to act as a smooth contacting surface. (5) (© Russell C. Hibbeler)

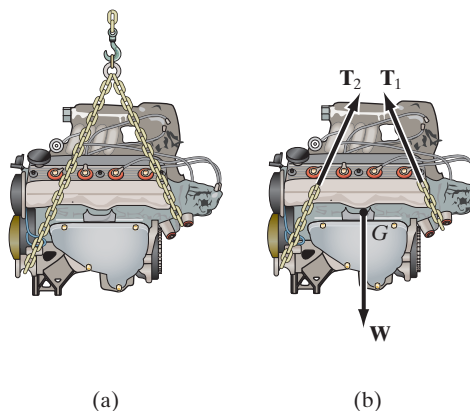


Typical pin support for a beam. (8)  
(© Russell C. Hibbeler)

The floor beams of this building are welded together and thus form fixed connections. (10)  
(© Russell C. Hibbeler)



**Internal Forces.** As stated in Sec. 5.1, the internal forces that act between adjacent particles in a body always occur in collinear pairs such that they have the same magnitude and act in opposite directions (Newton's third law). Since these forces cancel each other, they will not create an *external effect* on the body. It is for this reason that the internal forces should not be included on the free-body diagram if the entire body is to be considered. For example, the engine shown in Fig. 5-4a has a free-body diagram shown in Fig. 5-4b. The internal forces between all its connected parts, such as the screws and bolts, will cancel out because they form equal and opposite collinear pairs. Only the external forces  $\mathbf{T}_1$  and  $\mathbf{T}_2$ , exerted by the chains and the engine weight  $\mathbf{W}$ , are shown on the free-body diagram.



**Fig. 5-4**

**Weight and the Center of Gravity.** When a body is within a gravitational field, then each of its particles has a specified weight. It was shown in Sec. 4.8 that such a system of forces can be reduced to a single resultant force acting through a specified point. We refer to this force resultant as the *weight*  $\mathbf{W}$  of the body and to the location of its point of application as the *center of gravity*. The methods used for its determination will be developed in Chapter 9.

In the examples and problems that follow, if the weight of the body is important for the analysis, this force will be reported in the problem statement. Also, when the body is *uniform* or made from the same material, the center of gravity will be located at the body's *geometric center* or *centroid*; however, if the body consists of a nonuniform distribution of material, or has an unusual shape, then the location of its center of gravity  $G$  will be given.

**Idealized Models.** When an engineer performs a force analysis of any object, he or she considers a corresponding analytical or idealized model that gives results that approximate as closely as possible the actual situation. To do this, careful choices have to be made so that selection of the type of supports, the material behavior, and the object's dimensions can be justified. This way one can feel confident that any

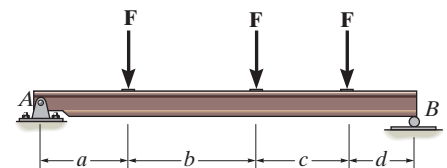


design or analysis will yield results which can be trusted. In complex cases this process may require developing several different models of the object that must be analyzed. In any case, this selection process requires both skill and experience.

The following two cases illustrate what is required to develop a proper model. In Fig. 5–5*a*, the steel beam is to be used to support the three roof joists of a building. For a force analysis it is reasonable to assume the material (steel) is rigid since only very small deflections will occur when the beam is loaded. A bolted connection at *A* will allow for any slight rotation that occurs here when the load is applied, and so a *pin* can be considered for this support. At *B* a *roller* can be considered since this support offers no resistance to horizontal movement. Building code is used to specify the roof loading *A* so that the joist loads **F** can be calculated. These forces will be larger than any actual loading on the beam since they account for extreme loading cases and for dynamic or vibrational effects. Finally, the weight of the beam is generally neglected when it is small compared to the load the beam supports. The idealized model of the beam is therefore shown with average dimensions *a*, *b*, *c*, and *d* in Fig. 5–5*b*.



(a)

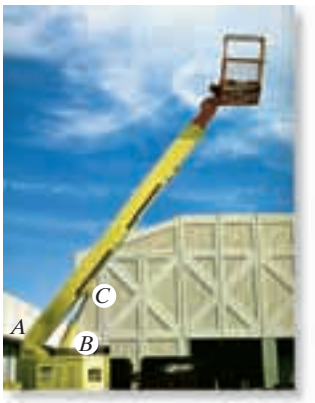


(b)

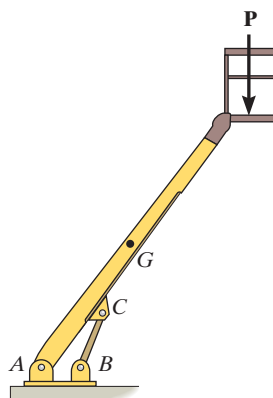
**Fig. 5–5** (© Russell C. Hibbeler)

As a second case, consider the lift boom in Fig. 5–6*a*. By inspection, it is supported by a pin at *A* and by the hydraulic cylinder *BC*, which can be approximated as a weightless link. The material can be assumed rigid, and with its density known, the weight of the boom and the location of its center of gravity *G* are determined. When a design loading **P** is specified, the idealized model shown in Fig. 5–6*b* can be used for a force analysis. Average dimensions (not shown) are used to specify the location of the loads and the supports.

Idealized models of specific objects will be given in some of the examples throughout the text. It should be realized, however, that each case represents the reduction of a practical situation using simplifying assumptions like the ones illustrated here.



(a)



(b)

**Fig. 5–6** (© Russell C. Hibbeler)



## Important Points

- No equilibrium problem should be solved without *first drawing the free-body diagram*, so as to account for all the forces and couple moments that act on the body.
- If a support *prevents translation* of a body in a particular direction, then the support, when it is removed, exerts a *force* on the body in that direction.
- If *rotation is prevented*, then the support, when it is removed, exerts a *couple moment* on the body.
- Study Table 5–1.
- Internal forces are *never shown* on the free-body diagram since they occur in equal but opposite collinear pairs and therefore cancel out.
- The weight of a body is an external force, and its effect is represented by a single resultant force acting through the body's center of gravity  $G$ .
- *Couple moments* can be placed anywhere on the free-body diagram since they are *free vectors*. *Forces* can act at any point along their lines of action since they are *sliding vectors*.

## Procedure for Analysis

To construct a free-body diagram for a rigid body or any group of bodies considered as a single system, the following steps should be performed:

### Draw Outlined Shape.

Imagine the body to be *isolated* or cut “free” from its constraints and connections and draw (sketch) its outlined shape. Be sure to *remove all the supports* from the body.

### Show All Forces and Couple Moments.

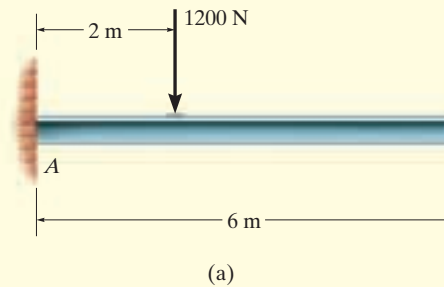
Identify all the known and unknown *external forces* and couple moments that *act on the body*. Those generally encountered are due to (1) applied loadings, (2) reactions occurring at the supports or at points of contact with other bodies (see Table 5–1), and (3) the weight of the body. To account for all these effects, it may help to trace over the boundary, carefully noting each force or couple moment acting on it.

### Identify Each Loading and Give Dimensions.

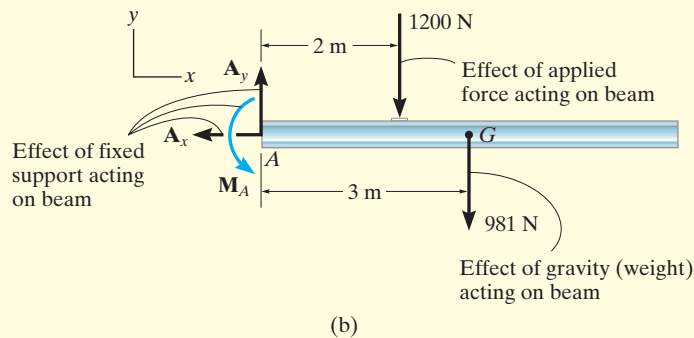
The forces and couple moments that are known should be labeled with their proper magnitudes and directions. Letters are used to represent the magnitudes and direction angles of forces and couple moments that are unknown. Establish an  $x, y$  coordinate system so that these unknowns,  $A_x, A_y$ , etc., can be identified. Finally, indicate the dimensions of the body necessary for calculating the moments of forces.

**EXAMPLE 5.1**

Draw the free-body diagram of the uniform beam shown in Fig. 5–7*a*. The beam has a mass of 100 kg.

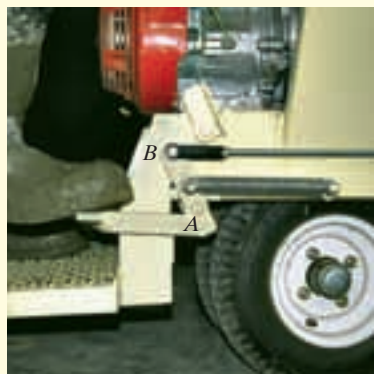
**SOLUTION**

The free-body diagram of the beam is shown in Fig. 5–7*b*. Since the support at  $A$  is fixed, the wall exerts three reactions *on the beam*, denoted as  $A_x$ ,  $A_y$ , and  $M_A$ . The magnitudes of these reactions are *unknown*, and their sense has been *assumed*. The weight of the beam,  $W = 100(9.81) \text{ N} = 981 \text{ N}$ , acts through the beam's center of gravity  $G$ , which is 3 m from  $A$  since the beam is uniform.

**Fig. 5–7**

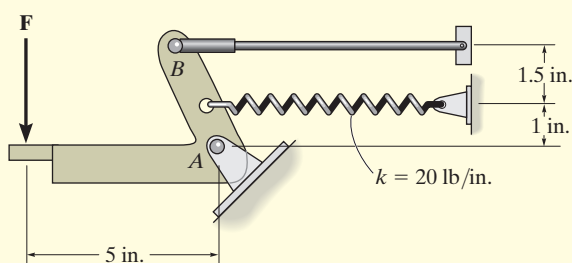
## EXAMPLE 5.2

Draw the free-body diagram of the foot lever shown in Fig. 5–8a. The operator applies a vertical force to the pedal so that the spring is stretched 1.5 in. and the force on the link at  $B$  is 20 lb.

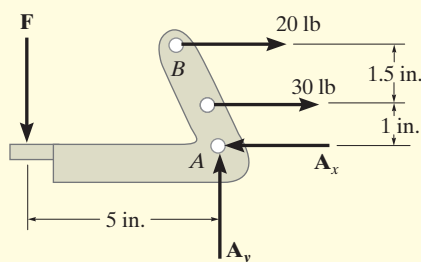


(a)

Fig. 5–8 (© Russell C. Hibbeler)



(b)



(c)

## SOLUTION

By inspection of the photo the lever is loosely bolted to the frame at  $A$  and so this bolt acts as a pin. (See (8) in Table 5–1.) Although not shown here the link at  $B$  is pinned at both ends and so it is like (2) in Table 5–1. After making the proper measurements, the idealized model of the lever is shown in Fig. 5–8b. From this, the free-body diagram is shown in Fig. 5–8c. Since the pin at  $A$  is *removed*, it exerts force components  $A_x$  and  $A_y$  on the lever. The link exerts a force of 20 lb, acting in the direction of the link. In addition the spring also exerts a horizontal force on the lever. If the stiffness is measured and found to be  $k = 20 \text{ lb/in.}$ , then since the stretch  $s = 1.5 \text{ in.}$ , using Eq. 3–2,  $F_s = ks = 20 \text{ lb/in.} (1.5 \text{ in.}) = 30 \text{ lb.}$  Finally, the operator's shoe applies a vertical force of  $F$  on the pedal. The dimensions of the lever are also shown on the free-body diagram, since this information will be useful when calculating the moments of the forces. As usual, the senses of the unknown forces at  $A$  have been assumed. The correct senses will become apparent after solving the equilibrium equations.

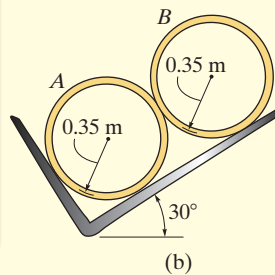
**EXAMPLE 5.3**

Two smooth pipes, each having a mass of 300 kg, are supported by the forked tines of the tractor in Fig. 5–9a. Draw the free-body diagrams for each pipe and both pipes together.

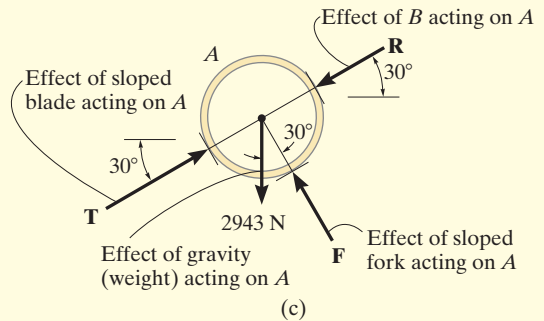


(a)

(© Russell C. Hibbeler)



(b)



(c)

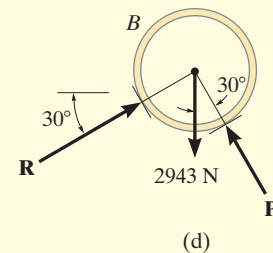
**SOLUTION**

The idealized model from which we must draw the free-body diagrams is shown in Fig. 5–9b. Here the pipes are identified, the dimensions have been added, and the physical situation reduced to its simplest form.

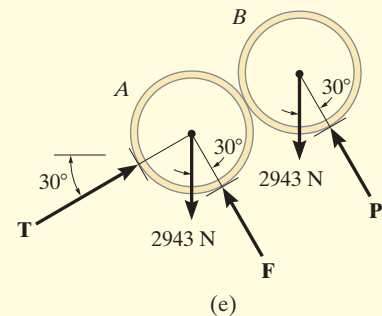
Removing the surfaces of contact, the free-body diagram for pipe A is shown in Fig. 5–9c. Its weight is  $W = 300(9.81) \text{ N} = 2943 \text{ N}$ . Assuming all contacting surfaces are *smooth*, the reactive forces  $\mathbf{T}$ ,  $\mathbf{F}$ ,  $\mathbf{R}$  act in a direction *normal* to the tangent at their surfaces of contact.

The free-body diagram of the isolated pipe B is shown in Fig. 5–9d. Can you identify each of the three forces acting *on this pipe*? In particular, note that  $\mathbf{R}$ , representing the force of A on B, Fig. 5–9d, is equal and opposite to  $\mathbf{R}$  representing the force of B on A, Fig. 5–9c. This is a consequence of Newton's third law of motion.

The free-body diagram of both pipes combined (“system”) is shown in Fig. 5–9e. Here the contact force  $\mathbf{R}$ , which acts between A and B, is considered as an *internal* force and hence is not shown on the free-body diagram. That is, it represents a pair of equal but opposite collinear forces which cancel each other.



(d)

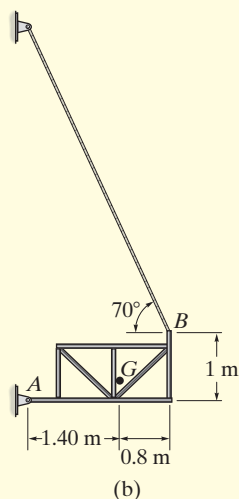


(e)

**Fig. 5–9**

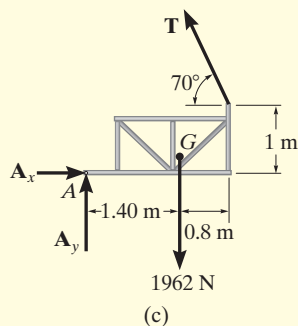
## EXAMPLE 5.4

Draw the free-body diagram of the unloaded platform that is suspended off the edge of the oil rig shown in Fig. 5–10a. The platform has a mass of 200 kg.



(a)

Fig. 5–10 (© Russell C. Hibbeler)



## SOLUTION

The idealized model of the platform will be considered in two dimensions because by observation the loading and the dimensions are all symmetrical about a vertical plane passing through its center, Fig. 5–10b. The connection at  $A$  is considered to be a pin, and the cable supports the platform at  $B$ . The direction of the cable and average dimensions of the platform are listed, and the center of gravity  $G$  has been determined. It is from this model that we have drawn the free-body diagram shown in Fig. 5–10c. The platform's weight is  $200(9.81) = 1962$  N. The supports have been *removed*, and the force components  $A_x$  and  $A_y$  along with the cable force  $T$  represent the reactions that *both* pins and *both* cables exert on the platform, Fig. 5–10a. As a result, half their magnitudes are developed on each side of the platform.

**PROBLEMS**

**5-1.** Draw the free-body diagram for the following problems.

- a) The cantilevered beam in Prob. 5-10.
- b) The beam in Prob. 5-11.
- c) The beam in Prob. 5-12.
- d) The beam in Prob. 5-14.

**5-2.** Draw the free-body diagram for the following problems.

- a) The truss in Prob. 5-15.
- b) The beam in Prob. 5-16.
- c) The man and load in Prob. 5-17.
- d) The beam in Prob. 5-18.

**5-3.** Draw the free-body diagram for the following problems.

- a) The man and beam in Prob. 5-19.
- b) The rod in Prob. 5-20.
- c) The rod in Prob. 5-21.
- d) The beam in Prob. 5-22.

**\*5-4.** Draw the free-body diagram for the following problems.

- a) The beam in Prob. 5-25.
- b) The crane and boom in Prob. 5-26.
- c) The bar in Prob. 5-27.
- d) The rod in Prob. 5-28.

**5-5.** Draw the free-body diagram for the following problems.

- a) The boom in Prob. 5-32.
- b) The jib crane in Prob. 5-33.
- c) The smooth pipe in Prob. 5-35.
- d) The beam in Prob. 5-36.

**5-6.** Draw the free-body diagram for the following problems.

- a) The jib crane in Prob. 5-37.
- b) The bar in Prob. 5-39.
- c) The bulkhead in Prob. 5-41.
- d) The boom in Prob. 5-42.

**5-7.** Draw the free-body diagram for the following problems.

- a) The rod in Prob. 5-44.
- b) The hand truck and load when it is lifted in Prob. 5-45.
- c) The beam in Prob. 5-47.
- d) The cantilever footing in Prob. 5-51.

**\*5-8.** Draw the free-body diagram for the following problems.

- a) The beam in Prob. 5-52.
- b) The boy and diving board in Prob. 5-53.
- c) The rod in Prob. 5-54.
- d) The rod in Prob. 5-56.

**5-9.** Draw the free-body diagram for the following problems.

- a) The beam in Prob. 5-57.
- b) The rod in Prob. 5-59.
- c) The bar in Prob. 5-60.

## 5.3 Equations of Equilibrium

In Sec. 5.1 we developed the two equations which are both necessary and sufficient for the equilibrium of a rigid body, namely,  $\Sigma \mathbf{F} = \mathbf{0}$  and  $\Sigma \mathbf{M}_O = \mathbf{0}$ . When the body is subjected to a system of forces, which all lie in the  $x$ - $y$  plane, then the forces can be resolved into their  $x$  and  $y$  components. Consequently, the conditions for equilibrium in two dimensions are

$$\begin{aligned}\Sigma F_x &= 0 \\ \Sigma F_y &= 0 \\ \Sigma M_O &= 0\end{aligned}\quad (5-2)$$

Here  $\Sigma F_x$  and  $\Sigma F_y$  represent, respectively, the algebraic sums of the  $x$  and  $y$  components of all the forces acting on the body, and  $\Sigma M_O$  represents the algebraic sum of the couple moments and the moments of all the force components about the  $z$  axis, which is perpendicular to the  $x$ - $y$  plane and passes through the arbitrary point  $O$ .

**Alternative Sets of Equilibrium Equations.** Although Eqs. 5-2 are *most often* used for solving coplanar equilibrium problems, two *alternative* sets of three independent equilibrium equations may also be used. One such set is

$$\begin{aligned}\Sigma F_x &= 0 \\ \Sigma M_A &= 0 \\ \Sigma M_B &= 0\end{aligned}\quad (5-3)$$

When using these equations it is required that a line passing through points  $A$  and  $B$  is *not parallel* to the  $y$  axis. To prove that Eqs. 5-3 provide the *conditions* for equilibrium, consider the free-body diagram of the plate shown in Fig. 5-11a. Using the methods of Sec. 4.7, all the forces on the free-body diagram may be replaced by an equivalent resultant force  $\mathbf{F}_R = \Sigma \mathbf{F}$ , acting at point  $A$ , and a resultant couple moment  $(\mathbf{M}_R)_A = \Sigma \mathbf{M}_A$ , Fig. 5-11b. If  $\Sigma M_A = 0$  is satisfied, it is necessary that  $(\mathbf{M}_R)_A = \mathbf{0}$ . Furthermore, in order that  $\mathbf{F}_R$  satisfy  $\Sigma F_x = 0$ , it must have *no component* along the  $x$  axis, and therefore  $\mathbf{F}_R$  must be parallel to the  $y$  axis, Fig. 5-11c. Finally, if it is required that  $\Sigma M_B = 0$ , where  $B$  does not lie on the line of action of  $\mathbf{F}_R$ , then  $\mathbf{F}_R = \mathbf{0}$ . Since Eqs. 5-3 show that both of these resultants are zero, indeed the body in Fig. 5-11a must be in equilibrium.

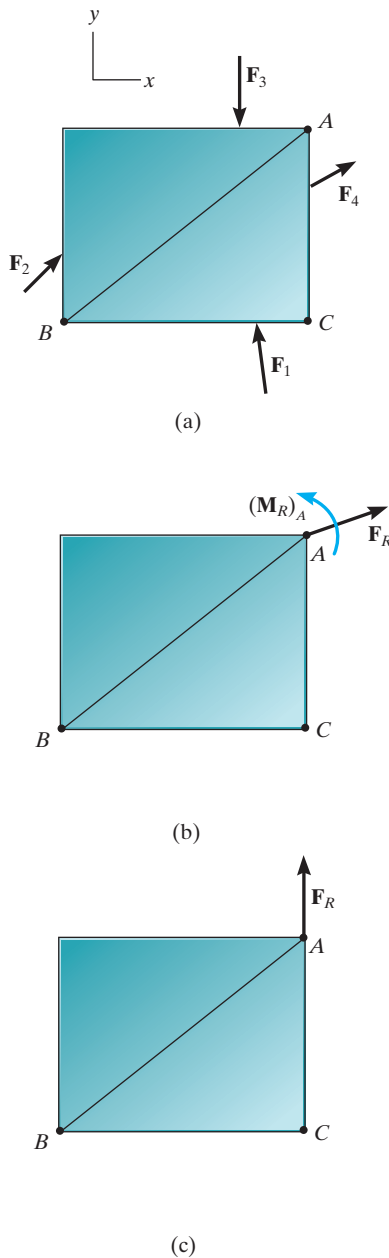


Fig. 5-11

A second alternative set of equilibrium equations is

$$\begin{aligned}\Sigma M_A &= 0 \\ \Sigma M_B &= 0 \\ \Sigma M_C &= 0\end{aligned}\tag{5-4}$$

Here it is necessary that points  $A$ ,  $B$ , and  $C$  do not lie on the same line. To prove that these equations, when satisfied, ensure equilibrium, consider again the free-body diagram in Fig. 5-11*b*. If  $\Sigma M_A = 0$  is to be satisfied, then  $(\mathbf{M}_R)_A = \mathbf{0}$ .  $\Sigma M_C = 0$  is satisfied if the line of action of  $\mathbf{F}_R$  passes through point  $C$  as shown in Fig. 5-11*c*. Finally, if we require  $\Sigma M_B = 0$ , it is necessary that  $\mathbf{F}_R = \mathbf{0}$ , and so the plate in Fig. 5-11*a* must then be in equilibrium.

## Procedure for Analysis

Coplanar force equilibrium problems for a rigid body can be solved using the following procedure.

### Free-Body Diagram.

- Establish the  $x$ ,  $y$  coordinate axes in any suitable orientation.
- Remove all supports and draw an outlined shape of the body.
- Show all the forces and couple moments acting on the body.
- Label all the loadings and specify their directions relative to the  $x$  or  $y$  axis. The sense of a force or couple moment having an *unknown* magnitude but known line of action can be *assumed*.
- Indicate the dimensions of the body necessary for computing the moments of forces.

### Equations of Equilibrium.

- Apply the moment equation of equilibrium,  $\Sigma M_O = 0$ , about a point ( $O$ ) that lies at the intersection of the lines of action of two unknown forces. In this way, the moments of these unknowns are zero about  $O$ , and a *direct solution* for the third unknown can be determined.
- When applying the force equilibrium equations,  $\Sigma F_x = 0$  and  $\Sigma F_y = 0$ , orient the  $x$  and  $y$  axes along lines that will provide the simplest resolution of the forces into their  $x$  and  $y$  components.
- If the solution of the equilibrium equations yields a negative scalar for a force or couple moment magnitude, this indicates that the sense is opposite to that which was assumed on the free-body diagram.



## EXAMPLE 5.5

Determine the horizontal and vertical components of reaction on the beam caused by the pin at  $B$  and the rocker at  $A$  as shown in Fig. 5–12a. Neglect the weight of the beam.

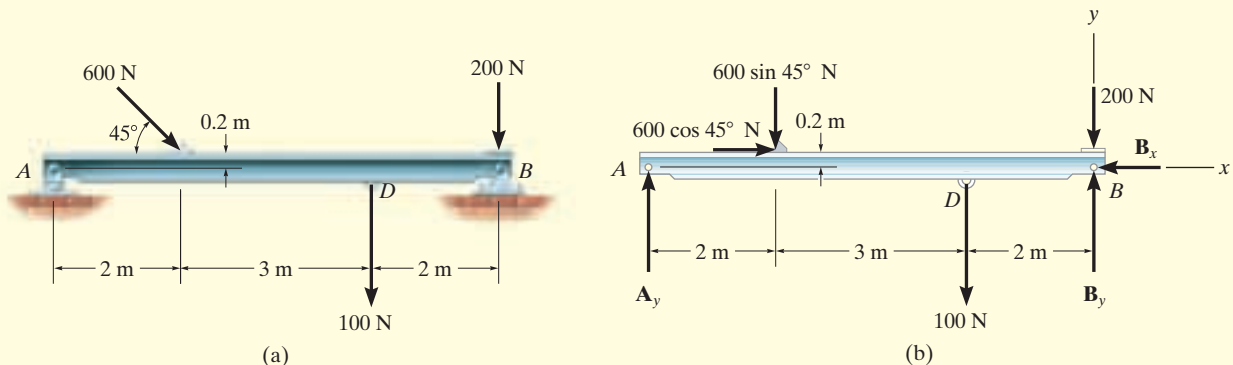


Fig. 5–12

## SOLUTION

**Free-Body Diagram.** The supports are *removed*, and the free-body diagram of the beam is shown in Fig. 5–12b. (See Example 5.1.) For simplicity, the 600-N force is represented by its  $x$  and  $y$  components as shown in Fig. 5–12b.

**Equations of Equilibrium.** Summing forces in the  $x$  direction yields

$$\rightarrow \Sigma F_x = 0; \quad 600 \cos 45^\circ \text{ N} - B_x = 0$$

$$B_x = 424 \text{ N} \quad \text{Ans.}$$

A direct solution for  $A_y$  can be obtained by applying the moment equation  $\Sigma M_B = 0$  about point  $B$ .

$$\begin{aligned} \zeta + \Sigma M_B = 0; \quad & 100 \text{ N}(2 \text{ m}) + (600 \sin 45^\circ \text{ N})(5 \text{ m}) \\ & - (600 \cos 45^\circ \text{ N})(0.2 \text{ m}) - A_y(7 \text{ m}) = 0 \\ A_y = & 319 \text{ N} \quad \text{Ans.} \end{aligned}$$

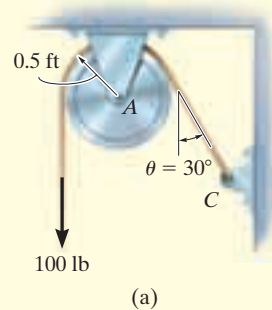
Summing forces in the  $y$  direction, using this result, gives

$$\begin{aligned} + \uparrow \Sigma F_y = 0; \quad & 319 \text{ N} - 600 \sin 45^\circ \text{ N} - 100 \text{ N} - 200 \text{ N} + B_y = 0 \\ B_y = & 405 \text{ N} \quad \text{Ans.} \end{aligned}$$

**NOTE:** Remember, the support forces in Fig. 5–12b are the result of pins that *act on the beam*. The opposite forces act on the pins. For example, Fig. 5–12c shows the equilibrium of the pin at  $A$  and the rocker.

**EXAMPLE 5.6**

The cord shown in Fig. 5–13a supports a force of 100 lb and wraps over the frictionless pulley. Determine the tension in the cord at  $C$  and the horizontal and vertical components of reaction at pin  $A$ .

**Fig. 5–13****SOLUTION**

**Free-Body Diagrams.** The free-body diagrams of the cord and pulley are shown in Fig. 5–13b. Note that the principle of action, equal but opposite reaction must be carefully observed when drawing each of these diagrams: the cord exerts an unknown load distribution  $p$  on the pulley at the contact surface, whereas the pulley exerts an equal but opposite effect on the cord. For the solution, however, it is simpler to *combine* the free-body diagrams of the pulley and this portion of the cord, so that the distributed load becomes *internal* to this “system” and is therefore eliminated from the analysis, Fig. 5–13c.

**Equations of Equilibrium.** Summing moments about point  $A$  to eliminate  $A_x$  and  $A_y$ , Fig. 5–13c, we have

$$\zeta + \Sigma M_A = 0; \quad 100 \text{ lb} (0.5 \text{ ft}) - T (0.5 \text{ ft}) = 0$$

$$T = 100 \text{ lb}$$

*Ans.*

Using this result,

$$\rightarrow \Sigma F_x = 0; \quad -A_x + 100 \sin 30^\circ \text{ lb} = 0$$

$$A_x = 50.0 \text{ lb}$$

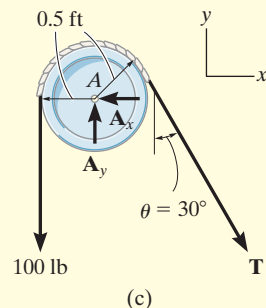
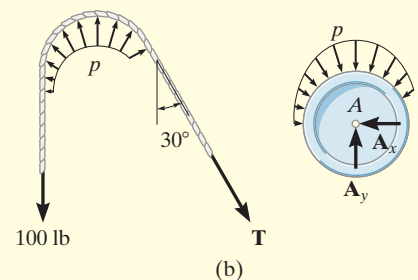
*Ans.*

$$+\uparrow \Sigma F_y = 0; \quad A_y - 100 \text{ lb} - 100 \cos 30^\circ \text{ lb} = 0$$

$$A_y = 187 \text{ lb}$$

*Ans.*

**NOTE:** From the moment equation, it is seen that the tension remains *constant* as the cord passes over the pulley. (This of course is true for *any* angle  $\theta$  at which the cord is directed and for *any* radius  $r$  of the pulley.)



## EXAMPLE 5.7

The member shown in Fig. 5–14a is pin connected at  $A$  and rests against a smooth support at  $B$ . Determine the horizontal and vertical components of reaction at the pin  $A$ .

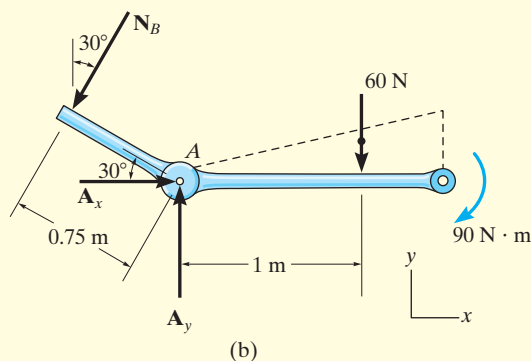
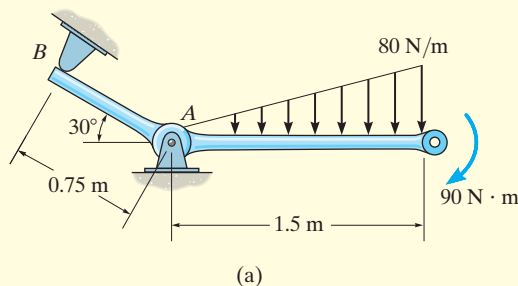


Fig. 5–14

## SOLUTION

**Free-Body Diagram.** As shown in Fig. 5–14b, the supports are removed and the reaction  $N_B$  is perpendicular to the member at  $B$ . Also, horizontal and vertical components of reaction are represented at  $A$ . The resultant of the distributed loading is  $\frac{1}{2}(1.5 \text{ m})(80 \text{ N/m}) = 60 \text{ N}$ . It acts through the centroid of the triangle, 1 m from  $A$  as shown.

**Equations of Equilibrium.** Summing moments about  $A$ , we obtain a direct solution for  $N_B$ ,

$$\zeta + \Sigma M_A = 0; \quad -90 \text{ N} \cdot \text{m} - 60 \text{ N}(1 \text{ m}) + N_B(0.75 \text{ m}) = 0$$

$$N_B = 200 \text{ N}$$

Using this result,

$$\rightarrow \Sigma F_x = 0; \quad A_x - 200 \sin 30^\circ \text{ N} = 0$$

$$A_x = 100 \text{ N} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad A_y - 200 \cos 30^\circ \text{ N} - 60 \text{ N} = 0$$

$$A_y = 233 \text{ N} \quad \text{Ans.}$$

**EXAMPLE 5.8**

The box wrench in Fig. 5–15*a* is used to tighten the bolt at *A*. If the wrench does not turn when the load is applied to the handle, determine the torque or moment applied to the bolt and the force of the wrench on the bolt.

**SOLUTION**

**Free-Body Diagram.** The free-body diagram for the wrench is shown in Fig. 5–15*b*. Since the bolt acts as a “fixed support,” when it is removed, it exerts force components  $A_x$  and  $A_y$  and a moment  $M_A$  on the wrench at *A*.

**Equations of Equilibrium.**

$$\begin{aligned} \rightarrow \Sigma F_x = 0; \quad A_x - 52\left(\frac{5}{13}\right) \text{ N} + 30 \cos 60^\circ \text{ N} &= 0 \\ A_x &= 5.00 \text{ N} \end{aligned} \quad \text{Ans.}$$

$$\begin{aligned} + \uparrow \Sigma F_y = 0; \quad A_y - 52\left(\frac{12}{13}\right) \text{ N} - 30 \sin 60^\circ \text{ N} &= 0 \\ A_y &= 74.0 \text{ N} \end{aligned} \quad \text{Ans.}$$

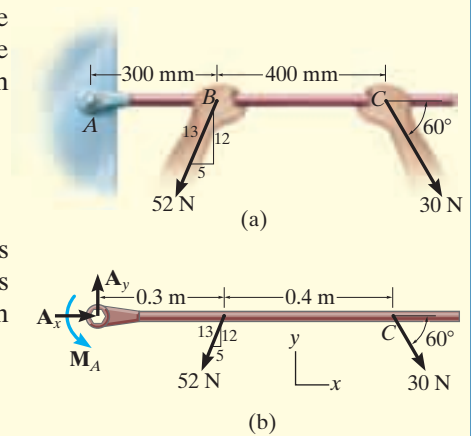
$$\begin{aligned} \zeta + \Sigma M_A = 0; \quad M_A - \left[ 52\left(\frac{12}{13}\right) \text{ N} \right] (0.3 \text{ m}) - (30 \sin 60^\circ \text{ N})(0.7 \text{ m}) &= 0 \\ M_A &= 32.6 \text{ N} \cdot \text{m} \end{aligned} \quad \text{Ans.}$$

Note that  $M_A$  must be *included* in this moment summation. This couple moment is a free vector and represents the twisting resistance of the bolt on the wrench. By Newton’s third law, the wrench exerts an equal but opposite moment or torque on the bolt. Furthermore, the resultant force on the wrench is

$$F_A = \sqrt{(5.00)^2 + (74.0)^2} = 74.1 \text{ N} \quad \text{Ans.}$$

**NOTE:** Although only *three* independent equilibrium equations can be written for a rigid body, it is a good practice to *check* the calculations using a fourth equilibrium equation. For example, the above computations may be verified in part by summing moments about point *C*:

$$\begin{aligned} \zeta + \Sigma M_C = 0; \quad \left[ 52\left(\frac{12}{13}\right) \text{ N} \right] (0.4 \text{ m}) + 32.6 \text{ N} \cdot \text{m} - 74.0 \text{ N}(0.7 \text{ m}) &= 0 \\ 19.2 \text{ N} \cdot \text{m} + 32.6 \text{ N} \cdot \text{m} - 51.8 \text{ N} \cdot \text{m} &= 0 \end{aligned}$$

**Fig. 5–15**

## EXAMPLE 5.9

Determine the horizontal and vertical components of reaction on the member at the pin  $A$ , and the normal reaction at the roller  $B$  in Fig. 5-16a.

## SOLUTION

**Free-Body Diagram.** All the supports are removed and so the free-body diagram is shown in Fig. 5-16b. The pin at  $A$  exerts two components of reaction on the member,  $A_x$  and  $A_y$ .

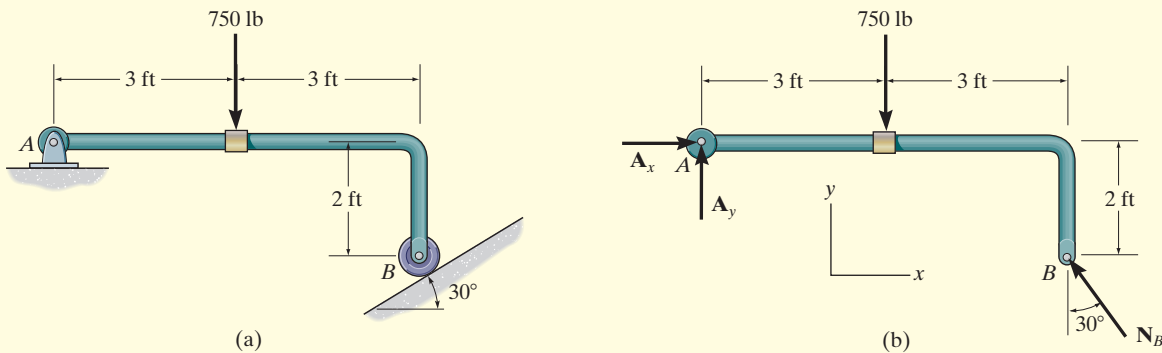


Fig. 5-16

**Equations of Equilibrium.** The reaction  $N_B$  can be obtained *directly* by summing moments about point  $A$ , since  $A_x$  and  $A_y$  produce no moment about  $A$ .

$$\zeta + \sum M_A = 0;$$

$$[N_B \cos 30^\circ](6 \text{ ft}) - [N_B \sin 30^\circ](2 \text{ ft}) - 750 \text{ lb}(3 \text{ ft}) = 0$$

$$N_B = 536.2 \text{ lb} = 536 \text{ lb}$$

Ans.

Using this result,

$$\rightarrow \sum F_x = 0; \quad A_x - (536.2 \text{ lb}) \sin 30^\circ = 0$$

$$A_x = 268 \text{ lb}$$

Ans.

$$+\uparrow \sum F_y = 0; \quad A_y + (536.2 \text{ lb}) \cos 30^\circ - 750 \text{ lb} = 0$$

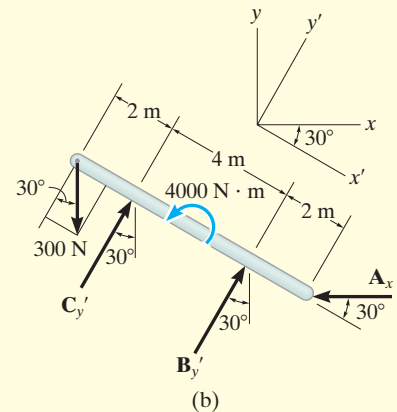
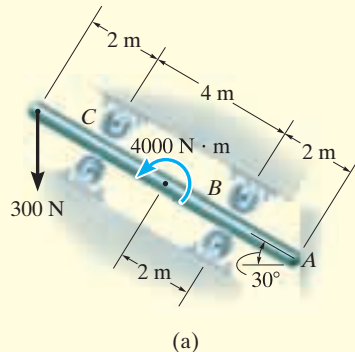
$$A_y = 286 \text{ lb}$$

Ans.

Details of the equilibrium of the pin at  $A$  are shown in Fig. 5-16c.

**EXAMPLE 5.10**

The uniform smooth rod shown in Fig. 5–17a is subjected to a force and couple moment. If the rod is supported at  $A$  by a smooth wall and at  $B$  and  $C$  either at the top or bottom by rollers, determine the reactions at these supports. Neglect the weight of the rod.

**Fig. 5–17****SOLUTION**

**Free-Body Diagram.** Removing the supports as shown in Fig. 5–17b, all the reactions act normal to the surfaces of contact since these surfaces are smooth. The reactions at  $B$  and  $C$  are shown acting in the positive  $y'$  direction. This assumes that only the rollers located on the bottom of the rod are used for support.

**Equations of Equilibrium.** Using the  $x, y$  coordinate system in Fig. 5–17b, we have

$$\rightarrow \Sigma F_x = 0; \quad C_{y'} \sin 30^\circ + B_{y'} \sin 30^\circ - A_x = 0 \quad (1)$$

$$+ \uparrow \Sigma F_y = 0; \quad -300 \text{ N} + C_{y'} \cos 30^\circ + B_{y'} \cos 30^\circ = 0 \quad (2)$$

$$\zeta + \Sigma M_A = 0; \quad -B_{y'}(2 \text{ m}) + 4000 \text{ N} \cdot \text{m} - C_{y'}(6 \text{ m}) + (300 \cos 30^\circ \text{ N})(8 \text{ m}) = 0 \quad (3)$$

When writing the moment equation, it should be noted that the line of action of the force component  $300 \sin 30^\circ \text{ N}$  passes through point  $A$ , and therefore this force is not included in the moment equation.

Solving Eqs. 2 and 3 simultaneously, we obtain

$$B_{y'} = -1000.0 \text{ N} = -1 \text{ kN} \quad \text{Ans.}$$

$$C_{y'} = 1346.4 \text{ N} = 1.35 \text{ kN} \quad \text{Ans.}$$

Since  $B_{y'}$  is a negative scalar, the sense of  $\mathbf{B}_{y'}$  is opposite to that shown on the free-body diagram in Fig. 5–17b. Therefore, the top roller at  $B$  serves as the support rather than the bottom one. Retaining the negative sign for  $B_{y'}$  (Why?) and substituting the results into Eq. 1, we obtain

$$1346.4 \sin 30^\circ \text{ N} + (-1000.0 \sin 30^\circ \text{ N}) - A_x = 0$$

$$A_x = 173 \text{ N} \quad \text{Ans.}$$

## EXAMPLE 5.11



(a)

(© Russell C. Hibbeler)

The uniform truck ramp shown in Fig. 5–18a has a weight of 400 lb and is pinned to the body of the truck at each side and held in the position shown by the two side cables. Determine the tension in the cables.

## SOLUTION

The idealized model of the ramp, which indicates all necessary dimensions and supports, is shown in Fig. 5–18b. Here the center of gravity is located at the midpoint since the ramp is considered to be uniform.

**Free-Body Diagram.** Removing the supports from the idealized model, the ramp's free-body diagram is shown in Fig. 5–18c.

**Equations of Equilibrium.** Summing moments about point  $A$  will yield a direct solution for the cable tension. Using the principle of moments, there are several ways of determining the moment of  $\mathbf{T}$  about  $A$ . If we use  $x$  and  $y$  components, with  $\mathbf{T}$  applied at  $B$ , we have

$$\begin{aligned}\zeta + \Sigma M_A &= 0; & -T \cos 20^\circ(7 \sin 30^\circ \text{ ft}) + T \sin 20^\circ(7 \cos 30^\circ \text{ ft}) \\ & & + 400 \text{ lb}(5 \cos 30^\circ \text{ ft}) = 0 \\ T &= 1425 \text{ lb}\end{aligned}$$

We can also determine the moment of  $\mathbf{T}$  about  $A$  by resolving it into components along and perpendicular to the ramp at  $B$ . Then the moment of the component along the ramp will be zero about  $A$ , so that

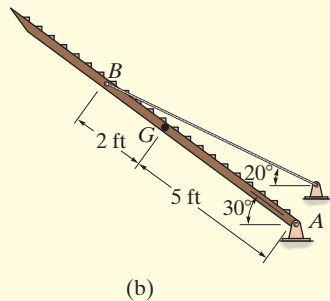
$$\begin{aligned}\zeta + \Sigma M_A &= 0; & -T \sin 10^\circ(7 \text{ ft}) + 400 \text{ lb}(5 \cos 30^\circ \text{ ft}) = 0 \\ T &= 1425 \text{ lb}\end{aligned}$$

Since there are two cables supporting the ramp,

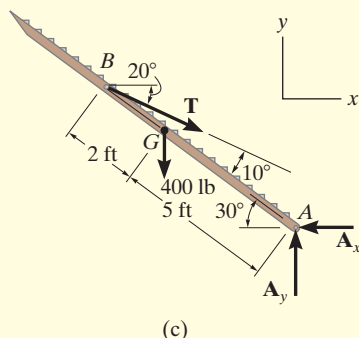
$$T' = \frac{T}{2} = 712 \text{ lb}$$

*Ans.*

**NOTE:** As an exercise, show that  $A_x = 1339 \text{ lb}$  and  $A_y = 887 \text{ lb}$ .



(b)

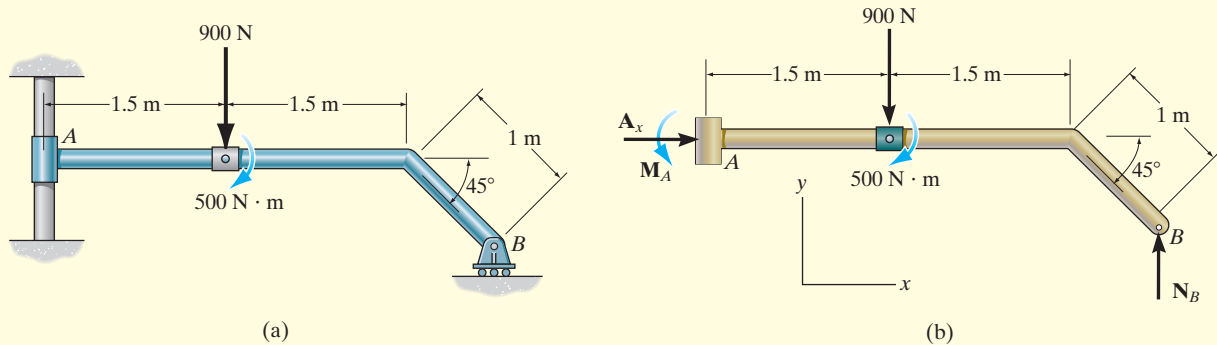


(c)

Fig. 5–18

**EXAMPLE 5.12**

Determine the support reactions on the member in Fig. 5–19*a*. The collar at *A* is fixed to the member and can slide vertically along the vertical shaft.

**Fig. 5–19****SOLUTION**

**Free-Body Diagram.** Removing the supports, the free-body diagram of the member is shown in Fig. 5–19*b*. The collar exerts a horizontal force  $A_x$  and moment  $M_A$  on the member. The reaction  $N_B$  of the roller on the member is vertical.

**Equations of Equilibrium.** The forces  $A_x$  and  $N_B$  can be determined directly from the force equations of equilibrium.

$$+\rightarrow \Sigma F_x = 0; \quad A_x = 0 \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad N_B - 900 \text{ N} = 0$$

$$N_B = 900 \text{ N} \quad \text{Ans.}$$

The moment  $M_A$  can be determined by summing moments either about point *A* or point *B*.

$$\zeta + \Sigma M_A = 0;$$

$$M_A - 900 \text{ N}(1.5 \text{ m}) - 500 \text{ N} \cdot \text{m} + 900 \text{ N}[3 \text{ m} + (1 \text{ m}) \cos 45^\circ] = 0$$

$$M_A = -1486 \text{ N} \cdot \text{m} = 1.49 \text{ kN} \cdot \text{m} \curvearrowright \quad \text{Ans.}$$

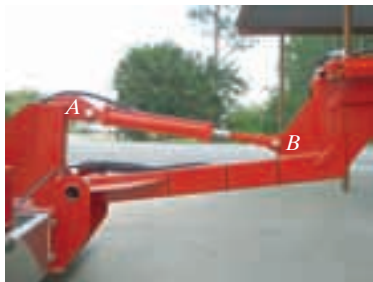
or

$$\zeta + \Sigma M_B = 0; \quad M_A + 900 \text{ N}[1.5 \text{ m} + (1 \text{ m}) \cos 45^\circ] - 500 \text{ N} \cdot \text{m} = 0$$

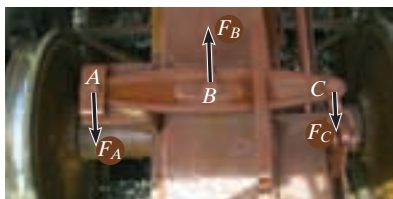
$$M_A = -1486 \text{ N} \cdot \text{m} = 1.49 \text{ kN} \cdot \text{m} \curvearrowright \quad \text{Ans.}$$

The negative sign indicates that  $M_A$  has the opposite sense of rotation to that shown on the free-body diagram.

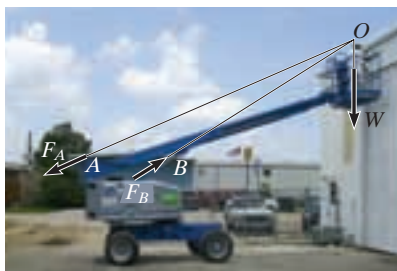




The hydraulic cylinder  $AB$  is a typical example of a two-force member since it is pin connected at its ends and, provided its weight is neglected, only the pin forces act on this member. (© Russell C. Hibbeler)



The link used for this railroad car brake is a three-force member. Since the force  $\mathbf{F}_B$  in the tie rod at  $B$  and  $\mathbf{F}_C$  from the link at  $C$  are parallel, then for equilibrium the resultant force  $\mathbf{F}_A$  at the pin  $A$  must also be parallel with these two forces. (© Russell C. Hibbeler)



The boom and bucket on this lift is a three-force member, provided its weight is neglected. Here the lines of action of the weight of the worker,  $\mathbf{W}$ , and the force of the two-force member (hydraulic cylinder) at  $B$ ,  $\mathbf{F}_B$ , intersect at  $O$ . For moment equilibrium, the resultant force at the pin  $A$ ,  $\mathbf{F}_A$ , must also be directed towards  $O$ . (© Russell C. Hibbeler)

## 5.4 Two- and Three-Force Members

The solutions to some equilibrium problems can be simplified by recognizing members that are subjected to only two or three forces.

**Two-Force Members.** As the name implies, a *two-force member* has forces applied at only two points on the member. An example of a two-force member is shown in Fig. 5–20a. To satisfy force equilibrium,  $\mathbf{F}_A$  and  $\mathbf{F}_B$  must be equal in magnitude,  $F_A = F_B = F$ , but opposite in direction ( $\Sigma \mathbf{F} = \mathbf{0}$ ), Fig. 5–20b. Furthermore, moment equilibrium requires that  $\mathbf{F}_A$  and  $\mathbf{F}_B$  share the same line of action, which can only happen if they are directed along the line joining points  $A$  and  $B$  ( $\Sigma \mathbf{M}_A = \mathbf{0}$  or  $\Sigma \mathbf{M}_B = \mathbf{0}$ ), Fig. 5–20c. Therefore, for any two-force member to be in equilibrium, the two forces acting on the member *must have the same magnitude, act in opposite directions, and have the same line of action, directed along the line joining the two points where these forces act.*

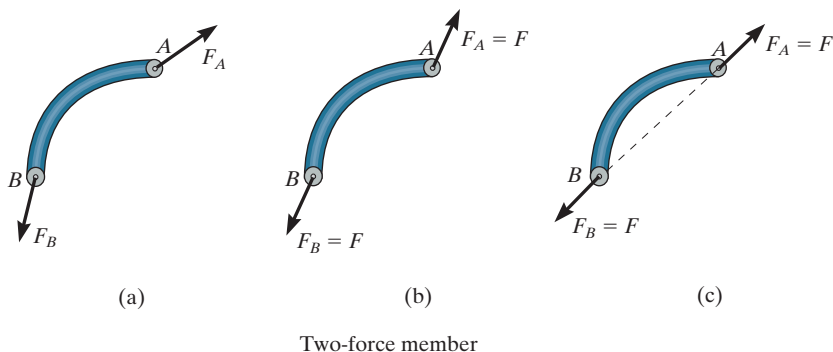


Fig. 5–20

**Three-Force Members.** If a member is subjected to only *three* forces, it is called a *three-force member*. Moment equilibrium can be satisfied only if the three forces form a *concurrent* or *parallel* force system. To illustrate, consider the member subjected to the three forces  $\mathbf{F}_1$ ,  $\mathbf{F}_2$ , and  $\mathbf{F}_3$ , shown in Fig. 5–21a. If the lines of action of  $\mathbf{F}_1$  and  $\mathbf{F}_2$  intersect at point  $O$ , then the line of action of  $\mathbf{F}_3$  must *also* pass through point  $O$  so that the forces satisfy  $\Sigma \mathbf{M}_O = \mathbf{0}$ . As a special case, if the three forces are all parallel, Fig. 5–21b, the location of the point of intersection,  $O$ , will approach infinity.

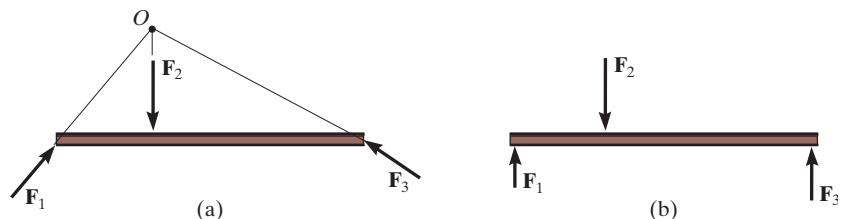


Fig. 5–21

## EXAMPLE 5.13

The lever  $ABC$  is pin supported at  $A$  and connected to a short link  $BD$  as shown in Fig. 5–22a. If the weight of the members is negligible, determine the force of the pin on the lever at  $A$ .

## SOLUTION

**Free-Body Diagrams.** As shown in Fig. 5–22b, the short link  $BD$  is a *two-force member*, so the *resultant forces* from the pins  $D$  and  $B$  must be equal, opposite, and collinear. Although the magnitude of the force is unknown, the line of action is known since it passes through  $B$  and  $D$ .

Lever  $ABC$  is a *three-force member*, and therefore, in order to satisfy moment equilibrium, the three nonparallel forces acting on it must be concurrent at  $O$ , Fig. 5–22c. In particular, note that the force  $\mathbf{F}$  on the lever at  $B$  is equal but opposite to the force  $\mathbf{F}$  acting at  $B$  on the link. Why? The distance  $CO$  must be  $0.5\text{ m}$  since the lines of action of  $\mathbf{F}$  and the  $400\text{-N}$  force are known.

**Equations of Equilibrium.** By requiring the force system to be concurrent at  $O$ , since  $\sum M_O = 0$ , the angle  $\theta$  which defines the line of action of  $\mathbf{F}_A$  can be determined from trigonometry,

$$\theta = \tan^{-1}\left(\frac{0.7}{0.4}\right) = 60.3^\circ$$

Using the  $x, y$  axes and applying the force equilibrium equations,

$$\rightarrow \sum F_x = 0; \quad F_A \cos 60.3^\circ - F \cos 45^\circ + 400\text{ N} = 0$$

$$+\uparrow \sum F_y = 0; \quad F_A \sin 60.3^\circ - F \sin 45^\circ = 0$$

Solving, we get

$$F_A = 1.07\text{ kN}$$

$$F = 1.32\text{ kN}$$

*Ans.*

**NOTE:** We can also solve this problem by representing the force at  $A$  by its two components  $A_x$  and  $A_y$  and applying  $\sum M_A = 0$ ,  $\sum F_x = 0$ ,  $\sum F_y = 0$  to the lever. Once  $A_x$  and  $A_y$  are determined, we can get  $F_A$  and  $\theta$ .

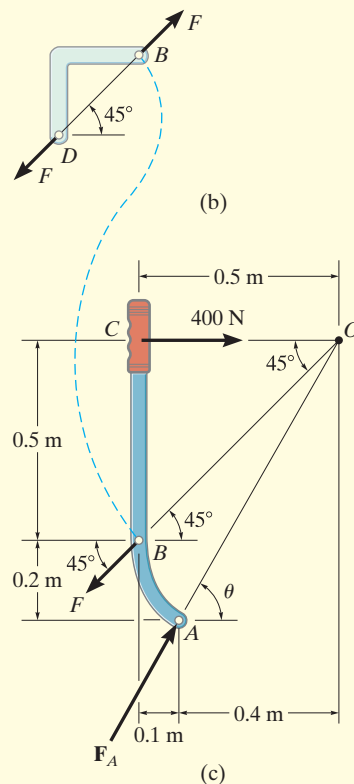
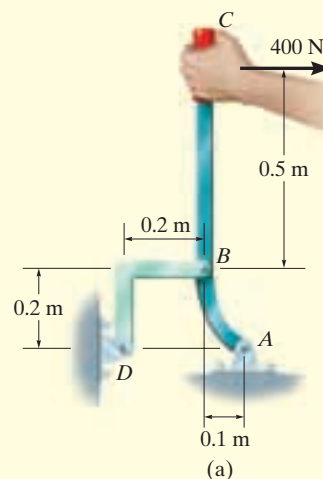
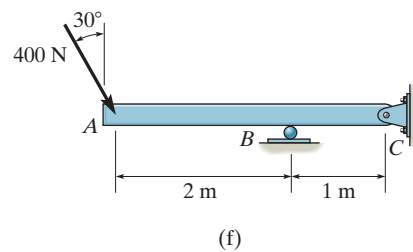
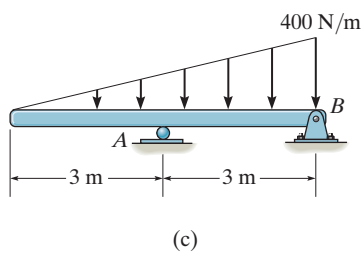
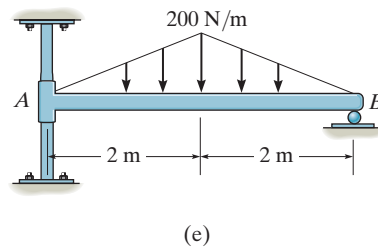
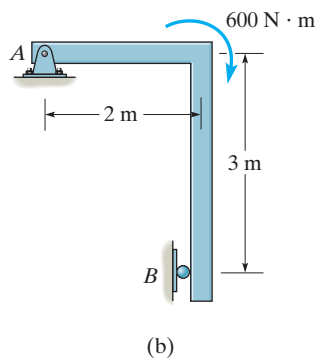
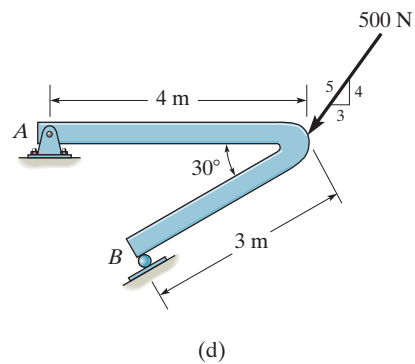
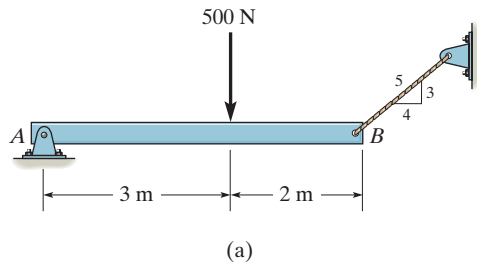


Fig. 5–22

## PRELIMINARY PROBLEMS

**P5-1.** Draw the free-body diagram of each object.

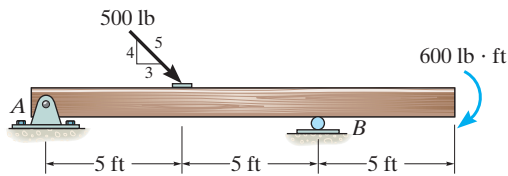


**Prob. P5-1**

## FUNDAMENTAL PROBLEMS

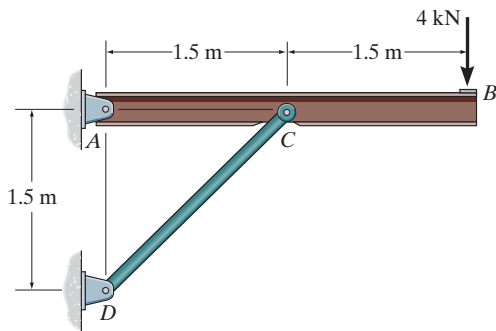
**All problem solutions must include an FBD.**

**F5-1.** Determine the horizontal and vertical components of reaction at the supports. Neglect the thickness of the beam.



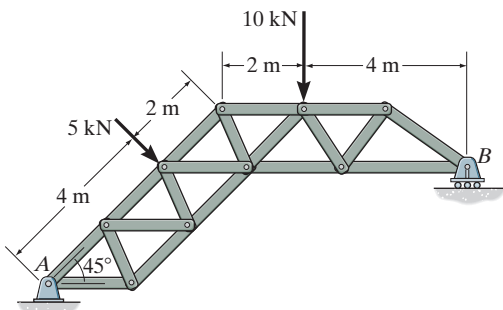
**Prob. F5-1**

**F5-2.** Determine the horizontal and vertical components of reaction at the pin A and the reaction on the beam at C.



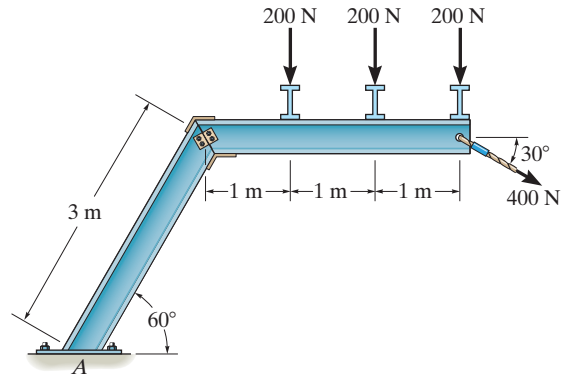
**Prob. F5-2**

**F5-3.** The truss is supported by a pin at A and a roller at B. Determine the support reactions.



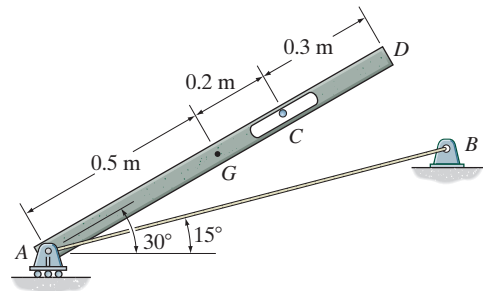
**Prob. F5-3**

**F5-4.** Determine the components of reaction at the fixed support A. Neglect the thickness of the beam.



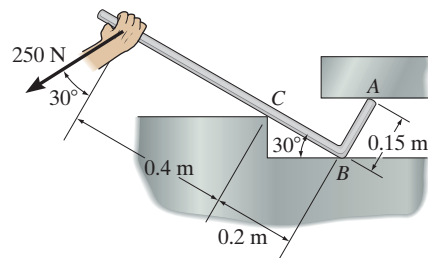
**Prob. F5-4**

**F5-5.** The 25-kg bar has a center of mass at G. If it is supported by a smooth peg at C, a roller at A, and cord AB, determine the reactions at these supports.



**Prob. F5-5**

**F5-6.** Determine the reactions at the smooth contact points A, B, and C on the bar.

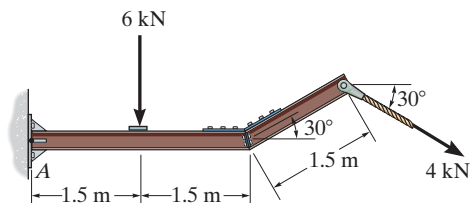


**Prob. F5-6**

# PROBLEMS

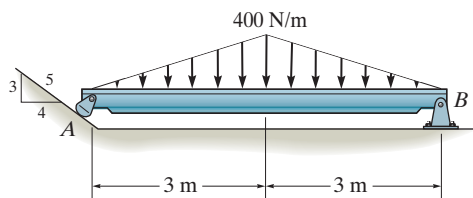
*All problem solutions must include an FBD.*

**5-10.** Determine the components of the support reactions at the fixed support  $A$  on the cantilevered beam.



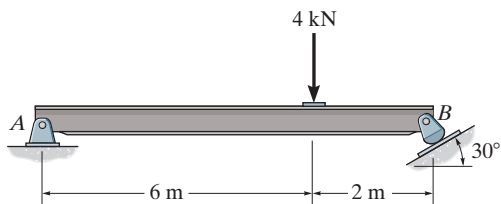
**Prob. 5-10**

**5-11.** Determine the reactions at the supports.



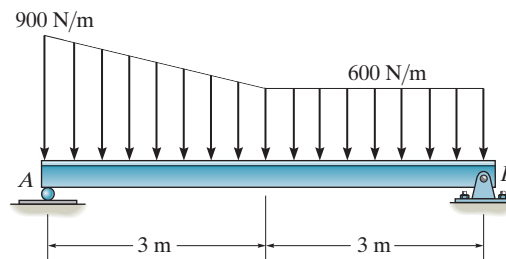
**Prob. 5-11**

**\*5-12.** Determine the horizontal and vertical components of reaction at the pin  $A$  and the reaction of the rocker  $B$  on the beam.



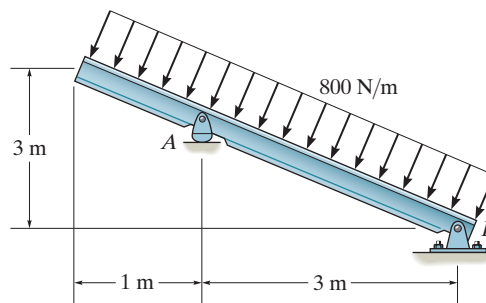
**Prob. 5-12**

**5-13.** Determine the reactions at the supports.



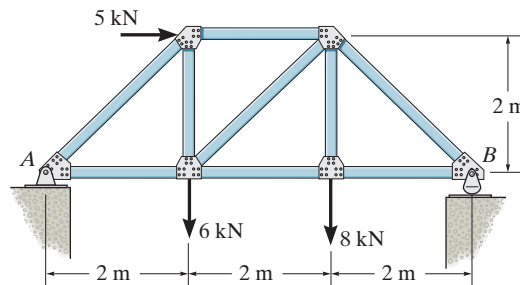
**Prob. 5-13**

**5-14.** Determine the reactions at the supports.



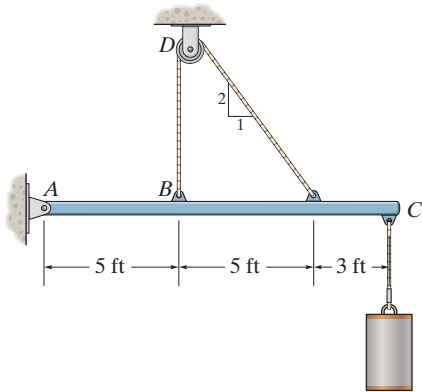
**Prob. 5-14**

**5-15.** Determine the reactions at the supports.



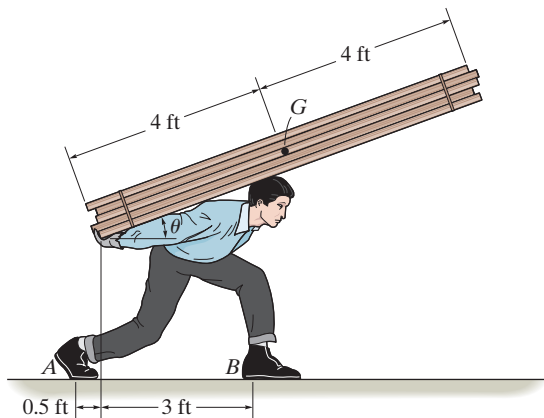
**Prob. 5-15**

**\*5-16.** Determine the tension in the cable and the horizontal and vertical components of reaction of the pin  $A$ . The pulley at  $D$  is frictionless and the cylinder weighs 80 lb.



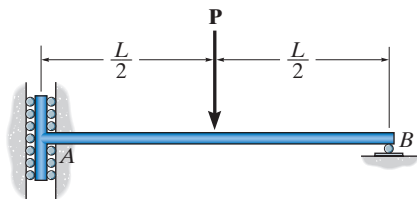
**Prob. 5-16**

**5-17.** The man attempts to support the load of boards having a weight  $W$  and a center of gravity at  $G$ . If he is standing on a smooth floor, determine the smallest angle  $\theta$  at which he can hold them up in the position shown. Neglect his weight.



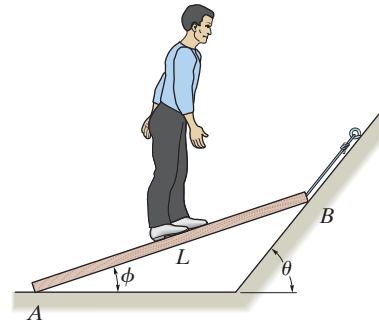
**Prob. 5-17**

**5-18.** Determine the components of reaction at the supports  $A$  and  $B$  on the rod.



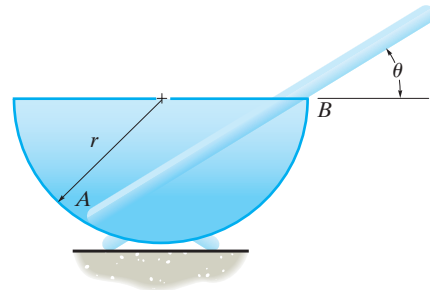
**Prob. 5-18**

**5-19.** The man has a weight  $W$  and stands at the center of the plank. If the planes at  $A$  and  $B$  are smooth, determine the tension in the cord in terms of  $W$  and  $\theta$ .



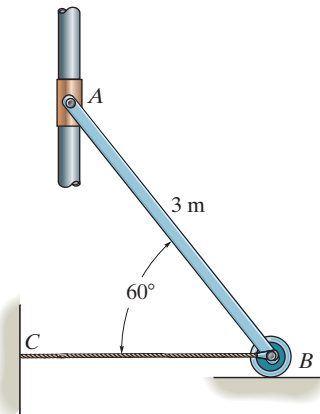
**Prob. 5-19**

**\*5-20.** A uniform glass rod having a length  $L$  is placed in the smooth hemispherical bowl having a radius  $r$ . Determine the angle of inclination  $\theta$  for equilibrium.



**Prob. 5-20**

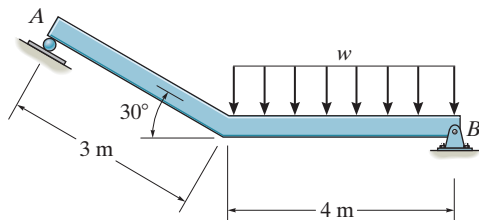
**5-21.** The uniform rod  $AB$  has a mass of 40 kg. Determine the force in the cable when the rod is in the position shown. There is a smooth collar at  $A$ .



**Prob. 5-21**

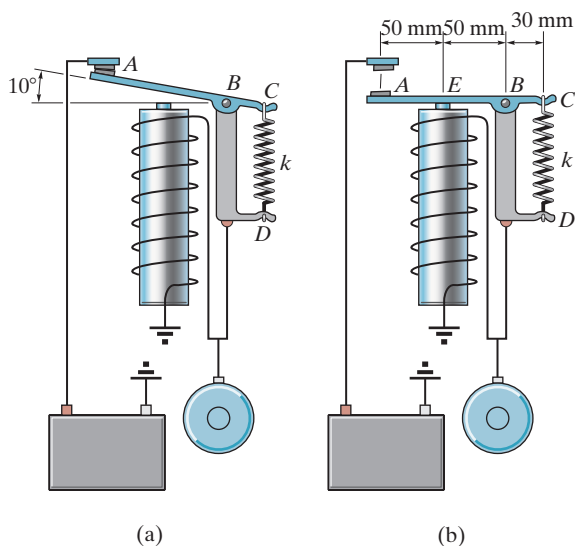
**5-22.** If the intensity of the distributed load acting on the beam is  $w = 3 \text{ kN/m}$ , determine the reactions at the roller  $A$  and pin  $B$ .

**5-23.** If the roller at  $A$  and the pin at  $B$  can support a load up to  $4 \text{ kN}$  and  $8 \text{ kN}$ , respectively, determine the maximum intensity of the distributed load  $w$ , measured in  $\text{kN/m}$ , so that failure of the supports does not occur.



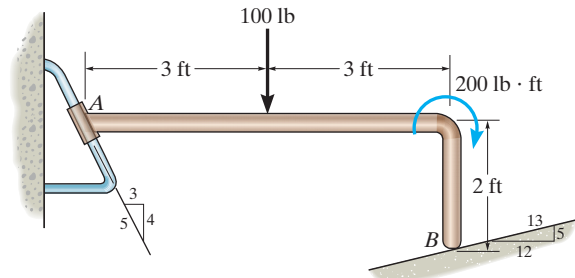
**Probs. 5-22/23**

**\*5-24.** The relay regulates voltage and current. Determine the force in the spring  $CD$ , which has a stiffness of  $k = 120 \text{ N/m}$ , so that it will allow the armature to make contact at  $A$  in figure (a) with a vertical force of  $0.4 \text{ N}$ . Also, determine the force in the spring when the coil is energized and attracts the armature to  $E$ , figure (b), thereby breaking contact at  $A$ .



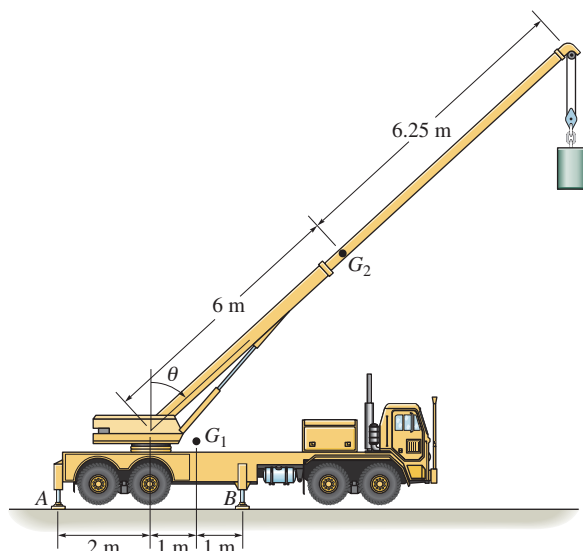
**Prob. 5-24**

**5-25.** Determine the reactions on the bent rod which is supported by a smooth surface at  $B$  and by a collar at  $A$ , which is fixed to the rod and is free to slide over the fixed inclined rod.



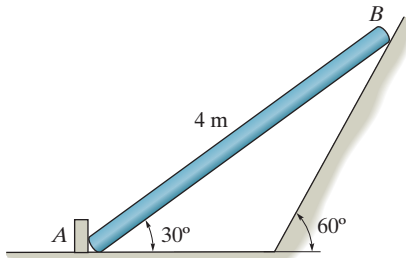
**Prob. 5-25**

**5-26.** The mobile crane is symmetrically supported by two outriggers at  $A$  and two at  $B$  in order to relieve the suspension of the truck upon which it rests and to provide greater stability. If the crane boom and truck have a mass of  $18 \text{ Mg}$  and center of mass at  $G_1$ , and the boom has a mass of  $1.8 \text{ Mg}$  and a center of mass at  $G_2$ , determine the vertical reactions at each of the four outriggers as a function of the boom angle  $\theta$  when the boom is supporting a load having a mass of  $1.2 \text{ Mg}$ . Plot the results measured from  $\theta = 0^\circ$  to the critical angle where tipping starts to occur.



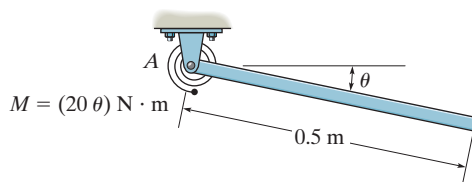
**Prob. 5-26**

**5-27.** Determine the reactions acting on the smooth uniform bar, which has a mass of 20 kg.



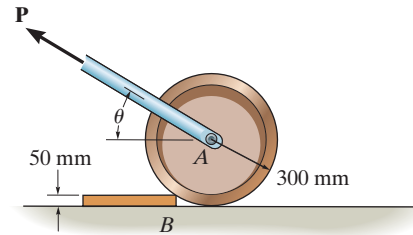
**Prob. 5-27**

**\*5-28.** A linear *torsional spring* deforms such that an applied couple moment  $M$  is related to the spring's rotation  $\theta$  in radians by the equation  $M = (20 \theta) \text{ N} \cdot \text{m}$ . If such a spring is attached to the end of a pin-connected uniform 10-kg rod, determine the angle  $\theta$  for equilibrium. The spring is undeformed when  $\theta = 0^\circ$ .



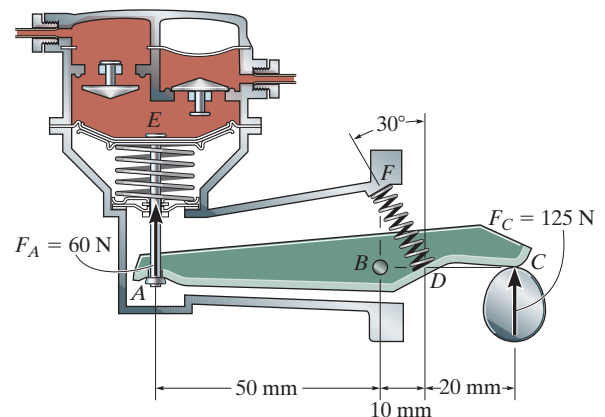
**Prob. 5-28**

**5-29.** Determine the force  $P$  needed to pull the 50-kg roller over the smooth step. Take  $\theta = 30^\circ$ .



**Probs. 5-29/30**

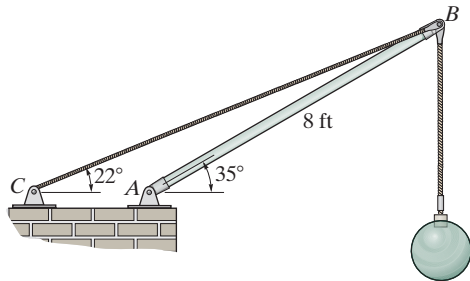
**5-31.** The operation of the fuel pump for an automobile depends on the reciprocating action of the rocker arm  $ABC$ , which is pinned at  $B$  and is spring loaded at  $A$  and  $D$ . When the smooth cam  $C$  is in the position shown, determine the horizontal and vertical components of force at the pin and the force along the spring  $DF$  for equilibrium. The vertical force acting on the rocker arm at  $A$  is  $F_A = 60 \text{ N}$ , and at  $C$  it is  $F_C = 125 \text{ N}$ .



**Prob. 5-31**



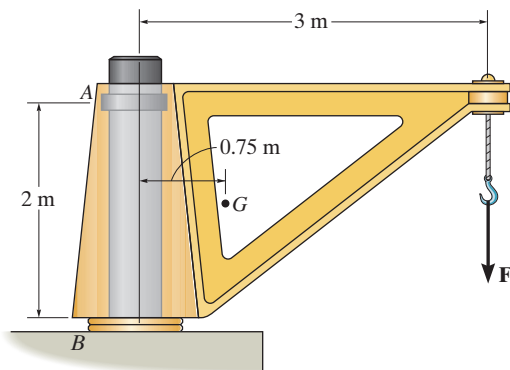
**\*5-32.** Determine the magnitude of force at the pin  $A$  and in the cable  $BC$  needed to support the 500-lb load. Neglect the weight of the boom  $AB$ .



**Prob. 5-32**

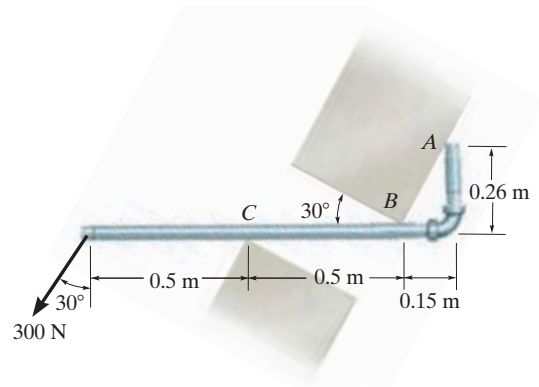
**5-33.** The dimensions of a jib crane, which is manufactured by the Basick Co., are given in the figure. If the crane has a mass of 800 kg and a center of mass at  $G$ , and the maximum rated force at its end is  $F = 15$  kN, determine the reactions at its bearings. The bearing at  $A$  is a journal bearing and supports only a horizontal force, whereas the bearing at  $B$  is a thrust bearing that supports both horizontal and vertical components.

**5-34.** The dimensions of a jib crane, which is manufactured by the Basick Co., are given in the figure. The crane has a mass of 800 kg and a center of mass at  $G$ . The bearing at  $A$  is a journal bearing and can support a horizontal force, whereas the bearing at  $B$  is a thrust bearing that supports both horizontal and vertical components. Determine the maximum load  $F$  that can be suspended from its end if the selected bearings at  $A$  and  $B$  can sustain a maximum resultant load of 24 kN and 34 kN, respectively.



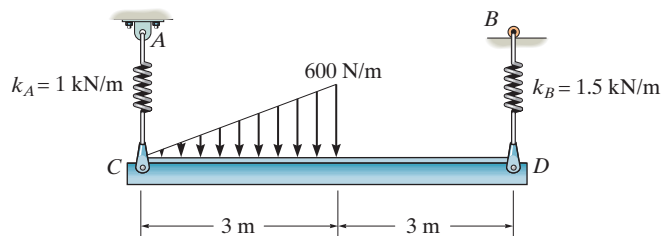
**Probs. 5-33/34**

**5-35.** The smooth pipe rests against the opening at the points of contact  $A$ ,  $B$ , and  $C$ . Determine the reactions at these points needed to support the force of 300 N. Neglect the pipe's thickness in the calculation.



**Prob. 5-35**

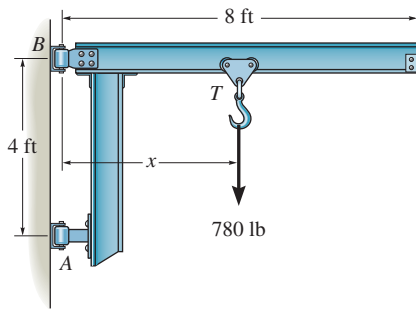
**\*5-36.** The beam of negligible weight is supported horizontally by two springs. If the beam is horizontal and the springs are unstretched when the load is removed, determine the angle of tilt of the beam when the load is applied.



**Prob. 5-36**

**5-37.** The cantilevered jib crane is used to support the load of 780 lb. If  $x = 5$  ft, determine the reactions at the supports. Note that the supports are collars that allow the crane to rotate freely about the vertical axis. The collar at  $B$  supports a force in the vertical direction, whereas the one at  $A$  does not.

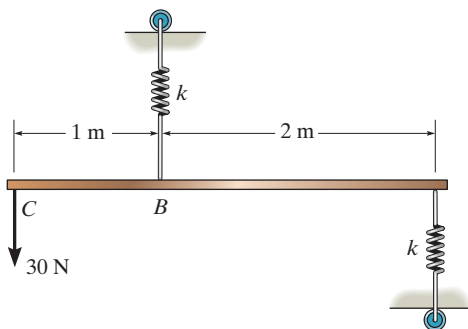
**5-38.** The cantilevered jib crane is used to support the load of 780 lb. If the trolley  $T$  can be placed anywhere between  $1.5 \text{ ft} \leq x \leq 7.5 \text{ ft}$ , determine the maximum magnitude of reaction at the supports  $A$  and  $B$ . Note that the supports are collars that allow the crane to rotate freely about the vertical axis. The collar at  $B$  supports a force in the vertical direction, whereas the one at  $A$  does not.



**Probs. 5-37/38**

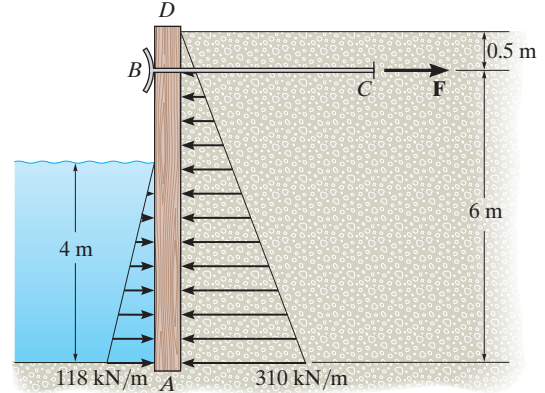
**5-39.** The bar of negligible weight is supported by two springs, each having a stiffness  $k = 100 \text{ N/m}$ . If the springs are originally unstretched, and the force is vertical as shown, determine the angle  $\theta$  the bar makes with the horizontal, when the 30-N force is applied to the bar.

**\*5-40.** Determine the stiffness  $k$  of each spring so that the 30-N force causes the bar to tip  $\theta = 15^\circ$  when the force is applied. Originally the bar is horizontal and the springs are unstretched. Neglect the weight of the bar.



**Probs. 5-39/40**

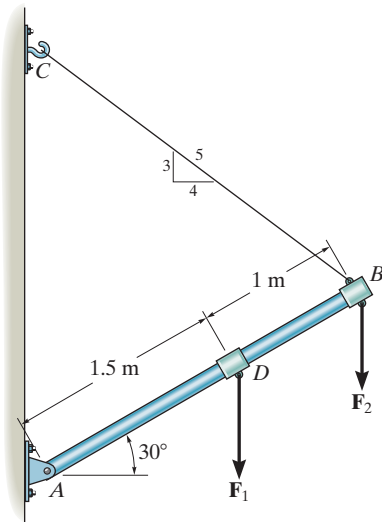
**5-41.** The bulk head  $AD$  is subjected to both water and soil-backfill pressures. Assuming  $AD$  is “pinned” to the ground at  $A$ , determine the horizontal and vertical reactions there and also the required tension in the ground anchor  $BC$  necessary for equilibrium. The bulk head has a mass of 800 kg.



**Prob. 5-41**

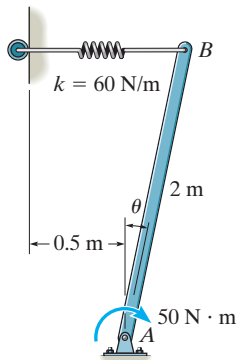
**5-42.** The boom supports the two vertical loads. Neglect the size of the collars at  $D$  and  $B$  and the thickness of the boom, and compute the horizontal and vertical components of force at the pin  $A$  and the force in cable  $CB$ . Set  $F_1 = 800 \text{ N}$  and  $F_2 = 350 \text{ N}$ .

**5-43.** The boom is intended to support two vertical loads,  $F_1$  and  $F_2$ . If the cable  $CB$  can sustain a maximum load of 1500 N before it fails, determine the critical loads if  $F_1 = 2F_2$ . Also, what is the magnitude of the maximum reaction at pin  $A$ ?



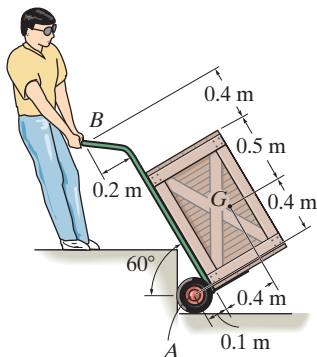
**Probs. 5-42/43**

**\*5-44.** The 10-kg uniform rod is pinned at end  $A$ . If it is also subjected to a couple moment of  $50 \text{ N} \cdot \text{m}$ , determine the smallest angle  $\theta$  for equilibrium. The spring is unstretched when  $\theta = 0$ , and has a stiffness of  $k = 60 \text{ N/m}$ .



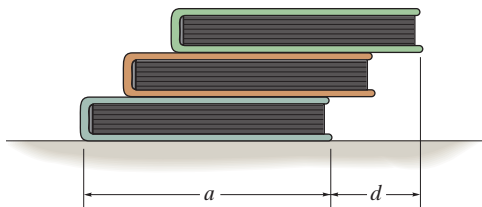
**Prob. 5-44**

**5-45.** The man uses the hand truck to move material up the step. If the truck and its contents have a mass of 50 kg with center of gravity at  $G$ , determine the normal reaction on both wheels and the magnitude and direction of the minimum force required at the grip  $B$  needed to lift the load.



**Prob. 5-45**

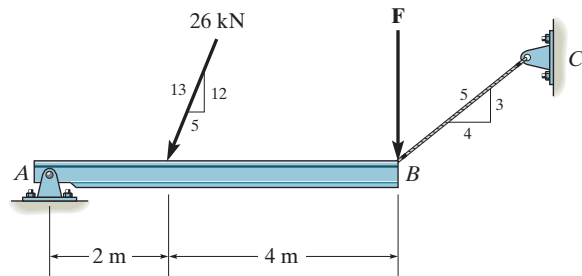
**5-46.** Three uniform books, each having a weight  $W$  and length  $a$ , are stacked as shown. Determine the maximum distance  $d$  that the top book can extend out from the bottom one so the stack does not topple over.



**Prob. 5-46**

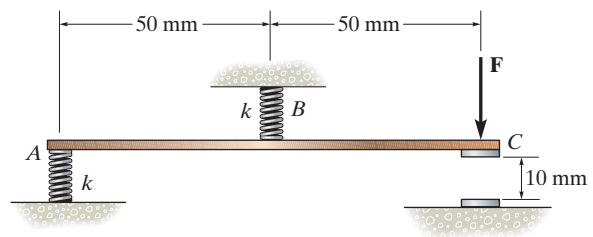
**5-47.** Determine the reactions at the pin  $A$  and the tension in cord  $BC$ . Set  $F = 40 \text{ kN}$ . Neglect the thickness of the beam.

**\*5-48.** If rope  $BC$  will fail when the tension becomes  $50 \text{ kN}$ , determine the greatest vertical load  $F$  that can be applied to the beam at  $B$ . What is the magnitude of the reaction at  $A$  for this loading? Neglect the thickness of the beam.



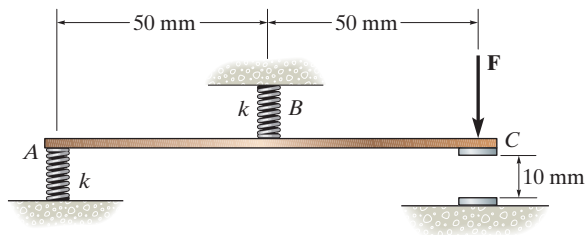
**Probs. 5-47/48**

**5-49.** The rigid metal strip of negligible weight is used as part of an electromagnetic switch. If the stiffness of the springs at  $A$  and  $B$  is  $k = 5 \text{ N/m}$  and the strip is originally horizontal when the springs are unstretched, determine the smallest force  $F$  needed to close the contact gap at  $C$ .



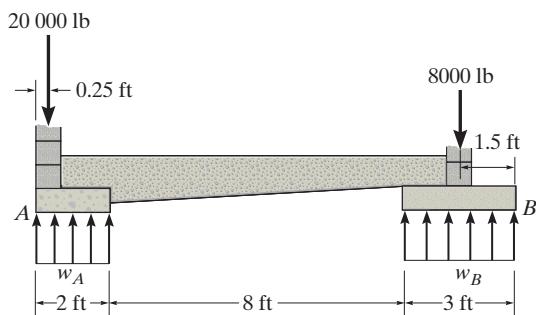
**Prob. 5-49**

**5-50.** The rigid metal strip of negligible weight is used as part of an electromagnetic switch. Determine the maximum stiffness  $k$  of the springs at  $A$  and  $B$  so that the contact at  $C$  closes when the vertical force developed there is  $F = 0.5$  N. Originally the strip is horizontal as shown.



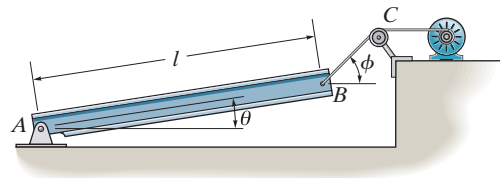
**Prob. 5-50**

**5-51.** The cantilever footing is used to support a wall near its edge  $A$  so that it causes a uniform soil pressure under the footing. Determine the uniform distribution loads,  $w_A$  and  $w_B$ , measured in lb/ft at pads  $A$  and  $B$ , necessary to support the wall forces of 8000 lb and 20 000 lb.



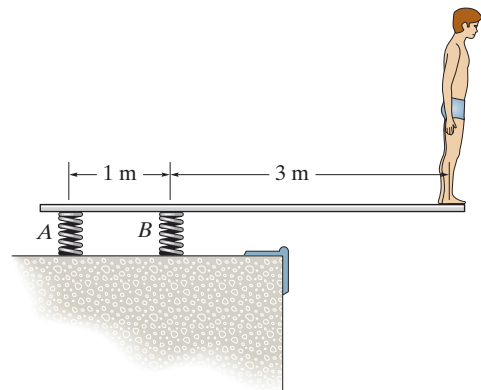
**Prob. 5-51**

**\*5-52.** The uniform beam has a weight  $W$  and length  $l$  and is supported by a pin at  $A$  and a cable  $BC$ . Determine the horizontal and vertical components of reaction at  $A$  and the tension in the cable necessary to hold the beam in the position shown.



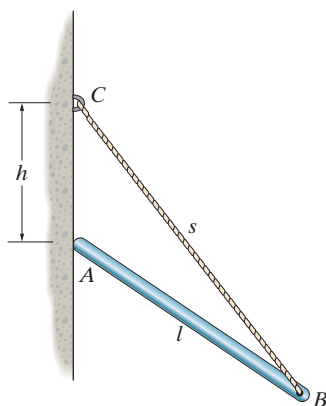
**Prob. 5-52**

**5-53.** A boy stands out at the end of the diving board, which is supported by two springs  $A$  and  $B$ , each having a stiffness of  $k = 15$  kN/m. In the position shown the board is horizontal. If the boy has a mass of 40 kg, determine the angle of tilt which the board makes with the horizontal after he jumps off. Neglect the weight of the board and assume it is rigid.



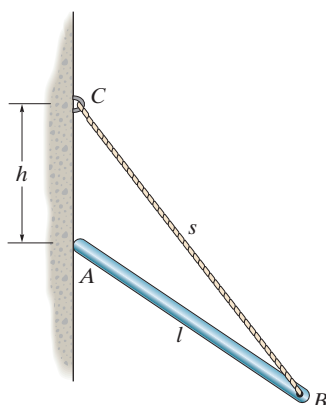
**Prob. 5-53**

**5-54.** The 30-N uniform rod has a length of  $l = 1$  m. If  $s = 1.5$  m, determine the distance  $h$  of placement at the end  $A$  along the smooth wall for equilibrium.



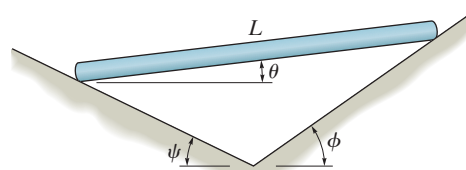
**Prob. 5-54**

**5-55.** The uniform rod has a length  $l$  and weight  $W$ . It is supported at one end  $A$  by a smooth wall and the other end by a cord of length  $s$  which is attached to the wall as shown. Determine the placement  $h$  for equilibrium.



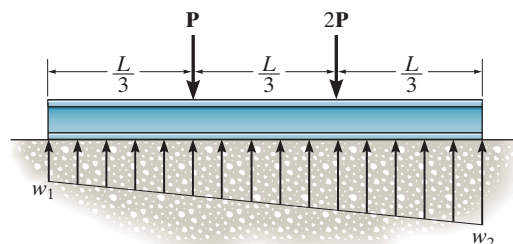
**Prob. 5-55**

**\*5-56.** The uniform rod of length  $L$  and weight  $W$  is supported on the smooth planes. Determine its position  $\theta$  for equilibrium. Neglect the thickness of the rod.



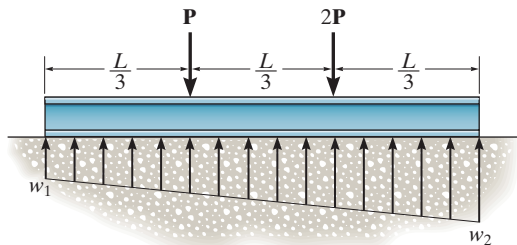
**Prob. 5-56**

**5-57.** The beam is subjected to the two concentrated loads. Assuming that the foundation exerts a linearly varying load distribution on its bottom, determine the load intensities  $w_1$  and  $w_2$  for equilibrium if  $P = 500$  lb and  $L = 12$  ft.



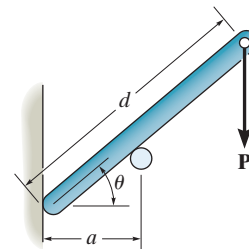
**Prob. 5-57**

**5-58.** The beam is subjected to the two concentrated loads. Assuming that the foundation exerts a linearly varying load distribution on its bottom, determine the load intensities  $w_1$  and  $w_2$  for equilibrium in terms of the parameters shown.



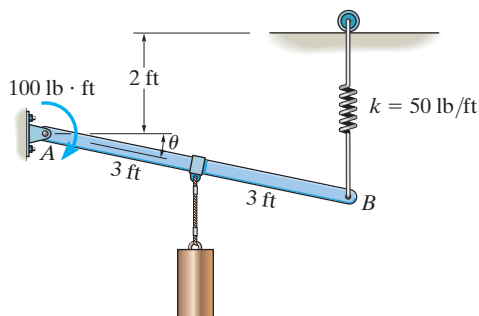
**Prob. 5-58**

**\*5-60.** Determine the distance  $d$  for placement of the load  $P$  for equilibrium of the smooth bar in the position  $\theta$  as shown. Neglect the weight of the bar.



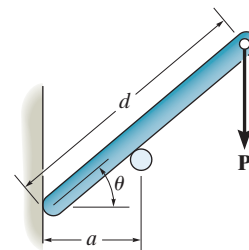
**Prob. 5-60**

**5-59.** The rod supports a weight of 200 lb and is pinned at its end A. If it is also subjected to a couple moment of 100 lb · ft, determine the angle  $\theta$  for equilibrium. The spring has an unstretched length of 2 ft and a stiffness of  $k = 50$  lb/ft.



**Prob. 5-59**

**5-61.** If  $d = 1$  m, and  $\theta = 30^\circ$ , determine the normal reaction at the smooth supports and the required distance  $a$  for the placement of the roller if  $P = 600$  N. Neglect the weight of the bar.



**Prob. 5-61**

## CONCEPTUAL PROBLEMS

**C5-1.** The tie rod is used to support this overhang at the entrance of a building. If it is pin connected to the building wall at  $A$  and to the center of the overhang  $B$ , determine if the force in the rod will increase, decrease, or remain the same if (a) the support at  $A$  is moved to a lower position  $D$ , and (b) the support at  $B$  is moved to the outer position  $C$ . Explain your answer with an equilibrium analysis, using dimensions and loads. Assume the overhang is pin supported from the building wall.



**Prob. C5-1** (© Russell C. Hibbeler)

**C5-2.** The man attempts to pull the four wheeler up the incline and onto the trailer. From the position shown, is it more effective to pull on the rope at  $A$ , or would it be better to pull on the rope at  $B$ ? Draw a free-body diagram for each case, and do an equilibrium analysis to explain your answer. Use appropriate numerical values to do your calculations.



**Prob. C5-2** (© Russell C. Hibbeler)

**C5-3.** Like all aircraft, this jet plane rests on three wheels. Why not use an additional wheel at the tail for better support? (Can you think of any other reason for not including this wheel?) If there was a fourth tail wheel, draw a free-body diagram of the plane from a side (2 D) view, and show why one would not be able to determine all the wheel reactions using the equations of equilibrium.



**Prob. C5-3** (© Russell C. Hibbeler)

**C5-4.** Where is the best place to arrange most of the logs in the wheelbarrow so that it minimizes the amount of force on the backbone of the person transporting the load? Do an equilibrium analysis to explain your answer.



**Prob. C5-4** (© Russell C. Hibbeler)

## EQUILIBRIUM IN THREE DIMENSIONS

### 5.5 Free-Body Diagrams

The first step in solving three-dimensional equilibrium problems, as in the case of two dimensions, is to draw a free-body diagram. Before we can do this, however, it is first necessary to discuss the types of reactions that can occur at the supports.

**Support Reactions.** The reactive forces and couple moments acting at various types of supports and connections, when the members are viewed in three dimensions, are listed in Table 5–2. It is important to recognize the symbols used to represent each of these supports and to understand clearly how the forces and couple moments are developed. As in the two-dimensional case:

- A force is developed by a support that restricts the translation of its attached member.
- A couple moment is developed when rotation of the attached member is prevented.


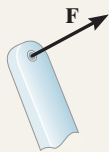





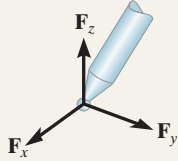

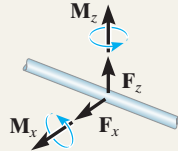
For example, in Table 5–2, item (4), the ball-and-socket joint prevents any translation of the connecting member; therefore, a force must act on the member at the point of connection. This force has three components having unknown magnitudes,  $F_x$ ,  $F_y$ ,  $F_z$ . Provided these components are known, one can obtain the magnitude of force,  $F = \sqrt{F_x^2 + F_y^2 + F_z^2}$ , and the force's orientation defined by its coordinate direction angles  $\alpha$ ,  $\beta$ ,  $\gamma$ , Eqs. 2–5.\* Since the connecting member is allowed to rotate freely about *any* axis, no couple moment is resisted by a ball-and-socket joint.

It should be noted that the *single* bearing supports in items (5) and (7), the *single* pin (8), and the *single* hinge (9) are shown to resist both force and couple-moment components. If, however, these supports are used in conjunction with *other* bearings, pins, or hinges to hold a rigid body in equilibrium and the supports are *properly aligned* when connected to the body, then the *force reactions* at these supports *alone* are adequate for supporting the body. In other words, the couple moments become redundant and are not shown on the free-body diagram. The reason for this should become clear after studying the examples which follow.

\* The three unknowns may also be represented as an unknown force magnitude  $F$  and two unknown coordinate direction angles. The third direction angle is obtained using the identity  $\cos^2 \alpha + \cos^2 \beta + \cos^2 \gamma = 1$ , Eq. 2–8


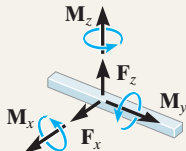

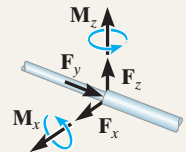

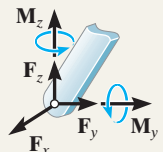

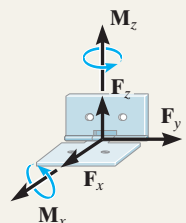

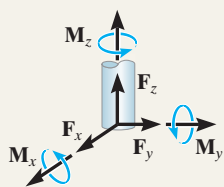


TABLE 5-2 Supports for Rigid Bodies Subjected to Three-Dimensional Force Systems

Types of Connection	Reaction	Number of Unknowns
(1)  cable		One unknown. The reaction is a force which acts away from the member in the known direction of the cable.
(2)  smooth surface support		One unknown. The reaction is a force which acts perpendicular to the surface at the point of contact.
(3)  roller		One unknown. The reaction is a force which acts perpendicular to the surface at the point of contact.
(4)  ball and socket		Three unknowns. The reactions are three rectangular force components.
(5)  single journal bearing		Four unknowns. The reactions are two force and two couple-moment components which act perpendicular to the shaft. Note: The couple moments are <i>generally not applied</i> if the body is supported elsewhere. See the examples.

continued

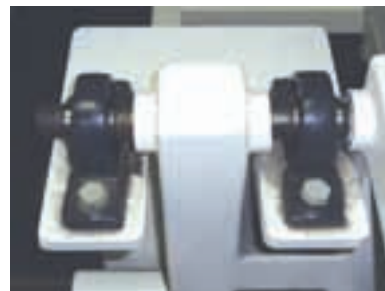
TABLE 5-2 Continued

Types of Connection	Reaction	Number of Unknowns
(6)  single journal bearing with square shaft		Five unknowns. The reactions are two force and three couple-moment components. <i>Note:</i> The couple moments are generally not applied if the body is supported elsewhere. See the examples.
(7)  single thrust bearing		Five unknowns. The reactions are three force and two couple-moment components. <i>Note:</i> The couple moments are generally not applied if the body is supported elsewhere. See the examples.
(8)  single smooth pin		Five unknowns. The reactions are three force and two couple-moment components. <i>Note:</i> The couple moments are generally not applied if the body is supported elsewhere. See the examples.
(9)  single hinge		Five unknowns. The reactions are three force and two couple-moment components. <i>Note:</i> The couple moments are generally not applied if the body is supported elsewhere. See the examples.
(10)  fixed support		Six unknowns. The reactions are three force and three couple-moment components.

Typical examples of actual supports that are referenced to Table 5–2 are shown in the following sequence of photos.



This ball-and-socket joint provides a connection for the housing of an earth grader to its frame. (4) (© Russell C. Hibbeler)



The journal bearings support the ends of the shaft. (5) (© Russell C. Hibbeler)



This thrust bearing is used to support the drive shaft on a machine. (7) (© Russell C. Hibbeler)



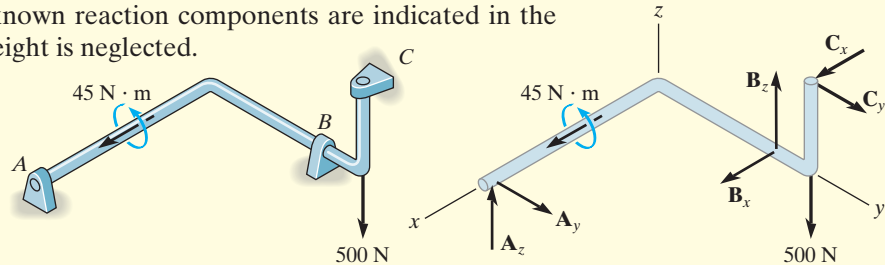
This pin is used to support the end of the strut used on a tractor. (8) (© Russell C. Hibbeler)

**Free-Body Diagrams.** The general procedure for establishing the free-body diagram of a rigid body has been outlined in Sec. 5.2. Essentially it requires first “isolating” the body by drawing its outlined shape. This is followed by a careful *labeling* of *all* the forces and couple moments with reference to an established  $x, y, z$  coordinate system. As a general rule, it is suggested to show the unknown components of reaction as acting on the free-body diagram in the *positive sense*. In this way, if any negative values are obtained, they will indicate that the components act in the negative coordinate directions.

## EXAMPLE 5.14

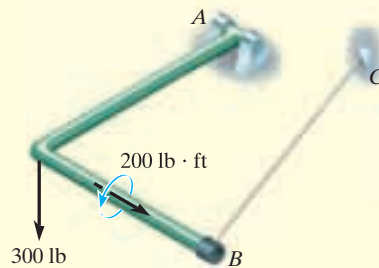
Consider the two rods and plate, along with their associated free-body diagrams, shown in Fig. 5–23. The  $x$ ,  $y$ ,  $z$  axes are established on the diagram and the unknown reaction components are indicated in the *positive sense*. The weight is neglected.

## SOLUTION

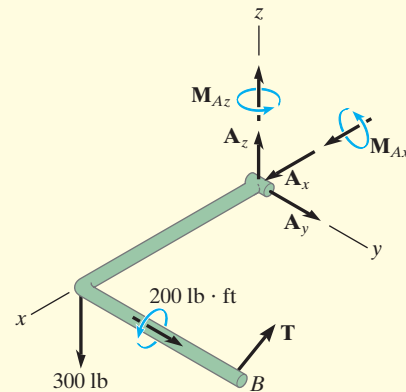


Properly aligned journal bearings at A, B, C.

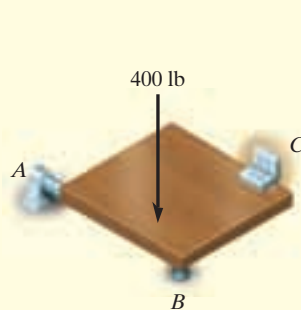
The force reactions developed by the bearings are *sufficient* for equilibrium since they prevent the shaft from rotating about each of the coordinate axes. No couple moments at each bearing are developed.



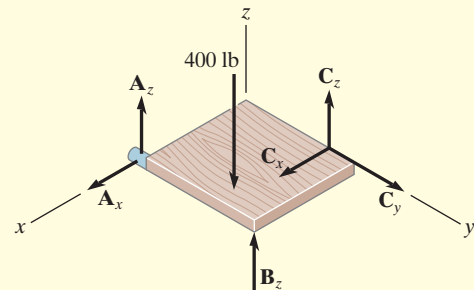
Pin at A and cable BC.



Moment components are developed by the pin on the rod to prevent rotation about the  $x$  and  $z$  axes.



Properly aligned journal bearing at A and hinge at C. Roller at B.



Only force reactions are developed by the bearing and hinge on the plate to prevent rotation about each coordinate axis. No moments are developed at the hinge.

Fig. 5–23

## 5.6 Equations of Equilibrium

As stated in Sec. 5.1, the conditions for equilibrium of a rigid body subjected to a three-dimensional force system require that both the *resultant* force and *resultant* couple moment acting on the body be equal to *zero*.

**Vector Equations of Equilibrium.** The two conditions for equilibrium of a rigid body may be expressed mathematically in vector form as

$$\begin{aligned}\Sigma \mathbf{F} &= \mathbf{0} \\ \Sigma \mathbf{M}_O &= \mathbf{0}\end{aligned}\tag{5-5}$$

where  $\Sigma \mathbf{F}$  is the vector sum of all the external forces acting on the body and  $\Sigma \mathbf{M}_O$  is the sum of the couple moments and the moments of all the forces about any point  $O$  located either on or off the body.

**Scalar Equations of Equilibrium.** If all the external forces and couple moments are expressed in Cartesian vector form and substituted into Eqs. 5-5, we have

$$\begin{aligned}\Sigma \mathbf{F} &= \Sigma F_x \mathbf{i} + \Sigma F_y \mathbf{j} + \Sigma F_z \mathbf{k} = \mathbf{0} \\ \Sigma \mathbf{M}_O &= \Sigma M_x \mathbf{i} + \Sigma M_y \mathbf{j} + \Sigma M_z \mathbf{k} = \mathbf{0}\end{aligned}$$

Since the  $\mathbf{i}$ ,  $\mathbf{j}$ , and  $\mathbf{k}$  components are independent from one another, the above equations are satisfied provided

$$\begin{aligned}\Sigma F_x &= 0 \\ \Sigma F_y &= 0 \\ \Sigma F_z &= 0\end{aligned}\tag{5-6a}$$

and

$$\begin{aligned}\Sigma M_x &= 0 \\ \Sigma M_y &= 0 \\ \Sigma M_z &= 0\end{aligned}\tag{5-6b}$$

These *six scalar equilibrium equations* may be used to solve for at most six unknowns shown on the free-body diagram. Equations 5-6a require the sum of the external force components acting in the  $x$ ,  $y$ , and  $z$  directions to be zero, and Eqs. 5-6b require the sum of the moment components about the  $x$ ,  $y$ , and  $z$  axes to be zero.

## 5.7 Constraints and Static Determinacy

To ensure the equilibrium of a rigid body, it is not only necessary to satisfy the equations of equilibrium, but the body must also be properly held or constrained by its supports. Some bodies may have more supports than are necessary for equilibrium, whereas others may not have enough or the supports may be arranged in a particular manner that could cause the body to move. Each of these cases will now be discussed.

**Redundant Constraints.** When a body has redundant supports, that is, more supports than are necessary to hold it in equilibrium, it becomes statically indeterminate. **Statically indeterminate** means that there will be more unknown loadings on the body than equations of equilibrium available for their solution. For example, the beam in Fig. 5–24a and the pipe assembly in Fig. 5–24b, shown together with their free-body diagrams, are both statically indeterminate because of additional (or redundant) support reactions. For the beam there are five unknowns,  $M_A$ ,  $A_x$ ,  $A_y$ ,  $B_y$ , and  $C_y$ , for which only three equilibrium equations can be written ( $\Sigma F_x = 0$ ,  $\Sigma F_y = 0$ , and  $\Sigma M_O = 0$ , Eq. 5–2). The pipe assembly has eight unknowns, for which only six equilibrium equations can be written, Eqs. 5–6.

The additional equations needed to solve statically indeterminate problems of the type shown in Fig. 5–24 are generally obtained from the deformation conditions at the points of support. These equations involve the physical properties of the body which are studied in subjects dealing with the mechanics of deformation, such as “mechanics of materials.”\*

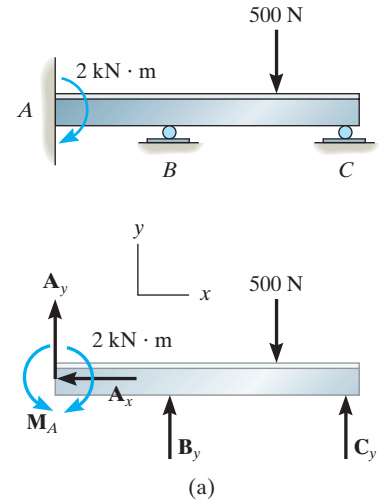
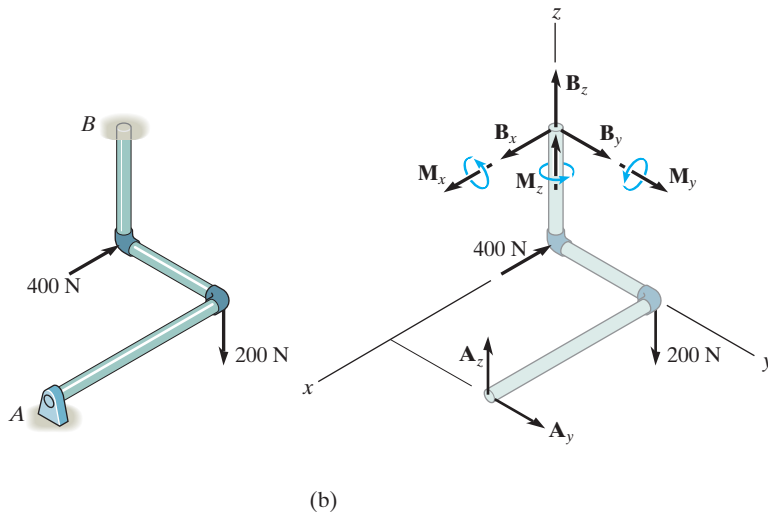


Fig. 5–24



\* See R. C. Hibbeler, *Mechanics of Materials*, 8th edition, Pearson Education/Prentice Hall, Inc.

**Improper Constraints.** Having the same number of unknown reactive forces as available equations of equilibrium does not always guarantee that a body will be stable when subjected to a particular loading. For example, the pin support at  $A$  and the roller support at  $B$  for the beam in Fig. 5–25a are placed in such a way that the lines of action of the reactive forces are *concurrent* at point  $A$ . Consequently, the applied loading  $\mathbf{P}$  will cause the beam to rotate slightly about  $A$ , and so the beam is improperly constrained,  $\Sigma M_A \neq 0$ .

In three dimensions, a body will be improperly constrained if the lines of action of all the reactive forces intersect a common axis. For example, the reactive forces at the ball-and-socket supports at  $A$  and  $B$  in Fig. 5–25b all intersect the axis passing through  $A$  and  $B$ . Since the moments of these forces about  $A$  and  $B$  are all zero, then the loading  $\mathbf{P}$  will rotate the member about the  $AB$  axis,  $\Sigma M_{AB} \neq 0$ .

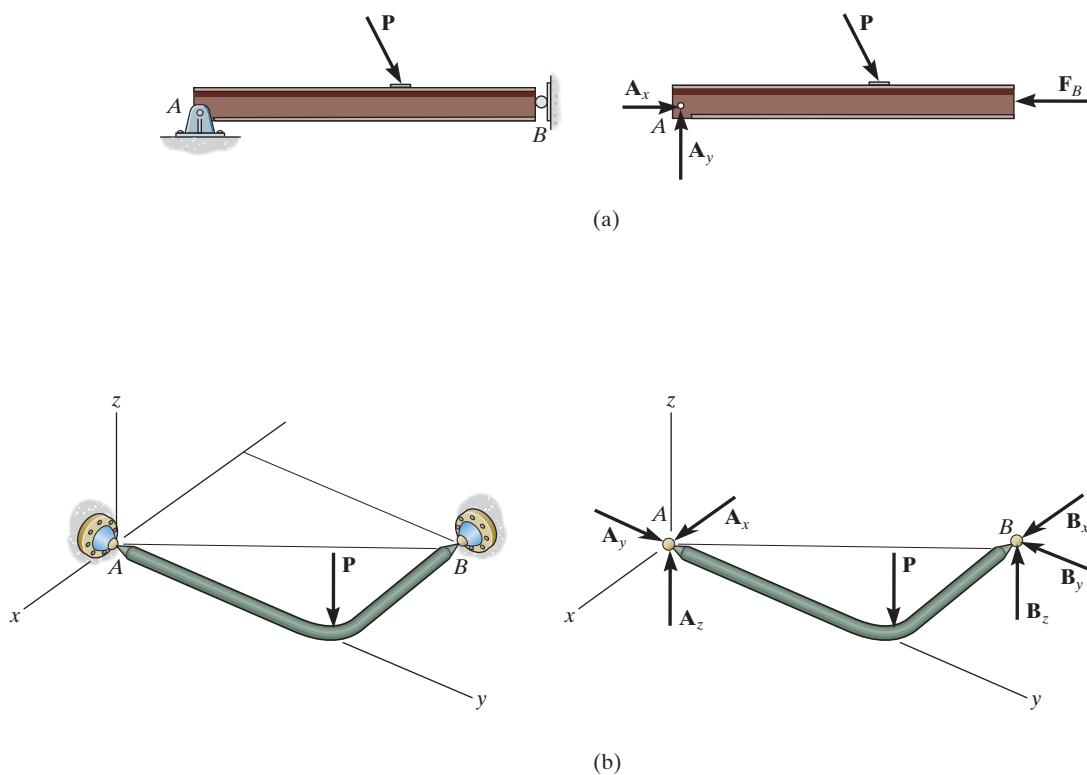


Fig. 5–25

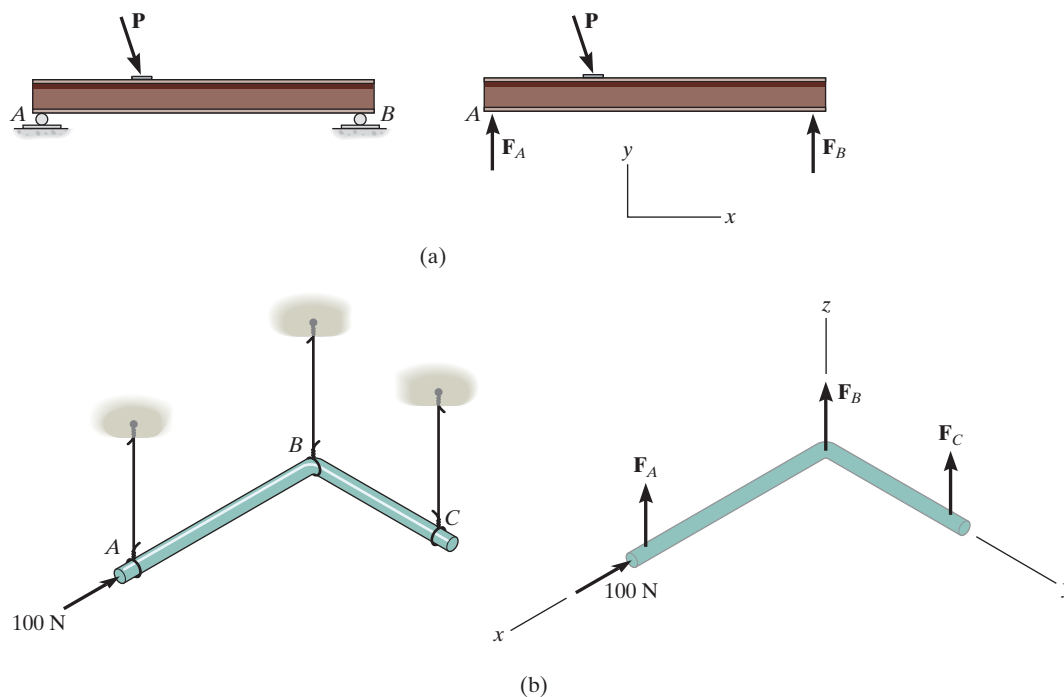


Fig. 5-26

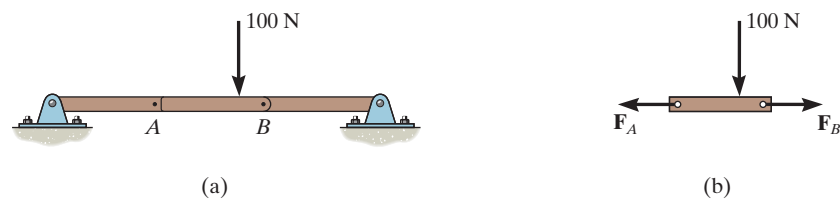


Fig. 5-27

Another way in which improper constraining leads to instability occurs when the *reactive forces* are all *parallel*. Two- and three-dimensional examples of this are shown in Fig. 5-26. In both cases, the summation of forces along the  $x$  axis will not equal zero.

In some cases, a body may have *fewer* reactive forces than equations of equilibrium that must be satisfied. The body then becomes only *partially constrained*. For example, consider member  $AB$  in Fig. 5-27a with its corresponding free-body diagram in Fig. 5-27b. Here  $\sum F_y = 0$  will not be satisfied for the loading conditions and therefore equilibrium will not be maintained.

To summarize these points, a body is considered **improperly constrained** if all the reactive forces intersect at a common point or pass through a common axis, or if all the reactive forces are parallel. In engineering practice, these situations should be avoided at all times since they will cause an unstable condition.



Stability is always an important concern when operating a crane, not only when lifting a load, but also when moving it about. (© Russell C. Hibbeler)



### Important Points

- Always draw the free-body diagram first when solving any equilibrium problem.
- If a support *prevents translation* of a body in a specific direction, then the support exerts a *force* on the body in that direction.
- If a support *prevents rotation about an axis*, then the support exerts a *couple moment* on the body about the axis.
- If a body is subjected to more unknown reactions than available equations of equilibrium, then the problem is *statically indeterminate*.
- A stable body requires that the lines of action of the reactive forces do not intersect a common axis and are not parallel to one another.

### Procedure for Analysis

Three-dimensional equilibrium problems for a rigid body can be solved using the following procedure.

#### Free-Body Diagram.

- Draw an outlined shape of the body.
- Show all the forces and couple moments acting on the body.
- Establish the origin of the  $x$ ,  $y$ ,  $z$  axes at a convenient point and orient the axes so that they are parallel to as many of the external forces and moments as possible.
- Label all the loadings and specify their directions. In general, show all the *unknown* components having a *positive sense* along the  $x$ ,  $y$ ,  $z$  axes.
- Indicate the dimensions of the body necessary for computing the moments of forces.

#### Equations of Equilibrium.

- If the  $x$ ,  $y$ ,  $z$  force and moment components seem easy to determine, then apply the six scalar equations of equilibrium; otherwise use the vector equations.
- It is not necessary that the set of axes chosen for force summation coincide with the set of axes chosen for moment summation. Actually, an axis in any arbitrary direction may be chosen for summing forces and moments.
- Choose the direction of an axis for moment summation such that it intersects the lines of action of as many unknown forces as possible. Realize that the moments of forces passing through points on this axis and the moments of forces which are parallel to the axis will then be zero.
- If the solution of the equilibrium equations yields a negative scalar for a force or couple moment magnitude, it indicates that the sense is opposite to that assumed on the free-body diagram.

**EXAMPLE 5.15**

The homogeneous plate shown in Fig. 5–28a has a mass of 100 kg and is subjected to a force and couple moment along its edges. If it is supported in the horizontal plane by a roller at  $A$ , a ball-and-socket joint at  $B$ , and a cord at  $C$ , determine the components of reaction at these supports.

**SOLUTION (SCALAR ANALYSIS)**

**Free-Body Diagram.** There are five unknown reactions acting on the plate, as shown in Fig. 5–28b. Each of these reactions is assumed to act in a positive coordinate direction.

**Equations of Equilibrium.** Since the three-dimensional geometry is rather simple, a *scalar analysis* provides a *direct solution* to this problem. A force summation along each axis yields

$$\Sigma F_x = 0; \quad B_x = 0$$

Ans.

$$\Sigma F_y = 0; \quad B_y = 0$$

Ans.

$$\Sigma F_z = 0; \quad A_z + B_z + T_C - 300 \text{ N} - 981 \text{ N} = 0 \quad (1)$$

Recall that the moment of a force about an axis is equal to the product of the force magnitude and the perpendicular distance (moment arm) from the line of action of the force to the axis. Also, forces that are parallel to an axis or pass through it create no moment about the axis. Hence, summing moments about the positive  $x$  and  $y$  axes, we have

$$\Sigma M_x = 0; \quad T_C(2 \text{ m}) - 981 \text{ N}(1 \text{ m}) + B_z(2 \text{ m}) = 0 \quad (2)$$

$$\Sigma M_y = 0; \quad 300 \text{ N}(1.5 \text{ m}) + 981 \text{ N}(1.5 \text{ m}) - B_z(3 \text{ m}) - A_z(3 \text{ m}) - 200 \text{ N} \cdot \text{m} = 0 \quad (3)$$

The components of the force at  $B$  can be eliminated if moments are summed about the  $x'$  and  $y'$  axes. We obtain

$$\Sigma M_{x'} = 0; \quad 981 \text{ N}(1 \text{ m}) + 300 \text{ N}(2 \text{ m}) - A_z(2 \text{ m}) = 0 \quad (4)$$

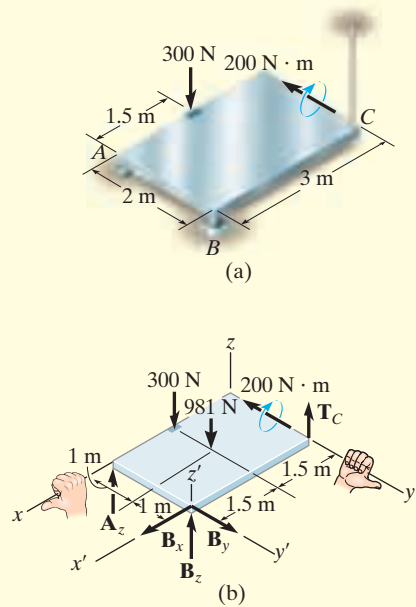
$$\Sigma M_{y'} = 0; \quad -300 \text{ N}(1.5 \text{ m}) - 981 \text{ N}(1.5 \text{ m}) - 200 \text{ N} \cdot \text{m} + T_C(3 \text{ m}) = 0 \quad (5)$$

Solving Eqs. 1 through 3 or the more convenient Eqs. 1, 4, and 5 yields

$$A_z = 790 \text{ N} \quad B_z = -217 \text{ N} \quad T_C = 707 \text{ N} \quad \text{Ans.}$$

The negative sign indicates that  $B_z$  acts downward.

**NOTE:** The solution of this problem does not require a summation of moments about the  $z$  axis. The plate is partially constrained since the supports cannot prevent it from turning about the  $z$  axis if a force is applied to it in the  $x$ - $y$  plane.



**Fig. 5–28**

## EXAMPLE 5.16

Determine the components of reaction that the ball-and-socket joint at  $A$ , the smooth journal bearing at  $B$ , and the roller support at  $C$  exert on the rod assembly in Fig. 5–29a.

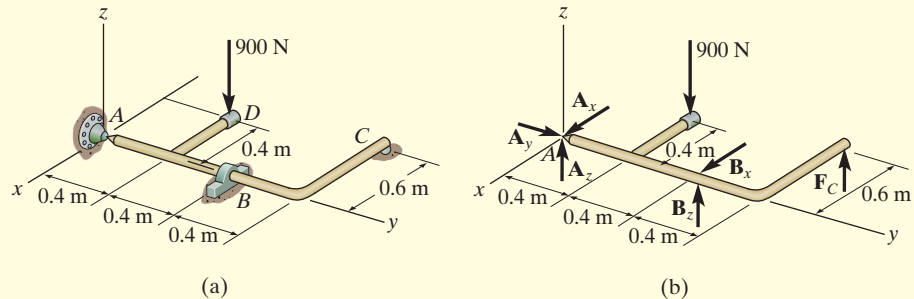


Fig. 5–29

## SOLUTION (SCALAR ANALYSIS)

**Free-Body Diagram.** As shown on the free-body diagram, Fig. 5–29b, the reactive forces of the supports will prevent the assembly from rotating about each coordinate axis, and so the journal bearing at  $B$  only exerts reactive forces on the member. No couple moments are required.

**Equations of Equilibrium.** Because all the forces are either horizontal or vertical, it is convenient to use a scalar analysis. A direct solution for  $A_y$  can be obtained by summing forces along the  $y$  axis.

$$\Sigma F_y = 0; \quad A_y = 0 \quad \text{Ans.}$$

The force  $F_C$  can be determined directly by summing moments about the  $y$  axis.

$$\begin{aligned} \Sigma M_y = 0; \quad F_C(0.6 \text{ m}) - 900 \text{ N}(0.4 \text{ m}) &= 0 \\ F_C &= 600 \text{ N} \quad \text{Ans.} \end{aligned}$$

Using this result,  $B_z$  can be determined by summing moments about the  $x$  axis.

$$\begin{aligned} \Sigma M_x = 0; \quad B_z(0.8 \text{ m}) + 600 \text{ N}(1.2 \text{ m}) - 900 \text{ N}(0.4 \text{ m}) &= 0 \\ B_z &= -450 \text{ N} \quad \text{Ans.} \end{aligned}$$

The negative sign indicates that  $B_z$  acts downward. The force  $B_x$  can be found by summing moments about the  $z$  axis.

$$\Sigma M_z = 0; \quad -B_x(0.8 \text{ m}) = 0 \quad B_x = 0 \quad \text{Ans.}$$

Thus,

$$\Sigma F_x = 0; \quad A_x + 0 = 0 \quad A_x = 0 \quad \text{Ans.}$$

Finally, using the results of  $B_z$  and  $F_C$ .

$$\begin{aligned} \Sigma F_z = 0; \quad A_z + (-450 \text{ N}) + 600 \text{ N} - 900 \text{ N} &= 0 \\ A_z &= 750 \text{ N} \quad \text{Ans.} \end{aligned}$$

**EXAMPLE 5.17**

The boom is used to support the 75-lb flowerpot in Fig. 5–30a. Determine the tension developed in wires  $AB$  and  $AC$ .

**SOLUTION (VECTOR ANALYSIS)**

**Free-Body Diagram.** The free-body diagram of the boom is shown in Fig. 5–30b.

**Equations of Equilibrium.** Here the cable forces are directed at angles with the coordinate axes, so we will use a vector analysis.

$$\begin{aligned}\mathbf{F}_{AB} &= F_{AB} \left( \frac{\mathbf{r}_{AB}}{r_{AB}} \right) = F_{AB} \left( \frac{\{2\mathbf{i} - 6\mathbf{j} + 3\mathbf{k}\} \text{ ft}}{\sqrt{(2 \text{ ft})^2 + (-6 \text{ ft})^2 + (3 \text{ ft})^2}} \right) \\ &= \frac{2}{7} F_{AB} \mathbf{i} - \frac{6}{7} F_{AB} \mathbf{j} + \frac{3}{7} F_{AB} \mathbf{k}\end{aligned}$$

$$\begin{aligned}\mathbf{F}_{AC} &= F_{AC} \left( \frac{\mathbf{r}_{AC}}{r_{AC}} \right) = F_{AC} \left( \frac{\{-2\mathbf{i} - 6\mathbf{j} + 3\mathbf{k}\} \text{ ft}}{\sqrt{(-2 \text{ ft})^2 + (-6 \text{ ft})^2 + (3 \text{ ft})^2}} \right) \\ &= -\frac{2}{7} F_{AC} \mathbf{i} - \frac{6}{7} F_{AC} \mathbf{j} + \frac{3}{7} F_{AC} \mathbf{k}\end{aligned}$$

We can eliminate the force reaction at  $O$  by writing the moment equation of equilibrium about point  $O$ .

$$\begin{aligned}\Sigma \mathbf{M}_O &= \mathbf{0}; & \mathbf{r}_A \times (\mathbf{F}_{AB} + \mathbf{F}_{AC} + \mathbf{W}) &= \mathbf{0} \\ (6\mathbf{j}) \times \left[ \left( \frac{2}{7} F_{AB} \mathbf{i} - \frac{6}{7} F_{AB} \mathbf{j} + \frac{3}{7} F_{AB} \mathbf{k} \right) + \left( -\frac{2}{7} F_{AC} \mathbf{i} - \frac{6}{7} F_{AC} \mathbf{j} + \frac{3}{7} F_{AC} \mathbf{k} \right) + (-75\mathbf{k}) \right] &= \mathbf{0}\end{aligned}$$

$$\left( \frac{18}{7} F_{AB} + \frac{18}{7} F_{AC} - 450 \right) \mathbf{i} + \left( -\frac{12}{7} F_{AB} + \frac{12}{7} F_{AC} \right) \mathbf{k} = \mathbf{0}$$

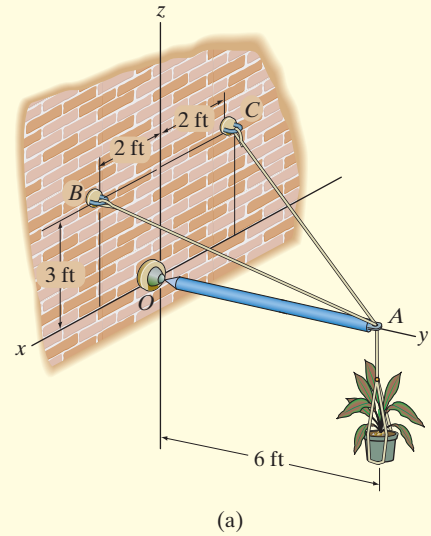
$$\Sigma M_x = 0; \quad \frac{18}{7} F_{AB} + \frac{18}{7} F_{AC} - 450 = 0$$

$$\Sigma M_y = 0; \quad 0 = 0$$

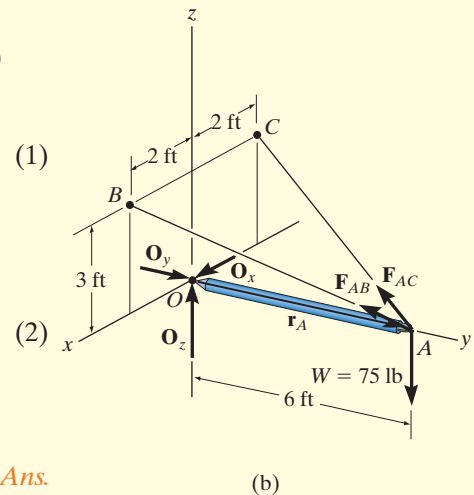
$$\Sigma M_z = 0; \quad -\frac{12}{7} F_{AB} + \frac{12}{7} F_{AC} = 0$$

Solving Eqs. (1) and (2) simultaneously,

$$F_{AB} = F_{AC} = 87.5 \text{ lb}$$



**Fig. 5–30**



*Ans.*

## EXAMPLE 5.18

Rod  $AB$  shown in Fig. 5–31a is subjected to the 200-N force. Determine the reactions at the ball-and-socket joint  $A$  and the tension in the cables  $BD$  and  $BE$ . The collar at  $C$  is fixed to the rod.

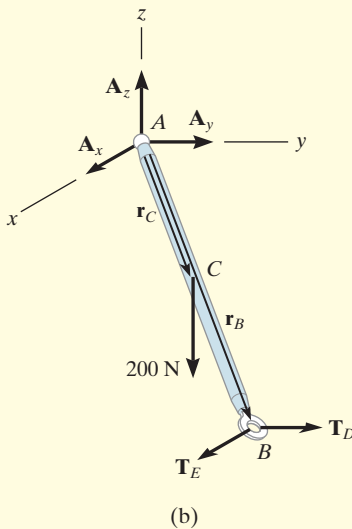
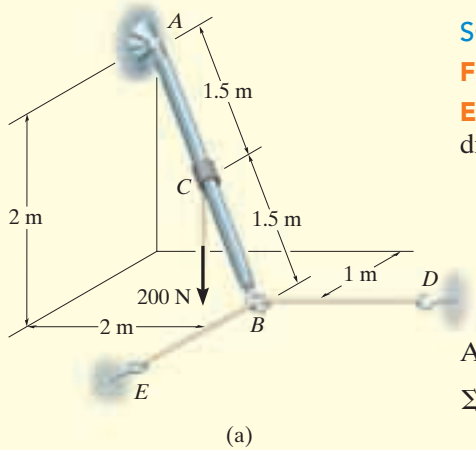


Fig. 5–31

## SOLUTION (VECTOR ANALYSIS)

**Free-Body Diagram.** Fig. 5–31b.

**Equations of Equilibrium.** Representing each force on the free-body diagram in Cartesian vector form, we have

$$\mathbf{F}_A = A_x \mathbf{i} + A_y \mathbf{j} + A_z \mathbf{k}$$

$$\mathbf{T}_E = T_E \mathbf{i}$$

$$\mathbf{T}_D = T_D \mathbf{j}$$

$$\mathbf{F} = \{-200\mathbf{k}\} \text{ N}$$

Applying the force equation of equilibrium.

$$\Sigma \mathbf{F} = \mathbf{0};$$

$$\mathbf{F}_A + \mathbf{T}_E + \mathbf{T}_D + \mathbf{F} = \mathbf{0}$$

$$(A_x + T_E)\mathbf{i} + (A_y + T_D)\mathbf{j} + (A_z - 200)\mathbf{k} = \mathbf{0}$$

$$\Sigma F_x = 0; \quad A_x + T_E = 0 \quad (1)$$

$$\Sigma F_y = 0; \quad A_y + T_D = 0 \quad (2)$$

$$\Sigma F_z = 0; \quad A_z - 200 = 0 \quad (3)$$

Summing moments about point  $A$  yields

$$\Sigma \mathbf{M}_A = \mathbf{0}; \quad \mathbf{r}_C \times \mathbf{F} + \mathbf{r}_B \times (\mathbf{T}_E + \mathbf{T}_D) = \mathbf{0}$$

Since  $\mathbf{r}_C = \frac{1}{2}\mathbf{r}_B$ , then

$$(0.5\mathbf{i} + 1\mathbf{j} - 1\mathbf{k}) \times (-200\mathbf{k}) + (1\mathbf{i} + 2\mathbf{j} - 2\mathbf{k}) \times (T_E \mathbf{i} + T_D \mathbf{j}) = \mathbf{0}$$

Expanding and rearranging terms gives

$$(2T_D - 200)\mathbf{i} + (-2T_E + 100)\mathbf{j} + (T_D - 2T_E)\mathbf{k} = \mathbf{0}$$

$$\Sigma M_x = 0; \quad 2T_D - 200 = 0 \quad (4)$$

$$\Sigma M_y = 0; \quad -2T_E + 100 = 0 \quad (5)$$

$$\Sigma M_z = 0; \quad T_D - 2T_E = 0 \quad (6)$$

Solving Eqs. 1 through 5, we get

$$T_D = 100 \text{ N} \quad \text{Ans.}$$

$$T_E = 50 \text{ N} \quad \text{Ans.}$$

$$A_x = -50 \text{ N} \quad \text{Ans.}$$

$$A_y = -100 \text{ N} \quad \text{Ans.}$$

$$A_z = 200 \text{ N} \quad \text{Ans.}$$

**NOTE:** The negative sign indicates that  $A_x$  and  $A_y$  have a sense which is opposite to that shown on the free-body diagram, Fig. 5–31b. Also, notice that Eqs. 1–6 can be set up *directly* using a scalar analysis.

**EXAMPLE 5.19**

The bent rod in Fig. 5–32a is supported at  $A$  by a journal bearing, at  $D$  by a ball-and-socket joint, and at  $B$  by means of cable  $BC$ . Using only *one equilibrium equation*, obtain a direct solution for the tension in cable  $BC$ . The bearing at  $A$  is capable of exerting force components only in the  $z$  and  $y$  directions since it is properly aligned on the shaft. In other words, no couple moments are required at this support.

**SOLUTION (VECTOR ANALYSIS)**

**Free-Body Diagram.** As shown in Fig. 5–32b, there are six unknowns.

**Equations of Equilibrium.** The cable tension  $\mathbf{T}_B$  may be obtained *directly* by summing moments about an axis that passes through points  $D$  and  $A$ . Why? The direction of this axis is defined by the unit vector  $\mathbf{u}$ , where

$$\begin{aligned}\mathbf{u} &= \frac{\mathbf{r}_{DA}}{r_{DA}} = -\frac{1}{\sqrt{2}}\mathbf{i} - \frac{1}{\sqrt{2}}\mathbf{j} \\ &= -0.7071\mathbf{i} - 0.7071\mathbf{j}\end{aligned}$$

Hence, the sum of the moments about this axis is zero provided

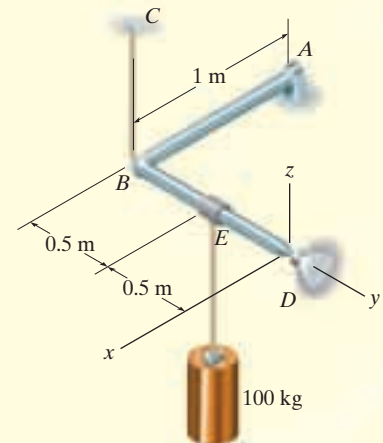
$$\Sigma M_{DA} = \mathbf{u} \cdot \Sigma(\mathbf{r} \times \mathbf{F}) = 0$$

Here  $\mathbf{r}$  represents a position vector drawn from *any point* on the axis  $DA$  to any point on the line of action of force  $\mathbf{F}$  (see Eq. 4–11). With reference to Fig. 5–32b, we can therefore write

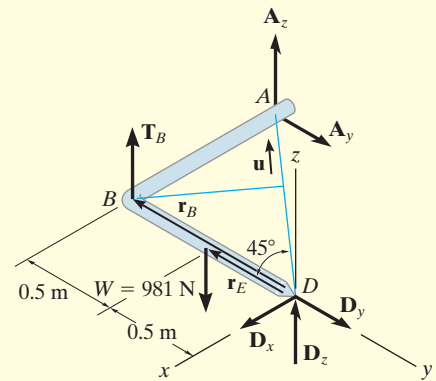
$$\begin{aligned}\mathbf{u} \cdot (\mathbf{r}_B \times \mathbf{T}_B + \mathbf{r}_E \times \mathbf{W}) &= 0 \\ (-0.7071\mathbf{i} - 0.7071\mathbf{j}) \cdot [(-1\mathbf{j}) \times (T_B\mathbf{k}) \\ &\quad + (-0.5\mathbf{j}) \times (-981\mathbf{k})] = 0 \\ (-0.7071\mathbf{i} - 0.7071\mathbf{j}) \cdot [(-T_B + 490.5)\mathbf{i}] &= 0 \\ -0.7071(-T_B + 490.5) + 0 + 0 &= 0 \\ T_B &= 490.5 \text{ N} \quad \text{Ans.}\end{aligned}$$

**NOTE:** Since the moment arms from the axis to  $\mathbf{T}_B$  and  $\mathbf{W}$  are easy to obtain, we can also determine this result using a scalar analysis. As shown in Fig. 5–32b,

$$\begin{aligned}\Sigma M_{DA} &= 0; \quad T_B(1 \text{ m} \sin 45^\circ) - 981 \text{ N}(0.5 \text{ m} \sin 45^\circ) = 0 \\ T_B &= 490.5 \text{ N} \quad \text{Ans.}\end{aligned}$$



(a)

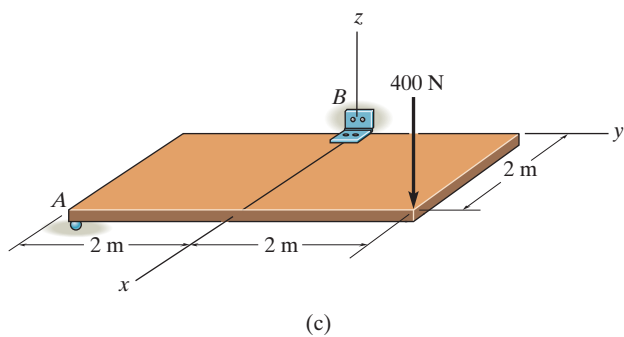
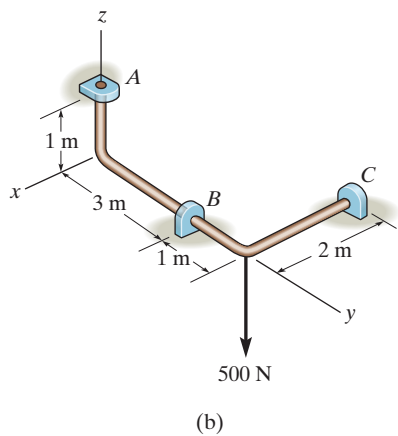
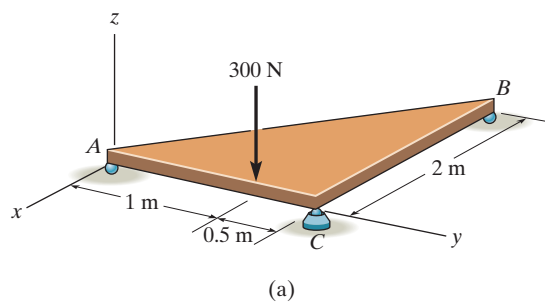


(b)

**Fig. 5–32**

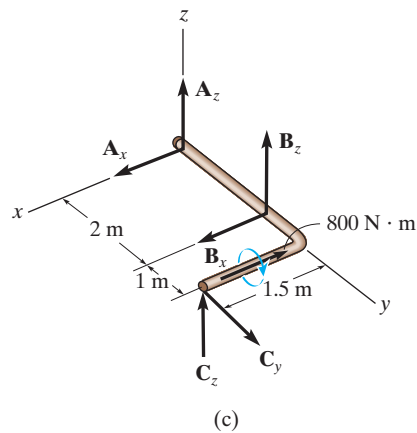
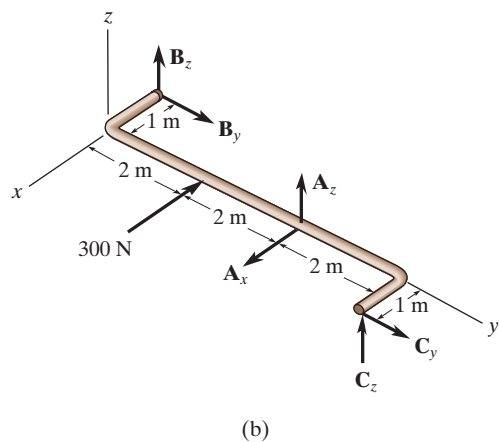
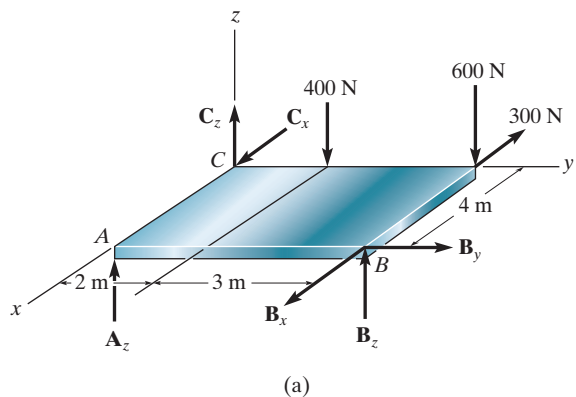
## PRELIMINARY PROBLEMS

**P5-2.** Draw the free-body diagram of each object.



**Prob. P5-2**

**P5-3.** In each case, write the moment equations about the  $x$ ,  $y$ , and  $z$  axes.

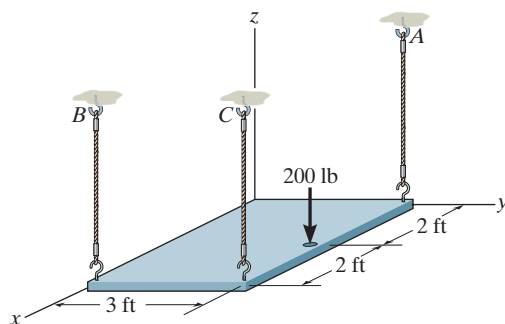


**Prob. P5-3**

## FUNDAMENTAL PROBLEMS

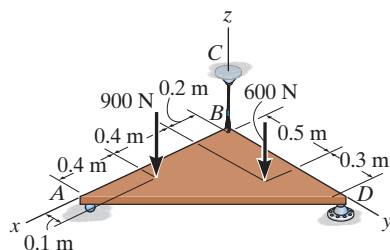
**All problem solutions must include an FBD.**

**F5-7.** The uniform plate has a weight of 500 lb. Determine the tension in each of the supporting cables.



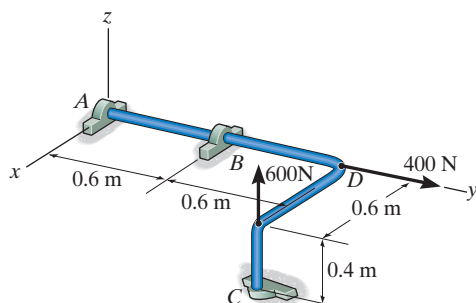
**Prob. F5-7**

**F5-8.** Determine the reactions at the roller support  $A$ , the ball-and-socket joint  $D$ , and the tension in cable  $BC$  for the plate.



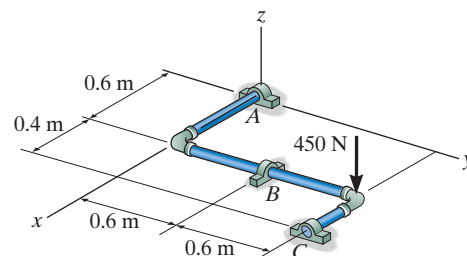
**Prob. F5-8**

**F5-9.** The rod is supported by smooth journal bearings at  $A$ ,  $B$ , and  $C$  and is subjected to the two forces. Determine the reactions at these supports.



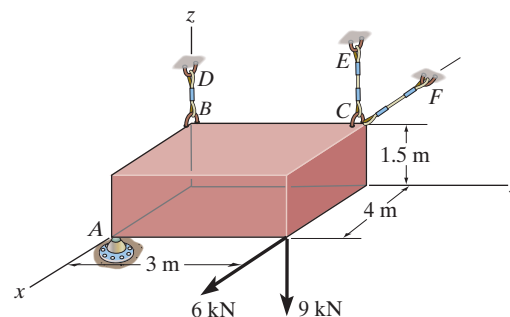
**Prob. F5-9**

**F5-10.** Determine the support reactions at the smooth journal bearings  $A$ ,  $B$ , and  $C$  of the pipe assembly.



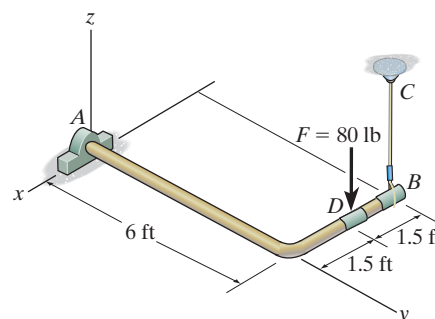
**Prob. F5-10**

**F5-11.** Determine the force developed in the short link  $BD$ , and the tension in the cords  $CE$  and  $CF$ , and the reactions of the ball-and-socket joint  $A$  on the block.



**Prob. F5-11**

**F5-12.** Determine the components of reaction that the thrust bearing  $A$  and cable  $BC$  exert on the bar.



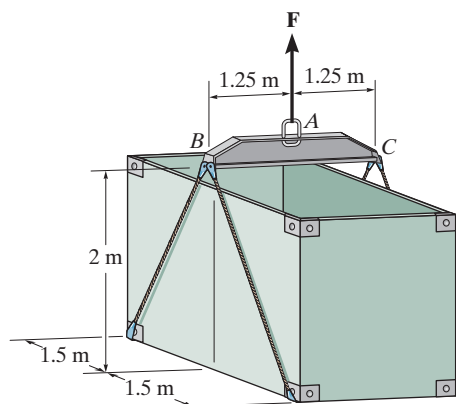
**Prob. F5-12**



## PROBLEMS

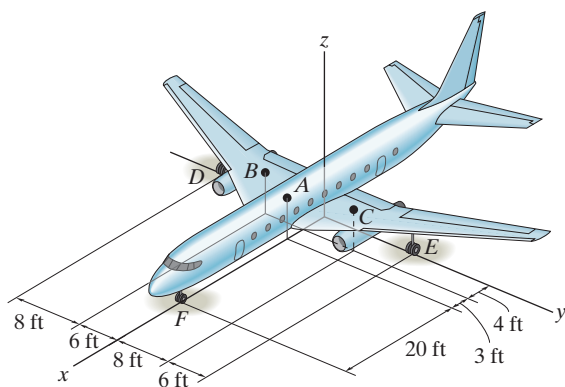
*All problem solutions must include an FBD.*

**5-62.** The uniform load has a mass of 600 kg and is lifted using a uniform 30-kg strongback beam  $BAC$  and four ropes as shown. Determine the tension in each rope and the force that must be applied at  $A$ .



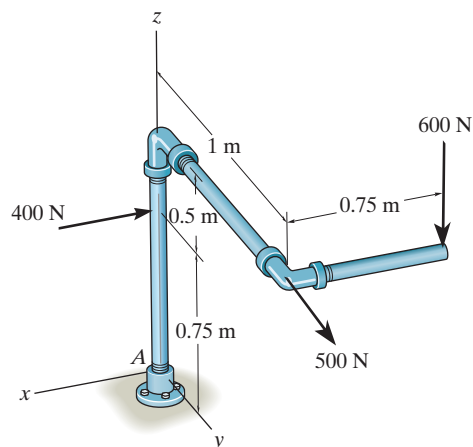
**Prob. 5-62**

**5-63.** Due to an unequal distribution of fuel in the wing tanks, the centers of gravity for the airplane fuselage  $A$  and wings  $B$  and  $C$  are located as shown. If these components have weights  $W_A = 45\,000$  lb,  $W_B = 8000$  lb, and  $W_C = 6000$  lb, determine the normal reactions of the wheels  $D$ ,  $E$ , and  $F$  on the ground.



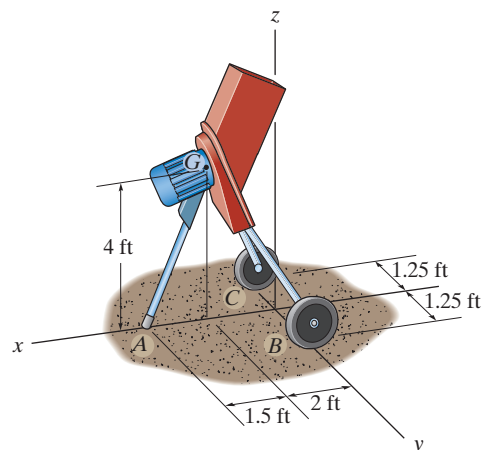
**Prob. 5-63**

**\*5-64.** Determine the components of reaction at the fixed support  $A$ . The 400 N, 500 N, and 600 N forces are parallel to the  $x$ ,  $y$ , and  $z$  axes, respectively.



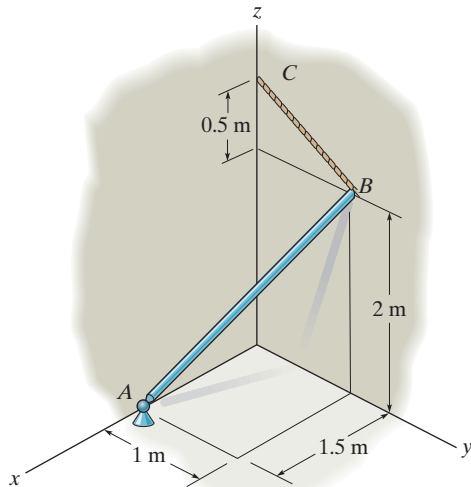
**Prob. 5-64**

**5-65.** The 50-lb mulching machine has a center of gravity at  $G$ . Determine the vertical reactions at the wheels  $C$  and  $B$  and the smooth contact point  $A$ .



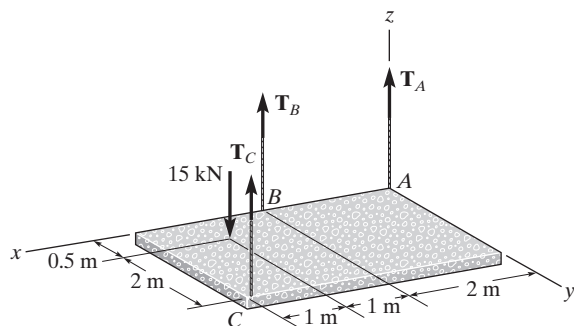
**Prob. 5-65**

**5–66.** The smooth uniform rod  $AB$  is supported by a ball-and-socket joint at  $A$ , the wall at  $B$ , and cable  $BC$ . Determine the components of reaction at  $A$ , the tension in the cable, and the normal reaction at  $B$  if the rod has a mass of 20 kg.



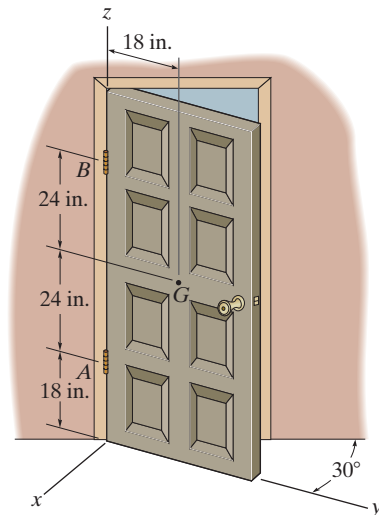
**Prob. 5–66**

**5–67.** The uniform concrete slab has a mass of 2400 kg. Determine the tension in each of the three parallel supporting cables when the slab is held in the horizontal plane as shown.



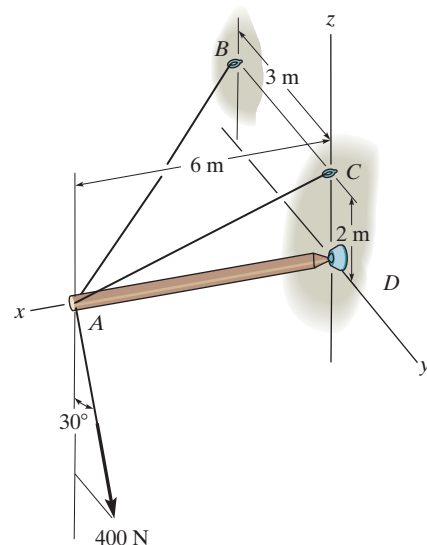
**Prob. 5–67**

**\*5–68.** The 100-lb door has its center of gravity at  $G$ . Determine the components of reaction at hinges  $A$  and  $B$  if hinge  $B$  resists only forces in the  $x$  and  $y$  directions and  $A$  resists forces in the  $x$ ,  $y$ ,  $z$  directions.



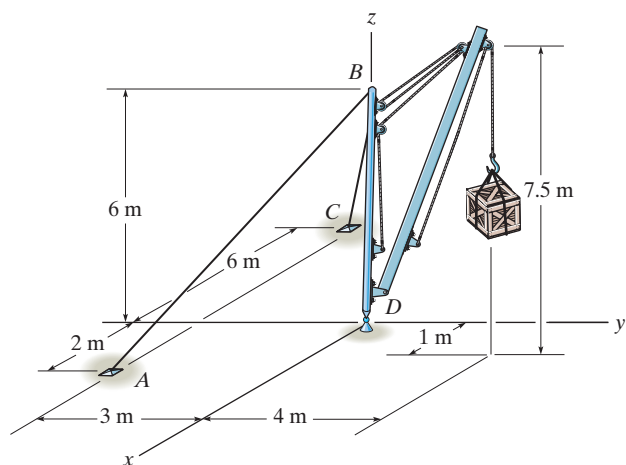
**Prob. 5–68**

**5–69.** Determine the tension in each cable and the components of reaction at  $D$  needed to support the load.



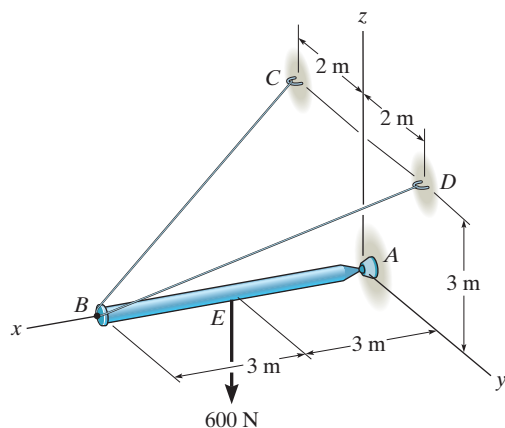
**Prob. 5–69**

**5-70.** The stiff-leg derrick used on ships is supported by a ball-and-socket joint at  $D$  and two cables  $BA$  and  $BC$ . The cables are attached to a smooth collar ring at  $B$ , which allows rotation of the derrick about  $z$  axis. If the derrick supports a crate having a mass of 200 kg, determine the tension in the cables and the  $x, y, z$  components of reaction at  $D$ .



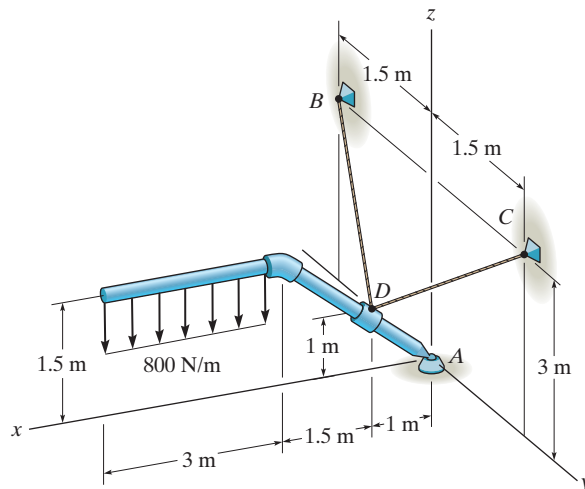
**Prob. 5-70**

**5-71.** Determine the components of reaction at the ball-and-socket joint  $A$  and the tension in each cable necessary for equilibrium of the rod.



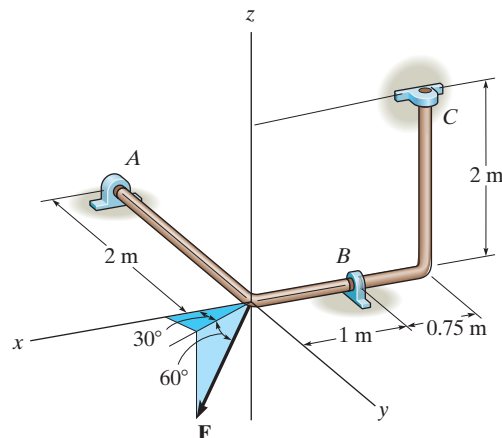
**Prob. 5-71**

**\*5-72.** Determine the components of reaction at the ball-and-socket joint  $A$  and the tension in the supporting cables  $DB$  and  $DC$ .



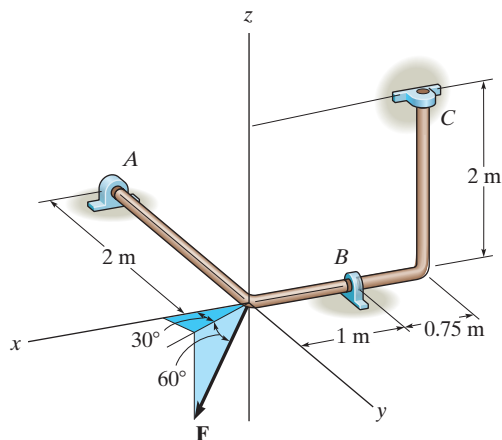
**Prob. 5-72**

**5-73.** The bent rod is supported at  $A, B$ , and  $C$  by smooth journal bearings. Determine the components of reaction at the bearings if the rod is subjected to the force  $F = 800$  N. The bearings are in proper alignment and exert only force reactions on the rod.



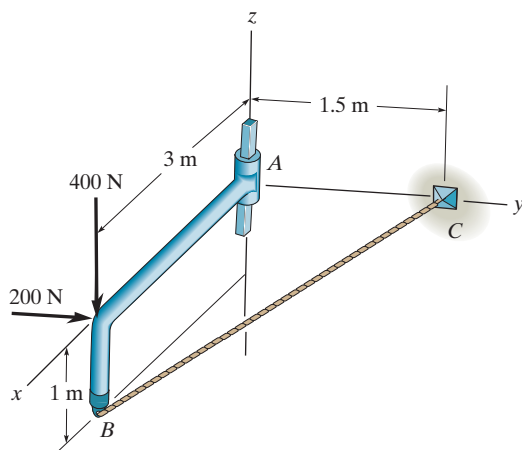
**Prob. 5-73**

**5-74.** The bent rod is supported at  $A$ ,  $B$ , and  $C$  by smooth journal bearings. Determine the magnitude of  $\mathbf{F}$  which will cause the positive  $x$  component of reaction at the bearing  $C$  to be  $C_x = 50$  N. The bearings are in proper alignment and exert only force reactions on the rod.



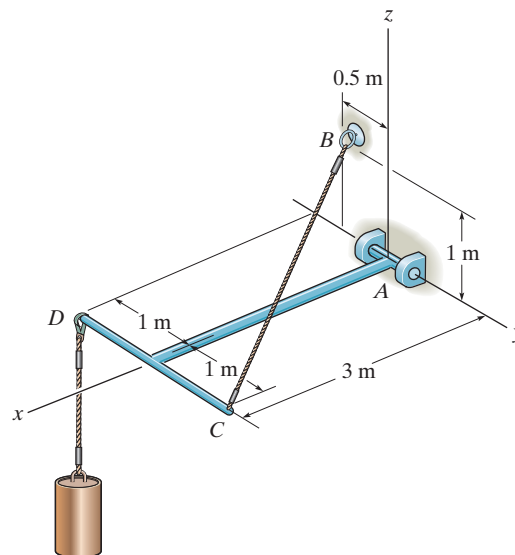
**Prob. 5-74**

**5-75.** Member  $AB$  is supported by a cable  $BC$  and at  $A$  by a square rod which fits loosely through the square hole in the collar fixed to the member as shown. Determine the components of reaction at  $A$  and the tension in the cable needed to hold the rod in equilibrium.



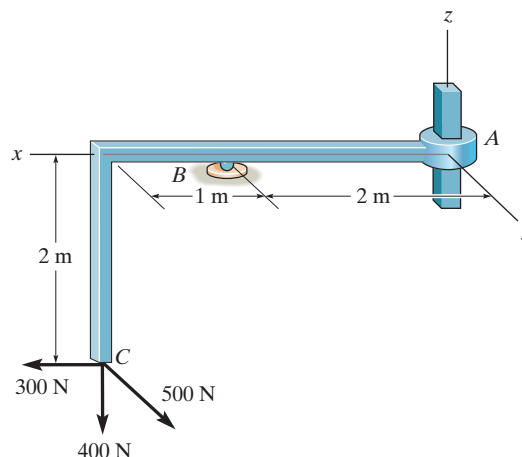
**Prob. 5-75**

**\*5-76.** The member is supported by a pin at  $A$  and cable  $BC$ . Determine the components of reaction at these supports if the cylinder has a mass of 40 kg.



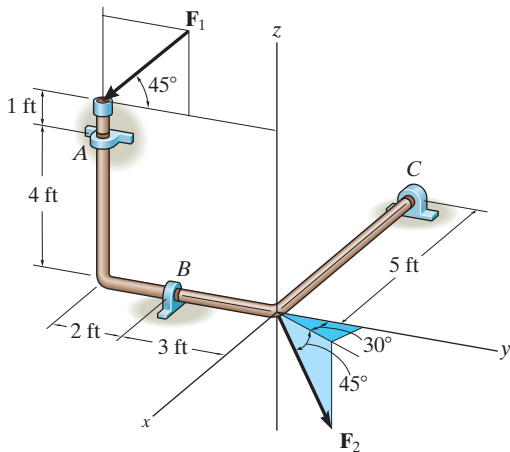
**Prob. 5-76**

**5-77.** The member is supported by a square rod which fits loosely through the smooth square hole of the attached collar at  $A$  and by a roller at  $B$ . Determine the components of reaction at these supports when the member is subjected to the loading shown.



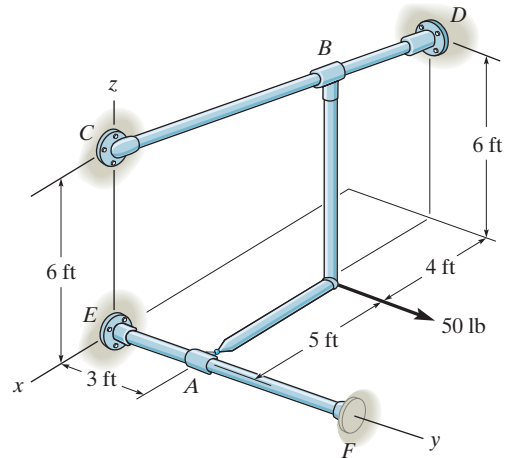
**Prob. 5-77**

**5-78.** The bent rod is supported at  $A$ ,  $B$ , and  $C$  by smooth journal bearings. Compute the  $x$ ,  $y$ ,  $z$  components of reaction at the bearings if the rod is subjected to forces  $F_1 = 300$  lb and  $F_2 = 250$  lb.  $F_1$  lies in the  $y$ - $z$  plane. The bearings are in proper alignment and exert only force reactions on the rod.



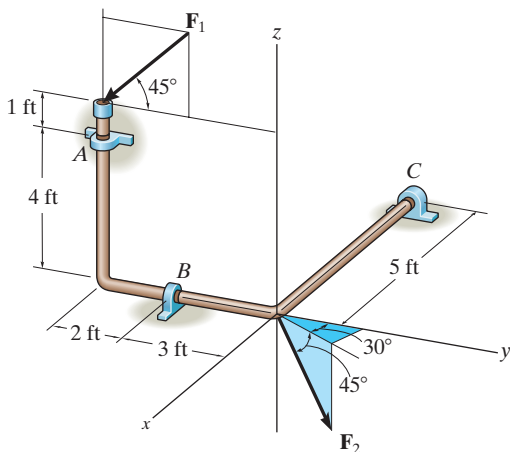
Prob. 5-78

**\*5-80.** The bar  $AB$  is supported by two smooth collars. At  $A$  the connection is with a ball-and-socket joint and at  $B$  it is a rigid attachment. If a 50-lb load is applied to the bar, determine the  $x$ ,  $y$ ,  $z$  components of reaction at  $A$  and  $B$ .



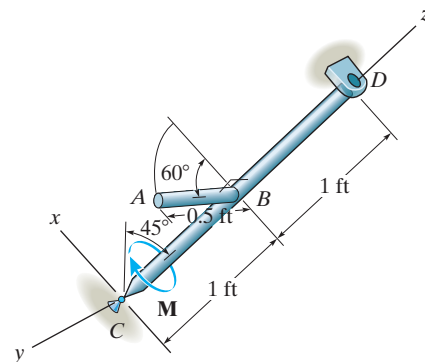
Prob. 5-80

**5-79.** The bent rod is supported at  $A$ ,  $B$ , and  $C$  by smooth journal bearings. Determine the magnitude of  $F_2$  which will cause the reaction  $C_y$  at the bearing  $C$  to be equal to zero. The bearings are in proper alignment and exert only force reactions on the rod. Set  $F_1 = 300$  lb.



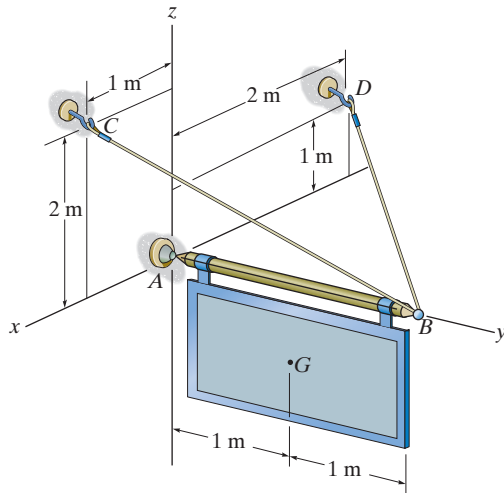
Prob. 5-79

**5-81.** The rod has a weight of 6 lb/ft. If it is supported by a ball-and-socket joint at  $C$  and a journal bearing at  $D$ , determine the  $x$ ,  $y$ ,  $z$  components of reaction at these supports and the moment  $M$  that must be applied along the axis of the rod to hold it in the position shown.



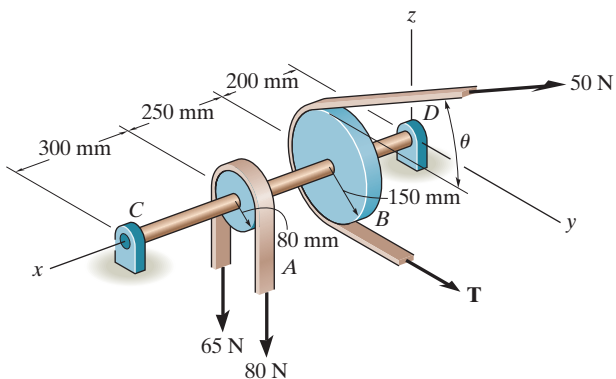
Prob. 5-81

**5-82.** The sign has a mass of 100 kg with center of mass at  $G$ . Determine the  $x$ ,  $y$ ,  $z$  components of reaction at the ball-and-socket joint  $A$  and the tension in wires  $BC$  and  $BD$ .



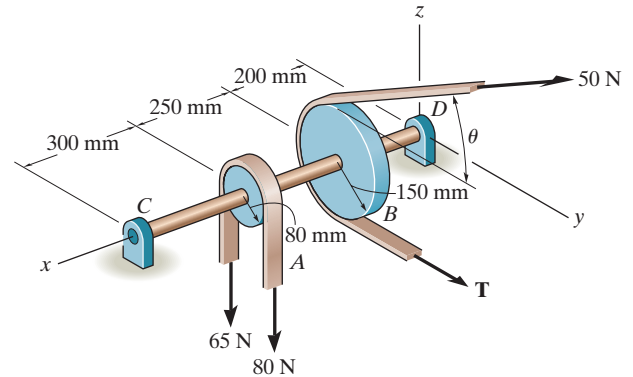
**Prob. 5-82**

**5-83.** Both pulleys are fixed to the shaft and as the shaft turns with constant angular velocity, the power of pulley  $A$  is transmitted to pulley  $B$ . Determine the horizontal tension  $T$  in the belt on pulley  $B$  and the  $x$ ,  $y$ ,  $z$  components of reaction at the journal bearing  $C$  and thrust bearing  $D$  if  $\theta = 0^\circ$ . The bearings are in proper alignment and exert only force reactions on the shaft.



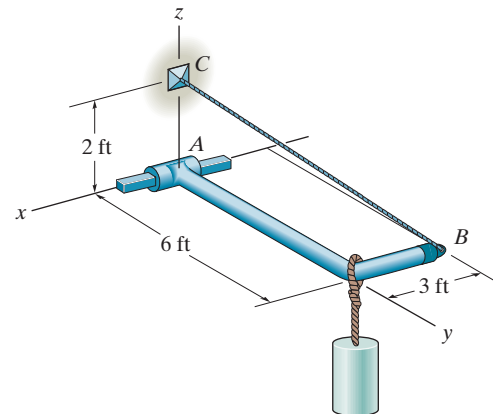
**Prob. 5-83**

**\*5-84.** Both pulleys are fixed to the shaft and as the shaft turns with constant angular velocity, the power of pulley  $A$  is transmitted to pulley  $B$ . Determine the horizontal tension  $T$  in the belt on pulley  $B$  and the  $x$ ,  $y$ ,  $z$  components of reaction at the journal bearing  $C$  and thrust bearing  $D$  if  $\theta = 45^\circ$ . The bearings are in proper alignment and exert only force reactions on the shaft.



**Prob. 5-84**

**5-85.** Member  $AB$  is supported by a cable  $BC$  and at  $A$  by a square rod which fits loosely through the square hole at the end joint of the member as shown. Determine the components of reaction at  $A$  and the tension in the cable needed to hold the 800-lb cylinder in equilibrium.



**Prob. 5-85**

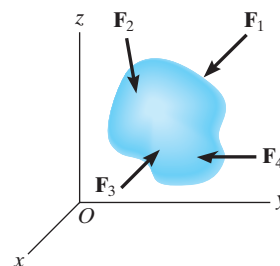
## CHAPTER REVIEW

### Equilibrium

A body in equilibrium is at rest or can translate with constant velocity.

$$\Sigma \mathbf{F} = \mathbf{0}$$

$$\Sigma \mathbf{M} = \mathbf{0}$$



### Two Dimensions

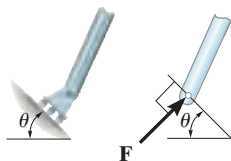
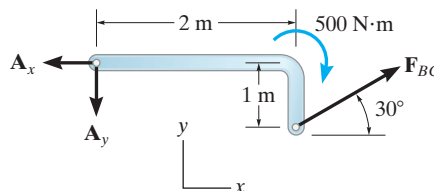
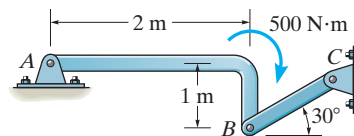
Before analyzing the equilibrium of a body, it is first necessary to draw its free-body diagram. This is an outlined shape of the body, which shows all the forces and couple moments that act on it.

Couple moments can be placed anywhere on a free-body diagram since they are free vectors. Forces can act at any point along their line of action since they are sliding vectors.

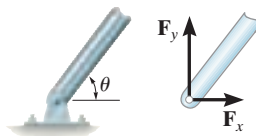
Angles used to resolve forces, and dimensions used to take moments of the forces, should also be shown on the free-body diagram.

Some common types of supports and their reactions are shown below in two dimensions.

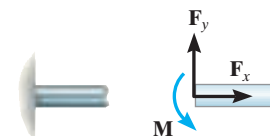
Remember that a support will exert a force on the body in a particular direction if it prevents translation of the body in that direction, and it will exert a couple moment on the body if it prevents rotation.



roller



smooth pin or hinge



fixed support

The three scalar equations of equilibrium can be applied when solving problems in two dimensions, since the geometry is easy to visualize.

$$\Sigma F_x = 0$$

$$\Sigma F_y = 0$$

$$\Sigma M_O = 0$$

For the most direct solution, try to sum forces along an axis that will eliminate as many unknown forces as possible. Sum moments about a point  $A$  that passes through the line of action of as many unknown forces as possible.

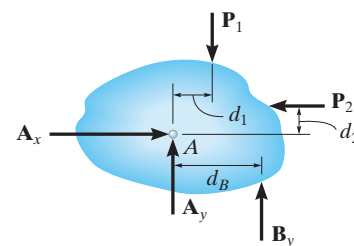
$$\Sigma F_x = 0;$$

$$A_x - P_2 = 0 \quad A_x = P_2$$

$$\Sigma M_A = 0;$$

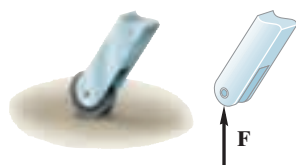
$$P_2 d_2 + B_y d_B - P_1 d_1 = 0$$

$$B_y = \frac{P_1 d_1 - P_2 d_2}{d_B}$$



### Three Dimensions

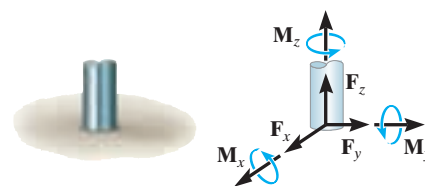
Some common types of supports and their reactions are shown here in three dimensions.



roller



ball and socket



fixed support

In three dimensions, it is often advantageous to use a Cartesian vector analysis when applying the equations of equilibrium. To do this, first express each known and unknown force and couple moment shown on the free-body diagram as a Cartesian vector. Then set the force summation equal to zero. Take moments about a point  $O$  that lies on the line of action of as many unknown force components as possible. From point  $O$  direct position vectors to each force, and then use the cross product to determine the moment of each force.

The six scalar equations of equilibrium are established by setting the respective  $\mathbf{i}$ ,  $\mathbf{j}$ , and  $\mathbf{k}$  components of these force and moment summations equal to zero.

$$\Sigma \mathbf{F} = \mathbf{0}$$

$$\Sigma \mathbf{M}_O = \mathbf{0}$$

$$\Sigma F_x = 0$$

$$\Sigma F_y = 0$$

$$\Sigma F_z = 0$$

$$\Sigma M_x = 0$$

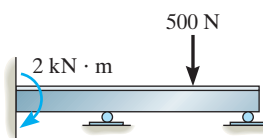
$$\Sigma M_y = 0$$

$$\Sigma M_z = 0$$

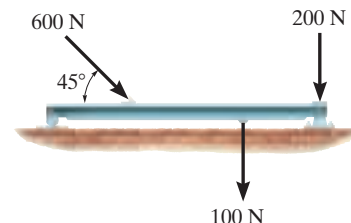
### Determinacy and Stability

If a body is supported by a minimum number of constraints to ensure equilibrium, then it is statically determinate. If it has more constraints than required, then it is statically indeterminate.

To properly constrain the body, the reactions must not all be parallel to one another or concurrent.



Statically indeterminate,  
five reactions, three  
equilibrium equations



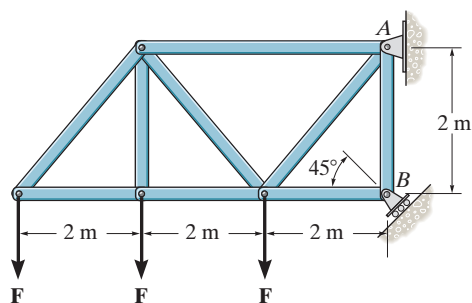
Proper constraint, statically determinate



## REVIEW PROBLEMS

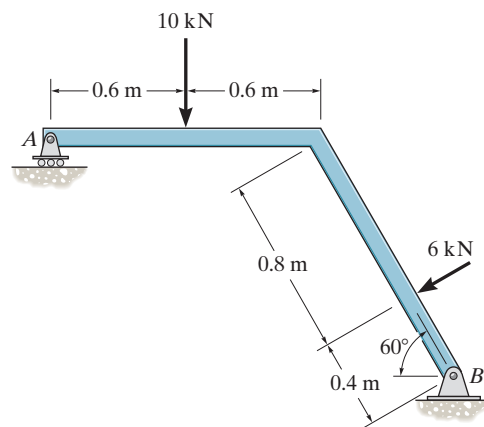
All problem solutions must include an FBD.

**R5-1.** If the roller at  $B$  can sustain a maximum load of 3 kN, determine the largest magnitude of each of the three forces  $\mathbf{F}$  that can be supported by the truss.



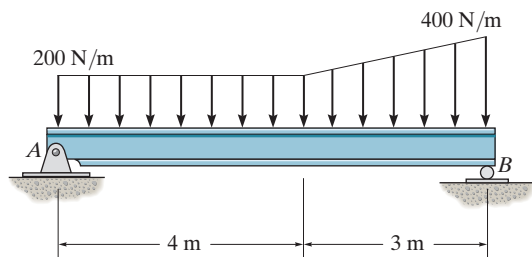
**Prob. R5-1**

**R5-3.** Determine the normal reaction at the roller  $A$  and horizontal and vertical components at pin  $B$  for equilibrium of the member.



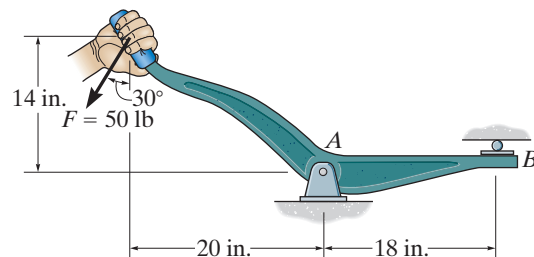
**Prob. R5-3**

**R5-2.** Determine the reactions at the supports  $A$  and  $B$  for equilibrium of the beam.



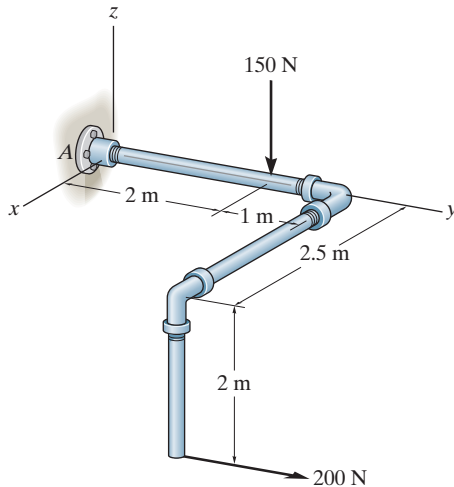
**Prob. R5-2**

**R5-4.** Determine the horizontal and vertical components of reaction at the pin at  $A$  and the reaction of the roller at  $B$  on the lever.



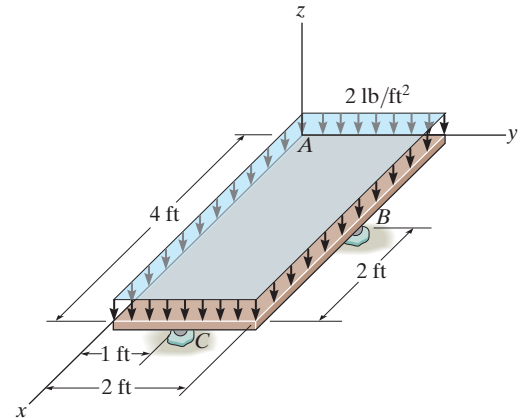
**Prob. R5-4**

**R5-5.** Determine the  $x, y, z$  components of reaction at the fixed wall  $A$ . The 150-N force is parallel to the  $z$  axis and the 200-N force is parallel to the  $y$  axis.



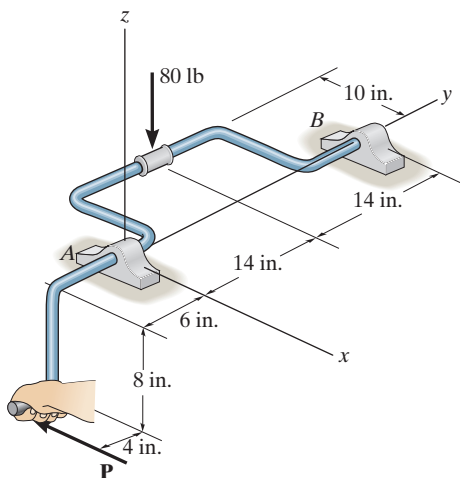
**Prob. R5-5**

**R5-7.** Determine the  $x, y, z$  components of reaction at the ball supports  $B$  and  $C$  and the ball-and-socket  $A$  (not shown) for the uniformly loaded plate.



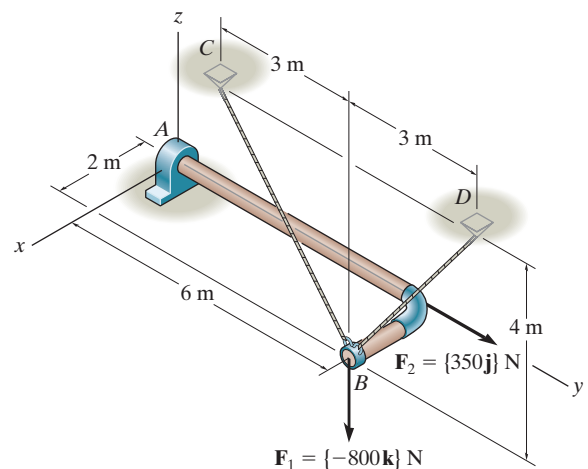
**Prob. R5-7**

**R5-6.** A vertical force of 80 lb acts on the crankshaft. Determine the horizontal equilibrium force  $\mathbf{P}$  that must be applied to the handle and the  $x, y, z$  components of reaction at the journal bearing  $A$  and thrust bearing  $B$ . The bearings are properly aligned and exert only force reactions on the shaft.



**Prob. R5-6**

**R5-8.** Determine the  $x$  and  $z$  components of reaction at the journal bearing  $A$  and the tension in cords  $BC$  and  $BD$  necessary for equilibrium of the rod.



**Prob. R5-8**

# Chapter 6



(© Tim Scrivener/Alamy)

In order to design the many parts of this boom assembly it is required that we know the forces that they must support. In this chapter we will show how to analyze such structures using the equations of equilibrium.

# Structural Analysis

## CHAPTER OBJECTIVES

- To show how to determine the forces in the members of a truss using the method of joints and the method of sections.
- To analyze the forces acting on the members of frames and machines composed of pin-connected members.

## 6.1 Simple Trusses

A **truss** is a structure composed of slender members joined together at their end points. The members commonly used in construction consist of wooden struts or metal bars. In particular, *planar* trusses lie in a single plane and are often used to support roofs and bridges. The truss shown in Fig. 6–1a is an example of a typical roof-supporting truss. In this figure, the roof load is transmitted to the truss *at the joints* by means of a series of *purlins*. Since this loading acts in the same plane as the truss, Fig. 6–1b, the analysis of the forces developed in the truss members will be two-dimensional.

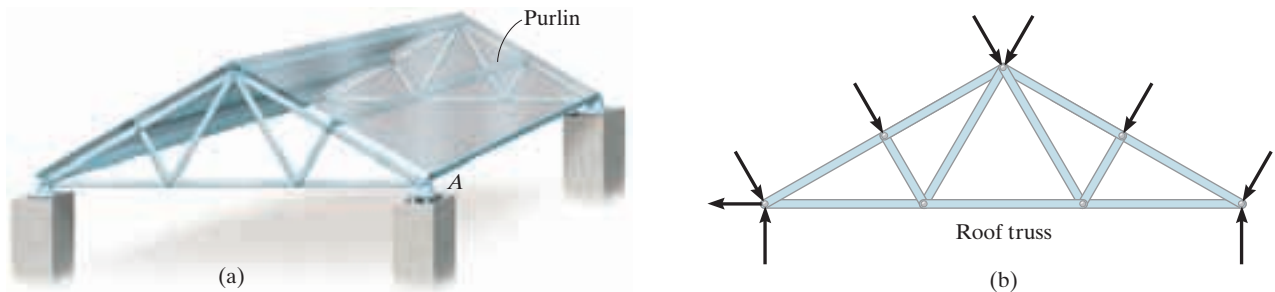


Fig. 6–1

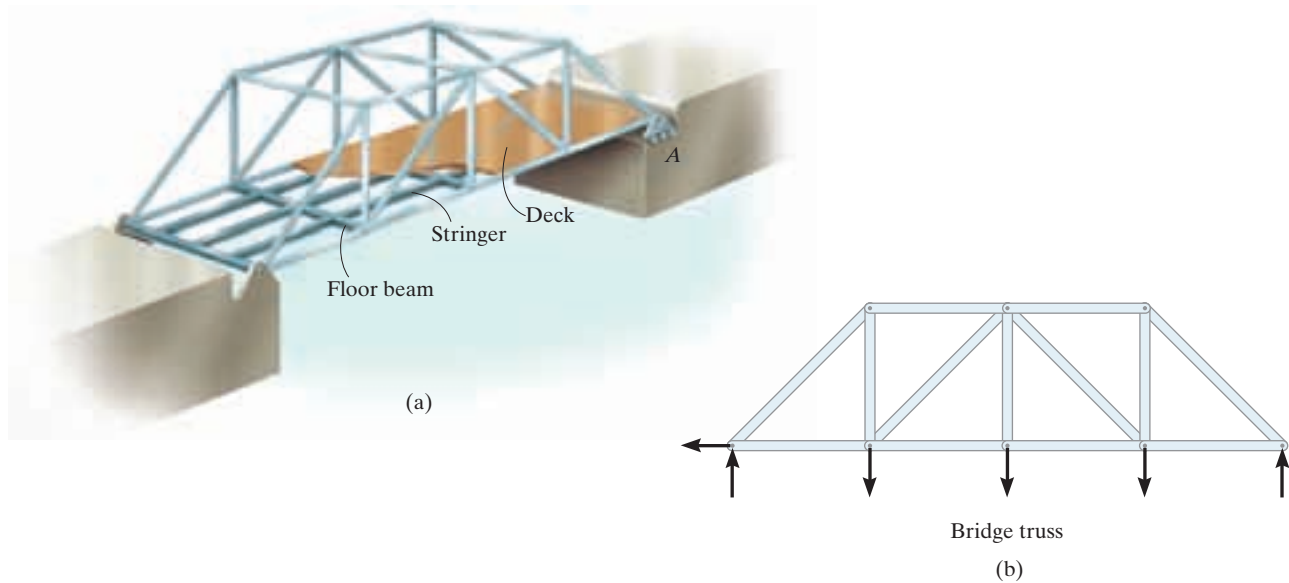


Fig. 6-2

In the case of a bridge, such as shown in Fig. 6-2a, the load on the deck is first transmitted to *stringers*, then to *floor beams*, and finally to the *joints* of the two supporting side trusses. Like the roof truss, the bridge truss loading is also coplanar, Fig. 6-2b.

When bridge or roof trusses extend over large distances, a rocker or roller is commonly used for supporting one end, for example, joint *A* in Figs. 6-1a and 6-2a. This type of support allows freedom for expansion or contraction of the members due to a change in temperature or application of loads.

**Assumptions for Design.** To design both the members and the connections of a truss, it is necessary first to determine the *force* developed in each member when the truss is subjected to a given loading. To do this we will make two important assumptions:

- **All loadings are applied at the joints.** In most situations, such as for bridge and roof trusses, this assumption is true. Frequently the weight of the members is neglected because the force supported by each member is usually much larger than its weight. However, if the weight is to be included in the analysis, it is generally satisfactory to apply it as a vertical force, with half of its magnitude applied at each end of the member.
- **The members are joined together by smooth pins.** The joint connections are usually formed by bolting or welding the ends of the members to a common plate, called a *gusset plate*, as shown in Fig. 6-3a, or by simply passing a large bolt or pin through each of the members, Fig. 6-3b. We can assume these connections act as pins provided the center lines of the joining members are *concurrent*, as in Fig. 6-3.

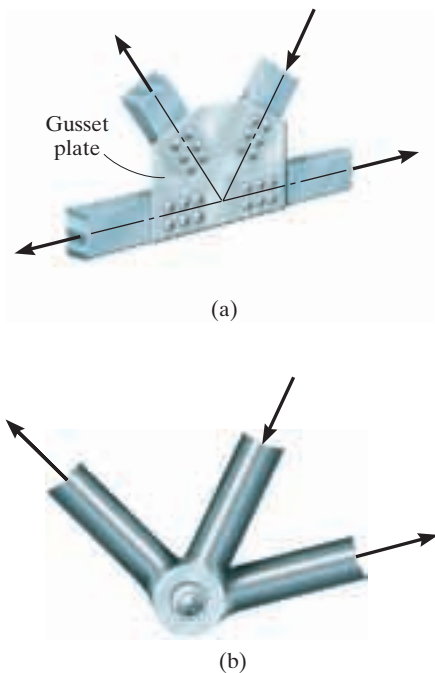


Fig. 6-3

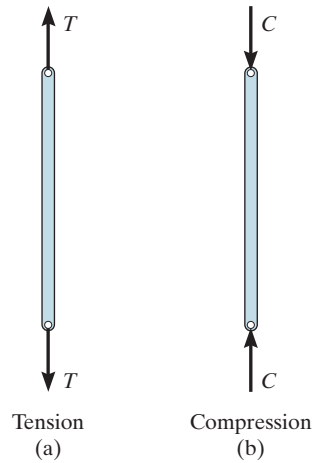


Fig. 6-4

Because of these two assumptions, *each truss member will act as a two-force member*, and therefore the force acting at each end of the member will be directed along the axis of the member. If the force tends to *elongate* the member, it is a **tensile force** ( $T$ ), Fig. 6-4a; whereas if it tends to *shorten* the member, it is a **compressive force** ( $C$ ), Fig. 6-4b. In the actual design of a truss it is important to state whether the nature of the force is tensile or compressive. Often, compression members must be made *thicker* than tension members because of the buckling or column effect that occurs when a member is in compression.

**Simple Truss.** If three members are pin connected at their ends, they form a *triangular truss* that will be *rigid*, Fig. 6-5. Attaching two more members and connecting these members to a new joint  $D$  forms a larger truss, Fig. 6-6. This procedure can be repeated as many times as desired to form an even larger truss. If a truss can be constructed by expanding the basic triangular truss in this way, it is called a **simple truss**.



The use of metal gusset plates in the construction of these Warren trusses is clearly evident. (© Russell C. Hibbeler)

6

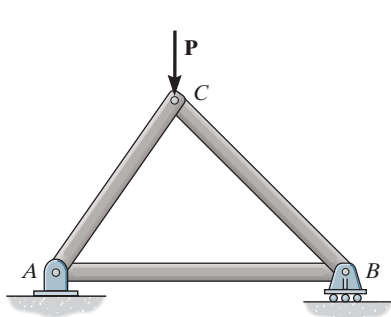


Fig. 6-5

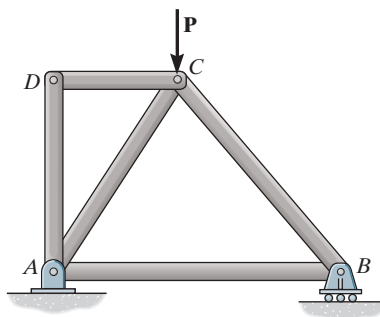


Fig. 6-6



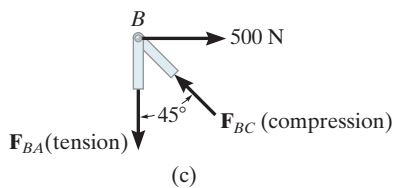
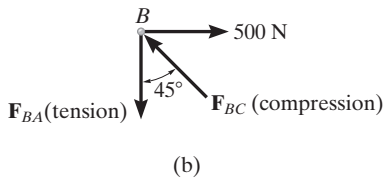
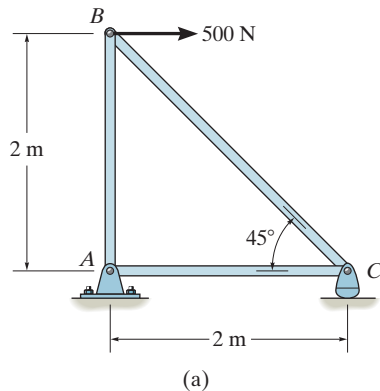


Fig. 6-7



The forces in the members of this simple roof truss can be determined using the method of joints. (© Russell C. Hibbeler)

## 6.2 The Method of Joints

In order to analyze or design a truss, it is necessary to determine the force in each of its members. One way to do this is to use the **method of joints**. This method is based on the fact that if the entire truss is in equilibrium, then each of its joints is also in equilibrium. Therefore, if the free-body diagram of each joint is drawn, the force equilibrium equations can then be used to obtain the member forces acting on each joint. Since the members of a *plane truss* are straight two-force members lying in a single plane, each joint is subjected to a force system that is *coplanar and concurrent*. As a result, only  $\Sigma F_x = 0$  and  $\Sigma F_y = 0$  need to be satisfied for equilibrium.

For example, consider the pin at joint *B* of the truss in Fig. 6-7a. Three forces act on the pin, namely, the 500-N force and the forces exerted by members *BA* and *BC*. The free-body diagram of the pin is shown in Fig. 6-7b. Here,  $\mathbf{F}_{BA}$  is “pulling” on the pin, which means that member *BA* is in *tension*; whereas  $\mathbf{F}_{BC}$  is “pushing” on the pin, and consequently member *BC* is in *compression*. These effects are clearly demonstrated by isolating the joint with small segments of the member connected to the pin, Fig. 6-7c. The pushing or pulling on these small segments indicates the effect of the member being either in compression or tension.

When using the method of joints, always start at a joint having at least one known force and at most two unknown forces, as in Fig. 6-7b. In this way, application of  $\Sigma F_x = 0$  and  $\Sigma F_y = 0$  yields two algebraic equations which can be solved for the two unknowns. When applying these equations, the correct sense of an unknown member force can be determined using one of two possible methods.

- The *correct* sense of direction of an unknown member force can, in many cases, be determined “by inspection.” For example,  $\mathbf{F}_{BC}$  in Fig. 6-7b must push on the pin (compression) since its horizontal component,  $F_{BC} \sin 45^\circ$ , must balance the 500-N force ( $\Sigma F_x = 0$ ). Likewise,  $\mathbf{F}_{BA}$  is a tensile force since it balances the vertical component,  $F_{BC} \cos 45^\circ$  ( $\Sigma F_y = 0$ ). In more complicated cases, the sense of an unknown member force can be *assumed*; then, after applying the equilibrium equations, the assumed sense can be verified from the numerical results. A *positive* answer indicates that the sense is *correct*, whereas a *negative* answer indicates that the sense shown on the free-body diagram must be *reversed*.
- *Always assume the unknown member forces acting on the joint’s free-body diagram to be in tension*; i.e., the forces “pull” on the pin. If this is done, then numerical solution of the equilibrium equations will yield *positive scalars for members in tension and negative scalars for members in compression*. Once an unknown member force is found, use its *correct* magnitude and sense (T or C) on subsequent joint free-body diagrams.

### Important Points

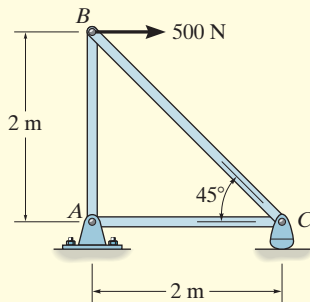
- Simple trusses are composed of triangular elements. The members are assumed to be pin connected at their ends and loads applied at the joints.
- If a truss is in equilibrium, then each of its joints is in equilibrium. The internal forces in the members become external forces when the free-body diagram of each joint of the truss is drawn. A force pulling on a joint is caused by tension in a member, and a force pushing on a joint is caused by compression.

### Procedure for Analysis

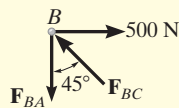
The following procedure provides a means for analyzing a truss using the method of joints.

- Draw the free-body diagram of a joint having at least one known force and at most two unknown forces. (If this joint is at one of the supports, then it may be necessary first to calculate the external reactions at the support.)
- Use one of the two methods described above for establishing the sense of an unknown force.
- Orient the  $x$  and  $y$  axes such that the forces on the free-body diagram can be easily resolved into their  $x$  and  $y$  components and then apply the two force equilibrium equations  $\Sigma F_x = 0$  and  $\Sigma F_y = 0$ . Solve for the two unknown member forces and verify their correct sense.
- Using the calculated results, continue to analyze each of the other joints. Remember that a member in *compression* “pushes” on the joint and a member in *tension* “pulls” on the joint. Also, be sure to choose a joint having at most two unknowns and at least one known force.

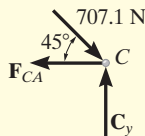


**EXAMPLE 6.1**

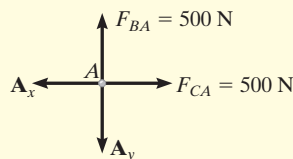
(a)



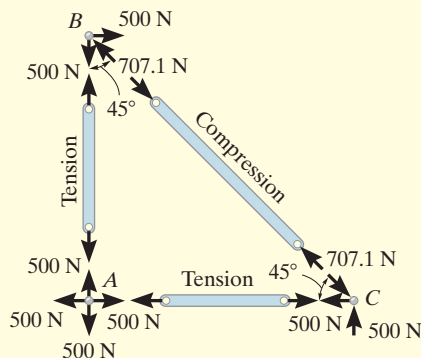
(b)



(c)



(d)



(e)

**Fig. 6-8**

Determine the force in each member of the truss shown in Fig. 6-8a and indicate whether the members are in tension or compression.

**SOLUTION**

Since we should have no more than two unknown forces at the joint and at least one known force acting there, we will begin our analysis at joint *B*.

**Joint B.** The free-body diagram of the joint at *B* is shown in Fig. 6-8b. Applying the equations of equilibrium, we have

$$\pm \Sigma F_x = 0; \quad 500 \text{ N} - F_{BC} \sin 45^\circ = 0 \quad F_{BC} = 707.1 \text{ N (C)} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad F_{BC} \cos 45^\circ - F_{BA} = 0 \quad F_{BA} = 500 \text{ N (T)} \quad \text{Ans.}$$

Since the force in member *BC* has been calculated, we can proceed to analyze joint *C* to determine the force in member *CA* and the support reaction at the rocker.

**Joint C.** From the free-body diagram of joint *C*, Fig. 6-8c, we have

$$\pm \Sigma F_x = 0; \quad -F_{CA} + 707.1 \cos 45^\circ \text{ N} = 0 \quad F_{CA} = 500 \text{ N (T)} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad C_y - 707.1 \sin 45^\circ \text{ N} = 0 \quad C_y = 500 \text{ N} \quad \text{Ans.}$$

**Joint A.** Although it is not necessary, we can determine the components of the support reactions at joint *A* using the results of  $F_{CA}$  and  $F_{BA}$ . From the free-body diagram, Fig. 6-8d, we have

$$\pm \Sigma F_x = 0; \quad 500 \text{ N} - A_x = 0 \quad A_x = 500 \text{ N}$$

$$+\uparrow \Sigma F_y = 0; \quad 500 \text{ N} - A_y = 0 \quad A_y = 500 \text{ N}$$

**NOTE:** The results of the analysis are summarized in Fig. 6-8e. Note that the free-body diagram of each joint (or pin) shows the effects of all the connected members and external forces applied to the joint, whereas the free-body diagram of each member shows only the effects of the end joints on the member.

**EXAMPLE 6.2**

Determine the forces acting in all the members of the truss shown in Fig. 6–9a.

**SOLUTION**

By inspection, there are more than two unknowns at each joint. Consequently, the support reactions on the truss must first be determined. Show that they have been correctly calculated on the free-body diagram in Fig. 6–9b. We can now begin the analysis at joint C. Why?

**Joint C.** From the free-body diagram, Fig. 6–9c,

$$\begin{aligned} \rightarrow \Sigma F_x &= 0; & -F_{CD} \cos 30^\circ + F_{CB} \sin 45^\circ &= 0 \\ + \uparrow \Sigma F_y &= 0; & 1.5 \text{ kN} + F_{CD} \sin 30^\circ - F_{CB} \cos 45^\circ &= 0 \end{aligned}$$

These two equations must be solved *simultaneously* for each of the two unknowns. Note, however, that a *direct solution* for one of the unknown forces may be obtained by applying a force summation along an axis that is *perpendicular* to the direction of the other unknown force. For example, summing forces along the  $y'$  axis, which is perpendicular to the direction of  $F_{CD}$ , Fig. 6–9d, yields a *direct solution* for  $F_{CB}$ .

$$\begin{aligned} + \nearrow \Sigma F_{y'} &= 0; & 1.5 \cos 30^\circ \text{ kN} - F_{CB} \sin 15^\circ &= 0 \\ & & F_{CB} &= 5.019 \text{ kN} = 5.02 \text{ kN (C)} \end{aligned}$$

*Ans.*

Then,

$$\begin{aligned} + \searrow \Sigma F_{x'} &= 0; \\ -F_{CD} + 5.019 \cos 15^\circ - 1.5 \sin 30^\circ &= 0; & F_{CD} &= 4.10 \text{ kN (T)} \end{aligned}$$

*Ans.*

**Joint D.** We can now proceed to analyze joint D. The free-body diagram is shown in Fig. 6–9e.

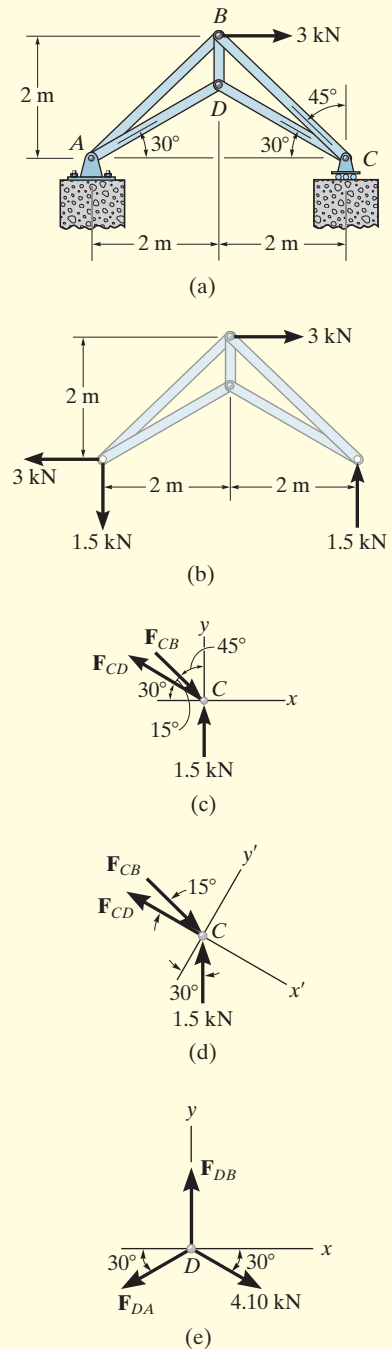
$$\begin{aligned} \rightarrow \Sigma F_x &= 0; & -F_{DA} \cos 30^\circ + 4.10 \cos 30^\circ \text{ kN} &= 0 \\ & & F_{DA} &= 4.10 \text{ kN (T)} \end{aligned}$$

*Ans.*

$$\begin{aligned} + \uparrow \Sigma F_y &= 0; & F_{DB} - 2(4.10 \sin 30^\circ \text{ kN}) &= 0 \\ & & F_{DB} &= 4.10 \text{ kN (T)} \end{aligned}$$

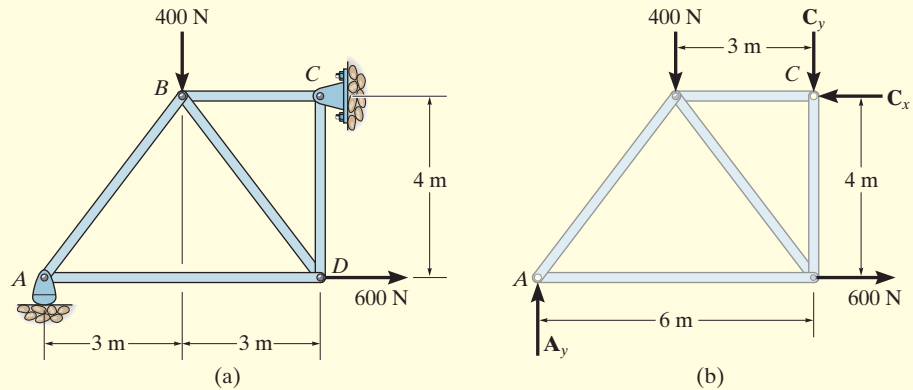
*Ans.*

**NOTE:** The force in the last member, BA, can be obtained from joint B or joint A. As an exercise, draw the free-body diagram of joint B, sum the forces in the horizontal direction, and show that  $F_{BA} = 0.776 \text{ kN (C)}$ .

**Fig. 6–9**

**EXAMPLE 6.3**

Determine the force in each member of the truss shown in Fig. 6–10a. Indicate whether the members are in tension or compression.

**Fig. 6–10****SOLUTION**

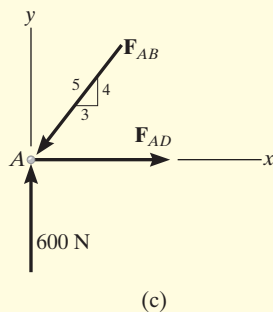
**Support Reactions.** No joint can be analyzed until the support reactions are determined, because each joint has at least three unknown forces acting on it. A free-body diagram of the entire truss is given in Fig. 6–10b. Applying the equations of equilibrium, we have

$$\begin{aligned} \rightarrow \Sigma F_x &= 0; & 600 \text{ N} - C_x &= 0 & C_x &= 600 \text{ N} \\ \zeta + \Sigma M_C &= 0; & -A_y(6 \text{ m}) + 400 \text{ N}(3 \text{ m}) + 600 \text{ N}(4 \text{ m}) &= 0 \\ & & A_y &= 600 \text{ N} \\ + \uparrow \Sigma F_y &= 0; & 600 \text{ N} - 400 \text{ N} - C_y &= 0 & C_y &= 200 \text{ N} \end{aligned}$$

The analysis can now start at either joint *A* or *C*. The choice is arbitrary since there are one known and two unknown member forces acting on the pin at each of these joints.

**Joint A.** (Fig. 6–10c). As shown on the free-body diagram,  $F_{AB}$  is assumed to be compressive and  $F_{AD}$  is tensile. Applying the equations of equilibrium, we have

$$\begin{aligned} + \uparrow \Sigma F_y &= 0; & 600 \text{ N} - \frac{4}{5}F_{AB} &= 0 & F_{AB} &= 750 \text{ N (C)} & \text{Ans.} \\ \rightarrow \Sigma F_x &= 0; & F_{AD} - \frac{3}{5}(750 \text{ N}) &= 0 & F_{AD} &= 450 \text{ N (T)} & \text{Ans.} \end{aligned}$$



**Joint D.** (Fig. 6–10d). Using the result for  $F_{AD}$  and summing forces in the horizontal direction, Fig. 6–10d, we have

$$\rightarrow \Sigma F_x = 0; \quad -450 \text{ N} + \frac{3}{5}F_{DB} + 600 \text{ N} = 0 \quad F_{DB} = -250 \text{ N}$$

The negative sign indicates that  $\mathbf{F}_{DB}$  acts in the *opposite sense* to that shown in Fig. 6–10d.\* Hence,

$$F_{DB} = 250 \text{ N (T)} \quad \text{Ans.}$$

To determine  $\mathbf{F}_{DC}$ , we can either correct the sense of  $\mathbf{F}_{DB}$  on the free-body diagram, and then apply  $\Sigma F_y = 0$ , or apply this equation and retain the negative sign for  $F_{DB}$ , i.e.,

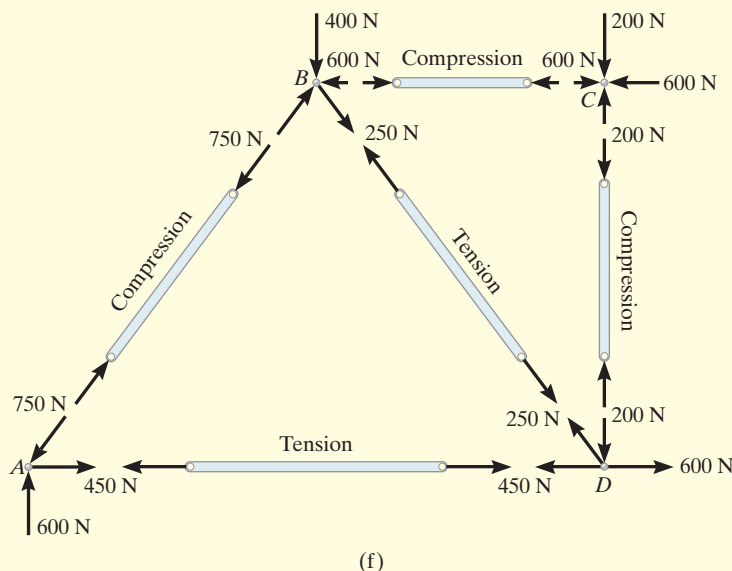
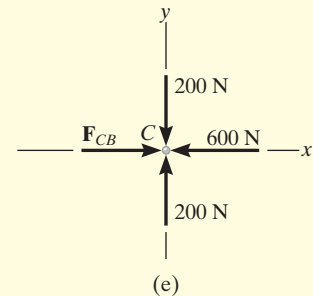
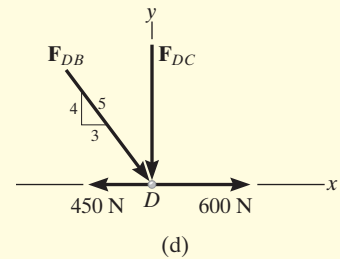
$$+\uparrow \Sigma F_y = 0; \quad -F_{DC} - \frac{4}{5}(-250 \text{ N}) = 0 \quad F_{DC} = 200 \text{ N (C)} \quad \text{Ans.}$$

**Joint C.** (Fig. 6–10e).

$$\rightarrow \Sigma F_x = 0; \quad F_{CB} - 600 \text{ N} = 0 \quad F_{CB} = 600 \text{ N (C)} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad 200 \text{ N} - 200 \text{ N} = 0 \quad (\text{check})$$

**NOTE:** The analysis is summarized in Fig. 6–10f, which shows the free-body diagram for each joint and member.



**Fig. 6–10 (cont.)**

\*The proper sense could have been determined by inspection, prior to applying  $\Sigma F_x = 0$ .

## 6.3 Zero-Force Members

Truss analysis using the method of joints is greatly simplified if we can first identify those members which support *no loading*. These *zero-force members* are used to increase the stability of the truss during construction and to provide added support if the loading is changed.

The zero-force members of a truss can generally be found by *inspection* of each of the joints. For example, consider the truss shown in Fig. 6–11a. If a free-body diagram of the pin at joint *A* is drawn, Fig. 6–11b, it is seen that members *AB* and *AF* are zero-force members. (We could not have come to this conclusion if we had considered the free-body diagrams of joints *F* or *B* simply because there are five unknowns at each of these joints.) In a similar manner, consider the free-body diagram of joint *D*, Fig. 6–11c. Here again it is seen that *DC* and *DE* are zero-force members. From these observations, we can conclude that *if only two non-collinear members form a truss joint and no external load or support reaction is applied to the joint, the two members must be zero-force members*. The load on the truss in Fig. 6–11a is therefore supported by only five members as shown in Fig. 6–11d.

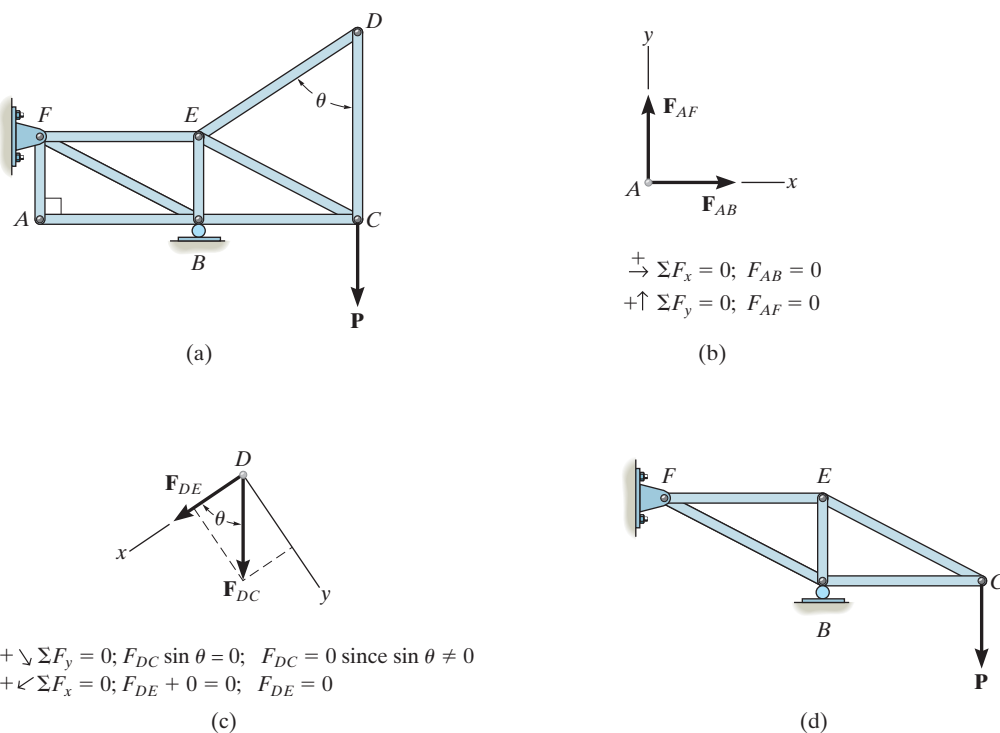
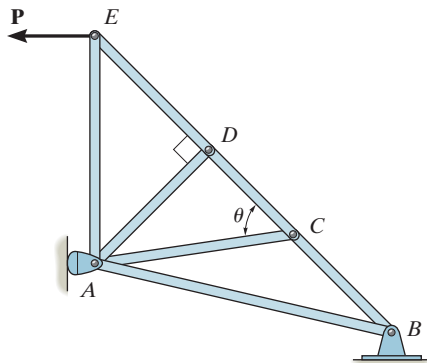
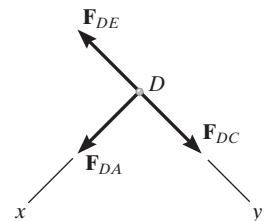


Fig. 6–11

Now consider the truss shown in Fig. 6–12*a*. The free-body diagram of the pin at joint *D* is shown in Fig. 6–12*b*. By orienting the *y* axis along members *DC* and *DE* and the *x* axis along member *DA*, it is seen that *DA* is a zero-force member. Note that this is also the case for member *CA*, Fig. 6–12*c*. In general then, *if three members form a truss joint for which two of the members are collinear, the third member is a zero-force member provided no external force or support reaction has a component that acts along this member*. The truss shown in Fig. 6–12*d* is therefore suitable for supporting the load **P**.

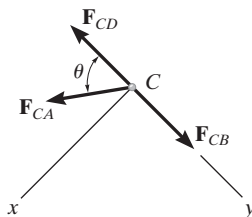


(a)



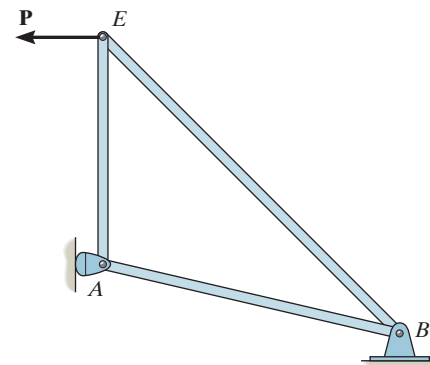
$$\begin{aligned}
 +\curvearrowright \Sigma F_x &= 0; & F_{DA} &= 0 \\
 +\searrow \Sigma F_y &= 0; & F_{DC} &= F_{DE}
 \end{aligned}$$

(b)



$$\begin{aligned}
 +\curvearrowright \Sigma F_x &= 0; & F_{CA} \sin \theta &= 0; & F_{CA} &= 0 \text{ since } \sin \theta \neq 0; \\
 +\searrow \Sigma F_y &= 0; & F_{CB} &= F_{CD}
 \end{aligned}$$

(c)



(d)

Fig. 6–12

### Important Point

- Zero-force members support no load; however, they are necessary for stability, and are available when additional loadings are applied to the joints of the truss. These members can usually be identified by inspection. They occur at joints where only two members are connected and no external load acts along either member. Also, at joints having two collinear members, a third member will be a zero-force member if no external force components act along this member.

## EXAMPLE 6.4

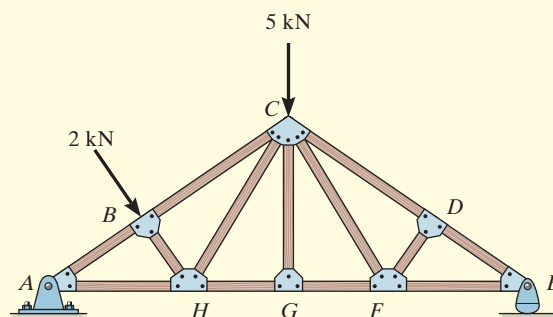
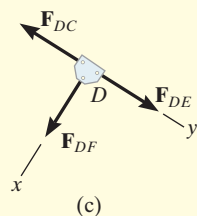
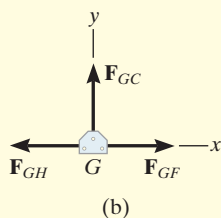


Fig. 6-13

## SOLUTION

Look for joint geometries that have three members for which two are collinear. We have

**Joint G.** (Fig. 6-13b).

$$+\uparrow \Sigma F_y = 0; \quad F_{GC} = 0 \quad \text{Ans.}$$

Realize that we could not conclude that  $GC$  is a zero-force member by considering joint  $C$ , where there are five unknowns. The fact that  $GC$  is a zero-force member means that the 5-kN load at  $C$  must be supported by members  $CB$ ,  $CH$ ,  $CF$ , and  $CD$ .

**Joint D.** (Fig. 6-13c).

$$+\swarrow \Sigma F_x = 0; \quad F_{DF} = 0 \quad \text{Ans.}$$

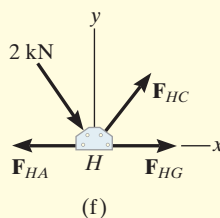
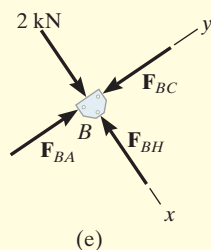
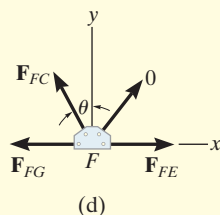
**Joint F.** (Fig. 6-13d).

$$+\uparrow \Sigma F_y = 0; \quad F_{FC} \cos \theta = 0 \quad \text{Since } \theta \neq 90^\circ, \quad F_{FC} = 0 \quad \text{Ans.}$$

**NOTE:** If joint  $B$  is analyzed, Fig. 6-13e,

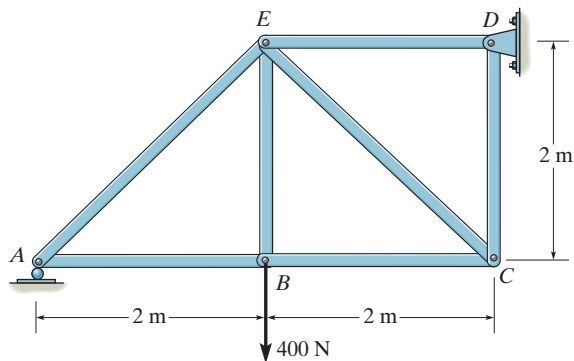
$$+\searrow \Sigma F_x = 0; \quad 2 \text{ kN} - F_{BH} = 0 \quad F_{BH} = 2 \text{ kN} \quad (\text{C})$$

Also,  $F_{HC}$  must satisfy  $\Sigma F_y = 0$ , Fig. 6-13f, and therefore  $HC$  is *not* a zero-force member.

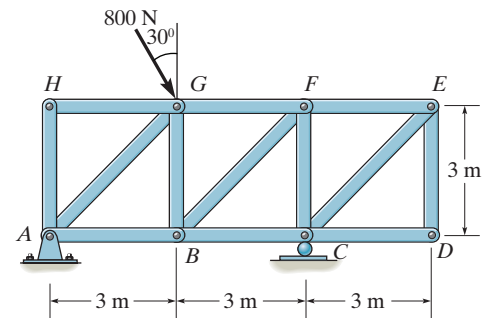


## PRELIMINARY PROBLEMS

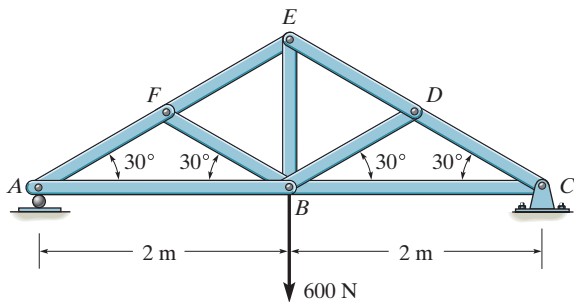
**P6-1.** In each case, calculate the support reactions and then draw the free-body diagrams of joints  $A$ ,  $B$ , and  $C$  of the truss.



(a)

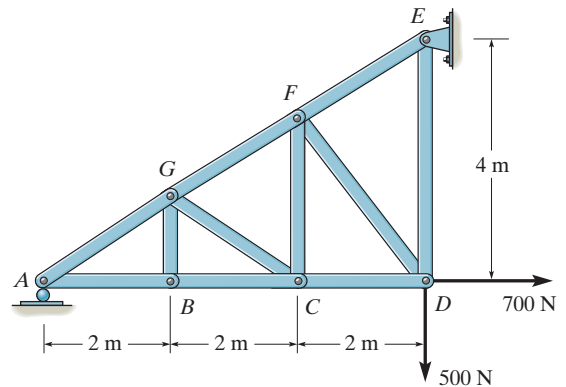


(a)



(b)

**Prob. P6-1**



(b)

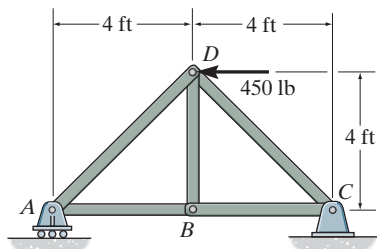
**Prob. P6-2**



## FUNDAMENTAL PROBLEMS

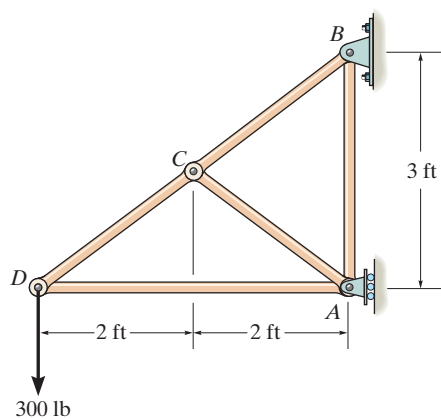
*All problem solutions must include FBDs.*

**F6-1.** Determine the force in each member of the truss. State if the members are in tension or compression.



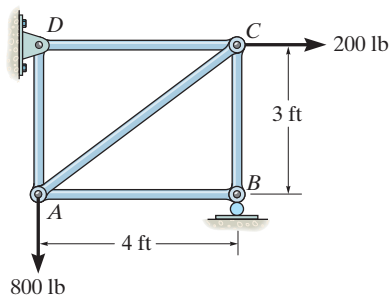
**Prob. F6-1**

**F6-2.** Determine the force in each member of the truss. State if the members are in tension or compression.



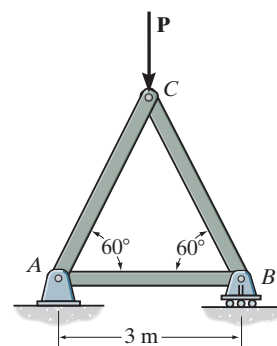
**Prob. F6-2**

**F6-3.** Determine the force in each member of the truss. State if the members are in tension or compression.



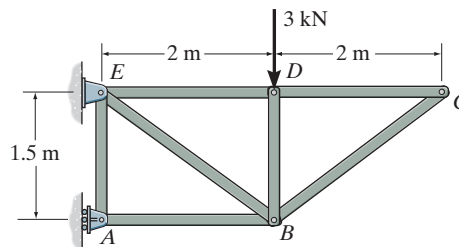
**Prob. F6-3**

**F6-4.** Determine the greatest load  $P$  that can be applied to the truss so that none of the members are subjected to a force exceeding either 2 kN in tension or 1.5 kN in compression.



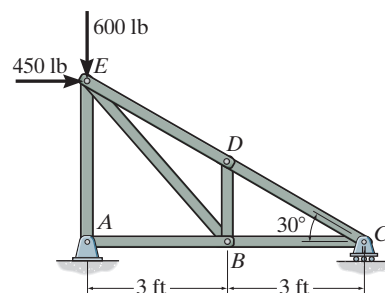
**Prob. F6-4**

**F6-5.** Identify the zero-force members in the truss.



**Prob. F6-5**

**F6-6.** Determine the force in each member of the truss. State if the members are in tension or compression.



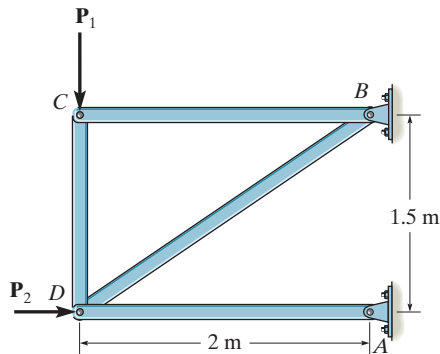
**Prob. F6-6**

## PROBLEMS

*All problem solutions must include FBDs.*

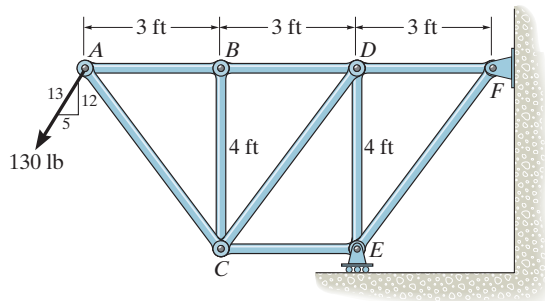
**6-1.** Determine the force in each member of the truss and state if the members are in tension or compression. Set  $P_1 = 20 \text{ kN}$ ,  $P_2 = 10 \text{ kN}$ .

**6-2.** Determine the force in each member of the truss and state if the members are in tension or compression. Set  $P_1 = 45 \text{ kN}$ ,  $P_2 = 30 \text{ kN}$ .



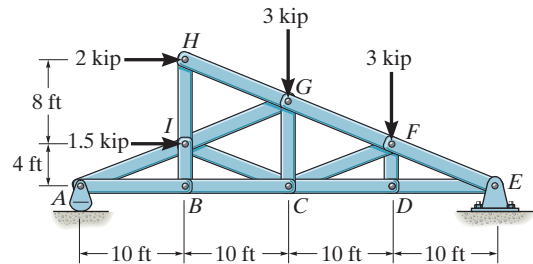
**Probs. 6-1/2**

**6-3.** Determine the force in each member of the truss. State if the members are in tension or compression.



**Prob. 6-3**

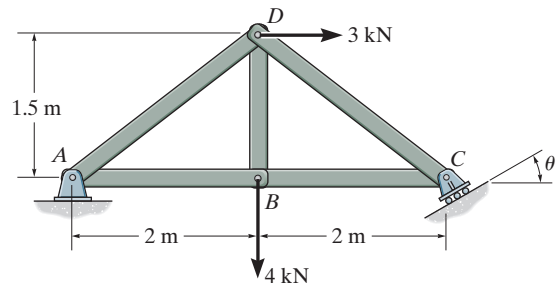
**\*6-4.** Determine the force in each member of the truss and state if the members are in tension or compression.



**Prob. 6-4**

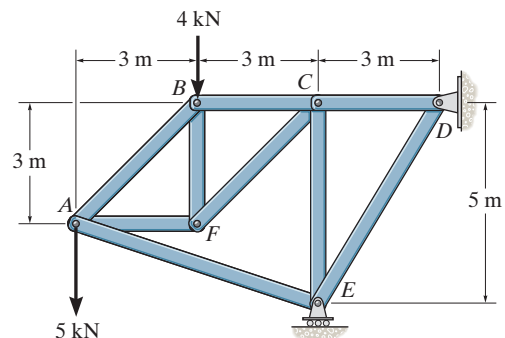
**6-5.** Determine the force in each member of the truss, and state if the members are in tension or compression. Set  $\theta = 0^\circ$ .

**6-6.** Determine the force in each member of the truss, and state if the members are in tension or compression. Set  $\theta = 30^\circ$ .



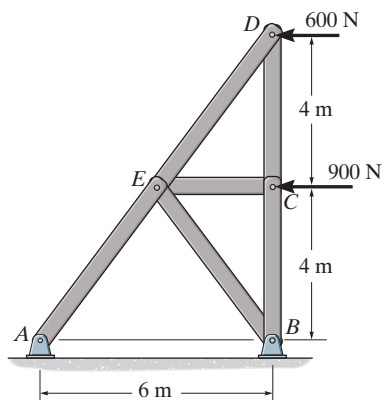
**Probs. 6-5/6**

**6-7.** Determine the force in each member of the truss and state if the members are in tension or compression.



**Prob. 6-7**

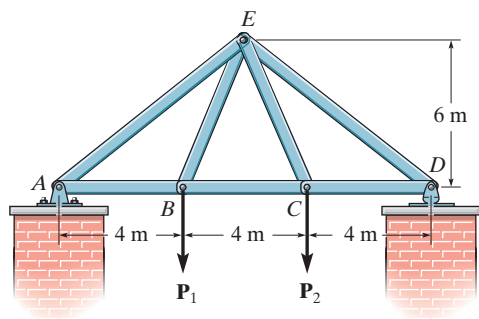
**\*6-8.** Determine the force in each member of the truss and state if the members are in tension or compression.



**Prob. 6-8**

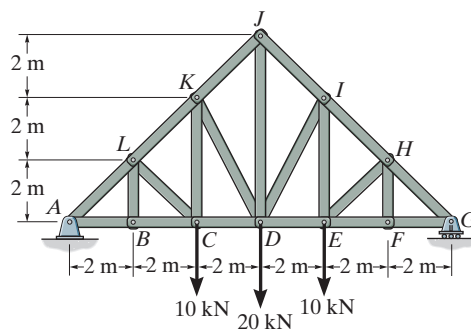
**6-9.** Determine the force in each member of the truss and state if the members are in tension or compression. Set  $P_1 = 3 \text{ kN}$ ,  $P_2 = 6 \text{ kN}$ .

**6-10.** Determine the force in each member of the truss and state if the members are in tension or compression. Set  $P_1 = 6 \text{ kN}$ ,  $P_2 = 9 \text{ kN}$ .



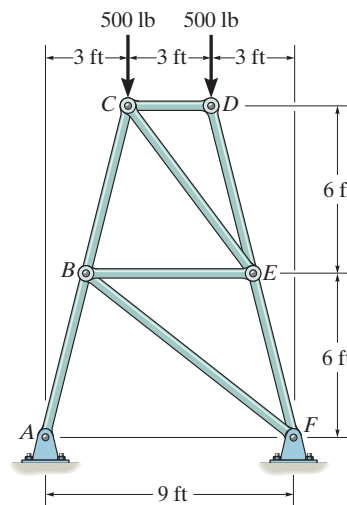
**Probs. 6-9/10**

**6-11.** Determine the force in each member of the *Pratt* truss, and state if the members are in tension or compression.



**Prob. 6-11**

**\*6-12.** Determine the force in each member of the truss and state if the members are in tension or compression.

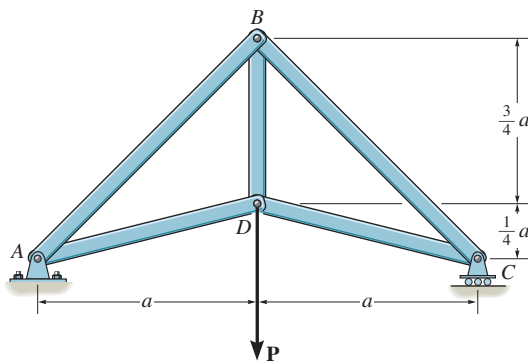


**Prob. 6-12**

**6-13.** Determine the force in each member of the truss in terms of the load  $P$  and state if the members are in tension or compression.

**6-14.** Members  $AB$  and  $BC$  can each support a maximum compressive force of 800 lb, and members  $AD$ ,  $DC$ , and  $BD$  can support a maximum tensile force of 1500 lb. If  $a = 10$  ft, determine the greatest load  $P$  the truss can support.

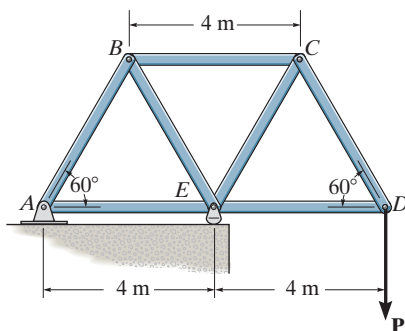
**6-15.** Members  $AB$  and  $BC$  can each support a maximum compressive force of 800 lb, and members  $AD$ ,  $DC$ , and  $BD$  can support a maximum tensile force of 2000 lb. If  $a = 6$  ft, determine the greatest load  $P$  the truss can support.



**Probs. 6-13/14/15**

**\*6-16.** Determine the force in each member of the truss. State whether the members are in tension or compression. Set  $P = 8$  kN.

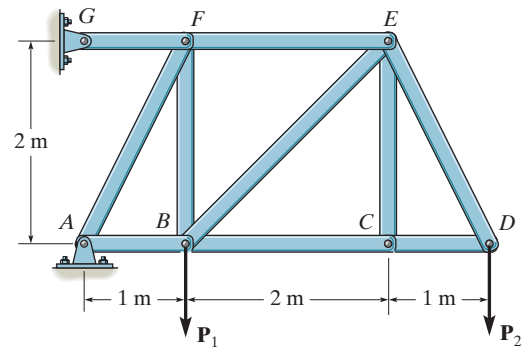
**6-17.** If the maximum force that any member can support is 8 kN in tension and 6 kN in compression, determine the maximum force  $P$  that can be supported at joint  $D$ .



**Probs. 6-16/17**

**6-18.** Determine the force in each member of the truss and state if the members are in tension or compression. Set  $P_1 = 10$  kN,  $P_2 = 8$  kN.

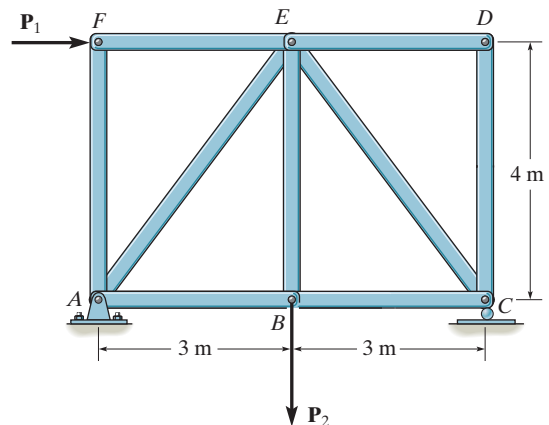
**6-19.** Determine the force in each member of the truss and state if the members are in tension or compression. Set  $P_1 = 8$  kN,  $P_2 = 12$  kN.



**Probs. 6-18/19**

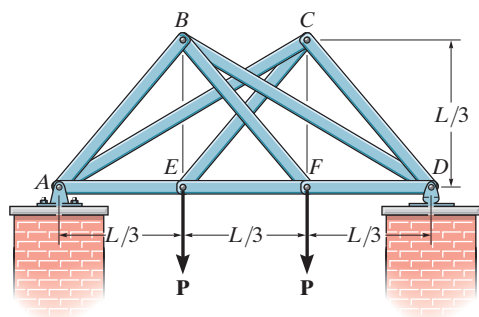
**\*6-20.** Determine the force in each member of the truss and state if the members are in tension or compression. Set  $P_1 = 9$  kN,  $P_2 = 15$  kN.

**6-21.** Determine the force in each member of the truss and state if the members are in tension or compression. Set  $P_1 = 30$  kN,  $P_2 = 15$  kN.



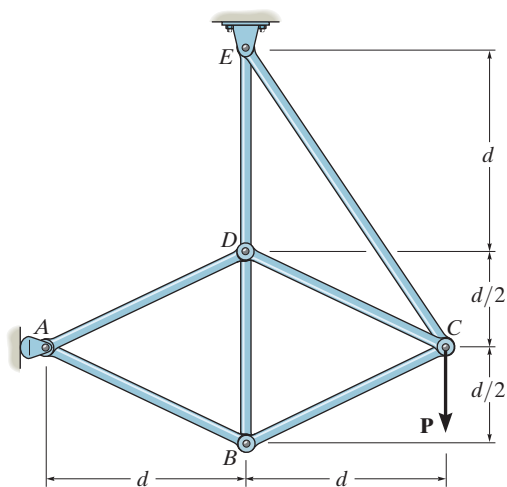
**Probs. 6-20/21**

**6-22.** Determine the force in each member of the double scissors truss in terms of the load  $P$  and state if the members are in tension or compression.



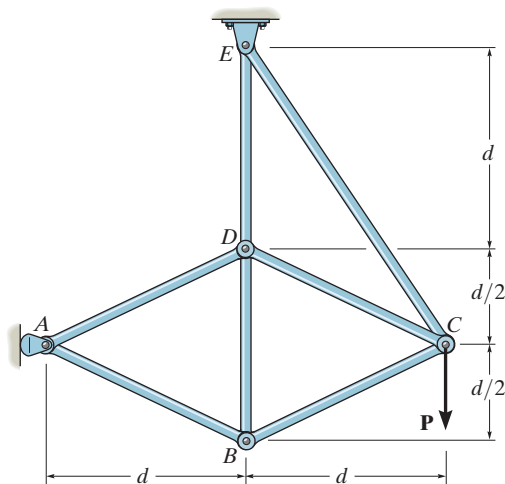
**Prob. 6-22**

**6-23.** Determine the force in each member of the truss in terms of the load  $P$  and state if the members are in tension or compression.



**Prob. 6-23**

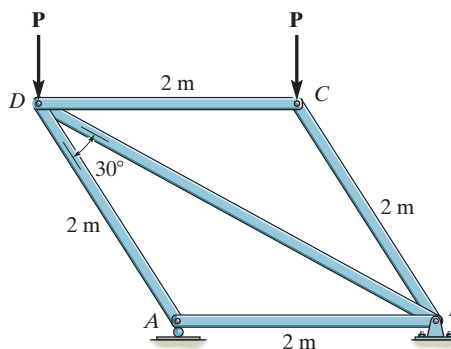
**\*6-24.** The maximum allowable tensile force in the members of the truss is  $(F_t)_{\max} = 5 \text{ kN}$ , and the maximum allowable compressive force is  $(F_c)_{\max} = 3 \text{ kN}$ . Determine the maximum magnitude of load  $P$  that can be applied to the truss. Take  $d = 2 \text{ m}$ .



**Prob. 6-24**

**6-25.** Determine the force in each member of the truss in terms of the external loading and state if the members are in tension or compression. Take  $P = 2 \text{ kN}$ .

**6-26.** The maximum allowable tensile force in the members of the truss is  $(F_t)_{\max} = 5 \text{ kN}$ , and the maximum allowable compressive force is  $(F_c)_{\max} = 3 \text{ kN}$ . Determine the maximum magnitude  $P$  of the two loads that can be applied to the truss.



**Probs. 6-25/26**

## 6.4 The Method of Sections

When we need to find the force in only a few members of a truss, we can analyze the truss using the **method of sections**. It is based on the principle that if the truss is in equilibrium then any segment of the truss is also in equilibrium. For example, consider the two truss members shown on the left in Fig. 6–14. If the forces within the members are to be determined, then an imaginary section, indicated by the blue line, can be used to cut each member into two parts and thereby “expose” each internal force as “external” to the free-body diagrams shown on the right. Clearly, it can be seen that equilibrium requires that the member in tension ( $T$ ) be subjected to a “pull,” whereas the member in compression ( $C$ ) is subjected to a “push.”

The method of sections can also be used to “cut” or section the members of an entire truss. If the section passes through the truss and the free-body diagram of either of its two parts is drawn, we can then apply the equations of equilibrium to that part to determine the member forces at the “cut section.” Since only *three* independent equilibrium equations ( $\Sigma F_x = 0$ ,  $\Sigma F_y = 0$ ,  $\Sigma M_O = 0$ ) can be applied to the free-body diagram of any segment, then we should try to select a section that, in general, passes through not more than *three* members in which the forces are unknown. For example, consider the truss in Fig. 6–15*a*. If the forces in members  $BC$ ,  $GC$ , and  $GF$  are to be determined, then section  $aa$  would be appropriate. The free-body diagrams of the two segments are shown in Figs. 6–15*b* and 6–15*c*. Note that the line of action of each member force is specified from the *geometry* of the truss, since the force in a member is along its axis. Also, the member forces acting on one part of the truss are equal but opposite to those acting on the other part—Newton’s third law. Members  $BC$  and  $GC$  are assumed to be in *tension* since they are subjected to a “pull,” whereas  $GF$  is in *compression* since it is subjected to a “push.”

The three unknown member forces  $F_{BC}$ ,  $F_{GC}$ , and  $F_{GF}$  can be obtained by applying the three equilibrium equations to the free-body diagram in Fig. 6–15*b*. If, however, the free-body diagram in Fig. 6–15*c* is considered, the three support reactions  $D_x$ ,  $D_y$  and  $E_x$  will have to be known, because only three equations of equilibrium are available. (This, of course, is done in the usual manner by considering a free-body diagram of the *entire truss*.)

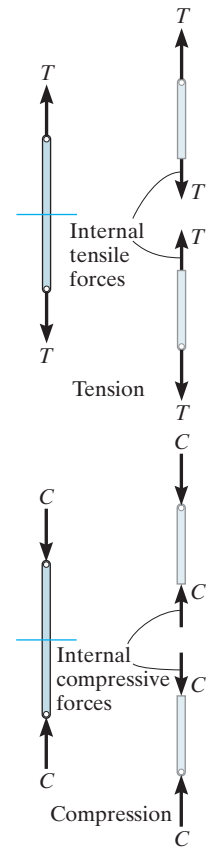


Fig. 6–14

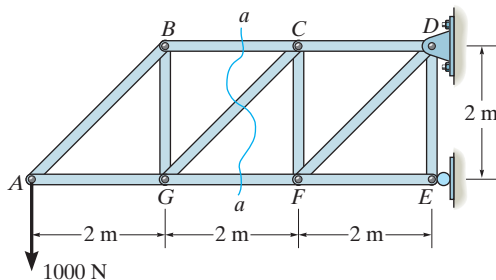


Fig. 6–15



The forces in selected members of this Pratt truss can readily be determined using the method of sections.  
(© Russell C. Hibbeler)

When applying the equilibrium equations, we should carefully consider ways of writing the equations so as to yield a *direct solution* for each of the unknowns, rather than having to solve simultaneous equations. For example, using the truss segment in Fig. 6–15b and summing moments about  $C$  would yield a direct solution for  $F_{GF}$  since  $F_{BC}$  and  $F_{GC}$  create zero moment about  $C$ . Likewise,  $F_{BC}$  can be directly obtained by summing moments about  $G$ . Finally,  $F_{GC}$  can be found directly from a force summation in the vertical direction since  $F_{GF}$  and  $F_{BC}$  have no vertical components. This ability to *determine directly* the force in a particular truss member is one of the main advantages of using the method of sections.\*

As in the method of joints, there are two ways in which we can determine the correct sense of an unknown member force:

- The correct sense of an unknown member force can in many cases be determined “by inspection.” For example,  $F_{BC}$  is a tensile force as represented in Fig. 6–15b since moment equilibrium about  $G$  requires that  $F_{BC}$  create a moment opposite to that of the 1000-N force. Also,  $F_{GC}$  is tensile since its vertical component must balance the 1000-N force which acts downward. In more complicated cases, the sense of an unknown member force may be *assumed*. If the solution yields a *negative* scalar, it indicates that the force’s sense is *opposite* to that shown on the free-body diagram.
- *Always assume* that the unknown member forces at the cut section are *tensile* forces, i.e., “pulling” on the member. By doing this, the numerical solution of the equilibrium equations will yield *positive* scalars for members in tension and *negative* scalars for members in compression.

\*Notice that if the method of joints were used to determine, say, the force in member  $GC$ , it would be necessary to analyze joints  $A$ ,  $B$ , and  $G$  in sequence.

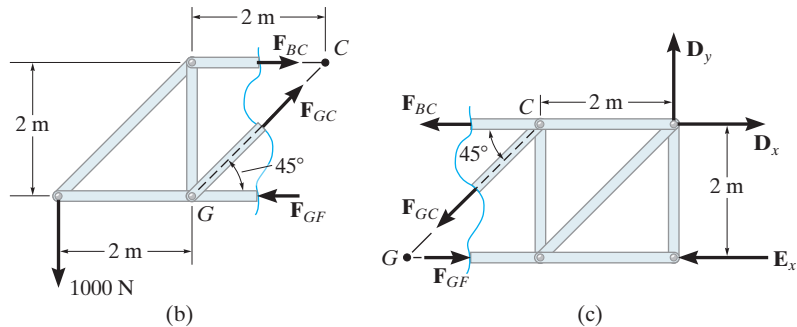


Fig. 6–15 (cont.)

### Important Point

- If a truss is in equilibrium, then each of its segments is in equilibrium. The internal forces in the members become external forces when the free-body diagram of a segment of the truss is drawn. A force pulling on a member causes tension in the member, and a force pushing on a member causes compression.



Simple trusses are often used in the construction of large cranes in order to reduce the weight of the boom and tower. (© Russell C. Hibbeler)

### Procedure for Analysis

The forces in the members of a truss may be determined by the method of sections using the following procedure.

#### Free-Body Diagram.

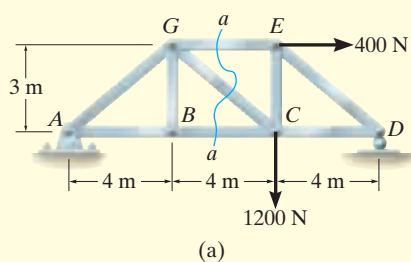
- Make a decision on how to “cut” or section the truss through the members where forces are to be determined.
- Before isolating the appropriate section, it may first be necessary to determine the truss’s support reactions. If this is done then the three equilibrium equations will be available to solve for member forces at the section.
- Draw the free-body diagram of that segment of the sectioned truss which has the least number of forces acting on it.
- Use one of the two methods described above for establishing the sense of the unknown member forces.

#### Equations of Equilibrium.

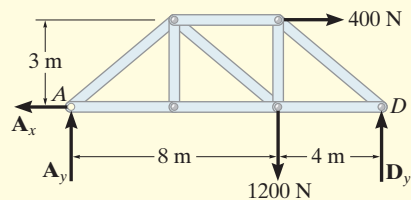
- Moments should be summed about a point that lies at the intersection of the lines of action of two unknown forces, so that the third unknown force can be determined directly from the moment equation.
- If two of the unknown forces are *parallel*, forces may be summed *perpendicular* to the direction of these unknowns to determine *directly* the third unknown force.



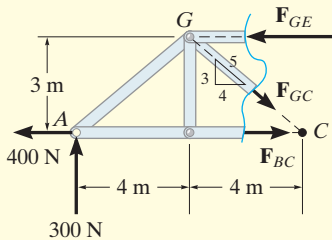
## EXAMPLE 6.5



(a)



(b)



(c)

Fig. 6-16

Determine the force in members  $GE$ ,  $GC$ , and  $BC$  of the truss shown in Fig. 6-16a. Indicate whether the members are in tension or compression.

## SOLUTION

Section  $aa$  in Fig. 6-16a has been chosen since it cuts through the *three* members whose forces are to be determined. In order to use the method of sections, however, it is *first* necessary to determine the external reactions at  $A$  or  $D$ . Why? A free-body diagram of the entire truss is shown in Fig. 6-16b. Applying the equations of equilibrium, we have

$$\pm \Sigma F_x = 0; \quad 400 \text{ N} - A_x = 0 \quad A_x = 400 \text{ N}$$

$$\zeta + \Sigma M_A = 0; \quad -1200 \text{ N}(8 \text{ m}) - 400 \text{ N}(3 \text{ m}) + D_y(12 \text{ m}) = 0$$

$$D_y = 900 \text{ N}$$

$$+\uparrow \Sigma F_y = 0; \quad A_y - 1200 \text{ N} + 900 \text{ N} = 0 \quad A_y = 300 \text{ N}$$

**Free-Body Diagram.** For the analysis the free-body diagram of the left portion of the sectioned truss will be used, since it involves the least number of forces, Fig. 6-16c.

**Equations of Equilibrium.** Summing moments about point  $G$  eliminates  $F_{GE}$  and  $F_{GC}$  and yields a direct solution for  $F_{BC}$ .

$$\zeta + \Sigma M_G = 0; \quad -300 \text{ N}(4 \text{ m}) - 400 \text{ N}(3 \text{ m}) + F_{BC}(3 \text{ m}) = 0$$

$$F_{BC} = 800 \text{ N} \quad (\text{T}) \quad \text{Ans.}$$

In the same manner, by summing moments about point  $C$  we obtain a direct solution for  $F_{GE}$ .

$$\zeta + \Sigma M_C = 0; \quad -300 \text{ N}(8 \text{ m}) + F_{GE}(3 \text{ m}) = 0$$

$$F_{GE} = 800 \text{ N} \quad (\text{C}) \quad \text{Ans.}$$

Since  $F_{BC}$  and  $F_{GE}$  have no vertical components, summing forces in the  $y$  direction directly yields  $F_{GC}$ , i.e.,

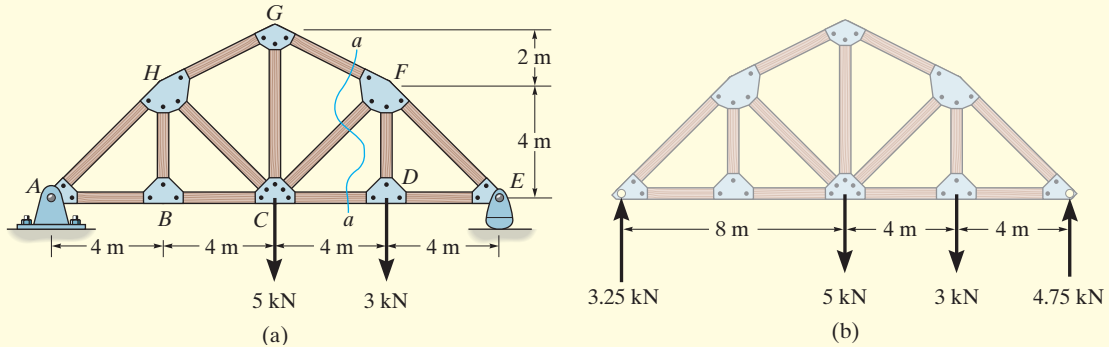
$$+\uparrow \Sigma F_y = 0; \quad 300 \text{ N} - \frac{3}{5} F_{GC} = 0$$

$$F_{GC} = 500 \text{ N} \quad (\text{T}) \quad \text{Ans.}$$

**NOTE:** Here it is possible to tell, by inspection, the proper direction for each unknown member force. For example,  $\Sigma M_C = 0$  requires  $F_{GE}$  to be *compressive* because it must balance the moment of the 300-N force about  $C$ .

**EXAMPLE 6.6**

Determine the force in member  $CF$  of the truss shown in Fig. 6-17a. Indicate whether the member is in tension or compression. Assume each member is pin connected.

**Fig. 6-17****SOLUTION**

**Free-Body Diagram.** Section  $aa$  in Fig. 6-17a will be used since this section will “expose” the internal force in member  $CF$  as “external” on the free-body diagram of either the right or left portion of the truss. It is first necessary, however, to determine the support reactions on either the left or right side. Verify the results shown on the free-body diagram in Fig. 6-17b.

The free-body diagram of the right portion of the truss, which is the easiest to analyze, is shown in Fig. 6-17c. There are three unknowns,  $F_{FG}$ ,  $F_{CF}$ , and  $F_{CD}$ .

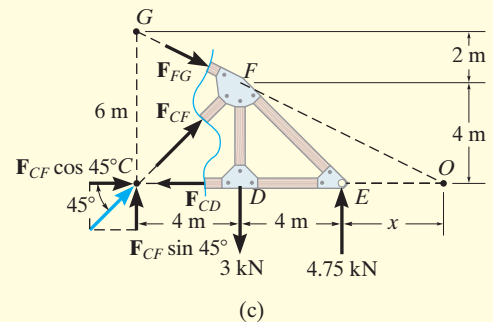
**Equations of Equilibrium.** We will apply the moment equation about point  $O$  in order to eliminate the two unknowns  $F_{FG}$  and  $F_{CD}$ . The location of point  $O$  measured from  $E$  can be determined from proportional triangles, i.e.,  $4/(4 + x) = 6/(8 + x)$ ,  $x = 4$  m. Or, stated in another manner, the slope of member  $GF$  has a drop of 2 m to a horizontal distance of 4 m. Since  $FD$  is 4 m, Fig. 6-17c, then from  $D$  to  $O$  the distance must be 8 m.

An easy way to determine the moment of  $\mathbf{F}_{CF}$  about point  $O$  is to use the principle of transmissibility and slide  $\mathbf{F}_{CF}$  to point  $C$ , and then resolve  $\mathbf{F}_{CF}$  into its two rectangular components. We have

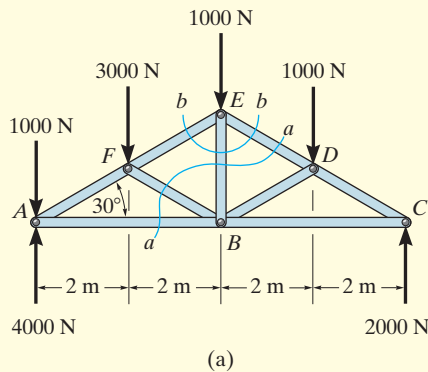
$$\zeta + \sum M_O = 0;$$

$$-F_{CF} \sin 45^\circ (12 \text{ m}) + (3 \text{ kN})(8 \text{ m}) - (4.75 \text{ kN})(4 \text{ m}) = 0$$

$$F_{CF} = 0.589 \text{ kN} \quad (\text{C})$$

*Ans.*

## EXAMPLE 6.7



Determine the force in member  $EB$  of the roof truss shown in Fig. 6-18a. Indicate whether the member is in tension or compression.

## SOLUTION

**Free-Body Diagrams.** By the method of sections, any imaginary section that cuts through  $EB$ , Fig. 6-18a, will also have to cut through three other members for which the forces are unknown. For example, section  $aa$  cuts through  $ED$ ,  $EB$ ,  $FB$ , and  $AB$ . If a free-body diagram of the left side of this section is considered, Fig. 6-18b, it is possible to obtain  $F_{ED}$  by summing moments about  $B$  to eliminate the other three unknowns; however,  $F_{EB}$  cannot be determined from the remaining two equilibrium equations. One possible way of obtaining  $F_{EB}$  is first to determine  $F_{ED}$  from section  $aa$ , then use this result on section  $bb$ , Fig. 6-18c, which is shown in Fig. 6-18c. Here the force system is concurrent and our sectioned free-body diagram is the same as the free-body diagram for the joint at  $E$ .

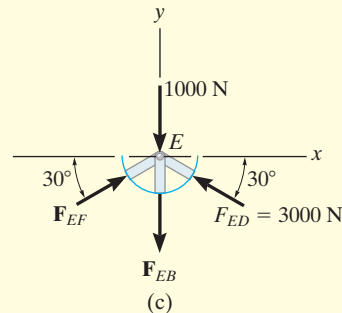
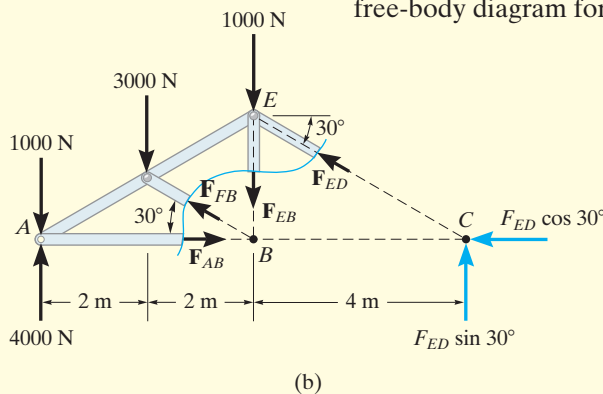


Fig. 6-18

**Equations of Equilibrium.** In order to determine the moment of  $F_{ED}$  about point  $B$ , Fig. 6-18b, we will use the principle of transmissibility and slide the force to point  $C$  and then resolve it into its rectangular components as shown. Therefore,

$$\begin{aligned} \zeta + \sum M_B = 0; \quad & 1000 \text{ N}(4 \text{ m}) + 3000 \text{ N}(2 \text{ m}) - 4000 \text{ N}(4 \text{ m}) \\ & + F_{ED} \sin 30^\circ(4 \text{ m}) = 0 \\ & F_{ED} = 3000 \text{ N} \quad (C) \end{aligned}$$

Considering now the free-body diagram of section  $bb$ , Fig. 6-18c, we have

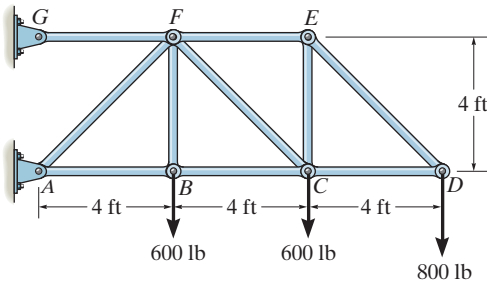
$$\begin{aligned} \rightarrow \sum F_x = 0; \quad & F_{EF} \cos 30^\circ - 3000 \cos 30^\circ \text{ N} = 0 \\ & F_{EF} = 3000 \text{ N} \quad (C) \\ + \uparrow \sum F_y = 0; \quad & 2(3000 \sin 30^\circ \text{ N}) - 1000 \text{ N} - F_{EB} = 0 \\ & F_{EB} = 2000 \text{ N} \quad (T) \end{aligned}$$

*Ans.*

## FUNDAMENTAL PROBLEMS

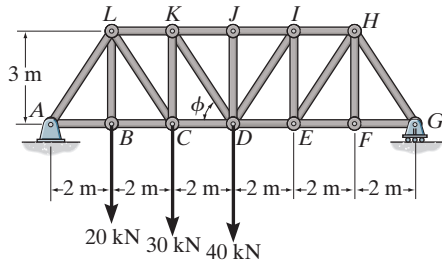
*All problem solutions must include FBDs.*

**F6-7.** Determine the force in members  $BC$ ,  $CF$ , and  $FE$ . State if the members are in tension or compression.



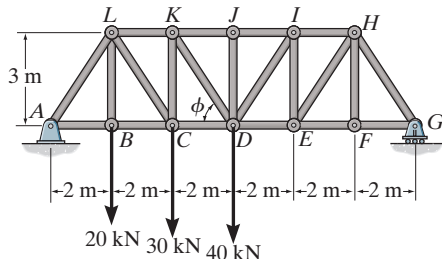
**Prob. F6-7**

**F6-8.** Determine the force in members  $LK$ ,  $KC$ , and  $CD$  of the Pratt truss. State if the members are in tension or compression.



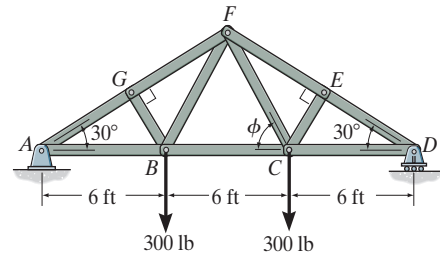
**Prob. F6-8**

**F6-9.** Determine the force in members  $KJ$ ,  $KD$ , and  $CD$  of the Pratt truss. State if the members are in tension or compression.



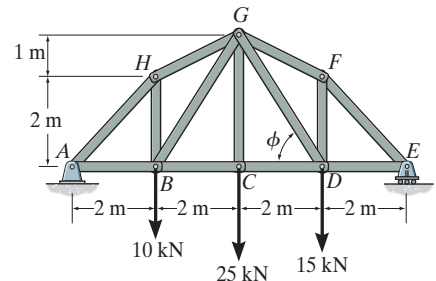
**Prob. F6-9**

**F6-10.** Determine the force in members  $EF$ ,  $CF$ , and  $BC$  of the truss. State if the members are in tension or compression.



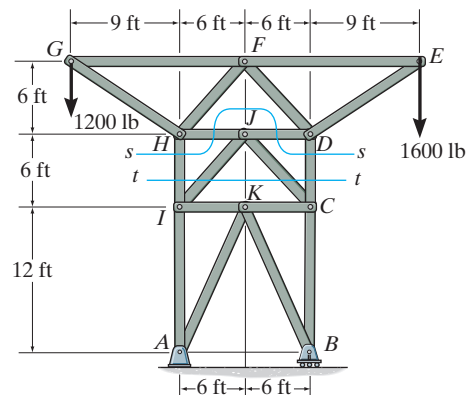
**Prob. F6-10**

**F6-11.** Determine the force in members  $GF$ ,  $GD$ , and  $CD$  of the truss. State if the members are in tension or compression.



**Prob. F6-11**

**F6-12.** Determine the force in members  $DC$ ,  $HI$ , and  $JI$  of the truss. State if the members are in tension or compression. *Suggestion:* Use the sections shown.



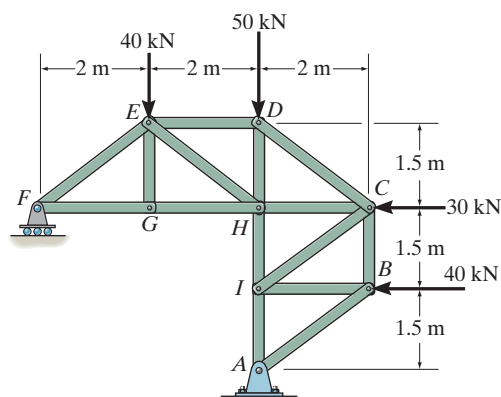
**Prob. F6-12**

# PROBLEMS

*All problem solutions must include FBDs.*

**6-27.** Determine the force in members  $DC$ ,  $HC$ , and  $HI$  of the truss, and state if the members are in tension or compression.

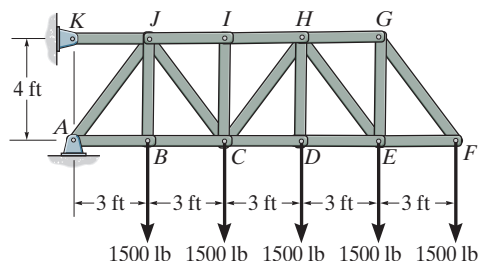
**\*6-28.** Determine the force in members  $ED$ ,  $EH$ , and  $GH$  of the truss, and state if the members are in tension or compression.



**Probs. 6-27/28**

**6-29.** Determine the force in members  $HG$ ,  $HE$  and  $DE$  of the truss, and state if the members are in tension or compression.

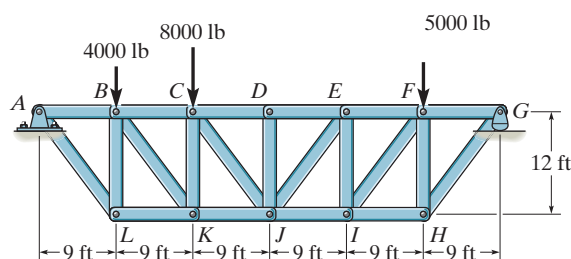
**6-30.** Determine the force in members  $CD$ ,  $HI$ , and  $CH$  of the truss, and state if the members are in tension or compression.



**Probs. 6-29/30**

**6-31.** Determine the force in members  $CD$ ,  $CJ$ ,  $KJ$ , and  $DJ$  of the truss which serves to support the deck of a bridge. State if these members are in tension or compression.

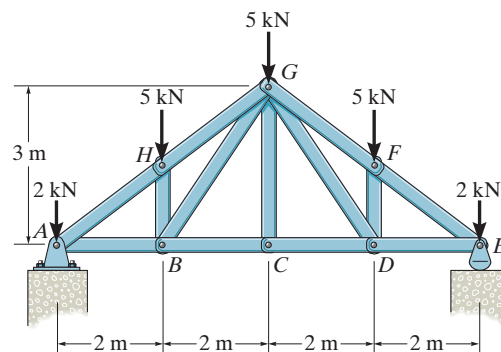
**\*6-32.** Determine the force in members  $EI$  and  $JI$  of the truss which serves to support the deck of a bridge. State if these members are in tension or compression.



**Probs. 6-31/32**

**6-33.** The *Howe truss* is subjected to the loading shown. Determine the force in members  $GF$ ,  $CD$ , and  $GC$ , and state if the members are in tension or compression.

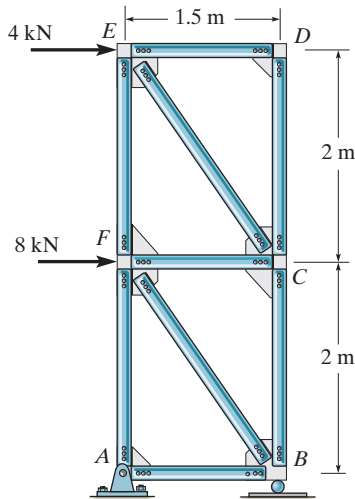
**6-34.** The *Howe truss* is subjected to the loading shown. Determine the force in members  $GH$ ,  $BC$ , and  $BG$  of the truss and state if the members are in tension or compression.



**Probs. 6-33/34**

**6-35.** Determine the force in members  $EF$ ,  $CF$ , and  $BC$ , and state if the members are in tension or compression.

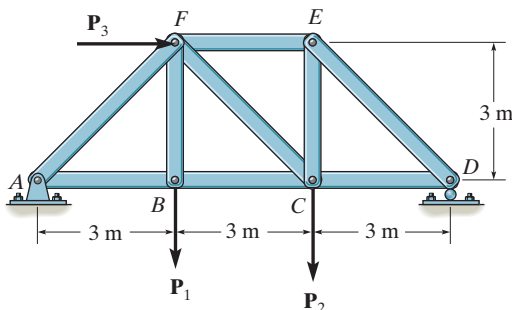
**\*6-36.** Determine the force in members  $AF$ ,  $BF$ , and  $BC$ , and state if the members are in tension or compression.



**Probs. 6-35/36**

**6-37.** Determine the force in members  $EF$ ,  $BE$ ,  $BC$  and  $BF$  of the truss and state if these members are in tension or compression. Set  $P_1 = 9$  kN,  $P_2 = 12$  kN, and  $P_3 = 6$  kN.

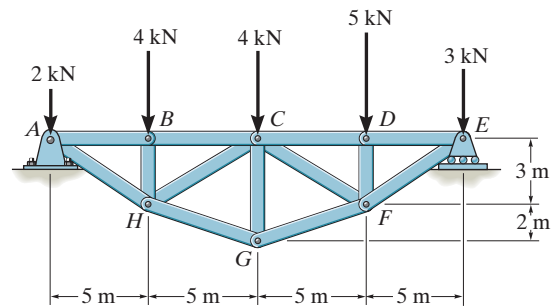
**6-38.** Determine the force in members  $BC$ ,  $BE$ , and  $EF$  of the truss and state if these members are in tension or compression. Set  $P_1 = 6$  kN,  $P_2 = 9$  kN, and  $P_3 = 12$  kN.



**Probs. 6-37/38**

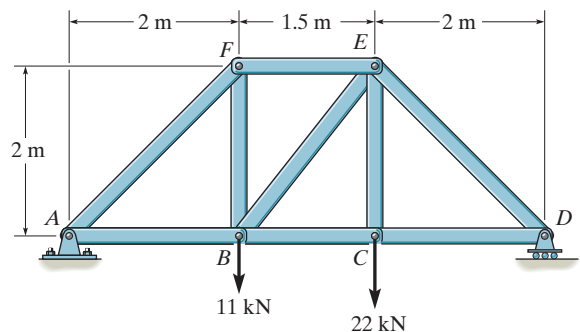
**6-39.** Determine the force in members  $BC$ ,  $HC$ , and  $HG$ . After the truss is sectioned use a single equation of equilibrium for the calculation of each force. State if these members are in tension or compression.

**\*6-40.** Determine the force in members  $CD$ ,  $CF$ , and  $CG$  and state if these members are in tension or compression.



**Probs. 6-39/40**

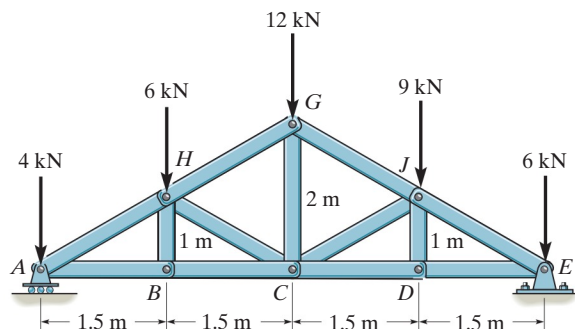
**6-41.** Determine the force developed in members  $FE$ ,  $EB$ , and  $BC$  of the truss and state if these members are in tension or compression.



**Prob. 6-41**

**6-42.** Determine the force in members  $BC$ ,  $HC$ , and  $HG$ . State if these members are in tension or compression.

**6-43.** Determine the force in members  $CD$ ,  $CJ$ ,  $GJ$ , and  $CG$  and state if these members are in tension or compression.

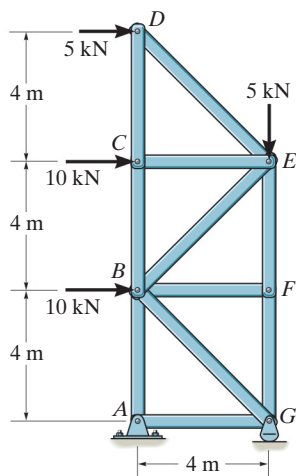


**Probs. 6-42/43**

6

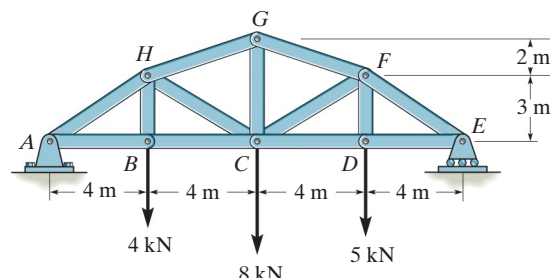
**\*6-44.** Determine the force in members  $BE$ ,  $EF$ , and  $CB$ , and state if the members are in tension or compression.

**6-45.** Determine the force in members  $BF$ ,  $BG$ , and  $AB$ , and state if the members are in tension or compression.



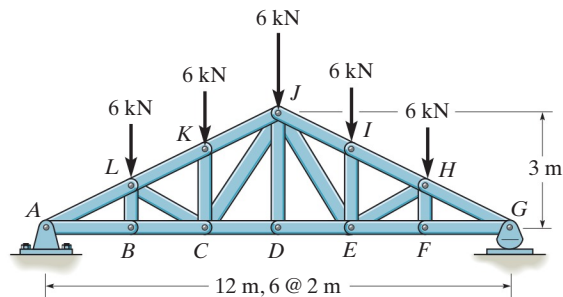
**Probs. 6-44/45**

**6-46.** Determine the force in members  $BC$ ,  $CH$ ,  $GH$ , and  $CG$  of the truss and state if the members are in tension or compression.



**Prob. 6-46**

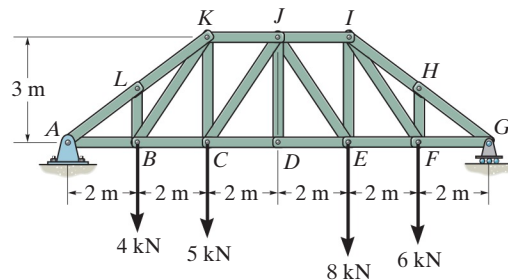
**6-47.** Determine the force in members  $CD$ ,  $CJ$ , and  $KJ$  and state if these members are in tension or compression.



**Prob. 6-47**

**\*6-48.** Determine the force in members  $JK$ ,  $CJ$ , and  $CD$  of the truss, and state if the members are in tension or compression.

**6-49.** Determine the force in members  $HI$ ,  $FI$ , and  $EF$  of the truss, and state if the members are in tension or compression.



**Probs. 6-48/49**

## \*6.5 Space Trusses

A **space truss** consists of members joined together at their ends to form a stable three-dimensional structure. The simplest form of a space truss is a **tetrahedron**, formed by connecting six members together, as shown in Fig. 6–19. Any additional members added to this basic element would be redundant in supporting the force **P**. A **simple space truss** can be built from this basic tetrahedral element by adding three additional members and a joint, and continuing in this manner to form a system of multiconnected tetrahedrons.

**Assumptions for Design.** The members of a space truss may be treated as two-force members provided the external loading is applied at the joints and the joints consist of ball-and-socket connections. These assumptions are justified if the welded or bolted connections of the joined members intersect at a common point and the weight of the members can be neglected. In cases where the weight of a member is to be included in the analysis, it is generally satisfactory to apply it as a vertical force, half of its magnitude applied at each end of the member.

### Procedure for Analysis

Either the method of joints or the method of sections can be used to determine the forces developed in the members of a simple space truss.

#### Method of Joints.

If the forces in *all* the members of the truss are to be determined, then the method of joints is most suitable for the analysis. Here it is necessary to apply the three equilibrium equations  $\Sigma F_x = 0$ ,  $\Sigma F_y = 0$ ,  $\Sigma F_z = 0$  to the forces acting at each joint. Remember that the solution of many simultaneous equations can be avoided if the force analysis begins at a joint having at least one known force and at most three unknown forces. Also, if the three-dimensional geometry of the force system at the joint is hard to visualize, it is recommended that a Cartesian vector analysis be used for the solution.

#### Method of Sections.

If only a *few* member forces are to be determined, the method of sections can be used. When an imaginary section is passed through a truss and the truss is separated into two parts, the force system acting on one of the segments must satisfy the *six* equilibrium equations:  $\Sigma F_x = 0$ ,  $\Sigma F_y = 0$ ,  $\Sigma F_z = 0$ ,  $\Sigma M_x = 0$ ,  $\Sigma M_y = 0$ ,  $\Sigma M_z = 0$  (Eqs. 5–6). By proper choice of the section and axes for summing forces and moments, many of the unknown member forces in a space truss can be computed *directly*, using a single equilibrium equation.

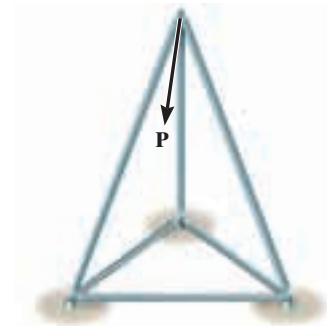
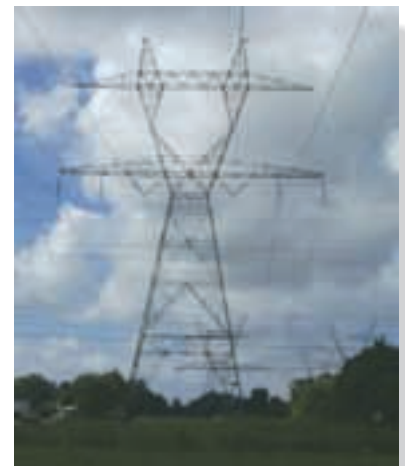


Fig. 6–19



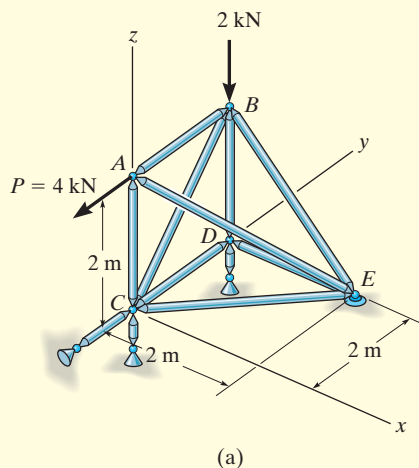
Typical roof-supporting space truss. Notice the use of ball-and-socket joints for the connections. (© Russell C. Hibbeler)



For economic reasons, large electrical transmission towers are often constructed using space trusses. (© Russell C. Hibbeler)



## EXAMPLE 6.8



Determine the forces acting in the members of the space truss shown in Fig. 6-20a. Indicate whether the members are in tension or compression.

## SOLUTION

Since there are one known force and three unknown forces acting at joint A, the force analysis of the truss will begin at this joint.

**Joint A.** (Fig. 6-20b). Expressing each force acting on the free-body diagram of joint A as a Cartesian vector, we have

$$\mathbf{P} = \{-4\mathbf{j}\} \text{ kN}, \quad \mathbf{F}_{AB} = F_{AB}\mathbf{j}, \quad \mathbf{F}_{AC} = -F_{AC}\mathbf{k},$$

$$\mathbf{F}_{AE} = F_{AE}\left(\frac{\mathbf{r}_{AE}}{r_{AE}}\right) = F_{AE}(0.577\mathbf{i} + 0.577\mathbf{j} - 0.577\mathbf{k})$$

For equilibrium,

$$\Sigma \mathbf{F} = \mathbf{0}; \quad \mathbf{P} + \mathbf{F}_{AB} + \mathbf{F}_{AC} + \mathbf{F}_{AE} = \mathbf{0}$$

$$-4\mathbf{j} + F_{AB}\mathbf{j} - F_{AC}\mathbf{k} + 0.577F_{AE}\mathbf{i} + 0.577F_{AE}\mathbf{j} - 0.577F_{AE}\mathbf{k} = \mathbf{0}$$

$$\Sigma F_x = 0; \quad 0.577F_{AE} = 0$$

$$\Sigma F_y = 0; \quad -4 + F_{AB} + 0.577F_{AE} = 0$$

$$\Sigma F_z = 0; \quad -F_{AC} - 0.577F_{AE} = 0$$

$$F_{AC} = F_{AE} = 0$$

Ans.

$$F_{AB} = 4 \text{ kN (T)}$$

Ans.

Since  $F_{AB}$  is known, joint B can be analyzed next.

**Joint B.** (Fig. 6-20c).

$$\Sigma F_x = 0; \quad F_{BE}\frac{1}{\sqrt{2}} = 0$$

$$\Sigma F_y = 0; \quad -4 + F_{CB}\frac{1}{\sqrt{2}} = 0$$

$$\Sigma F_z = 0; \quad -2 + F_{BD} - F_{BE}\frac{1}{\sqrt{2}} + F_{CB}\frac{1}{\sqrt{2}} = 0$$

$$F_{BE} = 0, \quad F_{CB} = 5.65 \text{ kN (C)} \quad F_{BD} = 2 \text{ kN (T)} \quad \text{Ans.}$$

The *scalar* equations of equilibrium can now be applied to the forces acting on the free-body diagrams of joints D and C. Show that

$$F_{DE} = F_{DC} = F_{CE} = 0 \quad \text{Ans.}$$

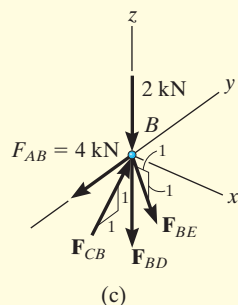
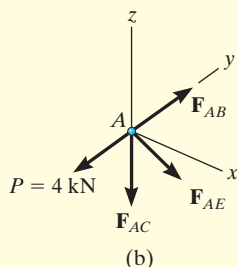
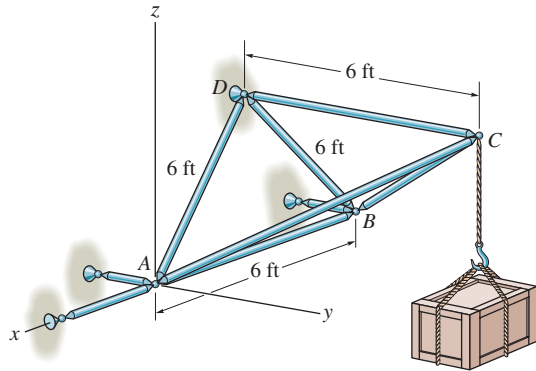


Fig. 6-20

## PROBLEMS

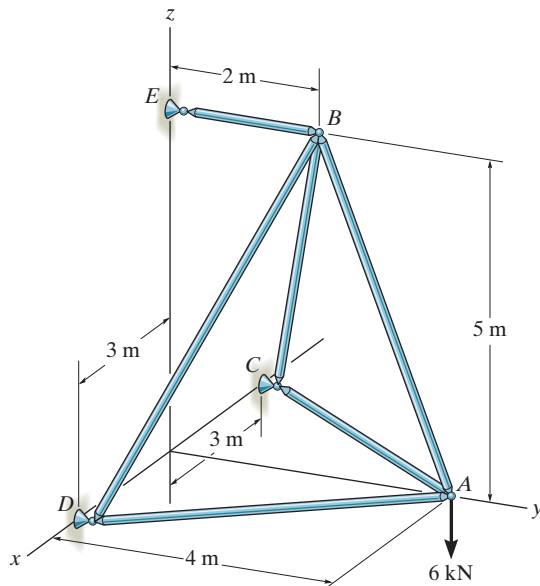
**All problem solutions must include FBDs.**

**6-50.** Determine the force developed in each member of the space truss and state if the members are in tension or compression. The crate has a weight of 150 lb.



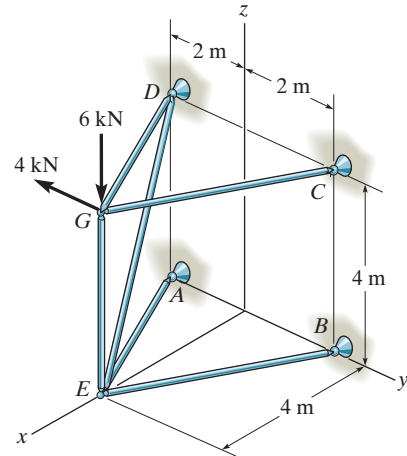
**Prob. 6-50**

**6-51.** Determine the force in each member of the space truss and state if the members are in tension or compression. *Hint:* The support reaction at *E* acts along member *EB*. Why?



**Prob. 6-51**

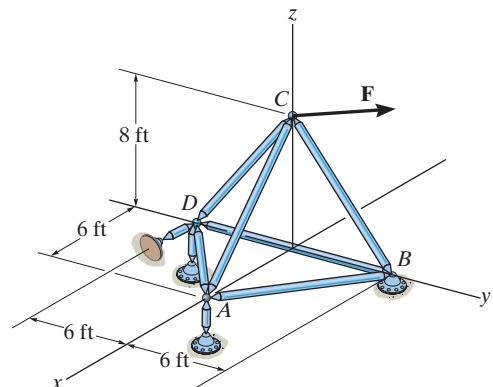
**\*6-52.** Determine the force in each member of the space truss and state if the members are in tension or compression. The truss is supported by ball-and-socket joints at *A*, *B*, *C*, and *D*.



**Prob. 6-52**

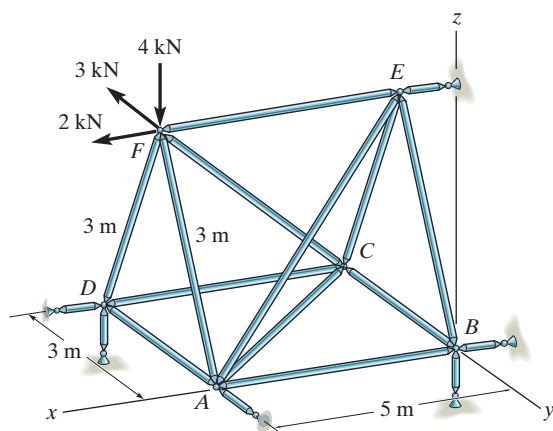
**6-53.** The space truss supports a force  $\mathbf{F} = \{-500\mathbf{i} + 600\mathbf{j} + 400\mathbf{k}\}$  lb. Determine the force in each member, and state if the members are in tension or compression.

**6-54.** The space truss supports a force  $\mathbf{F} = \{600\mathbf{i} + 450\mathbf{j} - 750\mathbf{k}\}$  lb. Determine the force in each member, and state if the members are in tension or compression.



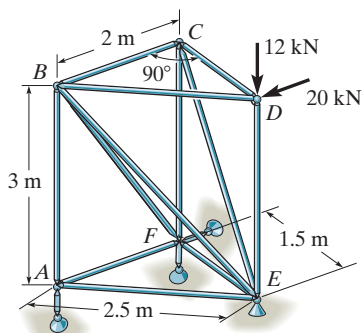
**Probs. 6-53/54**

**6-55.** Determine the force in members  $EF$ ,  $AF$ , and  $DF$  of the space truss and state if the members are in tension or compression. The truss is supported by short links at  $A$ ,  $B$ ,  $D$ , and  $E$ .



**Prob. 6-55**

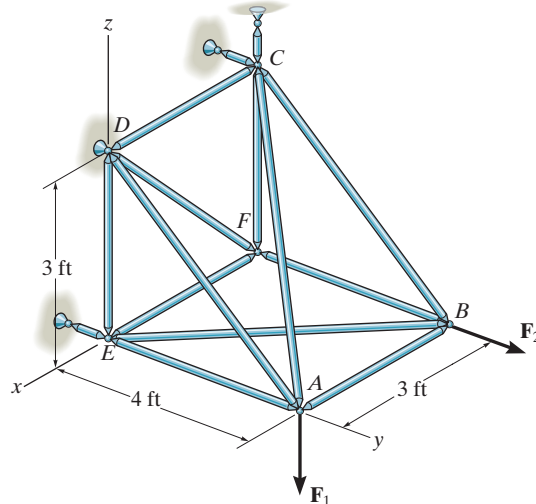
**\*6-56.** The space truss is used to support the forces at joints  $B$  and  $D$ . Determine the force in each member and state if the members are in tension or compression.



**Prob. 6-56**

**6-57.** The space truss is supported by a ball-and-socket joint at  $D$  and short links at  $C$  and  $E$ . Determine the force in each member and state if the members are in tension or compression. Take  $\mathbf{F}_1 = \{-500\mathbf{k}\}$  lb and  $\mathbf{F}_2 = \{400\mathbf{j}\}$  lb.

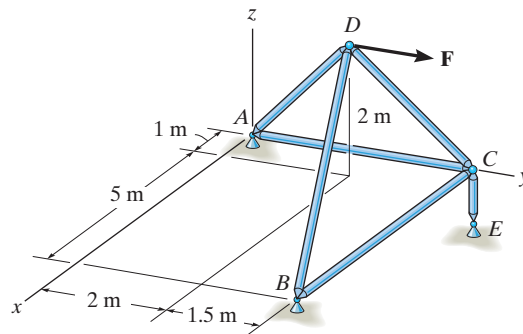
**6-58.** The space truss is supported by a ball-and-socket joint at  $D$  and short links at  $C$  and  $E$ . Determine the force in each member and state if the members are in tension or compression. Take  $\mathbf{F}_1 = \{200\mathbf{i} + 300\mathbf{j} - 500\mathbf{k}\}$  lb and  $\mathbf{F}_2 = \{400\mathbf{j}\}$  lb.



**Probs. 6-57/58**

**6-59.** Determine the force in each member of the space truss and state if the members are in tension or compression. The truss is supported by ball-and-socket joints at  $A$ ,  $B$ , and  $E$ . Set  $\mathbf{F} = \{800\mathbf{j}\}$  N. *Hint:* The support reaction at  $E$  acts along member  $EC$ . Why?

**\*6-60.** Determine the force in each member of the space truss and state if the members are in tension or compression. The truss is supported by ball-and-socket joints at  $A$ ,  $B$ , and  $E$ . Set  $\mathbf{F} = \{-200\mathbf{i} + 400\mathbf{j}\}$  N. *Hint:* The support reaction at  $E$  acts along member  $EC$ . Why?



**Probs. 6-59/60**

## 6.6 Frames and Machines

Frames and machines are two types of structures which are often composed of pin-connected **multiforce members**, i.e., members that are subjected to more than two forces. **Frames** are used to support loads, whereas **machines** contain moving parts and are designed to transmit and alter the effect of forces. Provided a frame or machine contains no more supports or members than are necessary to prevent its collapse, the forces acting at the joints and supports can be determined by applying the equations of equilibrium to each of its members. Once these forces are obtained, it is then possible to *design* the size of the members, connections, and supports using the theory of mechanics of materials and an appropriate engineering design code.

**Free-Body Diagrams.** In order to determine the forces acting at the joints and supports of a frame or machine, the structure must be disassembled and the free-body diagrams of its parts must be drawn. The following important points *must* be observed:

- Isolate each part by drawing its *outlined shape*. Then show all the forces and/or couple moments that act on the part. Make sure to *label* or *identify* each known and unknown force and couple moment with reference to an established  $x, y$  coordinate system. Also, indicate any dimensions used for taking moments. Most often the equations of equilibrium are easier to apply if the forces are represented by their rectangular components. As usual, the sense of an unknown force or couple moment can be assumed.
- Identify all the two-force members in the structure and represent their free-body diagrams as having two equal but opposite collinear forces acting at their points of application. (See Sec. 5.4.) By recognizing the two-force members, we can avoid solving an unnecessary number of equilibrium equations.
- Forces common to *any* two *contacting* members act with equal magnitudes but opposite sense on the respective members. If the two members are treated as a “system” of *connected members*, then these forces are “*internal*” and are *not shown* on the *free-body diagram of the system*; however, if the free-body diagram of *each member* is drawn, the forces are “*external*” and *must* be shown as equal in magnitude and opposite in direction on each of the two free-body diagrams.

The following examples graphically illustrate how to draw the free-body diagrams of a dismembered frame or machine. In all cases, the weight of the members is neglected.



This crane is a typical example of a framework. (© Russell C. Hibbeler)



Common tools such as these pliers act as simple machines. Here the applied force on the handles creates a much larger force at the jaws. (© Russell C. Hibbeler)

## EXAMPLE 6.9

For the frame shown in Fig. 6-21a, draw the free-body diagram of (a) each member, (b) the pins at  $B$  and  $A$ , and (c) the two members connected together.

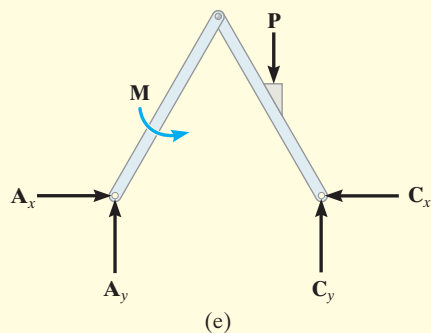
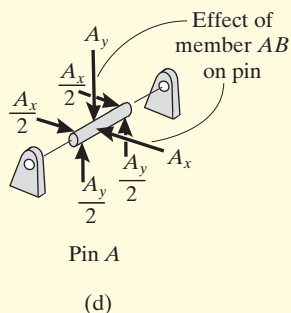
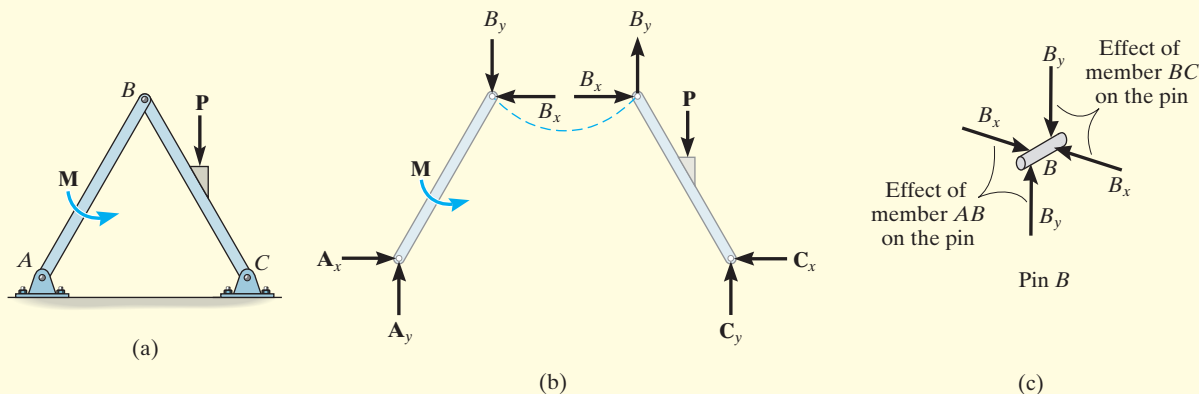


Fig. 6-21

## SOLUTION

**Part (a).** By inspection, members  $BA$  and  $BC$  are *not* two-force members. Instead, as shown on the free-body diagrams, Fig. 6-21b,  $BC$  is subjected to a force from each of the pins at  $B$  and  $C$  and the external force  $\mathbf{P}$ . Likewise,  $AB$  is subjected to a force from each of the pins at  $A$  and  $B$  and the external couple moment  $\mathbf{M}$ . The pin forces are represented by their  $x$  and  $y$  components.

**Part (b).** The pin at  $B$  is subjected to only *two* forces, i.e., the force of member  $BC$  and the force of member  $AB$ . For *equilibrium* these forces (or their respective components) must be equal but opposite, Fig. 6-21c. Realize that Newton's third law is applied between the pin and its connected members, i.e., the effect of the pin on the two members, Fig. 6-21b, and the equal but opposite effect of the two members on the pin, Fig. 6-21c. In the same manner, there are three forces on pin  $A$ , Fig. 6-21d, caused by the force components of member  $AB$  and each of the two pin leaves.

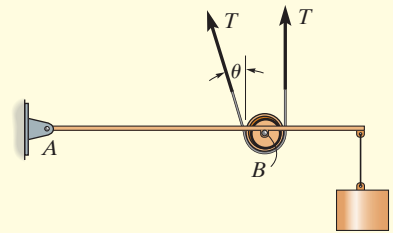
**Part (c).** The free-body diagram of both members connected together, yet removed from the supporting pins at  $A$  and  $C$ , is shown in Fig. 6-21e. The force components  $\mathbf{B}_x$  and  $\mathbf{B}_y$  are *not* shown on this diagram since they are *internal* forces (Fig. 6-21b) and therefore cancel out. Also, to be consistent when later applying the equilibrium equations, the unknown force components at  $A$  and  $C$  must act in the *same* sense as those shown in Fig. 6-21b.

**EXAMPLE 6.10**

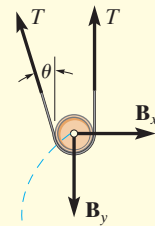
A constant tension in the conveyor belt is maintained by using the device shown in Fig. 6-22*a*. Draw the free-body diagrams of the frame and the cylinder (or pulley) that the belt surrounds. The suspended block has a weight of  $W$ .



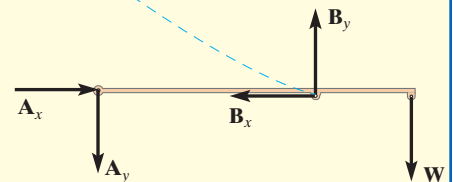
(a)

**Fig. 6-22** (© Russell C. Hibbeler)

(b)



(c)



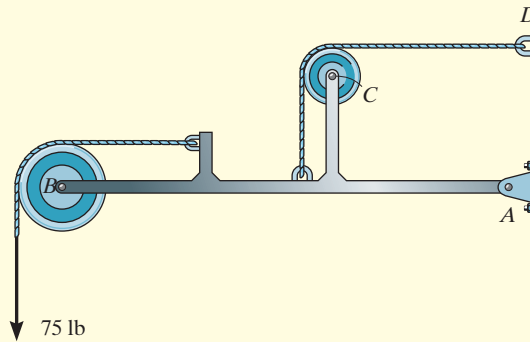
(d)

**SOLUTION**

The idealized model of the device is shown in Fig. 6-22*b*. Here the angle  $\theta$  is assumed to be known. From this model, the free-body diagrams of the pulley and frame are shown in Figs. 6-22*c* and 6-22*d*, respectively. Note that the force components  $B_x$  and  $B_y$  that the pin at  $B$  exerts on the pulley must be equal but opposite to the ones acting on the frame. See Fig. 6-21*c* of Example 6.9.

**EXAMPLE 6.11**

For the frame shown in Fig. 6–23a, draw the free-body diagrams of (a) the entire frame including the pulleys and cords, (b) the frame without the pulleys and cords, and (c) each of the pulleys.



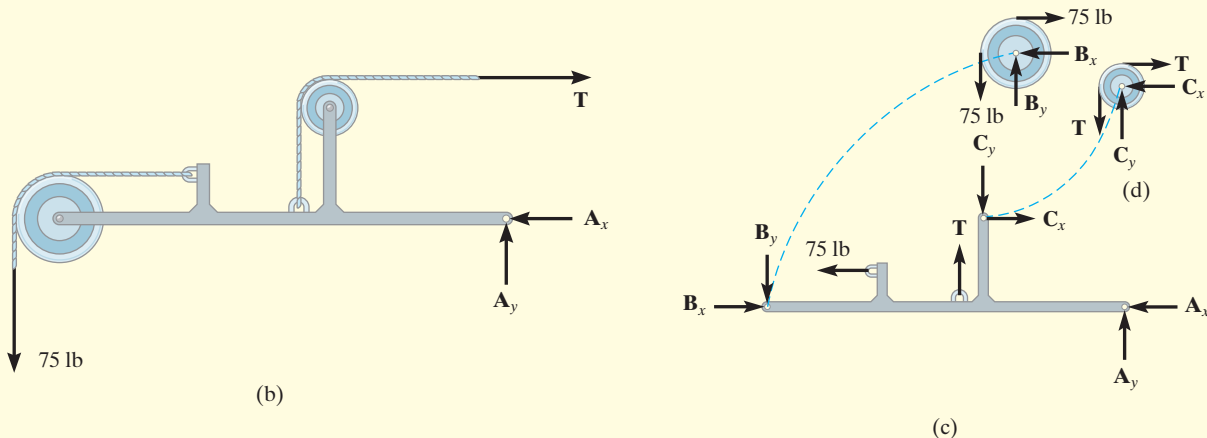
(a)

**SOLUTION**

**Part (a).** When the entire frame including the pulleys and cords is considered, the interactions at the points where the pulleys and cords are connected to the frame become pairs of *internal* forces which cancel each other and therefore are not shown on the free-body diagram, Fig. 6–23b.

**Part (b).** When the cords and pulleys are removed, their effect *on the frame* must be shown, Fig. 6–23c.

**Part (c).** The force components  $B_x$ ,  $B_y$ ,  $C_x$ ,  $C_y$  of the pins on the pulleys, Fig. 6–23d, are equal but opposite to the force components exerted by the pins on the frame, Fig. 6–23c. See Example 6.9.

**Fig. 6–23**

**EXAMPLE 6.12**

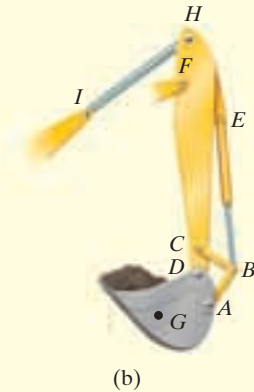
Draw the free-body diagrams of the members of the backhoe, shown in the photo, Fig. 6–24a. The bucket and its contents have a weight  $W$ .

**SOLUTION**

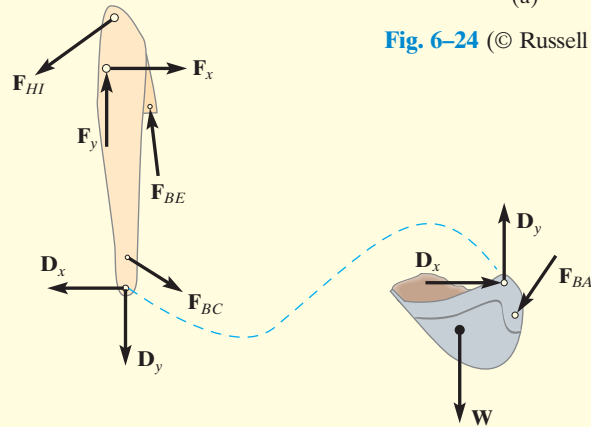
The idealized model of the assembly is shown in Fig. 6–24b. By inspection, members  $AB$ ,  $BC$ ,  $BE$ , and  $HI$  are all two-force members since they are pin connected at their end points and no other forces act on them. The free-body diagrams of the bucket and the stick are shown in Fig. 6–24c. Note that pin  $C$  is subjected to only two forces, whereas the pin at  $B$  is subjected to three forces, Fig. 6–24d. The free-body diagram of the entire assembly is shown in Fig. 6–24e.



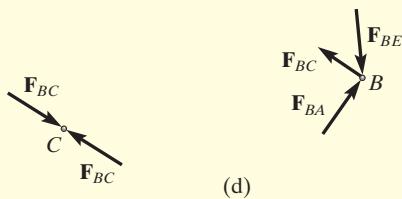
(a)

**Fig. 6–24** (© Russell C. Hibbeler)

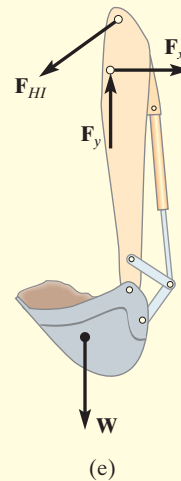
(b)



(c)



(d)

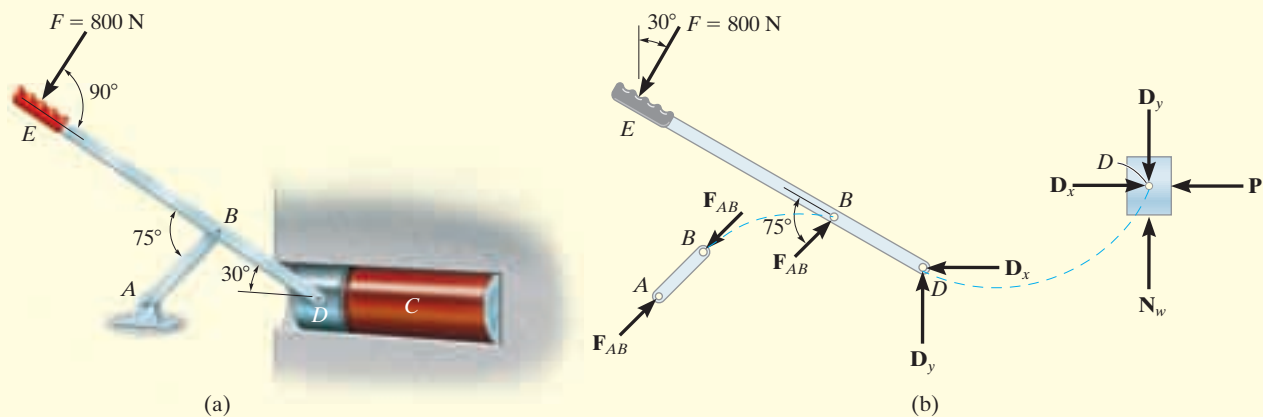


(e)



**EXAMPLE 6.13**

Draw the free-body diagram of each part of the smooth piston and link mechanism used to crush recycled cans, Fig. 6–25a.

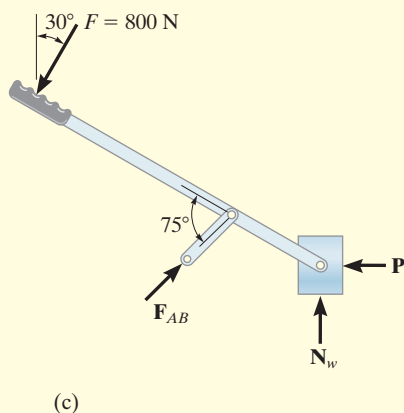


**Fig. 6–25**

**SOLUTION**

By inspection, member  $AB$  is a two-force member. The free-body diagrams of the three parts are shown in Fig. 6–25b. Since the pins at  $B$  and  $D$  connect *only two parts together*, the forces there are shown as equal but opposite on the separate free-body diagrams of their connected members. In particular, four components of force act on the piston:  $D_x$  and  $D_y$  represent the effect of the pin (or lever  $EBD$ ),  $N_w$  is the *resultant* force of the wall support, and  $P$  is the resultant compressive force caused by the can  $C$ . The directional sense of each of the unknown forces is assumed, and the correct sense will be established after the equations of equilibrium are applied.

**NOTE:** A free-body diagram of the entire assembly is shown in Fig. 6–25c. Here the forces between the components are internal and are not shown on the free-body diagram.



*Before proceeding, it is highly recommended that you cover the solutions of these examples and attempt to draw the requested free-body diagrams. When doing so, make sure the work is neat and that all the forces and couple moments are properly labeled.*

## Procedure for Analysis

The joint reactions on frames or machines (structures) composed of multiforce members can be determined using the following procedure.

### Free-Body Diagram.

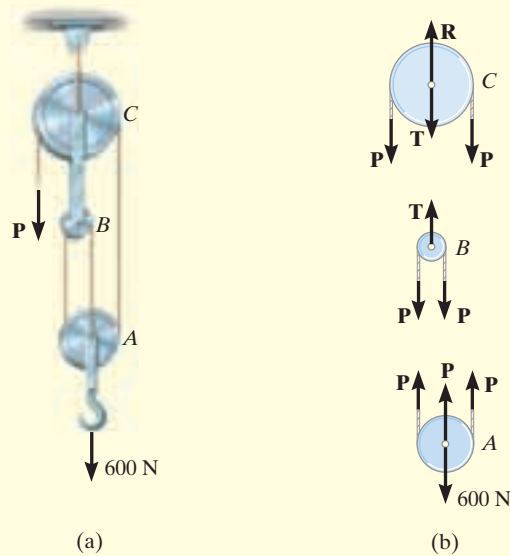
- Draw the free-body diagram of the entire frame or machine, a portion of it, or each of its members. The choice should be made so that it leads to the most direct solution of the problem.
- Identify the two-force members. Remember that regardless of their shape, they have equal but opposite collinear forces acting at their ends.
- When the free-body diagram of a group of members of a frame or machine is drawn, the forces between the connected parts of this group are internal forces and are not shown on the free-body diagram of the group.
- Forces common to two members which are in contact act with equal magnitude but opposite sense on the respective free-body diagrams of the members.
- In many cases it is possible to tell by inspection the proper sense of the unknown forces acting on a member; however, if this seems difficult, the sense can be assumed.
- Remember that once the free-body diagram is drawn, a couple moment is a free vector and can act at any point on the diagram. Also, a force is a sliding vector and can act at any point along its line of action.

### Equations of Equilibrium.

- Count the number of unknowns and compare it to the total number of equilibrium equations that are available. In two dimensions, there are three equilibrium equations that can be written for each member.
- Sum moments about a point that lies at the intersection of the lines of action of as many of the unknown forces as possible.
- If the solution of a force or couple moment magnitude is found to be negative, it means the sense of the force is the reverse of that shown on the free-body diagram.

**EXAMPLE 6.14**

Determine the tension in the cables and also the force **P** required to support the 600-N force using the frictionless pulley system shown in Fig. 6-26a.

**Fig. 6-26****SOLUTION**

**Free-Body Diagram.** A free-body diagram of each pulley *including* its pin and a portion of the contacting cable is shown in Fig. 6-26b. Since the cable is *continuous*, it has a *constant tension*  $P$  acting throughout its length. The link connection between pulleys  $B$  and  $C$  is a two-force member, and therefore it has an unknown tension  $T$  acting on it. Notice that the *principle of action, equal but opposite reaction* must be carefully observed for forces  $\mathbf{P}$  and  $\mathbf{T}$  when the *separate* free-body diagrams are drawn.

**Equations of Equilibrium.** The three unknowns are obtained as follows:

**Pulley A**

$$+\uparrow \Sigma F_y = 0; \quad 3P - 600 \text{ N} = 0 \quad P = 200 \text{ N} \quad \text{Ans.}$$

**Pulley B**

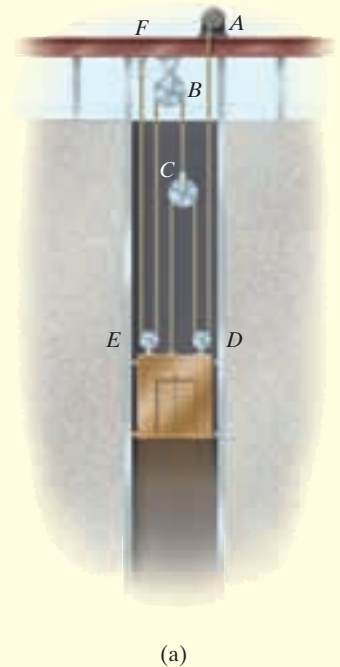
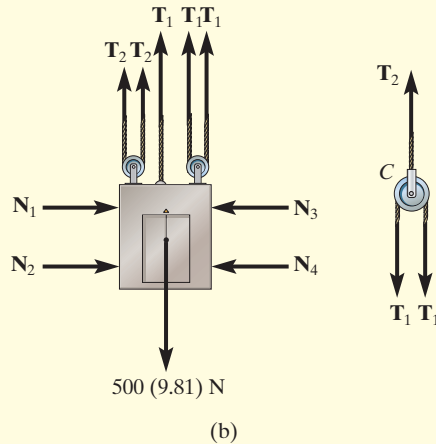
$$+\uparrow \Sigma F_y = 0; \quad T - 2P = 0 \quad T = 400 \text{ N} \quad \text{Ans.}$$

**Pulley C**

$$+\uparrow \Sigma F_y = 0; \quad R - 2P - T = 0 \quad R = 800 \text{ N} \quad \text{Ans.}$$

**EXAMPLE 6.15**

A 500-kg elevator car in Fig. 6-27a is being hoisted by motor *A* using the pulley system shown. If the car is traveling with a constant speed, determine the force developed in the two cables. Neglect the mass of the cable and pulleys.

**Fig. 6-27****SOLUTION**

**Free-Body Diagram.** We can solve this problem using the free-body diagrams of the elevator car and pulley *C*, Fig. 6-27b. The tensile forces developed in the cables are denoted as  $T_1$  and  $T_2$ .

**Equations of Equilibrium.** For pulley *C*,

$$+\uparrow \Sigma F_y = 0; \quad T_2 - 2T_1 = 0 \quad \text{or} \quad T_2 = 2T_1 \quad (1)$$

For the elevator car,

$$+\uparrow \Sigma F_y = 0; \quad 3T_1 + 2T_2 - 500(9.81) \text{ N} = 0 \quad (2)$$

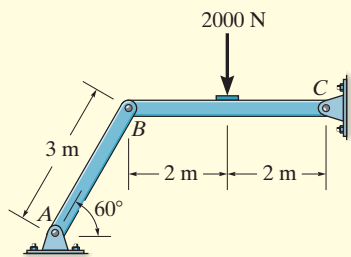
Substituting Eq. (1) into Eq. (2) yields

$$\begin{aligned} 3T_1 + 2(2T_1) - 500(9.81) \text{ N} &= 0 \\ T_1 &= 700.71 \text{ N} = 701 \text{ N} \quad \text{Ans.} \end{aligned}$$

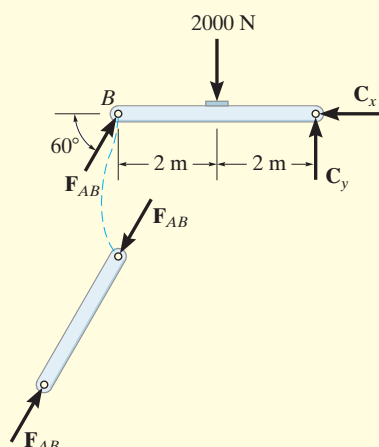
Substituting this result into Eq. (1),

$$T_2 = 2(700.71) \text{ N} = 1401 \text{ N} = 1.40 \text{ kN} \quad \text{Ans.}$$

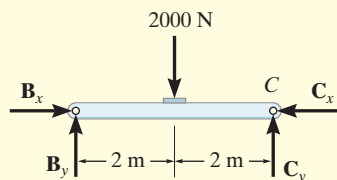
## EXAMPLE 6.16



(a)



(b)



(c)

Fig. 6-28

Determine the horizontal and vertical components of force which the pin at  $C$  exerts on member  $BC$  of the frame in Fig. 6-28a.

## SOLUTION I

**Free-Body Diagrams.** By inspection it can be seen that  $AB$  is a two-force member. The free-body diagrams are shown in Fig. 6-28b.

**Equations of Equilibrium.** The *three unknowns* can be determined by applying the three equations of equilibrium to member  $BC$ .

$$\zeta + \Sigma M_C = 0; 2000 \text{ N}(2 \text{ m}) - (F_{AB} \sin 60^\circ)(4 \text{ m}) = 0 \quad F_{AB} = 1154.7 \text{ N}$$

$$\rightarrow \Sigma F_x = 0; 1154.7 \cos 60^\circ \text{ N} - C_x = 0 \quad C_x = 577 \text{ N} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; 1154.7 \sin 60^\circ \text{ N} - 2000 \text{ N} + C_y = 0$$

$$C_y = 1000 \text{ N} \quad \text{Ans.}$$

## SOLUTION II

**Free-Body Diagrams.** If one does not recognize that  $AB$  is a two-force member, then more work is involved in solving this problem. The free-body diagrams are shown in Fig. 6-28c.

**Equations of Equilibrium.** The *six unknowns* are determined by applying the three equations of equilibrium to each member.

## Member AB

$$\zeta + \Sigma M_A = 0; B_x(3 \sin 60^\circ \text{ m}) - B_y(3 \cos 60^\circ \text{ m}) = 0 \quad (1)$$

$$\rightarrow \Sigma F_x = 0; A_x - B_x = 0 \quad (2)$$

$$+\uparrow \Sigma F_y = 0; A_y - B_y = 0 \quad (3)$$

## Member BC

$$\zeta + \Sigma M_C = 0; 2000 \text{ N}(2 \text{ m}) - B_y(4 \text{ m}) = 0 \quad (4)$$

$$\rightarrow \Sigma F_x = 0; B_x - C_x = 0 \quad (5)$$

$$+\uparrow \Sigma F_y = 0; B_y - 2000 \text{ N} + C_y = 0 \quad (6)$$

The results for  $C_x$  and  $C_y$  can be determined by solving these equations in the following sequence: 4, 1, 5, then 6. The results are

$$B_y = 1000 \text{ N}$$

$$B_x = 577 \text{ N}$$

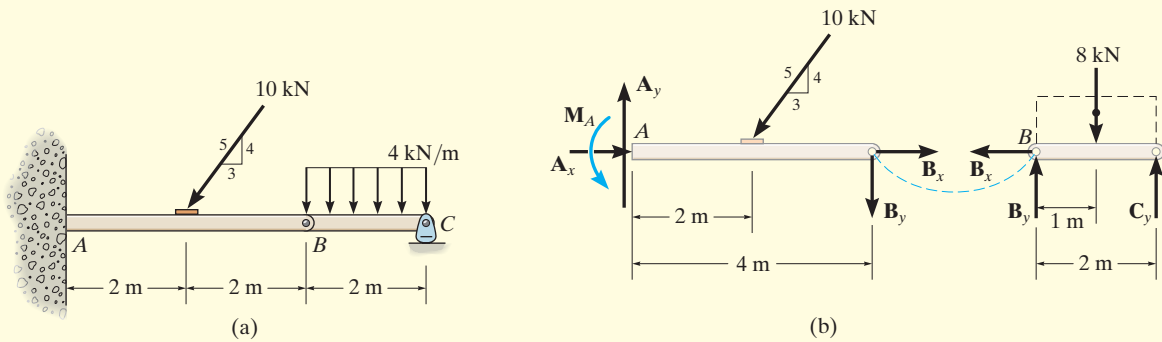
$$C_x = 577 \text{ N} \quad \text{Ans.}$$

$$C_y = 1000 \text{ N} \quad \text{Ans.}$$

By comparison, Solution I is simpler since the requirement that  $F_{AB}$  in Fig. 6-28b be equal, opposite, and collinear at the ends of member  $AB$  automatically satisfies Eqs. 1, 2, and 3 above and therefore eliminates the need to write these equations. **As a result, save yourself some time and effort by always identifying the two-force members before starting the analysis!**

**EXAMPLE 6.17**

The compound beam shown in Fig. 6–29a is pin connected at  $B$ . Determine the components of reaction at its supports. Neglect its weight and thickness.

**Fig. 6–29****SOLUTION**

**Free-Body Diagrams.** By inspection, if we consider a free-body diagram of the *entire beam*  $ABC$ , there will be three unknown reactions at  $A$  and one at  $C$ . These four unknowns cannot all be obtained from the three available equations of equilibrium, and so for the solution it will become necessary to dismember the beam into its two segments, as shown in Fig. 6–29b.

**Equations of Equilibrium.** The six unknowns are determined as follows:

*Segment BC*

$$\pm \Sigma F_x = 0; \quad B_x = 0$$

$$\zeta + \Sigma M_B = 0; \quad -8 \text{ kN}(1 \text{ m}) + C_y(2 \text{ m}) = 0$$

$$+ \uparrow \Sigma F_y = 0; \quad B_y - 8 \text{ kN} + C_y = 0$$

*Segment AB*

$$\pm \Sigma F_x = 0; \quad A_x - (10 \text{ kN})\left(\frac{3}{5}\right) + B_x = 0$$

$$\zeta + \Sigma M_A = 0; \quad M_A - (10 \text{ kN})\left(\frac{4}{5}\right)(2 \text{ m}) - B_y(4 \text{ m}) = 0$$

$$+ \uparrow \Sigma F_y = 0; \quad A_y - (10 \text{ kN})\left(\frac{4}{5}\right) - B_y = 0$$

Solving each of these equations successively, using previously calculated results, we obtain

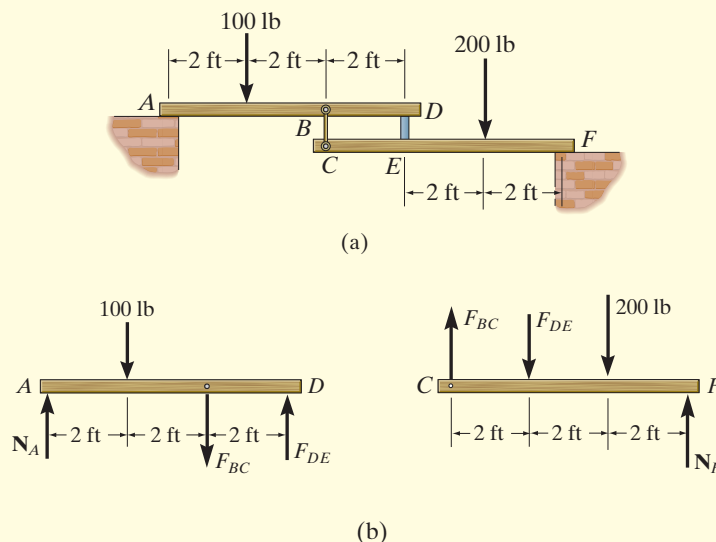
$$A_x = 6 \text{ kN} \quad A_y = 12 \text{ kN} \quad M_A = 32 \text{ kN} \cdot \text{m} \quad \text{Ans.}$$

$$B_x = 0 \quad B_y = 4 \text{ kN}$$

$$C_y = 4 \text{ kN} \quad \text{Ans.}$$

**EXAMPLE 6.18**

The two planks in Fig. 6–30a are connected together by cable  $BC$  and a smooth spacer  $DE$ . Determine the reactions at the smooth supports  $A$  and  $F$ , and also find the force developed in the cable and spacer.

**Fig. 6–30****SOLUTION**

**Free-Body Diagrams.** The free-body diagram of each plank is shown in Fig. 6–30b. It is important to apply Newton's third law to the interaction forces  $F_{BC}$  and  $F_{DE}$  as shown.

**Equations of Equilibrium.** For plank  $AD$ ,

$$\zeta + \Sigma M_A = 0; \quad F_{DE}(6 \text{ ft}) - F_{BC}(4 \text{ ft}) - 100 \text{ lb}(2 \text{ ft}) = 0$$

For plank  $CF$ ,

$$\zeta + \Sigma M_F = 0; \quad F_{DE}(4 \text{ ft}) - F_{BC}(6 \text{ ft}) + 200 \text{ lb}(2 \text{ ft}) = 0$$

Solving simultaneously,

$$F_{DE} = 140 \text{ lb} \quad F_{BC} = 160 \text{ lb} \quad \text{Ans.}$$

Using these results, for plank  $AD$ ,

$$+\uparrow \Sigma F_y = 0; \quad N_A + 140 \text{ lb} - 160 \text{ lb} - 100 \text{ lb} = 0$$

$$N_A = 120 \text{ lb} \quad \text{Ans.}$$

And for plank  $CF$ ,

$$+\uparrow \Sigma F_y = 0; \quad N_F + 160 \text{ lb} - 140 \text{ lb} - 200 \text{ lb} = 0$$

$$N_F = 180 \text{ lb} \quad \text{Ans.}$$

**NOTE:** Draw the free-body diagram of the system of *both* planks and apply  $\Sigma M_A = 0$  to determine  $N_F$ . Then use the free-body diagram of  $CEF$  to determine  $F_{DE}$  and  $F_{BC}$ .

**EXAMPLE 6.19**

The 75-kg man in Fig. 6–31*a* attempts to lift the 40-kg uniform beam off the roller support at *B*. Determine the tension developed in the cable attached to *B* and the normal reaction of the man on the beam when this is about to occur.

**SOLUTION**

**Free-Body Diagrams.** The tensile force in the cable will be denoted as  $T_1$ . The free-body diagrams of the pulley *E*, the man, and the beam are shown in Fig. 6–31*b*. Since the man must lift the beam off the roller support *B* then  $N_B = 0$ . When drawing each of these diagrams, it is very important to apply Newton's third law.

**Equations of Equilibrium.** Using the free-body diagram of pulley *E*,

$$+\uparrow \Sigma F_y = 0; \quad 2T_1 - T_2 = 0 \quad \text{or} \quad T_2 = 2T_1 \quad (1)$$

Referring to the free-body diagram of the man using this result,

$$+\uparrow \Sigma F_y = 0 \quad N_m + 2T_1 - 75(9.81) \text{ N} = 0 \quad (2)$$

Summing moments about point *A* on the beam,

$$\zeta + \Sigma M_A = 0; \quad T_1(3 \text{ m}) - N_m(0.8 \text{ m}) - [40(9.81) \text{ N}](1.5 \text{ m}) = 0 \quad (3)$$

Solving Eqs. 2 and 3 simultaneously for  $T_1$  and  $N_m$ , then using Eq. (1) for  $T_2$ , we obtain

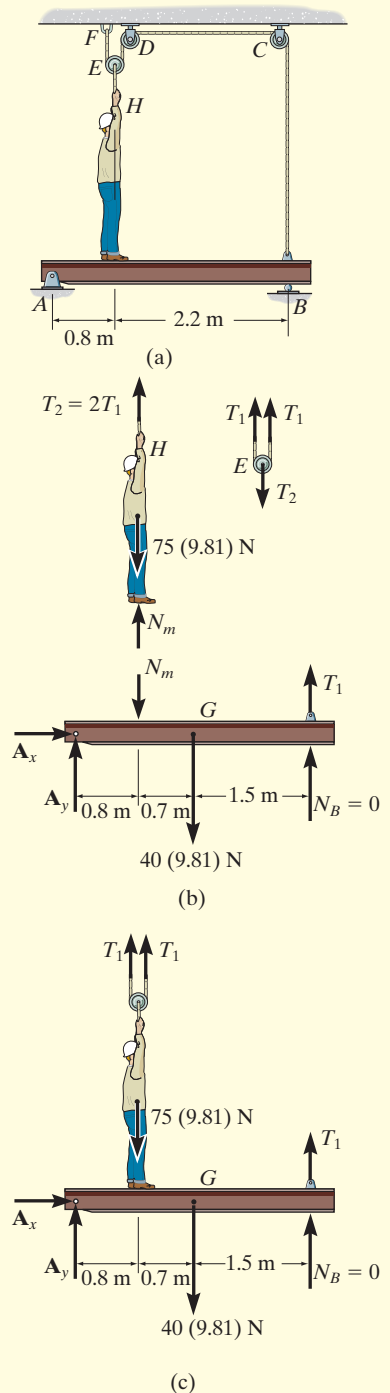
$$T_1 = 256 \text{ N} \quad N_m = 224 \text{ N} \quad T_2 = 512 \text{ N} \quad \text{Ans.}$$

**SOLUTION II**

A direct solution for  $T_1$  can be obtained by considering the beam, the man, and pulley *E* as a *single system*. The free-body diagram is shown in Fig. 6–31*c*. Thus,

$$\begin{aligned} \zeta + \Sigma M_A = 0; \quad & 2T_1(0.8 \text{ m}) - [75(9.81) \text{ N}](0.8 \text{ m}) \\ & - [40(9.81) \text{ N}](1.5 \text{ m}) + T_1(3 \text{ m}) = 0 \\ & T_1 = 256 \text{ N} \quad \text{Ans.} \end{aligned}$$

With this result Eqs. 1 and 2 can then be used to find  $N_m$  and  $T_2$ .

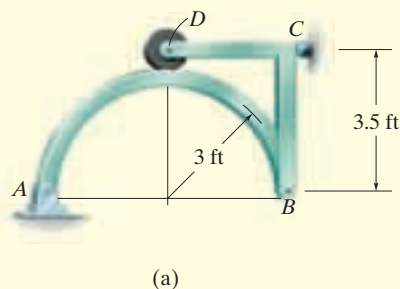
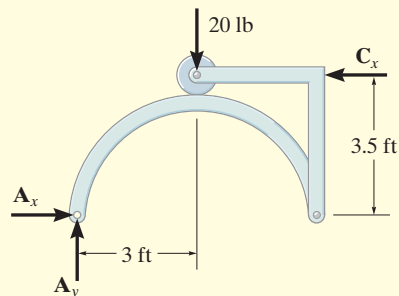


**Fig. 6–31**



## EXAMPLE 6.20

The smooth disk shown in Fig. 6–32a is pinned at  $D$  and has a weight of 20 lb. Neglecting the weights of the other members, determine the horizontal and vertical components of reaction at pins  $B$  and  $D$ .



## SOLUTION

**Free-Body Diagrams.** The free-body diagrams of the entire frame and each of its members are shown in Fig. 6–32b.

**Equations of Equilibrium.** The eight unknowns can of course be obtained by applying the eight equilibrium equations to each member—three to member  $AB$ , three to member  $BCD$ , and two to the disk. (Moment equilibrium is automatically satisfied for the disk.) If this is done, however, all the results can be obtained only from a simultaneous solution of some of the equations. (Try it and find out.) To avoid this situation, it is best first to determine the three support reactions on the *entire* frame; then, using these results, the remaining five equilibrium equations can be applied to two other parts in order to solve successively for the other unknowns.

*Entire Frame*

$$\zeta + \sum M_A = 0; \quad -20 \text{ lb}(3 \text{ ft}) + C_x(3.5 \text{ ft}) = 0 \quad C_x = 17.1 \text{ lb}$$

$$\rightarrow \sum F_x = 0; \quad A_x - 17.1 \text{ lb} = 0 \quad A_x = 17.1 \text{ lb}$$

$$+\uparrow \sum F_y = 0; \quad A_y - 20 \text{ lb} = 0 \quad A_y = 20 \text{ lb}$$

*Member AB*

$$\rightarrow \sum F_x = 0; \quad 17.1 \text{ lb} - B_x = 0 \quad B_x = 17.1 \text{ lb} \quad \text{Ans.}$$

$$\zeta + \sum M_B = 0; \quad -20 \text{ lb}(6 \text{ ft}) + N_D(3 \text{ ft}) = 0 \quad N_D = 40 \text{ lb}$$

$$+\uparrow \sum F_y = 0; \quad 20 \text{ lb} - 40 \text{ lb} + B_y = 0 \quad B_y = 20 \text{ lb} \quad \text{Ans.}$$

*Disk*

$$\rightarrow \sum F_x = 0; \quad D_x = 0 \quad \text{Ans.}$$

$$+\uparrow \sum F_y = 0; \quad 40 \text{ lb} - 20 \text{ lb} - D_y = 0 \quad D_y = 20 \text{ lb} \quad \text{Ans.}$$

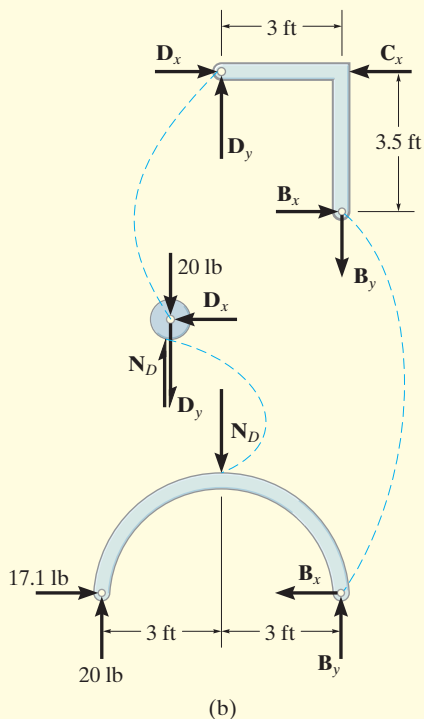
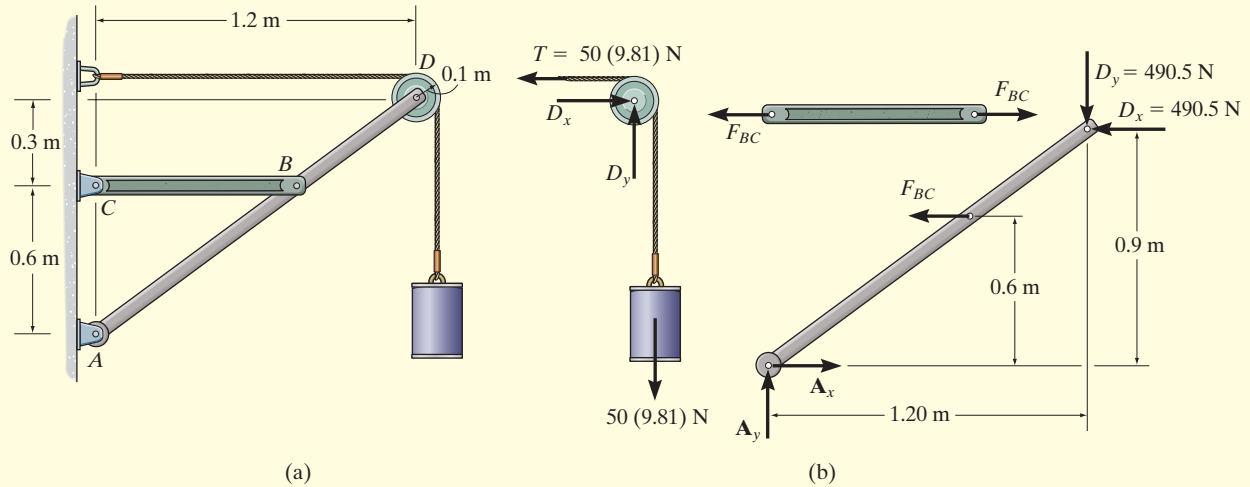


Fig. 6–32

**EXAMPLE 6.21**

The frame in Fig. 6–33a supports the 50-kg cylinder. Determine the horizontal and vertical components of reaction at A and the force at C.

**Fig. 6–33****SOLUTION**

**Free-Body Diagrams.** The free-body diagram of pulley *D*, along with the cylinder and a portion of the cord (a system), is shown in Fig. 6–33*b*. Member *BC* is a two-force member as indicated by its free-body diagram. The free-body diagram of member *ABD* is also shown.

**Equations of Equilibrium.** We will begin by analyzing the equilibrium of the pulley. The moment equation of equilibrium is automatically satisfied with  $T = 50(9.81)$  N, and so

$$\begin{aligned} \rightarrow \Sigma F_x &= 0; & D_x - 50(9.81) \text{ N} &= 0 & D_x &= 490.5 \text{ N} \\ + \uparrow \Sigma F_y &= 0; & D_y - 50(9.81) \text{ N} &= 0 & D_y &= 490.5 \text{ N} \end{aligned} \quad \text{Ans.}$$

Using these results,  $F_{BC}$  can be determined by summing moments about point *A* on member *ABD*.

$$\begin{aligned} \zeta + \Sigma M_A &= 0; & F_{BC}(0.6 \text{ m}) + 490.5 \text{ N}(0.9 \text{ m}) - 490.5 \text{ N}(1.20 \text{ m}) &= 0 \\ & & F_{BC} &= 245.25 \text{ N} \end{aligned} \quad \text{Ans.}$$

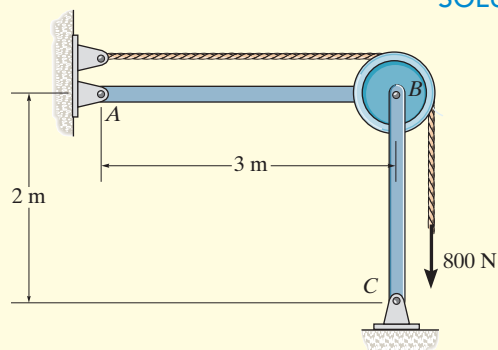
Now  $A_x$  and  $A_y$  can be determined by summing forces.

$$\begin{aligned} \rightarrow \Sigma F_x &= 0; & A_x - 245.25 \text{ N} - 490.5 \text{ N} &= 0 & A_x &= 736 \text{ N} \\ + \uparrow \Sigma F_y &= 0; & A_y - 490.5 \text{ N} &= 0 & A_y &= 490.5 \text{ N} \end{aligned} \quad \text{Ans.}$$

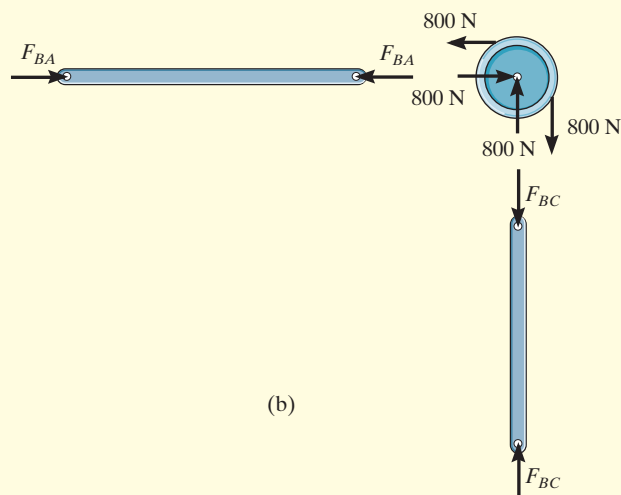
## EXAMPLE 6.22

Determine the force the pins at  $A$  and  $B$  exert on the two-member frame shown in Fig. 6–34a.

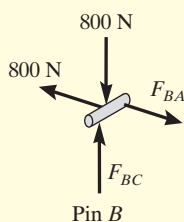
## SOLUTION I



(a)

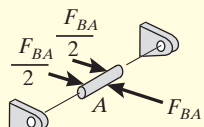


(b)



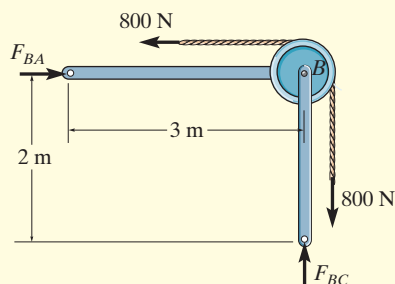
Pin B

(c)



Pin A

(d)



(e)

Fig. 6–34

**Free-Body Diagrams.** By inspection  $AB$  and  $BC$  are two-force members. Their free-body diagrams, along with that of the pulley, are shown in Fig. 6–34b. In order to solve this problem we must also include the free-body diagram of the pin at  $B$  because this pin connects all *three* members together, Fig. 6–34c.

**Equations of Equilibrium:** Apply the equations of force equilibrium to pin  $B$ .

$$+\rightarrow \Sigma F_x = 0; \quad F_{BA} - 800 \text{ N} = 0; \quad F_{BA} = 800 \text{ N} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad F_{BC} - 800 \text{ N} = 0; \quad F_{BC} = 800 \text{ N} \quad \text{Ans.}$$

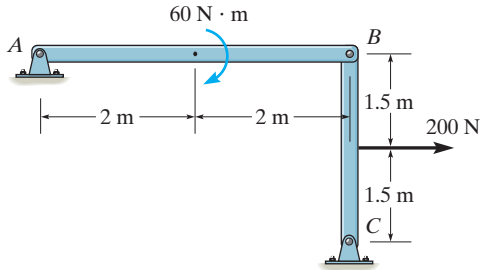
**NOTE:** The free-body diagram of the pin at  $A$ , Fig. 6–34d, indicates how the force  $F_{AB}$  is balanced by the force  $(F_{AB}/2)$  exerted on the pin by each of the two pin leaves.

## SOLUTION II

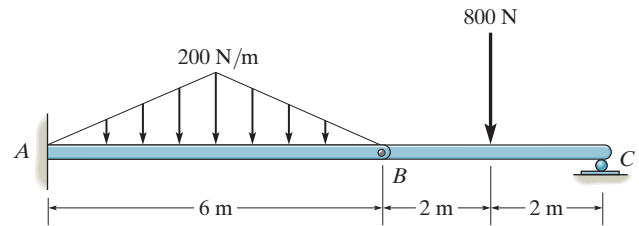
**Free-Body Diagram.** If we realize that  $AB$  and  $BC$  are two-force members, then the free-body diagram of the *entire* frame produces an easier solution, Fig. 6–34e. The force equations of equilibrium are the same as those above. Note that moment equilibrium will be satisfied, regardless of the radius of the pulley.

## PRELIMINARY PROBLEMS

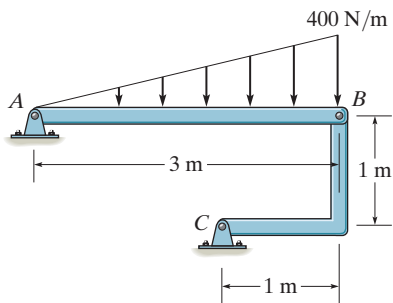
**P6-3.** In each case, identify any two-force members, and then draw the free-body diagrams of each member of the frame.



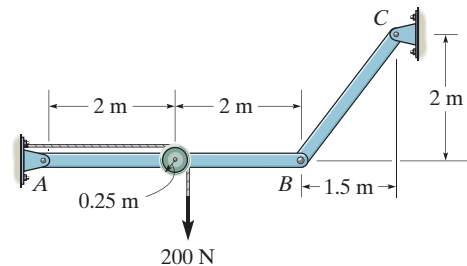
(a)



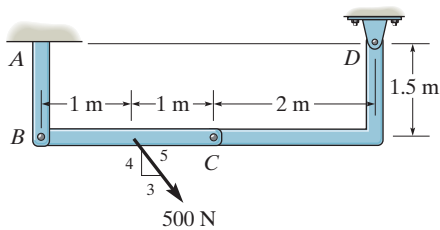
(d)



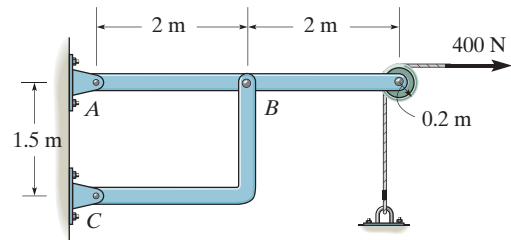
(b)



(e)



(c)



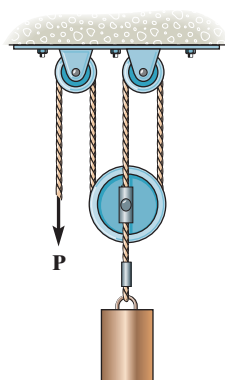
(f)

**Prob. P6-3**

## FUNDAMENTAL PROBLEMS

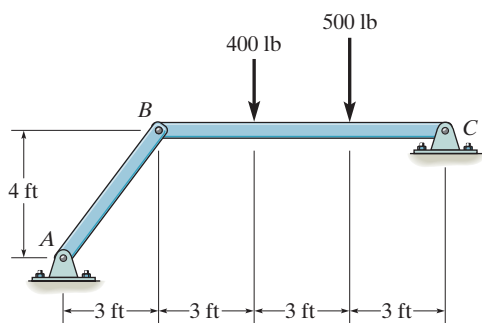
**All problem solutions must include FBDs.**

**F6-13.** Determine the force  $P$  needed to hold the 60-lb weight in equilibrium.



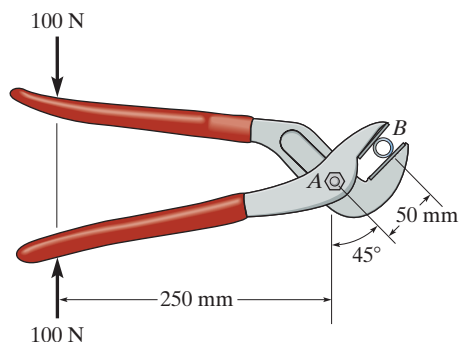
**Prob. F6-13**

**F6-14.** Determine the horizontal and vertical components of reaction at pin  $C$ .



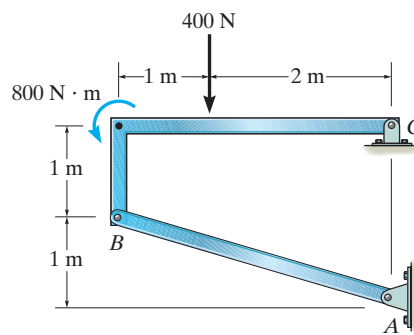
**Prob. F6-14**

**F6-15.** If a 100-N force is applied to the handles of the pliers, determine the clamping force exerted on the smooth pipe  $B$  and the magnitude of the resultant force that one of the members exerts on pin  $A$ .



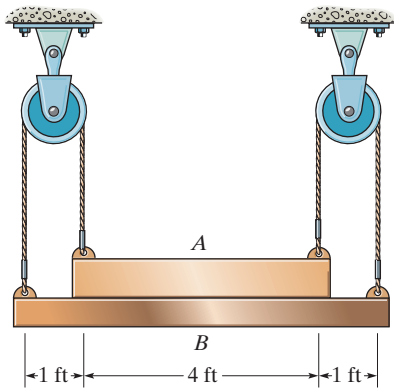
**Prob. F6-15**

**F6-16.** Determine the horizontal and vertical components of reaction at pin  $C$ .



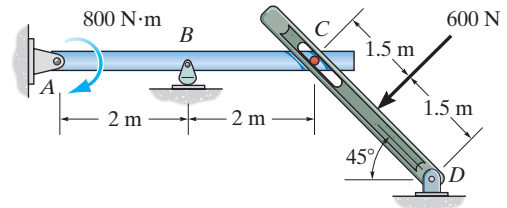
**Prob. F6-16**

**F6-17.** Determine the normal force that the 100-lb plate *A* exerts on the 30-lb plate *B*.



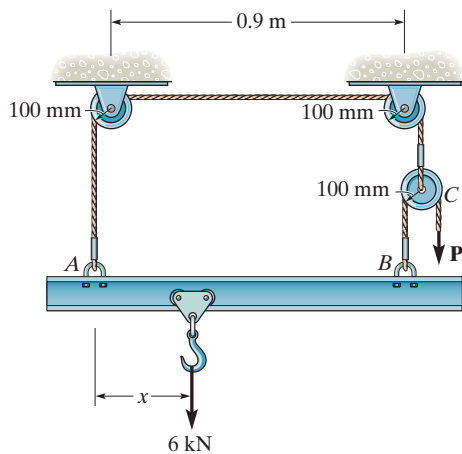
**Prob. F6-17**

**F6-19.** Determine the components of reaction at *A* and *B*.



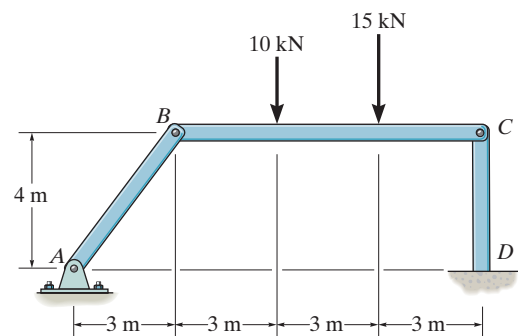
**Prob. F6-19**

**F6-18.** Determine the force *P* needed to lift the load. Also, determine the proper placement *x* of the hook for equilibrium. Neglect the weight of the beam.



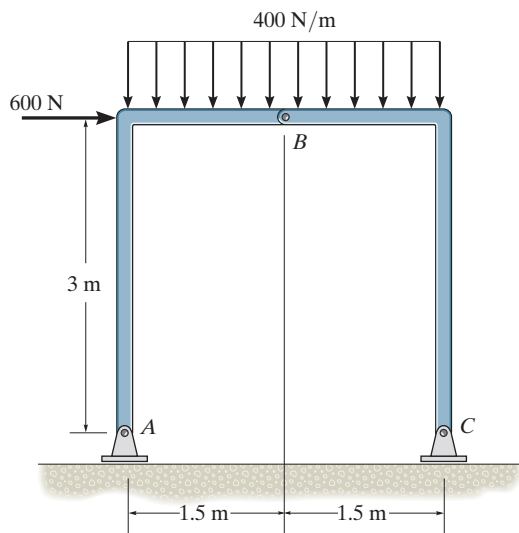
**Prob. F6-18**

**F6-20.** Determine the reactions at *D*.



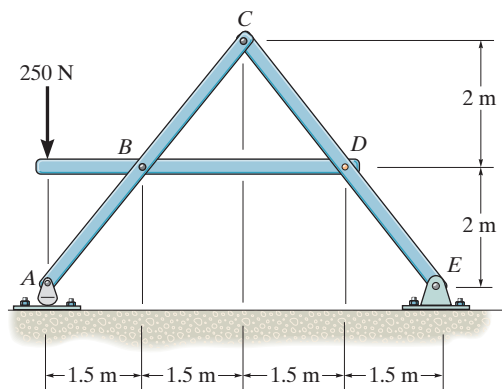
**Prob. F6-20**

**F6-21.** Determine the components of reaction at  $A$  and  $C$ .



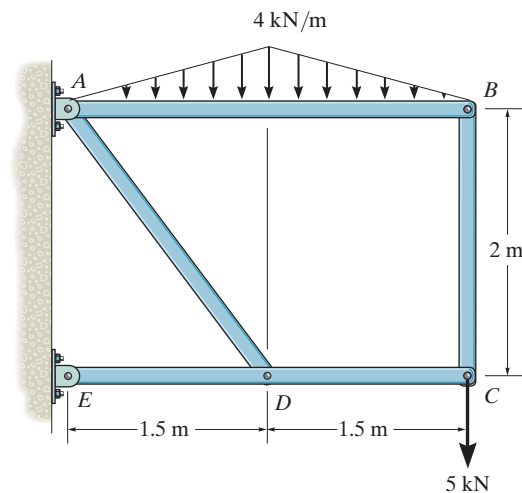
**Prob. F6-21**

**F6-22.** Determine the components of reaction at  $C$ .



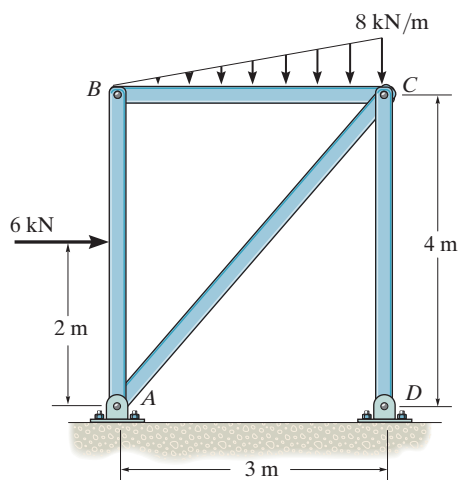
**Prob. F6-22**

**F6-23.** Determine the components of reaction at  $E$ .



**Prob. F6-23**

**F6-24.** Determine the components of reaction at  $D$  and the components of reaction the pin at  $A$  exerts on member  $BA$ .

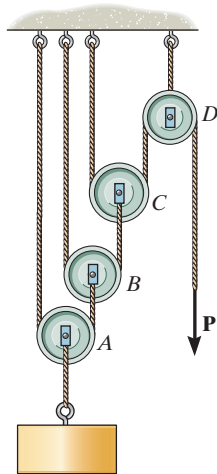


**Prob. F6-24**

## PROBLEMS

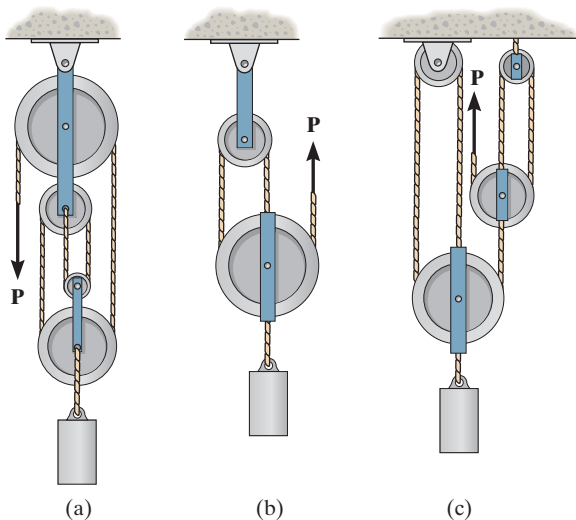
*All problem solutions must include FBDs.*

**6-61.** Determine the force  $\mathbf{P}$  required to hold the 100-lb weight in equilibrium.



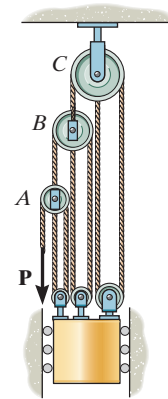
**Prob. 6-61**

**6-62.** In each case, determine the force  $\mathbf{P}$  required to maintain equilibrium. The block weighs 100 lb.



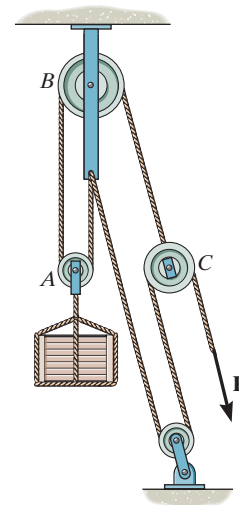
**Prob. 6-62**

**6-63.** Determine the force  $\mathbf{P}$  required to hold the 50-kg mass in equilibrium.



**Prob. 6-63**

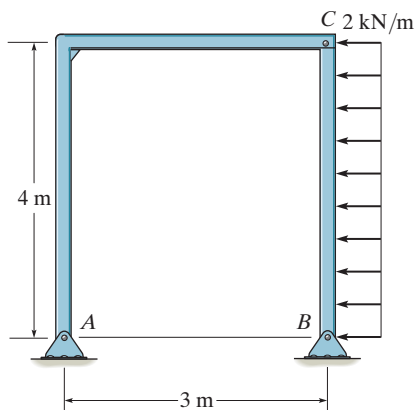
**\*6-64.** Determine the force  $\mathbf{P}$  required to hold the 150-kg crate in equilibrium.



**Prob. 6-64**

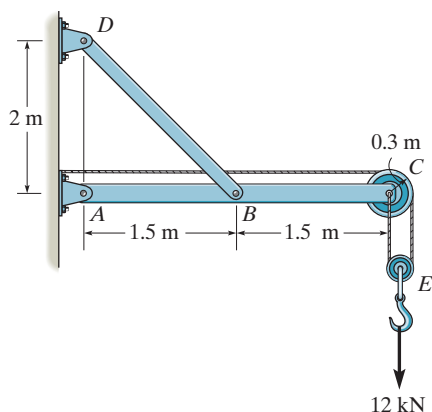


**6-65.** Determine the horizontal and vertical components of force that pins  $A$  and  $B$  exert on the frame.



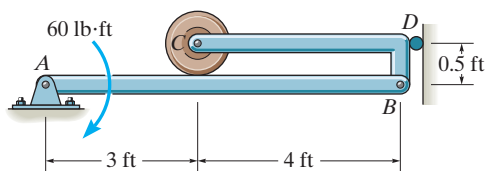
**Prob. 6-65**

**6-66.** Determine the horizontal and vertical components of force at pins  $A$  and  $D$ .



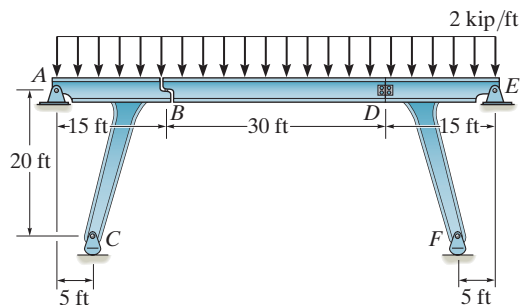
**Prob. 6-66**

**6-67.** Determine the force that the smooth roller  $C$  exerts on member  $AB$ . Also, what are the horizontal and vertical components of reaction at pin  $A$ ? Neglect the weight of the frame and roller.



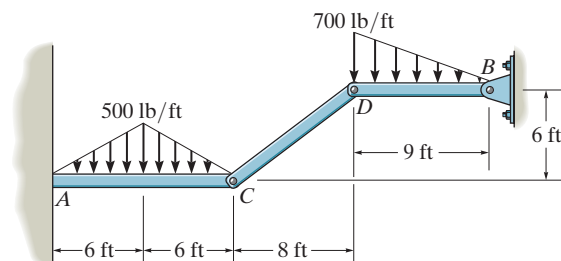
**Prob. 6-67**

**\*6-68.** The bridge frame consists of three segments which can be considered pinned at  $A$ ,  $D$ , and  $E$ , rocker supported at  $C$  and  $F$ , and roller supported at  $B$ . Determine the horizontal and vertical components of reaction at all these supports due to the loading shown.



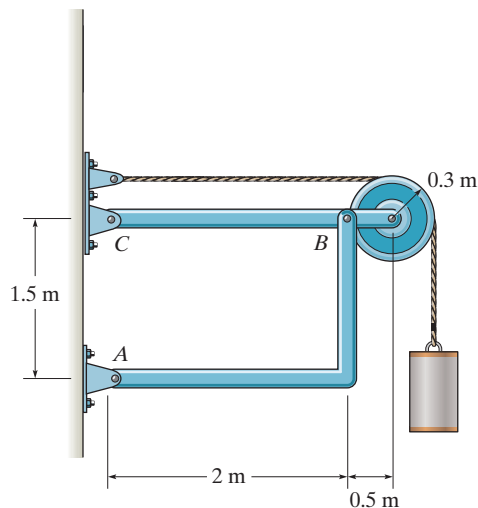
**Prob. 6-68**

**6-69.** Determine the reactions at supports  $A$  and  $B$ .



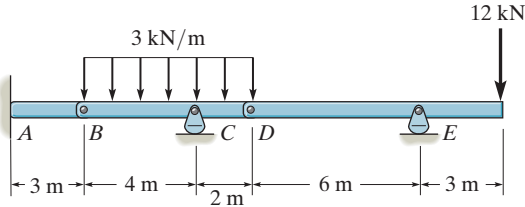
**Prob. 6-69**

**6-70.** Determine the horizontal and vertical components of force at pins  $B$  and  $C$ . The suspended cylinder has a mass of 75 kg.



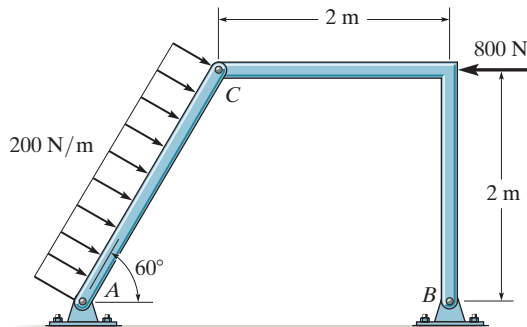
**Prob. 6-70**

**6-71.** Determine the reactions at the supports  $A$ ,  $C$ , and  $E$  of the compound beam.



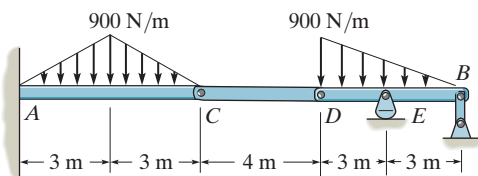
**Prob. 6-71**

**\*6-72.** Determine the resultant force at pins  $A$ ,  $B$ , and  $C$  on the three-member frame.



**Prob. 6-72**

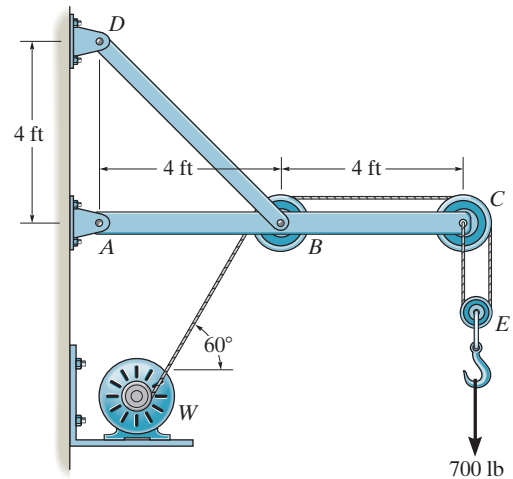
**6-73.** Determine the reactions at the supports at  $A$ ,  $E$ , and  $B$  of the compound beam.



**Prob. 6-73**

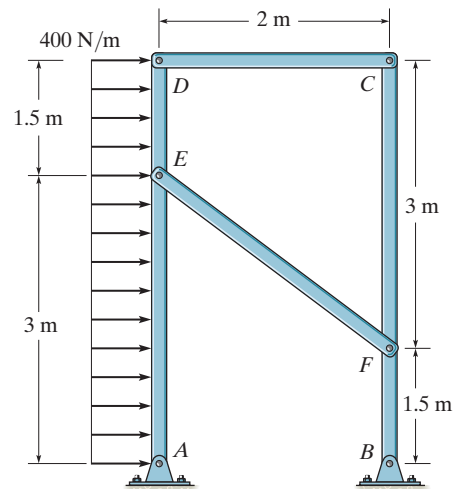
**6-74.** The wall crane supports a load of 700 lb. Determine the horizontal and vertical components of reaction at the pins  $A$  and  $D$ . Also, what is the force in the cable at the winch  $W$ ?

**6-75.** The wall crane supports a load of 700 lb. Determine the horizontal and vertical components of reaction at the pins  $A$  and  $D$ . Also, what is the force in the cable at the winch  $W$ ? The jib  $ABC$  has a weight of 100 lb and member  $BD$  has a weight of 40 lb. Each member is uniform and has a center of gravity at its center.



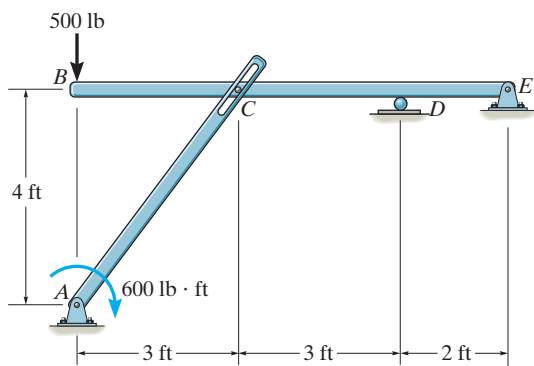
**Probs. 6-74/75**

**\*6-76.** Determine the horizontal and vertical components of force which the pins at  $A$  and  $B$  exert on the frame.



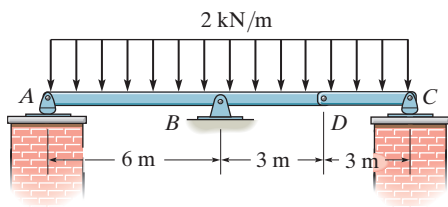
**Prob. 6-76**

**6-77.** The two-member structure is connected at  $C$  by a pin, which is fixed to  $BDE$  and passes through the smooth slot in member  $AC$ . Determine the horizontal and vertical components of reaction at the supports.



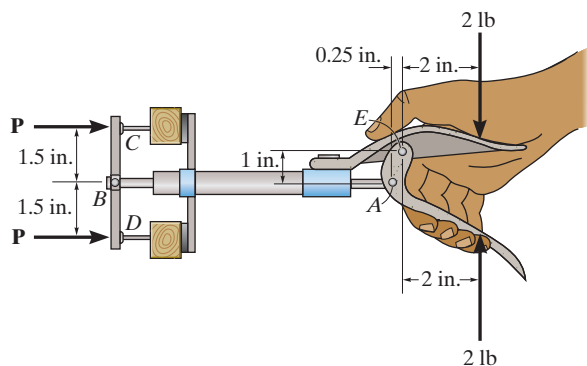
**Prob. 6-77**

**6-78.** The compound beam is pin supported at  $B$  and supported by rockers at  $A$  and  $C$ . There is a hinge (pin) at  $D$ . Determine the reactions at the supports.



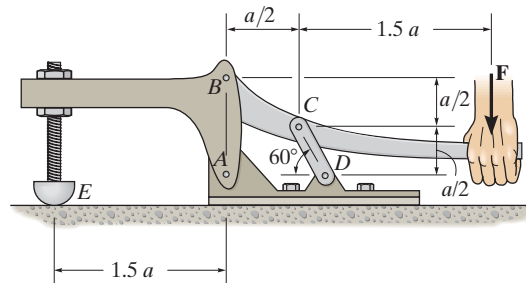
**Prob. 6-78**

**6-79.** When a force of 2 lb is applied to the handles of the brad squeezer, it pulls in the smooth rod  $AB$ . Determine the force  $P$  exerted on each of the smooth brads at  $C$  and  $D$ .



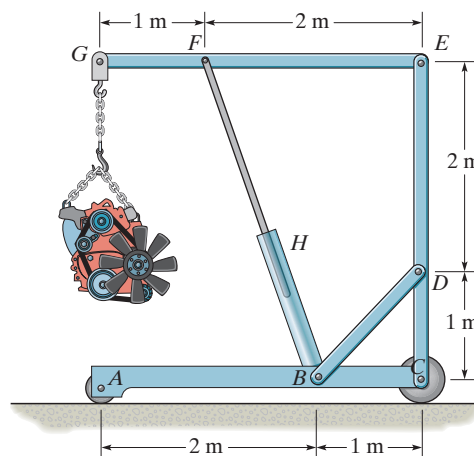
**Prob. 6-79**

**\*6-80.** The toggle clamp is subjected to a force  $F$  at the handle. Determine the vertical clamping force acting at  $E$ .



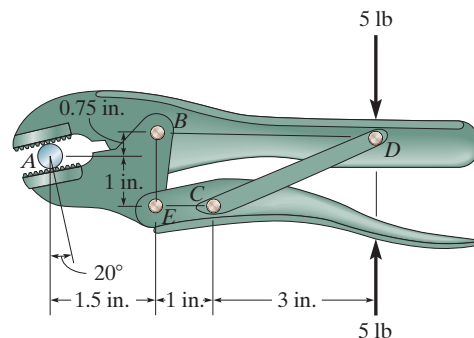
**Prob. 6-80**

**6-81.** The hoist supports the 125-kg engine. Determine the force the load creates in member  $DB$  and in member  $FB$ , which contains the hydraulic cylinder  $H$ .



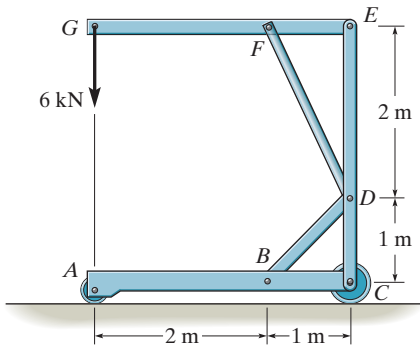
**Prob. 6-81**

**6-82.** A 5-lb force is applied to the handles of the vise grip. Determine the compressive force developed on the smooth bolt shank  $A$  at the jaws.



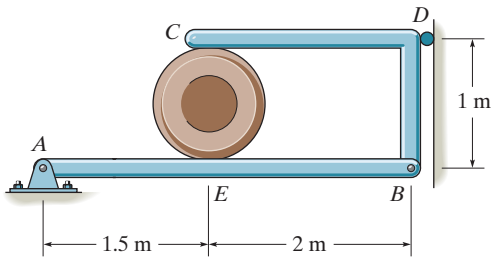
**Prob. 6-82**

**6-83.** Determine the force in members  $FD$  and  $DB$  of the frame. Also, find the horizontal and vertical components of reaction the pin at  $C$  exerts on member  $ABC$  and member  $EDC$ .



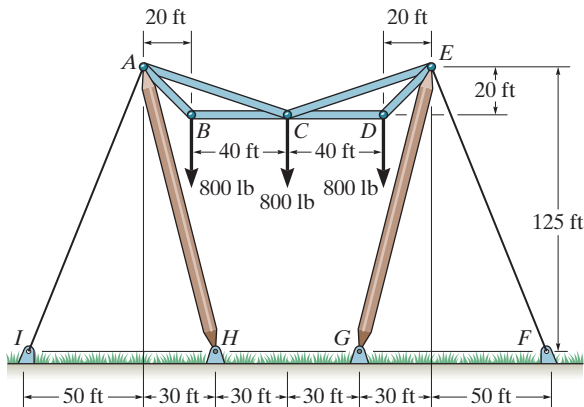
**Prob. 6-83**

**\*6-84.** Determine the force that the smooth 20-kg cylinder exerts on members  $AB$  and  $CDB$ . Also, what are the horizontal and vertical components of reaction at pin  $A$ ?



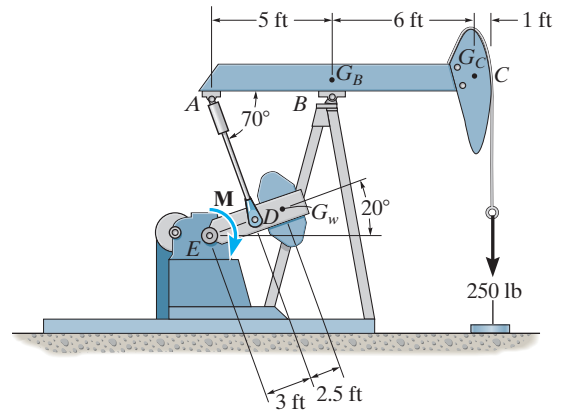
**Prob. 6-84**

**6-85.** The three power lines exert the forces shown on the pin-connected members at joints  $B$ ,  $C$ , and  $D$ , which in turn are pin connected to the poles  $AH$  and  $EG$ . Determine the force in the guy cable  $AI$  and the pin reaction at the support  $H$ .



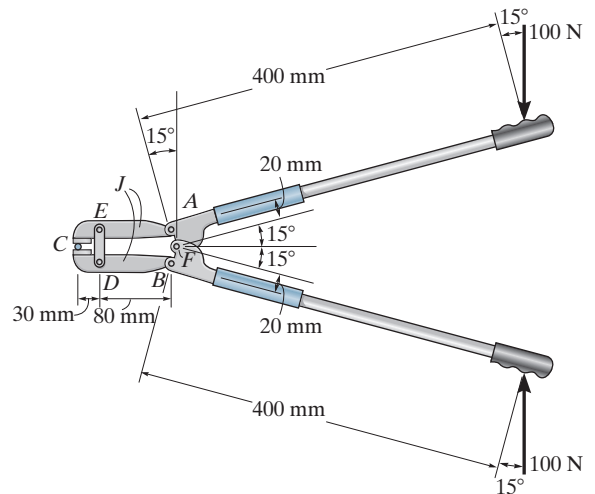
**Prob. 6-85**

**6-86.** The pumping unit is used to recover oil. When the walking beam  $ABC$  is horizontal, the force acting in the wireline at the well head is 250 lb. Determine the torque  $M$  which must be exerted by the motor in order to overcome this load. The horse-head  $C$  weighs 60 lb and has a center of gravity at  $G_C$ . The walking beam  $ABC$  has a weight of 130 lb and a center of gravity at  $G_B$ , and the counterweight has a weight of 200 lb and a center of gravity at  $G_W$ . The pitman,  $AD$ , is pin connected at its ends and has negligible weight.



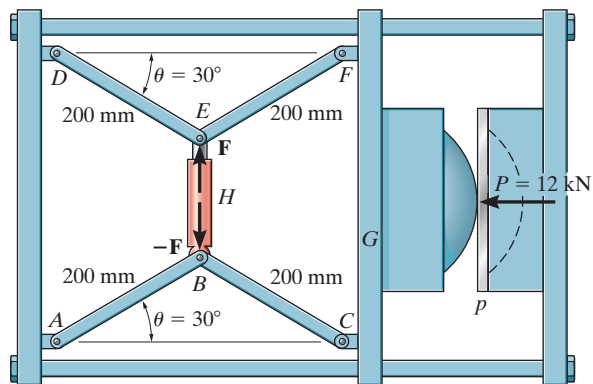
**Prob. 6-86**

**6-87.** Determine the force that the jaws  $J$  of the metal cutters exert on the smooth cable  $C$  if 100-N forces are applied to the handles. The jaws are pinned at  $E$  and  $A$ , and  $D$  and  $B$ . There is also a pin at  $F$ .



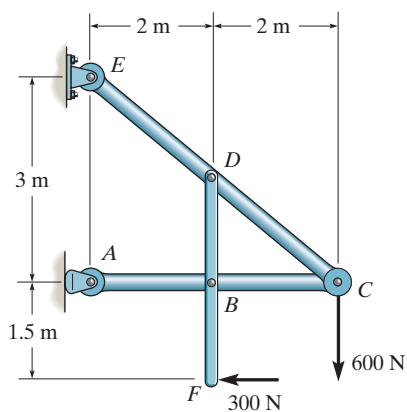
**Prob. 6-87**

**\*6-88.** The machine shown is used for forming metal plates. It consists of two toggles  $ABC$  and  $DEF$ , which are operated by the hydraulic cylinder  $H$ . The toggles push the movable bar  $G$  forward, pressing the plate  $p$  into the cavity. If the force which the plate exerts on the head is  $P = 12$  kN, determine the force  $F$  in the hydraulic cylinder when  $\theta = 30^\circ$ .



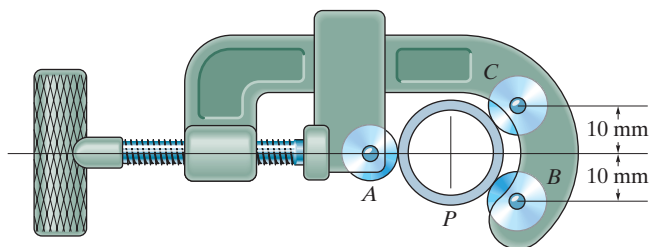
**Prob. 6-88**

**6-89.** Determine the horizontal and vertical components of force which pin  $C$  exerts on member  $ABC$ . The 600-N load is applied to the pin.



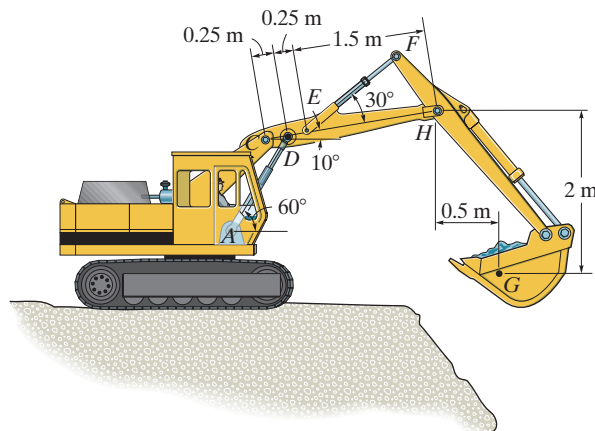
**Prob. 6-89**

**6-90.** The pipe cutter is clamped around the pipe  $P$ . If the wheel at  $A$  exerts a normal force of  $F_A = 80$  N on the pipe, determine the normal forces of wheels  $B$  and  $C$  on the pipe. Also compute the pin reaction on the wheel at  $C$ . The three wheels each have a radius of 7 mm and the pipe has an outer radius of 10 mm.



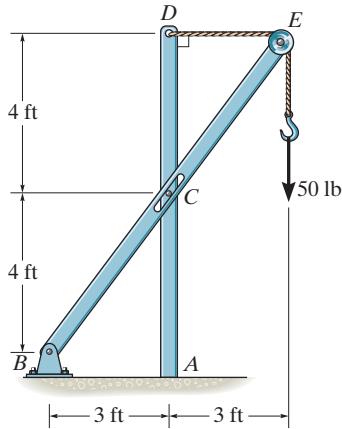
**Prob. 6-90**

**6-91.** Determine the force created in the hydraulic cylinders  $EF$  and  $AD$  in order to hold the shovel in equilibrium. The shovel load has a mass of 1.25 Mg and a center of gravity at  $G$ . All joints are pin connected.



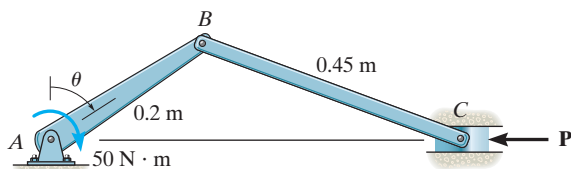
**Prob. 6-91**

**\*6-92.** Determine the horizontal and vertical components of force at pin  $B$  and the normal force the pin at  $C$  exerts on the smooth slot. Also, determine the moment and horizontal and vertical reactions of force at  $A$ . There is a pulley at  $E$ .



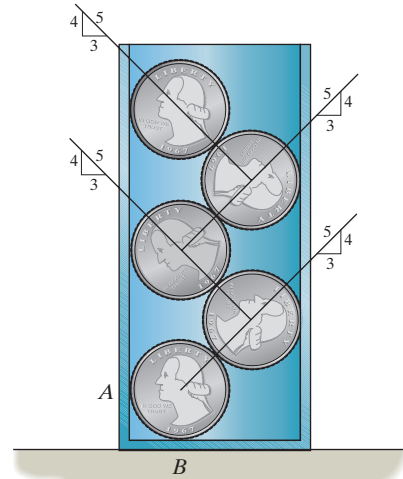
**Prob. 6-92**

**6-93.** The constant moment of  $50 \text{ N} \cdot \text{m}$  is applied to the crank shaft. Determine the compressive force  $P$  that is exerted on the piston for equilibrium as a function of  $\theta$ . Plot the results of  $P$  (vertical axis) versus  $\theta$  (horizontal axis) for  $0^\circ \leq \theta \leq 90^\circ$ .



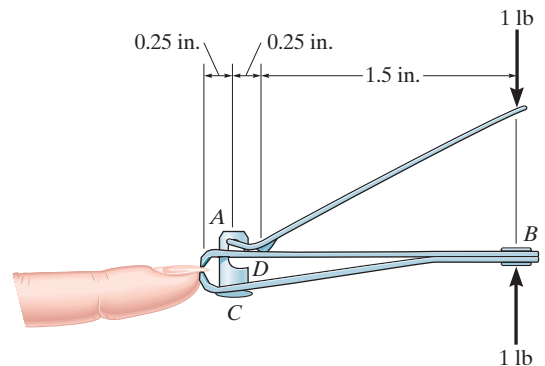
**Prob. 6-93**

**6-94.** Five coins are stacked in the smooth plastic container shown. If each coin weighs  $0.0235 \text{ lb}$ , determine the normal reactions of the bottom coin on the container at points  $A$  and  $B$ .



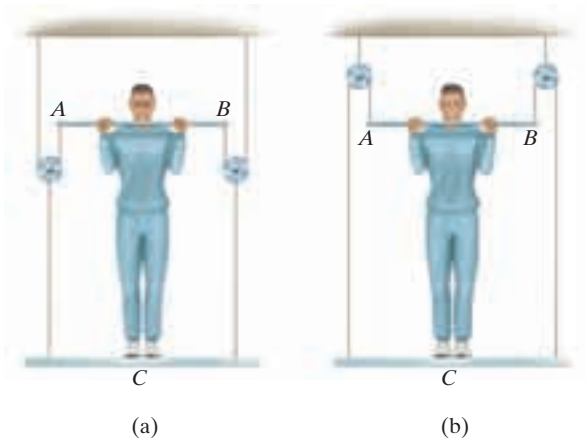
**Prob. 6-94**

**6-95.** The nail cutter consists of the handle and the two cutting blades. Assuming the blades are pin connected at  $B$  and the surface at  $D$  is smooth, determine the normal force on the fingernail when a force of  $1 \text{ lb}$  is applied to the handles as shown. The pin  $AC$  slides through a smooth hole at  $A$  and is attached to the bottom member at  $C$ .



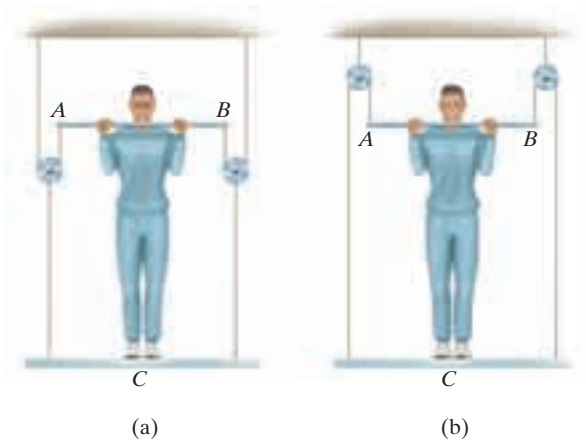
**Prob. 6-95**

**\*6-96.** A man having a weight of 175 lb attempts to hold himself using one of the two methods shown. Determine the total force he must exert on bar  $AB$  in each case and the normal reaction he exerts on the platform at  $C$ . Neglect the weight of the platform.



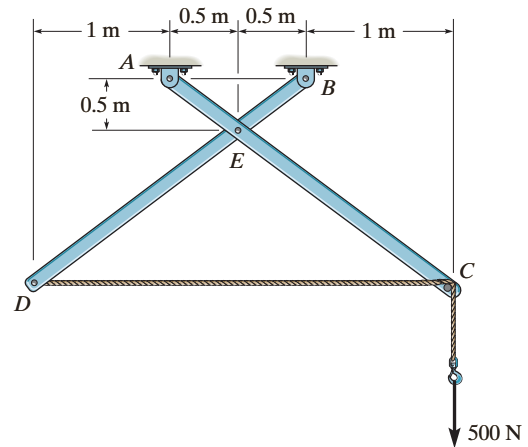
**Prob. 6-96**

**6-97.** A man having a weight of 175 lb attempts to hold himself using one of the two methods shown. Determine the total force he must exert on bar  $AB$  in each case and the normal reaction he exerts on the platform at  $C$ . The platform has a weight of 30 lb.



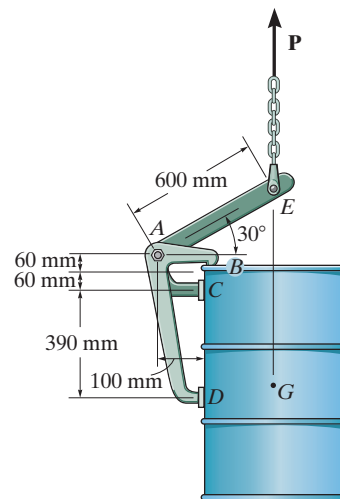
**Prob. 6-97**

**6-98.** The two-member frame is pin connected at  $E$ . The cable is attached to  $D$ , passes over the smooth peg at  $C$ , and supports the 500-N load. Determine the horizontal and vertical reactions at each pin.



**Prob. 6-98**

**6-99.** If the 300-kg drum has a center of mass at point  $G$ , determine the horizontal and vertical components of force acting at pin  $A$  and the reactions on the smooth pads  $C$  and  $D$ . The grip at  $B$  on member  $DAB$  resists both horizontal and vertical components of force at the rim of the drum.

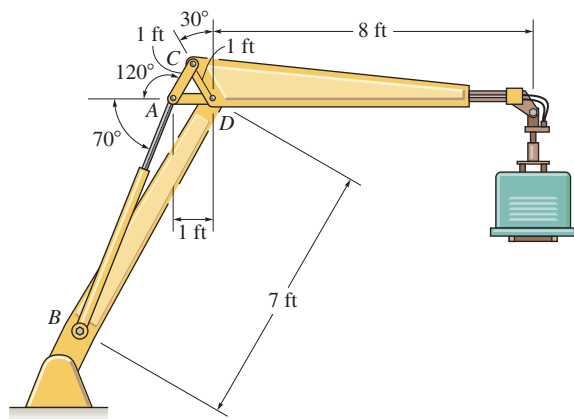


**Prob. 6-99**



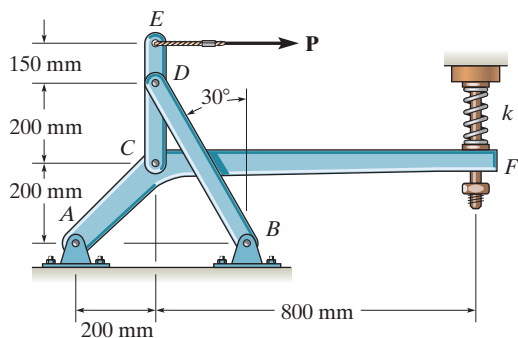


**\*6-104.** The hydraulic crane is used to lift the 1400-lb load. Determine the force in the hydraulic cylinder  $AB$  and the force in links  $AC$  and  $AD$  when the load is held in the position shown.



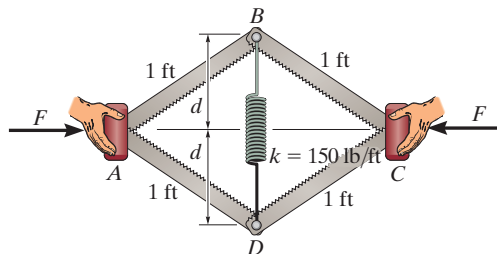
**Prob. 6-104**

**6-105.** Determine force  $P$  on the cable if the spring is compressed 0.025 m when the mechanism is in the position shown. The spring has a stiffness of  $k = 6 \text{ kN/m}$ .



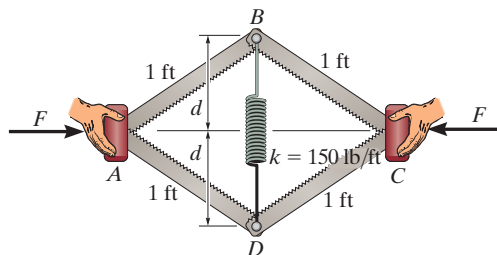
**Prob. 6-105**

**6-106.** If  $d = 0.75 \text{ ft}$  and the spring has an unstretched length of 1 ft, determine the force  $F$  required for equilibrium.



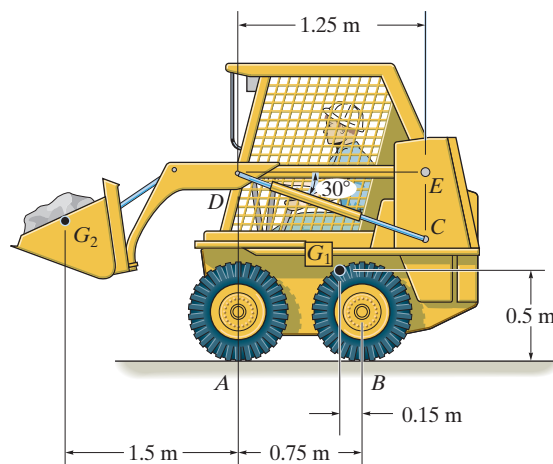
**Prob. 6-106**

**6-107.** If a force of  $F = 50 \text{ lb}$  is applied to the pads at  $A$  and  $C$ , determine the smallest dimension  $d$  required for equilibrium if the spring has an unstretched length of 1 ft.



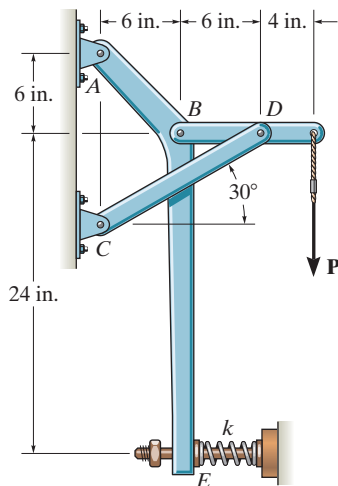
**Prob. 6-107**

**\*6-108.** The skid-steer loader has a mass of 1.18 Mg, and in the position shown the center of mass is at  $G_1$ . If there is a 300-kg stone in the bucket, with center of mass at  $G_2$ , determine the reactions of each pair of wheels  $A$  and  $B$  on the ground and the force in the hydraulic cylinder  $CD$  and at the pin  $E$ . There is a similar linkage on each side of the loader.



**Prob. 6-108**

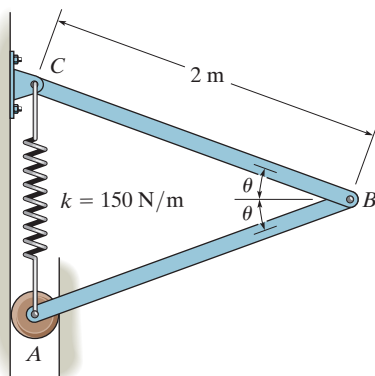
**6-109.** Determine the force  $\mathbf{P}$  on the cable if the spring is compressed 0.5 in. when the mechanism is in the position shown. The spring has a stiffness of  $k = 800 \text{ lb/ft}$ .



**Prob. 6-109**

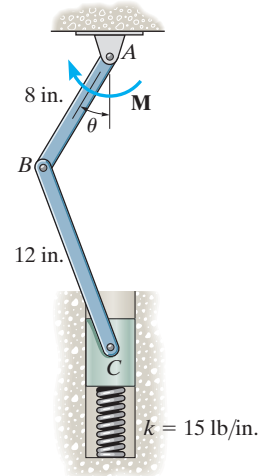
**6-110.** The spring has an unstretched length of 0.3 m. Determine the angle  $\theta$  for equilibrium if the uniform bars each have a mass of 20 kg.

**6-111.** The spring has an unstretched length of 0.3 m. Determine the mass  $m$  of each uniform bar if each angle  $\theta = 30^\circ$  for equilibrium.



**Probs. 6-110/111**

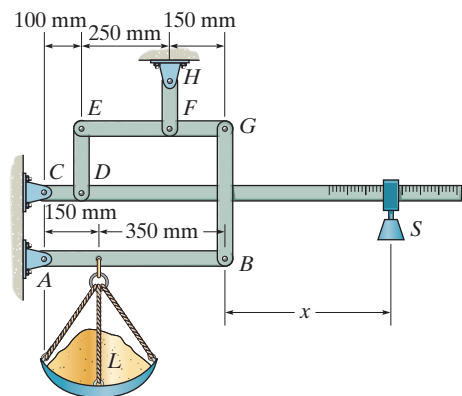
**\*6-112.** The piston  $C$  moves vertically between the two smooth walls. If the spring has a stiffness of  $k = 15 \text{ lb/in.}$ , and is unstretched when  $\theta = 0^\circ$ , determine the couple  $\mathbf{M}$  that must be applied to  $AB$  to hold the mechanism in equilibrium when  $\theta = 30^\circ$ .



**Prob. 6-112**

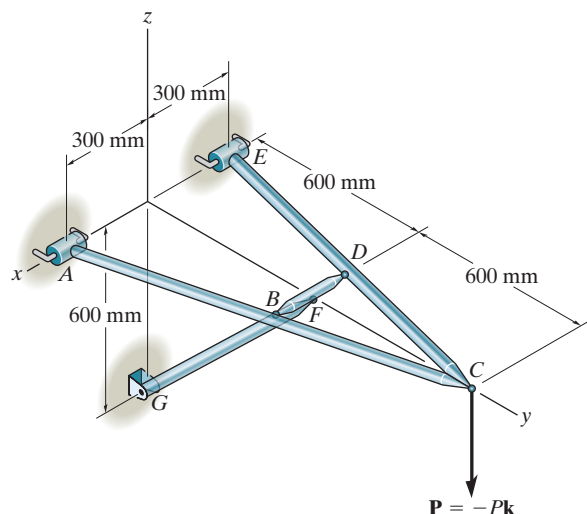
**6-113.** The platform scale consists of a combination of third and first class levers so that the load on one lever becomes the effort that moves the next lever. Through this arrangement, a small weight can balance a massive object. If  $x = 450 \text{ mm}$ , determine the required mass of the counterweight  $S$  required to balance a 90-kg load,  $L$ .

**6-114.** The platform scale consists of a combination of third and first class levers so that the load on one lever becomes the effort that moves the next lever. Through this arrangement, a small weight can balance a massive object. If  $x = 450 \text{ mm}$ , and the mass of the counterweight  $S$  is 2 kg, determine the mass of the load  $L$  required to maintain the balance.



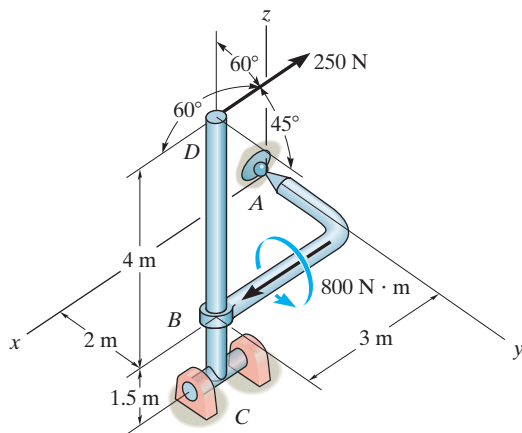
**Probs. 6-113/114**

**6-115.** The four-member “A” frame is supported at  $A$  and  $E$  by smooth collars and at  $G$  by a pin. All the other joints are ball-and-sockets. If the pin at  $G$  will fail when the resultant force there is 800 N, determine the largest vertical force  $P$  that can be supported by the frame. Also, what are the  $x, y, z$  force components which member  $BD$  exerts on members  $EDC$  and  $ABC$ ? The collars at  $A$  and  $E$  and the pin at  $G$  only exert force components on the frame.



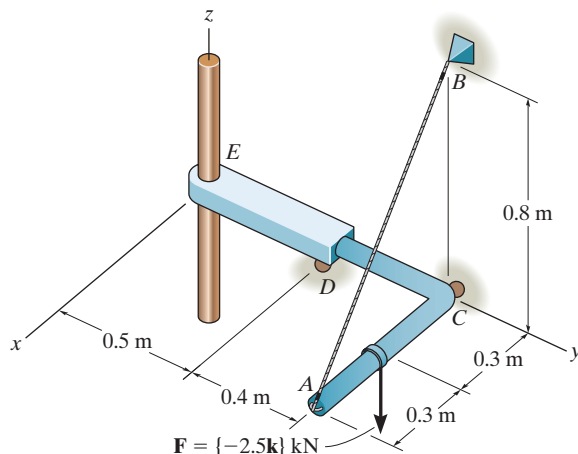
**Prob. 6-115**

**\*6-116.** The structure is subjected to the loadings shown. Member  $AB$  is supported by a ball-and-socket at  $A$  and smooth collar at  $B$ . Member  $CD$  is supported by a pin at  $C$ . Determine the  $x, y, z$  components of reaction at  $A$  and  $C$ .



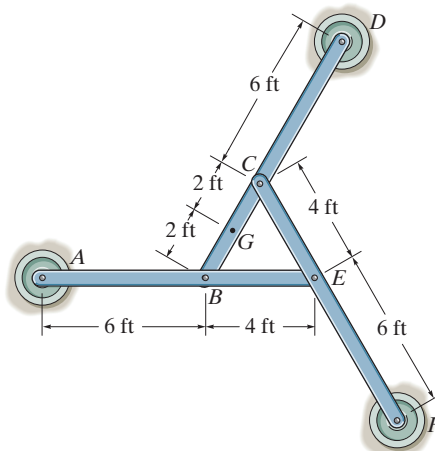
**Prob. 6-116**

**6-117.** The structure is subjected to the loading shown. Member  $AD$  is supported by a cable  $AB$  and roller at  $C$  and fits through a smooth circular hole at  $D$ . Member  $ED$  is supported by a roller at  $D$  and a pole that fits in a smooth snug circular hole at  $E$ . Determine the  $x, y, z$  components of reaction at  $E$  and the tension in cable  $AB$ .



**Prob. 6-117**

**6-118.** The three pin-connected members shown in the top view support a downward force of 60 lb at  $G$ . If only vertical forces are supported at the connections  $B, C, E$  and pad supports  $A, D, F$ , determine the reactions at each pad.

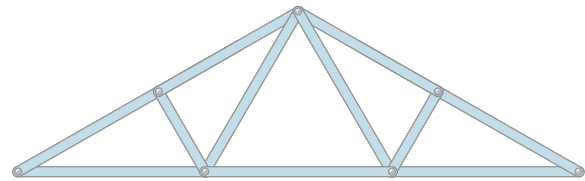


**Prob. 6-118**

## CHAPTER REVIEW

### Simple Truss

A simple truss consists of triangular elements connected together by pinned joints. The forces within its members can be determined by assuming the members are all two-force members, connected concurrently at each joint. The members are either in tension or compression, or carry no force.



Roof truss

### Method of Joints

The method of joints states that if a truss is in equilibrium, then each of its joints is also in equilibrium. For a plane truss, the concurrent force system at each joint must satisfy force equilibrium.

To obtain a numerical solution for the forces in the members, select a joint that has a free-body diagram with at most two unknown forces and one known force. (This may require first finding the reactions at the supports.)

Once a member force is determined, use its value and apply it to an adjacent joint.

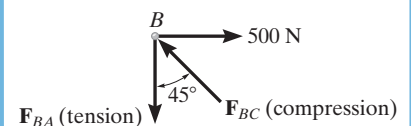
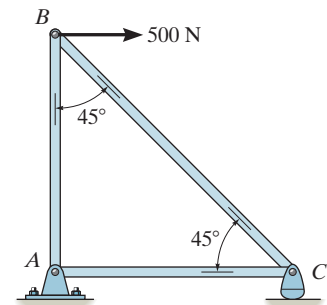
Remember that forces that are found to *pull* on the joint are *tensile forces*, and those that *push* on the joint are *compressive forces*.

To avoid a simultaneous solution of two equations, set one of the coordinate axes along the line of action of one of the unknown forces and sum forces perpendicular to this axis. This will allow a direct solution for the other unknown.

The analysis can also be simplified by first identifying all the zero-force members.

$$\Sigma F_x = 0$$

$$\Sigma F_y = 0$$



### Method of Sections

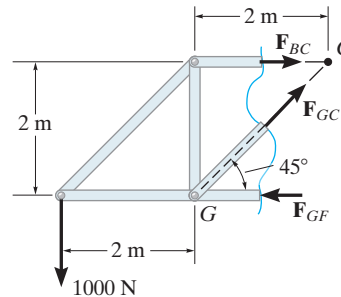
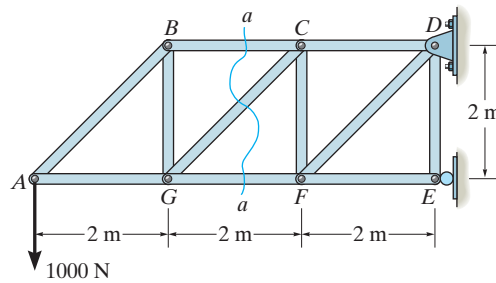
The method of sections states that if a truss is in equilibrium, then each segment of the truss is also in equilibrium. Pass a section through the truss and the member whose force is to be determined. Then draw the free-body diagram of the sectioned part having the least number of forces on it.

Sectioned members subjected to *pulling* are in *tension*, and those that are subjected to *pushing* are in *compression*.

Three equations of equilibrium are available to determine the unknowns.

If possible, sum forces in a direction that is perpendicular to two of the three unknown forces. This will yield a direct solution for the third force.

Sum moments about the point where the lines of action of two of the three unknown forces intersect, so that the third unknown force can be determined directly.



$$\Sigma F_x = 0$$

$$\Sigma F_y = 0$$

$$\Sigma M_O = 0$$

$$+\uparrow \Sigma F_y = 0$$

$$-1000 \text{ N} + F_{GC} \sin 45^\circ = 0$$

$$F_{GC} = 1.41 \text{ kN (T)}$$

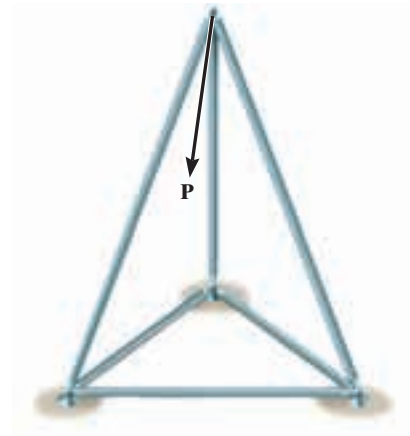
$$\zeta + \Sigma M_C = 0$$

$$1000 \text{ N}(4 \text{ m}) - F_{GF}(2 \text{ m}) = 0$$

$$F_{GF} = 2 \text{ kN (C)}$$

### Space Truss

A space truss is a three-dimensional truss built from tetrahedral elements, and is analyzed using the same methods as for plane trusses. The joints are assumed to be ball-and-socket connections.

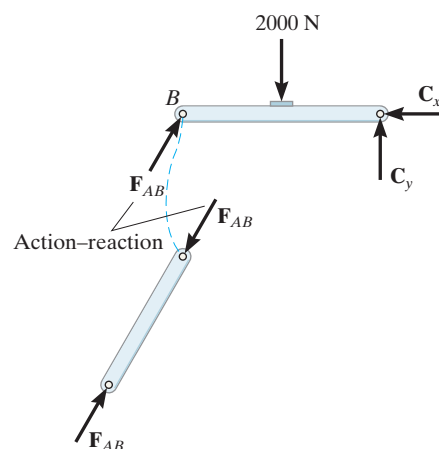
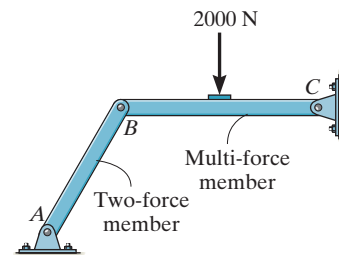


### Frames and Machines

Frames and machines are structures that contain one or more multiforce members, that is, members with three or more forces or couples acting on them. Frames are designed to support loads, and machines transmit and alter the effect of forces.

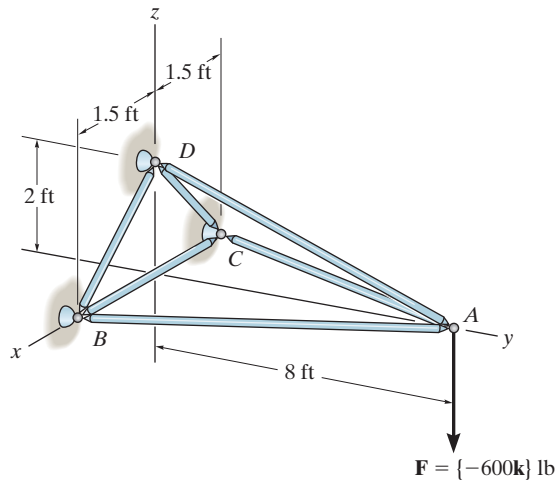
The forces acting at the joints of a frame or machine can be determined by drawing the free-body diagrams of each of its members or parts. The principle of action–reaction should be carefully observed when indicating these forces on the free-body diagram of each adjacent member or pin. For a coplanar force system, there are three equilibrium equations available for each member.

To simplify the analysis, be sure to recognize all two-force members. They have equal but opposite collinear forces at their ends.



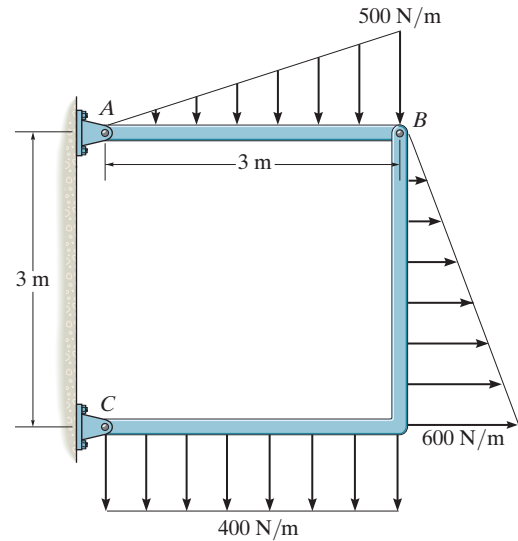


**R6-5.** Determine the force in members  $AB$ ,  $AD$ , and  $AC$  of the space truss and state if the members are in tension or compression.



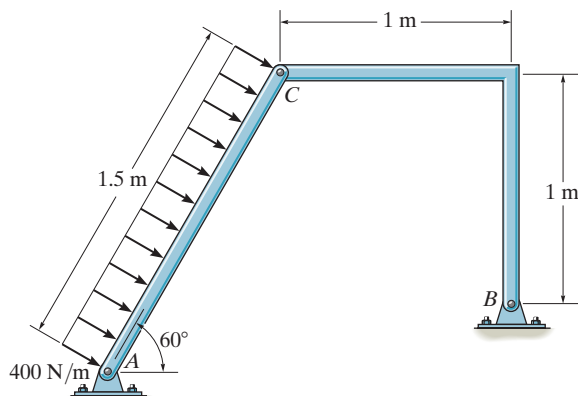
**Prob. R6-5**

**R6-7.** Determine the horizontal and vertical components of force at pins  $A$  and  $C$  of the two-member frame.



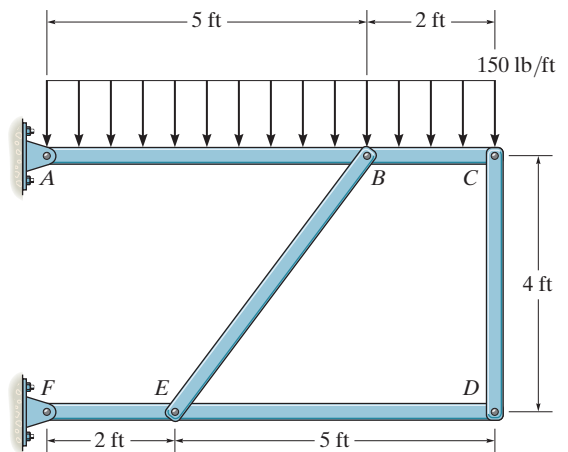
**Prob. R6-7**

**R6-6.** Determine the horizontal and vertical components of force that the pins  $A$  and  $B$  exert on the two-member frame.



**Prob. R6-6**

**R6-8.** Determine the resultant forces at pins  $B$  and  $C$  on member  $ABC$  of the four-member frame.



**Prob. R6-8**



# Chapter 7



(© Tony Freeman/Science Source)

When external loads are placed upon these beams and columns, the loads within them must be determined if they are to be properly designed. In this chapter we will study how to determine these internal loadings.

# Internal Forces

## CHAPTER OBJECTIVES

- To use the method of sections to determine the internal loadings in a member at a specific point.
- To show how to obtain the internal shear and moment throughout a member and express the result graphically in the form of shear and moment diagrams.
- To analyze the forces and the shape of cables supporting various types of loadings.

## 7.1 Internal Loadings Developed in Structural Members

To design a structural or mechanical member it is necessary to know the loading acting within the member in order to be sure the material can resist this loading. Internal loadings can be determined by using the **method of sections**. To illustrate this method, consider the cantilever beam in Fig. 7-1*a*. If the internal loadings acting on the cross section at point *B* are to be determined, we must pass an imaginary section *a-a* perpendicular to the axis of the beam through point *B* and then separate the beam into two segments. The internal loadings acting at *B* will then be exposed and become *external* on the free-body diagram of each segment, Fig. 7-1*b*.

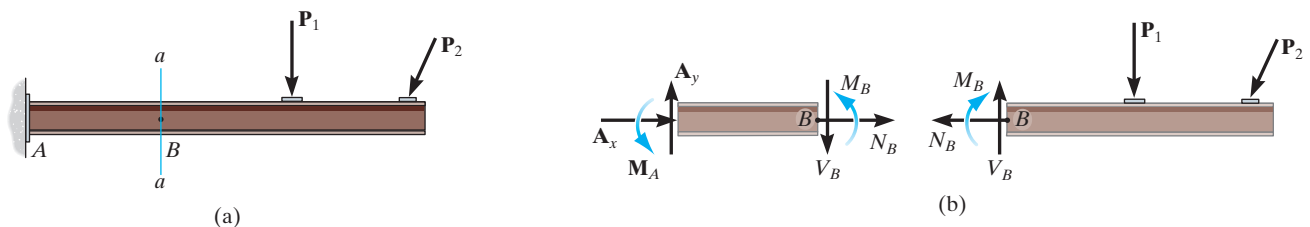


Fig. 7-1



In each case, the link on the backhoe is a two-force member. In the top photo it is subjected to both bending and an axial load at its center. It is more efficient to make the member straight, as in the bottom photo; then only an axial force acts within the member. (© Russell C. Hibbeler)

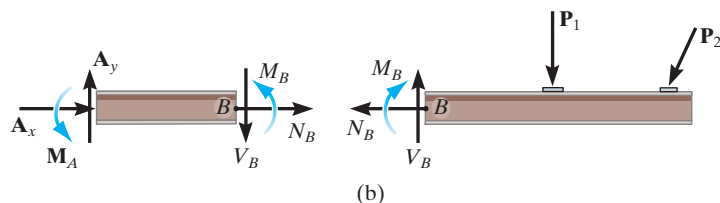


Fig. 7-1 (Repeated)

The force component  $N_B$  that acts *perpendicular* to the cross section is termed the **normal force**. The force component  $V_B$  that is tangent to the cross section is called the **shear force**, and the couple moment  $M_B$  is referred to as the **bending moment**. The force components prevent the relative translation between the two segments, and the couple moment prevents the relative rotation. According to Newton's third law, these loadings must act in opposite directions on each segment, as shown in Fig. 7-1b. They can be determined by applying the equations of equilibrium to the free-body diagram of either segment. In this case, however, the right segment is the better choice since it does not involve the unknown support reactions at A. A direct solution for  $N_B$  is obtained by applying  $\Sigma F_x = 0$ ,  $V_B$  is obtained from  $\Sigma F_y = 0$ , and  $M_B$  can be obtained by applying  $\Sigma M_B = 0$ , since the moments of  $N_B$  and  $V_B$  about B are zero.

In two dimensions, we have shown that three internal loading resultants exist, Fig. 7-2a; however in three dimensions, a general resultant internal force and couple moment resultant will act at the section. The  $x$ ,  $y$ ,  $z$  components of these loadings are shown in Fig. 7-2b. Here  $N_y$  is the **normal force**, and  $V_x$  and  $V_z$  are **shear force components**.  $M_y$  is a **torsional or twisting moment**, and  $M_x$  and  $M_z$  are **bending moment components**. For most applications, these *resultant loadings* will act at the geometric center or centroid (C) of the section's cross-sectional area. Although the magnitude for each loading generally will be different at various points along the axis of the member, the method of sections can always be used to determine their values.

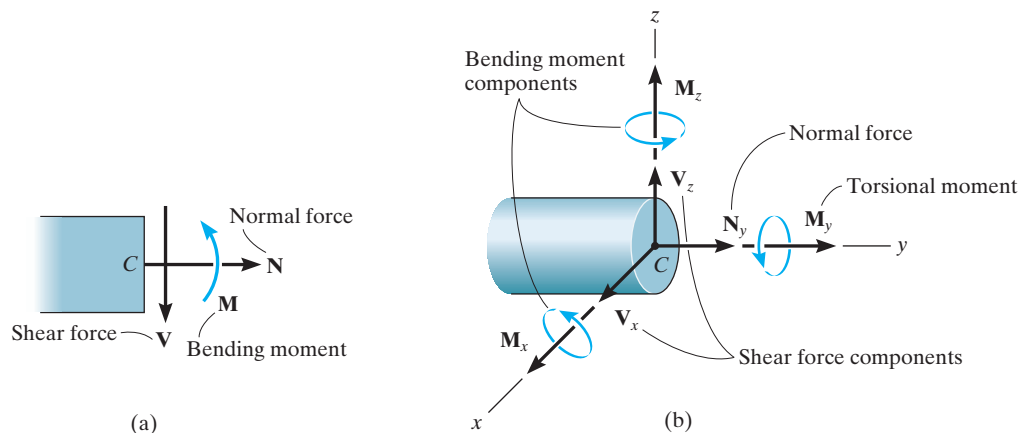


Fig. 7-2

**Sign Convention.** For problems in two dimensions engineers generally use a sign convention to report the three internal loadings  $N$ ,  $V$ , and  $M$ . Although this sign convention can be arbitrarily assigned, the one that is widely accepted will be used here, Fig. 7–3. The normal force is said to be positive if it creates *tension*, a positive shear force will cause the beam segment on which it acts to rotate clockwise, and a positive bending moment will tend to bend the segment on which it acts in a concave upward manner. Loadings that are opposite to these are considered negative.

### Important Point

- There can be four types of resultant internal loads in a member. They are the normal and shear forces and the bending and torsional moments. These loadings generally vary from point to point. They can be determined using the method of sections.

### Procedure for Analysis

The method of sections can be used to determine the internal loadings on the cross section of a member using the following procedure.

#### Support Reactions.

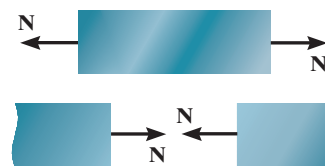
- Before the member is sectioned, it may first be necessary to determine its support reactions.

#### Free-Body Diagram.

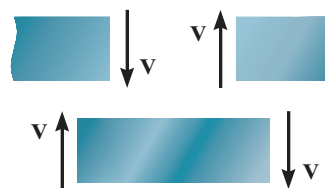
- It is important to *keep* all distributed loadings, couple moments, and forces acting on the member in their *exact locations*, then pass an imaginary section through the member, perpendicular to its axis at the point where the internal loadings are to be determined.
- After the section is made, draw a free-body diagram of the segment that has the least number of loads on it, and indicate the components of the internal force and couple moment resultants at the cross section acting in their positive directions in accordance with the established sign convention.

#### Equations of Equilibrium.

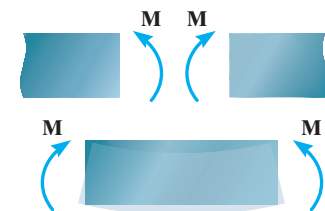
- Moments should be summed at the section. This way the normal and shear forces at the section are eliminated, and we can obtain a direct solution for the moment.
- If the solution of the equilibrium equations yields a negative scalar, the sense of the quantity is opposite to that shown on the free-body diagram.



Positive normal force



Positive shear



Positive moment

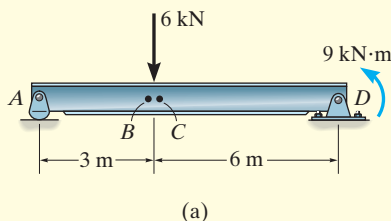
Fig. 7–3



The designer of this shop crane realized the need for additional reinforcement around the joint at  $A$  in order to prevent severe internal bending of the joint when a large load is suspended from the chain hoist. (© Russell C. Hibbeler)

## EXAMPLE 7.1

Determine the normal force, shear force, and bending moment acting just to the left, point  $B$ , and just to the right, point  $C$ , of the 6-kN force on the beam in Fig. 7-4a.



(a)

## SOLUTION

**Support Reactions.** The free-body diagram of the beam is shown in Fig. 7-4b. When determining the *external reactions*, realize that the 9-kN·m couple moment is a free vector and therefore it can be placed *anywhere* on the free-body diagram of the entire beam. Here we will only determine  $A_y$ , since the left segments will be used for the analysis.

$$\zeta + \Sigma M_D = 0; \quad 9 \text{ kN} \cdot \text{m} + (6 \text{ kN})(6 \text{ m}) - A_y(9 \text{ m}) = 0$$

$$A_y = 5 \text{ kN}$$

**Free-Body Diagrams.** The free-body diagrams of the left segments  $AB$  and  $AC$  of the beam are shown in Figs. 7-4c and 7-4d. In this case the 9-kN·m couple moment is *not included* on these diagrams since it must be kept in its *original position* until *after* the section is made and the appropriate segment is isolated.

## Equations of Equilibrium.

Segment  $AB$

$$\pm \Sigma F_x = 0; \quad N_B = 0 \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad 5 \text{ kN} - V_B = 0 \quad V_B = 5 \text{ kN} \quad \text{Ans.}$$

$$\zeta + \Sigma M_B = 0; \quad -(5 \text{ kN})(3 \text{ m}) + M_B = 0 \quad M_B = 15 \text{ kN} \cdot \text{m} \quad \text{Ans.}$$

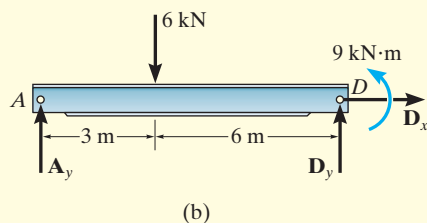
Segment  $AC$

$$\pm \Sigma F_x = 0; \quad N_C = 0 \quad \text{Ans.}$$

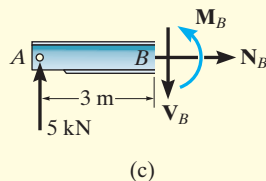
$$+\uparrow \Sigma F_y = 0; \quad 5 \text{ kN} - 6 \text{ kN} - V_C = 0 \quad V_C = -1 \text{ kN} \quad \text{Ans.}$$

$$\zeta + \Sigma M_C = 0; \quad -(5 \text{ kN})(3 \text{ m}) + M_C = 0 \quad M_C = 15 \text{ kN} \cdot \text{m} \quad \text{Ans.}$$

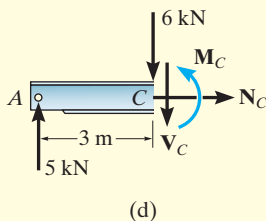
**NOTE:** The negative sign indicates that  $V_C$  acts in the opposite sense to that shown on the free-body diagram. Also, the moment arm for the 5-kN force in both cases is approximately 3 m since  $B$  and  $C$  are “almost” coincident.



(b)



(c)

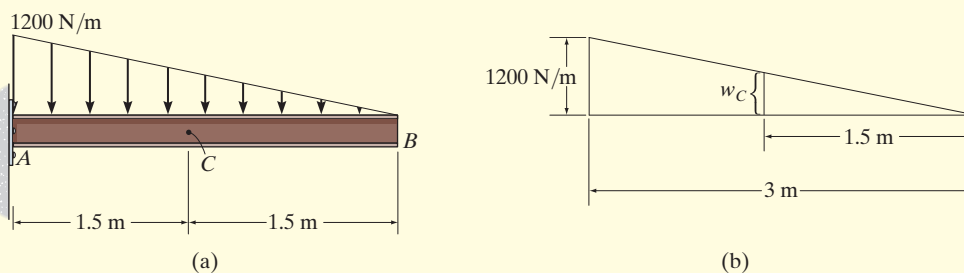


(d)

Fig. 7-4

**EXAMPLE 7.2**

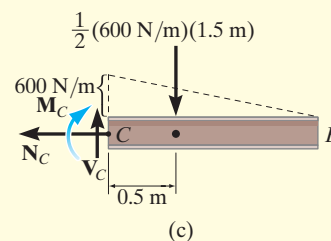
Determine the normal force, shear force, and bending moment at  $C$  of the beam in Fig. 7-5a.

**Fig. 7-5****SOLUTION**

**Free-Body Diagram.** It is not necessary to find the support reactions at  $A$  since segment  $BC$  of the beam can be used to determine the internal loadings at  $C$ . The intensity of the triangular distributed load at  $C$  is determined using similar triangles from the geometry shown in Fig. 7-5b, i.e.,

$$w_C = (1200 \text{ N/m}) \left( \frac{1.5 \text{ m}}{3 \text{ m}} \right) = 600 \text{ N/m}$$

The distributed load acting on segment  $BC$  can now be replaced by its resultant force, and its location is indicated on the free-body diagram, Fig. 7-5c.

**Equations of Equilibrium.**

$$\rightarrow \Sigma F_x = 0; \quad N_C = 0 \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad V_C - \frac{1}{2}(600 \text{ N/m})(1.5 \text{ m}) = 0$$

$$V_C = 450 \text{ N} \quad \text{Ans.}$$

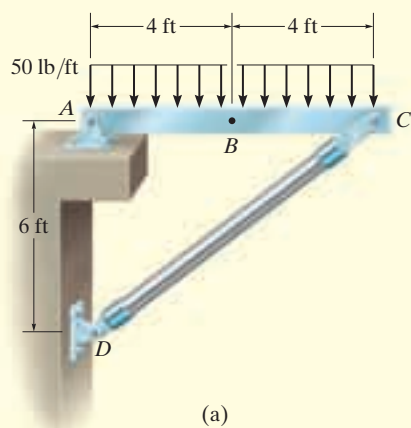
$$\curvearrowleft + \Sigma M_C = 0; \quad -M_C - \frac{1}{2}(600 \text{ N/m})(1.5 \text{ m})(0.5 \text{ m}) = 0$$

$$M_C = -225 \text{ N} \quad \text{Ans.}$$

The negative sign indicates that  $M_C$  acts in the opposite sense to that shown on the free-body diagram.



## EXAMPLE 7.3

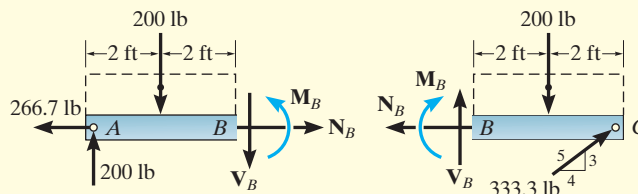
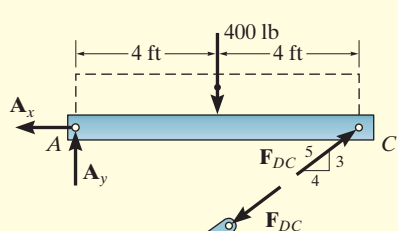


Determine the normal force, shear force, and bending moment acting at point  $B$  of the two-member frame shown in Fig. 7-6a.

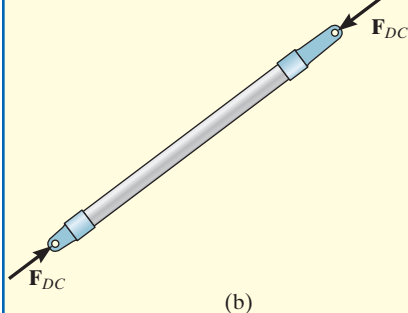
## SOLUTION

**Support Reactions.** A free-body diagram of each member is shown in Fig. 7-6b. Since  $CD$  is a two-force member, the equations of equilibrium need to be applied only to member  $AC$ .

$$\begin{aligned} \zeta + \Sigma M_A &= 0; & -400 \text{ lb} (4 \text{ ft}) + \left(\frac{3}{5}\right) F_{DC} (8 \text{ ft}) &= 0 & F_{DC} &= 333.3 \text{ lb} \\ \pm \Sigma F_x &= 0; & -A_x + \left(\frac{4}{5}\right) (333.3 \text{ lb}) &= 0 & A_x &= 266.7 \text{ lb} \\ + \uparrow \Sigma F_y &= 0; & A_y - 400 \text{ lb} + \left(\frac{3}{5}\right) (333.3 \text{ lb}) &= 0 & A_y &= 200 \text{ lb} \end{aligned}$$



(c)



(b)

Fig. 7-6

**Free-Body Diagrams.** Passing an imaginary section perpendicular to the axis of member  $AC$  through point  $B$  yields the free-body diagrams of segments  $AB$  and  $BC$  shown in Fig. 7-6c. When constructing these diagrams it is important to keep the distributed loading where it is until *after the section is made*. Only then can it be replaced by a single resultant force.

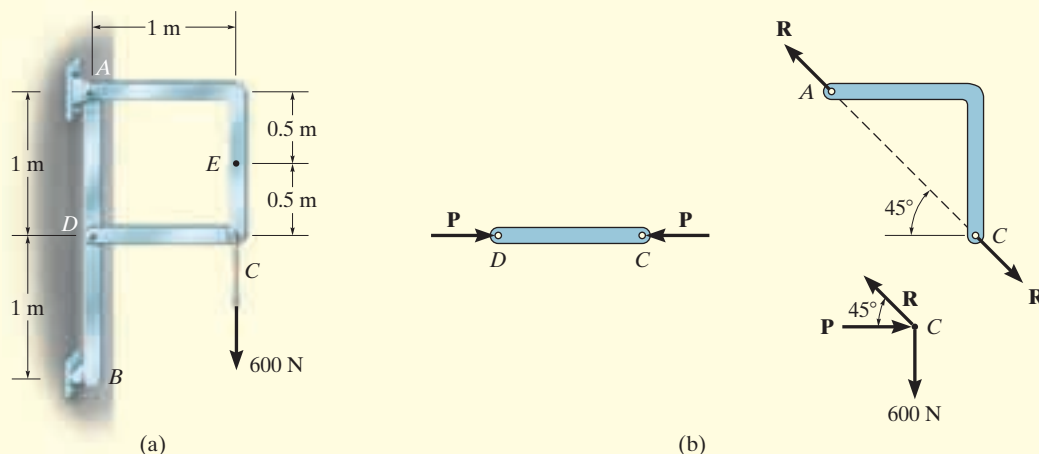
**Equations of Equilibrium.** Applying the equations of equilibrium to segment  $AB$ , we have

$$\begin{aligned} \pm \Sigma F_x &= 0; & N_B - 266.7 \text{ lb} &= 0 & N_B &= 267 \text{ lb} & \text{Ans.} \\ + \uparrow \Sigma F_y &= 0; & 200 \text{ lb} - 200 \text{ lb} - V_B &= 0 & V_B &= 0 & \text{Ans.} \\ \zeta + \Sigma M_B &= 0; & M_B - 200 \text{ lb} (4 \text{ ft}) + 200 \text{ lb} (2 \text{ ft}) &= 0 & M_B &= 400 \text{ lb} \cdot \text{ft} & \text{Ans.} \end{aligned}$$

**NOTE:** As an exercise, try to obtain these same results using segment  $BC$ .

**EXAMPLE 7.4**

Determine the normal force, shear force, and bending moment acting at point  $E$  of the frame loaded as shown in Fig. 7-7a.

**SOLUTION**

**Support Reactions.** By inspection, members  $AC$  and  $CD$  are two-force members, Fig. 7-7b. In order to determine the internal loadings at  $E$ , we must first determine the force  $\mathbf{R}$  acting at the end of member  $AC$ . To obtain it, we will analyze the equilibrium of the pin at  $C$ .

Summing forces in the vertical direction on the pin, Fig. 7-7b, we have

$$+\uparrow \Sigma F_y = 0; \quad R \sin 45^\circ - 600 \text{ N} = 0 \quad R = 848.5 \text{ N}$$

**Free-Body Diagram.** The free-body diagram of segment  $CE$  is shown in Fig. 7-7c.

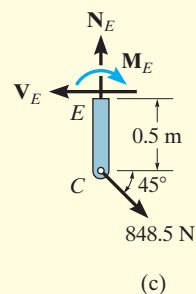
**Equations of Equilibrium.**

$$\pm \Sigma F_x = 0; \quad 848.5 \cos 45^\circ \text{ N} - V_E = 0 \quad V_E = 600 \text{ N} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad -848.5 \sin 45^\circ \text{ N} + N_E = 0 \quad N_E = 600 \text{ N} \quad \text{Ans.}$$

$$\zeta + \Sigma M_E = 0; \quad 848.5 \cos 45^\circ \text{ N}(0.5 \text{ m}) - M_E = 0$$

$$M_E = 300 \text{ N} \cdot \text{m} \quad \text{Ans.}$$

**Fig. 7-7**

**NOTE:** These results indicate a poor design. Member  $AC$  should be *straight* (from  $A$  to  $C$ ) so that bending within the member is *eliminated*. If  $AC$  were straight then the internal force would only create tension in the member.



## EXAMPLE 7.5



© Russell C. Hibbeler

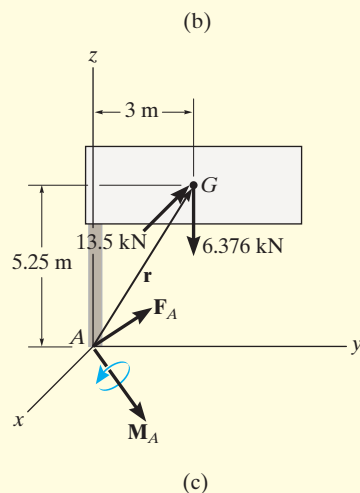
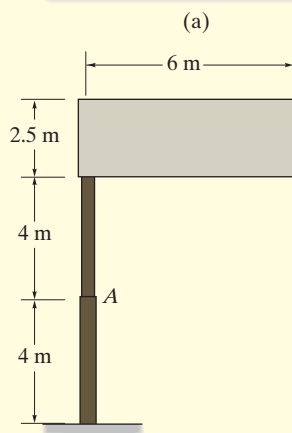


Fig. 7-8

The uniform sign shown in Fig. 7-8a has a mass of 650 kg and is supported on the fixed column. Design codes indicate that the expected maximum uniform wind loading that will occur in the area where it is located is 900 Pa. Determine the internal loadings at A.

## SOLUTION

The idealized model for the sign is shown in Fig. 7-8b. Here the necessary dimensions are indicated. We can consider the free-body diagram of a section above point A since it does not involve the support reactions.

**Free-Body Diagram.** The sign has a weight of  $W = 650(9.81) \text{ N} = 6.376 \text{ kN}$ , and the wind creates a resultant force of  $F_w = 900 \text{ N/m}^2(6 \text{ m})(2.5 \text{ m}) = 13.5 \text{ kN}$ , which acts perpendicular to the face of the sign. These loadings are shown on the free-body diagram, Fig. 7-8c.

**Equations of Equilibrium.** Since the problem is three dimensional, a vector analysis will be used.

$$\Sigma \mathbf{F} = \mathbf{0}; \quad \mathbf{F}_A - 13.5\mathbf{i} - 6.376\mathbf{k} = \mathbf{0}$$

$$\mathbf{F}_A = \{13.5\mathbf{i} + 6.38\mathbf{k}\} \text{ kN} \quad \text{Ans.}$$

$$\Sigma \mathbf{M}_A = \mathbf{0}; \quad \mathbf{M}_A + \mathbf{r} \times (\mathbf{F}_w + \mathbf{W}) = \mathbf{0}$$

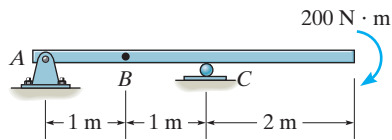
$$\mathbf{M}_A + \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 0 & 3 & 5.25 \\ -13.5 & 0 & -6.376 \end{vmatrix} = \mathbf{0}$$

$$\mathbf{M}_A = \{19.1\mathbf{i} + 70.9\mathbf{j} - 40.5\mathbf{k}\} \text{ kN} \cdot \text{m} \quad \text{Ans.}$$

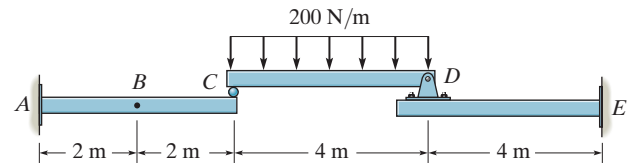
**NOTE:** Here  $\mathbf{F}_{A_z} = \{6.38\mathbf{k}\} \text{ kN}$  represents the normal force, whereas  $\mathbf{F}_{A_x} = \{13.5\mathbf{i}\} \text{ kN}$  is the shear force. Also, the torsional moment is  $\mathbf{M}_{A_z} = \{-40.5\mathbf{k}\} \text{ kN} \cdot \text{m}$ , and the bending moment is determined from its components  $\mathbf{M}_{A_x} = \{19.1\mathbf{i}\} \text{ kN} \cdot \text{m}$  and  $\mathbf{M}_{A_y} = \{70.9\mathbf{j}\} \text{ kN} \cdot \text{m}$ ; i.e.,  $(M_b)_A = \sqrt{(M_{A_x})^2 + (M_{A_y})^2} = 73.4 \text{ kN} \cdot \text{m}$ .

## PRELIMINARY PROBLEMS

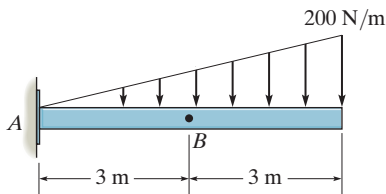
**P7-1.** In each case, calculate the reaction at  $A$  and then draw the free-body diagram of segment  $AB$  of the beam in order to determine the internal loading at  $B$ .



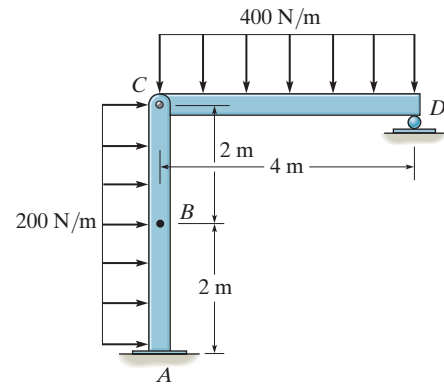
(a)



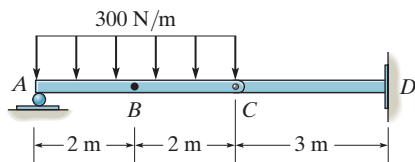
(d)



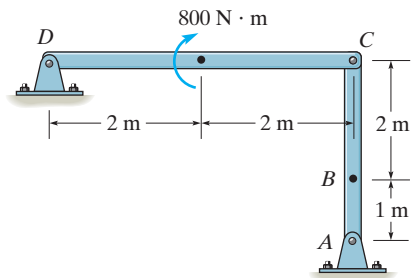
(b)



(e)



(c)



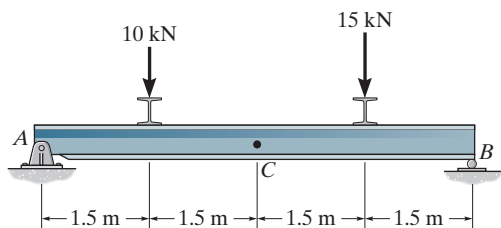
(f)

**Prob. P7-1**

# FUNDAMENTAL PROBLEMS

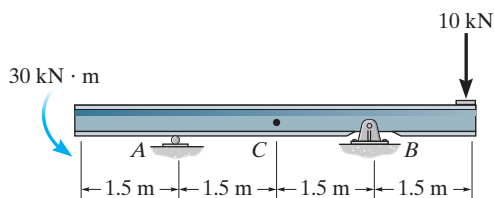
*All problem solutions must include FBDs.*

**F7-1.** Determine the normal force, shear force, and moment at point  $C$ .



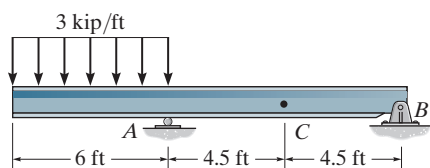
**Prob. F7-1**

**F7-2.** Determine the normal force, shear force, and moment at point  $C$ .



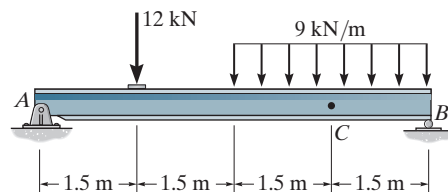
**Prob. F7-2**

**F7-3.** Determine the normal force, shear force, and moment at point  $C$ .



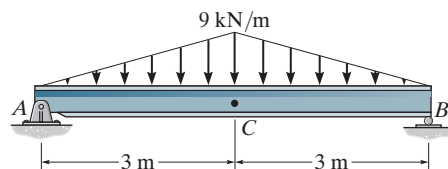
**Prob. F7-3**

**F7-4.** Determine the normal force, shear force, and moment at point  $C$ .



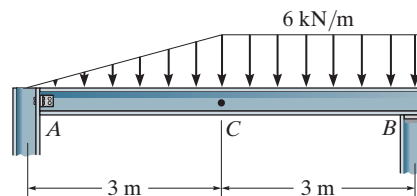
**Prob. F7-4**

**F7-5.** Determine the normal force, shear force, and moment at point  $C$ .



**Prob. F7-5**

**F7-6.** Determine the normal force, shear force, and moment at point  $C$ . Assume  $A$  is pinned and  $B$  is a roller.

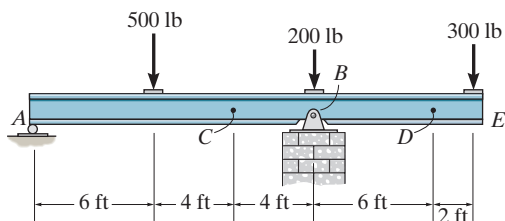


**Prob. F7-6**

## PROBLEMS

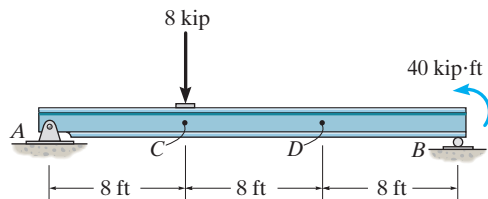
**All problem solutions must include FBDs.**

**7-1.** Determine the shear force and moment at points  $C$  and  $D$ .



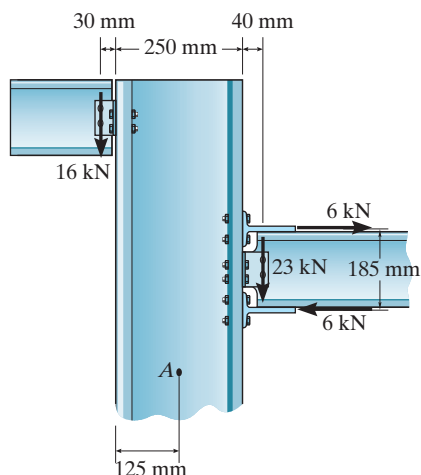
**Prob. 7-1**

**7-2.** Determine the internal normal force and shear force, and the bending moment in the beam at points  $C$  and  $D$ . Assume the support at  $B$  is a roller. Point  $C$  is located just to the right of the 8-kip load.



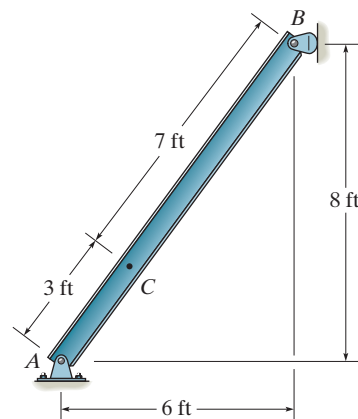
**Prob. 7-2**

**7-3.** Two beams are attached to the column such that structural connections transmit the loads shown. Determine the internal normal force, shear force, and moment acting in the column at a section passing horizontally through point  $A$ .



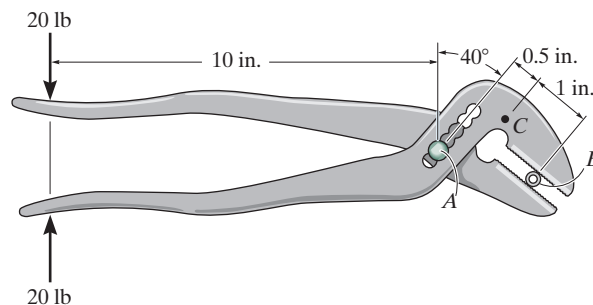
**Prob. 7-3**

**\*7-4.** The beam weighs 280 lb/ft. Determine the internal normal force, shear force, and moment at point  $C$ .



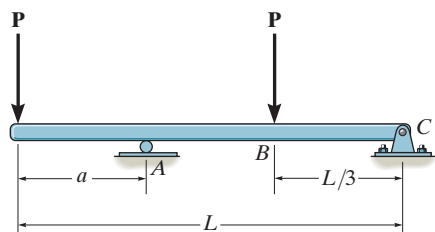
**Prob. 7-4**

**7-5.** The pliers are used to grip the tube at  $B$ . If a force of 20 lb is applied to the handles, determine the internal shear force and moment at point  $C$ . Assume the jaws of the pliers exert only normal forces on the tube.



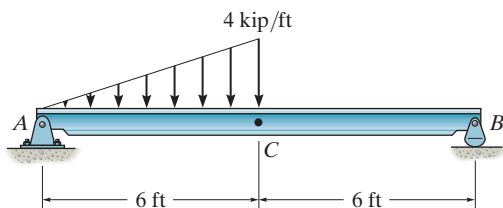
**Prob. 7-5**

**7-6.** Determine the distance  $a$  as a fraction of the beam's length  $L$  for locating the roller support so that the moment in the beam at  $B$  is zero.



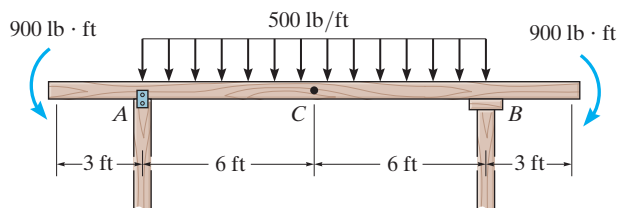
**Prob. 7-6**

**7-7.** Determine the internal shear force and moment acting at point  $C$  in the beam.



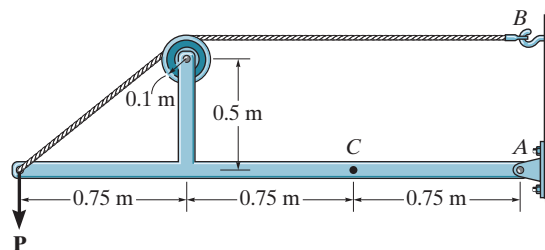
**Prob. 7-7**

**\*7-8.** Determine the internal shear force and moment acting at point  $C$  in the beam.



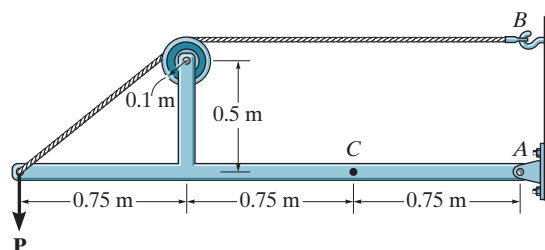
**Prob. 7-8**

**7-9.** Determine the normal force, shear force, and moment at a section passing through point  $C$ . Take  $P = 8$  kN.



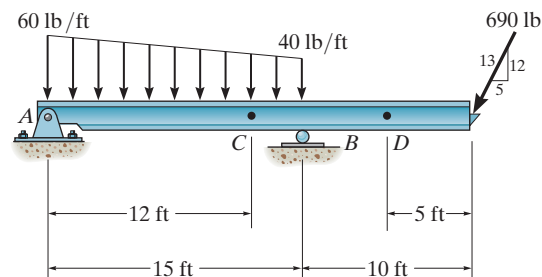
**Prob. 7-9**

**7-10.** The cable will fail when subjected to a tension of 2 kN. Determine the largest vertical load  $P$  the frame will support and calculate the internal normal force, shear force, and moment at a section passing through point  $C$  for this loading.



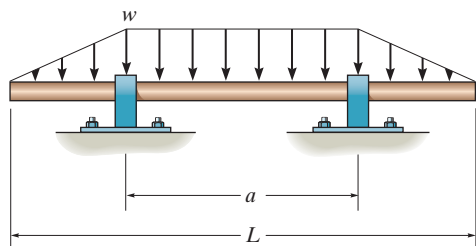
**Prob. 7-10**

**7-11.** Determine the internal normal force, shear force, and moment at points  $C$  and  $D$  of the beam.



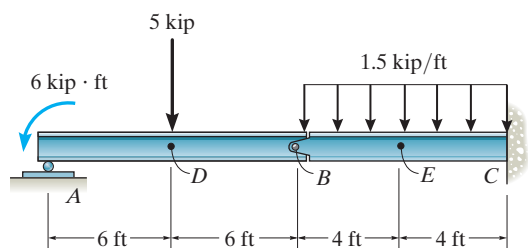
**Prob. 7-11**

**\*7-12.** Determine the distance  $a$  between the bearings in terms of the shaft's length  $L$  so that the moment in the *symmetric* shaft is zero at its center.



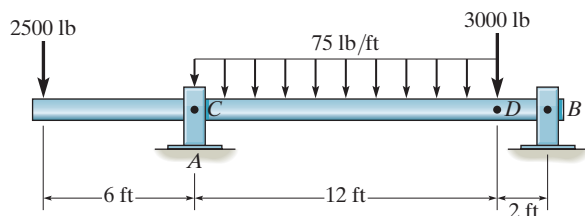
**Prob. 7-12**

**7-13.** Determine the internal normal force, shear force, and moment in the beam at sections passing through points  $D$  and  $E$ . Point  $D$  is located just to the left of the 5-kip load.



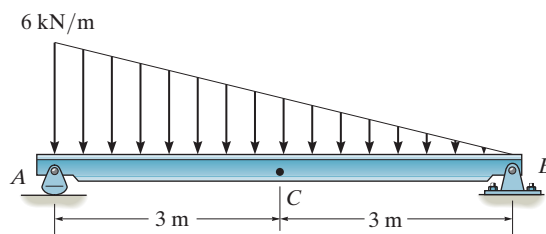
**Prob. 7-13**

**7-14.** The shaft is supported by a journal bearing at  $A$  and a thrust bearing at  $B$ . Determine the normal force, shear force, and moment at a section passing through (a) point  $C$ , which is just to the right of the bearing at  $A$ , and (b) point  $D$ , which is just to the left of the 3000-lb force.



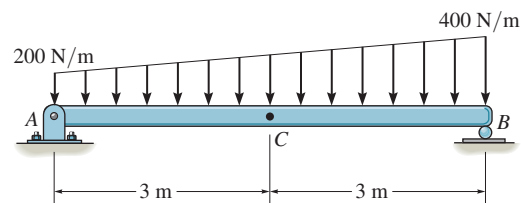
**Prob. 7-14**

**7-15.** Determine the internal normal force, shear force, and moment at point  $C$ .



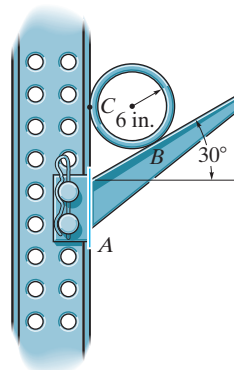
**Prob. 7-15**

**\*7-16.** Determine the internal normal force, shear force, and moment at point  $C$  of the beam.



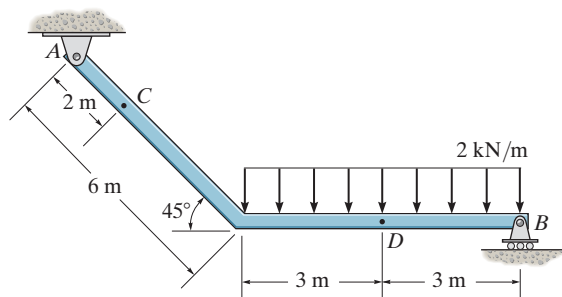
**Prob. 7-16**

**7-17.** The cantilevered rack is used to support each end of a smooth pipe that has a total weight of 300 lb. Determine the normal force, shear force, and moment that act in the arm at its fixed support  $A$  along a vertical section.



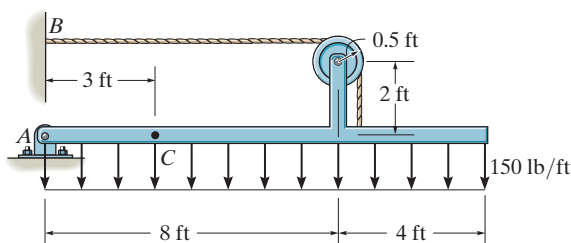
**Prob. 7-17**

**7-18.** Determine the internal normal force, shear force, and the moment at points  $C$  and  $D$ .



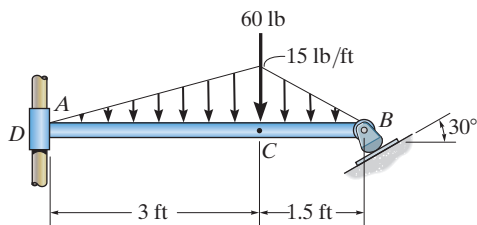
**Prob. 7-18**

**7-19.** Determine the internal normal force, shear force, and moment at point  $C$ .



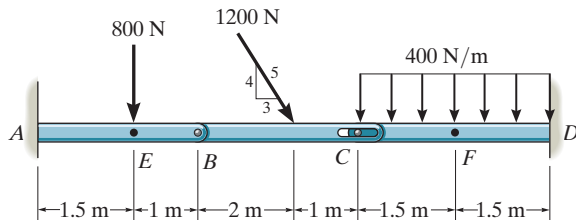
**Prob. 7-19**

**\*7-20.** Rod  $AB$  is fixed to a smooth collar  $D$ , which slides freely along the vertical guide. Determine the internal normal force, shear force, and moment at point  $C$ , which is located just to the left of the 60-lb concentrated load.



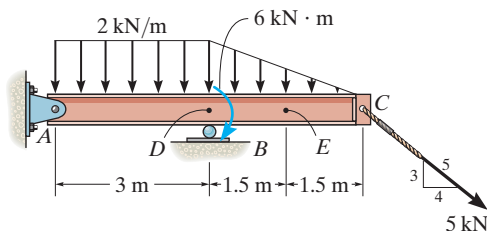
**Prob. 7-20**

**7-21.** Determine the internal normal force, shear force, and moment at points  $E$  and  $F$  of the compound beam. Point  $E$  is located just to the left of 800 N force.



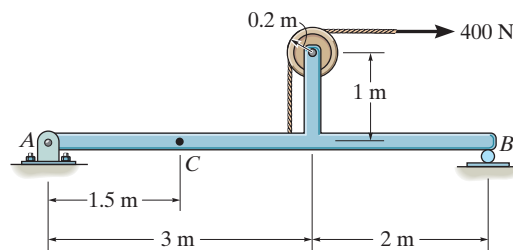
**Prob. 7-21**

**7-22.** Determine the internal normal force, shear force, and moment at points  $D$  and  $E$  in the overhang beam. Point  $D$  is located just to the left of the roller support at  $B$ , where the couple moment acts.



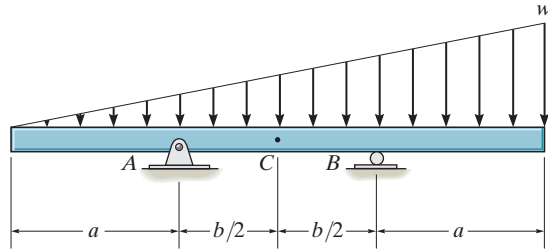
**Prob. 7-22**

**7-23.** Determine the internal normal force, shear force, and moment at point  $C$ .



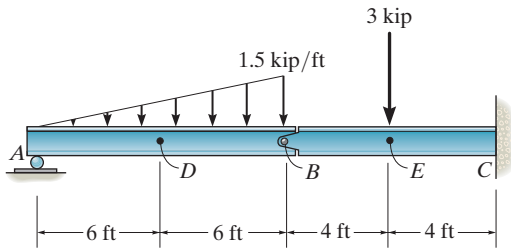
**Prob. 7-23**

**\*7-24.** Determine the ratio of  $a/b$  for which the shear force will be zero at the midpoint  $C$  of the beam.



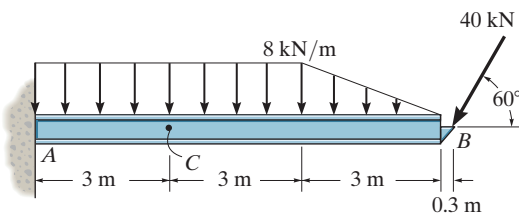
**Prob. 7-24**

**7-25.** Determine the normal force, shear force, and moment in the beam at sections passing through points  $D$  and  $E$ . Point  $E$  is just to the right of the 3-kip load.



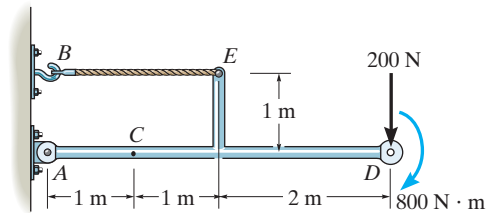
**Prob. 7-25**

**7-26.** Determine the internal normal force, shear force, and bending moment at point  $C$ .



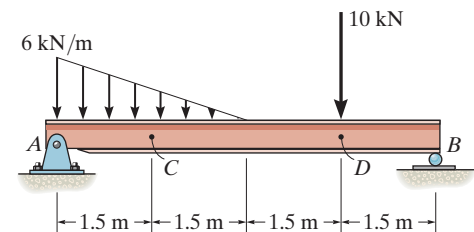
**Prob. 7-26**

**7-27.** Determine the internal normal force, shear force, and moment at point  $C$ .



**Prob. 7-27**

**\*7-28.** Determine the internal normal force, shear force, and moment at points  $C$  and  $D$  in the simply supported beam. Point  $D$  is located just to the left of the 10-kN concentrated load.

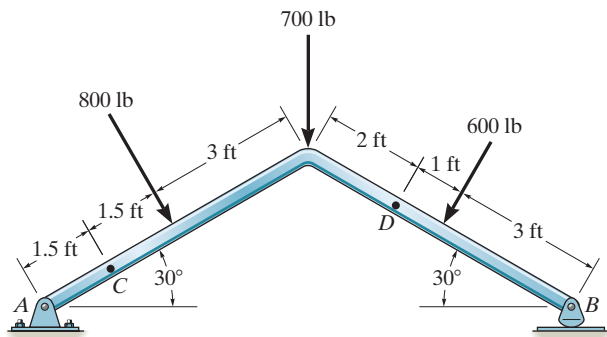


**Prob. 7-28**



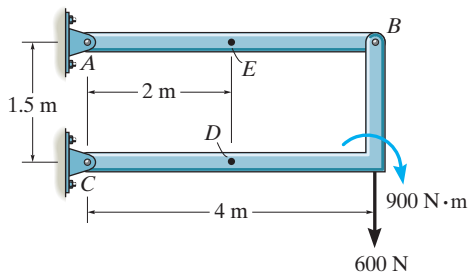
**7-29.** Determine the normal force, shear force, and moment acting at a section passing through point *C*.

**7-30.** Determine the normal force, shear force, and moment acting at a section passing through point *D*.



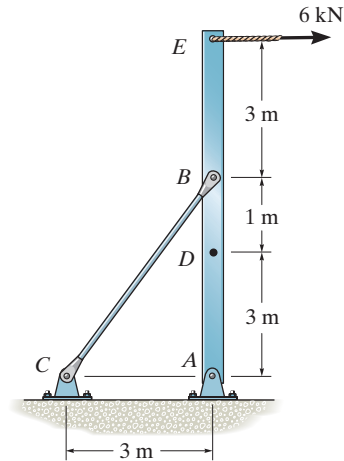
**Probs. 7-29/30**

**7-31.** Determine the internal normal force, shear force, and moment acting at points *D* and *E* of the frame.



**Prob. 7-31**

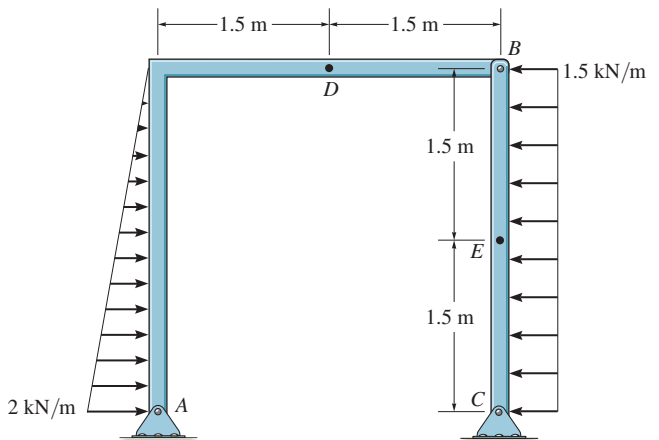
**\*7-32.** Determine the internal normal force, shear force, and moment at point *D*.



**Prob. 7-32**

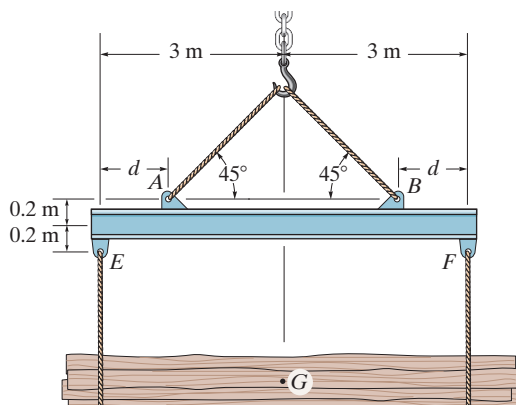
**7-33.** Determine the internal normal force, shear force, and moment at point *D* of the two-member frame.

**7-34.** Determine the internal normal force, shear force, and moment at point *E*.



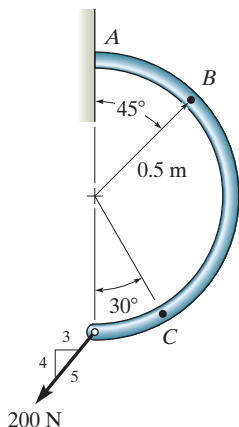
**Probs. 7-33/34**

**7–35.** The strongback or lifting beam is used for materials handling. If the suspended load has a weight of 2 kN and a center of gravity of  $G$ , determine the placement  $d$  of the padeyes on the top of the beam so that there is no moment developed within the length  $AB$  of the beam. The lifting bridle has two legs that are positioned at  $45^\circ$ , as shown.



**Prob. 7–35**

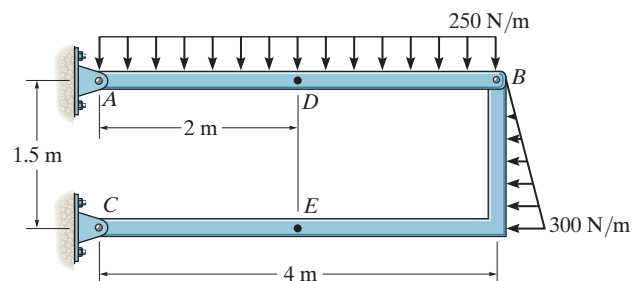
**\*7–36.** Determine the internal normal force, shear force, and moment acting at points  $B$  and  $C$  on the curved rod.



**Prob. 7–36**

**7–37.** Determine the internal normal force, shear force, and moment at point  $D$  of the two-member frame.

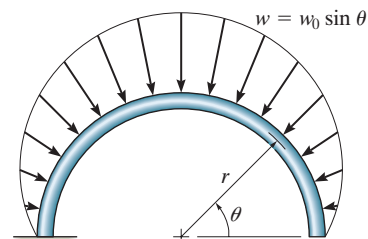
**7–38.** Determine the internal normal force, shear force, and moment at point  $E$  of the two-member frame.



**Probs. 7–37/38**

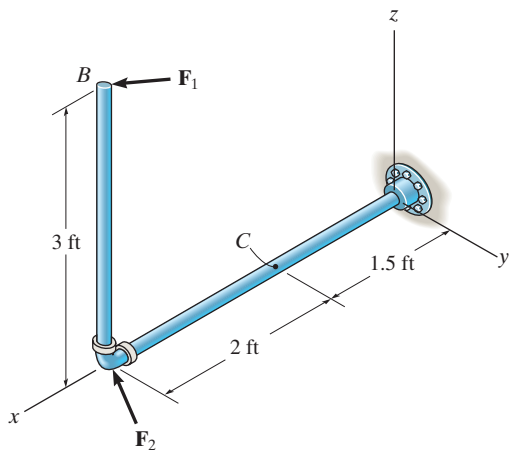
**7–39.** The distributed loading  $w = w_0 \sin \theta$ , measured per unit length, acts on the curved rod. Determine the internal normal force, shear force, and moment in the rod at  $\theta = 45^\circ$ .

**\*7–40.** Solve Prob. 7–39 for  $\theta = 120^\circ$ .



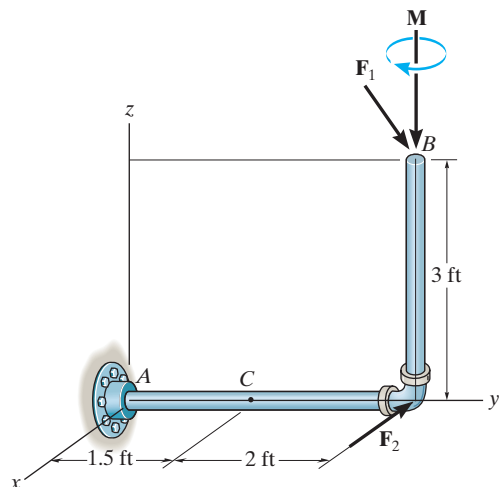
**Probs. 7–39/40**

**7-41.** Determine the  $x$ ,  $y$ ,  $z$  components of force and moment at point  $C$  in the pipe assembly. Neglect the weight of the pipe. Take  $\mathbf{F}_1 = \{350\mathbf{i} - 400\mathbf{j}\}$  lb and  $\mathbf{F}_2 = \{-300\mathbf{j} + 150\mathbf{k}\}$  lb.



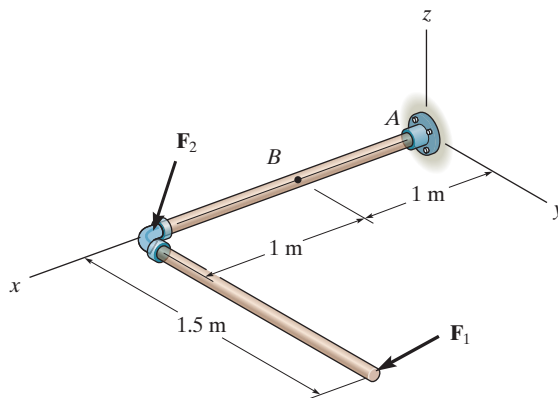
**Prob. 7-41**

**7-42.** Determine the  $x$ ,  $y$ ,  $z$  components of force and moment at point  $C$  in the pipe assembly. Neglect the weight of the pipe. The load acting at  $(0, 3.5 \text{ ft}, 3 \text{ ft})$  is  $\mathbf{F}_1 = \{-24\mathbf{i} - 10\mathbf{k}\}$  lb and  $\mathbf{M} = \{-30\mathbf{k}\}$  lb · ft and at point  $(0, 3.5 \text{ ft}, 0)$   $\mathbf{F}_2 = \{-80\mathbf{i}\}$  lb.



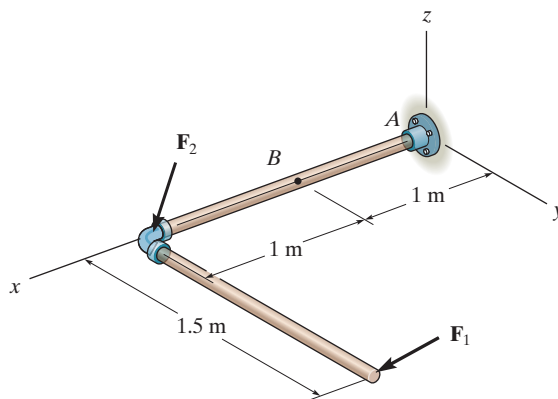
**Prob. 7-42**

**7-43.** Determine the  $x$ ,  $y$ ,  $z$  components of internal loading at a section passing through point  $B$  in the pipe assembly. Neglect the weight of the pipe. Take  $\mathbf{F}_1 = \{200\mathbf{i} - 100\mathbf{j} - 400\mathbf{k}\}$  N and  $\mathbf{F}_2 = \{300\mathbf{i} - 500\mathbf{k}\}$  N.



**Prob. 7-43**

**\*7-44.** Determine the  $x$ ,  $y$ ,  $z$  components of internal loading at a section passing through point  $B$  in the pipe assembly. Neglect the weight of the pipe. Take  $\mathbf{F}_1 = \{100\mathbf{i} - 200\mathbf{j} - 300\mathbf{k}\}$  N and  $\mathbf{F}_2 = \{100\mathbf{i} + 500\mathbf{j}\}$  N.



**Prob. 7-44**

## \*7.2 Shear and Moment Equations and Diagramms

**Beams** are structural members designed to support loadings applied perpendicular to their axes. In general, they are long and straight and have a constant cross-sectional area. They are often classified as to how they are supported. For example, a **simply supported beam** is pinned at one end and roller supported at the other, as in Fig. 7–9a, whereas a **cantilevered beam** is fixed at one end and free at the other. The actual design of a beam requires a detailed knowledge of the *variation* of the internal shear force  $V$  and bending moment  $M$  acting at *each point* along the axis of the beam.\*

These *variations* of  $V$  and  $M$  along the beam's axis can be obtained by using the method of sections discussed in Sec. 7.1. In this case, however, it is necessary to section the beam at an arbitrary distance  $x$  from one end and then apply the equations of equilibrium to the segment having the length  $x$ . Doing this we can then obtain  $V$  and  $M$  as functions of  $x$ .

In general, the internal shear and bending-moment functions will be discontinuous, or their slopes will be discontinuous, at points where a distributed load changes or where concentrated forces or couple moments are applied. Because of this, these functions must be determined for *each segment* of the beam located between any two discontinuities of loading. For example, segments having lengths  $x_1$ ,  $x_2$ , and  $x_3$  will have to be used to describe the variation of  $V$  and  $M$  along the length of the beam in Fig. 7–9a. These functions will be valid *only* within regions from 0 to  $a$  for  $x_1$ , from  $a$  to  $b$  for  $x_2$ , and from  $b$  to  $L$  for  $x_3$ . If the resulting functions of  $x$  are plotted, the graphs are termed the **shear diagram** and **bending-moment diagram**, Fig. 7–9b and Fig. 7–9c, respectively.



To save on material and thereby produce an efficient design, these beams, also called girders, have been tapered, since the internal moment in the beam will be larger at the supports, or piers, than at the center of the span. (© Russell C. Hibbeler)

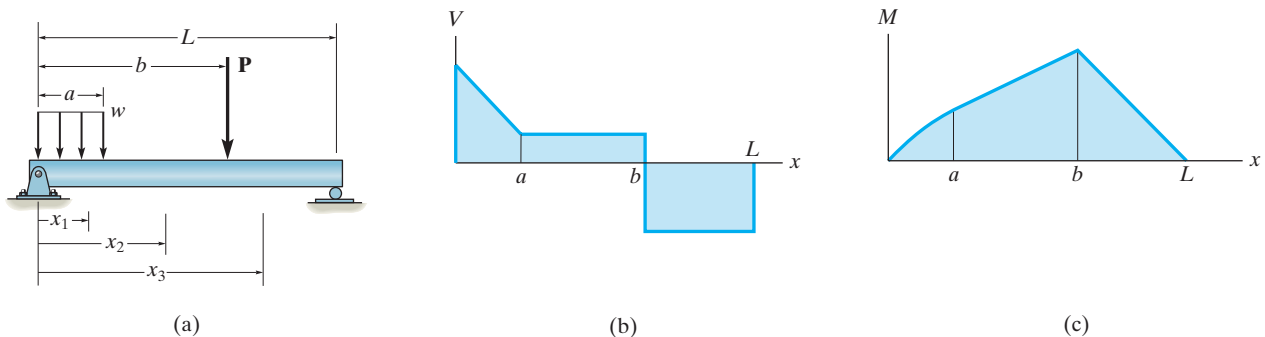


Fig. 7–9

\*The internal normal force is not considered for two reasons. In most cases, the loads applied to a beam act perpendicular to the beam's axis and hence produce only an internal shear force and bending moment. And for design purposes, the beam's resistance to shear, and particularly to bending, is more important than its ability to resist a normal force.

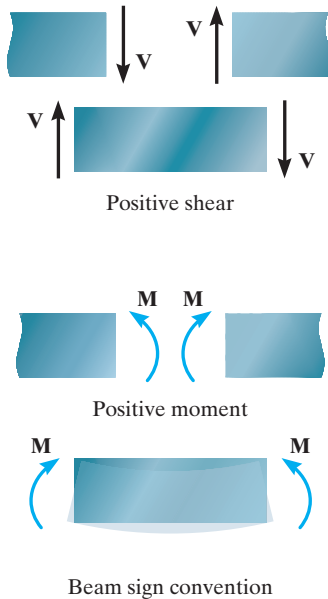


Fig. 7-10

## Important Points

- Shear and moment diagrams for a beam provide graphical descriptions of how the internal shear and moment vary throughout the beam's length.
- To obtain these diagrams, the method of sections is used to determine  $V$  and  $M$  as functions of  $x$ . These results are then plotted. If the load on the beam suddenly changes, then regions between each load must be selected to obtain each function of  $x$ .

## Procedure for Analysis

The shear and bending-moment diagrams for a beam can be constructed using the following procedure.

### Support Reactions.

- Determine all the reactive forces and couple moments acting on the beam and resolve all the forces into components acting perpendicular and parallel to the beam's axis.

### Shear and Moment Functions.

- Specify separate coordinates  $x$  having an origin at the beam's left end and extending to regions of the beam *between* concentrated forces and/or couple moments, or where the distributed loading is continuous.
- Section the beam at each distance  $x$  and draw the free-body diagram of one of the segments. Be sure  $\mathbf{V}$  and  $\mathbf{M}$  are shown acting in their *positive sense*, in accordance with the sign convention given in Fig. 7-10.
- The shear  $V$  is obtained by summing forces perpendicular to the beam's axis, and the moment  $M$  is obtained by summing moments about the sectioned end of the segment.

### Shear and Moment Diagrams.

- Plot the shear diagram ( $V$  versus  $x$ ) and the moment diagram ( $M$  versus  $x$ ). If computed values of the functions describing  $V$  and  $M$  are *positive*, the values are plotted above the  $x$  axis, whereas *negative* values are plotted below the  $x$  axis.



The shelving arms must be designed to resist the internal loading in the arms caused by the lumber. (© Russell C. Hibbeler)

## EXAMPLE 7.6

Draw the shear and moment diagrams for the shaft shown in Fig. 7–11a. The support at  $A$  is a thrust bearing and the support at  $C$  is a journal bearing.

## SOLUTION

**Support Reactions.** The support reactions are shown on the shaft's free-body diagram, Fig. 7–11d.

**Shear and Moment Functions.** The shaft is sectioned at an arbitrary distance  $x$  from point  $A$ , extending within the region  $AB$ , and the free-body diagram of the left segment is shown in Fig. 7–11b. The unknowns  $\mathbf{V}$  and  $\mathbf{M}$  are assumed to act in the *positive sense* on the right-hand face of the segment according to the established sign convention. Applying the equilibrium equations yields

$$+\uparrow \Sigma F_y = 0; \quad V = 2.5 \text{ kN} \quad (1)$$

$$\zeta + \Sigma M = 0; \quad M = 2.5x \text{ kN} \cdot \text{m} \quad (2)$$

A free-body diagram for a left segment of the shaft extending from  $A$  a distance  $x$ , within the region  $BC$  is shown in Fig. 7–11c. As always,  $\mathbf{V}$  and  $\mathbf{M}$  are shown acting in the positive sense. Hence,

$$+\uparrow \Sigma F_y = 0; \quad 2.5 \text{ kN} - 5 \text{ kN} - V = 0$$

$$V = -2.5 \text{ kN} \quad (3)$$

$$\zeta + \Sigma M = 0; \quad M + 5 \text{ kN}(x - 2 \text{ m}) - 2.5 \text{ kN}(x) = 0$$

$$M = (10 - 2.5x) \text{ kN} \cdot \text{m} \quad (4)$$

**Shear and Moment Diagrams.** When Eqs. 1 through 4 are plotted within the regions in which they are valid, the shear and moment diagrams shown in Fig. 7–11d are obtained. The shear diagram indicates that the internal shear force is always 2.5 kN (positive) within segment  $AB$ . Just to the right of point  $B$ , the shear force changes sign and remains at a constant value of  $-2.5$  kN for segment  $BC$ . The moment diagram starts at zero, increases linearly to point  $B$  at  $x = 2$  m, where  $M_{\max} = 2.5 \text{ kN}(2 \text{ m}) = 5 \text{ kN} \cdot \text{m}$ , and thereafter decreases back to zero.

**NOTE:** It is seen in Fig. 7–11d that the graphs of the shear and moment diagrams “jump” or changes abruptly where the concentrated force acts, i.e., at points  $A$ ,  $B$ , and  $C$ . For this reason, as stated earlier, it is necessary to express both the shear and moment functions separately for regions between concentrated loads. It should be realized, however, that all loading discontinuities are mathematical, arising from the *idealization of a concentrated force and couple moment*. Physically, loads are always applied over a finite area, and if the actual load variation could be accounted for, the shear and moment diagrams would then be continuous over the shaft's entire length.

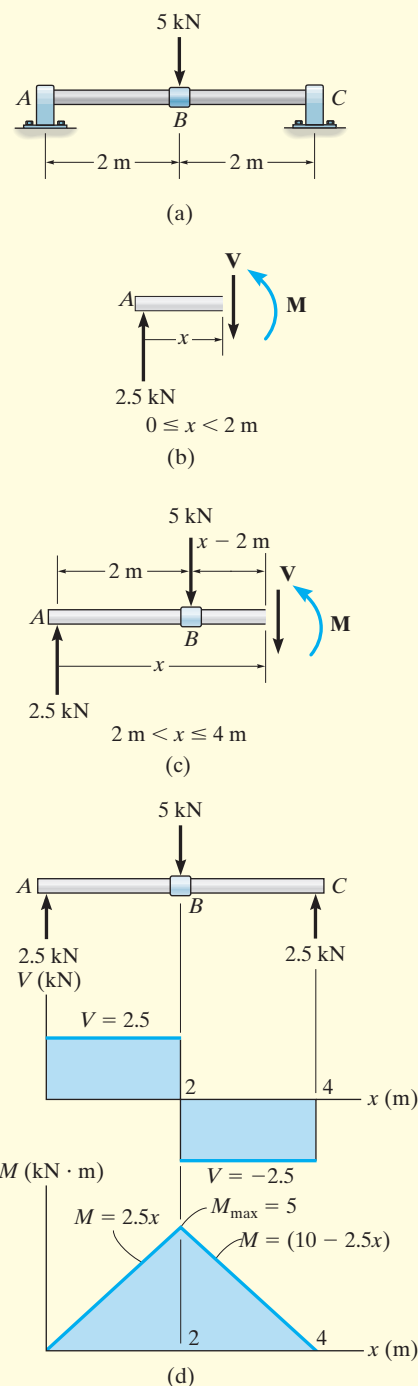


Fig. 7-11

## EXAMPLE 7.7

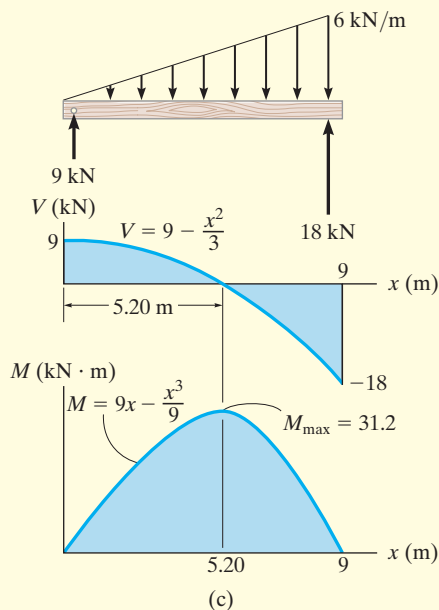
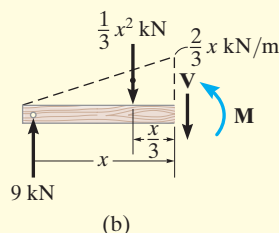
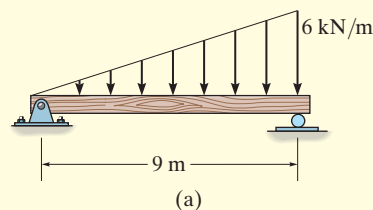


Fig. 7-12

Draw the shear and moment diagrams for the beam shown in Fig. 7-12a.

## SOLUTION

**Support Reactions.** The support reactions are shown on the beam's free-body diagram, Fig. 7-12c.

**Shear and Moment Functions.** A free-body diagram for a left segment of the beam having a length  $x$  is shown in Fig. 7-12b. Due to proportional triangles, the distributed loading acting at the end of this segment has an intensity of  $w/x = 6/9$  or  $w = (2/3)x$ . It is replaced by a resultant force *after* the segment is isolated as a free-body diagram. The *magnitude* of the resultant force is equal to  $\frac{1}{2}(x)(\frac{2}{3}x) = \frac{1}{3}x^2$ . This force *acts through the centroid* of the distributed loading area, a distance  $\frac{1}{3}x$  from the right end. Applying the two equations of equilibrium yields

$$\begin{aligned}
 +\uparrow \Sigma F_y &= 0; & 9 - \frac{1}{3}x^2 - V &= 0 \\
 V &= \left(9 - \frac{x^2}{3}\right) \text{ kN} & & (1)
 \end{aligned}$$

$$\begin{aligned}
 \zeta + \Sigma M &= 0; & M + \frac{1}{3}x^2\left(\frac{x}{3}\right) - 9x &= 0 \\
 M &= \left(9x - \frac{x^3}{9}\right) \text{ kN} \cdot \text{m} & & (2)
 \end{aligned}$$

**Shear and Moment Diagrams.** The shear and moment diagrams shown in Fig. 7-12c are obtained by plotting Eqs. 1 and 2.

The point of *zero shear* can be found using Eq. 1:

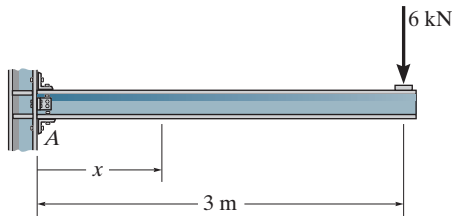
$$\begin{aligned}
 V &= 9 - \frac{x^2}{3} = 0 \\
 x &= 5.20 \text{ m}
 \end{aligned}$$

**NOTE:** It will be shown in Sec. 7.3 that this value of  $x$  happens to represent the point on the beam where the *maximum moment* occurs. Using Eq. 2, we have

$$\begin{aligned}
 M_{\max} &= \left(9(5.20) - \frac{(5.20)^3}{9}\right) \text{ kN} \cdot \text{m} \\
 &= 31.2 \text{ kN} \cdot \text{m}
 \end{aligned}$$

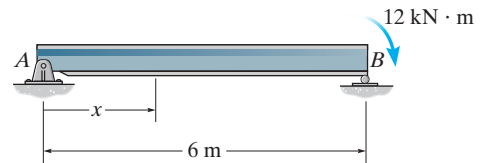
# FUNDAMENTAL PROBLEMS

**F7-7.** Determine the shear and moment as a function of  $x$ , and then draw the shear and moment diagrams.



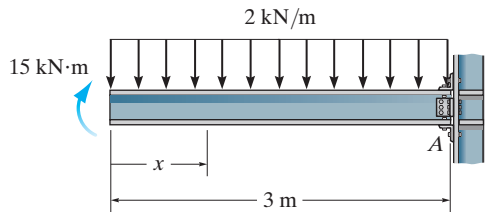
**Prob. F7-7**

**F7-10.** Determine the shear and moment as a function of  $x$ , and then draw the shear and moment diagrams.



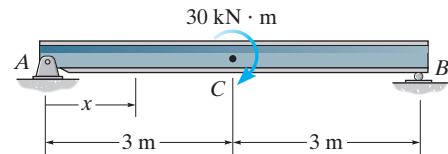
**Prob. F7-10**

**F7-8.** Determine the shear and moment as a function of  $x$ , and then draw the shear and moment diagrams.



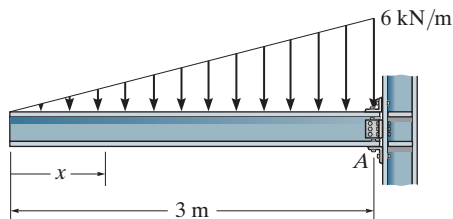
**Prob. F7-8**

**F7-11.** Determine the shear and moment as a function of  $x$ , where  $0 \leq x < 3$  m and  $3 \text{ m} < x \leq 6$  m, and then draw the shear and moment diagrams.



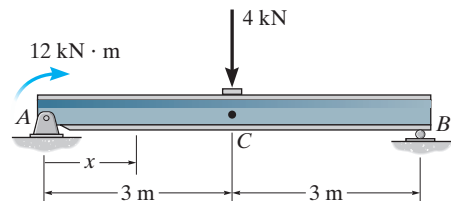
**Prob. F7-11**

**F7-9.** Determine the shear and moment as a function of  $x$ , and then draw the shear and moment diagrams.



**Prob. F7-9**

**F7-12.** Determine the shear and moment as a function of  $x$ , where  $0 \leq x < 3$  m and  $3 \text{ m} < x \leq 6$  m, and then draw the shear and moment diagrams.

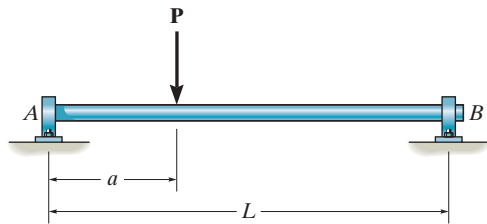


**Prob. F7-12**



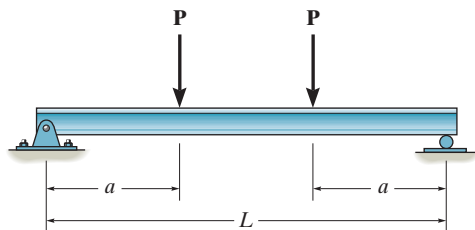
## PROBLEMS

**7-45.** Draw the shear and moment diagrams for the shaft (a) in terms of the parameters shown; (b) set  $P = 9 \text{ kN}$ ,  $a = 2 \text{ m}$ ,  $L = 6 \text{ m}$ . There is a thrust bearing at  $A$  and a journal bearing at  $B$ .



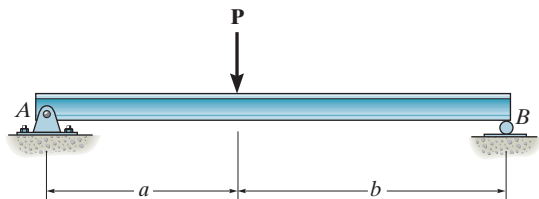
Prob. 7-45

**7-46.** Draw the shear and moment diagrams for the beam (a) in terms of the parameters shown; (b) set  $P = 800 \text{ lb}$ ,  $a = 5 \text{ ft}$ ,  $L = 12 \text{ ft}$ .



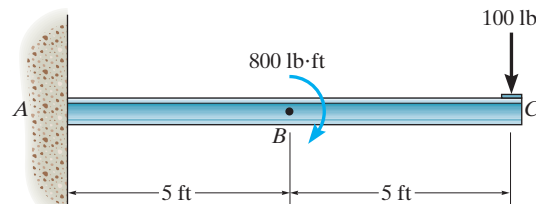
Prob. 7-46

**7-47.** Draw the shear and moment diagrams for the beam (a) in terms of the parameters shown; (b) set  $P = 600 \text{ lb}$ ,  $a = 5 \text{ ft}$ ,  $b = 7 \text{ ft}$ .



Prob. 7-47

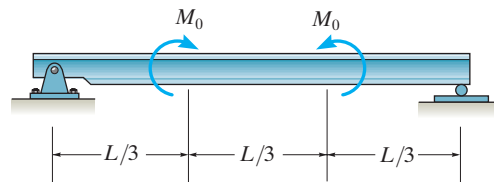
**\*7-48.** Draw the shear and moment diagrams for the cantilevered beam.



Prob. 7-48

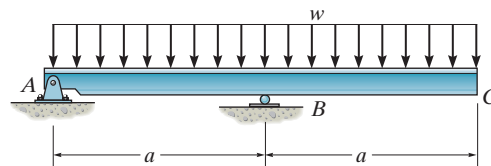
**7-49.** Draw the shear and moment diagrams of the beam (a) in terms of the parameters shown; (b) set  $M_0 = 500 \text{ N} \cdot \text{m}$ ,  $L = 8 \text{ m}$ .

**7-50.** If  $L = 9 \text{ m}$ , the beam will fail when the maximum shear force is  $V_{\max} = 5 \text{ kN}$  or the maximum bending moment is  $M_{\max} = 2 \text{ kN} \cdot \text{m}$ . Determine the magnitude  $M_0$  of the largest couple moments it will support.



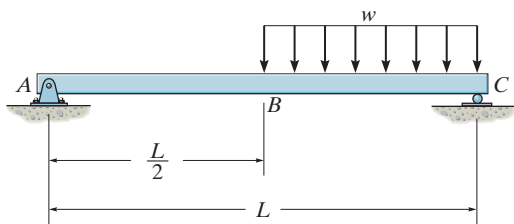
Probs. 7-49/50

**7-51.** Draw the shear and moment diagrams for the beam.



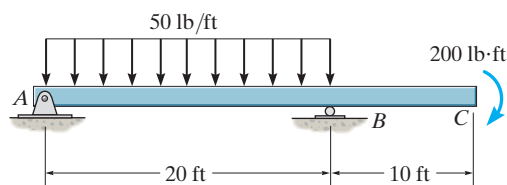
Prob. 7-51

**\*7-52.** Draw the shear and moment diagrams for the beam.



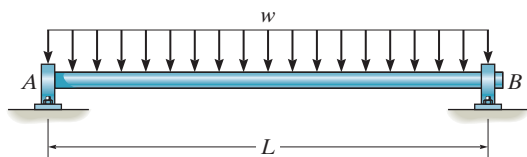
**Prob. 7-52**

**7-53.** Draw the shear and bending-moment diagrams for the beam.



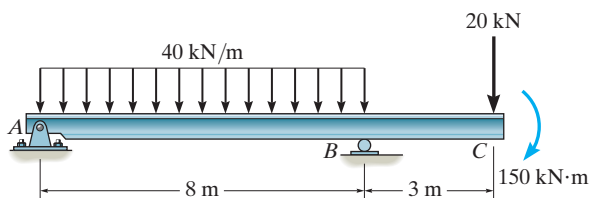
**Prob. 7-53**

**7-54.** The shaft is supported by a smooth thrust bearing at A and a smooth journal bearing at B. Draw the shear and moment diagrams for the shaft (a) in terms of the parameters shown; (b) set  $w = 500 \text{ lb/ft}$ ,  $L = 10 \text{ ft}$ .



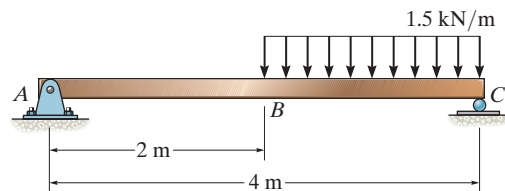
**Prob. 7-54**

**7-55.** Draw the shear and moment diagrams for the beam.



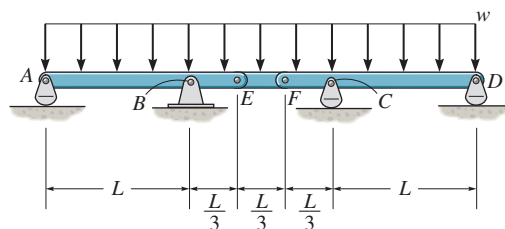
**Prob. 7-55**

**\*7-56.** Draw the shear and moment diagrams for the beam.



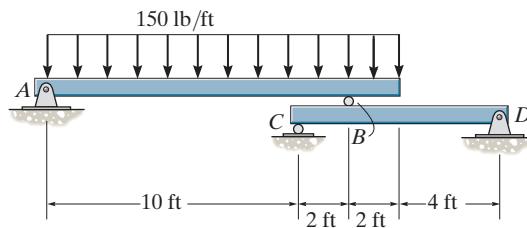
**Prob. 7-56**

**7-57.** Draw the shear and moment diagrams for the compound beam. The beam is pin connected at E and F.



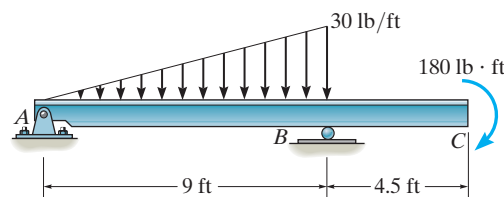
**Prob. 7-57**

**7-58.** Draw the shear and bending-moment diagrams for each of the two segments of the compound beam.



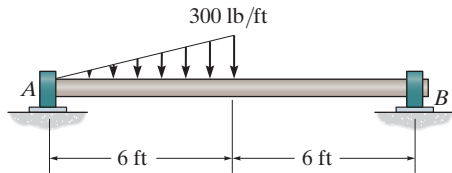
**Prob. 7-58**

**7-59.** Draw the shear and moment diagrams for the beam.



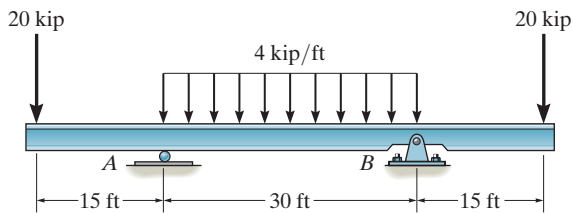
**Prob. 7-59**

**\*7-60.** The shaft is supported by a smooth thrust bearing at  $A$  and a smooth journal bearing at  $B$ . Draw the shear and moment diagrams for the shaft.



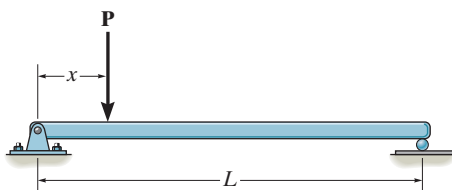
**Prob. 7-60**

**7-61.** Draw the shear and moment diagrams for the beam.



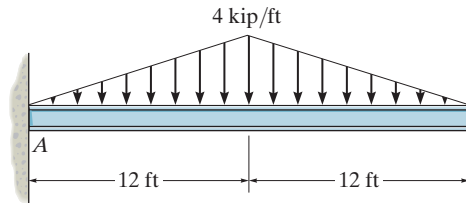
**Prob. 7-61**

**7-62.** The beam will fail when the maximum internal moment is  $M_{\max}$ . Determine the position  $x$  of the concentrated force  $P$  and its smallest magnitude that will cause failure.



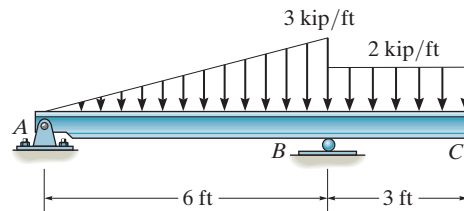
**Prob. 7-62**

**7-63.** Draw the shear and moment diagrams for the beam.



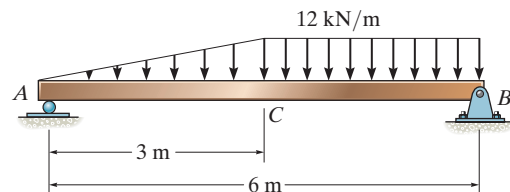
**Prob. 7-63**

**\*7-64.** Draw the shear and moment diagrams for the beam.



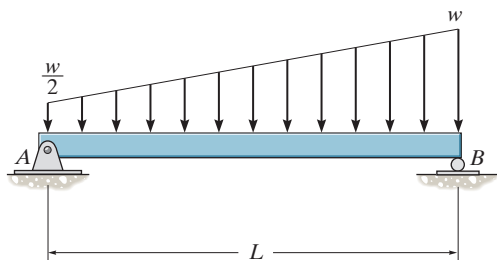
**Prob. 7-64**

**7-65.** Draw the shear and moment diagrams for the beam.



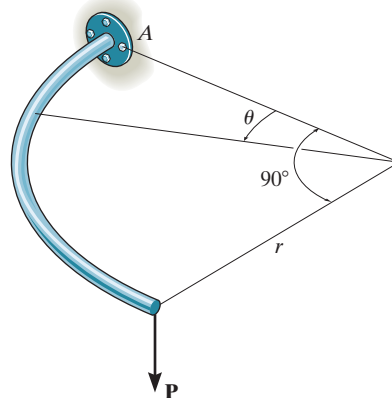
**Prob. 7-65**

7-66. Draw the shear and moment diagrams for the beam.



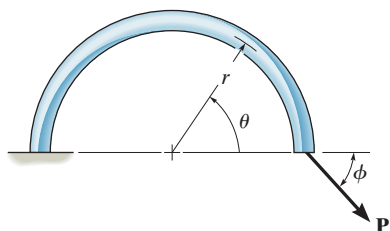
Prob. 7-66

\*7-68. The quarter circular rod lies in the horizontal plane and supports a vertical force  $\mathbf{P}$  at its end. Determine the magnitudes of the components of the internal shear force, moment, and torque acting in the rod as a function of the angle  $\theta$ .



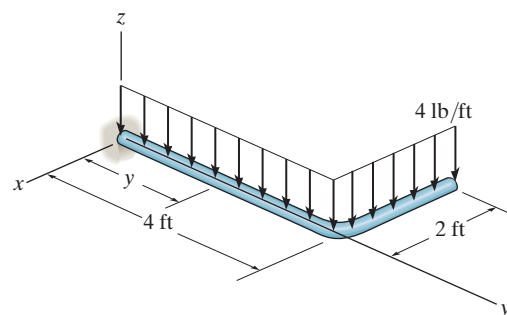
Prob. 7-68

7-67. Determine the internal normal force, shear force, and moment in the curved rod as a function of  $\theta$ . The force  $\mathbf{P}$  acts at the constant angle  $\phi$ .



Prob. 7-67

7-69. Express the internal shear and moment components acting in the rod as a function of  $y$ , where  $0 \leq y \leq 4$  ft.



Prob. 7-69



In order to design the beam used to support these power lines, it is important to first draw the shear and moment diagrams for the beam.  
(© Russell C. Hibbeler)

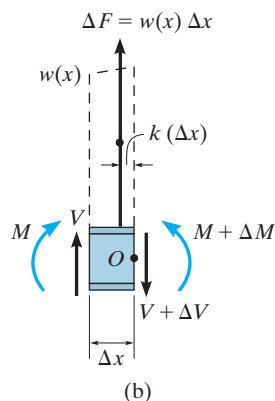
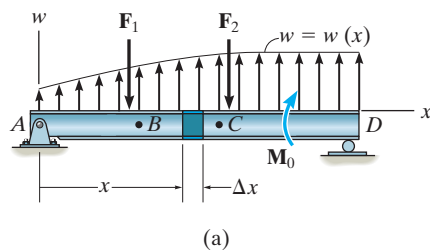


Fig. 7-13

## \*7.3 Relations between Distributed Load, Shear, and Moment

If a beam is subjected to several concentrated forces, couple moments, and distributed loads, the method of constructing the shear and bending-moment diagrams discussed in Sec. 7.2 may become quite tedious. In this section a simpler method for constructing these diagrams is discussed—a method based on differential relations that exist between the load, shear, and bending moment.

**Distributed Load.** Consider the beam  $AD$  shown in Fig. 7-13a, which is subjected to an arbitrary load  $w = w(x)$  and a series of concentrated forces and couple moments. In the following discussion, the *distributed load* will be considered *positive* when the *loading acts upward* as shown. A free-body diagram for a small segment of the beam having a length  $\Delta x$  is chosen at a point  $x$  along the beam which is *not* subjected to a concentrated force or couple moment, Fig. 7-13b. Hence any results obtained will not apply at these points of concentrated loading. The internal shear force and bending moment shown on the free-body diagram are assumed to act in the *positive sense* according to the established sign convention. Note that both the shear force and moment acting on the right-hand face must be increased by a small, finite amount in order to keep the segment in equilibrium. The distributed loading has been replaced by a resultant force  $\Delta F = w(x) \Delta x$  that acts at a fractional distance  $k(\Delta x)$  from the right end, where  $0 < k < 1$  [for example, if  $w(x)$  is *uniform*,  $k = \frac{1}{2}$ ].

**Relation between the Distributed Load and Shear.** If we apply the force equation of equilibrium to the segment, then

$$+\uparrow \Sigma F_y = 0; \quad V + w(x)\Delta x - (V + \Delta V) = 0$$

$$\Delta V = w(x)\Delta x$$

Dividing by  $\Delta x$ , and letting  $\Delta x \rightarrow 0$ , we get

$$\frac{dV}{dx} = w(x)$$

Slope of  
shear diagram = Distributed load  
intensity

(7-1)

If we rewrite the above equation in the form  $dV = w(x)dx$  and perform an integration between any two points  $B$  and  $C$  on the beam, we see that

$$\Delta V = \int w(x) dx$$

Change in shear = Area under loading curve

(7-2)

**Relation between the Shear and Moment.** If we apply the moment equation of equilibrium about point  $O$  on the free-body diagram in Fig. 7-13b, we get

$$\zeta + \Sigma M_O = 0; \quad (M + \Delta M) - [w(x)\Delta x]k\Delta x - V\Delta x - M = 0$$

$$\Delta M = V\Delta x + k w(x)\Delta x^2$$

Dividing both sides of this equation by  $\Delta x$ , and letting  $\Delta x \rightarrow 0$ , yields

$$\frac{dM}{dx} = V$$

Slope of moment diagram = Shear

(7-3)

In particular, notice that a maximum bending moment  $|M|_{\max}$  will occur at the point where the slope  $dM/dx = 0$ , since this is where the shear is equal to zero.

If Eq. 7-3 is rewritten in the form  $dM = V dx$  and integrated between any two points  $B$  and  $C$  on the beam, we have

$$\Delta M = \int V dx$$

Change in moment = Area under shear diagram

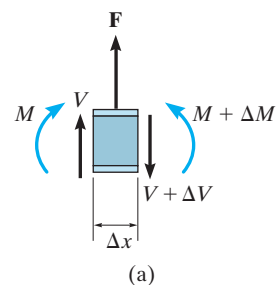
(7-4)

As stated previously, the above equations do not apply at points where a *concentrated* force or couple moment acts. These two special cases create *discontinuities* in the shear and moment diagrams, and as a result, each deserves separate treatment.

**Force.** A free-body diagram of a small segment of the beam in Fig. 7-13a, taken from under one of the forces, is shown in Fig. 7-14a. Here force equilibrium requires

$$+\uparrow \Sigma F_y = 0; \quad \Delta V = F$$
(7-5)

Since the *change in shear is positive*, the shear diagram will “jump” *upward* when  $\mathbf{F}$  acts *upward* on the beam. Likewise, the jump in shear ( $\Delta V$ ) is downward when  $\mathbf{F}$  acts downward.



**Fig. 7-14**

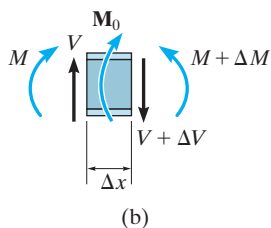


Fig. 7-14 (cont.)

**Couple Moment.** If we remove a segment of the beam in Fig. 7-13a that is located at the couple moment  $\mathbf{M}_0$ , the free-body diagram shown in Fig. 7-14b results. In this case letting  $\Delta x \rightarrow 0$ , moment equilibrium requires

$$\zeta + \Sigma M = 0; \quad \Delta M = M_0 \quad (7-6)$$

Thus, the *change in moment is positive*, or the moment diagram will “jump” *upward* if  $\mathbf{M}_0$  is *clockwise*. Likewise, the jump  $\Delta M$  is downward when  $\mathbf{M}_0$  is counterclockwise.

The examples which follow illustrate application of the above equations when used to construct the shear and moment diagrams. After working through these examples, it is recommended that you also go back and solve Examples 7.6 and 7.7 using this method.



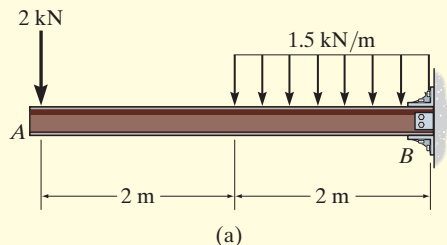
This concrete beam is used to support the deck. Its size and the placement of steel reinforcement within it can be determined once the shear and moment diagrams have been established. (© Russell C. Hibbeler)

## Important Points

- The slope of the shear diagram at a point is equal to the intensity of the distributed loading, where positive distributed loading is upward, i.e.,  $dV/dx = w(x)$ .
- The change in the shear  $\Delta V$  between two points is equal to *the area* under the distributed-loading curve between the points.
- If a concentrated force acts upward on the beam, the shear will jump upward by the same amount.
- The slope of the moment diagram at a point is equal to the shear, i.e.,  $dM/dx = V$ .
- The change in the moment  $\Delta M$  between two points is equal to *the area* under the shear diagram between the two points.
- If a *clockwise* couple moment acts on the beam, the shear will not be affected; however, the moment diagram will jump *upward* by the amount of the moment.
- Points of *zero shear* represent points of *maximum or minimum moment* since  $dM/dx = 0$ .
- Because two integrations of  $w = w(x)$  are involved to first determine the change in shear,  $\Delta V = \int w(x) dx$ , then to determine the change in moment,  $\Delta M = \int V dx$ , then if the loading curve  $w = w(x)$  is a polynomial of degree  $n$ ,  $V = V(x)$  will be a curve of degree  $n + 1$ , and  $M = M(x)$  will be a curve of degree  $n + 2$ .

**EXAMPLE 7.8**

Draw the shear and moment diagrams for the cantilever beam in Fig. 7-15a.

**Fig. 7-15****SOLUTION**

The support reactions at the fixed support  $B$  are shown in Fig. 7-15b.

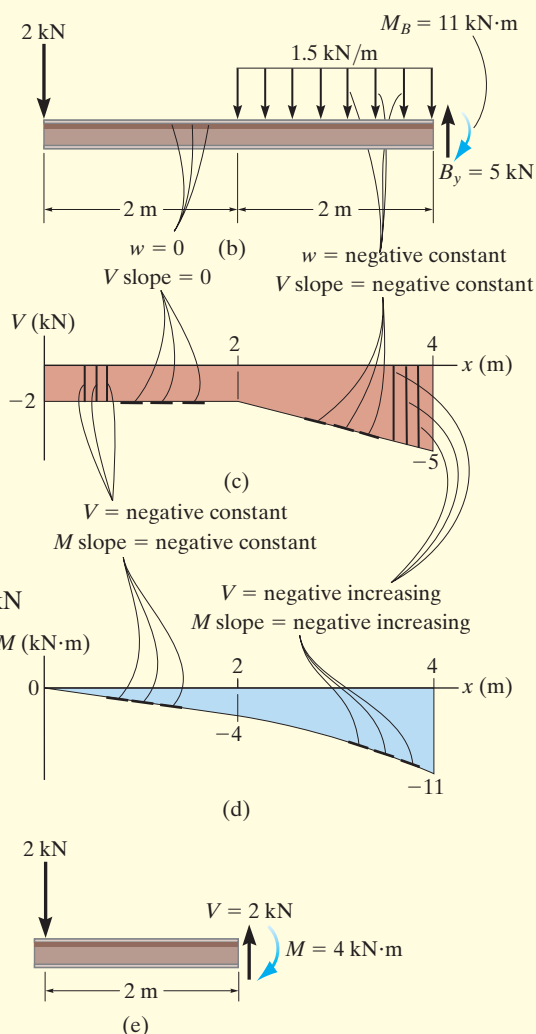
**Shear Diagram.** The shear at end  $A$  is  $-2$  kN. This value is plotted at  $x = 0$ , Fig. 7-15c. Notice how the shear diagram is constructed by following the slopes defined by the loading  $w$ . The shear at  $x = 4$  m is  $-5$  kN, the reaction on the beam. This value can be verified by finding the area under the distributed loading; i.e.,

$$V|_{x=4\text{ m}} = V|_{x=2\text{ m}} + \Delta V = -2\text{ kN} - (1.5\text{ kN/m})(2\text{ m}) = -5\text{ kN}$$

**Moment Diagram.** The moment of zero at  $x = 0$  is plotted in Fig. 7-15d. Construction of the moment diagram is based on knowing that its slope is equal to the shear at each point. The change of moment from  $x = 0$  to  $x = 2$  m is determined from the area under the shear diagram. Hence, the moment at  $x = 2$  m is

$$M|_{x=2\text{ m}} = M|_{x=0} + \Delta M = 0 + [-2\text{ kN}(2\text{ m})] = -4\text{ kN}\cdot\text{m}$$

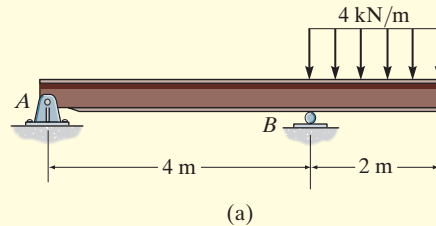
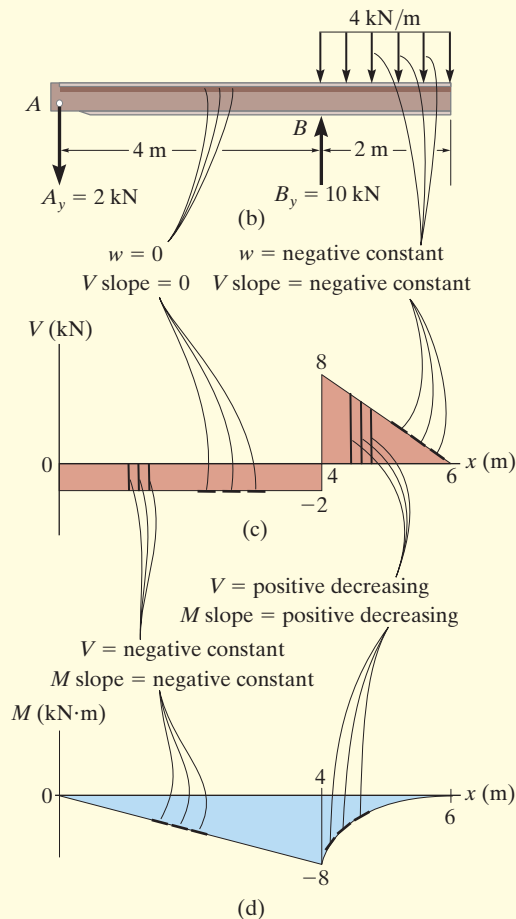
This same value can be determined from the method of sections, Fig. 7-15e.





**EXAMPLE 7.9**

Draw the shear and moment diagrams for the overhang beam in Fig. 7-16a.

**Fig. 7-16****SOLUTION**

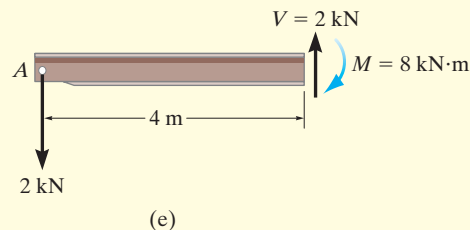
The support reactions are shown in Fig. 7-16b.

**Shear Diagram.** The shear of  $-2$  kN at end A of the beam is plotted at  $x = 0$ , Fig. 7-16c. The slopes are determined from the loading and from this the shear diagram is constructed, as indicated in the figure. In particular, notice the positive jump of 10 kN at  $x = 4$  m due to the force  $B_y$ , as indicated in the figure.

**Moment Diagram.** The moment of zero at  $x = 0$  is plotted, Fig. 7-16d, then following the behavior of the slope found from the shear diagram, the moment diagram is constructed. The moment at  $x = 4$  m is found from the area under the shear diagram.

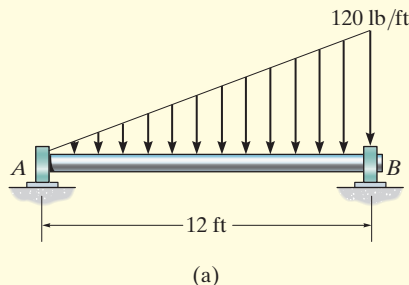
$$M|_{x=4\text{ m}} = M|_{x=0} + \Delta M = 0 + [-2\text{ kN}(4\text{ m})] = -8\text{ kN}\cdot\text{m}$$

We can also obtain this value by using the method of sections, as shown in Fig. 7-16e.



**EXAMPLE 7.10**

The shaft in Fig. 7-17a is supported by a thrust bearing at  $A$  and a journal bearing at  $B$ . Draw the shear and moment diagrams.

**Fig. 7-17****SOLUTION**

The support reactions are shown in Fig. 7-17b.

**Shear Diagram.** As shown in Fig. 7-17c, the shear at  $x=0$  is +240. Following the slope defined by the loading, the shear diagram is constructed, where at  $B$  its value is -480 lb. Since the shear changes sign, the point where  $V=0$  must be located. To do this we will use the method of sections. The free-body diagram of the left segment of the shaft, sectioned at an arbitrary position  $x$  within the region  $0 \leq x < 12$  ft, is shown in Fig. 7-17e. Notice that the intensity of the distributed load at  $x$  is  $w=10x$ , which has been found by proportional triangles, i.e.,  $120/12 = w/x$ .

Thus, for  $V=0$ ,

$$+\uparrow \Sigma F_y = 0; \quad 240 \text{ lb} - \frac{1}{2}(10x)x = 0$$

$$x = 6.93 \text{ ft}$$

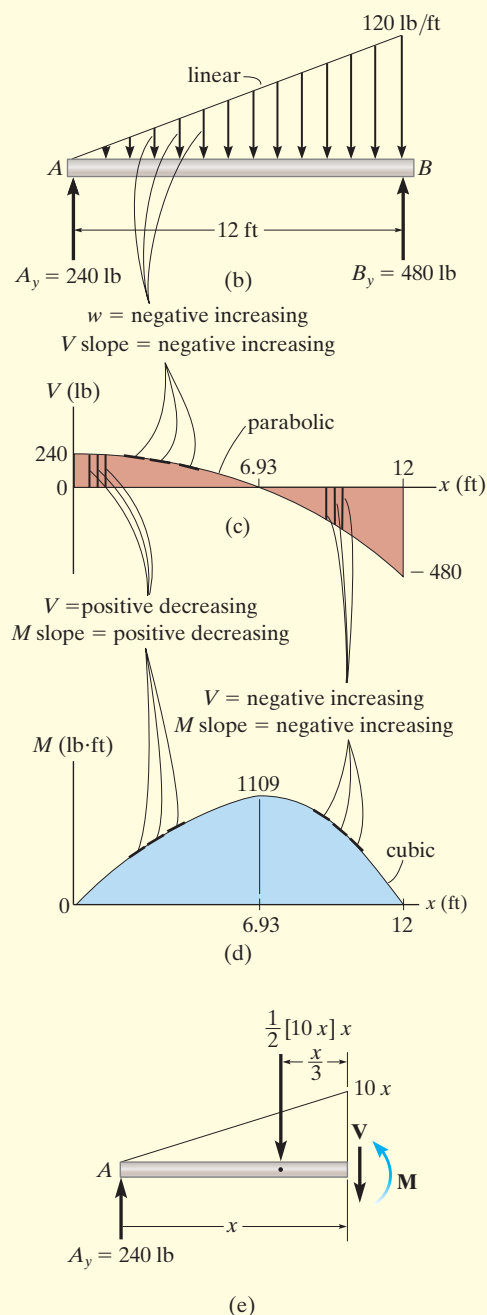
**Moment Diagram.** The moment diagram starts at 0 since there is no moment at  $A$ , then it is constructed based on the slope as determined from the shear diagram. The maximum moment occurs at  $x=6.93$  ft, where the shear is equal to zero, since  $dM/dx = V = 0$ , Fig. 7-17e,

$$\zeta + \Sigma M = 0;$$

$$M_{\max} + \frac{1}{2}[(10)(6.93)] 6.93 \left(\frac{1}{3}(6.93)\right) - 240(6.93) = 0$$

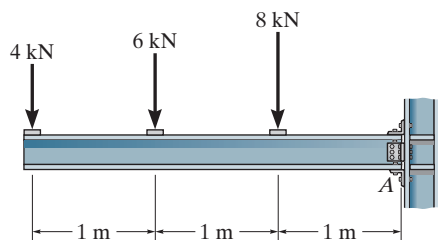
$$M_{\max} = 1109 \text{ lb} \cdot \text{ft}$$

Finally, notice how integration, first of the loading  $w$  which is linear, produces a shear diagram which is parabolic, and then a moment diagram which is cubic.



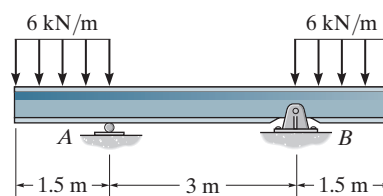
## FUNDAMENTAL PROBLEMS

**F7-13.** Draw the shear and moment diagrams for the beam.



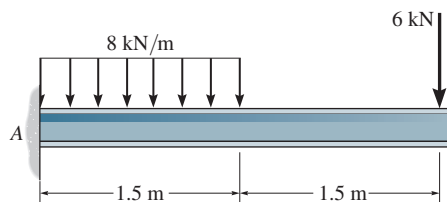
**Prob. F7-13**

**F7-16.** Draw the shear and moment diagrams for the beam.



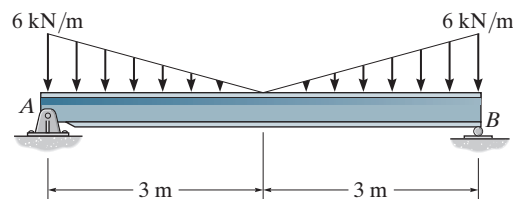
**Prob. F7-16**

**F7-14.** Draw the shear and moment diagrams for the beam.



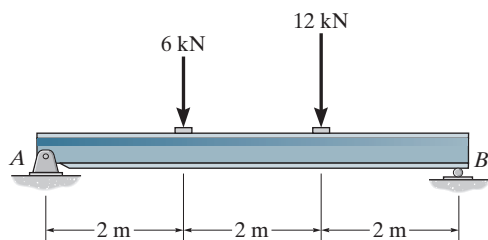
**Prob. F7-14**

**F7-17.** Draw the shear and moment diagrams for the beam.



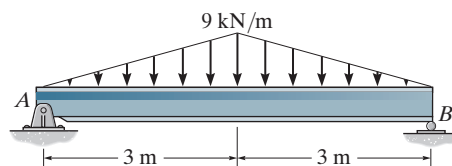
**Prob. F7-17**

**F7-15.** Draw the shear and moment diagrams for the beam.



**Prob. F7-15**

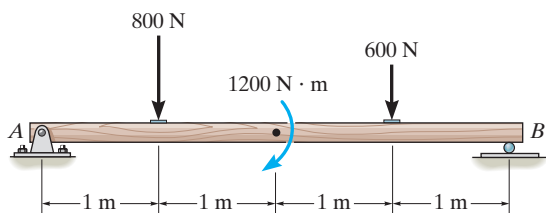
**F7-18.** Draw the shear and moment diagrams for the beam.



**Prob. F7-18**

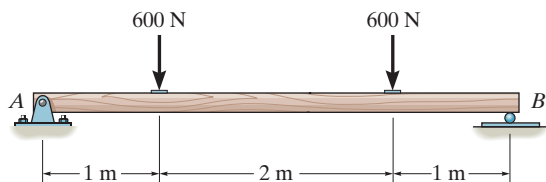
## PROBLEMS

**7-70.** Draw the shear and moment diagrams for the beam.



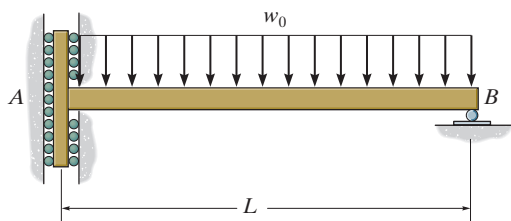
**Prob. 7-70**

**7-71.** Draw the shear and moment diagrams for the beam.



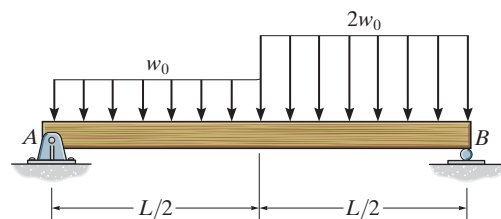
**Prob. 7-71**

**\*7-72.** Draw the shear and moment diagrams for the beam. The support at A offers no resistance to vertical load.



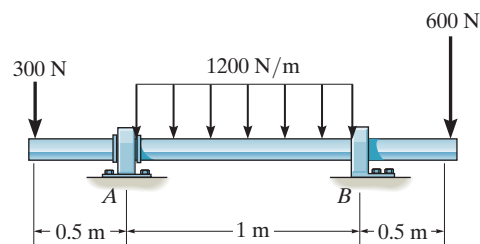
**Prob. 7-72**

**7-73.** Draw the shear and moment diagrams for the simply-supported beam.



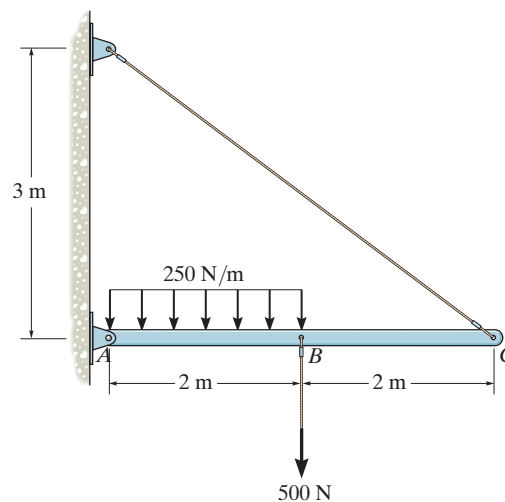
**Prob. 7-73**

**7-74.** Draw the shear and moment diagrams for the beam. The supports at A and B are a thrust bearing and journal bearing, respectively.



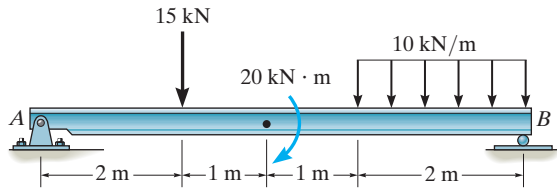
**Prob. 7-74**

**7-75.** Draw the shear and moment diagrams for the beam.



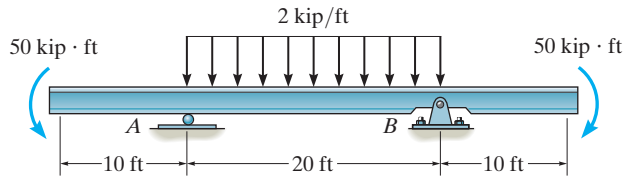
**Prob. 7-75**

**\*7-76.** Draw the shear and moment diagrams for the beam.



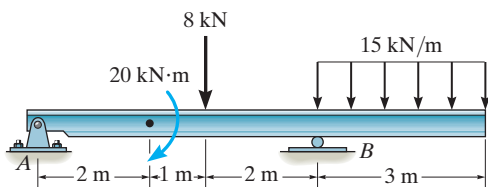
**Prob. 7-76**

**7-77.** Draw the shear and moment diagrams for the beam.



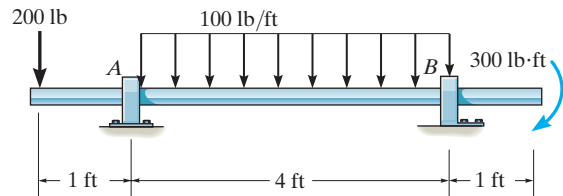
**Prob. 7-77**

**7-78.** Draw the shear and moment diagrams for the beam.



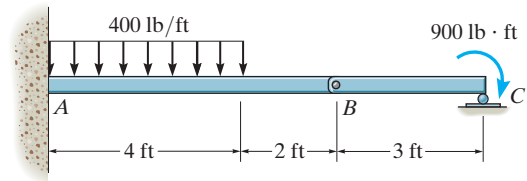
**Prob. 7-78**

**7-79.** Draw the shear and moment diagrams for the shaft. The support at *A* is a journal bearing and at *B* it is a thrust bearing.



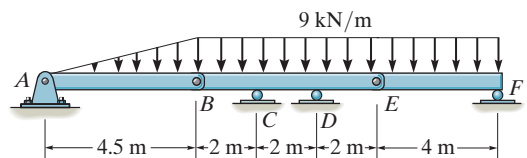
**Prob. 7-79**

**\*7-80.** Draw the shear and moment diagrams for the beam.



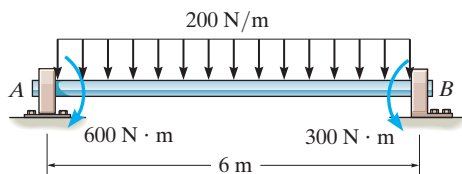
**Prob. 7-80**

**7-81.** The beam consists of three segments pin connected at *B* and *E*. Draw the shear and moment diagrams for the beam.



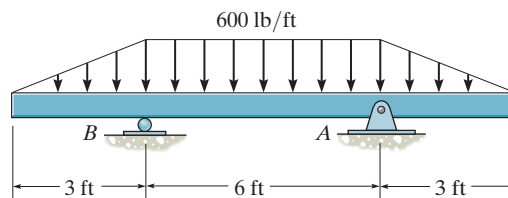
**Prob. 7-81**

**7-82.** Draw the shear and moment diagrams for the beam. The supports at  $A$  and  $B$  are a thrust and journal bearing, respectively.



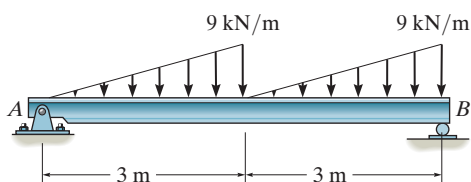
**Prob. 7-82**

**7-85.** Draw the shear and moment diagrams for the beam.



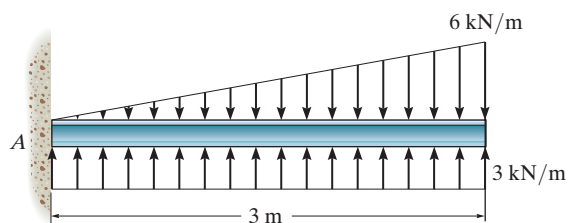
**Prob. 7-85**

**7-83.** Draw the shear and moment diagrams for the beam.



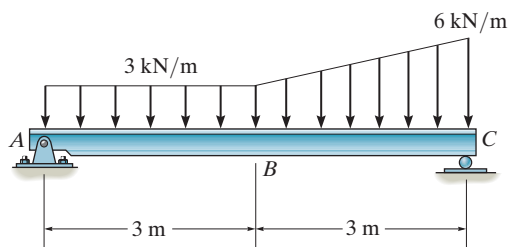
**Prob. 7-83**

**7-86.** Draw the shear and moment diagrams for the beam.



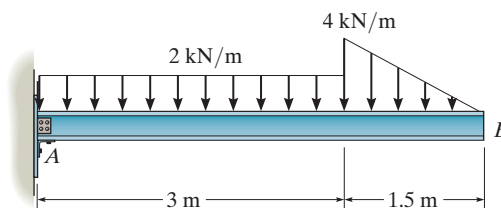
**Prob. 7-86**

**\*7-84.** Draw the shear and moment diagrams for the beam.



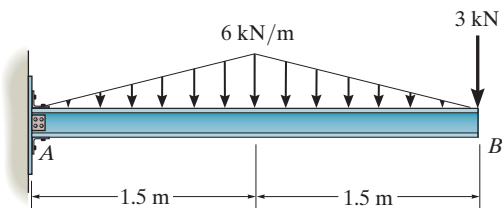
**Prob. 7-84**

**7-87.** Draw the shear and moment diagrams for the beam.



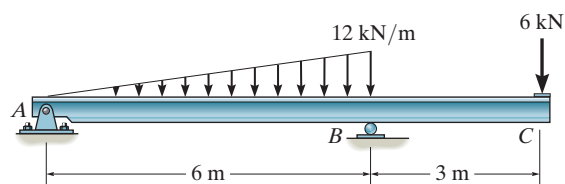
**Prob. 7-87**

\*7-88. Draw the shear and moment diagrams for the beam.



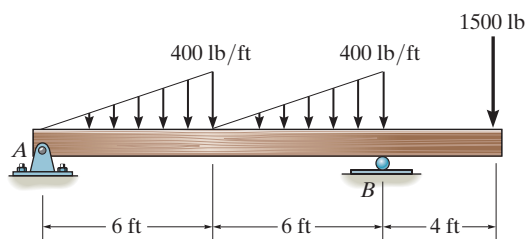
Prob. 7-88

7-91. Draw the shear and moment diagrams for the beam.



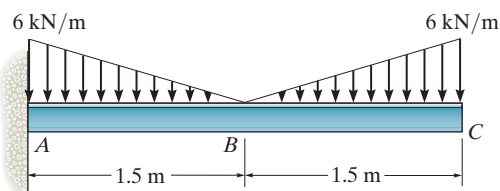
Prob. 7-91

7-89. Draw the shear and moment diagrams for the beam.



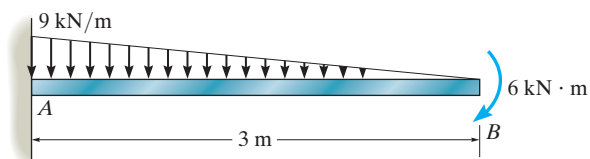
Prob. 7-89

\*7-92. Draw the shear and moment diagrams for the beam.



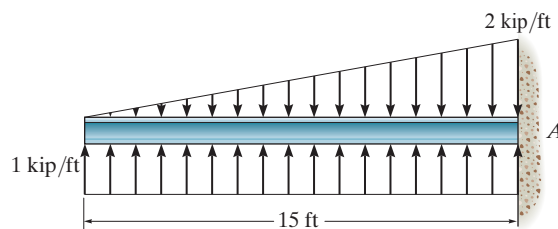
Prob. 7-92

7-90. Draw the shear and moment diagrams for the beam.



Prob. 7-90

7-93. Draw the shear and moment diagrams for the beam.



Prob. 7-93

## \*7.4 Cables

Flexible cables and chains combine strength with lightness and often are used in structures for support and to transmit loads from one member to another. When used to support suspension bridges and trolley wheels, cables form the main load-carrying element of the structure. In the force analysis of such systems, the weight of the cable itself may be neglected because it is often small compared to the load it carries. On the other hand, when cables are used as transmission lines and guys for radio antennas and derricks, the cable weight may become important and must be included in the structural analysis.

Three cases will be considered in the analysis that follows. In each case we will make the assumption that the cable is *perfectly flexible* and *inextensible*. Due to its flexibility, the cable offers no resistance to bending, and therefore, the tensile force acting in the cable is always tangent to the cable at points along its length. Being inextensible, the cable has a constant length both before and after the load is applied. As a result, once the load is applied, the geometry of the cable remains unchanged, and the cable or a segment of it can be treated as a rigid body.

**Cable Subjected to Concentrated Loads.** When a cable of negligible weight supports several concentrated loads, the cable takes the form of several straight-line segments, each of which is subjected to a constant tensile force. Consider, for example, the cable shown in Fig. 7–18, where the distances  $h$ ,  $L_1$ ,  $L_2$ , and  $L_3$  and the loads  $\mathbf{P}_1$  and  $\mathbf{P}_2$  are known. The problem here is to determine the *nine unknowns* consisting of the tension in each of the *three* segments, the *four* components of reaction at  $A$  and  $B$ , and the *two* sags  $y_C$  and  $y_D$  at points  $C$  and  $D$ . For the solution we can write *two* equations of force equilibrium at each of points  $A$ ,  $B$ ,  $C$ , and  $D$ . This results in a total of *eight equations*.<sup>\*</sup> To complete the solution, we need to know something about the geometry of the cable in order to obtain the necessary ninth equation. For example, if the cable's total *length*  $L$  is specified, then the Pythagorean theorem can be used to relate each of the three segmental lengths, written in terms of  $h$ ,  $y_C$ ,  $y_D$ ,  $L_1$ ,  $L_2$ , and  $L_3$ , to the total length  $L$ . Unfortunately, this type of problem cannot be solved easily by hand. Another possibility, however, is to specify one of the sags, either  $y_C$  or  $y_D$ , instead of the cable length. By doing this, the equilibrium equations are then sufficient for obtaining the unknown forces and the remaining sag. Once the sag at each point of loading is obtained, the length of the cable can then be determined by trigonometry. The following example illustrates a procedure for performing the equilibrium analysis for a problem of this type.

<sup>\*</sup>As will be shown in the following example, the eight equilibrium equations *also* can be written for the entire cable, or any part thereof. But *no more than eight* independent equations are available.



Each of the cable segments remains approximately straight as they support the weight of these traffic lights. (© Russell C. Hibbeler)

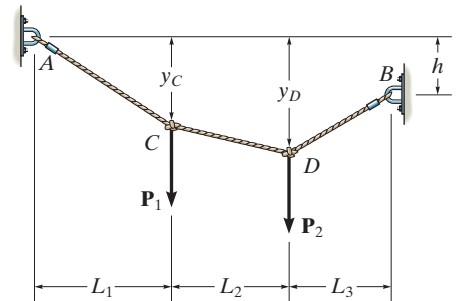
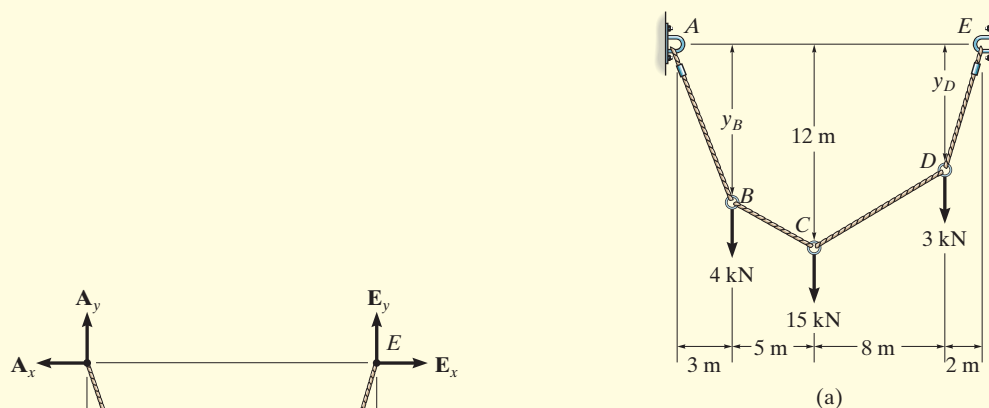


Fig. 7–18



## EXAMPLE 7.11

Determine the tension in each segment of the cable shown in Fig. 7–19a.



## SOLUTION

By inspection, there are four unknown external reactions ( $A_x$ ,  $A_y$ ,  $E_x$ , and  $E_y$ ) and four unknown cable tensions, one in each cable segment. These eight unknowns along with the two unknown sags  $y_B$  and  $y_D$  can be determined from *ten* available equilibrium equations. One method is to apply the force equations of equilibrium ( $\sum F_x = 0$ ,  $\sum F_y = 0$ ) to each of the five points A through E. Here, however, we will take a more direct approach.

Consider the free-body diagram for the entire cable, Fig. 7–19b. Thus,

$$\pm \sum F_x = 0;$$

$$-A_x + E_x = 0$$

$$\zeta + \sum M_E = 0;$$

$$-A_y(18 \text{ m}) + 4 \text{ kN}(15 \text{ m}) + 15 \text{ kN}(10 \text{ m}) + 3 \text{ kN}(2 \text{ m}) = 0$$

$$A_y = 12 \text{ kN}$$

$$+\uparrow \sum F_y = 0;$$

$$12 \text{ kN} - 4 \text{ kN} - 15 \text{ kN} - 3 \text{ kN} + E_y = 0$$

$$E_y = 10 \text{ kN}$$

Since the sag  $y_C = 12 \text{ m}$  is known, we will now consider the leftmost section, which cuts cable BC, Fig. 7–19c.

$$\zeta + \sum M_C = 0; A_x(12 \text{ m}) - 12 \text{ kN}(8 \text{ m}) + 4 \text{ kN}(5 \text{ m}) = 0$$

$$A_x = E_x = 6.33 \text{ kN}$$

$$\pm \sum F_x = 0;$$

$$T_{BC} \cos \theta_{BC} - 6.33 \text{ kN} = 0$$

$$+\uparrow \sum F_y = 0;$$

$$12 \text{ kN} - 4 \text{ kN} - T_{BC} \sin \theta_{BC} = 0$$

Thus,

$$\theta_{BC} = 51.6^\circ$$

$$T_{BC} = 10.2 \text{ kN}$$

*Ans.*

Fig. 7–19

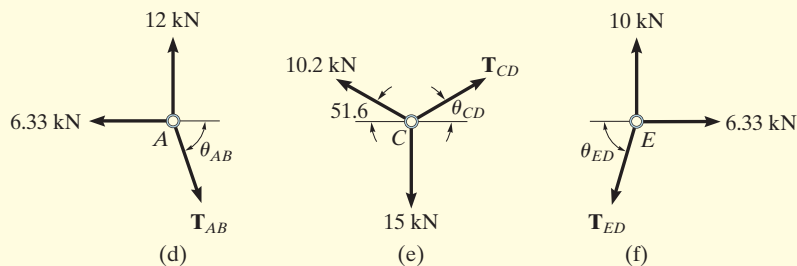


Fig. 7-19 (cont.)

Proceeding now to analyze the equilibrium of points  $A$ ,  $C$ , and  $E$  in sequence, we have

**Point A.** (Fig. 7-19d).

$$\begin{aligned}\pm \Sigma F_x &= 0; & T_{AB} \cos \theta_{AB} - 6.33 \text{ kN} &= 0 \\ + \uparrow \Sigma F_y &= 0; & -T_{AB} \sin \theta_{AB} + 12 \text{ kN} &= 0 \\ & & \theta_{AB} &= 62.2^\circ \\ & & T_{AB} &= 13.6 \text{ kN} \quad \text{Ans.}\end{aligned}$$

**Point C.** (Fig. 7-19e).

$$\begin{aligned}\pm \Sigma F_x &= 0; & T_{CD} \cos \theta_{CD} - 10.2 \cos 51.6^\circ \text{ kN} &= 0 \\ + \uparrow \Sigma F_y &= 0; & T_{CD} \sin \theta_{CD} + 10.2 \sin 51.6^\circ \text{ kN} - 15 \text{ kN} &= 0 \\ & & \theta_{CD} &= 47.9^\circ \\ & & T_{CD} &= 9.44 \text{ kN} \quad \text{Ans.}\end{aligned}$$

**Point E.** (Fig. 7-19f).

$$\begin{aligned}\pm \Sigma F_x &= 0; & 6.33 \text{ kN} - T_{ED} \cos \theta_{ED} &= 0 \\ + \uparrow \Sigma F_y &= 0; & 10 \text{ kN} - T_{ED} \sin \theta_{ED} &= 0 \\ & & \theta_{ED} &= 57.7^\circ \\ & & T_{ED} &= 11.8 \text{ kN} \quad \text{Ans.}\end{aligned}$$

**NOTE:** By comparison, the maximum cable tension is in segment  $AB$  since this segment has the greatest slope ( $\theta$ ) and it is required that for any cable segment the horizontal component  $T \cos \theta = A_x = E_x$  (a constant). Also, since the slope angles that the cable segments make with the horizontal have now been determined, it is possible to determine the sags  $y_B$  and  $y_D$ , Fig. 7-19a, using trigonometry.



The cable and suspenders are used to support the uniform load of a gas pipe which crosses the river. (© Russell C. Hibbeler)

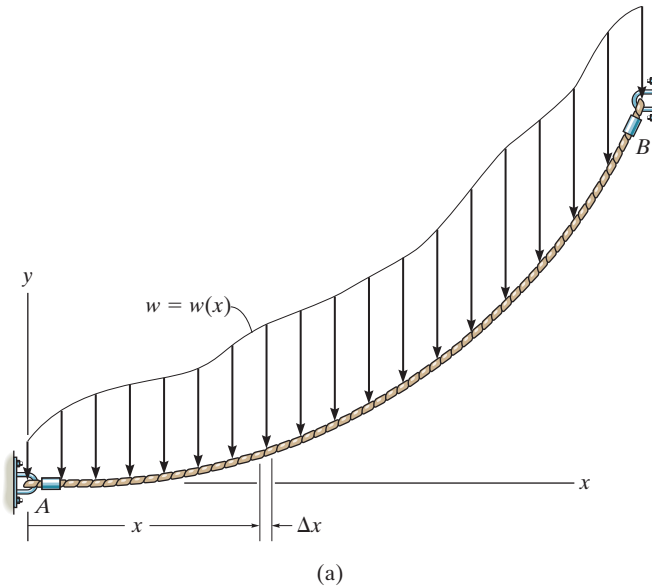


Fig. 7-20

**Cable Subjected to a Distributed Load.** Let us now consider the weightless cable shown in Fig. 7-20a, which is subjected to a distributed loading  $w = w(x)$  that is *measured in the  $x$  direction*. The free-body diagram of a small segment of the cable having a length  $\Delta s$  is shown in Fig. 7-20b. Since the tensile force changes in both magnitude and direction along the cable's length, we will denote this change on the free-body diagram by  $\Delta T$ . Finally, the distributed load is represented by its resultant force  $w(x)(\Delta x)$ , which acts at a fractional distance  $k(\Delta x)$  from point  $O$ , where  $0 < k < 1$ . Applying the equations of equilibrium, we have

$$\begin{aligned} \rightarrow \Sigma F_x &= 0; & -T \cos \theta + (T + \Delta T) \cos(\theta + \Delta \theta) &= 0 \\ + \uparrow \Sigma F_y &= 0; & -T \sin \theta - w(x)(\Delta x) + (T + \Delta T) \sin(\theta + \Delta \theta) &= 0 \\ \zeta + \Sigma M_O &= 0; & w(x)(\Delta x)k(\Delta x) - T \cos \theta \Delta y + T \sin \theta \Delta x &= 0 \end{aligned}$$

Dividing each of these equations by  $\Delta x$  and taking the limit as  $\Delta x \rightarrow 0$ , and therefore  $\Delta y \rightarrow 0$ ,  $\Delta \theta \rightarrow 0$ , and  $\Delta T \rightarrow 0$ , we obtain

$$\frac{d(T \cos \theta)}{dx} = 0 \quad (7-7)$$

$$\frac{d(T \sin \theta)}{dx} - w(x) = 0 \quad (7-8)$$

$$\frac{dy}{dx} = \tan \theta \quad (7-9)$$

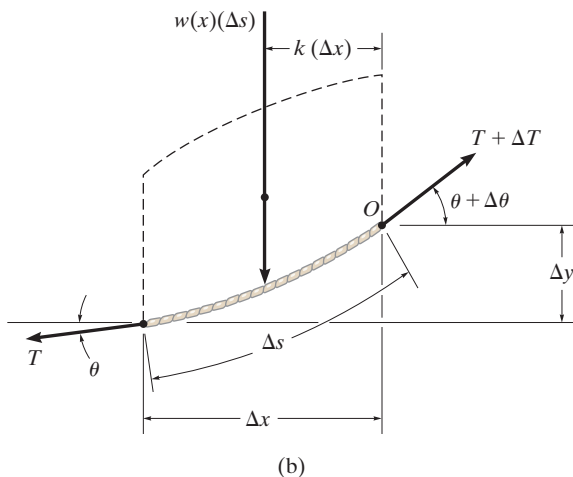


Fig. 7-20 (cont.)

Integrating Eq. 7-7, we have

$$T \cos \theta = \text{constant} = F_H \quad (7-10)$$

where  $F_H$  represents the horizontal component of tensile force at *any* point along the cable.

Integrating Eq. 7-8 gives

$$T \sin \theta = \int w(x) dx \quad (7-11)$$

Dividing Eq. 7-11 by Eq. 7-10 eliminates  $T$ . Then, using Eq. 7-9, we can obtain the slope of the cable.

$$\tan \theta = \frac{dy}{dx} = \frac{1}{F_H} \int w(x) dx$$

Performing a second integration yields

$$y = \frac{1}{F_H} \int \left( \int w(x) dx \right) dx \quad (7-12)$$

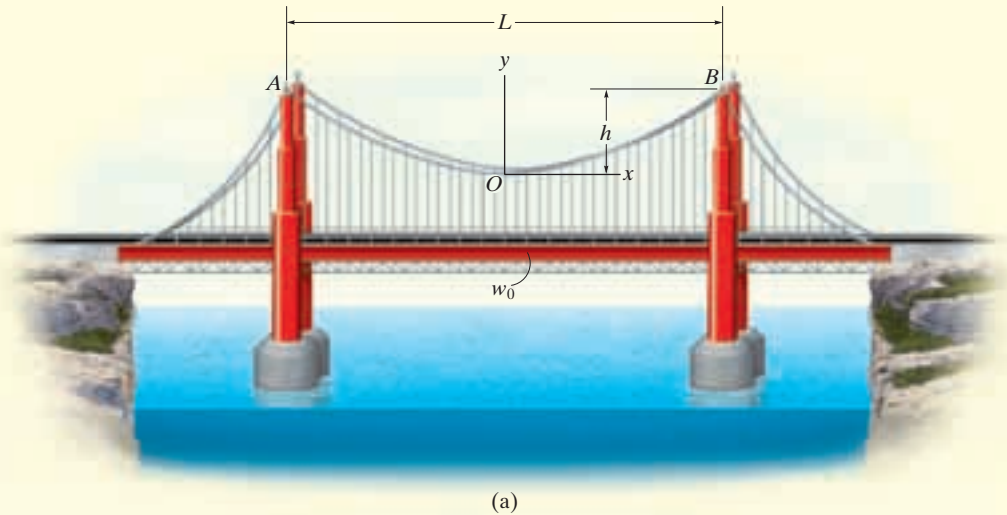
This equation is used to determine the curve for the cable,  $y = f(x)$ . The horizontal force component  $F_H$  and the additional two constants, say  $C_1$  and  $C_2$ , resulting from the integration are determined by applying the boundary conditions for the curve.



The cables of the suspension bridge exert very large forces on the tower and the foundation block which have to be accounted for in their design. (© Russell C. Hibbeler)

**EXAMPLE 7.12**

The cable of a suspension bridge supports half of the uniform road surface between the two towers at  $A$  and  $B$ , Fig. 7-21a. If this distributed loading is  $w_0$ , determine the maximum force developed in the cable and the cable's required length. The span length  $L$  and sag  $h$  are known.

**Fig. 7-21****SOLUTION**

We can determine the unknowns in the problem by first finding the equation of the curve that defines the shape of the cable using Eq. 7-12. For reasons of symmetry, the origin of coordinates has been placed at the cable's center. Noting that  $w(x) = w_0$ , we have

$$y = \frac{1}{F_H} \int \left( \int w_0 dx \right) dx$$

Performing the two integrations gives

$$y = \frac{1}{F_H} \left( \frac{w_0 x^2}{2} + C_1 x + C_2 \right) \quad (1)$$

The constants of integration may be determined using the boundary conditions  $y = 0$  at  $x = 0$  and  $dy/dx = 0$  at  $x = 0$ . Substituting into Eq. 1 and its derivative yields  $C_1 = C_2 = 0$ . The equation of the curve then becomes

$$y = \frac{w_0}{2F_H} x^2 \quad (2)$$

This is the equation of a *parabola*. The constant  $F_H$  may be obtained using the boundary condition  $y = h$  at  $x = L/2$ . Thus,

$$F_H = \frac{w_0 L^2}{8h} \quad (3)$$

Therefore, Eq. 2 becomes

$$y = \frac{4h}{L^2} x^2 \quad (4)$$

Since  $F_H$  is known, the tension in the cable may now be determined using Eq. 7-10, written as  $T = F_H / \cos \theta$ . For  $0 \leq \theta < \pi/2$ , the maximum tension will occur when  $\theta$  is *maximum*, i.e., at point *B*, Fig. 7-21*a*. From Eq. 2, the slope at this point is

$$\left. \frac{dy}{dx} \right|_{x=L/2} = \tan \theta_{\max} = \left. \frac{w_0}{F_H} x \right|_{x=L/2}$$

or

$$\theta_{\max} = \tan^{-1} \left( \frac{w_0 L}{2F_H} \right) \quad (5)$$

Therefore,

$$T_{\max} = \frac{F_H}{\cos(\theta_{\max})} \quad (6)$$

Using the triangular relationship shown in Fig. 7-21*b*, which is based on Eq. 5, Eq. 6 may be written as

$$T_{\max} = \frac{\sqrt{4F_H^2 + w_0^2 L^2}}{2}$$

Substituting Eq. 3 into the above equation yields

$$T_{\max} = \frac{w_0 L}{2} \sqrt{1 + \left( \frac{L}{4h} \right)^2} \quad \text{Ans.}$$

For a differential segment of cable length  $ds$ , we can write

$$ds = \sqrt{(dx)^2 + (dy)^2} = \sqrt{1 + \left( \frac{dy}{dx} \right)^2} dx$$

Hence, the total length of the cable can be determined by integration. Using Eq. 4, we have

$$\mathcal{L} = \int ds = 2 \int_0^{L/2} \sqrt{1 + \left( \frac{8h}{L^2} x \right)^2} dx \quad (7)$$

Integrating yields

$$\mathcal{L} = \frac{L}{2} \left[ \sqrt{1 + \left( \frac{4h}{L} \right)^2} + \frac{L}{4h} \sinh^{-1} \left( \frac{4h}{L} \right) \right] \quad \text{Ans.}$$

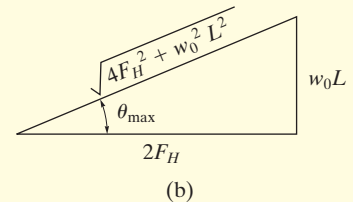


Fig. 7-21 (cont.)

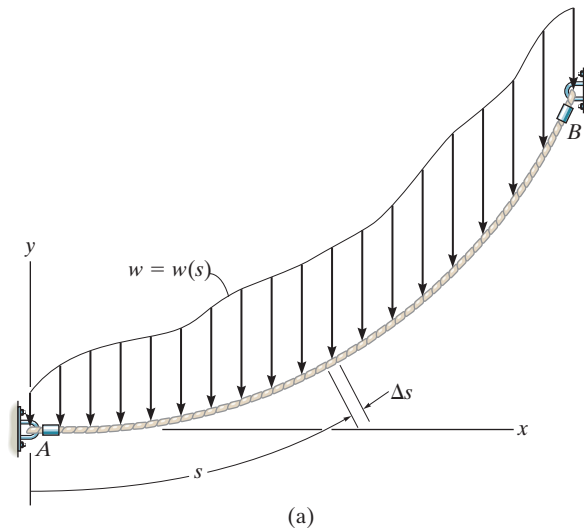


Fig. 7-22

**Cable Subjected to Its Own Weight.** When the weight of a cable becomes important in the force analysis, the loading function along the cable will be a function of the arc length  $s$  rather than the projected length  $x$ . To analyze this problem, we will consider a generalized loading function  $w = w(s)$  acting along the cable, as shown in Fig. 7-22a. The free-body diagram for a small segment  $\Delta s$  of the cable is shown in Fig. 7-22b. Applying the equilibrium equations to the force system on this diagram, one obtains relationships identical to those given by Eqs. 7-7 through 7-9, but with  $s$  replacing  $x$  in Eqs. 7-7 and 7-8. Therefore, we can show that

$$T \cos \theta = F_H$$

$$T \sin \theta = \int w(s) ds \quad (7-13)$$

$$\frac{dy}{dx} = \frac{1}{F_H} \int w(s) ds \quad (7-14)$$

To perform a direct integration of Eq. 7-14, it is necessary to replace  $dy/dx$  by  $ds/dx$ . Since

$$ds = \sqrt{dx^2 + dy^2}$$

then

$$\frac{dy}{dx} = \sqrt{\left(\frac{ds}{dx}\right)^2 - 1}$$

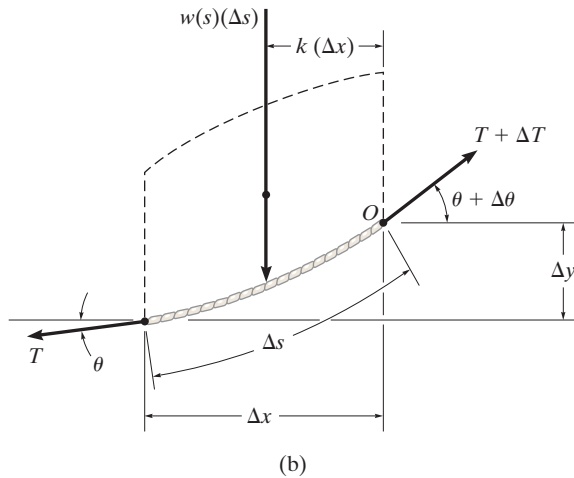


Fig. 7-22 (cont.)

Therefore,

$$\frac{ds}{dx} = \left[ 1 + \frac{1}{F_H^2} \left( \int w(s) ds \right)^2 \right]^{1/2}$$

Separating the variables and integrating we obtain

$$x = \int \frac{ds}{\left[ 1 + \frac{1}{F_H^2} \left( \int w(s) ds \right)^2 \right]^{1/2}} \quad (7-15)$$

The two constants of integration, say  $C_1$  and  $C_2$ , are found using the boundary conditions for the curve.



Electrical transmission towers must be designed to support the weights of the suspended power lines. The weight and length of the cables can be determined since they each form a catenary curve. (© Russell C. Hibbeler)

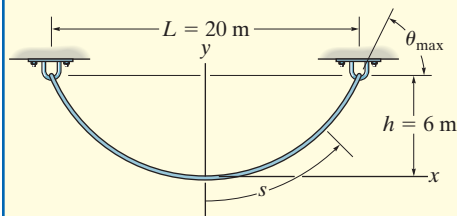


**EXAMPLE 7.13**

Determine the deflection curve, the length, and the maximum tension in the uniform cable shown in Fig. 7-23. The cable has a weight per unit length of  $w_0 = 5 \text{ N/m}$ .

**SOLUTION**

For reasons of symmetry, the origin of coordinates is located at the center of the cable. The deflection curve is expressed as  $y = f(x)$ . We can determine it by first applying Eq. 7-15, where  $w(s) = w_0$ .

**Fig. 7-23**

$$x = \int \frac{ds}{\left[1 + (1/F_H^2) \left(\int w_0 ds\right)^2\right]^{1/2}}$$

Integrating the term under the integral sign in the denominator, we have

$$x = \int \frac{ds}{[1 + (1/F_H^2)(w_0 s + C_1)^2]^{1/2}}$$

Substituting  $u = (1/F_H)(w_0 s + C_1)$  so that  $du = (w_0/F_H) ds$ , a second integration yields

$$x = \frac{F_H}{w_0} (\sinh^{-1} u + C_2)$$

or

$$x = \frac{F_H}{w_0} \left\{ \sinh^{-1} \left[ \frac{1}{F_H} (w_0 s + C_1) \right] + C_2 \right\} \quad (1)$$

To evaluate the constants note that, from Eq. 7-14,

$$\frac{dy}{dx} = \frac{1}{F_H} \int w_0 ds \quad \text{or} \quad \frac{dy}{dx} = \frac{1}{F_H} (w_0 s + C_1)$$

Since  $dy/dx = 0$  at  $s = 0$ , then  $C_1 = 0$ . Thus,

$$\frac{dy}{dx} = \frac{w_0 s}{F_H} \quad (2)$$

The constant  $C_2$  may be evaluated by using the condition  $s = 0$  at  $x = 0$  in Eq. 1, in which case  $C_2 = 0$ . To obtain the deflection curve, solve for  $s$  in Eq. 1, which yields

$$s = \frac{F_H}{w_0} \sinh \left( \frac{w_0}{F_H} x \right) \quad (3)$$

Now substitute into Eq. 2, in which case

$$\frac{dy}{dx} = \sinh \left( \frac{w_0}{F_H} x \right)$$

Hence,

$$y = \frac{F_H}{w_0} \cosh\left(\frac{w_0}{F_H}x\right) + C_3$$

If the boundary condition  $y = 0$  at  $x = 0$  is applied, the constant  $C_3 = -F_H/w_0$ , and therefore the deflection curve becomes

$$y = \frac{F_H}{w_0} \left[ \cosh\left(\frac{w_0}{F_H}x\right) - 1 \right] \quad (4)$$

This equation defines the shape of a **catenary curve**. The constant  $F_H$  is obtained by using the boundary condition that  $y = h$  at  $x = L/2$ , in which case

$$h = \frac{F_H}{w_0} \left[ \cosh\left(\frac{w_0 L}{2F_H}\right) - 1 \right] \quad (5)$$

Since  $w_0 = 5 \text{ N/m}$ ,  $h = 6 \text{ m}$ , and  $L = 20 \text{ m}$ , Eqs. 4 and 5 become

$$y = \frac{F_H}{5 \text{ N/m}} \left[ \cosh\left(\frac{5 \text{ N/m}}{F_H}x\right) - 1 \right] \quad (6)$$

$$6 \text{ m} = \frac{F_H}{5 \text{ N/m}} \left[ \cosh\left(\frac{50 \text{ N}}{F_H}\right) - 1 \right] \quad (7)$$

Equation 7 can be solved for  $F_H$  by using a trial-and-error procedure. The result is

$$F_H = 45.9 \text{ N}$$

and therefore the deflection curve, Eq. 6, becomes

$$y = 9.19 [\cosh(0.109x) - 1] \text{ m} \quad \text{Ans.}$$

Using Eq. 3, with  $x = 10 \text{ m}$ , the half-length of the cable is

$$\frac{\mathcal{L}}{2} = \frac{45.9 \text{ N}}{5 \text{ N/m}} \sinh\left[\frac{5 \text{ N/m}}{45.9 \text{ N}}(10 \text{ m})\right] = 12.1 \text{ m}$$

Hence,

$$\mathcal{L} = 24.2 \text{ m} \quad \text{Ans.}$$

Since  $T = F_H/\cos \theta$ , the maximum tension occurs when  $\theta$  is maximum, i.e., at  $s = \mathcal{L}/2 = 12.1 \text{ m}$ . Using Eq. 2 yields

$$\begin{aligned} \left. \frac{dy}{dx} \right|_{s=12.1 \text{ m}} &= \tan \theta_{\max} = \frac{5 \text{ N/m}(12.1 \text{ m})}{45.9 \text{ N}} = 1.32 \\ \theta_{\max} &= 52.8^\circ \end{aligned}$$

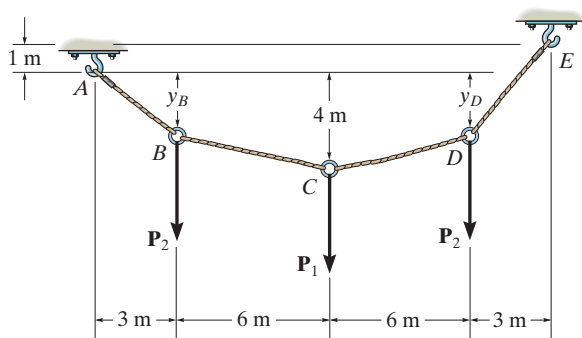
And so,

$$T_{\max} = \frac{F_H}{\cos \theta_{\max}} = \frac{45.9 \text{ N}}{\cos 52.8^\circ} = 75.9 \text{ N} \quad \text{Ans.}$$

## PROBLEMS

**7-94.** The cable supports the three loads shown. Determine the sags  $y_B$  and  $y_D$  of  $B$  and  $D$ . Take  $P_1 = 800$  N,  $P_2 = 500$  N.

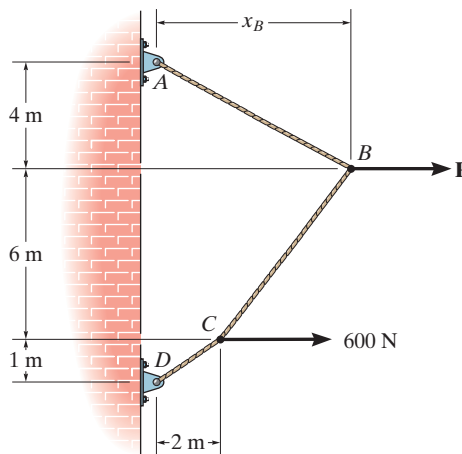
**7-95.** The cable supports the three loads shown. Determine the magnitude of  $P_1$  if  $P_2 = 600$  N and  $y_B = 3$  m. Also find sag  $y_D$ .



Probs. 7-94/95

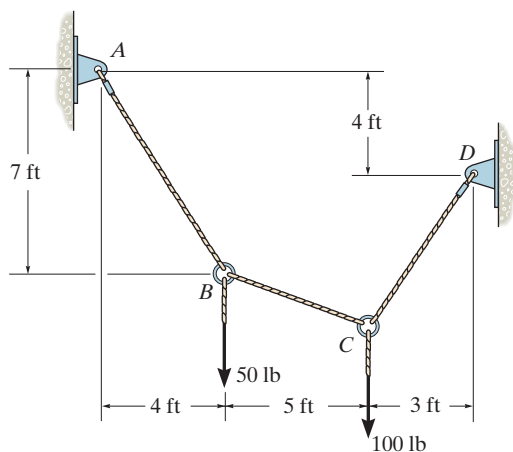
**7-97.** The cable supports the loading shown. Determine the distance  $x_B$  the force at  $B$  acts from  $A$ . Set  $P = 800$  N.

**7-98.** The cable supports the loading shown. Determine the magnitude of the horizontal force  $P$  so that  $x_B = 5$  m.



Probs. 7-97/98

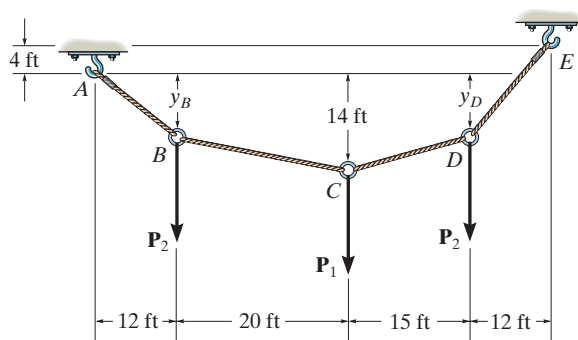
**\*7-96.** Determine the tension in each segment of the cable and the cable's total length.



Prob. 7-96

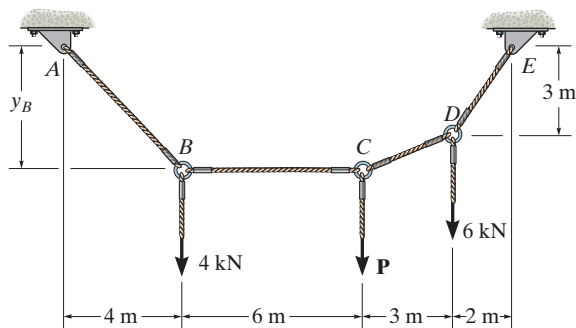
**7-99.** The cable supports the three loads shown. Determine the sags  $y_B$  and  $y_D$  of points  $B$  and  $D$ . Take  $P_1 = 400$  lb,  $P_2 = 250$  lb.

**\*7-100.** The cable supports the three loads shown. Determine the magnitude of  $P_1$  if  $P_2 = 300$  lb and  $y_B = 8$  ft. Also find the sag  $y_D$ .



Probs. 7-99/100

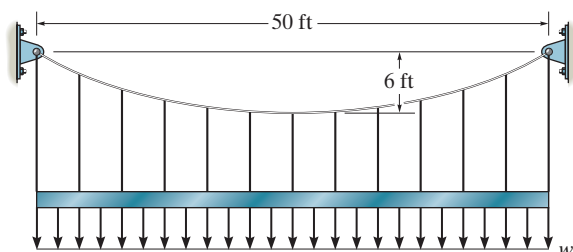
**7-101.** Determine the force  $P$  needed to hold the cable in the position shown, i.e., so segment  $BC$  remains horizontal. Also, compute the sag  $y_B$  and the maximum tension in the cable.



**Prob. 7-101**

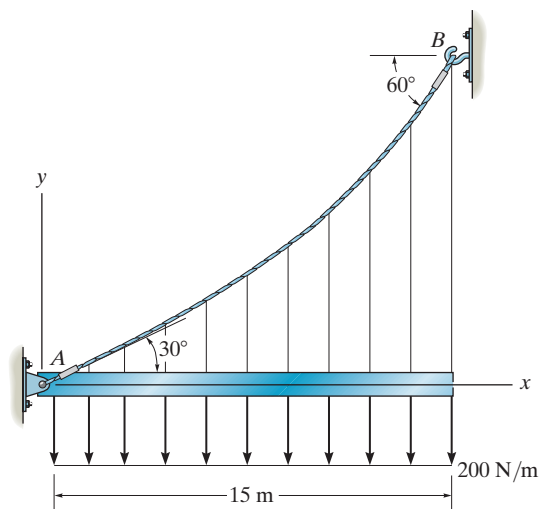
**7-102.** Determine the maximum uniform loading  $w$ , measured in lb/ft, that the cable can support if it is capable of sustaining a maximum tension of 3000 lb before it will break.

**7-103.** The cable is subjected to a uniform loading of  $w = 250$  lb/ft. Determine the maximum and minimum tension in the cable.



**Probs. 7-102/103**

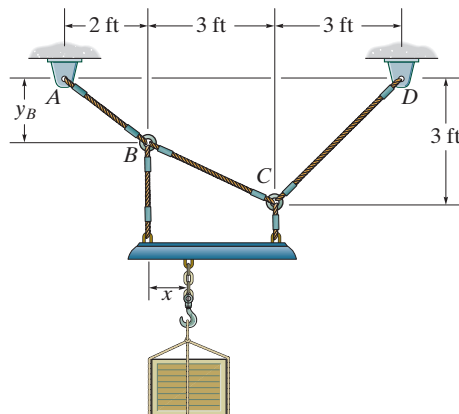
**\*7-104.** The cable  $AB$  is subjected to a uniform loading of  $200$  N/m. If the weight of the cable is neglected and the slope angles at points  $A$  and  $B$  are  $30^\circ$  and  $60^\circ$ , respectively, determine the curve that defines the cable shape and the maximum tension developed in the cable.



**Prob. 7-104**

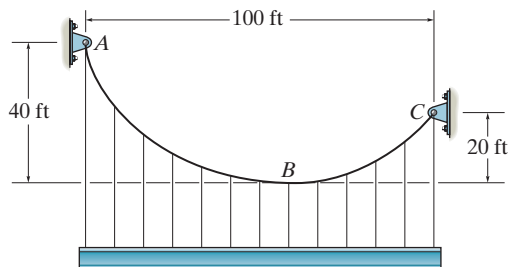
**7-105.** If  $x = 2$  ft and the crate weighs 300 lb, which cable segment  $AB$ ,  $BC$ , or  $CD$  has the greatest tension? What is this force and what is the sag  $y_B$ ?

**7-106.** If  $y_B = 1.5$  ft, determine the largest weight of the crate and its placement  $x$  so that neither cable segment  $AB$ ,  $BC$ , or  $CD$  is subjected to a tension that exceeds 200 lb.



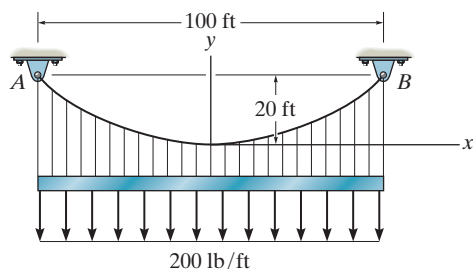
**Probs. 7-105/106**

**7-107.** The cable supports a girder which weighs 850 lb/ft. Determine the tension in the cable at points *A*, *B*, and *C*.



**Prob. 7-107**

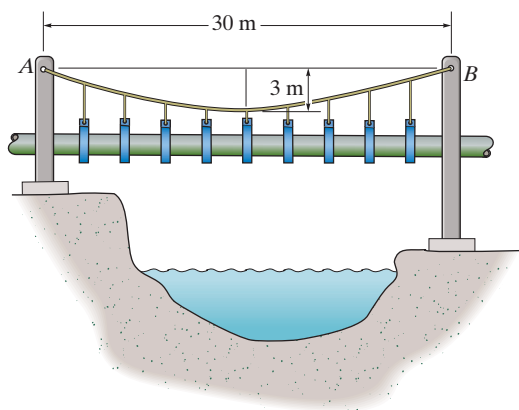
**\*7-108.** The cable is subjected to a uniform loading of  $w = 200$  lb/ft. Determine the maximum and minimum tension in the cable.



**Prob. 7-108**

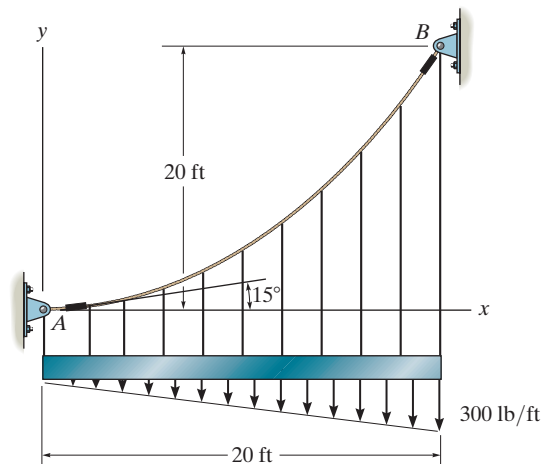
**7-109.** If the pipe has a mass per unit length of 1500 kg/m, determine the maximum tension developed in the cable.

**7-110.** If the pipe has a mass per unit length of 1500 kg/m, determine the minimum tension developed in the cable.



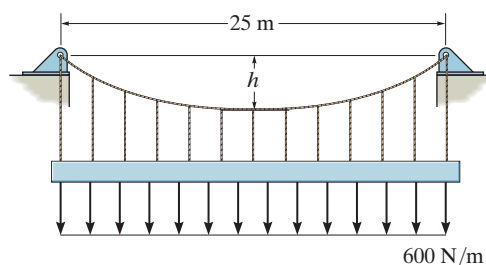
**Probs. 7-109/110**

**7-111.** Determine the maximum tension developed in the cable if it is subjected to the triangular distributed load.



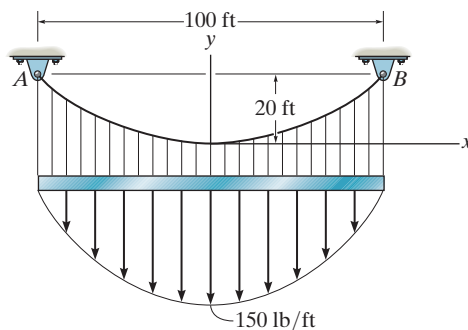
**Prob. 7-111**

**\*7-112.** The cable will break when the maximum tension reaches  $T_{\max} = 10$  kN. Determine the minimum sag  $h$  if it supports the uniform distributed load of  $w = 600$  N/m.



**Prob. 7-112**

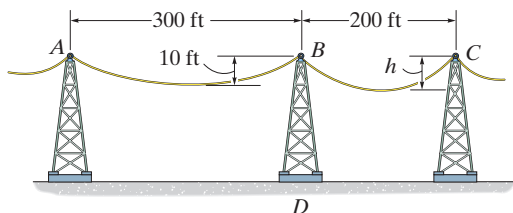
**7-113.** The cable is subjected to the parabolic loading  $w = 150(1 - (x/50)^2)$  lb/ft, where  $x$  is in ft. Determine the equation  $y = f(x)$  which defines the cable shape *AB* and the maximum tension in the cable.



**Prob. 7-113**

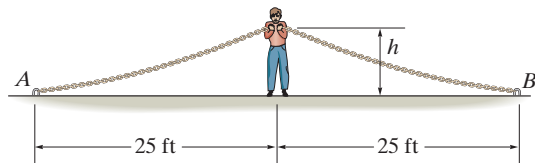
**7-114.** The power transmission cable weighs 10 lb/ft. If the resultant horizontal force on tower  $BD$  is required to be zero, determine the sag  $h$  of cable  $BC$ .

**7-115.** The power transmission cable weighs 10 lb/ft. If  $h = 10$  ft, determine the resultant horizontal and vertical forces the cables exert on tower  $BD$ .



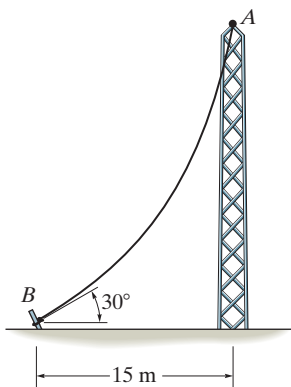
**Probs. 7-114/115**

**\*7-116.** The man picks up the 52-ft chain and holds it just high enough so it is completely off the ground. The chain has points of attachment  $A$  and  $B$  that are 50 ft apart. If the chain has a weight of 3 lb/ft, and the man weighs 150 lb, determine the force he exerts on the ground. Also, how high  $h$  must he lift the chain? *Hint:* The slopes at  $A$  and  $B$  are zero.



**Prob. 7-116**

**7-117.** The cable has a mass of 0.5 kg/m and is 25 m long. Determine the vertical and horizontal components of force it exerts on the top of the tower.



**Prob. 7-117**

**7-118.** A 50-ft cable is suspended between two points a distance of 15 ft apart and at the same elevation. If the minimum tension in the cable is 200 lb, determine the total weight of the cable and the maximum tension developed in the cable.

**7-119.** Show that the deflection curve of the cable discussed in Example 7.13 reduces to Eq. 4 in Example 7.12 when the *hyperbolic cosine function* is expanded in terms of a series and only the first two terms are retained. (The answer indicates that the *catenary* may be replaced by a *parabola* in the analysis of problems in which the sag is small. In this case, the cable weight is assumed to be uniformly distributed along the horizontal.)

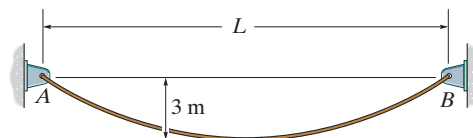
**\*7-120.** A telephone line (cable) stretches between two points which are 150 ft apart and at the same elevation. The line sags 5 ft and the cable has a weight of 0.3 lb/ft. Determine the length of the cable and the maximum tension in the cable.

**7-121.** A cable has a weight of 2 lb/ft. If it can span 100 ft and has a sag of 12 ft, determine the length of the cable. The ends of the cable are supported from the same elevation.

**7-122.** A cable has a weight of 3 lb/ft and is supported at points that are 500 ft apart and at the same elevation. If it has a length of 600 ft, determine the sag.

**7-123.** A cable has a weight of 5 lb/ft. If it can span 300 ft and has a sag of 15 ft, determine the length of the cable. The ends of the cable are supported at the same elevation.

**\*7-124.** The 10 kg/m cable is suspended between the supports  $A$  and  $B$ . If the cable can sustain a maximum tension of 1.5 kN and the maximum sag is 3 m, determine the maximum distance  $L$  between the supports.



**Prob. 7-124**

## CHAPTER REVIEW

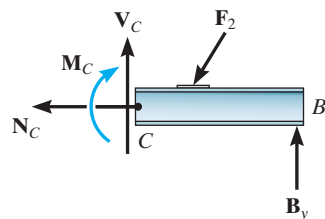
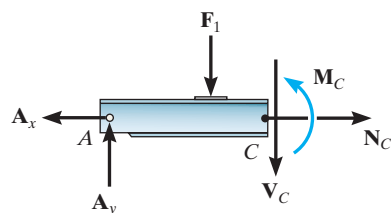
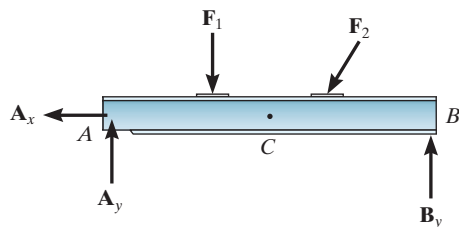
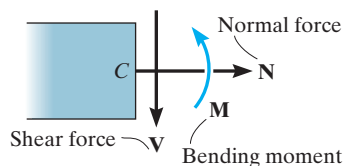
### Internal Loadings

If a coplanar force system acts on a member, then in general a resultant internal *normal force*  $\mathbf{N}$ , *shear force*  $\mathbf{V}$ , and *bending moment*  $\mathbf{M}$  will act at any cross section along the member. For two-dimensional problems the positive directions of these loadings are shown in the figure.

The resultant internal normal force, shear force, and bending moment are determined using the method of sections. To find them, the member is sectioned at the point  $C$  where the internal loadings are to be determined. A free-body diagram of one of the sectioned parts is then drawn and the internal loadings are shown in their positive directions.

The resultant normal force is determined by summing forces normal to the cross section. The resultant shear force is found by summing forces tangent to the cross section, and the resultant bending moment is found by summing moments about the geometric center or centroid of the cross-sectional area.

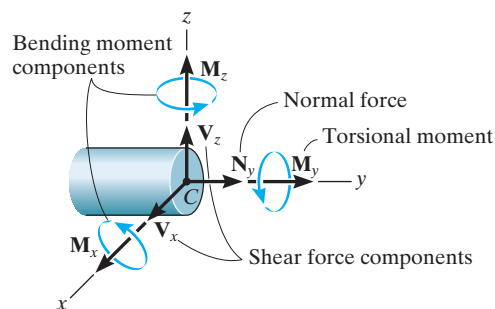
If the member is subjected to a three-dimensional loading, then, in general, a *torsional moment* will also act on the cross section. It can be determined by summing moments about an axis that is perpendicular to the cross section and passes through its centroid.



$$\Sigma F_x = 0$$

$$\Sigma F_y = 0$$

$$\Sigma M_C = 0$$



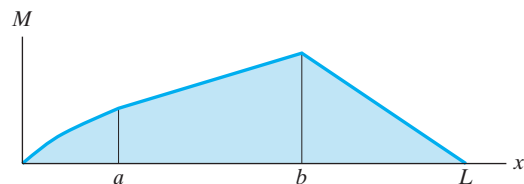
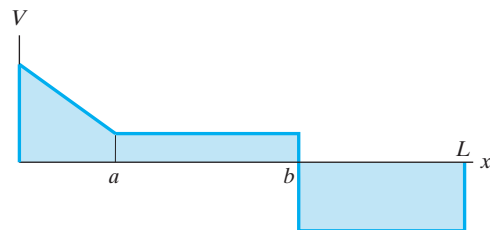
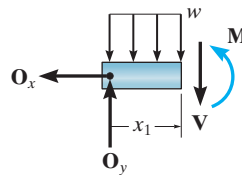
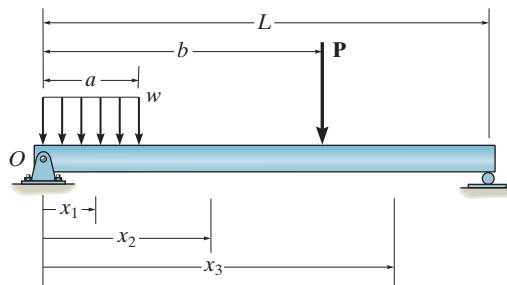
### Shear and Moment Diagrams

To construct the shear and moment diagrams for a member, it is necessary to section the member at an arbitrary point, located a distance  $x$  from the left end.

If the external loading consists of changes in the distributed load, or a series of concentrated forces and couple moments act on the member, then different expressions for  $V$  and  $M$  must be determined within regions between any load discontinuities.

The unknown shear and moment are indicated on the cross section in the positive direction according to the established sign convention, and then the internal shear and moment are determined as functions of  $x$ .

Each of the functions of the shear and moment is then plotted to create the shear and moment diagrams.





**Relations between Shear and Moment**

It is possible to plot the shear and moment diagrams quickly by using differential relationships that exist between the distributed loading  $w$ ,  $V$  and  $M$ .

The slope of the shear diagram is equal to the distributed loading at any point. The slope is positive if the distributed load acts upward, and vice-versa.

The slope of the moment diagram is equal to the shear at any point. The slope is positive if the shear is positive, or vice-versa.

The change in shear between any two points is equal to the area under the distributed loading between the points.

The change in the moment is equal to the area under the shear diagram between the points.

$$\frac{dV}{dx} = w$$

$$\frac{dM}{dx} = V$$

$$\Delta V = \int w \, dx$$

$$\Delta M = \int V \, dx$$

**Cables**

When a flexible and inextensible cable is subjected to a series of concentrated forces, then the analysis of the cable can be performed by using the equations of equilibrium applied to free-body diagrams of either segments or points of application of the loading.

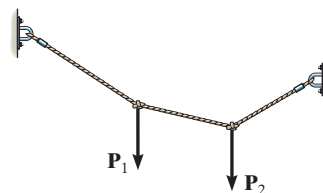
If external distributed loads or the weight of the cable are to be considered, then the shape of the cable must be determined by first analyzing the forces on a differential segment of the cable and then integrating this result. The two constants, say  $C_1$  and  $C_2$ , resulting from the integration are determined by applying the boundary conditions for the cable.

$$y = \frac{1}{F_H} \int \left( \int w(x) \, dx \right) dx$$

Distributed load

$$x = \int \frac{ds}{\left[ 1 + \frac{1}{F_H^2} \left( \int w(s) \, ds \right)^2 \right]^{1/2}}$$

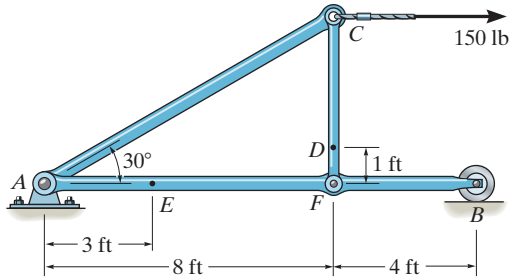
Cable weight



## REVIEW PROBLEMS

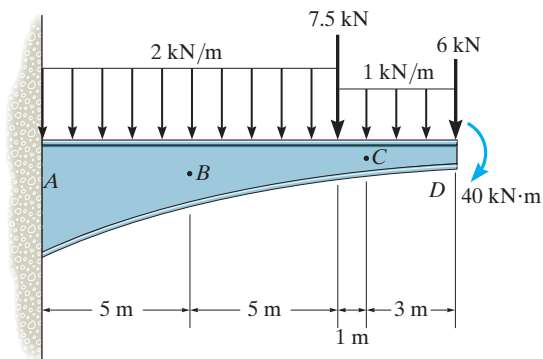
*All problem solutions must include FBDs.*

**R7-1.** Determine the internal normal force, shear force, and moment at points *D* and *E* of the frame.



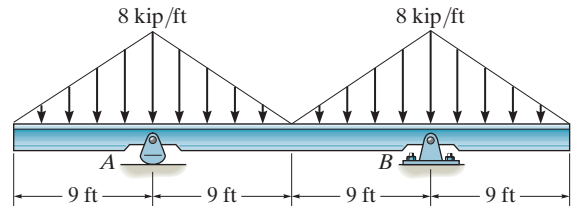
**Prob. R7-1**

**R7-2.** Determine the normal force, shear force, and moment at points *B* and *C* of the beam.



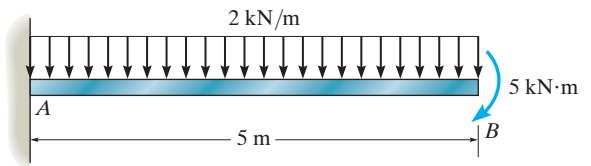
**Prob. R7-2**

**R7-3.** Draw the shear and moment diagrams for the beam.



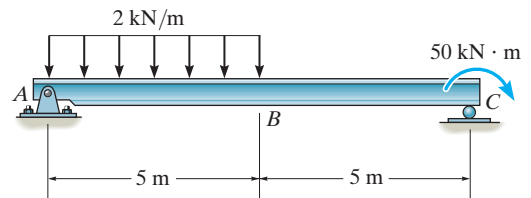
**Prob. R7-3**

**R7-4.** Draw the shear and moment diagrams for the beam.



**Prob. R7-4**

**R7-5.** Draw the shear and moment diagrams for the beam.



**Prob. R7-5**

**R7-6.** A chain is suspended between points at the same elevation and spaced a distance of 60 ft apart. If it has a weight per unit length of 0.5 lb/ft and the sag is 3 ft, determine the maximum tension in the chain.

# Chapter 8



(© Pavel Polkovnikov/Shutterstock)

The effective design of this brake requires that it resist the frictional forces developed between it and the wheel. In this chapter we will study dry friction, and show how to analyze friction forces for various engineering applications.

# Friction

## CHAPTER OBJECTIVES

- To introduce the concept of dry friction and show how to analyze the equilibrium of rigid bodies subjected to this force.
- To present specific applications of frictional force analysis on wedges, screws, belts, and bearings.
- To investigate the concept of rolling resistance.

## 8.1 Characteristics of Dry Friction

*Friction* is a force that resists the movement of two contacting surfaces that slide relative to one another. This force always acts *tangent* to the surface at the points of contact and is directed so as to oppose the possible or existing motion between the surfaces.

In this chapter, we will study the effects of ***dry friction***, which is sometimes called *Coulomb friction* since its characteristics were studied extensively by the French physicist Charles-Augustin de Coulomb in 1781. Dry friction occurs between the contacting surfaces of bodies when there is no lubricating fluid.\*



The heat generated by the abrasive action of friction can be noticed when using this grinder to sharpen a metal blade. (© Russell C. Hibbeler)

\*Another type of friction, called fluid friction, is studied in fluid mechanics.

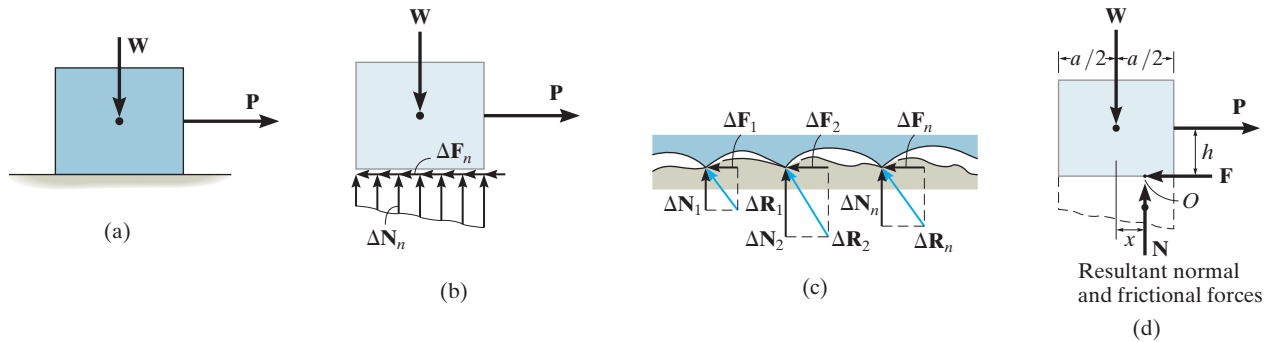
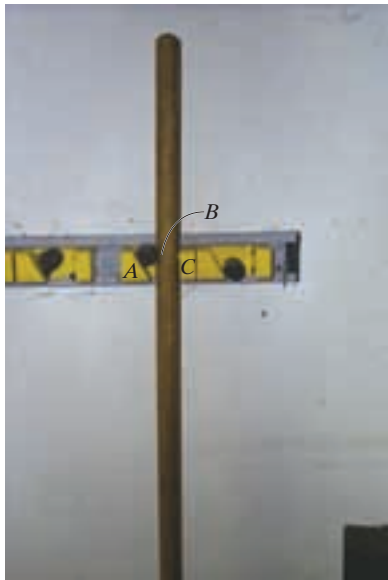


Fig. 8-1



Regardless of the weight of the rake or shovel that is suspended, the device has been designed so that the small roller holds the handle in equilibrium due to frictional forces that develop at the points of contact, *A, B, C*. (© Russell C. Hibbeler)

**Theory of Dry Friction.** The theory of dry friction can be explained by considering the effects caused by pulling horizontally on a block of uniform weight  $W$  which is resting on a rough horizontal surface that is *nonrigid or deformable*, Fig. 8-1*a*. The upper portion of the block, however, can be considered rigid. As shown on the free-body diagram of the block, Fig. 8-1*b*, the floor exerts an uneven *distribution* of both *normal force*  $\Delta N_n$  and *frictional force*  $\Delta F_n$  along the contacting surface. For equilibrium, the normal forces must act *upward* to balance the block's weight  $W$ , and the frictional forces act to the left to prevent the applied force  $P$  from moving the block to the right. Close examination of the contacting surfaces between the floor and block reveals how these frictional and normal forces develop, Fig. 8-1*c*. It can be seen that many microscopic irregularities exist between the two surfaces and, as a result, reactive forces  $\Delta R_n$  are developed at each point of contact.\* As shown, each reactive force contributes both a frictional component  $\Delta F_n$  and a normal component  $\Delta N_n$ .

**Equilibrium.** The effect of the *distributed* normal and frictional loadings is indicated by their *resultants*  $N$  and  $F$  on the free-body diagram, Fig. 8-1*d*. Notice that  $N$  acts a distance  $x$  to the right of the line of action of  $W$ , Fig. 8-1*d*. This location, which coincides with the centroid or geometric center of the normal force distribution in Fig. 8-1*b*, is necessary in order to balance the “tipping effect” caused by  $P$ . For example, if  $P$  is applied at a height  $h$  from the surface, Fig. 8-1*d*, then moment equilibrium about point  $O$  is satisfied if  $Wx = Ph$  or  $x = Ph/W$ .

\*Besides mechanical interactions as explained here, which is referred to as a classical approach, a detailed treatment of the nature of frictional forces must also include the effects of temperature, density, cleanliness, and atomic or molecular attraction between the contacting surfaces. See J. Krim, *Scientific American*, October, 1996.

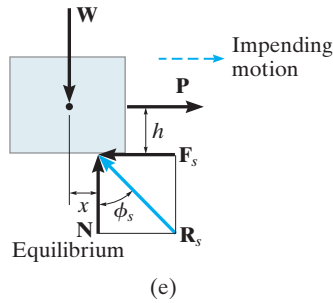


Fig. 8–1 (cont.)

**Impending Motion.** In cases where the surfaces of contact are rather “slippery,” the frictional force  $\mathbf{F}$  may *not* be great enough to balance  $\mathbf{P}$ , and consequently the block will tend to slip. In other words, as  $P$  is slowly increased,  $F$  correspondingly increases until it attains a certain *maximum value*  $F_s$ , called the *limiting static frictional force*, Fig. 8–1*e*. When this value is reached, the block is in *unstable equilibrium* since any further increase in  $P$  will cause the block to move. Experimentally, it has been determined that this limiting static frictional force  $F_s$  is *directly proportional* to the resultant normal force  $N$ . Expressed mathematically,

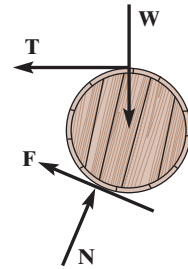
$$F_s = \mu_s N \quad (8-1)$$

where the constant of proportionality,  $\mu_s$  (mu “sub”  $s$ ), is called the **coefficient of static friction**.

Thus, when the block is on the *verge of sliding*, the normal force  $\mathbf{N}$  and frictional force  $\mathbf{F}_s$  combine to create a resultant  $\mathbf{R}_s$ , Fig. 8–1*e*. The angle  $\phi_s$  (phi “sub”  $s$ ) that  $\mathbf{R}_s$  makes with  $\mathbf{N}$  is called the **angle of static friction**. From the figure,

$$\phi_s = \tan^{-1}\left(\frac{F_s}{N}\right) = \tan^{-1}\left(\frac{\mu_s N}{N}\right) = \tan^{-1} \mu_s$$

Typical values for  $\mu_s$  are given in Table 8–1. Note that these values can vary since experimental testing was done under variable conditions of roughness and cleanliness of the contacting surfaces. For applications, therefore, it is important that both caution and judgment be exercised when selecting a coefficient of friction for a given set of conditions. When a more accurate calculation of  $F_s$  is required, the coefficient of friction should be determined directly by an experiment that involves the two materials to be used.



Some objects, such as this barrel, may not be on the verge of slipping, and therefore the friction force  $\mathbf{F}$  must be determined strictly from the equations of equilibrium. (© Russell C. Hibbeler)

Table 8–1 Typical Values for  $\mu_s$ 

Contact Materials	Coefficient of Static Friction ( $\mu_s$ )
Metal on ice	0.03–0.05
Wood on wood	0.30–0.70
Leather on wood	0.20–0.50
Leather on metal	0.30–0.60
Copper on copper	0.74–1.21

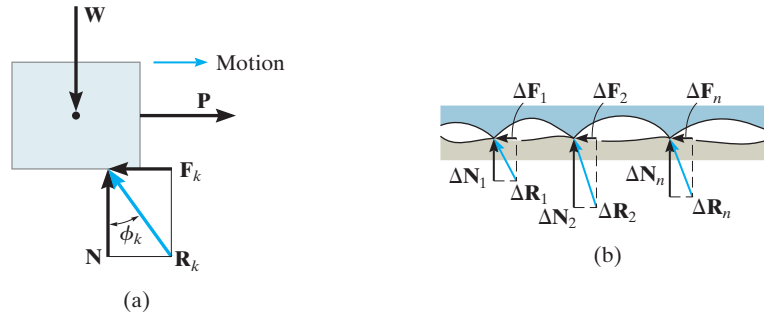


Fig. 8-2

**Motion.** If the magnitude of  $\mathbf{P}$  acting on the block is increased so that it becomes slightly greater than  $F_s$ , the frictional force at the contacting surface will drop to a smaller value  $F_k$ , called the *kinetic frictional force*. The block will begin to slide with increasing speed, Fig. 8-2a. As this occurs, the block will “ride” on top of these peaks at the points of contact, as shown in Fig. 8-2b. The continued breakdown of the surface is the dominant mechanism creating kinetic friction.

Experiments with sliding blocks indicate that the magnitude of the kinetic friction force is directly proportional to the magnitude of the resultant normal force, expressed mathematically as

$$F_k = \mu_k N \quad (8-2)$$

Here the constant of proportionality,  $\mu_k$ , is called the **coefficient of kinetic friction**. Typical values for  $\mu_k$  are approximately 25 percent smaller than those listed in Table 8-1 for  $\mu_s$ .

As shown in Fig. 8-2a, in this case, the resultant force at the surface of contact,  $\mathbf{R}_k$ , has a line of action defined by  $\phi_k$ . This angle is referred to as the **angle of kinetic friction**, where

$$\phi_k = \tan^{-1}\left(\frac{F_k}{N}\right) = \tan^{-1}\left(\frac{\mu_k N}{N}\right) = \tan^{-1} \mu_k$$

By comparison,  $\phi_s \geq \phi_k$ .

The above effects regarding friction can be summarized by referring to the graph in Fig. 8–3, which shows the variation of the frictional force  $F$  versus the applied load  $P$ . Here the frictional force is categorized in three different ways:

- $F$  is a *static frictional force* if equilibrium is maintained.
- $F$  is a *limiting static frictional force*  $F_s$  when it reaches a maximum value needed to maintain equilibrium.
- $F$  is a *kinetic frictional force*  $F_k$  when sliding occurs at the contacting surface.

Notice also from the graph that for very large values of  $P$  or for high speeds, aerodynamic effects will cause  $F_k$  and likewise  $\mu_k$  to begin to decrease.

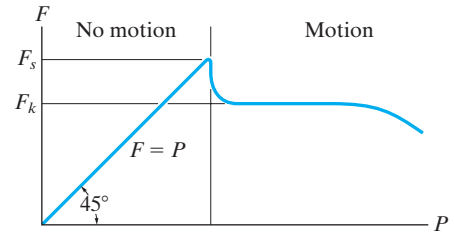


Fig. 8–3

**Characteristics of Dry Friction.** As a result of *experiments* that pertain to the foregoing discussion, we can state the following rules which apply to bodies subjected to dry friction.

- The frictional force acts *tangent* to the contacting surfaces in a direction *opposed* to the *motion* or tendency for motion of one surface relative to another.
- The maximum static frictional force  $F_s$  that can be developed is independent of the area of contact, provided the normal pressure is not very low nor great enough to severely deform or crush the contacting surfaces of the bodies.
- The maximum static frictional force is generally greater than the kinetic frictional force for any two surfaces of contact. However, if one of the bodies is moving with a *very low velocity* over the surface of another,  $F_k$  becomes approximately equal to  $F_s$ , i.e.,  $\mu_s \approx \mu_k$ .
- When *slipping* at the surface of contact is *about to occur*, the maximum static frictional force is proportional to the normal force, such that  $F_s = \mu_s N$ .
- When *slipping* at the surface of contact is *occurring*, the kinetic frictional force is proportional to the normal force, such that  $F_k = \mu_k N$ .



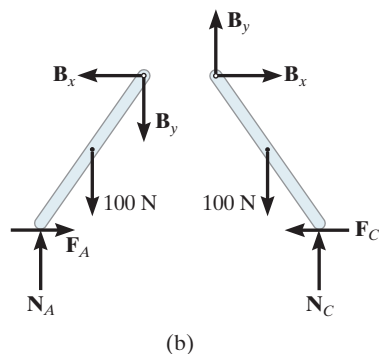
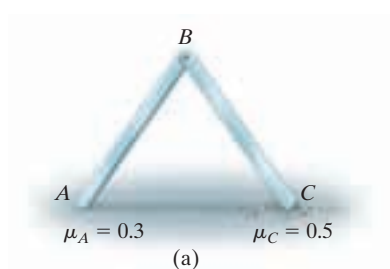


Fig. 8-4

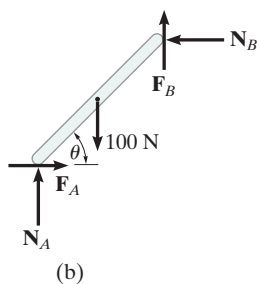
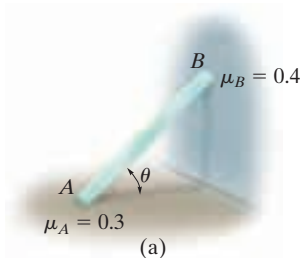


Fig. 8-5

## 8.2 Problems Involving Dry Friction

If a rigid body is in equilibrium when it is subjected to a system of forces that includes the effect of friction, the force system must satisfy not only the equations of equilibrium but *also* the laws that govern the frictional forces.

**Types of Friction Problems.** In general, there are three types of static problems involving dry friction. They can easily be classified once free-body diagrams are drawn and the total number of unknowns are identified and compared with the total number of available equilibrium equations.

**No Apparent Impending Motion.** Problems in this category are strictly equilibrium problems, which require the number of unknowns to be *equal* to the number of available equilibrium equations. Once the frictional forces are determined from the solution, however, their numerical values must be checked to be sure they satisfy the inequality  $F \leq \mu_s N$ ; otherwise, slipping will occur and the body will not remain in equilibrium. A problem of this type is shown in Fig. 8-4a. Here we must determine the frictional forces at A and C to check if the equilibrium position of the two-member frame can be maintained. If the bars are uniform and have known weights of 100 N each, then the free-body diagrams are as shown in Fig. 8-4b. There are six unknown force components which can be determined *strictly* from the six equilibrium equations (three for each member). Once  $F_A$ ,  $N_A$ ,  $F_C$ , and  $N_C$  are determined, then the bars will remain in equilibrium provided  $F_A \leq 0.3N_A$  and  $F_C \leq 0.5N_C$  are satisfied.

**Impending Motion at All Points of Contact.** In this case the total number of unknowns will *equal* the total number of available equilibrium equations *plus* the total number of available frictional equations,  $F = \mu N$ . When *motion is impending* at the points of contact, then  $F_s = \mu_s N$ ; whereas if the body is *slipping*, then  $F_k = \mu_k N$ . For example, consider the problem of finding the smallest angle  $\theta$  at which the 100-N bar in Fig. 8-5a can be placed against the wall without slipping. The free-body diagram is shown in Fig. 8-5b. Here the *five* unknowns are determined from the *three* equilibrium equations and *two* static frictional equations which apply at *both* points of contact, so that  $F_A = 0.3N_A$  and  $F_B = 0.4N_B$ .

**Impending Motion at Some Points of Contact.** Here the number of unknowns will be *less* than the number of available equilibrium equations plus the number of available frictional equations or conditional equations for tipping. As a result, several possibilities for motion or impending motion will exist and the problem will involve a determination of the kind of motion which actually occurs. For example, consider the two-member frame in Fig. 8–6a. In this problem we wish to determine the horizontal force  $P$  needed to cause movement. If each member has a weight of 100 N, then the free-body diagrams are as shown in Fig. 8–6b. There are *seven* unknowns. For a unique solution we must satisfy the *six* equilibrium equations (three for each member) and only *one* of two possible static frictional equations. This means that as  $P$  increases it will either cause slipping at  $A$  and no slipping at  $C$ , so that  $F_A = 0.3N_A$  and  $F_C \leq 0.5N_C$ ; or slipping occurs at  $C$  and no slipping at  $A$ , in which case  $F_C = 0.5N_C$  and  $F_A \leq 0.3N_A$ . The actual situation can be determined by calculating  $P$  for each case and then choosing the case for which  $P$  is *smaller*. If in both cases the *same value* for  $P$  is calculated, which would be highly improbable, then slipping at both points occurs simultaneously; i.e., the *seven unknowns* would satisfy *eight equations*.

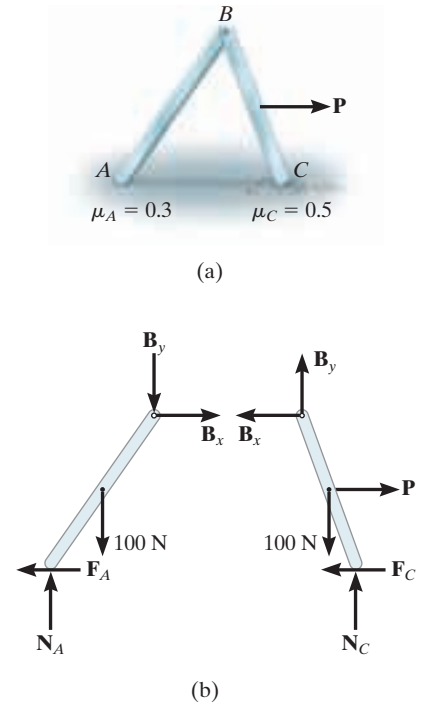


Fig. 8–6

**Equilibrium Versus Frictional Equations.** Whenever we solve a problem such as the one in Fig. 8–4, where the friction force  $F$  is to be an “equilibrium force” and satisfies the inequality  $F < \mu_s N$ , then we can assume the sense of direction of  $F$  on the free-body diagram. The correct sense is made known *after* solving the equations of equilibrium for  $F$ . If  $F$  is a negative scalar the sense of  $\mathbf{F}$  is the reverse of that which was assumed. This convenience of *assuming* the sense of  $\mathbf{F}$  is possible because the equilibrium equations equate to zero the *components of vectors* acting in the *same direction*. However, in cases where the frictional equation  $F = \mu N$  is used in the solution of a problem, as in the case shown in Fig. 8–5, then the convenience of *assuming* the sense of  $\mathbf{F}$  is *lost*, since the frictional equation relates only the *magnitudes* of two *perpendicular* vectors. Consequently,  $\mathbf{F}$  *must always* be shown acting with its *correct sense* on the free-body diagram, *whenever* the frictional equation is used for the solution of a problem.



Depending upon where the man pushes on the crate, it will either tip or slip.  
(© Russell C. Hibbeler)

### Important Points

- Friction is a tangential force that resists the movement of one surface relative to another.
- If no sliding occurs, the maximum value for the friction force is equal to the product of the coefficient of static friction and the normal force at the surface.
- If sliding occurs at a slow speed, then the friction force is the product of the coefficient of kinetic friction and the normal force at the surface.
- There are three types of static friction problems. Each of these problems is analyzed by first drawing the necessary free-body diagrams, and then applying the equations of equilibrium, while satisfying the conditions of friction or the possibility of tipping.

## Procedure for Analysis

Equilibrium problems involving dry friction can be solved using the following procedure.

### Free-Body Diagrams.

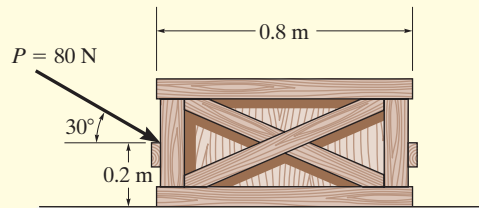
- Draw the necessary free-body diagrams, and unless it is stated in the problem that impending motion or slipping occurs, *always* show the frictional forces as unknowns (i.e., *do not assume*  $F = \mu N$ ).
- Determine the number of unknowns and compare this with the number of available equilibrium equations.
- If there are more unknowns than equations of equilibrium, it will be necessary to apply the frictional equation at some, if not all, points of contact to obtain the extra equations needed for a complete solution.
- If the equation  $F = \mu N$  is to be used, it will be necessary to show  $\mathbf{F}$  acting in the correct sense of direction on the free-body diagram.

### Equations of Equilibrium and Friction.

- Apply the equations of equilibrium and the necessary frictional equations (or conditional equations if tipping is possible) and solve for the unknowns.
- If the problem involves a three-dimensional force system such that it becomes difficult to obtain the force components or the necessary moment arms, apply the equations of equilibrium using Cartesian vectors.

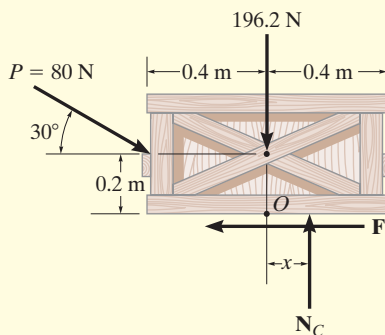
## EXAMPLE 8.1

The uniform crate shown in Fig. 8–7a has a mass of 20 kg. If a force  $P = 80$  N is applied to the crate, determine if it remains in equilibrium. The coefficient of static friction is  $\mu_s = 0.3$ .



(a)

Fig. 8–7



(b)

## SOLUTION

**Free-Body Diagram.** As shown in Fig. 8–7b, the *resultant* normal force  $N_C$  must act a distance  $x$  from the crate's center line in order to counteract the tipping effect caused by  $\mathbf{P}$ . There are *three unknowns*,  $F$ ,  $N_C$ , and  $x$ , which can be determined strictly from the *three* equations of equilibrium.

## Equations of Equilibrium.

$$\rightarrow \Sigma F_x = 0; \quad 80 \cos 30^\circ \text{ N} - F = 0$$

$$+\uparrow \Sigma F_y = 0; \quad -80 \sin 30^\circ \text{ N} + N_C - 196.2 \text{ N} = 0$$

$$\zeta + \Sigma M_O = 0; \quad 80 \sin 30^\circ \text{ N}(0.4 \text{ m}) - 80 \cos 30^\circ \text{ N}(0.2 \text{ m}) + N_C(x) = 0$$

Solving,

$$F = 69.3 \text{ N}$$

$$N_C = 236.2 \text{ N}$$

$$x = -0.00908 \text{ m} = -9.08 \text{ mm}$$

Since  $x$  is negative it indicates the *resultant* normal force acts (slightly) to the *left* of the crate's center line. No tipping will occur since  $x < 0.4$  m. Also, the *maximum* frictional force which can be developed at the surface of contact is  $F_{\max} = \mu_s N_C = 0.3(236.2 \text{ N}) = 70.9 \text{ N}$ . Since  $F = 69.3 \text{ N} < 70.9 \text{ N}$ , the crate will *not slip*, although it is very close to doing so.

**EXAMPLE 8.2**

It is observed that when the bed of the dump truck is raised to an angle of  $\theta = 25^\circ$  the vending machines will begin to slide off the bed, Fig. 8–8a. Determine the static coefficient of friction between a vending machine and the surface of the truckbed.



(a)

(© Russell C. Hibbeler)

**SOLUTION**

An idealized model of a vending machine resting on the truckbed is shown in Fig. 8–8b. The dimensions have been measured and the center of gravity has been located. We will assume that the vending machine weighs  $W$ .

**Free-Body Diagram.** As shown in Fig. 8–8c, the dimension  $x$  is used to locate the position of the resultant normal force  $\mathbf{N}$ . There are four unknowns,  $N$ ,  $F$ ,  $\mu_s$ , and  $x$ .

**Equations of Equilibrium.**

$$+\swarrow \Sigma F_x = 0; \quad W \sin 25^\circ - F = 0 \quad (1)$$

$$+\nearrow \Sigma F_y = 0; \quad N - W \cos 25^\circ = 0 \quad (2)$$

$$\zeta + \Sigma M_O = 0; \quad -W \sin 25^\circ(2.5 \text{ ft}) + W \cos 25^\circ(x) = 0 \quad (3)$$

Since slipping impends at  $\theta = 25^\circ$ , using Eqs. 1 and 2, we have

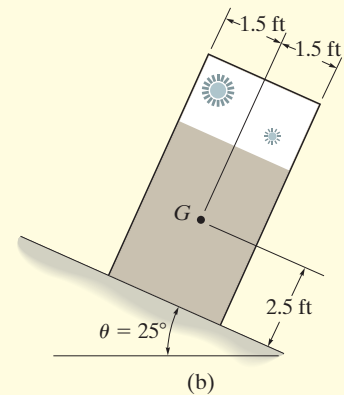
$$F_s = \mu_s N; \quad W \sin 25^\circ = \mu_s (W \cos 25^\circ)$$

$$\mu_s = \tan 25^\circ = 0.466$$

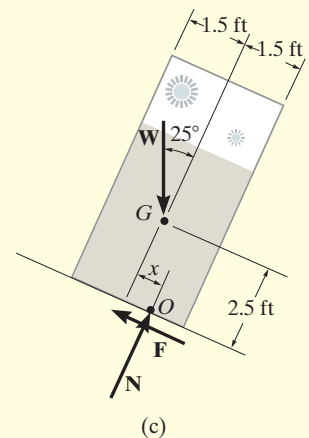
*Ans.*

The angle of  $\theta = 25^\circ$  is referred to as the **angle of repose**, and by comparison, it is equal to the angle of static friction,  $\theta = \phi_s$ . Notice from the calculation that  $\theta$  is independent of the weight of the vending machine, and so knowing  $\theta$  provides a convenient method for determining the coefficient of static friction.

**NOTE:** From Eq. 3, we find  $x = 1.17 \text{ ft}$ . Since  $1.17 \text{ ft} < 1.5 \text{ ft}$ , indeed the vending machine will slip before it can tip as observed in Fig. 8–8a.



(b)

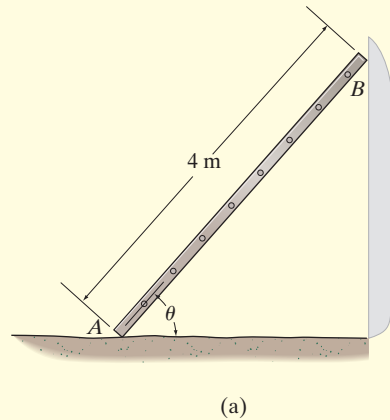
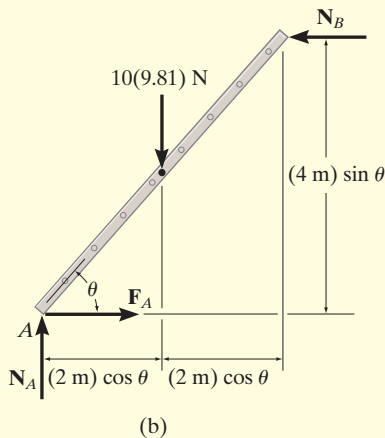


(c)

**Fig. 8–8**

**EXAMPLE 8.3**

The uniform 10-kg ladder in Fig. 8–9*a* rests against the smooth wall at *B*, and the end *A* rests on the rough horizontal plane for which the coefficient of static friction is  $\mu_s = 0.3$ . Determine the angle of inclination  $\theta$  of the ladder and the normal reaction at *B* if the ladder is on the verge of slipping.

**Fig. 8–9****SOLUTION**

**Free-Body Diagram.** As shown on the free-body diagram, Fig. 8–9*b*, the frictional force  $F_A$  must act to the right since impending motion at *A* is to the left.

**Equations of Equilibrium and Friction.** Since the ladder is on the verge of slipping, then  $F_A = \mu_s N_A = 0.3N_A$ . By inspection,  $N_A$  can be obtained directly.

$$+\uparrow \Sigma F_y = 0; \quad N_A - 10(9.81) \text{ N} = 0 \quad N_A = 98.1 \text{ N}$$

Using this result,  $F_A = 0.3(98.1 \text{ N}) = 29.43 \text{ N}$ . Now  $N_B$  can be found.

$$\rightarrow \Sigma F_x = 0; \quad 29.43 \text{ N} - N_B = 0$$

$$N_B = 29.43 \text{ N} = 29.4 \text{ N} \quad \text{Ans.}$$

Finally, the angle  $\theta$  can be determined by summing moments about point *A*.

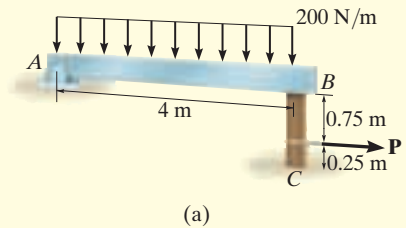
$$\zeta + \Sigma M_A = 0; \quad (29.43 \text{ N})(4 \text{ m}) \sin \theta - [10(9.81 \text{ N})](2 \text{ m}) \cos \theta = 0$$

$$\frac{\sin \theta}{\cos \theta} = \tan \theta = 1.6667$$

$$\theta = 59.04^\circ = 59.0^\circ \quad \text{Ans.}$$

**EXAMPLE 8.4**

Beam  $AB$  is subjected to a uniform load of  $200 \text{ N/m}$  and is supported at  $B$  by post  $BC$ , Fig. 8–10a. If the coefficients of static friction at  $B$  and  $C$  are  $\mu_B = 0.2$  and  $\mu_C = 0.5$ , determine the force  $\mathbf{P}$  needed to pull the post out from under the beam. Neglect the weight of the members and the thickness of the beam.

**SOLUTION**

**Free-Body Diagrams.** The free-body diagram of the beam is shown in Fig. 8–10b. Applying  $\Sigma M_A = 0$ , we obtain  $N_B = 400 \text{ N}$ . This result is shown on the free-body diagram of the post, Fig. 8–10c. Referring to this member, the *four* unknowns  $F_B$ ,  $P$ ,  $F_C$ , and  $N_C$  are determined from the *three* equations of equilibrium and *one* frictional equation applied either at  $B$  or  $C$ .

**Equations of Equilibrium and Friction.**

$$\rightarrow \Sigma F_x = 0; \quad P - F_B - F_C = 0 \quad (1)$$

$$+\uparrow \Sigma F_y = 0; \quad N_C - 400 \text{ N} = 0 \quad (2)$$

$$\curvearrowright \Sigma M_C = 0; \quad -P(0.25 \text{ m}) + F_B(1 \text{ m}) = 0 \quad (3)$$

**(Post Slips at B and Rotates about C.)** This requires  $F_C \leq \mu_C N_C$  and

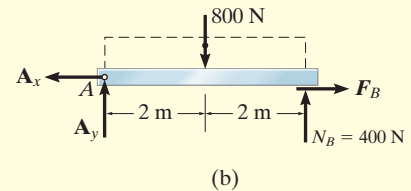
$$F_B = \mu_B N_B; \quad F_B = 0.2(400 \text{ N}) = 80 \text{ N}$$

Using this result and solving Eqs. 1 through 3, we obtain

$$P = 320 \text{ N}$$

$$F_C = 240 \text{ N}$$

$$N_C = 400 \text{ N}$$



Since  $F_C = 240 \text{ N} > \mu_C N_C = 0.5(400 \text{ N}) = 200 \text{ N}$ , slipping at  $C$  occurs. Thus the other case of movement must be investigated.

**(Post Slips at C and Rotates about B.)** Here  $F_B \leq \mu_B N_B$  and

$$F_C = \mu_C N_C; \quad F_C = 0.5 N_C \quad (4)$$

Solving Eqs. 1 through 4 yields

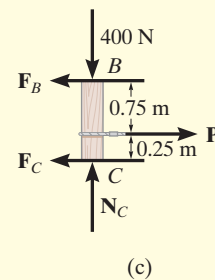
$$P = 267 \text{ N}$$

$$N_C = 400 \text{ N}$$

$$F_C = 200 \text{ N}$$

$$F_B = 66.7 \text{ N}$$

*Ans.*

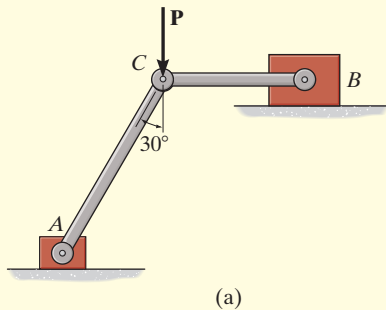


Obviously, this case occurs first since it requires a *smaller* value for  $P$ .

**Fig. 8–10**



## EXAMPLE 8.5



Blocks *A* and *B* have a mass of 3 kg and 9 kg, respectively, and are connected to the weightless links shown in Fig. 8–11*a*. Determine the largest vertical force **P** that can be applied at the pin *C* without causing any movement. The coefficient of static friction between the blocks and the contacting surfaces is  $\mu_s = 0.3$ .

## SOLUTION

**Free-Body Diagram.** The links are two-force members and so the free-body diagrams of pin *C* and blocks *A* and *B* are shown in Fig. 8–11*b*. Since the horizontal component of  $\mathbf{F}_{AC}$  tends to move block *A* to the left,  $\mathbf{F}_A$  must act to the right. Similarly,  $\mathbf{F}_B$  must act to the left to oppose the tendency of motion of block *B* to the right, caused by  $\mathbf{F}_{BC}$ . There are seven unknowns and six available force equilibrium equations, two for the pin and two for each block, so that *only one* frictional equation is needed.

**Equations of Equilibrium and Friction.** The force in links *AC* and *BC* can be related to *P* by considering the equilibrium of pin *C*.

$$+\uparrow \Sigma F_y = 0; \quad F_{AC} \cos 30^\circ - P = 0; \quad F_{AC} = 1.155P$$

$$\rightarrow \Sigma F_x = 0; \quad 1.155P \sin 30^\circ - F_{BC} = 0; \quad F_{BC} = 0.5774P$$

Using the result for  $F_{AC}$ , for block *A*,

$$\rightarrow \Sigma F_x = 0; \quad F_A - 1.155P \sin 30^\circ = 0; \quad F_A = 0.5774P \quad (1)$$

$$+\uparrow \Sigma F_y = 0; \quad N_A - 1.155P \cos 30^\circ - 3(9.81 \text{ N}) = 0; \quad N_A = P + 29.43 \text{ N} \quad (2)$$

Using the result for  $F_{BC}$ , for block *B*,

$$\rightarrow \Sigma F_x = 0; \quad (0.5774P) - F_B = 0; \quad F_B = 0.5774P \quad (3)$$

$$+\uparrow \Sigma F_y = 0; \quad N_B - 9(9.81 \text{ N}) = 0; \quad N_B = 88.29 \text{ N}$$

Movement of the system may be caused by the initial slipping of *either* block *A* or block *B*. If we assume that block *A* slips first, then

$$F_A = \mu_s N_A = 0.3N_A \quad (4)$$

Substituting Eqs. 1 and 2 into Eq. 4,

$$0.5774P = 0.3(P + 29.43)$$

$$P = 31.8 \text{ N}$$

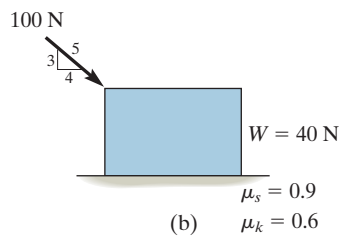
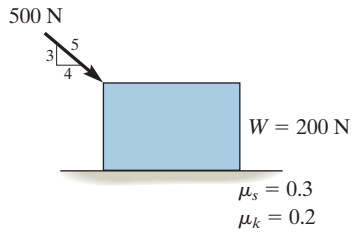
*Ans.*

Substituting this result into Eq. 3, we obtain  $F_B = 18.4 \text{ N}$ . Since the maximum static frictional force at *B* is  $(F_B)_{\max} = \mu_s N_B = 0.3(88.29 \text{ N}) = 26.5 \text{ N} > F_B$ , block *B* will not slip. Thus, the above assumption is correct. Notice that if the inequality were not satisfied, we would have to assume slipping of block *B* and then solve for *P*.

Fig. 8–11

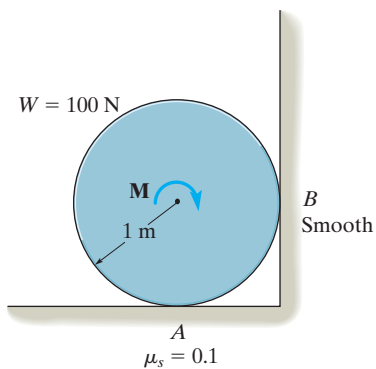
## PRELIMINARY PROBLEMS

**P8-1.** Determine the friction force at the surface of contact.



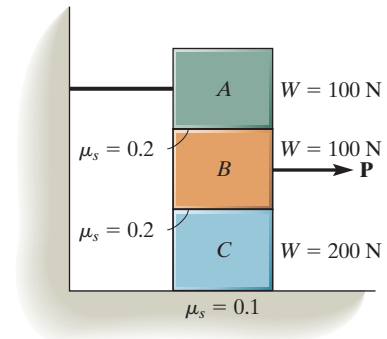
**Prob. P8-1**

**P8-2.** Determine  $M$  to cause impending motion of the cylinder.



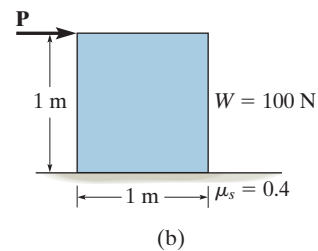
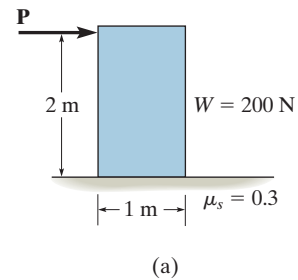
**Prob. P8-2**

**P8-3.** Determine the force  $P$  to move block  $B$ .



**Prob. P8-3**

**P8-4.** Determine the force  $P$  needed to cause impending motion of the block.

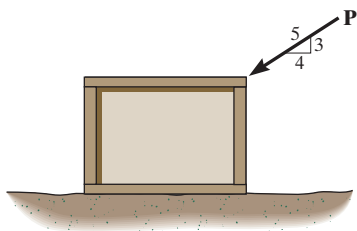


**Prob. P8-4**

## FUNDAMENTAL PROBLEMS

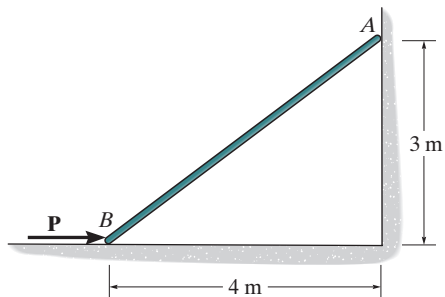
*All problem solutions must include FBDs.*

**F8-1.** Determine the friction developed between the 50-kg crate and the ground if a)  $P = 200$  N, and b)  $P = 400$  N. The coefficients of static and kinetic friction between the crate and the ground are  $\mu_s = 0.3$  and  $\mu_k = 0.2$ .



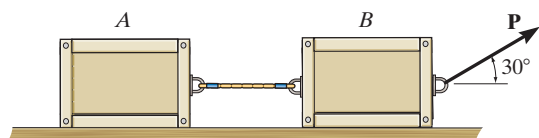
**Prob. F8-1**

**F8-2.** Determine the minimum force  $P$  to prevent the 30-kg rod  $AB$  from sliding. The contact surface at  $B$  is smooth, whereas the coefficient of static friction between the rod and the wall at  $A$  is  $\mu_s = 0.2$ .



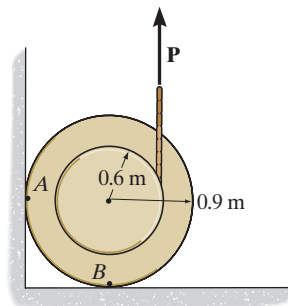
**Prob. F8-2**

**F8-3.** Determine the maximum force  $P$  that can be applied without causing the two 50-kg crates to move. The coefficient of static friction between each crate and the ground is  $\mu_s = 0.25$ .



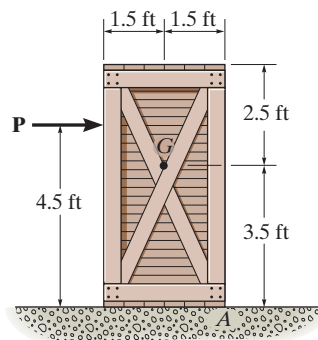
**Prob. F8-3**

**F8-4.** If the coefficient of static friction at contact points  $A$  and  $B$  is  $\mu_s = 0.3$ , determine the maximum force  $P$  that can be applied without causing the 100-kg spool to move.



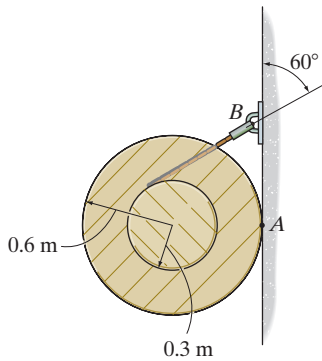
**Prob. F8-4**

**F8-5.** Determine the maximum force  $P$  that can be applied without causing movement of the 250-lb crate that has a center of gravity at  $G$ . The coefficient of static friction at the floor is  $\mu_s = 0.4$ .



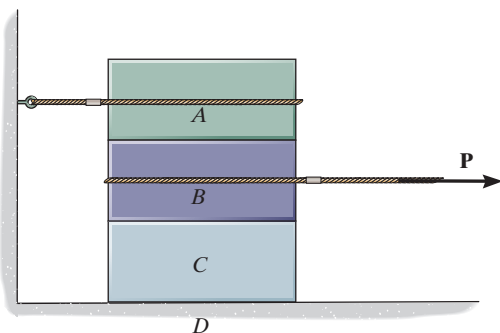
**Prob. F8-5**

**F8-6.** Determine the minimum coefficient of static friction between the uniform 50-kg spool and the wall so that the spool does not slip.



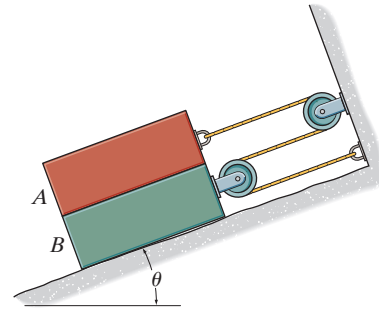
**Prob. F8-6**

**F8-7.** Blocks *A*, *B*, and *C* have weights of 50 N, 25 N, and 15 N, respectively. Determine the smallest horizontal force *P* that will cause impending motion. The coefficient of static friction between *A* and *B* is  $\mu_s = 0.3$ , between *B* and *C*,  $\mu'_s = 0.4$ , and between block *C* and the ground,  $\mu''_s = 0.35$ .



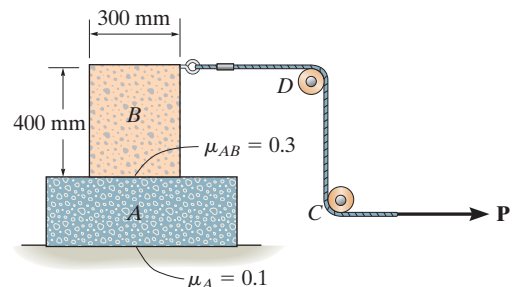
**Prob. F8-7**

**F8-8.** If the coefficient of static friction at all contacting surfaces is  $\mu_s$ , determine the inclination  $\theta$  at which the identical blocks, each of weight *W*, begin to slide.



**Prob. F8-8**

**F8-9.** Blocks *A* and *B* have a mass of 7 kg and 10 kg, respectively. Using the coefficients of static friction indicated, determine the largest force *P* which can be applied to the cord without causing motion. There are pulleys at *C* and *D*.

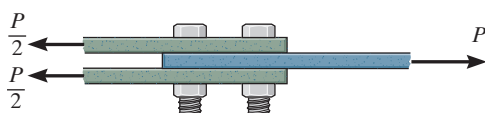


**Prob. F8-9**

## PROBLEMS

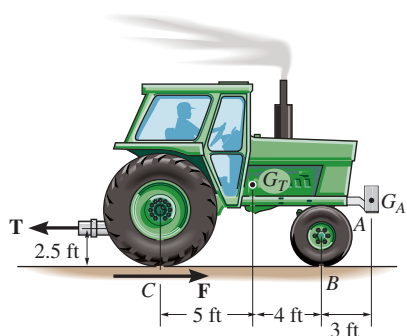
All problem solutions must include FBDs.

**8-1.** Determine the maximum force  $P$  the connection can support so that no slipping occurs between the plates. There are four bolts used for the connection and each is tightened so that it is subjected to a tension of 4 kN. The coefficient of static friction between the plates is  $\mu_s = 0.4$ .



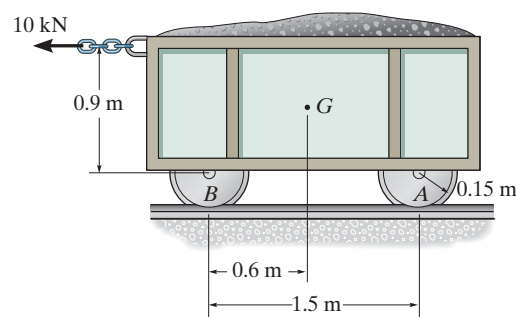
Prob. 8-1

**8-2.** The tractor exerts a towing force  $T = 400$  lb. Determine the normal reactions at each of the two front and two rear tires and the tractive frictional force  $\mathbf{F}$  on each rear tire needed to pull the load forward at constant velocity. The tractor has a weight of 7500 lb and a center of gravity located at  $G_T$ . An additional weight of 600 lb is added to its front having a center of gravity at  $G_A$ . Take  $\mu_s = 0.4$ . The front wheels are free to roll.



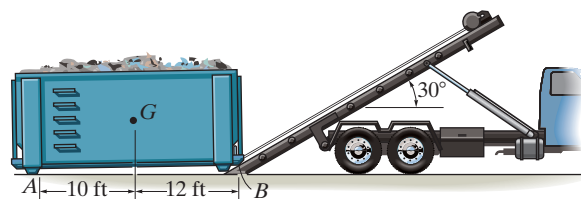
Prob. 8-2

**8-3.** The mine car and its contents have a total mass of 6 Mg and a center of gravity at  $G$ . If the coefficient of static friction between the wheels and the tracks is  $\mu_s = 0.4$  when the wheels are locked, find the normal force acting on the front wheels at  $B$  and the rear wheels at  $A$  when the brakes at both  $A$  and  $B$  are locked. Does the car move?



Prob. 8-3

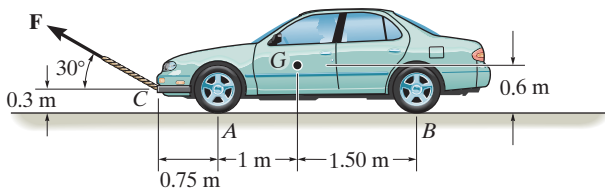
**\*8-4.** The winch on the truck is used to hoist the garbage bin onto the bed of the truck. If the loaded bin has a weight of 8500 lb and center of gravity at  $G$ , determine the force in the cable needed to begin the lift. The coefficients of static friction at  $A$  and  $B$  are  $\mu_A = 0.3$  and  $\mu_B = 0.2$ , respectively. Neglect the height of the support at  $A$ .



Prob. 8-4

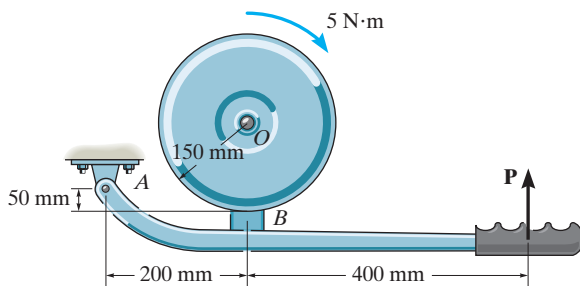
**8-5.** The automobile has a mass of 2 Mg and center of mass at  $G$ . Determine the towing force  $\mathbf{F}$  required to move the car if the back brakes are locked, and the front wheels are free to roll. Take  $\mu_s = 0.3$ .

**8-6.** The automobile has a mass of 2 Mg and center of mass at  $G$ . Determine the towing force  $\mathbf{F}$  required to move the car. Both the front and rear brakes are locked. Take  $\mu_s = 0.3$ .



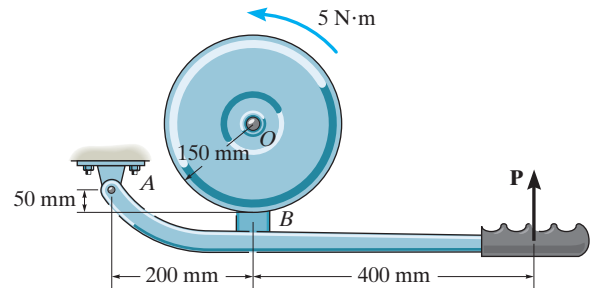
**Probs. 8-5/6**

**8-7.** The block brake consists of a pin-connected lever and friction block at  $B$ . The coefficient of static friction between the wheel and the lever is  $\mu_s = 0.3$ , and a torque of  $5 \text{ N} \cdot \text{m}$  is applied to the wheel. Determine if the brake can hold the wheel stationary when the force applied to the lever is (a)  $P = 30 \text{ N}$ , (b)  $P = 70 \text{ N}$ .



**Prob. 8-7**

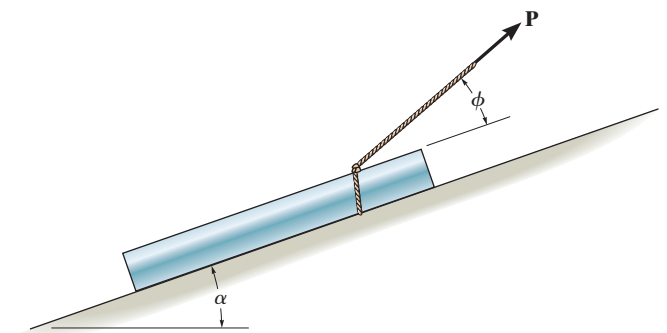
**\*8-8.** The block brake consists of a pin-connected lever and friction block at  $B$ . The coefficient of static friction between the wheel and the lever is  $\mu_s = 0.3$ , and a torque of  $5 \text{ N} \cdot \text{m}$  is applied to the wheel. Determine if the brake can hold the wheel stationary when the force applied to the lever is (a)  $P = 30 \text{ N}$ , (b)  $P = 70 \text{ N}$ .



**Prob. 8-8**

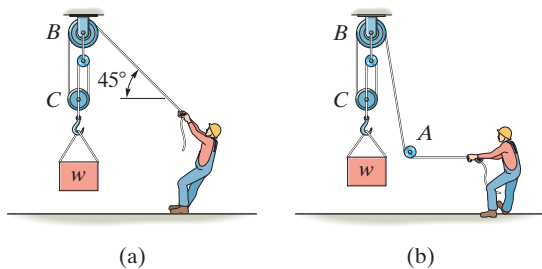
**8-9.** The pipe of weight  $W$  is to be pulled up the inclined plane of slope  $\alpha$  using a force  $\mathbf{P}$ . If  $\mathbf{P}$  acts at an angle  $\phi$ , show that for slipping  $P = W \sin(\alpha + \theta) / \cos(\phi - \theta)$ , where  $\theta$  is the angle of static friction;  $\theta = \tan^{-1} \mu_s$ .

**8-10.** Determine the angle  $\phi$  at which the applied force  $\mathbf{P}$  should act on the pipe so that the magnitude of  $\mathbf{P}$  is as small as possible for pulling the pipe up the incline. What is the corresponding value of  $P$ ? The pipe weighs  $W$  and the slope  $\alpha$  is known. Express the answer in terms of the angle of kinetic friction,  $\theta = \tan^{-1} \mu_k$ .



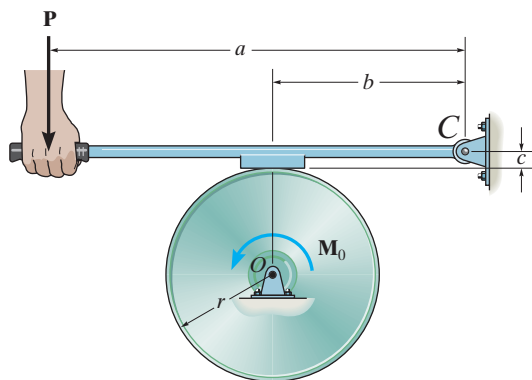
**Probs. 8-9/10**

**8–11.** Determine the maximum weight  $W$  the man can lift with constant velocity using the pulley system, without and then with the “leading block” or pulley at  $A$ . The man has a weight of 200 lb and the coefficient of static friction between his feet and the ground is  $\mu_s = 0.6$ .



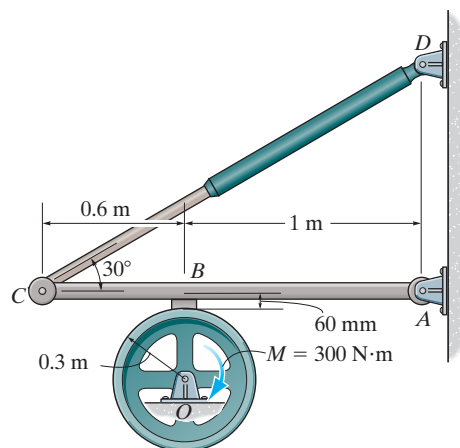
**Prob. 8–11**

**\*8–12.** The block brake is used to stop the wheel from rotating when the wheel is subjected to a couple moment  $M_0$ . If the coefficient of static friction between the wheel and the block is  $\mu_s$ , determine the smallest force  $P$  that should be applied.



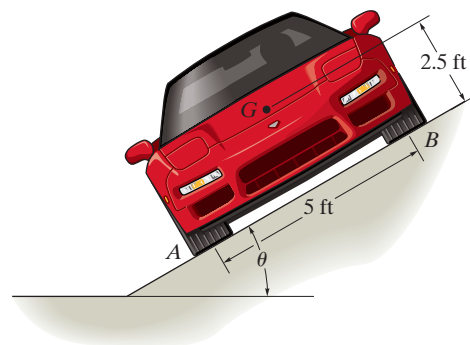
**Prob. 8–12**

**8–13.** If a torque of  $M = 300 \text{ N} \cdot \text{m}$  is applied to the flywheel, determine the force that must be developed in the hydraulic cylinder  $CD$  to prevent the flywheel from rotating. The coefficient of static friction between the friction pad at  $B$  and the flywheel is  $\mu_s = 0.4$ .



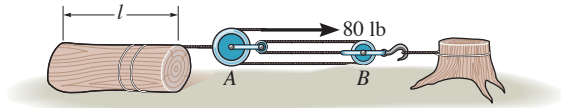
**Prob. 8–13**

**8–14.** The car has a mass of 1.6 Mg and center of mass at  $G$ . If the coefficient of static friction between the shoulder of the road and the tires is  $\mu_s = 0.4$ , determine the greatest slope  $\theta$  the shoulder can have without causing the car to slip or tip over if the car travels along the shoulder at constant velocity.



**Prob. 8–14**

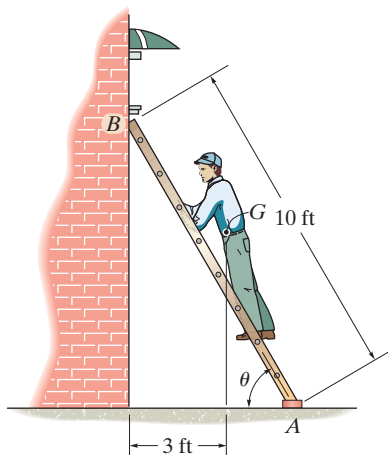
**8-15.** The log has a coefficient of static friction of  $\mu_s = 0.3$  with the ground and a weight of 40 lb/ft. If a man can pull on the rope with a maximum force of 80 lb, determine the greatest length  $l$  of log he can drag.



**Prob. 8-15**

**\*8-16.** The 180-lb man climbs up the ladder and stops at the position shown after he senses that the ladder is on the verge of slipping. Determine the inclination  $\theta$  of the ladder if the coefficient of static friction between the friction pad A and the ground is  $\mu_s = 0.4$ . Assume the wall at B is smooth. The center of gravity for the man is at G. Neglect the weight of the ladder.

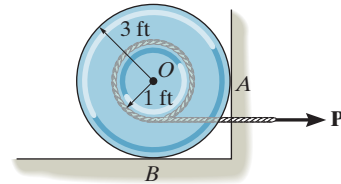
**8-17.** The 180-lb man climbs up the ladder and stops at the position shown after he senses that the ladder is on the verge of slipping. Determine the coefficient of static friction between the friction pad at A and ground if the inclination of the ladder is  $\theta = 60^\circ$  and the wall at B is smooth. The center of gravity for the man is at G. Neglect the weight of the ladder.



**Probs. 8-16/17**

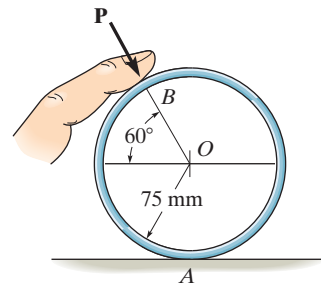
**8-18.** The spool of wire having a weight of 300 lb rests on the ground at B and against the wall at A. Determine the force  $P$  required to begin pulling the wire horizontally off the spool. The coefficient of static friction between the spool and its points of contact is  $\mu_s = 0.25$ .

**8-19.** The spool of wire having a weight of 300 lb rests on the ground at B and against the wall at A. Determine the normal force acting on the spool at A if  $P = 300$  lb. The coefficient of static friction between the spool and the ground at B is  $\mu_s = 0.35$ . The wall at A is smooth.



**Probs. 8-18/19**

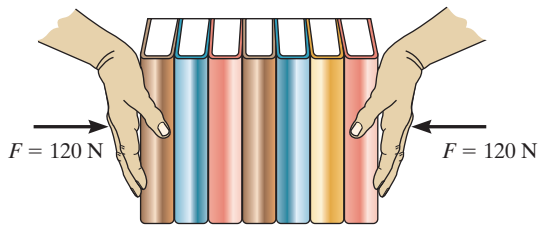
**\*8-20.** The ring has a mass of 0.5 kg and is resting on the surface of the table. In an effort to move the ring a normal force  $\mathbf{P}$  from the finger is exerted on it. If this force is directed towards the ring's center  $O$  as shown, determine its magnitude when the ring is on the verge of slipping at A. The coefficient of static friction at A is  $\mu_A = 0.2$  and at B,  $\mu_B = 0.3$ .



**Prob. 8-20**

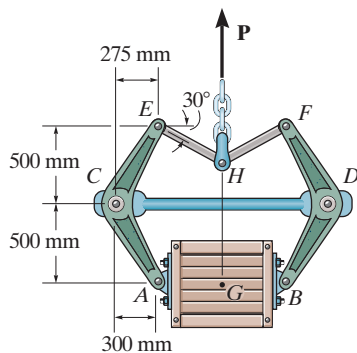


**8-21.** A man attempts to support a stack of books horizontally by applying a compressive force of  $F = 120 \text{ N}$  to the ends of the stack with his hands. If each book has a mass of  $0.95 \text{ kg}$ , determine the greatest number of books that can be supported in the stack. The coefficient of static friction between his hands and a book is  $(\mu_s)_h = 0.6$  and between any two books  $(\mu_s)_b = 0.4$ .



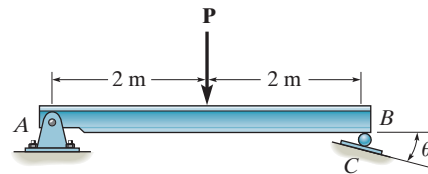
**Prob. 8-21**

**8-22.** The tongs are used to lift the  $150\text{-kg}$  crate, whose center of mass is at  $G$ . Determine the least coefficient of static friction at the pivot blocks so that the crate can be lifted.



**Prob. 8-22**

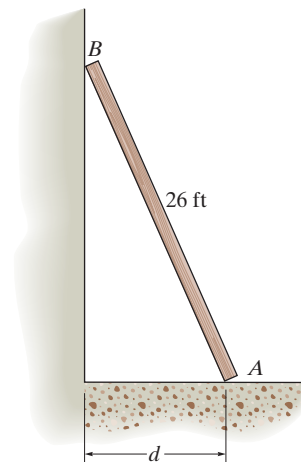
**8-23.** The beam is supported by a pin at  $A$  and a roller at  $B$  which has negligible weight and a radius of  $15 \text{ mm}$ . If the coefficient of static friction is  $\mu_B = \mu_C = 0.3$ , determine the largest angle  $\theta$  of the incline so that the roller does not slip for any force  $\mathbf{P}$  applied to the beam.



**Prob. 8-23**

**\*8-24.** The uniform thin pole has a weight of  $30 \text{ lb}$  and a length of  $26 \text{ ft}$ . If it is placed against the smooth wall and on the rough floor in the position  $d = 10 \text{ ft}$ , will it remain in this position when it is released? The coefficient of static friction is  $\mu_s = 0.3$ .

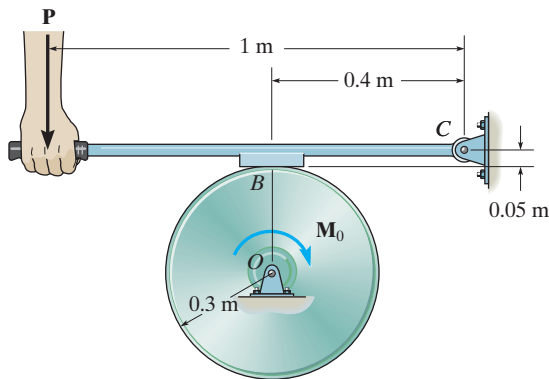
**8-25.** The uniform pole has a weight of  $30 \text{ lb}$  and a length of  $26 \text{ ft}$ . Determine the maximum distance  $d$  it can be placed from the smooth wall and not slip. The coefficient of static friction between the floor and the pole is  $\mu_s = 0.3$ .



**Probs. 8-24/25**

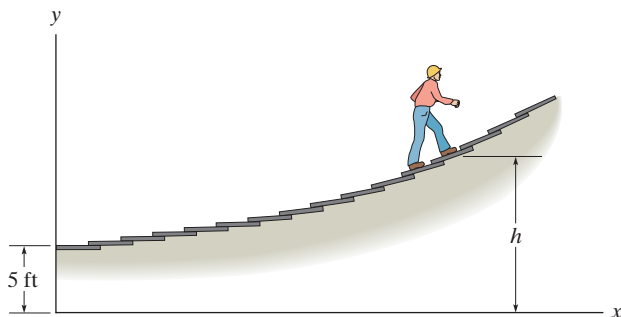
**8–26.** The block brake is used to stop the wheel from rotating when the wheel is subjected to a couple moment  $M_0 = 360 \text{ N} \cdot \text{m}$ . If the coefficient of static friction between the wheel and the block is  $\mu_s = 0.6$ , determine the smallest force  $P$  that should be applied.

**8–27.** Solve Prob. 8–26 if the couple moment  $M_0$  is applied counterclockwise.



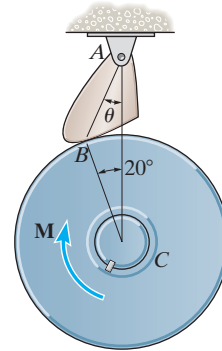
**Probs. 8–26/27**

**\*8–28.** A worker walks up the sloped roof that is defined by the curve  $y = (5e^{0.01x}) \text{ ft}$ , where  $x$  is in feet. Determine how high  $h$  he can go without slipping. The coefficient of static friction is  $\mu_s = 0.6$ .



**Prob. 8–28**

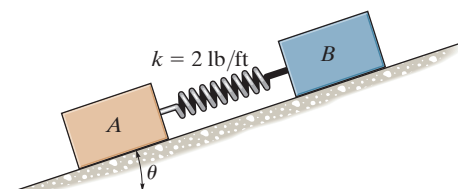
**8–29.** The friction pawl is pinned at  $A$  and rests against the wheel at  $B$ . It allows freedom of movement when the wheel is rotating counterclockwise about  $C$ . Clockwise rotation is prevented due to friction of the pawl which tends to bind the wheel. If  $(\mu_s)_B = 0.6$ , determine the design angle  $\theta$  which will prevent clockwise motion for any value of applied moment  $M$ . *Hint:* Neglect the weight of the pawl so that it becomes a two-force member.



**Prob. 8–29**

**8–30.** Two blocks  $A$  and  $B$  have a weight of 10 lb and 6 lb, respectively. They are resting on the incline for which the coefficients of static friction are  $\mu_A = 0.15$  and  $\mu_B = 0.25$ . Determine the incline angle  $\theta$  for which both blocks begin to slide. Also find the required stretch or compression in the connecting spring for this to occur. The spring has a stiffness of  $k = 2 \text{ lb/ft}$ .

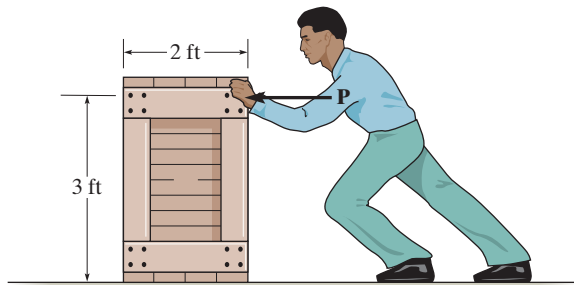
**8–31.** Two blocks  $A$  and  $B$  have a weight of 10 lb and 6 lb, respectively. They are resting on the incline for which the coefficients of static friction are  $\mu_A = 0.15$  and  $\mu_B = 0.25$ . Determine the angle  $\theta$  which will cause motion of one of the blocks. What is the friction force under each of the blocks when this occurs? The spring has a stiffness of  $k = 2 \text{ lb/ft}$  and is originally unstretched.



**Probs. 8–30/31**

**\*8-32.** Determine the smallest force  $P$  that must be applied in order to cause the 150-lb uniform crate to move. The coefficient of static friction between the crate and the floor is  $\mu_s = 0.5$ .

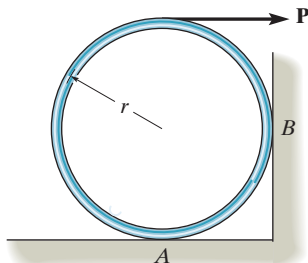
**8-33.** The man having a weight of 200 lb pushes horizontally on the crate. If the coefficient of static friction between the 450-lb crate and the floor is  $\mu_s = 0.3$  and between his shoes and the floor is  $\mu'_s = 0.6$ , determine if he can move the crate.



**Probs. 8-32/33**

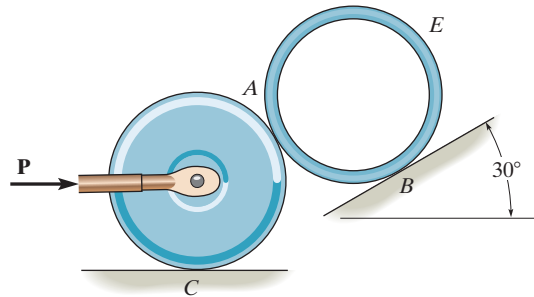
**8-34.** The uniform hoop of weight  $W$  is subjected to the horizontal force  $P$ . Determine the coefficient of static friction between the hoop and the surface of  $A$  and  $B$  if the hoop is on the verge of rotating.

**8-35.** Determine the maximum horizontal force  $P$  that can be applied to the 30-lb hoop without causing it to rotate. The coefficient of static friction between the hoop and the surfaces  $A$  and  $B$  is  $\mu_s = 0.2$ . Take  $r = 300$  mm.



**Probs. 8-34/35**

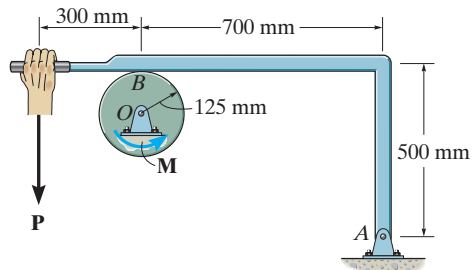
**\*8-36.** Determine the minimum force  $P$  needed to push the tube  $E$  up the incline. The force acts parallel to the plane, and the coefficients of static friction at the contacting surfaces are  $\mu_A = 0.2$ ,  $\mu_B = 0.3$ , and  $\mu_C = 0.4$ . The 100-kg roller and 40-kg tube each have a radius of 150 mm.



**Prob. 8-36**

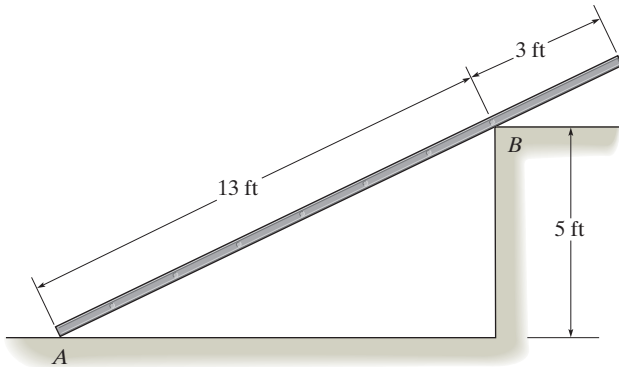
**8-37.** The coefficients of static and kinetic friction between the drum and brake bar are  $\mu_s = 0.4$  and  $\mu_k = 0.3$ , respectively. If  $M = 50$  N·m and  $P = 85$  N, determine the horizontal and vertical components of reaction at the pin  $O$ . Neglect the weight and thickness of the brake. The drum has a mass of 25 kg.

**8-38.** The coefficient of static friction between the drum and brake bar is  $\mu_s = 0.4$ . If the moment  $M = 35$  N·m, determine the smallest force  $P$  that needs to be applied to the brake bar in order to prevent the drum from rotating. Also determine the corresponding horizontal and vertical components of reaction at pin  $O$ . Neglect the weight and thickness of the brake bar. The drum has a mass of 25 kg.



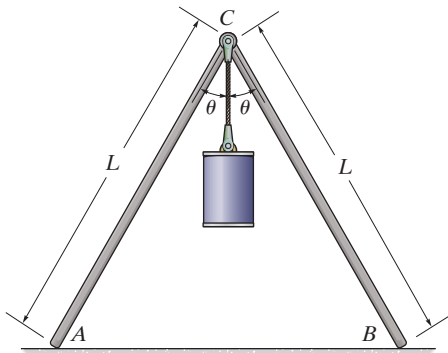
**Probs. 8-37/38**

**8–39.** Determine the smallest coefficient of static friction at both  $A$  and  $B$  needed to hold the uniform 100-lb bar in equilibrium. Neglect the thickness of the bar. Take  $\mu_A = \mu_B = \mu$ .



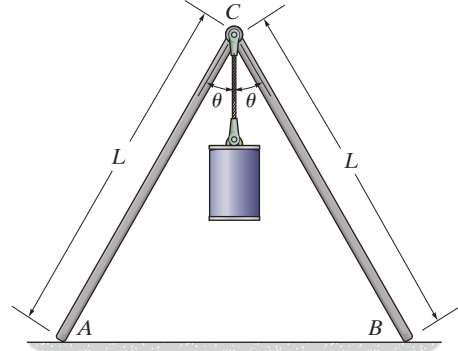
**Prob. 8–39**

**\*8–40.** If  $\theta = 30^\circ$ , determine the minimum coefficient of static friction at  $A$  and  $B$  so that equilibrium of the supporting frame is maintained regardless of the mass of the cylinder. Neglect the mass of the rods.



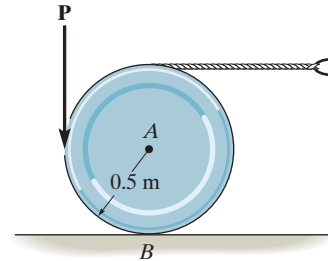
**Prob. 8–40**

**8–41.** If the coefficient of static friction at  $A$  and  $B$  is  $\mu_s = 0.6$ , determine the maximum angle  $\theta$  so that the frame remains in equilibrium, regardless of the mass of the cylinder. Neglect the mass of the rods.



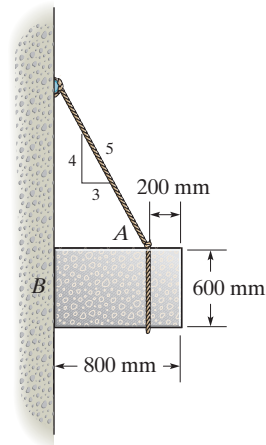
**Prob. 8–41**

**8–42.** The 100-kg disk rests on a surface for which  $\mu_B = 0.2$ . Determine the smallest vertical force  $\mathbf{P}$  that can be applied tangentially to the disk which will cause motion to impend.



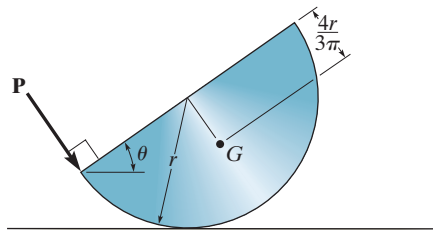
**Prob. 8–42**

**8–43.** Investigate whether the equilibrium can be maintained. The uniform block has a mass of 500 kg, and the coefficient of static friction is  $\mu_s = 0.3$ .



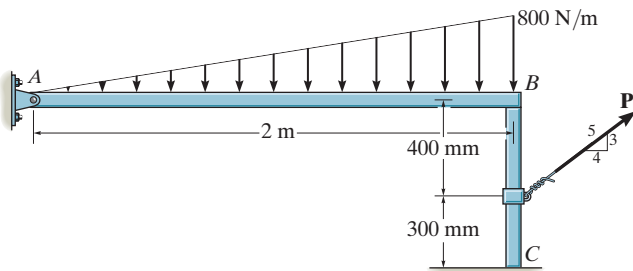
**Prob. 8–43**

**\*8-44.** The homogenous semicylinder has a mass of 20 kg and mass center at  $G$ . If force  $\mathbf{P}$  is applied at the edge, and  $r = 300$  mm, determine the angle  $\theta$  at which the semicylinder is on the verge of slipping. The coefficient of static friction between the plane and the cylinder is  $\mu_s = 0.3$ . Also, what is the corresponding force  $P$  for this case?



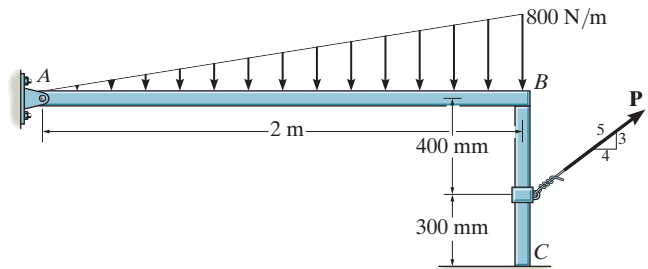
**Prob. 8-44**

**8-45.** The beam  $AB$  has a negligible mass and thickness and is subjected to a triangular distributed loading. It is supported at one end by a pin and at the other end by a post having a mass of 50 kg and negligible thickness. Determine the minimum force  $P$  needed to move the post. The coefficients of static friction at  $B$  and  $C$  are  $\mu_B = 0.4$  and  $\mu_C = 0.2$ , respectively.



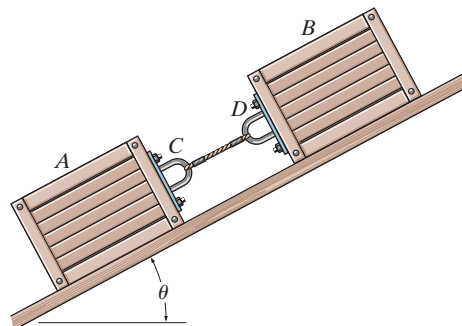
**Prob. 8-45**

**8-46.** The beam  $AB$  has a negligible mass and thickness and is subjected to a triangular distributed loading. It is supported at one end by a pin and at the other end by a post having a mass of 50 kg and negligible thickness. Determine the two coefficients of static friction at  $B$  and at  $C$  so that when the magnitude of the applied force is increased to  $P = 150$  N, the post slips at both  $B$  and  $C$  simultaneously.



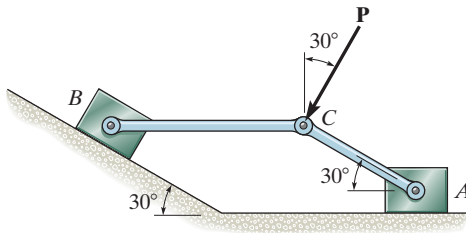
**Prob. 8-46**

**8-47.** Crates  $A$  and  $B$  weigh 200 lb and 150 lb, respectively. They are connected together with a cable and placed on the inclined plane. If the angle  $\theta$  is gradually increased, determine  $\theta$  when the crates begin to slide. The coefficients of static friction between the crates and the plane are  $\mu_A = 0.25$  and  $\mu_B = 0.35$ .



**Prob. 8-47**

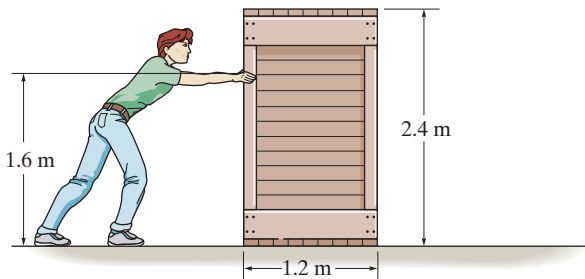
**\*8-48.** Two blocks  $A$  and  $B$ , each having a mass of 5 kg, are connected by the linkage shown. If the coefficient of static friction at the contacting surfaces is  $\mu_s = 0.5$ , determine the largest force  $P$  that can be applied to pin  $C$  of the linkage without causing the blocks to move. Neglect the weight of the links.



**Prob. 8-48**

**8-49.** The uniform crate has a mass of 150 kg. If the coefficient of static friction between the crate and the floor is  $\mu_s = 0.2$ , determine whether the 85-kg man can move the crate. The coefficient of static friction between his shoes and the floor is  $\mu'_s = 0.4$ . Assume the man only exerts a horizontal force on the crate.

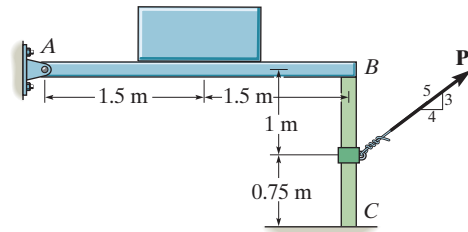
**8-50.** The uniform crate has a mass of 150 kg. If the coefficient of static friction between the crate and the floor is  $\mu_s = 0.2$ , determine the smallest mass of the man so he can move the crate. The coefficient of static friction between his shoes and the floor is  $\mu'_s = 0.45$ . Assume the man exerts only a horizontal force on the crate.



**Probs. 8-49/50**

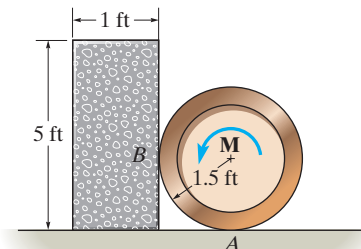
**8-51.** Beam  $AB$  has a negligible mass and thickness, and supports the 200-kg uniform block. It is pinned at  $A$  and rests on the top of a post, having a mass of 20 kg and negligible thickness. Determine the minimum force  $P$  needed to move the post. The coefficients of static friction at  $B$  and  $C$  are  $\mu_B = 0.4$  and  $\mu_C = 0.2$ , respectively.

**\*8-52.** Beam  $AB$  has a negligible mass and thickness, and supports the 200-kg uniform block. It is pinned at  $A$  and rests on the top of a post, having a mass of 20 kg and negligible thickness. Determine the two coefficients of static friction at  $B$  and at  $C$  so that when the magnitude of the applied force is increased to  $P = 300$  N, the post slips at both  $B$  and  $C$  simultaneously.



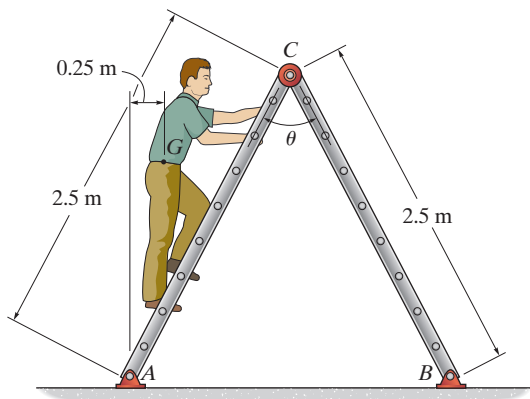
**Probs. 8-51/52**

**8-53.** Determine the smallest couple moment that can be applied to the 150-lb wheel that will cause impending motion. The uniform concrete block has a weight of 300 lb. The coefficients of static friction are  $\mu_A = 0.2$ ,  $\mu_B = 0.3$ , and between the concrete block and the floor,  $\mu = 0.4$ .



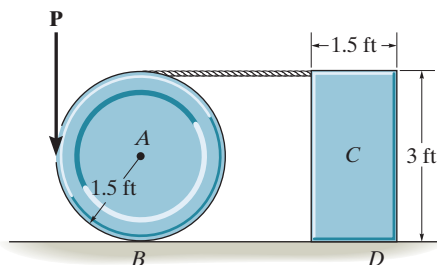
**Prob. 8-53**

**8-54.** Determine the greatest angle  $\theta$  so that the ladder does not slip when it supports the 75-kg man in the position shown. The surface is rather slippery, where the coefficient of static friction at  $A$  and  $B$  is  $\mu_s = 0.3$ .



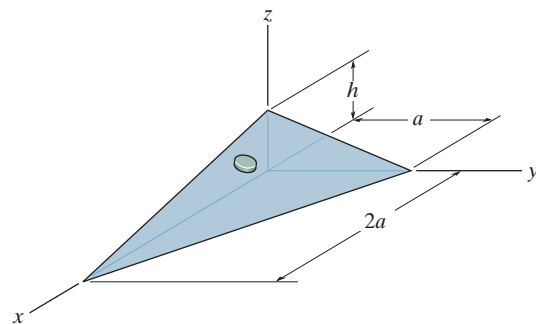
**Prob. 8-54**

**8-55.** The wheel weighs 20 lb and rests on a surface for which  $\mu_B = 0.2$ . A cord wrapped around it is attached to the top of the 30-lb homogeneous block. If the coefficient of static friction at  $D$  is  $\mu_D = 0.3$ , determine the smallest vertical force that can be applied tangentially to the wheel which will cause motion to impend.



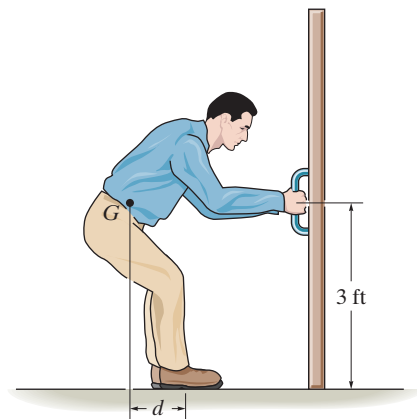
**Prob. 8-55**

**\*8-56.** The disk has a weight  $W$  and lies on a plane that has a coefficient of static friction  $\mu$ . Determine the maximum height  $h$  to which the plane can be lifted without causing the disk to slip.



**Prob. 8-56**

**8-57.** The man has a weight of 200 lb, and the coefficient of static friction between his shoes and the floor is  $\mu_s = 0.5$ . Determine where he should position his center of gravity  $G$  at  $d$  in order to exert the maximum horizontal force on the door. What is this force?



**Prob. 8-57**

## CONCEPTUAL PROBLEMS

**C8-1.** Draw the free-body diagrams of each of the two members of this friction tong used to lift the 100-kg block.



**C8-1** (© Russell C. Hibbeler)

**C8-2.** Show how to find the force needed to move the top block. Use reasonable data and use an equilibrium analysis to explain your answer.



**C8-2** (© Russell C. Hibbeler)

**C8-3.** The rope is used to tow the refrigerator. Is it best to pull slightly up on the rope as shown, pull horizontally, or pull somewhat downwards? Also, is it best to attach the rope at a high position as shown, or at a lower position? Do an equilibrium analysis to explain your answer.

**C8-4.** The rope is used to tow the refrigerator. In order to prevent yourself from slipping while towing, is it best to pull up as shown, pull horizontally, or pull downwards on the rope? Do an equilibrium analysis to explain your answer.



**C8-3/4** (© Russell C. Hibbeler)

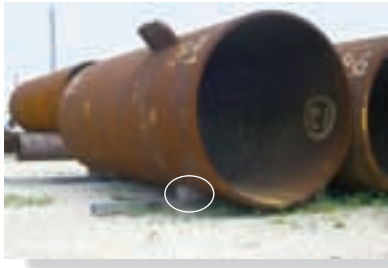
**C8-5.** Explain how to find the maximum force this man can exert on the vehicle. Use reasonable data and use an equilibrium analysis to explain your answer.



**C8-5** (© Russell C. Hibbeler)



## 8.3 Wedges



Wedges are often used to adjust the elevation of structural or mechanical parts. Also, they provide stability for objects such as this pipe. (© Russell C. Hibbeler)

A **wedge** is a simple machine that is often used to transform an applied force into much larger forces, directed at approximately right angles to the applied force. Wedges also can be used to slightly move or adjust heavy loads.

Consider, for example, the wedge shown in Fig. 8–12a, which is used to *lift* the block by applying a force to the wedge. Free-body diagrams of the block and wedge are shown in Fig. 8–12b. Here we have excluded the weight of the wedge since it is usually *small* compared to the weight  $\mathbf{W}$  of the block. Also, note that the frictional forces  $\mathbf{F}_1$  and  $\mathbf{F}_2$  must oppose the motion of the wedge. Likewise, the frictional force  $\mathbf{F}_3$  of the wall on the block must act downward so as to oppose the block's upward motion. The locations of the resultant normal forces are not important in the force analysis since neither the block nor wedge will “tip.” Hence the moment equilibrium equations will not be considered. There are seven unknowns, consisting of the applied force  $\mathbf{P}$ , needed to cause motion of the wedge, and six normal and frictional forces. The seven available equations consist of four force equilibrium equations,  $\Sigma F_x = 0$ ,  $\Sigma F_y = 0$  applied to the wedge and block, and three frictional equations,  $F = \mu N$ , applied at each surface of contact.

If the block is to be *lowered*, then the frictional forces will all act in a sense opposite to that shown in Fig. 8–12b. Provided the coefficient of friction is very *small* or the wedge angle  $\theta$  is *large*, then the applied force  $\mathbf{P}$  must act to the right to hold the block. Otherwise,  $\mathbf{P}$  may have a reverse sense of direction in order to *pull* on the wedge to remove it. If  $\mathbf{P}$  is *not applied* and friction forces hold the block in place, then the wedge is referred to as *self-locking*.

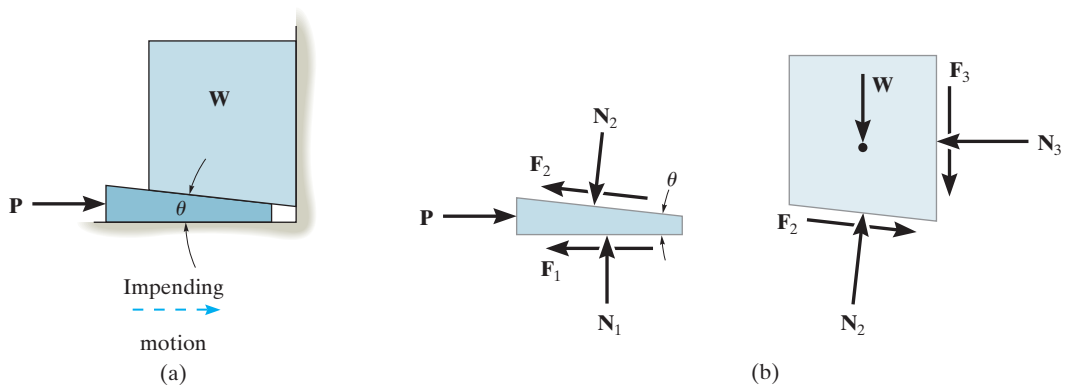
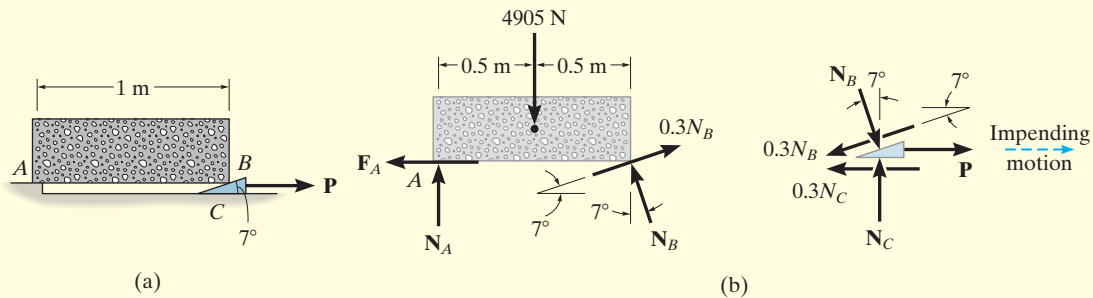


Fig. 8–12

**EXAMPLE 8.6**

The uniform stone in Fig. 8–13*a* has a mass of 500 kg and is held in the horizontal position using a wedge at *B*. If the coefficient of static friction is  $\mu_s = 0.3$  at the surfaces of contact, determine the minimum force  $P$  needed to remove the wedge. Assume that the stone does not slip at *A*.

**Fig. 8–13****SOLUTION**

The minimum force  $P$  requires  $F = \mu_s N$  at the surfaces of contact with the wedge. The free-body diagrams of the stone and wedge are shown in Fig. 8–13*b*. On the wedge the friction force opposes the impending motion, and on the stone at *A*,  $F_A \leq \mu_s N_A$ , since slipping does not occur there. There are five unknowns. Three equilibrium equations for the stone and two for the wedge are available for solution. From the free-body diagram of the stone,

$$\begin{aligned} \zeta + \Sigma M_A = 0; \quad & -4905 \text{ N}(0.5 \text{ m}) + (N_B \cos 7^\circ \text{ N})(1 \text{ m}) \\ & + (0.3N_B \sin 7^\circ \text{ N})(1 \text{ m}) = 0 \\ & N_B = 2383.1 \text{ N} \end{aligned}$$

Using this result for the wedge, we have

$$\begin{aligned} + \uparrow \Sigma F_y = 0; \quad & N_C - 2383.1 \cos 7^\circ \text{ N} - 0.3(2383.1 \sin 7^\circ \text{ N}) = 0 \\ & N_C = 2452.5 \text{ N} \\ \pm \Sigma F_x = 0; \quad & 2383.1 \sin 7^\circ \text{ N} - 0.3(2383.1 \cos 7^\circ \text{ N}) + \\ & P - 0.3(2452.5 \text{ N}) = 0 \\ & P = 1154.9 \text{ N} = 1.15 \text{ kN} \quad \text{Ans.} \end{aligned}$$

**NOTE:** Since  $P$  is positive, indeed the wedge must be pulled out. If  $P$  were zero, the wedge would remain in place (self-locking) and the frictional forces developed at *B* and *C* would satisfy  $F_B < \mu_s N_B$  and  $F_C < \mu_s N_C$ .



Square-threaded screws find applications on valves, jacks, and vises, where particularly large forces must be developed along the axis of the screw. (© Russell C. Hibbeler)

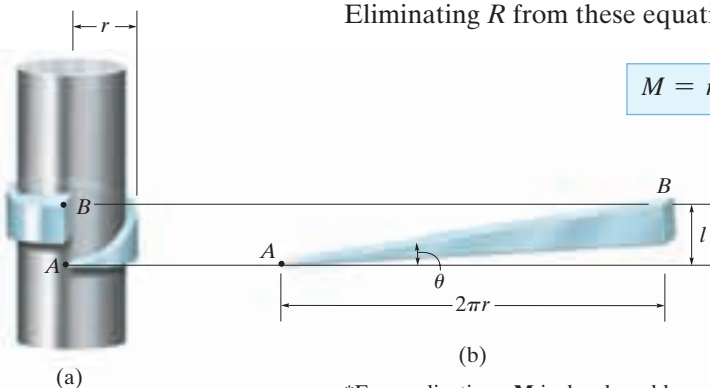
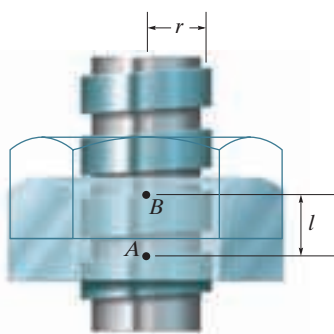


Fig. 8–14

## 8.4 Frictional Forces on Screws

In most cases, screws are used as fasteners; however, in many types of machines they are incorporated to transmit power or motion from one part of the machine to another. A **square-threaded screw** is commonly used for the latter purpose, especially when large forces are applied along its axis. In this section, we will analyze the forces acting on square-threaded screws. The analysis of other types of screws, such as the V-thread, is based on these same principles.

For analysis, a square-threaded screw, as in Fig. 8–14, can be considered a cylinder having an inclined square ridge or **thread** wrapped around it. If we unwind the thread by one revolution, as shown in Fig. 8–14b, the slope or the **lead angle**  $\theta$  is determined from  $\theta = \tan^{-1}(l/2\pi r)$ . Here  $l$  and  $2\pi r$  are the vertical and horizontal distances between  $A$  and  $B$ , where  $r$  is the mean radius of the thread. The distance  $l$  is called the **lead** of the screw and it is equivalent to the distance the screw advances when it turns one revolution.

**Upward Impending Motion.** Let us now consider the case of the square-threaded screw jack in Fig. 8–15 that is subjected to upward impending motion caused by the applied torsional moment  $\mathbf{M}$ . A free-body diagram of the *entire unraveled thread*  $h$  in contact with the jack can be represented as a block, as shown in Fig. 8–16a. The force  $\mathbf{W}$  is the vertical force acting on the thread or the axial force applied to the shaft, Fig. 8–15, and  $M/r$  is the resultant horizontal force produced by the couple moment  $M$  about the axis of the shaft. The reaction  $\mathbf{R}$  of the groove on the thread has both frictional and normal components, where  $F = \mu_s N$ . The angle of static friction is  $\phi_s = \tan^{-1}(F/N) = \tan^{-1}\mu_s$ . Applying the force equations of equilibrium along the horizontal and vertical axes, we have

$$\rightarrow \Sigma F_x = 0; \quad M/r - R \sin(\theta + \phi_s) = 0$$

$$+\uparrow \Sigma F_y = 0; \quad R \cos(\theta + \phi_s) - W = 0$$

Eliminating  $R$  from these equations, we obtain

$$M = rW \tan(\theta + \phi_s) \quad (8-3)$$

\*For applications,  $\mathbf{M}$  is developed by applying a horizontal force  $\mathbf{P}$  at a right angle to the end of a lever that would be fixed to the screw.

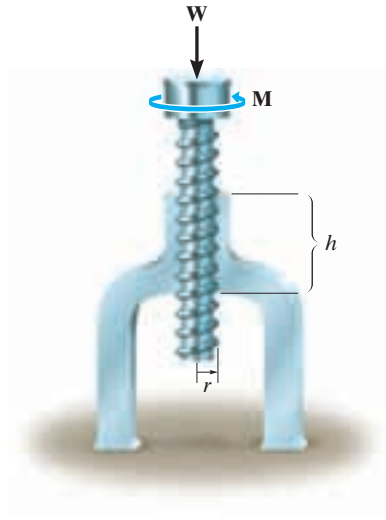


Fig. 8-15

**Self-Locking Screw.** A screw is said to be *self-locking* if it remains in place under any axial load  $\mathbf{W}$  when the moment  $\mathbf{M}$  is removed. For this to occur, the direction of the frictional force must be reversed so that  $\mathbf{R}$  acts on the other side of  $\mathbf{N}$ . Here the angle of static friction  $\phi_s$  becomes greater than or equal to  $\theta$ , Fig. 8-16d. If  $\phi_s = \theta$ , Fig. 8-16b, then  $\mathbf{R}$  will act vertically to balance  $\mathbf{W}$ , and the screw will be on the verge of winding downward.

**Downward Impending Motion, ( $\theta > \phi_s$ ).** If the screw is not self-locking, it is necessary to apply a moment  $\mathbf{M}'$  to prevent the screw from winding downward. Here, a horizontal force  $M'/r$  is required to push against the thread to prevent it from sliding down the plane, Fig. 8-16c. Using the same procedure as before, the magnitude of the moment  $\mathbf{M}'$  required to prevent this unwinding is

$$M' = rW \tan (\theta - \phi_s) \quad (8-4)$$

**Downward Impending Motion, ( $\phi_s > \theta$ ).** If a screw is self-locking, a couple moment  $\mathbf{M}''$  must be applied to the screw in the opposite direction to wind the screw downward ( $\phi_s > \theta$ ). This causes a reverse horizontal force  $M''/r$  that pushes the thread down as indicated in Fig. 8-16d. In this case, we obtain

$$M'' = rW \tan (\phi_s - \theta) \quad (8-5)$$

If motion of the screw occurs, Eqs. 8-3, 8-4, and 8-5 can be applied by simply replacing  $\phi_s$  with  $\phi_k$ .

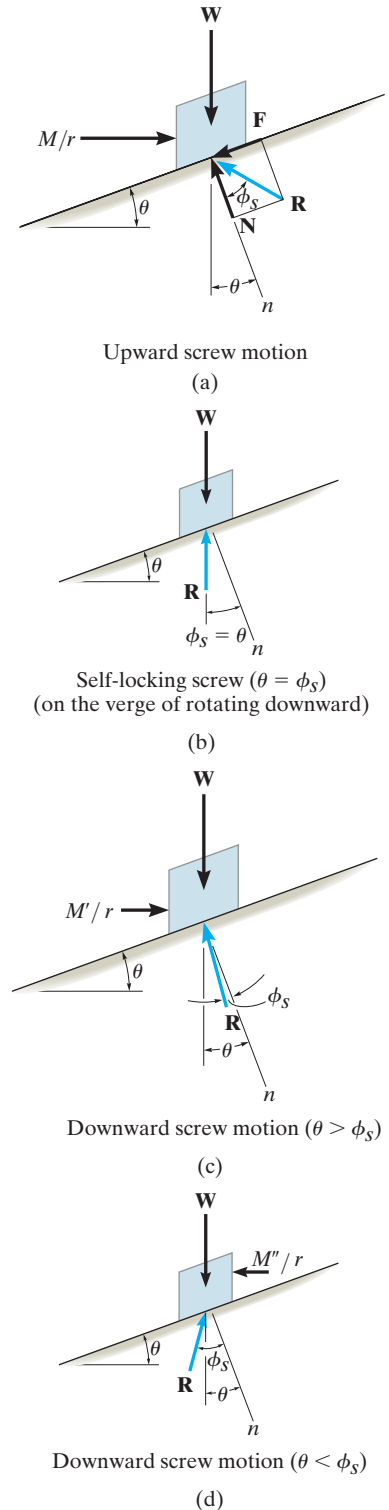


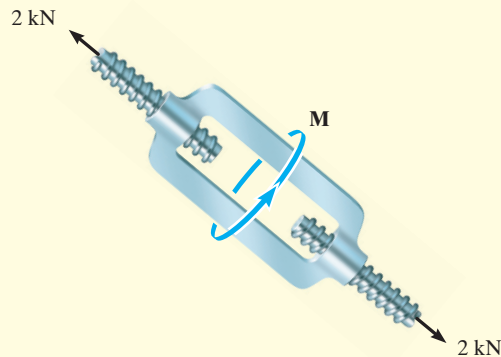
Fig. 8-16

**EXAMPLE 8.7**

The turnbuckle shown in Fig. 8–17 has a square thread with a mean radius of 5 mm and a lead of 2 mm. If the coefficient of static friction between the screw and the turnbuckle is  $\mu_s = 0.25$ , determine the moment  $\mathbf{M}$  that must be applied to draw the end screws closer together.



(© Russell C. Hibbeler)



**Fig. 8–17**

**SOLUTION**

The moment can be obtained by applying Eq. 8–3. Since friction at *two screws* must be overcome, this requires

$$M = 2[rW \tan(\theta + \phi_s)] \quad (1)$$

Here  $W = 2000 \text{ N}$ ,  $\phi_s = \tan^{-1}\mu_s = \tan^{-1}(0.25) = 14.04^\circ$ ,  $r = 5 \text{ mm}$ , and  $\theta = \tan^{-1}(l/2\pi r) = \tan^{-1}(2 \text{ mm}/[2\pi(5 \text{ mm})]) = 3.64^\circ$ . Substituting these values into Eq. 1 and solving gives

$$\begin{aligned} M &= 2[(2000 \text{ N})(5 \text{ mm}) \tan(14.04^\circ + 3.64^\circ)] \\ &= 6374.7 \text{ N} \cdot \text{mm} = 6.37 \text{ N} \cdot \text{m} \end{aligned} \quad \text{Ans.}$$

**NOTE:** When the moment is *removed*, the turnbuckle will be self-locking; i.e., it will not unscrew since  $\phi_s > \theta$ .

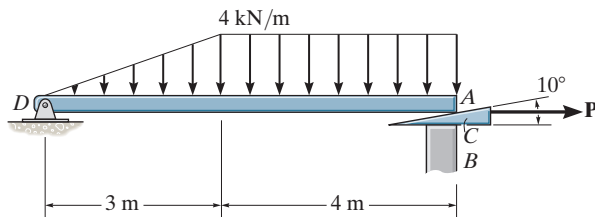
## PROBLEMS

**8-58.** Determine the largest angle  $\theta$  that will cause the wedge to be self-locking regardless of the magnitude of horizontal force  $P$  applied to the blocks. The coefficient of static friction between the wedge and the blocks is  $\mu_s = 0.3$ . Neglect the weight of the wedge.



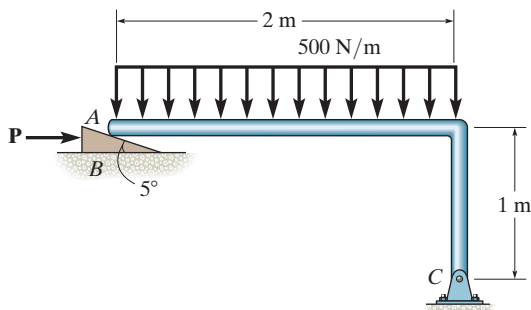
Prob. 8-58

**8-59.** If the beam  $AD$  is loaded as shown, determine the horizontal force  $P$  which must be applied to the wedge in order to remove it from under the beam. The coefficients of static friction at the wedge's top and bottom surfaces are  $\mu_{CA} = 0.25$  and  $\mu_{CB} = 0.35$ , respectively. If  $P = 0$ , is the wedge self-locking? Neglect the weight and size of the wedge and the thickness of the beam.



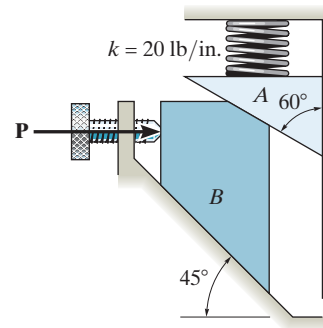
Prob. 8-59

**\*8-60.** The wedge is used to level the member. Determine the horizontal force  $P$  that must be applied to begin to push the wedge forward. The coefficient of static friction between the wedge and the two surfaces of contact is  $\mu_s = 0.2$ . Neglect the weight of the wedge.



Prob. 8-60

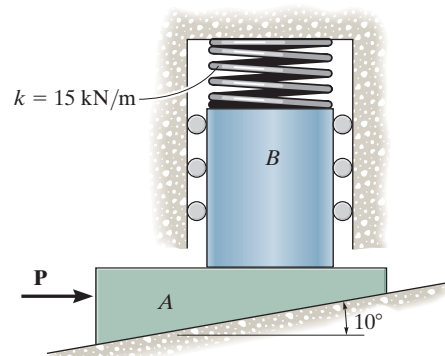
**8-61.** The two blocks used in a measuring device have negligible weight. If the spring is compressed 5 in. when in the position shown, determine the smallest axial force  $P$  which the adjustment screw must exert on  $B$  in order to start the movement of  $B$  downward. The end of the screw is *smooth* and the coefficient of static friction at all other points of contact is  $\mu_s = 0.3$ .



Prob. 8-61

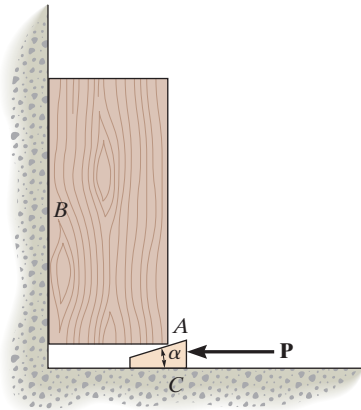
**8-62.** If  $P = 250$  N, determine the required minimum compression in the spring so that the wedge will not move to the right. Neglect the weight of  $A$  and  $B$ . The coefficient of static friction for all contacting surfaces is  $\mu_s = 0.35$ . Neglect friction at the rollers.

**8-63.** Determine the minimum applied force  $P$  required to move wedge  $A$  to the right. The spring is compressed a distance of 175 mm. Neglect the weight of  $A$  and  $B$ . The coefficient of static friction for all contacting surfaces is  $\mu_s = 0.35$ . Neglect friction at the rollers.



Probs. 8-62/63

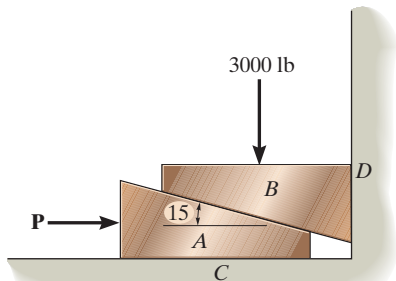
**\*8-64.** If the coefficient of static friction between all the surfaces of contact is  $\mu_s$ , determine the force  $\mathbf{P}$  that must be applied to the wedge in order to lift the block having a weight  $W$ .



**Prob. 8-64**

**8-65.** Determine the smallest force  $P$  needed to lift the 3000-lb load. The coefficient of static friction between  $A$  and  $C$  and between  $B$  and  $D$  is  $\mu_s = 0.3$ , and between  $A$  and  $B$   $\mu_s' = 0.4$ . Neglect the weight of each wedge.

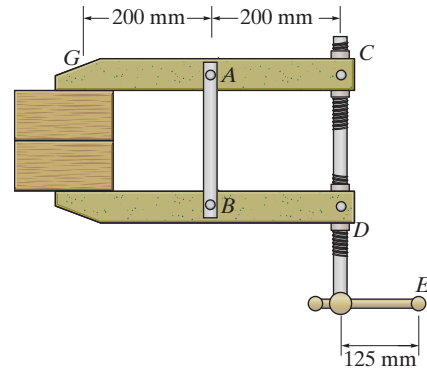
**8-66.** Determine the reversed horizontal force  $-\mathbf{P}$  needed to pull out wedge  $A$ . The coefficient of static friction between  $A$  and  $C$  and between  $B$  and  $D$  is  $\mu_s = 0.2$ , and between  $A$  and  $B$   $\mu_s' = 0.1$ . Neglect the weight of each wedge.



**Probs. 8-65/66**

**8-67.** If the clamping force at  $G$  is 900 N, determine the horizontal force  $\mathbf{F}$  that must be applied perpendicular to the handle of the lever at  $E$ . The mean diameter and lead of both single square-threaded screws at  $C$  and  $D$  are 25 mm and 5 mm, respectively. The coefficient of static friction is  $\mu_s = 0.3$ .

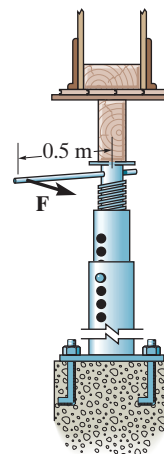
**\*8-68.** If a horizontal force of  $F = 50$  N is applied perpendicular to the handle of the lever at  $E$ , determine the clamping force developed at  $G$ . The mean diameter and lead of the single square-threaded screw at  $C$  and  $D$  are 25 mm and 5 mm, respectively. The coefficient of static friction is  $\mu_s = 0.3$ .



**Probs. 8-67/68**

**8-69.** The column is used to support the upper floor. If a force  $F = 80$  N is applied perpendicular to the handle to tighten the screw, determine the compressive force in the column. The square-threaded screw on the jack has a coefficient of static friction of  $\mu_s = 0.4$ , mean diameter of 25 mm, and a lead of 3 mm.

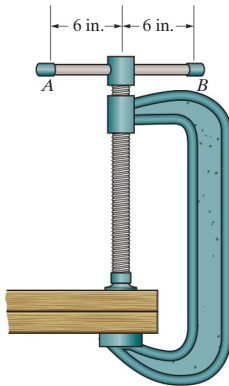
**8-70.** If the force  $\mathbf{F}$  is removed from the handle of the jack in Prob. 8-69, determine if the screw is self-locking.



**Probs. 8-69/70**

**8-71.** If couple forces of  $F = 10$  lb are applied perpendicular to the lever of the clamp at  $A$  and  $B$ , determine the clamping force on the boards. The single square-threaded screw of the clamp has a mean diameter of 1 in. and a lead of 0.25 in. The coefficient of static friction is  $\mu_s = 0.3$ .

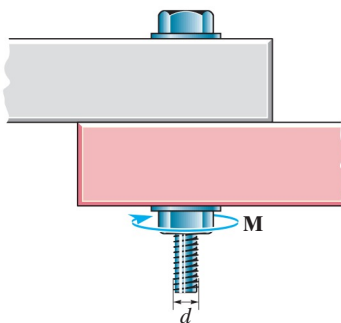
**\*8-72.** If the clamping force on the boards is 600 lb, determine the required magnitude of the couple forces that must be applied perpendicular to the lever  $AB$  of the clamp at  $A$  and  $B$  in order to loosen the screw. The single square-threaded screw has a mean diameter of 1 in. and a lead of 0.25 in. The coefficient of static friction is  $\mu_s = 0.3$ .



**Probs. 8-71/72**

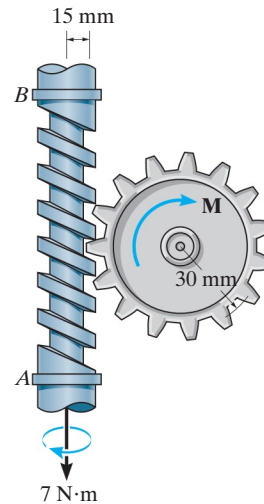
**8-73.** Prove that the lead  $l$  must be less than  $2\pi r\mu_s$  for the jack screw shown in Fig. 8-15 to be “self-locking.”

**8-74.** The square-threaded bolt is used to join two plates together. If the bolt has a mean diameter of  $d = 20$  mm and a lead of  $l = 3$  mm, determine the smallest torque  $M$  required to loosen the bolt if the tension in the bolt is  $T = 40$  kN. The coefficient of static friction between the threads and the bolt is  $\mu_s = 0.15$ .



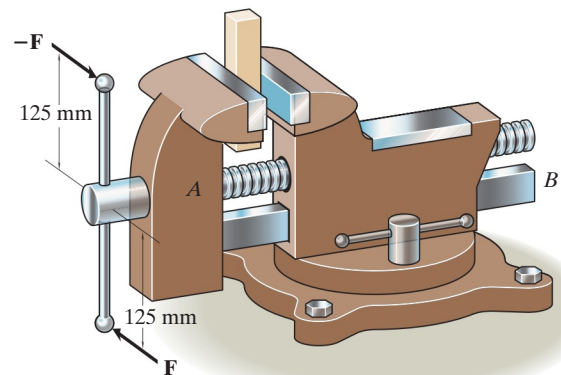
**Prob. 8-74**

**8-75.** The shaft has a square-threaded screw with a lead of 8 mm and a mean radius of 15 mm. If it is in contact with a plate gear having a mean radius of 30 mm, determine the resisting torque  $M$  on the plate gear which can be overcome if a torque of  $7 \text{ N}\cdot\text{m}$  is applied to the shaft. The coefficient of static friction at the screw is  $\mu_B = 0.2$ . Neglect friction of the bearings located at  $A$  and  $B$ .



**Prob. 8-75**

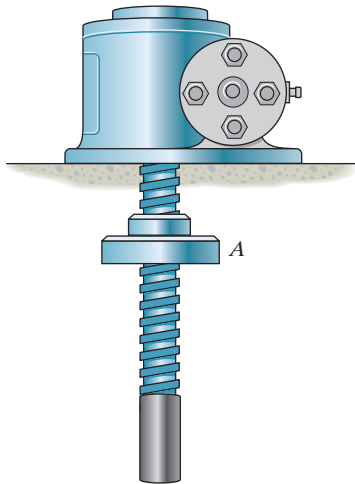
**\*8-76.** If couple forces of  $F = 35$  N are applied to the handle of the machinist's vise, determine the compressive force developed in the block. Neglect friction at the bearing  $A$ . The guide at  $B$  is smooth. The single square-threaded screw has a mean radius of 6 mm and a lead of 8 mm, and the coefficient of static friction is  $\mu_s = 0.27$ .



**Prob. 8-76**

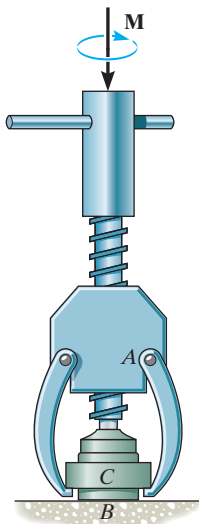


**8-77.** The square-threaded screw has a mean diameter of 20 mm and a lead of 4 mm. If the weight of the plate  $A$  is 5 lb, determine the smallest coefficient of static friction between the screw and the plate so that the plate does not travel down the screw when the plate is suspended as shown.



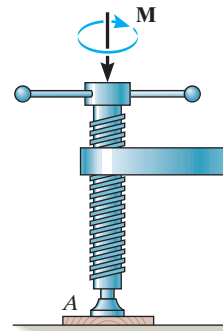
**Prob. 8-77**

**8-78.** The device is used to pull the battery cable terminal  $C$  from the post of a battery. If the required pulling force is 85 lb, determine the torque  $M$  that must be applied to the handle on the screw to tighten it. The screw has square threads, a mean diameter of 0.2 in., a lead of 0.08 in., and the coefficient of static friction is  $\mu_s = 0.5$ .



**Prob. 8-78**

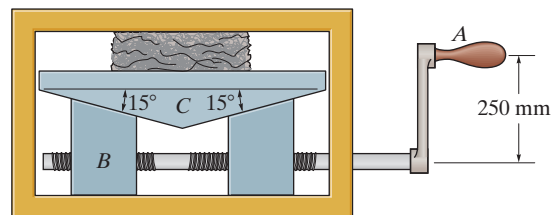
**8-79.** Determine the clamping force on the board  $A$  if the screw is tightened with a torque of  $M = 8 \text{ N}\cdot\text{m}$ . The square-threaded screw has a mean radius of 10 mm and a lead of 3 mm, and the coefficient of static friction is  $\mu_s = 0.35$ .



**Probs. 8-79/80**

**8-81.** If a horizontal force of  $P = 100 \text{ N}$  is applied perpendicular to the handle of the lever at  $A$ , determine the compressive force  $F$  exerted on the material. Each single square-threaded screw has a mean diameter of 25 mm and a lead of 7.5 mm. The coefficient of static friction at all contacting surfaces of the wedges is  $\mu_s = 0.2$ , and the coefficient of static friction at the screw is  $\mu'_s = 0.15$ .

**8-82.** Determine the horizontal force  $P$  that must be applied perpendicular to the handle of the lever at  $A$  in order to develop a compressive force of 12 kN on the material. Each single square-threaded screw has a mean diameter of 25 mm and a lead of 7.5 mm. The coefficient of static friction at all contacting surfaces of the wedges is  $\mu_s = 0.2$ , and the coefficient of static friction at the screw is  $\mu'_s = 0.15$ .



**Probs. 8-81/82**

## 8.5 Frictional Forces on Flat Belts

Whenever belt drives or band brakes are designed, it is necessary to determine the frictional forces developed between the belt and its contacting surface. In this section we will analyze the frictional forces acting on a flat belt, although the analysis of other types of belts, such as the V-belt, is based on similar principles.

Consider the flat belt shown in Fig. 8–18*a*, which passes over a fixed curved surface. The total angle of belt-to-surface contact in radians is  $\beta$ , and the coefficient of friction between the two surfaces is  $\mu$ . We wish to determine the tension  $T_2$  in the belt, which is needed to pull the belt counterclockwise over the surface, and thereby overcome both the frictional forces at the surface of contact and the tension  $T_1$  in the other end of the belt. Obviously,  $T_2 > T_1$ .

**Frictional Analysis.** A free-body diagram of the belt segment in contact with the surface is shown in Fig. 8–18*b*. As shown, the normal and frictional forces, acting at different points along the belt, will vary both in magnitude and direction. Due to this *unknown* distribution, the analysis of the problem will first require a study of the forces acting on a differential element of the belt.

A free-body diagram of an element having a length  $ds$  is shown in Fig. 8–18*c*. Assuming either impending motion or motion of the belt, the magnitude of the frictional force  $dF = \mu dN$ . This force opposes the sliding motion of the belt, and so it will increase the magnitude of the tensile force acting in the belt by  $dT$ . Applying the two force equations of equilibrium, we have

$$\downarrow + \Sigma F_x = 0; \quad T \cos\left(\frac{d\theta}{2}\right) + \mu dN - (T + dT) \cos\left(\frac{d\theta}{2}\right) = 0$$

$$+ \nearrow \Sigma F_y = 0; \quad dN - (T + dT) \sin\left(\frac{d\theta}{2}\right) - T \sin\left(\frac{d\theta}{2}\right) = 0$$

Since  $d\theta$  is of *infinitesimal size*,  $\sin(d\theta/2) = d\theta/2$  and  $\cos(d\theta/2) = 1$ . Also, the *product* of the two infinitesimals  $dT$  and  $d\theta/2$  may be neglected when compared to infinitesimals of the first order. As a result, these two equations become

$$\mu dN = dT$$

and

$$dN = T d\theta$$

Eliminating  $dN$  yields

$$\frac{dT}{T} = \mu d\theta$$

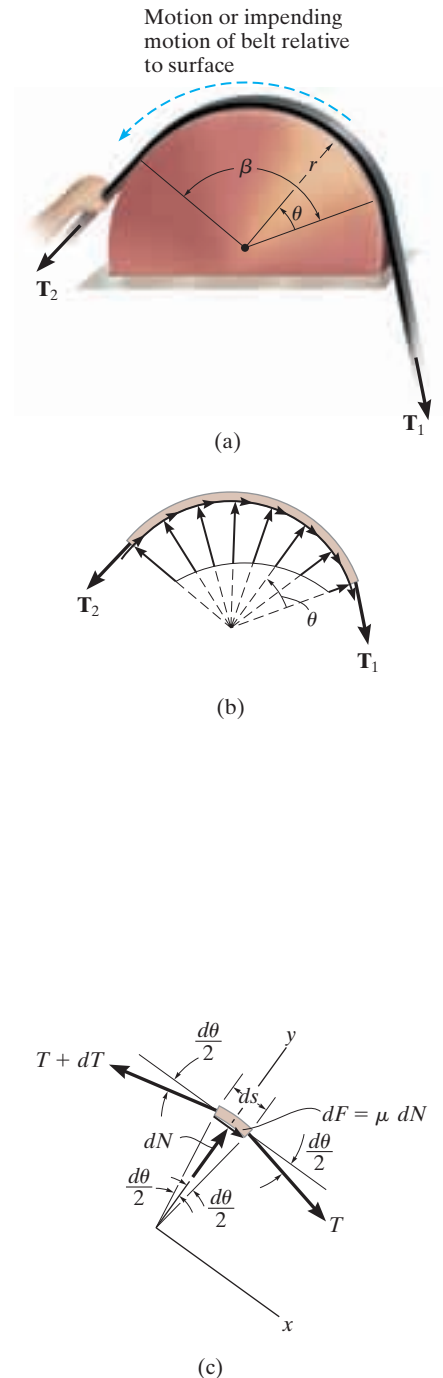


Fig. 8–18

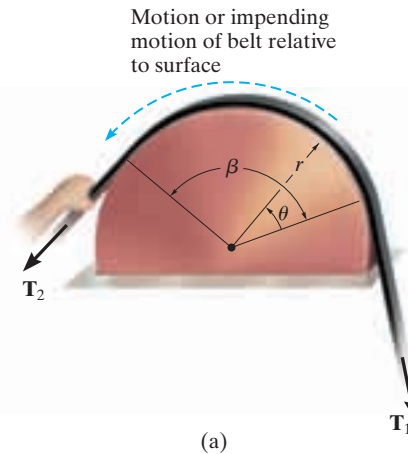


Fig. 8-18 (Repeated)



Flat or V-belts are often used to transmit the torque developed by a motor to a wheel attached to a pump, fan, or blower. (© Russell C. Hibbeler)

Integrating this equation between all the points of contact that the belt makes with the drum, and noting that  $T = T_1$  at  $\theta = 0$  and  $T = T_2$  at  $\theta = \beta$ , yields

$$\int_{T_1}^{T_2} \frac{dT}{T} = \mu \int_0^\beta d\theta$$

$$\ln \frac{T_2}{T_1} = \mu\beta$$

Solving for  $T_2$ , we obtain

$$T_2 = T_1 e^{\mu\beta} \quad (8-6)$$

where

$T_2, T_1$  = belt tensions;  $T_1$  opposes the direction of motion (or impending motion) of the belt measured relative to the surface, while  $T_2$  acts in the direction of the relative belt motion (or impending motion); because of friction,  $T_2 > T_1$

$\mu$  = coefficient of static or kinetic friction between the belt and the surface of contact

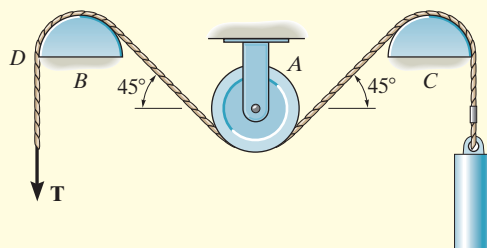
$\beta$  = angle of belt-to-surface contact, measured in radians

$e = 2.718 \dots$ , base of the natural logarithm

Note that  $T_2$  is *independent* of the *radius* of the drum, and instead it is a function of the angle of belt to surface contact,  $\beta$ . As a result, this equation is valid for flat belts passing over any curved contacting surface.

**EXAMPLE 8.8**

The maximum tension that can be developed in the cord shown in Fig. 8–19a is 500 N. If the pulley at  $A$  is free to rotate and the coefficient of static friction at the fixed drums  $B$  and  $C$  is  $\mu_s = 0.25$ , determine the largest mass of the cylinder that can be lifted by the cord.



(a)

**SOLUTION**

Lifting the cylinder, which has a weight  $W = mg$ , causes the cord to move counterclockwise over the drums at  $B$  and  $C$ ; hence, the maximum tension  $T_2$  in the cord occurs at  $D$ . Thus,  $F = T_2 = 500$  N. A section of the cord passing over the drum at  $B$  is shown in Fig. 8–19b. Since  $180^\circ = \pi$  rad the angle of contact between the drum and the cord is  $\beta = (135^\circ/180^\circ)\pi = 3\pi/4$  rad. Using Eq. 8–6, we have

$$T_2 = T_1 e^{\mu_s \beta}; \quad 500 \text{ N} = T_1 e^{0.25[(3/4)\pi]}$$

Hence,

$$T_1 = \frac{500 \text{ N}}{e^{0.25[(3/4)\pi]}} = \frac{500 \text{ N}}{1.80} = 277.4 \text{ N}$$

Since the pulley at  $A$  is free to rotate, equilibrium requires that the tension in the cord remains the *same* on both sides of the pulley.

The section of the cord passing over the drum at  $C$  is shown in Fig. 8–19c. The weight  $W < 277.4$  N. Why? Applying Eq. 8–6, we obtain

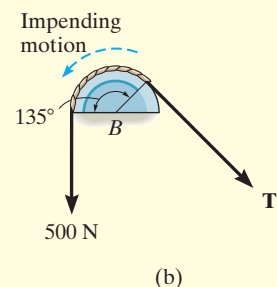
$$T_2 = T_1 e^{\mu_s \beta}; \quad 277.4 \text{ N} = W e^{0.25[(3/4)\pi]}$$

$$W = 153.9 \text{ N}$$

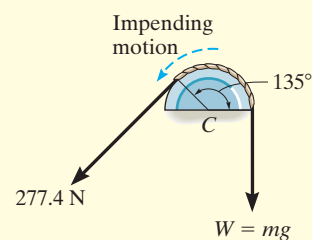
so that

$$\begin{aligned} m &= \frac{W}{g} = \frac{153.9 \text{ N}}{9.81 \text{ m/s}^2} \\ &= 15.7 \text{ kg} \end{aligned}$$

*Ans.*



(b)



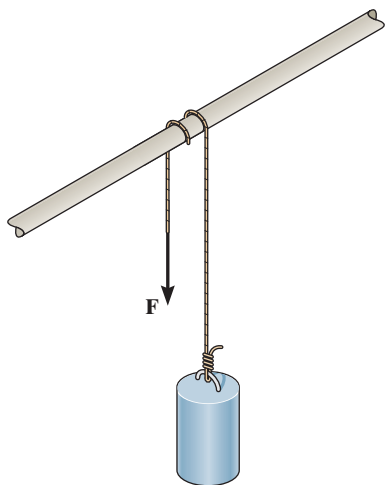
(c)

**Fig. 8–19**

## PROBLEMS

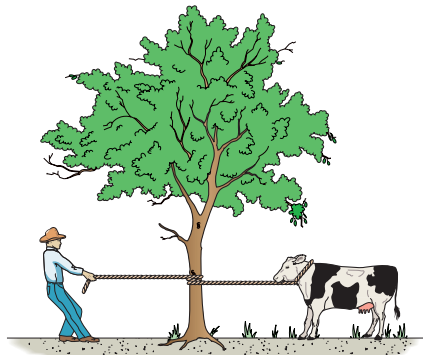
**8–83.** A cylinder having a mass of 250 kg is to be supported by the cord that wraps over the pipe. Determine the smallest vertical force  $\mathbf{F}$  needed to support the load if the cord passes (a) once over the pipe,  $\beta = 180^\circ$ , and (b) two times over the pipe,  $\beta = 540^\circ$ . Take  $\mu_s = 0.2$ .

**\*8–84.** A cylinder having a mass of 250 kg is to be supported by the cord that wraps over the pipe. Determine the largest vertical force  $\mathbf{F}$  that can be applied to the cord without moving the cylinder. The cord passes (a) once over the pipe,  $\beta = 180^\circ$ , and (b) two times over the pipe,  $\beta = 540^\circ$ . Take  $\mu_s = 0.2$ .



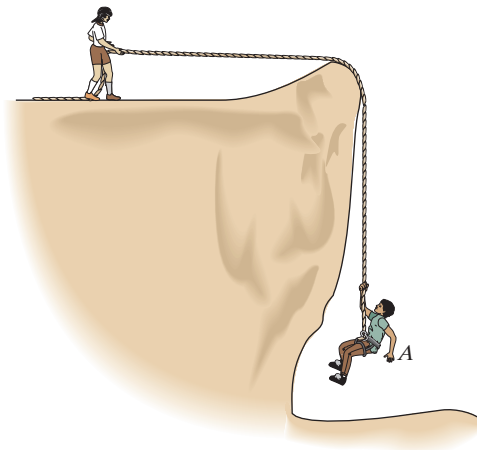
**Probs. 8–83/84**

**8–85.** A 180-lb farmer tries to restrain the cow from escaping by wrapping the rope two turns around the tree trunk as shown. If the cow exerts a force of 250 lb on the rope, determine if the farmer can successfully restrain the cow. The coefficient of static friction between the rope and the tree trunk is  $\mu_s = 0.15$ , and between the farmer's shoes and the ground  $\mu'_s = 0.3$ .



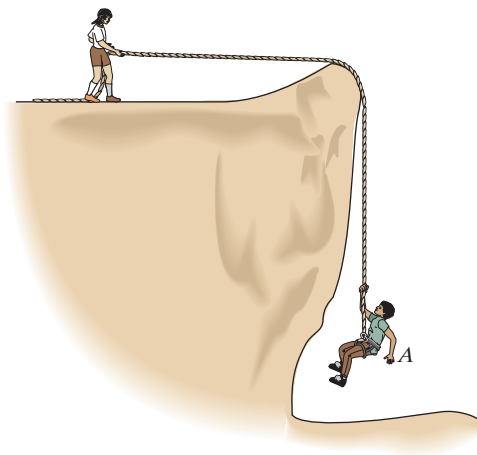
**Prob. 8–85**

**8–86.** The 100-lb boy at  $A$  is suspended from the cable that passes over the quarter circular cliff rock. Determine if it is possible for the 185-lb woman to hoist him up; and if this is possible, what smallest force must she exert on the horizontal cable? The coefficient of static friction between the cable and the rock is  $\mu_s = 0.2$ , and between the shoes of the woman and the ground  $\mu'_s = 0.8$ .



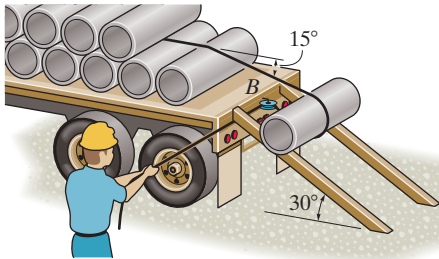
**Prob. 8–86**

**8–87.** The 100-lb boy at  $A$  is suspended from the cable that passes over the quarter circular cliff rock. What horizontal force must the woman at  $A$  exert on the cable in order to let the boy descend at constant velocity? The coefficients of static and kinetic friction between the cable and the rock are  $\mu_s = 0.4$  and  $\mu_k = 0.35$ , respectively.



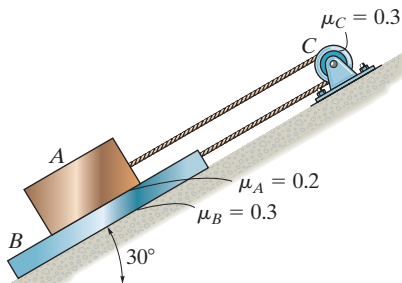
**Prob. 8–87**

**\*8–88.** The uniform concrete pipe has a weight of 800 lb and is unloaded slowly from the truck bed using the rope and skids shown. If the coefficient of kinetic friction between the rope and pipe is  $\mu_k = 0.3$ , determine the force the worker must exert on the rope to lower the pipe at constant speed. There is a pulley at  $B$ , and the pipe does not slip on the skids. The lower portion of the rope is parallel to the skids.



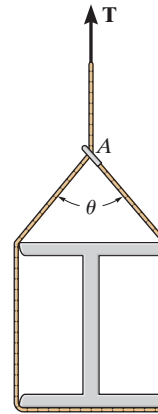
**Prob. 8–88**

**8–89.** A cable is attached to the 20-kg plate  $B$ , passes over a fixed peg at  $C$ , and is attached to the block at  $A$ . Using the coefficients of static friction shown, determine the smallest mass of block  $A$  so that it will prevent sliding motion of  $B$  down the plane.



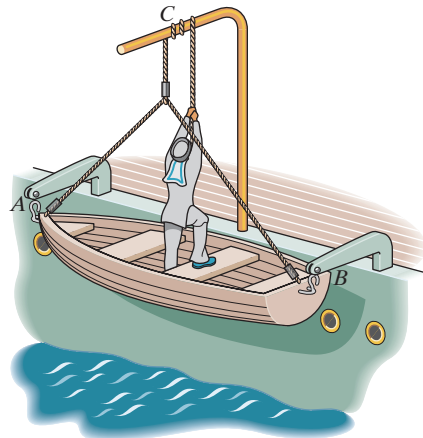
**Prob. 8–89**

**8–90.** The smooth beam is being hoisted using a rope that is wrapped around the beam and passes through a ring at  $A$  as shown. If the end of the rope is subjected to a tension  $T$  and the coefficient of static friction between the rope and ring is  $\mu_s = 0.3$ , determine the smallest angle of  $\theta$  for equilibrium.



**Prob. 8–90**

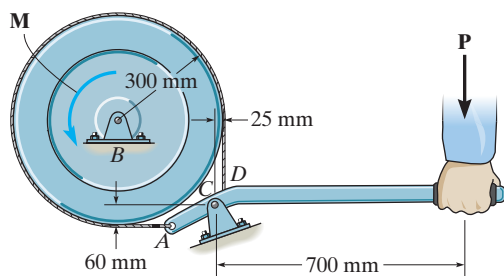
**8–91.** The boat has a weight of 500 lb and is held in position off the side of a ship by the spars at  $A$  and  $B$ . A man having a weight of 130 lb gets in the boat, wraps a rope around an overhead boom at  $C$ , and ties it to the end of the boat as shown. If the boat is disconnected from the spars, determine the *minimum number of half turns* the rope must make around the boom so that the boat can be safely lowered into the water at constant velocity. Also, what is the normal force between the boat and the man? The coefficient of kinetic friction between the rope and the boom is  $\mu_s = 0.15$ . *Hint:* The problem requires that the normal force between the man's feet and the boat be as small as possible.



**Prob. 8–91**

**\*8-92.** Determine the force  $P$  that must be applied to the handle of the lever so that the wheel is on the verge of turning if  $M = 300 \text{ N} \cdot \text{m}$ . The coefficient of static friction between the belt and the wheel is  $\mu_s = 0.3$ .

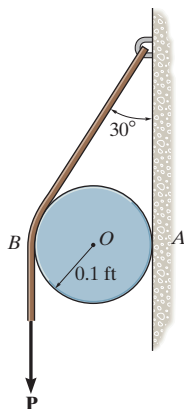
**8-93.** If a force of  $P = 30 \text{ N}$  is applied to the handle of the lever, determine the largest couple moment  $\mathbf{M}$  that can be resisted so that the wheel does not turn. The coefficient of static friction between the belt and the wheel is  $\mu_s = 0.3$ .



Probs. 8-92/93

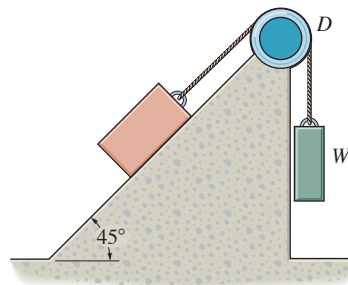
**8-94.** A minimum force of  $P = 50 \text{ lb}$  is required to hold the cylinder from slipping against the belt and the wall. Determine the weight of the cylinder if the coefficient of friction between the belt and cylinder is  $\mu_s = 0.3$  and slipping does not occur at the wall.

**8-95.** The cylinder weighs  $10 \text{ lb}$  and is held in equilibrium by the belt and wall. If slipping does not occur at the wall, determine the minimum vertical force  $P$  which must be applied to the belt for equilibrium. The coefficient of static friction between the belt and the cylinder is  $\mu_s = 0.25$ .



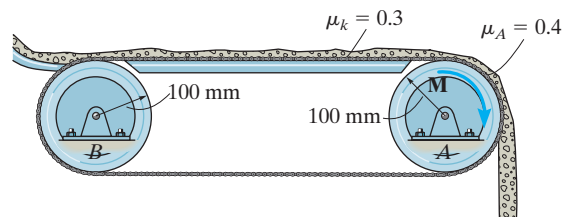
Probs. 8-94/95

**\*8-96.** Determine the maximum and the minimum values of weight  $W$  which may be applied without causing the  $50\text{-lb}$  block to slip. The coefficient of static friction between the block and the plane is  $\mu_s = 0.2$ , and between the rope and the drum  $D$  is  $\mu'_s = 0.3$ .



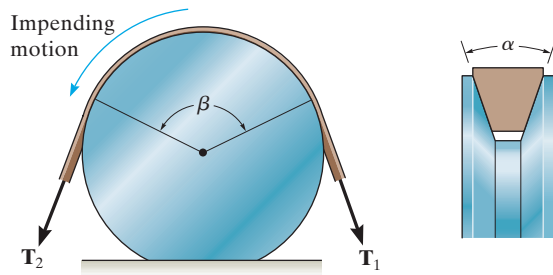
Prob. 8-96

**8-97.** Granular material, having a density of  $1.5 \text{ Mg/m}^3$ , is transported on a conveyor belt that slides over the fixed surface, having a coefficient of kinetic friction of  $\mu_k = 0.3$ . Operation of the belt is provided by a motor that supplies a torque  $\mathbf{M}$  to wheel  $A$ . The wheel at  $B$  is free to turn, and the coefficient of static friction between the wheel at  $A$  and the belt is  $\mu_A = 0.4$ . If the belt is subjected to a pretension of  $300 \text{ N}$  when no load is on the belt, determine the greatest volume  $V$  of material that is permitted on the belt at any time without allowing the belt to stop. What is the torque  $\mathbf{M}$  required to drive the belt when it is subjected to this maximum load?



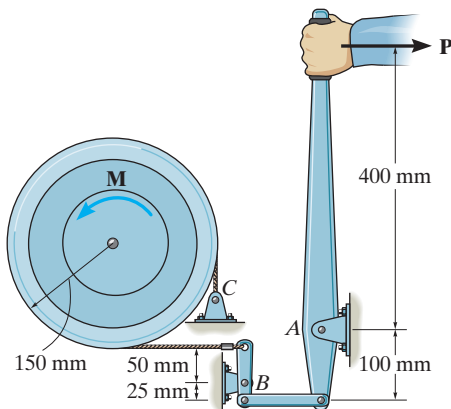
Prob. 8-97

**8–98.** Show that the frictional relationship between the belt tensions, the coefficient of friction  $\mu$ , and the angular contacts  $\alpha$  and  $\beta$  for the V-belt is  $T_2 = T_1 e^{\mu\beta/\sin(\alpha/2)}$ .



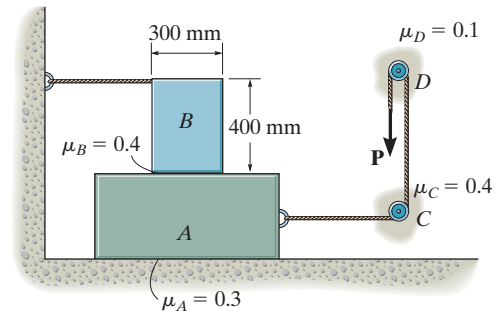
**Prob. 8–98**

**8–99.** The wheel is subjected to a torque of  $M = 50 \text{ N} \cdot \text{m}$ . If the coefficient of kinetic friction between the band brake and the rim of the wheel is  $\mu_k = 0.3$ , determine the smallest horizontal force  $P$  that must be applied to the lever to stop the wheel.



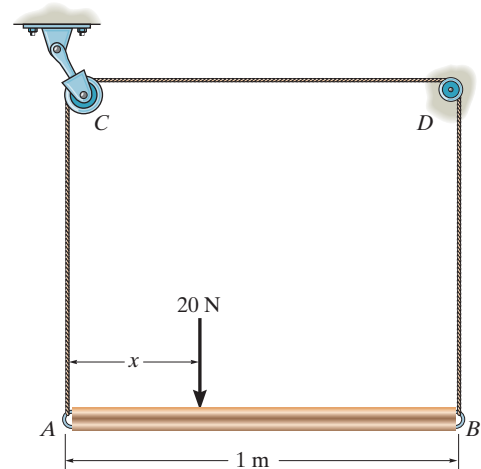
**Prob. 8–99**

**\*8–100.** Blocks  $A$  and  $B$  have a mass of 7 kg and 10 kg, respectively. Using the coefficients of static friction indicated, determine the largest vertical force  $P$  which can be applied to the cord without causing motion.



**Prob. 8–100**

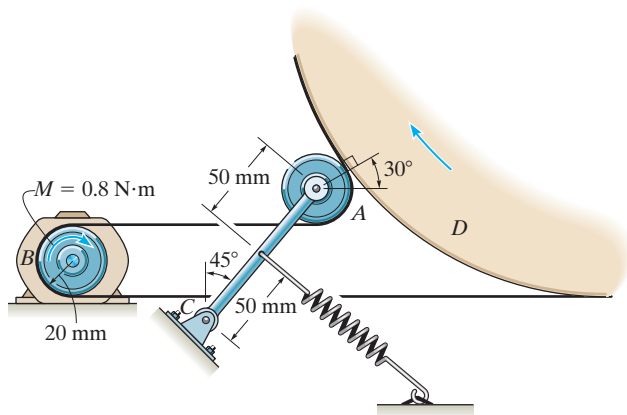
**8–101.** The uniform bar  $AB$  is supported by a rope that passes over a frictionless pulley at  $C$  and a fixed peg at  $D$ . If the coefficient of static friction between the rope and the peg is  $\mu_D = 0.3$ , determine the smallest distance  $x$  from the end of the bar at which a 20-N force may be placed and not cause the bar to move.



**Prob. 8–101**

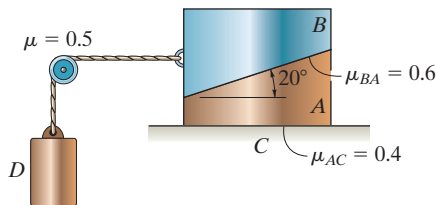


**8-102.** The belt on the portable dryer wraps around the drum  $D$ , idler pulley  $A$ , and motor pulley  $B$ . If the motor can develop a maximum torque of  $M = 0.80 \text{ N} \cdot \text{m}$ , determine the smallest spring tension required to hold the belt from slipping. The coefficient of static friction between the belt and the drum and motor pulley is  $\mu_s = 0.3$ .



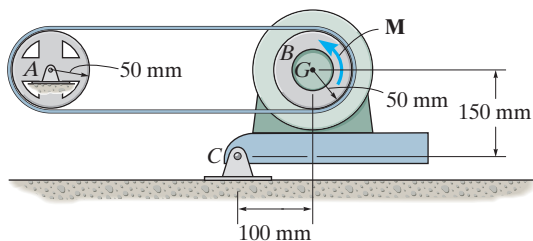
**Prob. 8-102**

**8-103.** Blocks  $A$  and  $B$  weigh 50 lb and 30 lb, respectively. Using the coefficients of static friction indicated, determine the greatest weight of block  $D$  without causing motion.



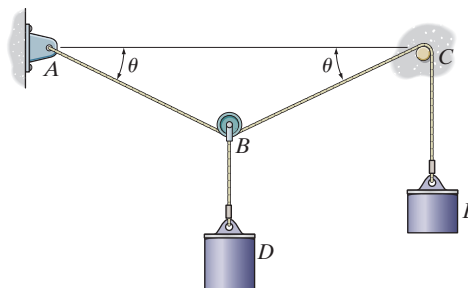
**Prob. 8-103**

**\*8-104.** The 20-kg motor has a center of gravity at  $G$  and is pin connected at  $C$  to maintain a tension in the drive belt. Determine the smallest counterclockwise twist or torque  $\mathbf{M}$  that must be supplied by the motor to turn the disk  $B$  if wheel  $A$  locks and causes the belt to slip over the disk. No slipping occurs at  $A$ . The coefficient of static friction between the belt and the disk is  $\mu_s = 0.3$ .



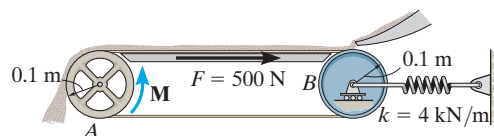
**Prob. 8-104**

**8-105.** A 10-kg cylinder  $D$ , which is attached to a small pulley  $B$ , is placed on the cord as shown. Determine the largest angles  $\theta$  so that the cord does not slip over the peg at  $C$ . The cylinder at  $E$  also has a mass of 10 kg, and the coefficient of static friction between the cord and the peg is  $\mu_s = 0.1$ .



**Prob. 8-105**

**8-106.** A conveyor belt is used to transfer granular material and the frictional resistance on the top of the belt is  $F = 500 \text{ N}$ . Determine the smallest stretch of the spring attached to the moveable axle of the idle pulley  $B$  so that the belt does not slip at the drive pulley  $A$  when the torque  $\mathbf{M}$  is applied. What minimum torque  $\mathbf{M}$  is required to keep the belt moving? The coefficient of static friction between the belt and the wheel at  $A$  is  $\mu_s = 0.2$ .



**Prob. 8-106**

## \*8.6 Frictional Forces on Collar Bearings, Pivot Bearings, and Disks

**Pivot** and **collar bearings** are commonly used in machines to support an *axial load* on a rotating shaft. Typical examples are shown in Fig. 8–20. Provided these bearings are not lubricated, or are only partially lubricated, the laws of dry friction may be applied to determine the moment needed to turn the shaft when it supports an axial force.

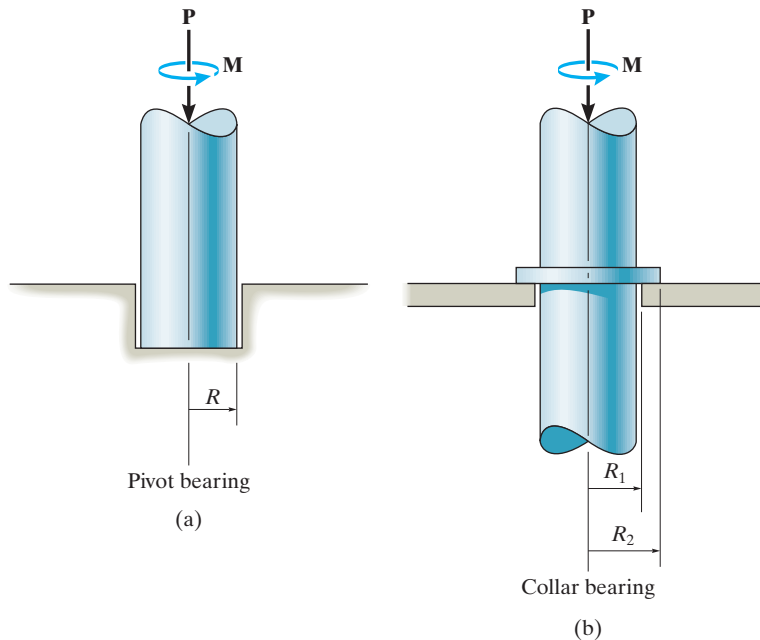


Fig. 8–20

**Frictional Analysis.** The collar bearing on the shaft shown in Fig. 8–21 is subjected to an axial force  $\mathbf{P}$  and has a total bearing or contact area  $\pi(R_2^2 - R_1^2)$ . Provided the bearing is new and evenly supported, then the normal pressure  $p$  on the bearing will be *uniformly distributed* over this area. Since  $\Sigma F_z = 0$ , then  $p$ , measured as a force per unit area, is  $p = P/\pi(R_2^2 - R_1^2)$ .

The moment needed to cause impending rotation of the shaft can be determined from moment equilibrium about the  $z$  axis. A differential area element  $dA = (r d\theta)(dr)$ , shown in Fig. 8–21, is subjected to both a normal force  $dN = p dA$  and an associated frictional force,

$$dF = \mu_s dN = \mu_s p dA = \frac{\mu_s P}{\pi(R_2^2 - R_1^2)} dA$$

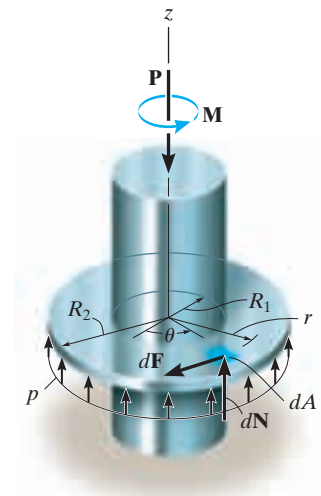


Fig. 8–21

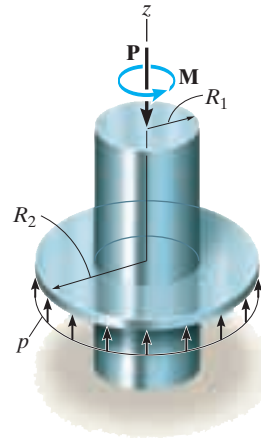


Fig. 8-21 (Repeated)

The normal force does not create a moment about the  $z$  axis of the shaft; however, the frictional force does; namely,  $dM = r dF$ . Integration is needed to compute the applied moment  $\mathbf{M}$  needed to overcome all the frictional forces. Therefore, for impending rotational motion,

$$\Sigma M_z = 0; \quad M - \int_A r dF = 0$$

Substituting for  $dF$  and  $dA$  and integrating over the entire bearing area yields

$$M = \int_{R_1}^{R_2} \int_0^{2\pi} r \left[ \frac{\mu_s P}{\pi(R_2^2 - R_1^2)} \right] (r d\theta dr) = \frac{\mu_s P}{\pi(R_2^2 - R_1^2)} \int_{R_1}^{R_2} r^2 dr \int_0^{2\pi} d\theta$$

or

$$M = \frac{2}{3} \mu_s P \left( \frac{R_2^3 - R_1^3}{R_2^2 - R_1^2} \right) \quad (8-7)$$

The moment developed at the end of the shaft, when it is *rotating* at constant speed, can be found by substituting  $\mu_k$  for  $\mu_s$  in Eq. 8-7.

In the case of a pivot bearing, Fig. 8-20a, then  $R_2 = R$  and  $R_1 = 0$ , and Eq. 8-7 reduces to

$$M = \frac{2}{3} \mu_s PR \quad (8-8)$$

Remember that Eqs. 8-7 and 8-8 apply only for bearing surfaces subjected to *constant pressure*. If the pressure is not uniform, a variation of the pressure as a function of the bearing area must be determined before integrating to obtain the moment. The following example illustrates this concept.



The motor that turns the disk of this sanding machine develops a torque that must overcome the frictional forces acting on the disk. (© Russell C. Hibbeler)

**EXAMPLE 8.9**

The uniform bar shown in Fig. 8–22a has a weight of 4 lb. If it is assumed that the normal pressure acting at the contacting surface varies linearly along the length of the bar as shown, determine the couple moment  $\mathbf{M}$  required to rotate the bar. Assume that the bar's width is negligible in comparison to its length. The coefficient of static friction is equal to  $\mu_s = 0.3$ .

**SOLUTION**

A free-body diagram of the bar is shown in Fig. 8–22b. The intensity  $w_0$  of the distributed load at the center ( $x = 0$ ) is determined from vertical force equilibrium, Fig. 8–22a.

$$+\uparrow \Sigma F_z = 0; \quad -4 \text{ lb} + 2 \left[ \frac{1}{2} (2 \text{ ft}) w_0 \right] = 0 \quad w_0 = 2 \text{ lb/ft}$$

Since  $w = 0$  at  $x = 2 \text{ ft}$ , the distributed load expressed as a function of  $x$  is

$$w = (2 \text{ lb/ft}) \left( 1 - \frac{x}{2 \text{ ft}} \right) = 2 - x$$

The magnitude of the normal force acting on a differential segment of area having a length  $dx$  is therefore

$$dN = w dx = (2 - x) dx$$

The magnitude of the frictional force acting on the same element of area is

$$dF = \mu_s dN = 0.3(2 - x) dx$$

Hence, the moment created by this force about the  $z$  axis is

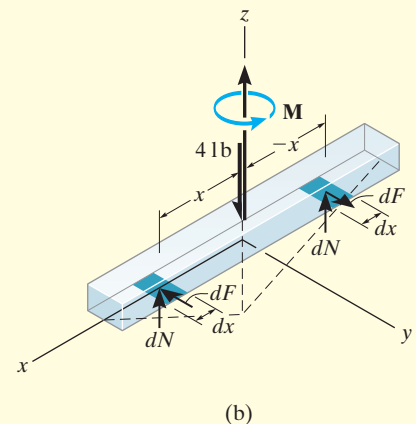
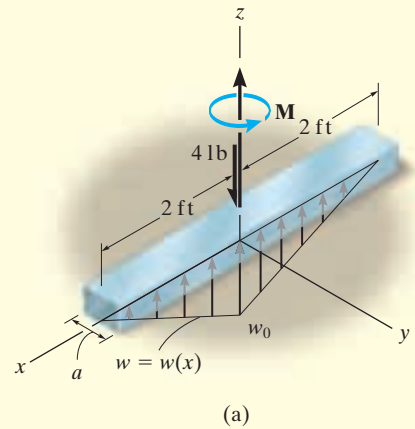
$$dM = x dF = 0.3(2x - x^2) dx$$

The summation of moments about the  $z$  axis of the bar is determined by integration, which yields

$$\Sigma M_z = 0; \quad M - 2 \int_0^2 (0.3)(2x - x^2) dx = 0$$

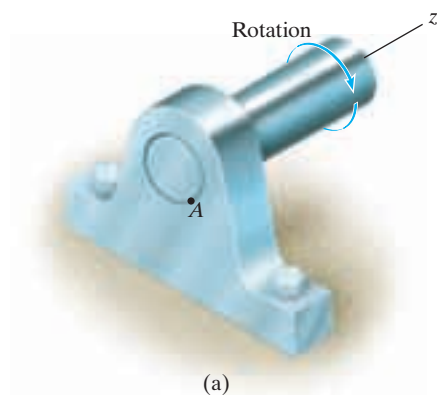
$$M = 0.6 \left( x^2 - \frac{x^3}{3} \right) \bigg|_0^2$$

$$M = 0.8 \text{ lb} \cdot \text{ft}$$

*Ans.***Fig. 8–22**



Unwinding the cable from this spool requires overcoming friction from the supporting shaft. (© Russell C. Hibbeler)



(a)

Fig. 8–23

## 8.7 Frictional Forces on Journal Bearings

When a shaft or axle is subjected to lateral loads, a **journal bearing** is commonly used for support. Provided the bearing is not lubricated, or is only partially lubricated, a reasonable analysis of the frictional resistance on the bearing can be based on the laws of dry friction.

**Frictional Analysis.** A typical journal-bearing support is shown in Fig. 8–23a. As the shaft rotates, the contact point moves up the wall of the bearing to some point *A* where slipping occurs. If the vertical load acting at the end of the shaft is **P**, then the bearing reactive force **R** acting at *A* will be equal and opposite to **P**, Fig. 8–23b. The moment needed to maintain constant rotation of the shaft can be found by summing moments about the *z* axis of the shaft; i.e.,

$$\Sigma M_z = 0; \quad M - (R \sin \phi_k)r = 0$$

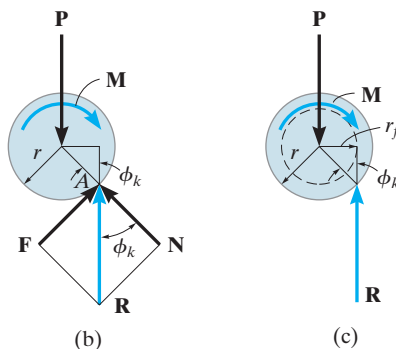
or

$$M = Rr \sin \phi_k \quad (8-9)$$

where  $\phi_k$  is the angle of kinetic friction defined by  $\tan \phi_k = F/N = \mu_k N/N = \mu_k$ . In Fig. 8–23c, it is seen that  $r \sin \phi_k = r_f$ . The dashed circle with radius  $r_f$  is called the **friction circle**, and as the shaft rotates, the reaction **R** will always be tangent to it. If the bearing is partially lubricated,  $\mu_k$  is small, and therefore  $\sin \phi_k \approx \tan \phi_k \approx \mu_k$ . Under these conditions, a reasonable *approximation* to the moment needed to overcome the frictional resistance becomes

$$M \approx Rr\mu_k \quad (8-10)$$

Notice that to minimize friction the bearing radius *r* should be as small as possible. In practice, however, this type of journal bearing is not suitable for long service since friction between the shaft and bearing will eventually wear down the surfaces. Instead, designers will incorporate “ball bearings” or “rollers” in journal bearings to minimize frictional losses.

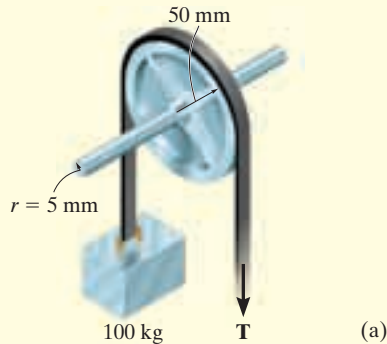


(b)

(c)

**EXAMPLE 8.10**

The 100-mm-diameter pulley shown in Fig. 8–24a fits loosely on a 10-mm-diameter shaft for which the coefficient of static friction is  $\mu_s = 0.4$ . Determine the minimum tension  $T$  in the belt needed to (a) raise the 100-kg block and (b) lower the block. Assume that no slipping occurs between the belt and pulley and neglect the weight of the pulley.

**SOLUTION**

**Part (a).** A free-body diagram of the pulley is shown in Fig. 8–24b. When the pulley is subjected to belt tensions of 981 N each, it makes contact with the shaft at point  $P_1$ . As the tension  $T$  is *increased*, the contact point will move around the shaft to point  $P_2$  before motion impends. From the figure, the friction circle has a radius  $r_f = r \sin \phi_s$ . Using the simplification that  $\sin \phi_s \approx \tan \phi_s \approx \mu_s$  then  $r_f \approx r\mu_s = (5 \text{ mm})(0.4) = 2 \text{ mm}$ , so that summing moments about  $P_2$  gives

$$\zeta + \Sigma M_{P_2} = 0; \quad 981 \text{ N}(52 \text{ mm}) - T(48 \text{ mm}) = 0$$

$$T = 1063 \text{ N} = 1.06 \text{ kN} \quad \text{Ans.}$$

If a more exact analysis is used, then  $\phi_s = \tan^{-1} 0.4 = 21.8^\circ$ . Thus, the radius of the friction circle would be  $r_f = r \sin \phi_s = 5 \sin 21.8^\circ = 1.86 \text{ mm}$ . Therefore,

$$\zeta + \Sigma M_{P_2} = 0;$$

$$981 \text{ N}(50 \text{ mm} + 1.86 \text{ mm}) - T(50 \text{ mm} - 1.86 \text{ mm}) = 0$$

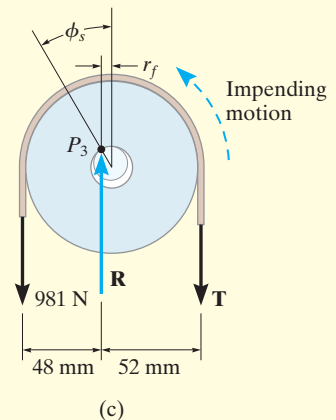
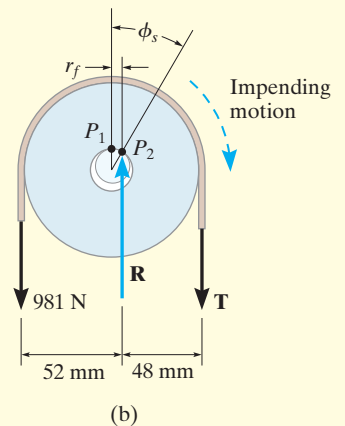
$$T = 1057 \text{ N} = 1.06 \text{ kN} \quad \text{Ans.}$$

**Part (b).** When the block is lowered, the resultant force  $\mathbf{R}$  acting on the shaft passes through point as shown in Fig. 8–24c. Summing moments about this point yields

$$\zeta + \Sigma M_{P_3} = 0; \quad 981 \text{ N}(48 \text{ mm}) - T(52 \text{ mm}) = 0$$

$$T = 906 \text{ N} \quad \text{Ans.}$$

**NOTE:** Using the approximate analysis, the difference between raising and lowering the block is thus 157 N.

**Fig. 8–24**

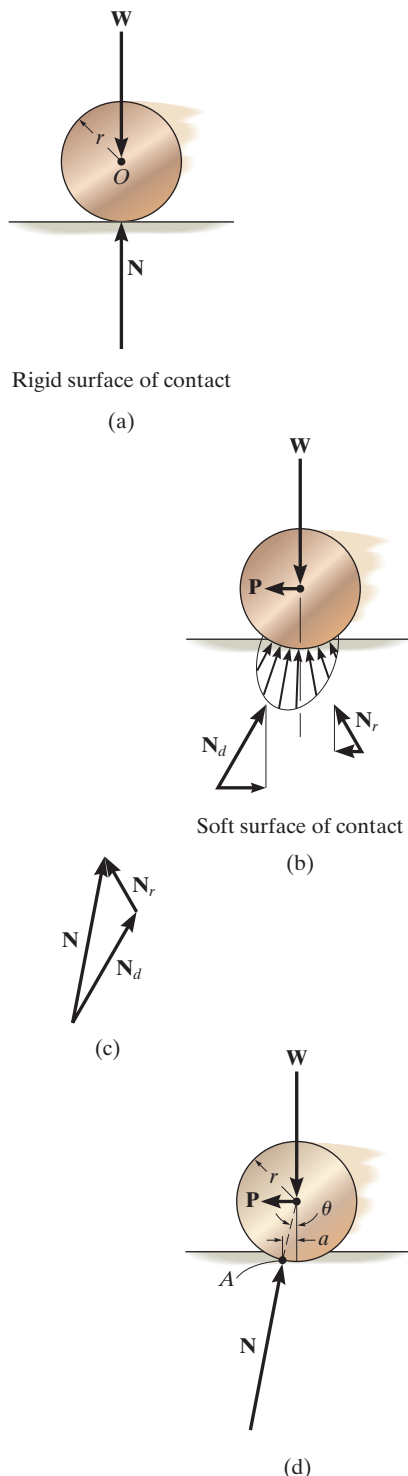


Fig. 8-25

## \*8.8 Rolling Resistance

When a *rigid* cylinder rolls at constant velocity along a *rigid* surface, the normal force exerted by the surface on the cylinder acts perpendicular to the tangent at the point of contact, as shown in Fig. 8-25a. Actually, however, no materials are perfectly rigid, and therefore the reaction of the surface on the cylinder consists of a distribution of normal pressure. For example, consider the cylinder to be made of a very hard material, and the surface on which it rolls to be relatively soft. Due to its weight, the cylinder compresses the surface underneath it, Fig. 8-25b. As the cylinder rolls, the surface material in front of the cylinder *retards* the motion since it is being *deformed*, whereas the material in the rear is *restored* from the deformed state and therefore tends to *push* the cylinder forward. The normal pressures acting on the cylinder in this manner are represented in Fig. 8-25b by their resultant forces  $N_d$  and  $N_r$ . The magnitude of the force of *deformation*,  $N_d$ , and its horizontal component is *always greater* than that of *restoration*,  $N_r$ , and consequently a horizontal driving force  $P$  must be applied to the cylinder to maintain the motion. Fig. 8-25b.\*

Rolling resistance is caused primarily by this effect, although it is also, to a lesser degree, the result of surface adhesion and relative micro-sliding between the surfaces of contact. Because the actual force  $P$  needed to overcome these effects is difficult to determine, a simplified method will be developed here to explain one way engineers have analyzed this phenomenon. To do this, we will consider the resultant of the *entire* normal pressure,  $N = N_d + N_r$ , acting on the cylinder, Fig. 8-25c. As shown in Fig. 8-25d, this force acts at an angle  $\theta$  with the vertical. To keep the cylinder in equilibrium, i.e., rolling at a constant rate, it is necessary that  $N$  be *concurrent* with the driving force  $P$  and the weight  $W$ . Summing moments about point  $A$  gives  $Wa = P(r \cos \theta)$ . Since the deformations are generally very small in relation to the cylinder's radius,  $\cos \theta \approx 1$ ; hence,

$$Wa \approx Pr$$

or

$$P \approx \frac{Wa}{r} \quad (8-11)$$

The distance  $a$  is termed the **coefficient of rolling resistance**, which has the dimension of length. For instance,  $a \approx 0.5$  mm for a wheel rolling on a rail, both of which are made of mild steel. For hardened steel ball

\*Actually, the deformation force  $N_d$  causes *energy* to be stored in the material as its magnitude is increased, whereas the restoration force  $N_r$ , as its magnitude is decreased, allows some of this energy to be released. The remaining energy is *lost* since it is used to heat up the surface, and if the cylinder's weight is very large, it accounts for permanent deformation of the surface. Work must be done by the horizontal force  $P$  to make up for this loss.



bearings on steel,  $a \approx 0.1$  mm. Experimentally, though, this factor is difficult to measure, since it depends on such parameters as the rate of rotation of the cylinder, the elastic properties of the contacting surfaces, and the surface finish. For this reason, little reliance is placed on the data for determining  $a$ . The analysis presented here does, however, indicate why a heavy load ( $W$ ) offers greater resistance to motion ( $P$ ) than a light load under the same conditions. Furthermore, since  $Wa/r$  is generally very small compared to  $\mu_k W$ , the force needed to *roll* a cylinder over the surface will be much less than that needed to *slide* it across the surface. It is for this reason that a roller or ball bearings are often used to minimize the frictional resistance between moving parts.



Rolling resistance of railroad wheels on the rails is small since steel is very stiff. By comparison, the rolling resistance of the wheels of a tractor in a wet field is very large. (© Russell C. Hibbeler)

### EXAMPLE 8.11

A 10-kg steel wheel shown in Fig. 8–26a has a radius of 100 mm and rests on an inclined plane made of soft wood. If  $\theta$  is increased so that the wheel begins to roll down the incline with constant velocity when  $\theta = 1.2^\circ$ , determine the coefficient of rolling resistance.

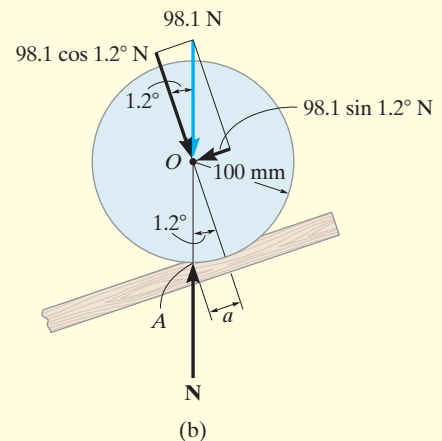
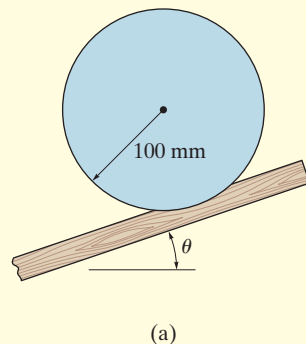


Fig. 8–26

### SOLUTION

As shown on the free-body diagram, Fig. 8–26b, when the wheel has impending motion, the normal reaction  $\mathbf{N}$  acts at point  $A$  defined by the dimension  $a$ . Resolving the weight into components parallel and perpendicular to the incline, and summing moments about point  $A$ , yields

$$\zeta + \sum M_A = 0;$$

$$-(98.1 \cos 1.2^\circ \text{ N})(a) + (98.1 \sin 1.2^\circ \text{ N})(100 \cos 1.2^\circ \text{ mm}) = 0$$

Solving, we obtain

$$a = 2.09 \text{ mm}$$

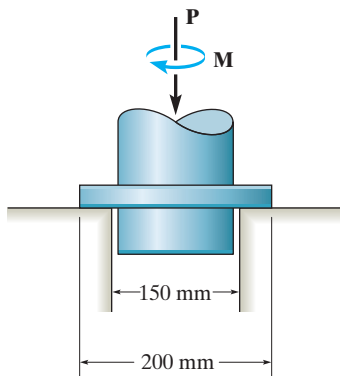
*Ans.*



## PROBLEMS

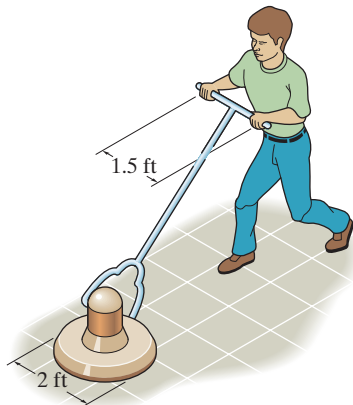
**8-107.** The collar bearing uniformly supports an axial force of  $P = 5$  kN. If the coefficient of static friction is  $\mu_s = 0.3$ , determine the smallest torque  $M$  required to overcome friction.

**\*8-108.** The collar bearing uniformly supports an axial force of  $P = 8$  kN. If a torque of  $M = 200$  N·m is applied to the shaft and causes it to rotate at constant velocity, determine the coefficient of kinetic friction at the surface of contact.



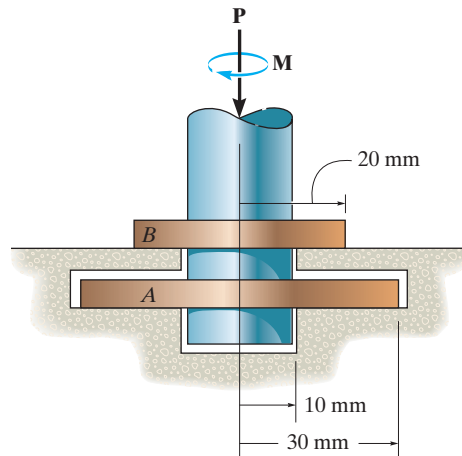
**Probs. 8-107/108**

**8-109.** The floor-polishing machine rotates at a constant angular velocity. If it has a weight of 80 lb, determine the couple forces  $F$  the operator must apply to the handles to hold the machine stationary. The coefficient of kinetic friction between the floor and brush is  $\mu_k = 0.3$ . Assume the brush exerts a uniform pressure on the floor.



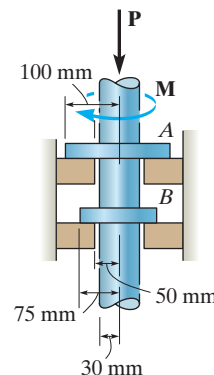
**Prob. 8-109**

**8-110.** The *double-collar bearing* is subjected to an axial force  $P = 4$  kN. Assuming that collar  $A$  supports  $0.75P$  and collar  $B$  supports  $0.25P$ , both with a uniform distribution of pressure, determine the maximum frictional moment  $M$  that may be resisted by the bearing. Take  $\mu_s = 0.2$  for both collars.



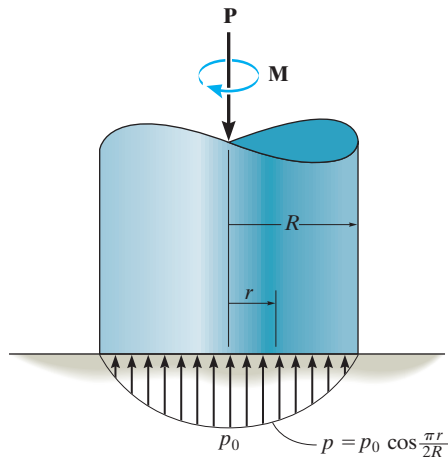
**Prob. 8-110**

**8-111.** The *double-collar bearing* is subjected to an axial force  $P = 16$  kN. Assuming that collar  $A$  supports  $0.75P$  and collar  $B$  supports  $0.25P$ , both with a uniform distribution of pressure, determine the smallest torque  $M$  that must be applied to overcome friction. Take  $\mu_s = 0.2$  for both collars.



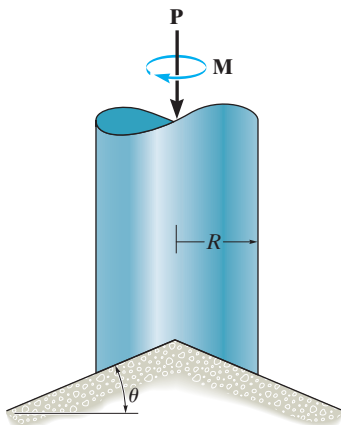
**Prob. 8-111**

**\*8–112.** The pivot bearing is subjected to a pressure distribution at its surface of contact which varies as shown. If the coefficient of static friction is  $\mu$ , determine the torque  $M$  required to overcome friction if the shaft supports an axial force  $P$ .



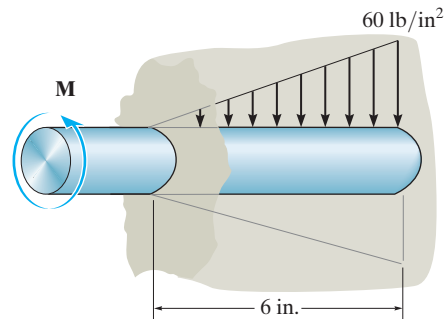
**Prob. 8–112**

**8–113.** The conical bearing is subjected to a constant pressure distribution at its surface of contact. If the coefficient of static friction is  $\mu_s$ , determine the torque  $M$  required to overcome friction if the shaft supports an axial force  $P$ .



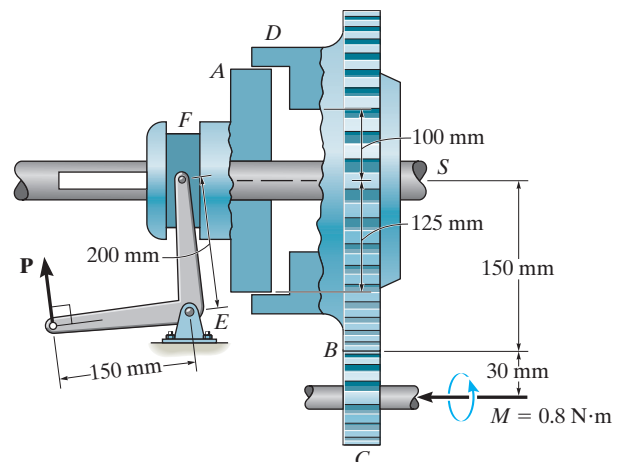
**Prob. 8–113**

**8–114.** The 4-in.-diameter shaft is held in the hole such that the normal pressure acting around the shaft varies linearly with its depth as shown. Determine the frictional torque that must be overcome to rotate the shaft. Take  $\mu_s = 0.2$ .



**Prob. 8–114**

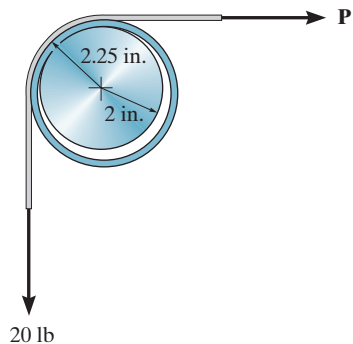
**8–115.** The plate clutch consists of a flat plate  $A$  that slides over the rotating shaft  $S$ . The shaft is fixed to the driving plate gear  $B$ . If the gear  $C$ , which is in mesh with  $B$ , is subjected to a torque of  $M = 0.8 \text{ N}\cdot\text{m}$ , determine the smallest force  $P$ , that must be applied via the control arm, to stop the rotation. The coefficient of static friction between the plates  $A$  and  $D$  is  $\mu_s = 0.4$ . Assume the bearing pressure between  $A$  and  $D$  to be uniform.



**Prob. 8–115**

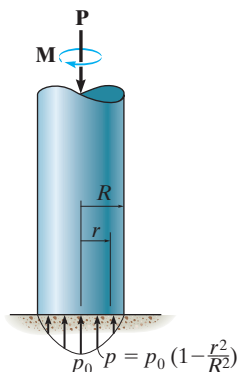
**\*8-116.** The collar fits *loosely* around a fixed shaft that has a radius of 2 in. If the coefficient of kinetic friction between the shaft and the collar is  $\mu_k = 0.3$ , determine the force  $P$  on the horizontal segment of the belt so that the collar rotates *counterclockwise* with a constant angular velocity. Assume that the belt does not slip on the collar; rather, the collar slips on the shaft. Neglect the weight and thickness of the belt and collar. The radius, measured from the center of the collar to the mean thickness of the belt, is 2.25 in.

**8-117.** The collar fits *loosely* around a fixed shaft that has a radius of 2 in. If the coefficient of kinetic friction between the shaft and the collar is  $\mu_k = 0.3$ , determine the force  $P$  on the horizontal segment of the belt so that the collar rotates *clockwise* with a constant angular velocity. Assume that the belt does not slip on the collar; rather, the collar slips on the shaft. Neglect the weight and thickness of the belt and collar. The radius, measured from the center of the collar to the mean thickness of the belt, is 2.25 in.



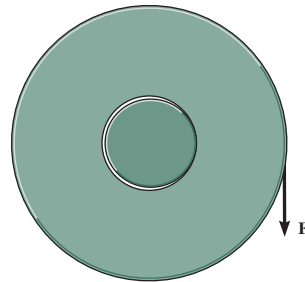
**Probs. 8-116/117**

**8-118.** The pivot bearing is subjected to a parabolic pressure distribution at its surface of contact. If the coefficient of static friction is  $\mu_k$ , determine the torque  $M$  required to overcome friction and turn the shaft if it supports an axial force  $P$ .



**Prob. 8-118**

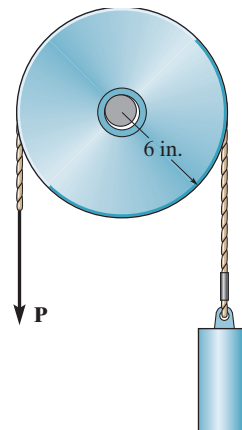
**8-119.** A disk having an outer diameter of 120 mm fits loosely over a fixed shaft having a diameter of 30 mm. If the coefficient of static friction between the disk and the shaft is  $\mu_s = 0.15$  and the disk has a mass of 50 kg, determine the smallest vertical force  $F$  acting on the rim which must be applied to the disk to cause it to slip over the shaft.



**Prob. 8-119**

**\*8-120.** The 4-lb pulley has a diameter of 1 ft and the axle has a diameter of 1 in. If the coefficient of kinetic friction between the axle and the pulley is  $\mu_k = 0.20$ , determine the vertical force  $P$  on the rope required to lift the 20-lb block at constant velocity.

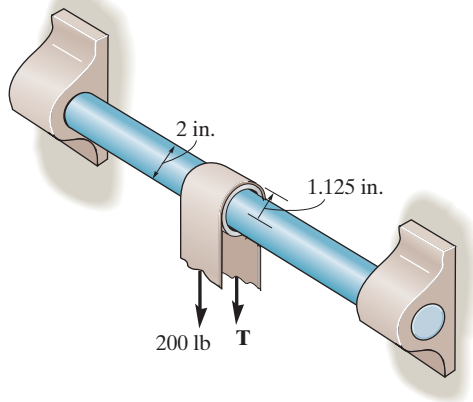
**8-121.** Solve Prob. 8-120 if the force  $P$  is applied horizontally to the left.



**Probs. 8-120/121**

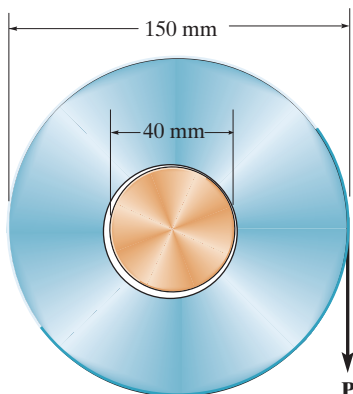
**8-122.** Determine the tension  $T$  in the belt needed to overcome the tension of 200 lb created on the other side. Also, what are the normal and frictional components of force developed on the collar bushing? The coefficient of static friction is  $\mu_s = 0.21$ .

**8-123.** If a tension force  $T = 215$  lb is required to pull the 200-lb force around the collar bushing, determine the coefficient of static friction at the contacting surface. The belt does not slip on the collar.



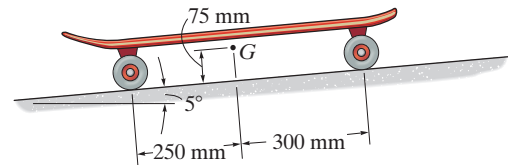
**Probs. 8-122/123**

**\*8-124.** The uniform disk fits loosely over a fixed shaft having a diameter of 40 mm. If the coefficient of static friction between the disk and the shaft is  $\mu_s = 0.15$ , determine the smallest vertical force  $P$ , acting on the rim, which must be applied to the disk to cause it to slip on the shaft. The disk has a mass of 20 kg.



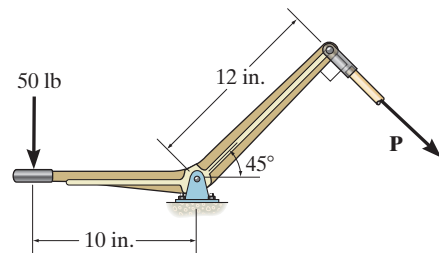
**Prob. 8-124**

**8-125.** The 5-kg skateboard rolls down the  $5^\circ$  slope at constant speed. If the coefficient of kinetic friction between the 12.5-mm-diameter axles and the wheels is  $\mu_k = 0.3$ , determine the radius of the wheels. Neglect rolling resistance of the wheels on the surface. The center of mass for the skateboard is at  $G$ .



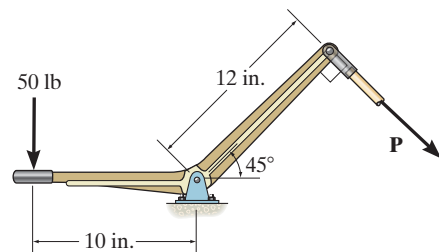
**Prob. 8-125**

**8-126.** The bell crank fits loosely into a 0.5-in-diameter pin. Determine the required force  $P$  which is just sufficient to rotate the bell crank clockwise. The coefficient of static friction between the pin and the bell crank is  $\mu_s = 0.3$ .



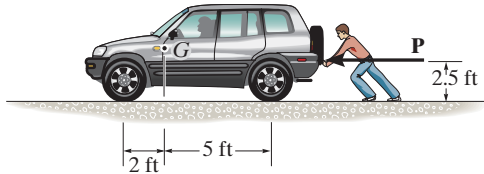
**Prob. 8-126**

**8-127.** The bell crank fits loosely into a 0.5-in-diameter pin. If  $P = 41$  lb, the bell crank is then on the verge of rotating counterclockwise. Determine the coefficient of static friction between the pin and the bell crank.



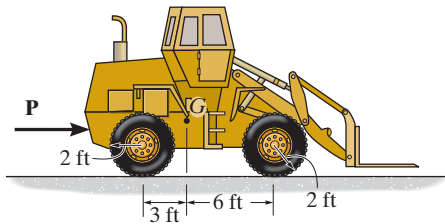
**Prob. 8-127**

**\*8–128.** The vehicle has a weight of 2600 lb and center of gravity at  $G$ . Determine the horizontal force  $\mathbf{P}$  that must be applied to overcome the rolling resistance of the wheels. The coefficient of rolling resistance is 0.5 in. The tires have a diameter of 2.75 ft.



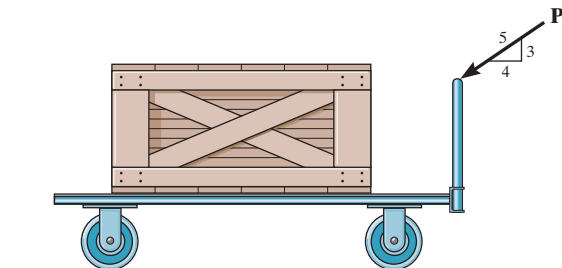
**Prob. 8–128**

**8–129.** The tractor has a weight of 16 000 lb and the coefficient of rolling resistance is  $a = 2$  in. Determine the force  $\mathbf{P}$  needed to overcome rolling resistance at all four wheels and push it forward.



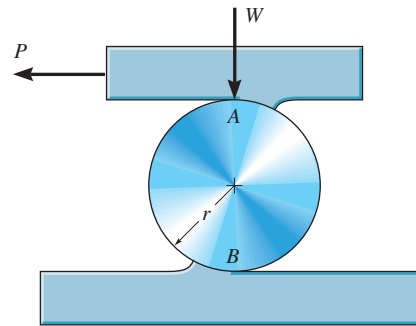
**Prob. 8–129**

**8–130.** The handcart has wheels with a diameter of 6 in. If a crate having a weight of 1500 lb is placed on the cart, determine the force  $P$  that must be applied to the handle to overcome the rolling resistance. The coefficient of rolling resistance is 0.04 in. Neglect the weight of the cart.



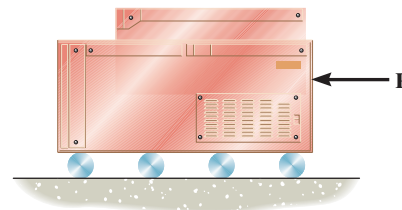
**Prob. 8–130**

**8–131.** The cylinder is subjected to a load that has a weight  $W$ . If the coefficients of rolling resistance for the cylinder's top and bottom surfaces are  $a_A$  and  $a_B$ , respectively, show that a horizontal force having a magnitude of  $P = [W(a_A + a_B)]/2r$  is required to move the load and thereby roll the cylinder forward. Neglect the weight of the cylinder.



**Prob. 8–131**

**\*8–132.** The 1.4-Mg machine is to be moved over a level surface using a series of rollers for which the coefficient of rolling resistance is 0.5 mm at the ground and 0.2 mm at the bottom surface of the machine. Determine the appropriate diameter of the rollers so that the machine can be pushed forward with a horizontal force of  $P = 250$  N. *Hint:* Use the result of Prob. 8–131.



**Prob. 8–132**

## CHAPTER REVIEW

### Dry Friction

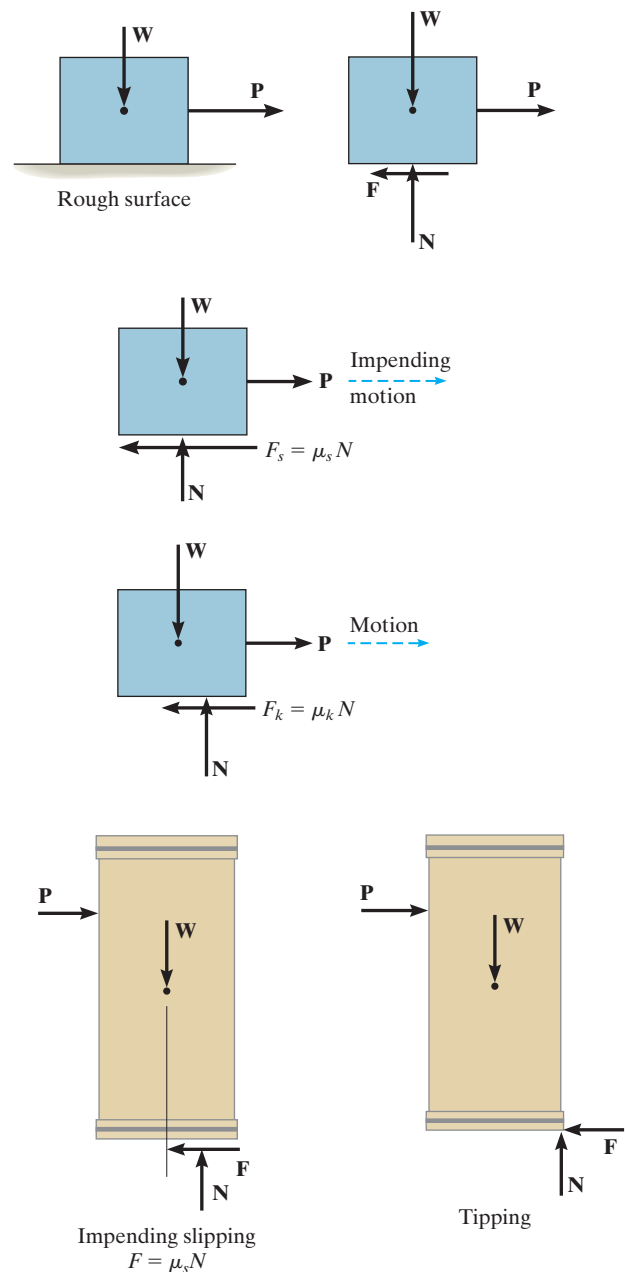
Frictional forces exist between two rough surfaces of contact. These forces act on a body so as to oppose its motion or tendency of motion.

A static frictional force approaches a maximum value of  $F_s = \mu_s N$ , where  $\mu_s$  is the *coefficient of static friction*. In this case, motion between the contacting surfaces is *impending*.

If slipping occurs, then the friction force remains essentially constant and equal to  $F_k = \mu_k N$ . Here  $\mu_k$  is the *coefficient of kinetic friction*.

The solution of a problem involving friction requires first drawing the free-body diagram of the body. If the unknowns cannot be determined strictly from the equations of equilibrium, and the possibility of slipping occurs, then the friction equation should be applied at the appropriate points of contact in order to complete the solution.

It may also be possible for slender objects, like crates, to tip over, and this situation should also be investigated.



### Wedges

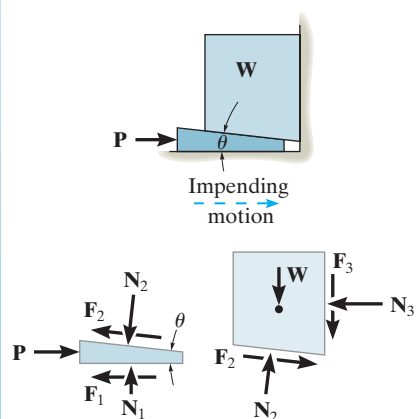
Wedges are inclined planes used to increase the application of a force. The two force equilibrium equations are used to relate the forces acting on the wedge.

An applied force **P** must push on the wedge to move it to the right.

If the coefficients of friction between the surfaces are large enough, then **P** can be removed, and the wedge will be self-locking and remain in place.

$$\Sigma F_x = 0$$

$$\Sigma F_y = 0$$



### Screws

Square-threaded screws are used to move heavy loads. They represent an inclined plane, wrapped around a cylinder.

The moment needed to turn a screw depends upon the coefficient of friction and the screw's lead angle  $\theta$ .

If the coefficient of friction between the surfaces is large enough, then the screw will support the load without tending to turn, i.e., it will be self-locking.

$$M = rW \tan(\theta + \phi_s)$$

Upward Impending Screw Motion

$$M' = rW \tan(\theta - \phi_s)$$

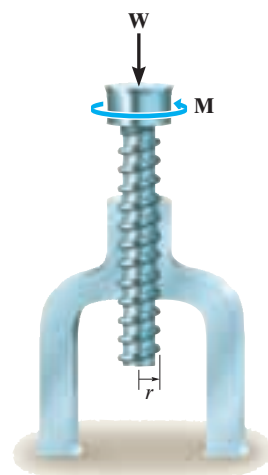
Downward Impending Screw Motion

$$\theta > \phi_s$$

$$M'' = rW \tan(\phi_s - \theta)$$

Downward Screw Motion

$$\phi_s > \theta$$

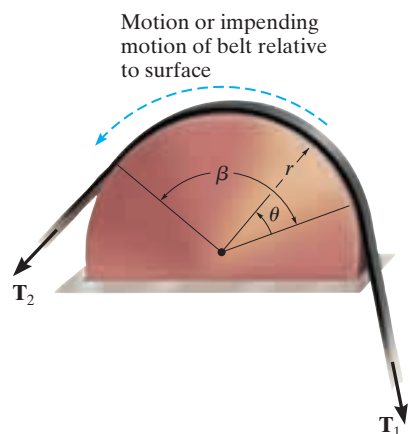


### Flat Belts

The force needed to move a flat belt over a rough curved surface depends only on the angle of belt contact,  $\beta$ , and the coefficient of friction.

$$T_2 = T_1 e^{\mu\beta}$$

$$T_2 > T_1$$

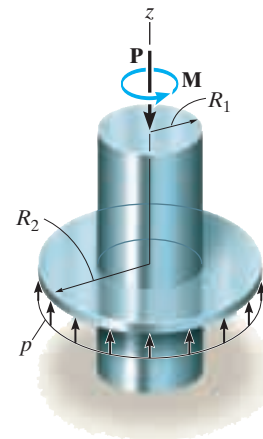


### Collar Bearings and Disks

The frictional analysis of a collar bearing or disk requires looking at a differential element of the contact area. The normal force acting on this element is determined from force equilibrium along the shaft, and the moment needed to turn the shaft at a constant rate is determined from moment equilibrium about the shaft's axis.

If the pressure on the surface of a collar bearing is uniform, then integration gives the result shown.

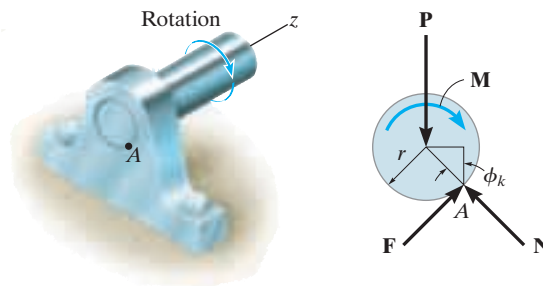
$$M = \frac{2}{3} \mu_s P \left( \frac{R_2^3 - R_1^3}{R_2^2 - R_1^2} \right)$$



### Journal Bearings

When a moment is applied to a shaft in a nonlubricated or partially lubricated journal bearing, the shaft will tend to roll up the side of the bearing until slipping occurs. This defines the radius of a friction circle, and from it the moment needed to turn the shaft can be determined.

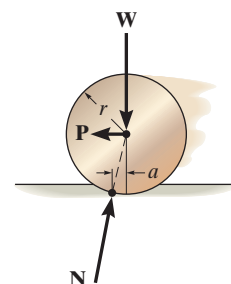
$$M = Rr \sin \phi_k$$



### Rolling Resistance

The resistance of a wheel to rolling over a surface is caused by localized *deformation* of the two materials in contact. This causes the resultant normal force acting on the rolling body to be inclined so that it provides a component that acts in the opposite direction of the applied force **P** causing the motion. This effect is characterized using the *coefficient of rolling resistance*, *a*, which is determined from experiment.

$$P \approx \frac{Wa}{r}$$

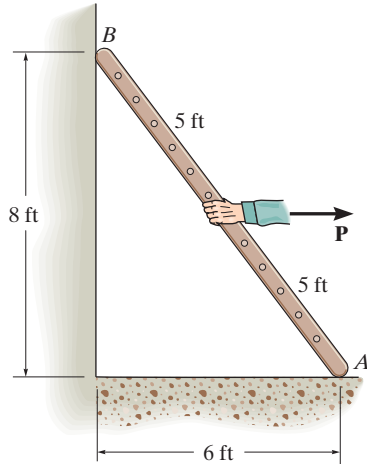




## REVIEW PROBLEMS

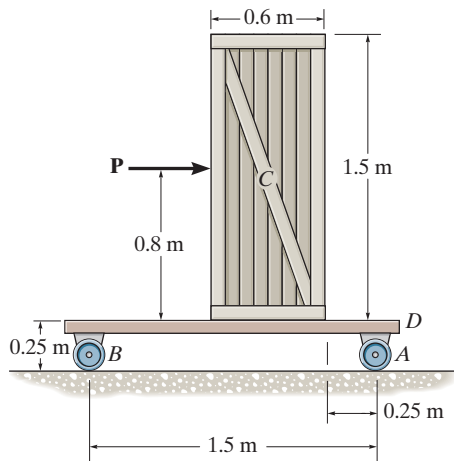
*All problem solutions must include FBDs.*

**R8-1.** The uniform 20-lb ladder rests on the rough floor for which the coefficient of static friction is  $\mu_s = 0.4$  and against the smooth wall at  $B$ . Determine the horizontal force  $P$  the man must exert on the ladder in order to cause it to move.



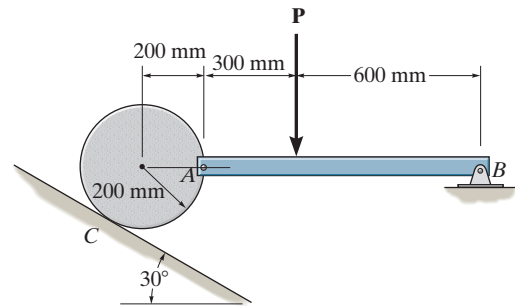
**Prob. R8-1**

**R8-2.** The uniform 60-kg crate  $C$  rests uniformly on a 10-kg dolly  $D$ . If the front casters of the dolly at  $A$  are locked to prevent rolling while the casters at  $B$  are free to roll, determine the maximum force  $P$  that may be applied without causing motion of the crate. The coefficient of static friction between the casters and the floor is  $\mu_f = 0.35$  and between the dolly and the crate,  $\mu_d = 0.5$ .



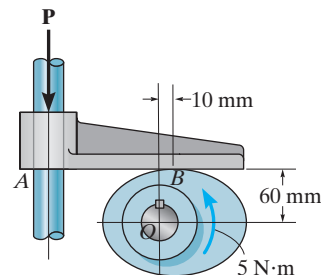
**Prob. R8-2**

**R8-3.** A 35-kg disk rests on an inclined surface for which  $\mu_s = 0.2$ . Determine the maximum vertical force  $P$  that may be applied to bar  $AB$  without causing the disk to slip at  $C$ . Neglect the mass of the bar.



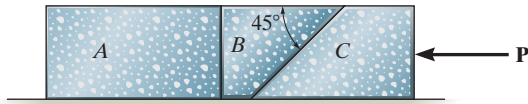
**Prob. R8-3**

**R8-4.** The cam is subjected to a couple moment of  $5 \text{ N} \cdot \text{m}$ . Determine the minimum force  $P$  that should be applied to the follower in order to hold the cam in the position shown. The coefficient of static friction between the cam and the follower is  $\mu = 0.4$ . The guide at  $A$  is smooth.



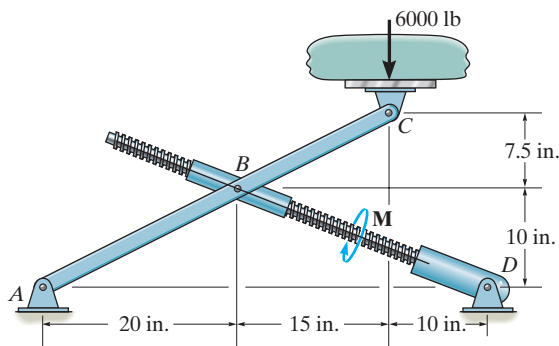
**Prob. R8-4**

**R8-5.** The three stone blocks have weights of  $W_A = 600$  lb,  $W_B = 150$  lb, and  $W_C = 500$  lb. Determine the smallest horizontal force  $P$  that must be applied to block  $C$  in order to move this block. The coefficient of static friction between the blocks is  $\mu_s = 0.3$ , and between the floor and each block  $\mu'_s = 0.5$ .



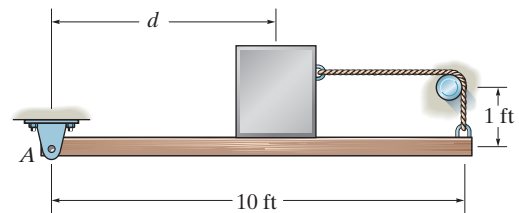
**Prob. R8-5**

**R8-6.** The jacking mechanism consists of a link that has a square-threaded screw with a mean diameter of 0.5 in. and a lead of 0.20 in., and the coefficient of static friction is  $\mu_s = 0.4$ . Determine the torque  $M$  that should be applied to the screw to start lifting the 6000-lb load acting at the end of member  $ABC$ .



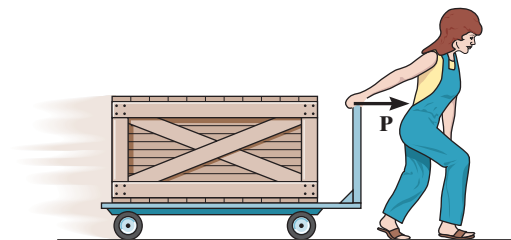
**Prob. R8-6**

**R8-7.** The uniform 50-lb beam is supported by the rope that is attached to the end of the beam, wraps over the rough peg, and is then connected to the 100-lb block. If the coefficient of static friction between the beam and the block, and between the rope and the peg, is  $\mu_s = 0.4$ , determine the maximum distance that the block can be placed from  $A$  and still remain in equilibrium. Assume the block will not tip.



**Prob. R8-7**

**R8-8.** The hand cart has wheels with a diameter of 80 mm. If a crate having a mass of 500 kg is placed on the cart so that each wheel carries an equal load, determine the horizontal force  $P$  that must be applied to the handle to overcome the rolling resistance. The coefficient of rolling resistance is 2 mm. Neglect the mass of the cart.



**Prob. R8-8**

# Chapter 9



(© Heather Reeder/Shutterstock)

When a tank of any shape is designed, it is important to be able to determine its center of gravity, calculate its volume and surface area, and determine the forces of the liquids they contain. These topics will be covered in this chapter.

# Center of Gravity and Centroid

## CHAPTER OBJECTIVES

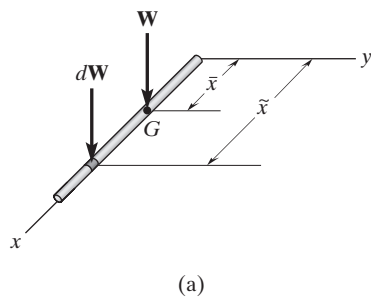
- To discuss the concept of the center of gravity, center of mass, and the centroid.
- To show how to determine the location of the center of gravity and centroid for a body of arbitrary shape and one composed of composite parts.
- To use the theorems of Pappus and Guldinus for finding the surface area and volume for a body having axial symmetry.
- To present a method for finding the resultant of a general distributed loading and to show how it applies to finding the resultant force of a pressure loading caused by a fluid.

## 9.1 Center of Gravity, Center of Mass, and the Centroid of a Body

Knowing the resultant or total weight of a body and its location is important when considering the effect this force produces on the body. The point of location is called the center of gravity, and in this section we will show how to find it for an irregularly shaped body. We will then extend this method to show how to find the body's center of mass, and its geometric center or centroid.

**Center of Gravity.** A body is composed of an infinite number of particles of differential size, and so if the body is located within a gravitational field, then each of these particles will have a weight  $dW$ . These weights will form a parallel force system, and the resultant of this system is the total weight of the body, which passes through a single point called the *center of gravity*,  $G^*$ .

\*In a strict sense this is true as long as the gravity field is assumed to have the same magnitude and direction everywhere. Although the actual force of gravity is directed toward the center of the earth, and this force varies with its distance from the center, for most engineering applications we can assume the gravity field has the same magnitude and direction everywhere.



To show how to determine the location of the center of gravity, consider the rod in Fig. 9-1a, where the segment having the weight  $dW$  is located at the arbitrary position  $\tilde{x}$ . Using the methods outlined in Sec. 4.8, the total weight of the rod is the sum of the weights of all of its particles, that is

$$+\downarrow F_R = \Sigma F_z; \quad W = \int dW$$

The location of the center of gravity, measured from the  $y$  axis, is determined by equating the moment of  $W$  about the  $y$  axis, Fig. 9-1b, to the sum of the moments of the weights of all its particles about this same axis. Therefore,

$$(M_R)_y = \Sigma M_y; \quad \bar{x}W = \int \tilde{x} dW$$

$$\bar{x} = \frac{\int \tilde{x} dW}{\int dW}$$

In a similar manner, if the body represents a plate, Fig. 9-1b, then a moment balance about the  $x$  and  $y$  axes would be required to determine the location  $(\bar{x}, \bar{y})$  of point  $G$ . Finally we can generalize this idea to a three-dimensional body, Fig. 9-1c, and perform a moment balance about all three axes to locate  $G$  for any rotated position of the axes. This results in the following equations.

$$\bar{x} = \frac{\int \tilde{x} dW}{\int dW} \quad \bar{y} = \frac{\int \tilde{y} dW}{\int dW} \quad \bar{z} = \frac{\int \tilde{z} dW}{\int dW} \quad (9-1)$$

where

$\bar{x}, \bar{y}, \bar{z}$  are the coordinates of the center of gravity  $G$ .

$\tilde{x}, \tilde{y}, \tilde{z}$  are the coordinates of an arbitrary particle in the body.

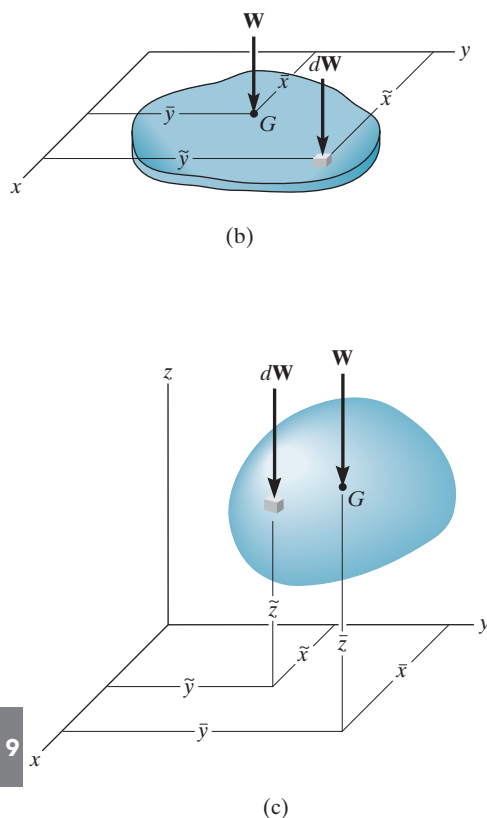


Fig. 9-1

**Center of Mass of a Body.** In order to study the *dynamic response* or accelerated motion of a body, it becomes important to locate the body's **center of mass**  $C_m$ , Fig. 9-2. This location can be determined by substituting  $dW = g \, dm$  into Eqs. 9-1. Provided  $g$  is constant, it cancels out, and so

$$\bar{x} = \frac{\int \tilde{x} \, dm}{\int dm} \quad \bar{y} = \frac{\int \tilde{y} \, dm}{\int dm} \quad \bar{z} = \frac{\int \tilde{z} \, dm}{\int dm} \quad (9-2)$$

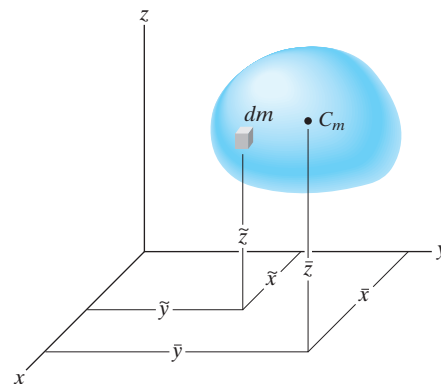


Fig. 9-2

**Centroid of a Volume.** If the body in Fig. 9-3 is made from a *homogeneous material*, then its density  $\rho$  (rho) will be *constant*. Therefore, a differential element of volume  $dV$  has a mass  $dm = \rho \, dV$ . Substituting this into Eqs. 9-2 and canceling out  $\rho$ , we obtain formulas that locate the **centroid**  $C$  or geometric center of the body; namely

$$\bar{x} = \frac{\int_V \tilde{x} \, dV}{\int_V dV} \quad \bar{y} = \frac{\int_V \tilde{y} \, dV}{\int_V dV} \quad \bar{z} = \frac{\int_V \tilde{z} \, dV}{\int_V dV} \quad (9-3)$$

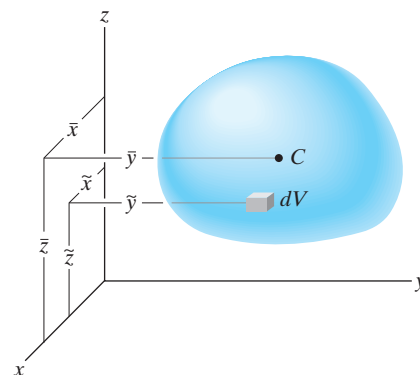


Fig. 9-3

These equations represent a balance of the moments of the volume of the body. Therefore, if the volume possesses two planes of symmetry, then its centroid must lie along the line of intersection of these two planes. For example, the cone in Fig. 9-4 has a centroid that lies on the  $y$  axis so that  $\bar{x} = \bar{z} = 0$ . The location  $\bar{y}$  can be found using a single integration by choosing a differential element represented by a *thin disk* having a thickness  $dy$  and radius  $r = z$ . Its volume is  $dV = \pi r^2 \, dy = \pi z^2 \, dy$  and its centroid is at  $\tilde{x} = 0, \tilde{y} = y, \tilde{z} = 0$ .

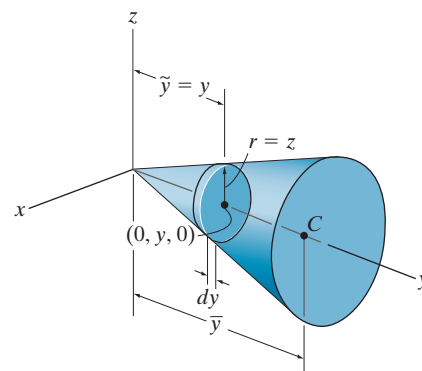


Fig. 9-4

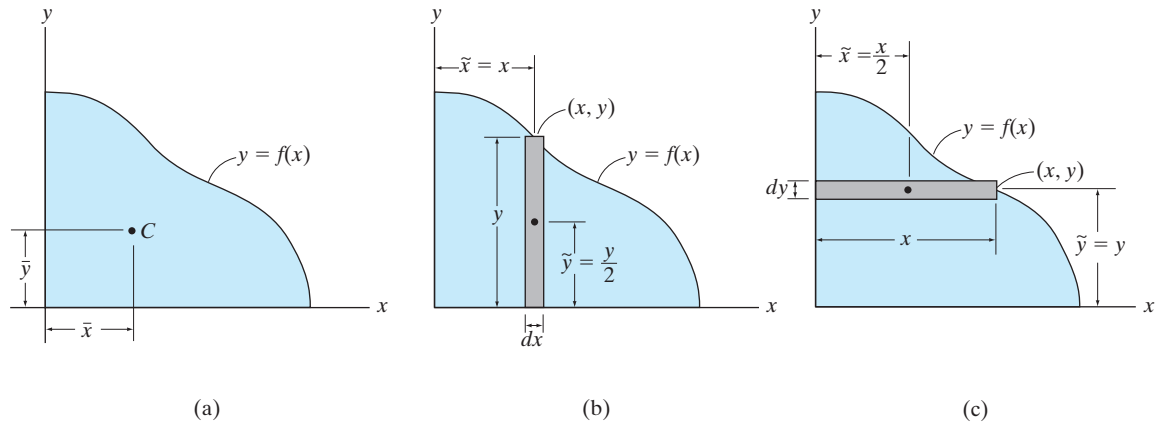


Fig. 9-5



Integration must be used to determine the location of the center of gravity of this lamp post due to the curvature of the member. (© Russell C. Hibbeler)

**Centroid of an Area.** If an area lies in the  $x$ - $y$  plane and is bounded by the curve  $y = f(x)$ , as shown in Fig. 9-5a, then its centroid will be in this plane and can be determined from integrals similar to Eqs. 9-3, namely,

$$\bar{x} = \frac{\int_A \tilde{x} dA}{\int_A dA} \quad \bar{y} = \frac{\int_A \tilde{y} dA}{\int_A dA} \quad (9-4)$$

These integrals can be evaluated by performing a *single integration* if we use a *rectangular strip* for the differential area element. For example, if a vertical strip is used, Fig. 9-5b, the area of the element is  $dA = y dx$ , and its centroid is located at  $\tilde{x} = x$  and  $\tilde{y} = y/2$ . If we consider a horizontal strip, Fig. 9-5c, then  $dA = x dy$ , and its centroid is located at  $\tilde{x} = x/2$  and  $\tilde{y} = y$ .

**Centroid of a Line.** If a line segment (or rod) lies within the  $x$ - $y$  plane and it can be described by a thin curve  $y = f(x)$ , Fig. 9-6a, then its centroid is determined from

$$\bar{x} = \frac{\int_L \tilde{x} dL}{\int_L dL} \quad \bar{y} = \frac{\int_L \tilde{y} dL}{\int_L dL} \quad (9-5)$$

Here, the length of the differential element is given by the Pythagorean theorem,  $dL = \sqrt{(dx)^2 + (dy)^2}$ , which can also be written in the form

$$\begin{aligned} dL &= \sqrt{\left(\frac{dx}{dx}\right)^2 dx^2 + \left(\frac{dy}{dx}\right)^2 dx^2} \\ &= \left(\sqrt{1 + \left(\frac{dy}{dx}\right)^2}\right) dx \end{aligned}$$

or

$$\begin{aligned} dL &= \sqrt{\left(\frac{dx}{dy}\right)^2 dy^2 + \left(\frac{dy}{dy}\right)^2 dy^2} \\ &= \left(\sqrt{\left(\frac{dx}{dy}\right)^2 + 1}\right) dy \end{aligned}$$

Either one of these expressions can be used; however, for application, the one that will result in a simpler integration should be selected. For example, consider the rod in Fig. 9–6*b*, defined by  $y = 2x^2$ . The length of the element is  $dL = \sqrt{1 + (dy/dx)^2} dx$ , and since  $dy/dx = 4x$ , then  $dL = \sqrt{1 + (4x)^2} dx$ . The centroid for this element is located at  $\tilde{x} = x$  and  $\tilde{y} = y$ .

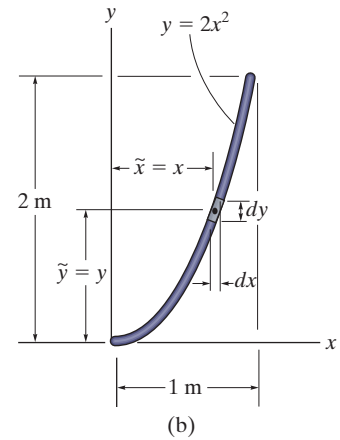
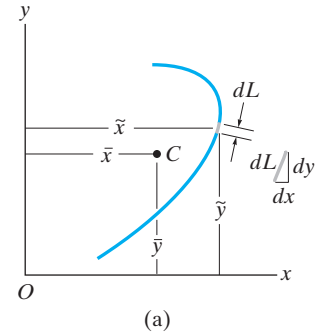


Fig. 9–6

### Important Points

- The centroid represents the geometric center of a body. This point coincides with the center of mass or the center of gravity only if the material composing the body is uniform or homogeneous.
- Formulas used to locate the center of gravity or the centroid simply represent a balance between the sum of moments of all the parts of the system and the moment of the “resultant” for the system.
- In some cases the centroid is located at a point that is not on the object, as in the case of a ring, where the centroid is at its center. Also, this point will lie on any axis of symmetry for the body, Fig. 9–7.

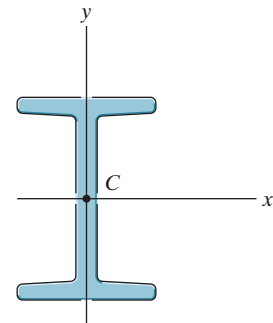


Fig. 9–7



## Procedure for Analysis

The center of gravity or centroid of an object or shape can be determined by single integrations using the following procedure.

### Differential Element.

- Select an appropriate coordinate system, specify the coordinate axes, and then choose a differential element for integration.
- For lines the element is represented by a differential line segment of length  $dL$ .
- For areas the element is generally a rectangle of area  $dA$ , having a finite length and differential width.
- For volumes the element can be a circular disk of volume  $dV$ , having a finite radius and differential thickness.
- Locate the element so that it touches the arbitrary point  $(x, y, z)$  on the curve that defines the boundary of the shape.

### Size and Moment Arms.

- Express the length  $dL$ , area  $dA$ , or volume  $dV$  of the element in terms of the coordinates describing the curve.
- Express the moment arms  $\tilde{x}$ ,  $\tilde{y}$ ,  $\tilde{z}$  for the centroid or center of gravity of the element in terms of the coordinates describing the curve.

### Integrations.

- Substitute the formulations for  $\tilde{x}$ ,  $\tilde{y}$ ,  $\tilde{z}$  and  $dL$ ,  $dA$ , or  $dV$  into the appropriate equations (Eqs. 9–1 through 9–5).
- Express the function in the integrand in terms of the *same variable as the differential thickness of the element*.
- The limits of the integral are defined from the two extreme locations of the element's differential thickness, so that when the elements are “summed” or the integration performed, the entire region is covered.

**EXAMPLE 9.1**

Locate the centroid of the rod bent into the shape of a parabolic arc as shown in Fig. 9–8.

**SOLUTION**

**Differential Element.** The differential element is shown in Fig. 9–8. It is located on the curve at the *arbitrary point*  $(x, y)$ .

**Area and Moment Arms.** The differential element of length  $dL$  can be expressed in terms of the differentials  $dx$  and  $dy$  using the Pythagorean theorem.

$$dL = \sqrt{(dx)^2 + (dy)^2} = \sqrt{\left(\frac{dx}{dy}\right)^2 + 1} dy$$

Since  $x = y^2$ , then  $dx/dy = 2y$ . Therefore, expressing  $dL$  in terms of  $y$  and  $dy$ , we have

$$dL = \sqrt{(2y)^2 + 1} dy$$

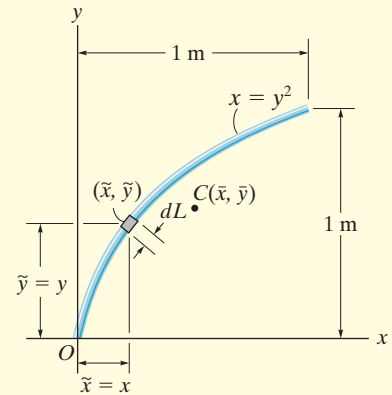
As shown in Fig. 9–8, the centroid of the element is located at  $\tilde{x} = x$ ,  $\tilde{y} = y$ .

**Integrations.** Applying Eq. 9–5 and using the integration formula to evaluate the integrals, we get

$$\begin{aligned} \bar{x} &= \frac{\int_L \tilde{x} dL}{\int_L dL} = \frac{\int_0^{1\text{ m}} x \sqrt{4y^2 + 1} dy}{\int_0^{1\text{ m}} \sqrt{4y^2 + 1} dy} = \frac{\int_0^{1\text{ m}} y^2 \sqrt{4y^2 + 1} dy}{\int_0^{1\text{ m}} \sqrt{4y^2 + 1} dy} \\ &= \frac{0.6063}{1.479} = 0.410 \text{ m} \quad \text{Ans.} \end{aligned}$$

$$\begin{aligned} \bar{y} &= \frac{\int_L \tilde{y} dL}{\int_L dL} = \frac{\int_0^{1\text{ m}} y \sqrt{4y^2 + 1} dy}{\int_0^{1\text{ m}} \sqrt{4y^2 + 1} dy} = \frac{0.8484}{1.479} = 0.574 \text{ m} \quad \text{Ans.} \end{aligned}$$

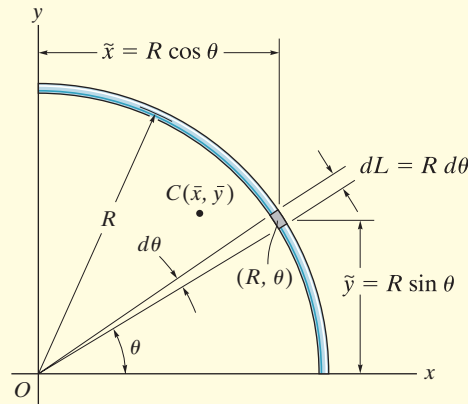
**NOTE:** These results for  $C$  seem reasonable when they are plotted on Fig. 9–8.



**Fig. 9–8**

**EXAMPLE 9.2**

Locate the centroid of the circular wire segment shown in Fig. 9–9.



**Fig. 9–9**

**SOLUTION**

Polar coordinates will be used to solve this problem since the arc is circular.

**Differential Element.** A differential circular arc is selected as shown in the figure. This element lies on the curve at  $(R, \theta)$ .

**Length and Moment Arm.** The length of the differential element is  $dL = R d\theta$ , and its centroid is located at  $\tilde{x} = R \cos \theta$  and  $\tilde{y} = R \sin \theta$ .

**Integrations.** Applying Eqs. 9–5 and integrating with respect to  $\theta$ , we obtain

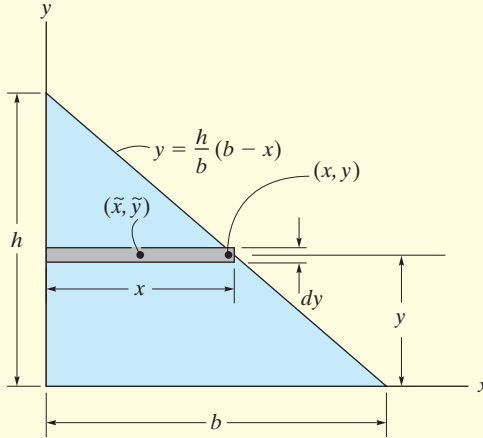
$$\bar{x} = \frac{\int_L \tilde{x} dL}{\int_L dL} = \frac{\int_0^{\pi/2} (R \cos \theta) R d\theta}{\int_0^{\pi/2} R d\theta} = \frac{R^2 \int_0^{\pi/2} \cos \theta d\theta}{R \int_0^{\pi/2} d\theta} = \frac{2R}{\pi} \quad \text{Ans.}$$

$$\bar{y} = \frac{\int_L \tilde{y} dL}{\int_L dL} = \frac{\int_0^{\pi/2} (R \sin \theta) R d\theta}{\int_0^{\pi/2} R d\theta} = \frac{R^2 \int_0^{\pi/2} \sin \theta d\theta}{R \int_0^{\pi/2} d\theta} = \frac{2R}{\pi} \quad \text{Ans.}$$

**NOTE:** As expected, the two coordinates are numerically the same due to the symmetry of the wire.

**EXAMPLE 9.3**

Determine the distance  $\bar{y}$  measured from the  $x$  axis to the centroid of the area of the triangle shown in Fig. 9–10.



**Fig. 9–10**

**SOLUTION**

**Differential Element.** Consider a rectangular element having a thickness  $dy$ , and located in an arbitrary position so that it intersects the boundary at  $(x, y)$ , Fig. 9–10.

**Area and Moment Arms.** The area of the element is  $dA = x \, dy = \frac{b}{h}(h - y) \, dy$ , and its centroid is located a distance  $\tilde{y} = y$  from the  $x$  axis.

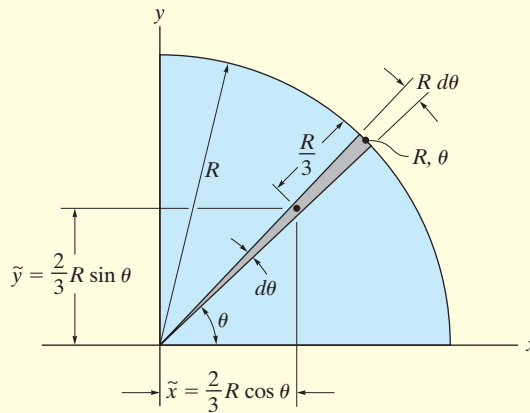
**Integration.** Applying the second of Eqs. 9–4 and integrating with respect to  $y$  yields

$$\begin{aligned} \bar{y} &= \frac{\int_A \tilde{y} \, dA}{\int_A dA} = \frac{\int_0^h y \left[ \frac{b}{h}(h - y) \, dy \right]}{\int_0^h \frac{b}{h}(h - y) \, dy} = \frac{\frac{1}{6}bh^2}{\frac{1}{2}bh} \\ &= \frac{h}{3} \qquad \text{Ans.} \end{aligned}$$

**NOTE:** This result is valid for any shape of triangle. It states that the centroid is located at one-third the height, measured from the base of the triangle.

**EXAMPLE 9.4**

Locate the centroid for the area of a quarter circle shown in Fig. 9–11.

**Fig. 9–11****SOLUTION**

**Differential Element.** Polar coordinates will be used, since the boundary is circular. We choose the element in the shape of a *triangle*, Fig. 9–11. (Actually the shape is a circular sector; however, neglecting higher-order differentials, the element becomes triangular.) The element intersects the curve at point  $(R, \theta)$ .

**Area and Moment Arms.** The area of the element is

$$dA = \frac{1}{2}(R)(R d\theta) = \frac{R^2}{2} d\theta$$

and using the results of Example 9.3, the centroid of the (triangular) element is located at  $\tilde{x} = \frac{2}{3}R \cos \theta$ ,  $\tilde{y} = \frac{2}{3}R \sin \theta$ .

**Integrations.** Applying Eqs. 9–4 and integrating with respect to  $\theta$ , we obtain

$$\bar{x} = \frac{\int_A \tilde{x} dA}{\int_A dA} = \frac{\int_0^{\pi/2} \left( \frac{2}{3}R \cos \theta \right) \frac{R^2}{2} d\theta}{\int_0^{\pi/2} \frac{R^2}{2} d\theta} = \frac{\left( \frac{2}{3}R \right) \int_0^{\pi/2} \cos \theta d\theta}{\int_0^{\pi/2} d\theta} = \frac{4R}{3\pi} \quad \text{Ans.}$$

$$\bar{y} = \frac{\int_A \tilde{y} dA}{\int_A dA} = \frac{\int_0^{\pi/2} \left( \frac{2}{3}R \sin \theta \right) \frac{R^2}{2} d\theta}{\int_0^{\pi/2} \frac{R^2}{2} d\theta} = \frac{\left( \frac{2}{3}R \right) \int_0^{\pi/2} \sin \theta d\theta}{\int_0^{\pi/2} d\theta} = \frac{4R}{3\pi} \quad \text{Ans.}$$

**EXAMPLE 9.5**

Locate the centroid of the area shown in Fig. 9–12a.

**SOLUTION I**

**Differential Element.** A differential element of thickness  $dx$  is shown in Fig. 9–12a. The element intersects the curve at the *arbitrary point*  $(x, y)$ , and so it has a height  $y$ .

**Area and Moment Arms.** The area of the element is  $dA = y dx$ , and its centroid is located at  $\tilde{x} = x, \tilde{y} = y/2$ .

**Integrations.** Applying Eqs. 9–4 and integrating with respect to  $x$  yields

$$\bar{x} = \frac{\int_A \tilde{x} dA}{\int_A dA} = \frac{\int_0^{1\text{ m}} xy dx}{\int_0^{1\text{ m}} y dx} = \frac{\int_0^{1\text{ m}} x^3 dx}{\int_0^{1\text{ m}} x^2 dx} = \frac{0.250}{0.333} = 0.75\text{ m}$$

$$\bar{y} = \frac{\int_A \tilde{y} dA}{\int_A dA} = \frac{\int_0^{1\text{ m}} (y/2)y dx}{\int_0^{1\text{ m}} y dx} = \frac{\int_0^{1\text{ m}} (x^2/2)x^2 dx}{\int_0^{1\text{ m}} x^2 dx} = \frac{0.100}{0.333} = 0.3\text{ m} \quad \text{Ans.}$$

**SOLUTION II**

**Differential Element.** The differential element of thickness  $dy$  is shown in Fig. 9–12b. The element intersects the curve at the *arbitrary point*  $(x, y)$ , and so it has a length  $(1 - x)$ .

**Area and Moment Arms.** The area of the element is  $dA = (1 - x) dy$ , and its centroid is located at

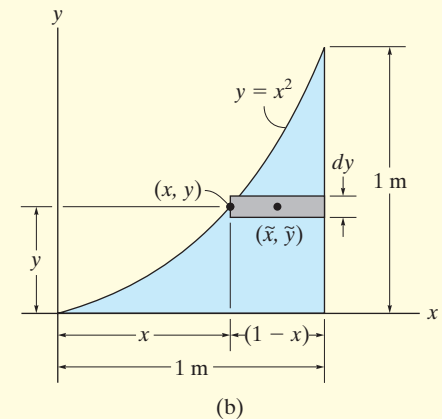
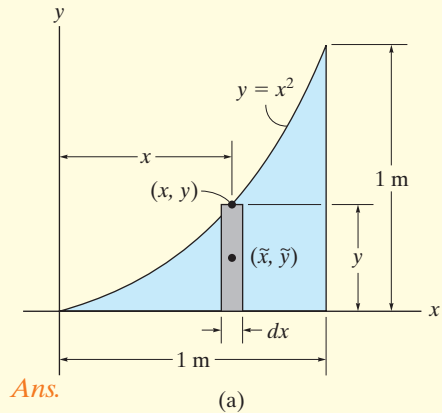
$$\tilde{x} = x + \left(\frac{1 - x}{2}\right) = \frac{1 + x}{2}, \quad \tilde{y} = y$$

**Integrations.** Applying Eqs. 9–4 and integrating with respect to  $y$ , we obtain

$$\bar{x} = \frac{\int_A \tilde{x} dA}{\int_A dA} = \frac{\int_0^{1\text{ m}} [(1 + x)/2](1 - x) dy}{\int_0^{1\text{ m}} (1 - x) dy} = \frac{\frac{1}{2} \int_0^{1\text{ m}} (1 - y) dy}{\int_0^{1\text{ m}} (1 - \sqrt{y}) dy} = \frac{0.250}{0.333} = 0.75\text{ m} \quad \text{Ans.}$$

$$\bar{y} = \frac{\int_A \tilde{y} dA}{\int_A dA} = \frac{\int_0^{1\text{ m}} y(1 - x) dy}{\int_0^{1\text{ m}} (1 - x) dy} = \frac{\int_0^{1\text{ m}} (y - y^{3/2}) dy}{\int_0^{1\text{ m}} (1 - \sqrt{y}) dy} = \frac{0.100}{0.333} = 0.3\text{ m} \quad \text{Ans.}$$

**NOTE:** Plot these results and notice that they seem reasonable. Also, for this problem, elements of thickness  $dx$  offer a simpler solution.



**Fig. 9–12**

## EXAMPLE 9.6

Locate the centroid of the semi-elliptical area shown in Fig. 9-13a.

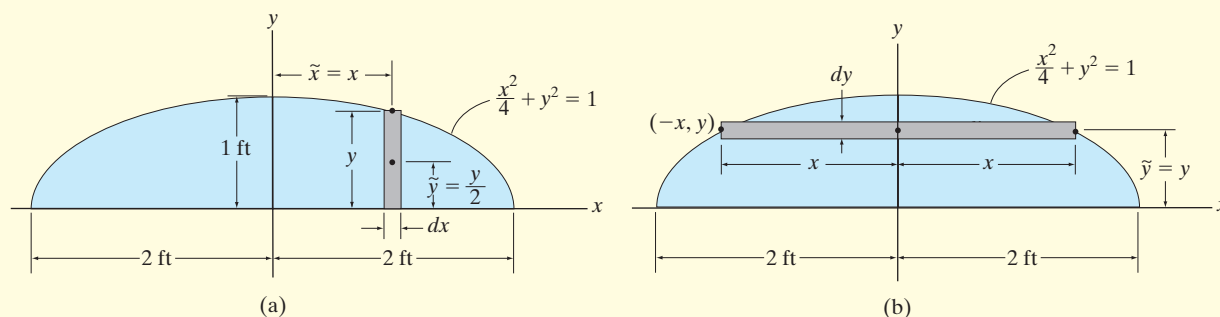


Fig. 9-13

## SOLUTION I

**Differential Element.** The rectangular differential element parallel to the  $y$  axis shown shaded in Fig. 9-13a will be considered. This element has a thickness of  $dx$  and a height of  $y$ .

**Area and Moment Arms.** Thus, the area is  $dA = y dx$ , and its centroid is located at  $\tilde{x} = x$  and  $\tilde{y} = y/2$ .

**Integration.** Since the area is symmetrical about the  $y$  axis,

$$\bar{x} = 0 \quad \text{Ans.}$$

Applying the second of Eqs. 9-4 with  $y = \sqrt{1 - \frac{x^2}{4}}$ , we have

$$\bar{y} = \frac{\int_A \tilde{y} dA}{\int_A dA} = \frac{\int_{-2\text{ ft}}^{2\text{ ft}} \frac{y}{2} (y dx)}{\int_{-2\text{ ft}}^{2\text{ ft}} y dx} = \frac{\frac{1}{2} \int_{-2\text{ ft}}^{2\text{ ft}} \left(1 - \frac{x^2}{4}\right) dx}{\int_{-2\text{ ft}}^{2\text{ ft}} \sqrt{1 - \frac{x^2}{4}} dx} = \frac{4/3}{\pi} = 0.424 \text{ ft} \quad \text{Ans.}$$

## SOLUTION II

**Differential Element.** The shaded rectangular differential element of thickness  $dy$  and width  $2x$ , parallel to the  $x$  axis, will be considered, Fig. 9-13b.

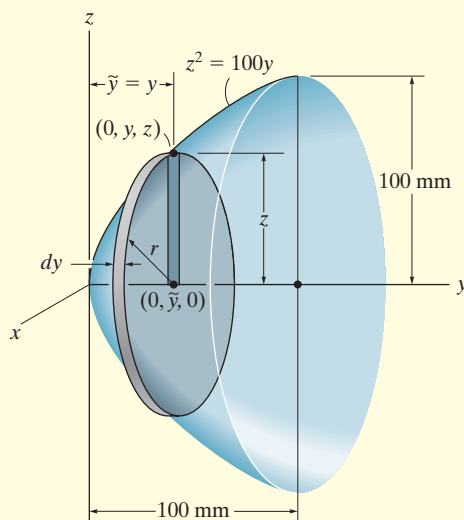
**Area and Moment Arms.** The area is  $dA = 2x dy$ , and its centroid is at  $\tilde{x} = 0$  and  $\tilde{y} = y$ .

**Integration.** Applying the second of Eqs. 9-4, with  $x = 2\sqrt{1 - y^2}$ , we have

$$\bar{y} = \frac{\int_A \tilde{y} dA}{\int_A dA} = \frac{\int_0^{1\text{ ft}} y(2x dy)}{\int_0^{1\text{ ft}} 2x dy} = \frac{\int_0^{1\text{ ft}} 4y\sqrt{1 - y^2} dy}{\int_0^{1\text{ ft}} 4\sqrt{1 - y^2} dy} = \frac{4/3}{\pi} \text{ ft} = 0.424 \text{ ft} \quad \text{Ans.}$$

**EXAMPLE 9.7**

Locate the  $\bar{y}$  centroid for the paraboloid of revolution, shown in Fig. 9–14.



**Fig. 9–14**

**SOLUTION**

**Differential Element.** An element having the shape of a *thin disk* is chosen. This element has a thickness  $dy$ , it intersects the generating curve at the *arbitrary point*  $(0, y, z)$ , and so its radius is  $r = z$ .

**Volume and Moment Arm.** The volume of the element is  $dV = (\pi z^2) dy$ , and its centroid is located at  $\tilde{y} = y$ .

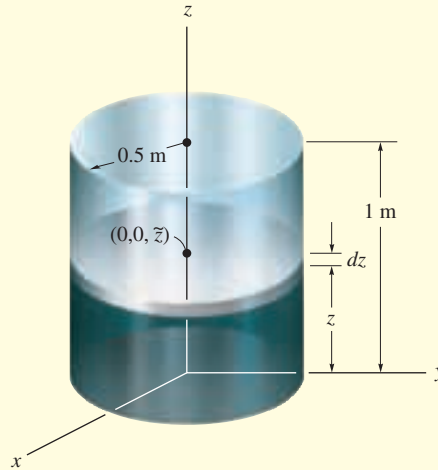
**Integration.** Applying the second of Eqs. 9–3 and integrating with respect to  $y$  yields.

$$\bar{y} = \frac{\int_V \tilde{y} dV}{\int_V dV} = \frac{\int_0^{100 \text{ mm}} y(\pi z^2) dy}{\int_0^{100 \text{ mm}} (\pi z^2) dy} = \frac{100\pi \int_0^{100 \text{ mm}} y^2 dy}{100\pi \int_0^{100 \text{ mm}} y dy} = 66.7 \text{ mm} \quad \text{Ans.}$$



**EXAMPLE 9.8**

Determine the location of the center of mass of the cylinder shown in Fig. 9–15 if its density varies directly with the distance from its base, i.e.,  $\rho = 200z \text{ kg/m}^3$ .

**Fig. 9–15****SOLUTION**

For reasons of material symmetry,

$$\bar{x} = \bar{y} = 0$$

*Ans.*

**Differential Element.** A disk element of radius 0.5 m and thickness  $dz$  is chosen for integration, Fig. 9–15, since the *density of the entire element is constant* for a given value of  $z$ . The element is located along the  $z$  axis at the *arbitrary point*  $(0, 0, z)$ .

**Volume and Moment Arm.** The volume of the element is  $dV = \pi(0.5)^2 dz$ , and its centroid is located at  $\tilde{z} = z$ .

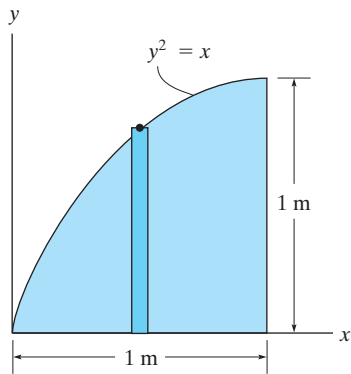
**Integrations.** Using the third of Eqs. 9–2 with  $dm = \rho dV$  and integrating with respect to  $z$ , noting that  $\rho = 200z$ , we have

$$\begin{aligned} \bar{z} &= \frac{\int_V \tilde{z} \rho dV}{\int_V \rho dV} = \frac{\int_0^{1 \text{ m}} z(200z) [\pi(0.5)^2 dz]}{\int_0^{1 \text{ m}} (200z) \pi(0.5)^2 dz} \\ &= \frac{\int_0^{1 \text{ m}} z^2 dz}{\int_0^{1 \text{ m}} z dz} = 0.667 \text{ m} \end{aligned}$$

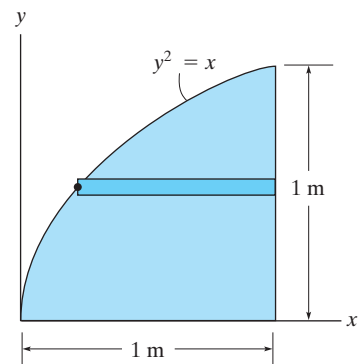
*Ans.*

**PRELIMINARY PROBLEM**

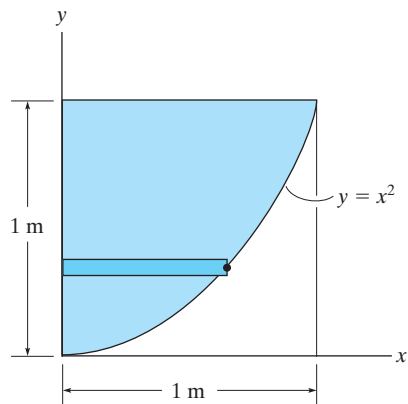
**P9-1.** In each case, use the element shown and specify  $\tilde{x}$ ,  $\tilde{y}$ , and  $dA$ .



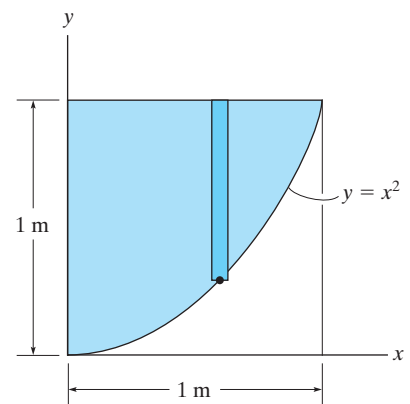
(a)



(b)



(c)

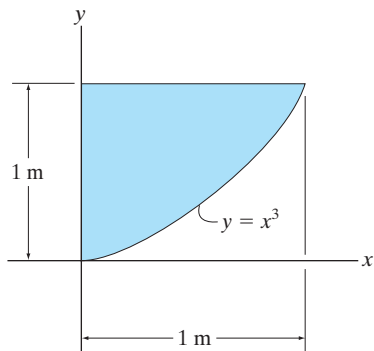


(d)

**Prob. P9-1**

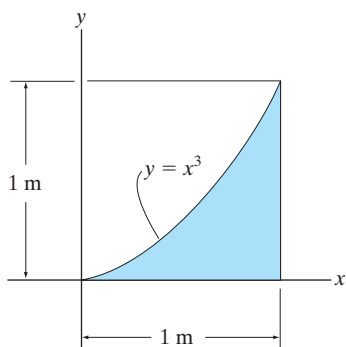
## FUNDAMENTAL PROBLEMS

**F9-1.** Determine the centroid  $(\bar{x}, \bar{y})$  of the shaded area.



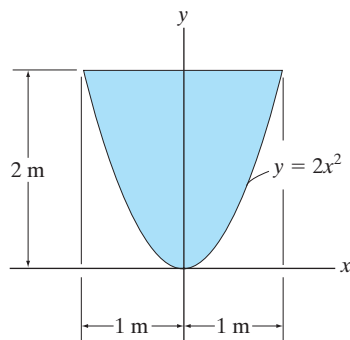
**Prob. F9-1**

**F9-2.** Determine the centroid  $(\bar{x}, \bar{y})$  of the shaded area.



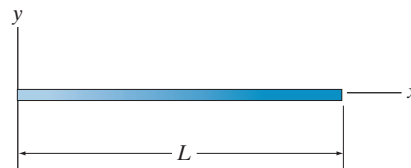
**Prob. F9-2**

**F9-3.** Determine the centroid  $\bar{y}$  of the shaded area.



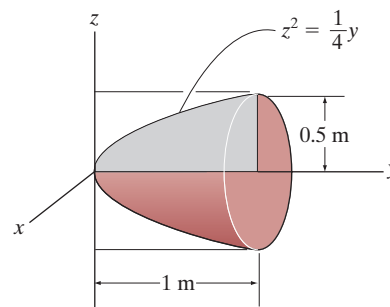
**Prob. F9-3**

**F9-4.** Locate the center of mass  $\bar{x}$  of the straight rod if its mass per unit length is given by  $m = m_0(1 + x^2/L^2)$ .



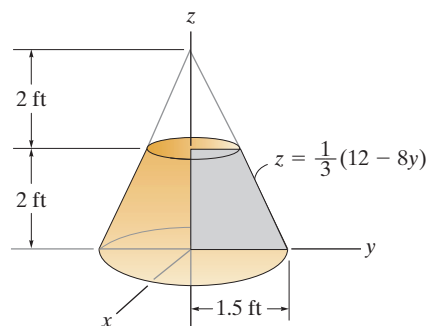
**Prob. F9-4**

**F9-5.** Locate the centroid  $\bar{y}$  of the homogeneous solid formed by revolving the shaded area about the y axis.



**Prob. F9-5**

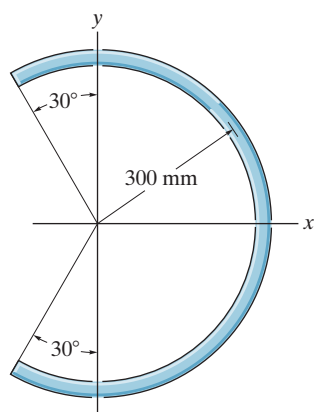
**F9-6.** Locate the centroid  $\bar{z}$  of the homogeneous solid formed by revolving the shaded area about the z axis.



**Prob. F9-6**

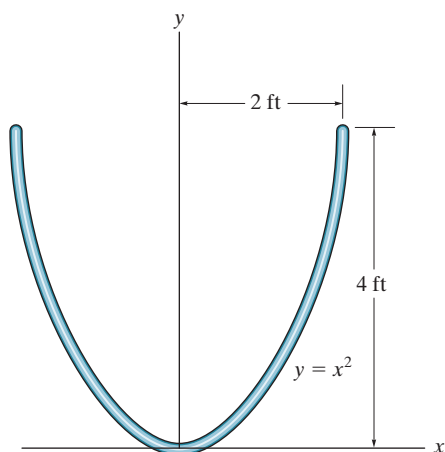
## PROBLEMS

**9-1.** Locate the center of mass of the homogeneous rod bent into the shape of a circular arc.



**Prob. 9-1**

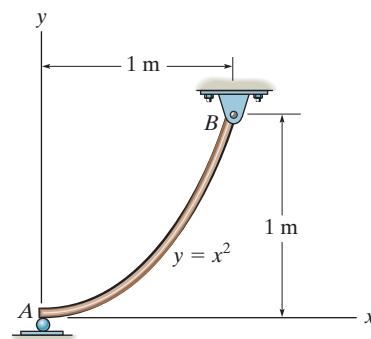
**9-2.** Determine the location  $(\bar{x}, \bar{y})$  of the centroid of the wire.



**Prob. 9-2**

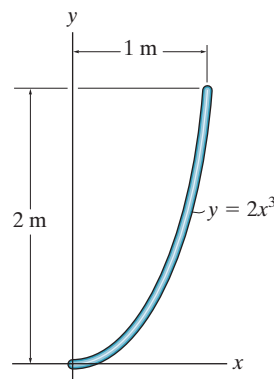
**9-3.** Locate the center of gravity  $\bar{x}$  of the homogeneous rod. If the rod has a weight per unit length of 100 N/m, determine the vertical reaction at  $A$  and the  $x$  and  $y$  components of reaction at the pin  $B$ .

**\*9-4.** Locate the center of gravity  $\bar{y}$  of the homogeneous rod.



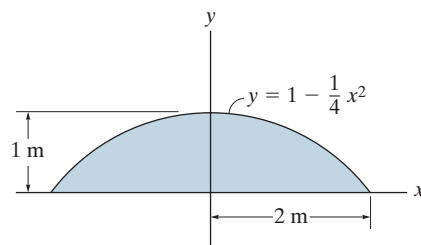
**Probs. 9-3/4**

**9-5.** Determine the distance  $\bar{y}$  to the center of gravity of the homogeneous rod.



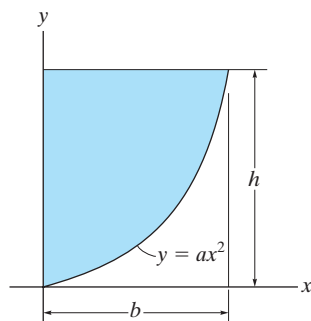
**Prob. 9-5**

**9-6.** Locate the centroid  $\bar{y}$  of the area.



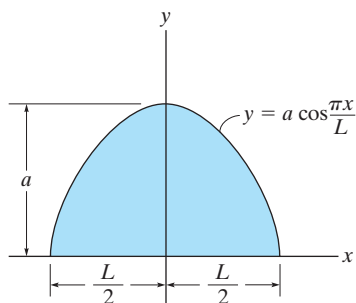
**Prob. 9-6**

**9-7.** Locate the centroid  $\bar{x}$  of the parabolic area.



**Prob. 9-7**

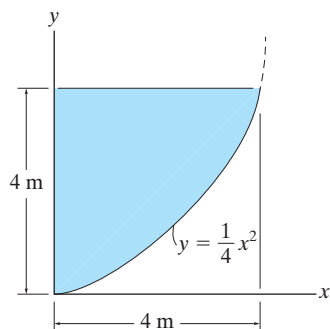
**\*9-8.** Locate the centroid of the shaded area.



**Prob. 9-8**

**9-9.** Locate the centroid  $\bar{x}$  of the shaded area.

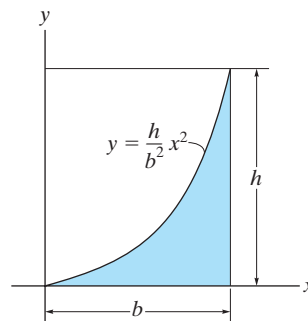
**9-10.** Locate the centroid  $\bar{y}$  of the shaded area.



**Probs. 9-9/10**

**9-11.** Locate the centroid  $\bar{x}$  of the area.

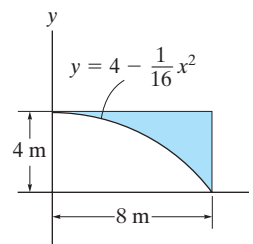
**\*9-12.** Locate the centroid  $\bar{y}$  of the area.



**Probs. 9-11/12**

**9-13.** Locate the centroid  $\bar{x}$  of the area.

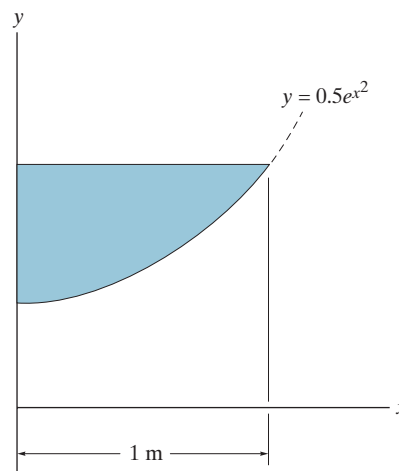
**9-14.** Locate the centroid  $\bar{y}$  of the area.



**Probs. 9-13/14**

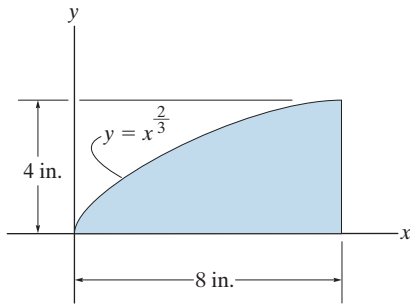
**9-15.** Locate the centroid  $\bar{x}$  of the shaded area. Solve the problem by evaluating the integrals using Simpson's rule.

**\*9-16.** Locate the centroid  $\bar{y}$  of the shaded area. Solve the problem by evaluating the integrals using Simpson's rule.



**Probs. 9-15/16**

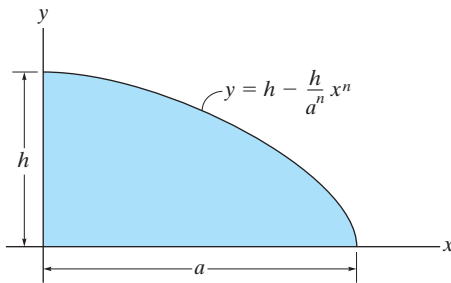
**9–17.** Locate the centroid  $\bar{y}$  of the area.



**Prob. 9–17**

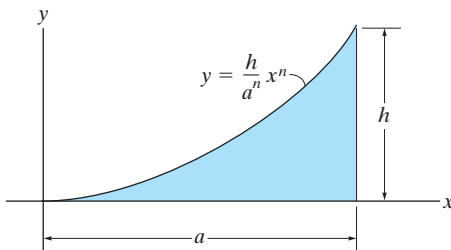
**9–18.** Locate the centroid  $\bar{x}$  of the area.

**9–19.** Locate the centroid  $\bar{y}$  of the area.



**Probs. 9–18/19**

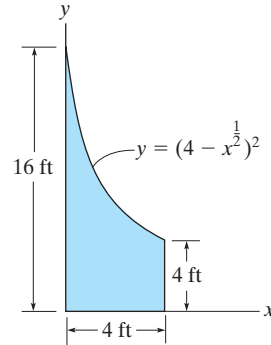
**\*9–20.** Locate the centroid  $\bar{y}$  of the shaded area.



**Prob. 9–20**

**9–21.** Locate the centroid  $\bar{x}$  of the shaded area.

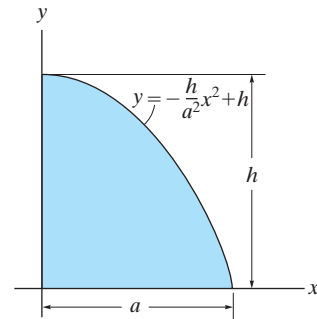
**9–22.** Locate the centroid  $\bar{y}$  of the shaded area.



**Probs. 9–21/22**

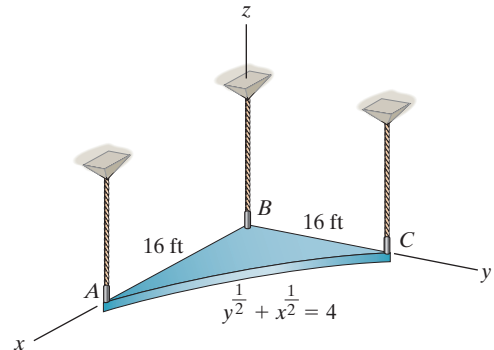
**9–23.** Locate the centroid  $\bar{x}$  of the shaded area.

**\*9–24.** Locate the centroid  $\bar{y}$  of the shaded area.



**Probs. 9–23/24**

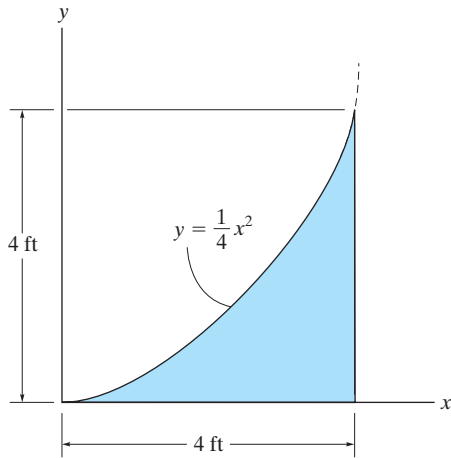
**9–25.** The plate has a thickness of 0.25 ft and a specific weight of  $\gamma = 180 \text{ lb/ft}^3$ . Determine the location of its center of gravity. Also, find the tension in each of the cords used to support it.



**Prob. 9–25**

**9-26.** Locate the centroid  $\bar{x}$  of the shaded area.

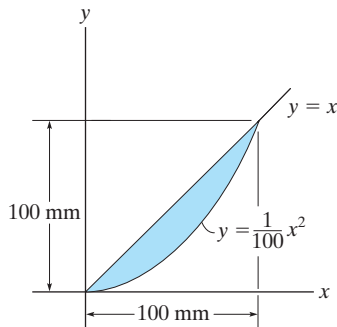
**9-27.** Locate the centroid  $\bar{y}$  of the shaded area.



**Probs. 9-26/27**

**\*9-28.** Locate the centroid  $\bar{x}$  of the shaded area.

**9-29.** Locate the centroid  $\bar{y}$  of the shaded area.

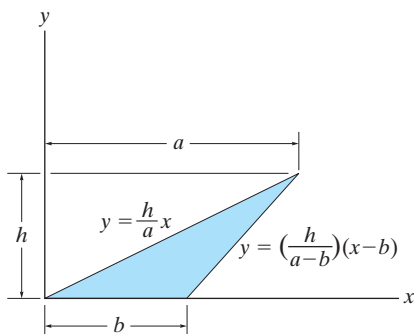


**Probs. 9-28/29**

**9-30.** Locate the centroid  $\bar{x}$  of the shaded area.

9

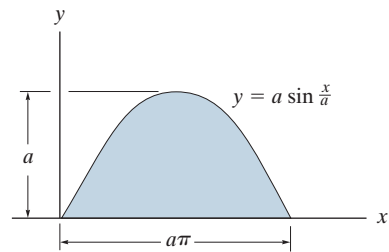
**9-31.** Locate the centroid  $\bar{y}$  of the shaded area.



**Probs. 9-30/31**

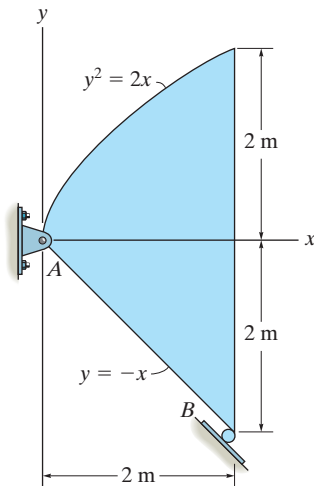
**\*9-32.** Locate the centroid  $\bar{x}$  of the area.

**9-33.** Locate the centroid  $\bar{y}$  of the area.



**Probs. 9-32/33**

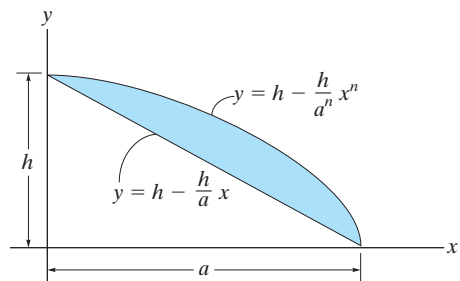
**9-34.** The steel plate is 0.3 m thick and has a density of  $7850 \text{ kg/m}^3$ . Determine the location of its center of mass. Also find the reactions at the pin and roller support.



**Prob. 9-34**

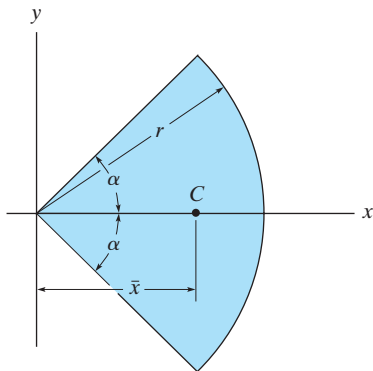
**9-35.** Locate the centroid  $\bar{x}$  of the shaded area.

**\*9-36.** Locate the centroid  $\bar{y}$  of the shaded area.



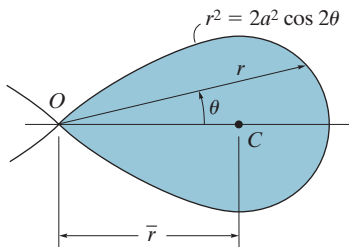
**Probs. 9-35/36**

9-37. Locate the centroid  $\bar{x}$  of the circular sector.



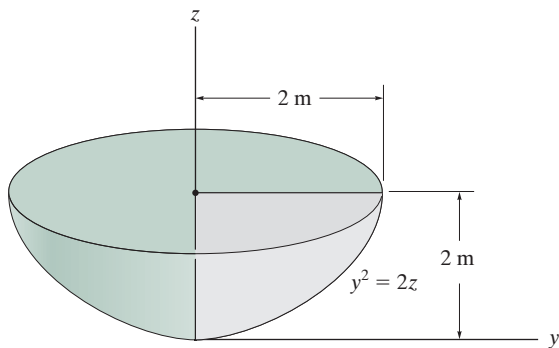
Prob. 9-37

9-38. Determine the location  $\bar{r}$  of the centroid  $C$  for the loop of the lemniscate,  $r^2 = 2a^2 \cos 2\theta$ ,  $(-45^\circ \leq \theta \leq 45^\circ)$ .



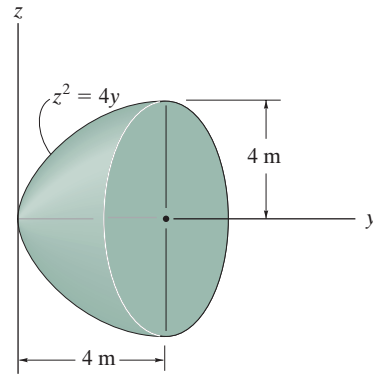
Prob. 9-38

9-39. Locate the center of gravity of the volume. The material is homogeneous.



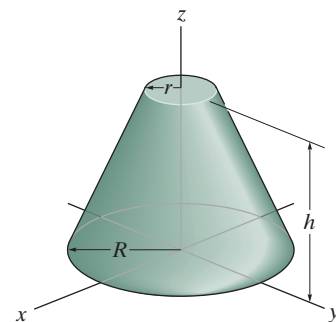
Prob. 9-39

\*9-40. Locate the centroid  $\bar{y}$  of the paraboloid.



Prob. 9-40

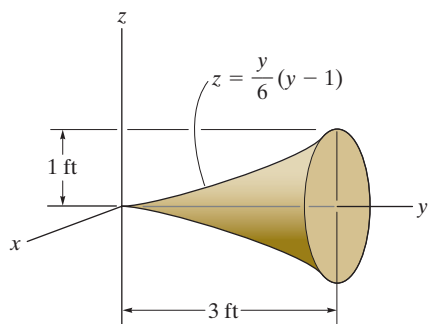
9-41. Locate the centroid  $\bar{z}$  of the frustum of the right-circular cone.



Prob. 9-41

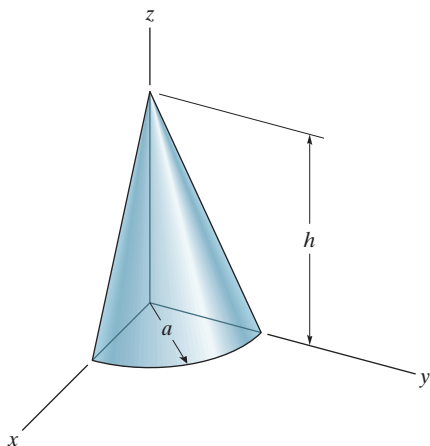


**9-42.** Determine the centroid  $\bar{y}$  of the solid.



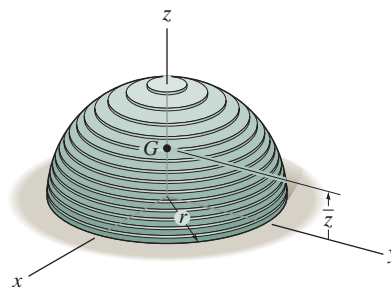
**Prob. 9-42**

**9-43.** Locate the centroid of the quarter-cone.



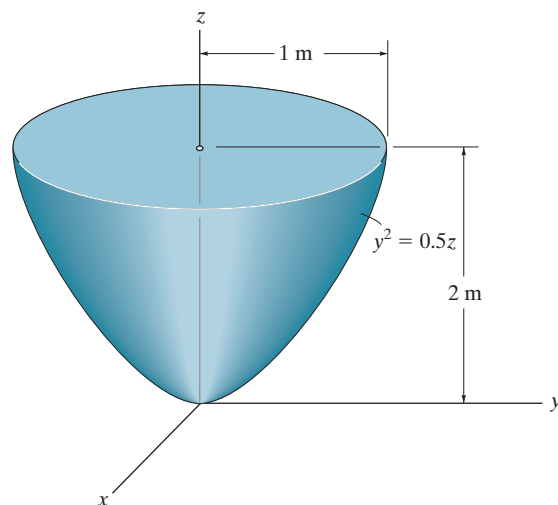
**Prob. 9-43**

**\*9-44.** The hemisphere of radius  $r$  is made from a stack of very thin plates such that the density varies with height,  $\rho = kz$ , where  $k$  is a constant. Determine its mass and the distance  $\bar{z}$  to the center of mass  $G$ .



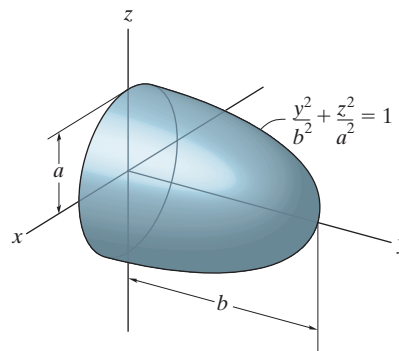
**Prob. 9-44**

**9-45.** Locate the centroid  $\bar{z}$  of the volume.



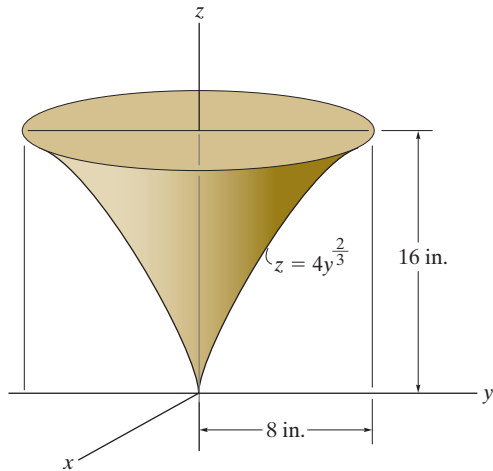
**Prob. 9-45**

**9-46.** Locate the centroid of the ellipsoid of revolution.



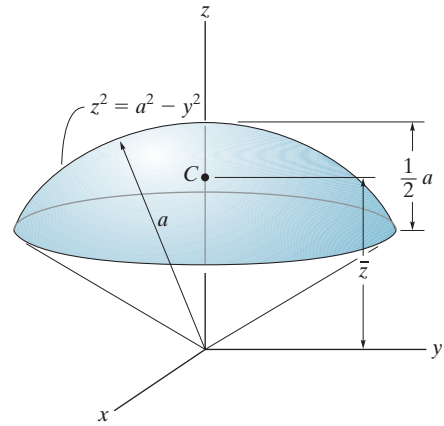
**Prob. 9-46**

**9-47.** Locate the center of gravity  $\bar{z}$  of the solid.



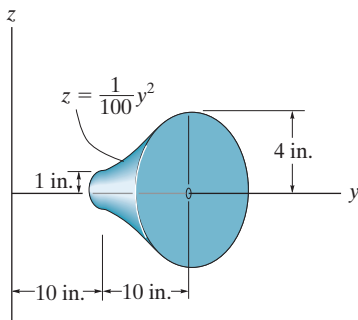
**Prob. 9-47**

**9-49.** Locate the centroid  $\bar{z}$  of the spherical segment.



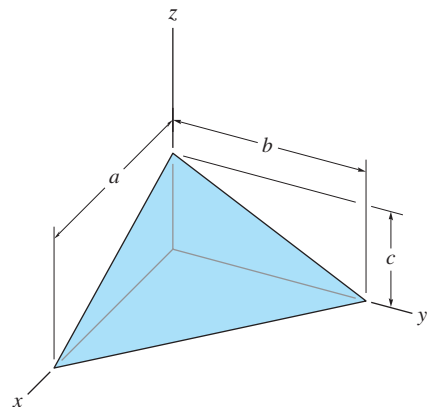
**Prob. 9-49**

**\*9-48.** Locate the center of gravity  $\bar{y}$  of the volume. The material is homogeneous.



**Prob. 9-48**

**9-50.** Determine the location  $\bar{z}$  of the centroid for the tetrahedron. *Suggestion:* Use a triangular “plate” element parallel to the  $x$ - $y$  plane and of thickness  $dz$ .



**Prob. 9-50**

## 9.2 Composite Bodies

A **composite body** consists of a series of connected “simpler” shaped bodies, which may be rectangular, triangular, semicircular, etc. Such a body can often be sectioned or divided into its composite parts and, provided the *weight* and location of the center of gravity of each of these parts are known, we can then eliminate the need for integration to determine the center of gravity for the entire body. The method for doing this follows the same procedure outlined in Sec. 9.1. Formulas analogous to Eqs. 9–1 result; however, rather than account for an infinite number of differential weights, we have instead a finite number of weights. Therefore,

$$\bar{x} = \frac{\sum \tilde{x}W}{\sum W} \quad \bar{y} = \frac{\sum \tilde{y}W}{\sum W} \quad \bar{z} = \frac{\sum \tilde{z}W}{\sum W} \quad (9-6)$$

Here

$\bar{x}, \bar{y}, \bar{z}$  represent the coordinates of the center of gravity  $G$  of the composite body.

$\tilde{x}, \tilde{y}, \tilde{z}$  represent the coordinates of the center of gravity of each composite part of the body.

$\sum W$  is the sum of the weights of all the composite parts of the body, or simply the total weight of the body.

When the body has a *constant density or specific weight*, the center of gravity *coincides* with the centroid of the body. The centroid for composite lines, areas, and volumes can be found using relations analogous to Eqs. 9–6; however, the  $W$ 's are replaced by  $L$ 's,  $A$ 's, and  $V$ 's, respectively. Centroids for common shapes of lines, areas, shells, and volumes that often make up a composite body are given in the table on the inside back cover.



A stress analysis of this angle requires that the centroid of its cross-sectional area be located. (© Russell C. Hibbeler)



In order to determine the force required to tip over this concrete barrier it is first necessary to determine the location of its center of gravity  $G$ . Due to symmetry,  $G$  will lie on the vertical axis of symmetry. (© Russell C. Hibbeler)

## Procedure for Analysis

The location of the center of gravity of a body or the centroid of a composite geometrical object represented by a line, area, or volume can be determined using the following procedure.

### Composite Parts.

- Using a sketch, divide the body or object into a finite number of composite parts that have simpler shapes.
- If a composite body has a *hole*, or a geometric region having no material, then consider the composite body without the hole and consider the hole as an *additional* composite part having *negative* weight or size.

### Moment Arms.

- Establish the coordinate axes on the sketch and determine the coordinates  $\tilde{x}$ ,  $\tilde{y}$ ,  $\tilde{z}$  of the center of gravity or centroid of each part.

### Summations.

- Determine  $\bar{x}$ ,  $\bar{y}$ ,  $\bar{z}$  by applying the center of gravity equations, Eqs. 9–6, or the analogous centroid equations.
- If an object is *symmetrical* about an axis, the centroid of the object lies on this axis.

If desired, the calculations can be arranged in tabular form, as indicated in the following three examples.



The center of gravity of this water tank can be determined by dividing it into composite parts and applying Eqs. 9–6. (© Russell C. Hibbeler)

**EXAMPLE 9.9**

Locate the centroid of the wire shown in Fig. 9–16a.

**SOLUTION**

**Composite Parts.** The wire is divided into three segments as shown in Fig. 9–16b.

**Moment Arms.** The location of the centroid for each segment is determined and indicated in the figure. In particular, the centroid of segment ① is determined either by integration or by using the table on the inside back cover.

**Summations.** For convenience, the calculations can be tabulated as follows:

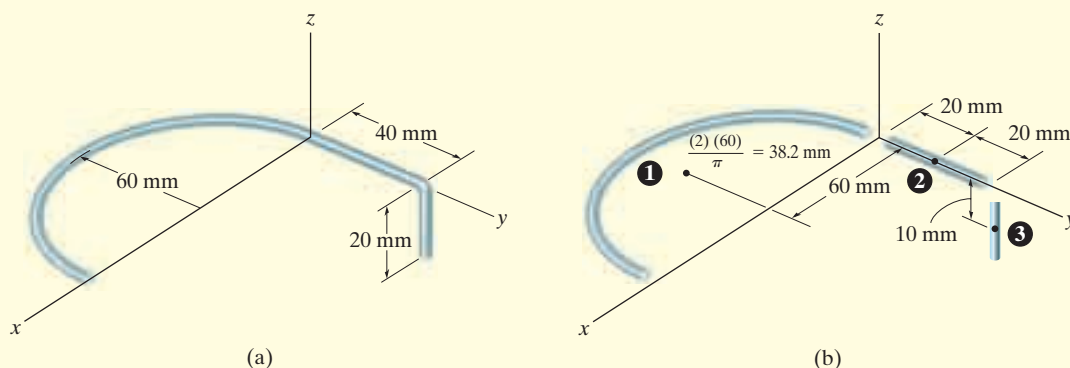
Segment	$L$ (mm)	$\tilde{x}$ (mm)	$\tilde{y}$ (mm)	$\tilde{z}$ (mm)	$\tilde{x}L$ (mm <sup>2</sup> )	$\tilde{y}L$ (mm <sup>2</sup> )	$\tilde{z}L$ (mm <sup>2</sup> )
1	$\pi(60) = 188.5$	60	-38.2	0	11 310	-7200	0
2	40	0	20	0	0	800	0
3	20	0	40	-10	0	800	-200
	$\Sigma L = 248.5$				$\Sigma \tilde{x}L = 11\,310$	$\Sigma \tilde{y}L = -5600$	$\Sigma \tilde{z}L = -200$

Thus,

$$\bar{x} = \frac{\Sigma \tilde{x}L}{\Sigma L} = \frac{11\,310}{248.5} = 45.5 \text{ mm} \quad \text{Ans.}$$

$$\bar{y} = \frac{\Sigma \tilde{y}L}{\Sigma L} = \frac{-5600}{248.5} = -22.5 \text{ mm} \quad \text{Ans.}$$

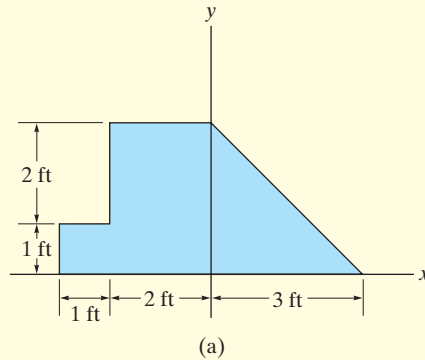
$$\bar{z} = \frac{\Sigma \tilde{z}L}{\Sigma L} = \frac{-200}{248.5} = -0.805 \text{ mm} \quad \text{Ans.}$$



**Fig. 9–16**

**EXAMPLE 9.10**

Locate the centroid of the plate area shown in Fig. 9–17a.



**Fig. 9–17**

**SOLUTION**

**Composite Parts.** The plate is divided into three segments as shown in Fig. 9–17b. Here the area of the small rectangle (3) is considered “negative” since it must be subtracted from the larger one (2).

**Moment Arms.** The centroid of each segment is located as indicated in the figure. Note that the  $\tilde{x}$  coordinates of (2) and (3) are *negative*.

**Summations.** Taking the data from Fig. 9–17b, the calculations are tabulated as follows:

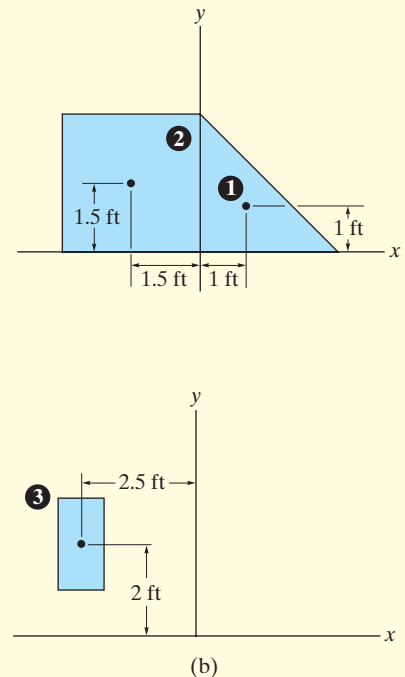
Segment	$A$ (ft <sup>2</sup> )	$\tilde{x}$ (ft)	$\tilde{y}$ (ft)	$\tilde{x}A$ (ft <sup>3</sup> )	$\tilde{y}A$ (ft <sup>3</sup> )
1	$\frac{1}{2}(3)(3) = 4.5$	1	1	4.5	4.5
2	$(3)(3) = 9$	-1.5	1.5	-13.5	13.5
3	$-(2)(1) = -2$	-2.5	2	5	-4
	$\Sigma A = 11.5$			$\Sigma \tilde{x}A = -4$	$\Sigma \tilde{y}A = 14$

Thus,

$$\bar{x} = \frac{\Sigma \tilde{x}A}{\Sigma A} = \frac{-4}{11.5} = -0.348 \text{ ft} \quad \text{Ans.}$$

$$\bar{y} = \frac{\Sigma \tilde{y}A}{\Sigma A} = \frac{14}{11.5} = 1.22 \text{ ft} \quad \text{Ans.}$$

**NOTE:** If these results are plotted in Fig. 9–17a, the location of point C seems reasonable.



## EXAMPLE 9.11

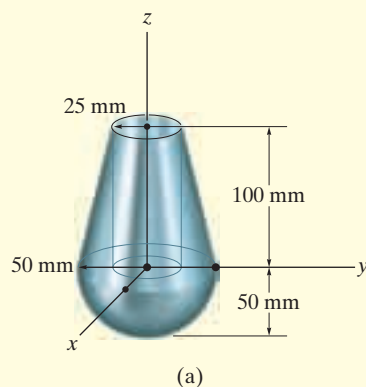


Fig. 9-18

Locate the center of mass of the assembly shown in Fig. 9-18a. The conical frustum has a density of  $\rho_c = 8 \text{ Mg/m}^3$ , and the hemisphere has a density of  $\rho_h = 4 \text{ Mg/m}^3$ . There is a 25-mm-radius cylindrical hole in the center of the frustum.

## SOLUTION

**Composite Parts.** The assembly can be thought of as consisting of four segments as shown in Fig. 9-18b. For the calculations, (3) and (4) must be considered as “negative” segments in order that the four segments, when added together, yield the total composite shape shown in Fig. 9-18a.

**Moment Arm.** Using the table on the inside back cover, the computations for the centroid  $\tilde{z}$  of each piece are shown in the figure.

**Summations.** Because of *symmetry*, note that

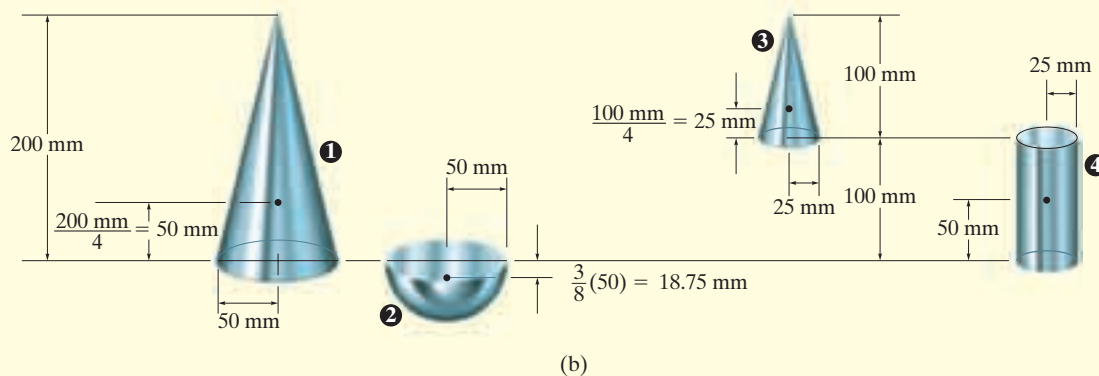
$$\bar{x} = \bar{y} = 0$$

Ans.

Since  $W = mg$ , and  $g$  is constant, the third of Eqs. 9-6 becomes  $\bar{z} = \Sigma \tilde{z}m / \Sigma m$ . The mass of each piece can be computed from  $m = \rho V$  and used for the calculations. Also,  $1 \text{ Mg/m}^3 = 10^{-6} \text{ kg/mm}^3$ , so that

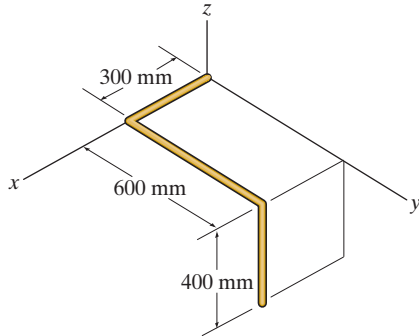
Segment	$m$ (kg)	$\tilde{z}$ (mm)	$\tilde{z}m$ (kg · mm)
1	$8(10^{-6})\left(\frac{1}{3}\right)\pi(50)^2(200) = 4.189$	50	209.440
2	$4(10^{-6})\left(\frac{2}{3}\right)\pi(50)^3 = 1.047$	-18.75	-19.635
3	$-8(10^{-6})\left(\frac{1}{3}\right)\pi(25)^2(100) = -0.524$	$100 + 25 = 125$	-65.450
4	$-8(10^{-6})\pi(25)^2(100) = -1.571$	50	-78.540
$\Sigma m = 3.142$			$\Sigma \tilde{z}m = 45.815$

$$\text{Thus, } \bar{z} = \frac{\Sigma \tilde{z}m}{\Sigma m} = \frac{45.815}{3.142} = 14.6 \text{ mm} \quad \text{Ans.}$$



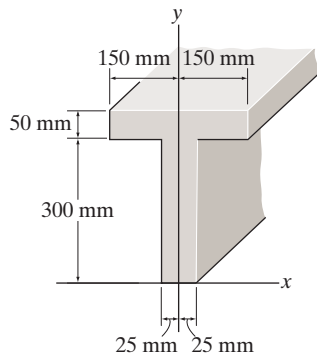
## FUNDAMENTAL PROBLEMS

**F9-7.** Locate the centroid  $(\bar{x}, \bar{y}, \bar{z})$  of the wire bent in the shape shown.



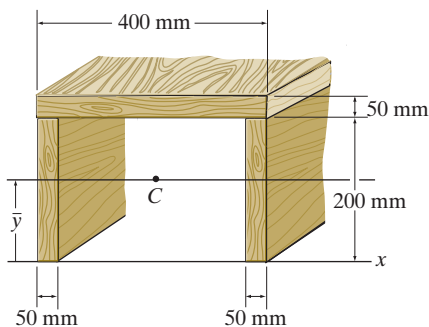
**Prob. F9-7**

**F9-8.** Locate the centroid  $\bar{y}$  of the beam's cross-sectional area.



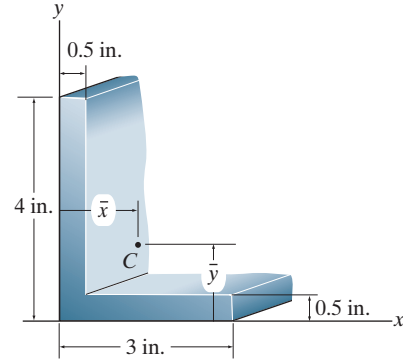
**Prob. F9-8**

**F9-9.** Locate the centroid  $\bar{y}$  of the beam's cross-sectional area.



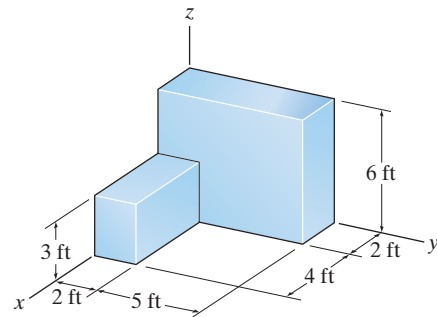
**Prob. F9-9**

**F9-10.** Locate the centroid  $(\bar{x}, \bar{y})$  of the cross-sectional area.



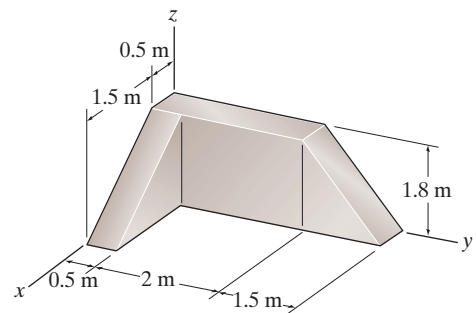
**Prob. F9-10**

**F9-11.** Locate the center of mass  $(\bar{x}, \bar{y}, \bar{z})$  of the homogeneous solid block.



**Prob. F9-11**

**F9-12.** Determine the center of mass  $(\bar{x}, \bar{y}, \bar{z})$  of the homogeneous solid block.

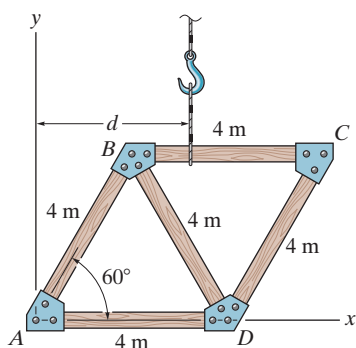


**Prob. F9-12**



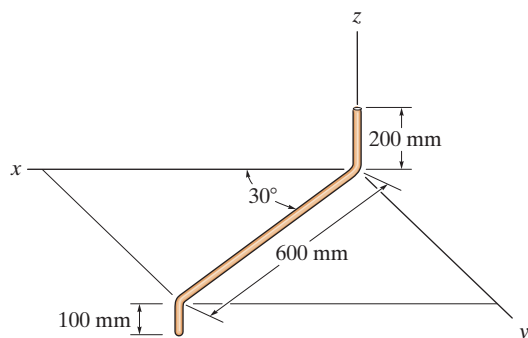
## PROBLEMS

**9-51.** The truss is made from five members, each having a length of 4 m and a mass of 7 kg/m. If the mass of the gusset plates at the joints and the thickness of the members can be neglected, determine the distance  $d$  to where the hoisting cable must be attached, so that the truss does not tip (rotate) when it is lifted.



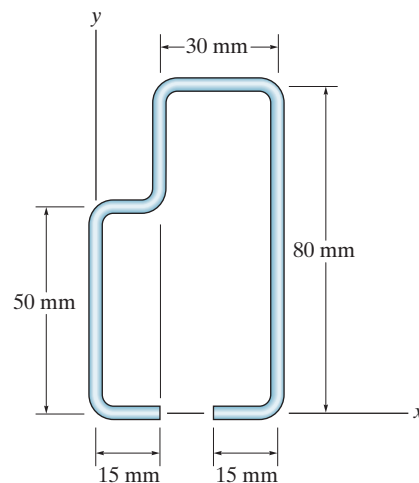
Prob. 9-51

**\*9-52.** Determine the location  $(\bar{x}, \bar{y}, \bar{z})$  of the centroid of the homogeneous rod.



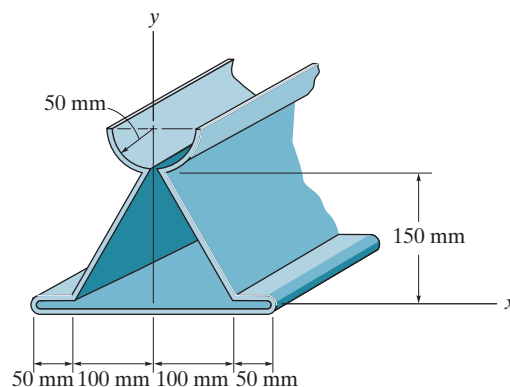
Prob. 9-52

**9-53.** A rack is made from roll-formed sheet steel and has the cross section shown. Determine the location  $(\bar{x}, \bar{y})$  of the centroid of the cross section. The dimensions are indicated at the center thickness of each segment.



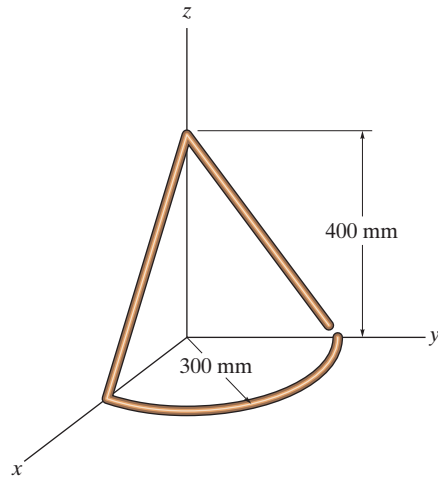
Prob. 9-53

**9-54.** Locate the centroid  $(\bar{x}, \bar{y})$  of the metal cross section. Neglect the thickness of the material and slight bends at the corners.



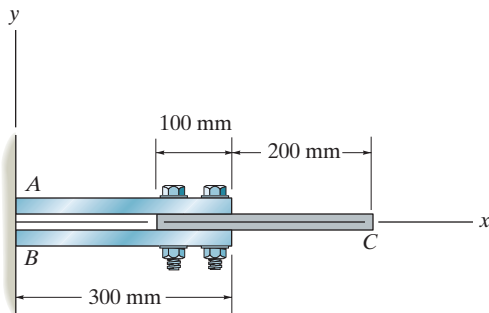
Prob. 9-54

**9–55.** Locate the center of gravity  $(\bar{x}, \bar{y}, \bar{z})$  of the homogeneous wire.



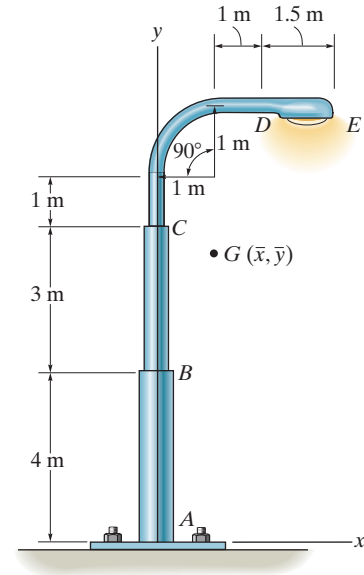
**Prob. 9–55**

**\*9–56.** The steel and aluminum plate assembly is bolted together and fastened to the wall. Each plate has a constant width in the  $z$  direction of 200 mm and thickness of 20 mm. If the density of  $A$  and  $B$  is  $\rho_s = 7.85 \text{ Mg/m}^3$ , and for  $C$ ,  $\rho_{al} = 2.71 \text{ Mg/m}^3$ , determine the location  $\bar{x}$  of the center of mass. Neglect the size of the bolts.



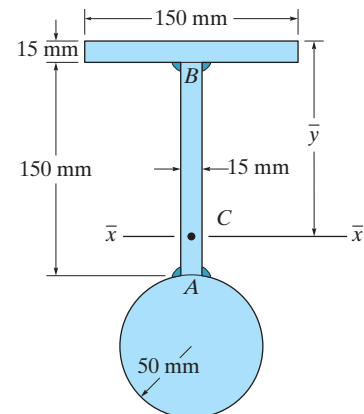
**Prob. 9–56**

**9–57.** Locate the center of gravity  $G(\bar{x}, \bar{y})$  of the streetlight. Neglect the thickness of each segment. The mass per unit length of each segment is as follows:  $\rho_{AB} = 12 \text{ kg/m}$ ,  $\rho_{BC} = 8 \text{ kg/m}$ ,  $\rho_{CD} = 5 \text{ kg/m}$ , and  $\rho_{DE} = 2 \text{ kg/m}$ .



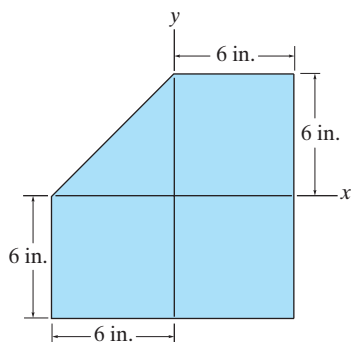
**Prob. 9–57**

**9–58.** Determine the location  $\bar{y}$  of the centroidal axis  $\bar{x}-\bar{x}$  of the beam's cross-sectional area. Neglect the size of the corner welds at  $A$  and  $B$  for the calculation.



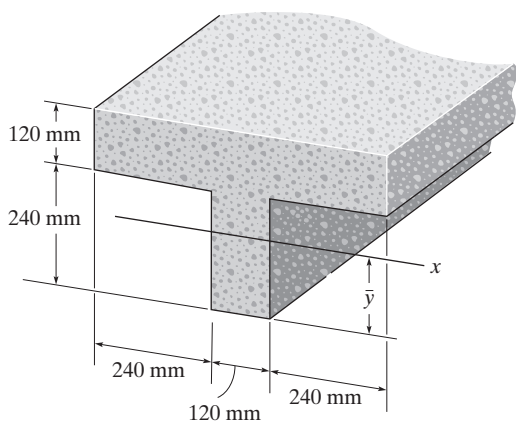
**Prob. 9–58**

**9-59.** Locate the centroid  $(\bar{x}, \bar{y})$  of the shaded area.



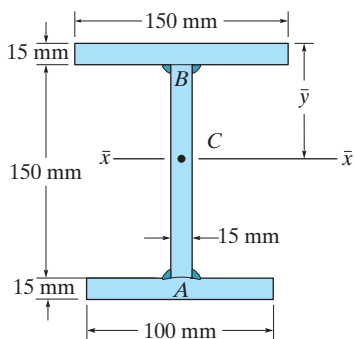
**Prob. 9-59**

**\*9-60.** Locate the centroid  $\bar{y}$  for the beam's cross-sectional area.



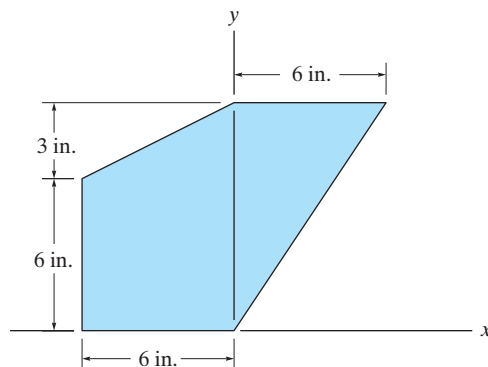
**Prob. 9-60**

**9-61.** Determine the location  $\bar{y}$  of the centroid  $C$  of the beam having the cross-sectional area shown.



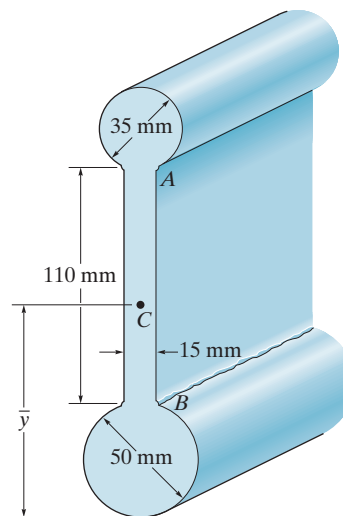
**Prob. 9-61**

**9-62.** Locate the centroid  $(\bar{x}, \bar{y})$  of the shaded area.



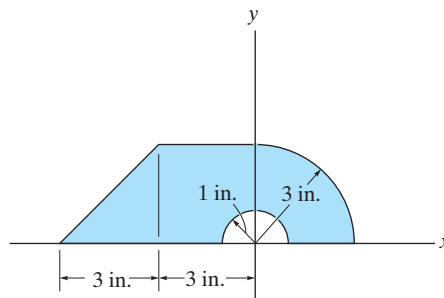
**Prob. 9-62**

**9-63.** Determine the location  $\bar{y}$  of the centroid of the beam's cross-sectional area. Neglect the size of the corner welds at  $A$  and  $B$  for the calculation.



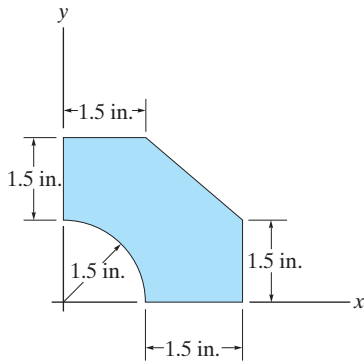
**Prob. 9-63**

**\*9-64.** Locate the centroid  $(\bar{x}, \bar{y})$  of the shaded area.



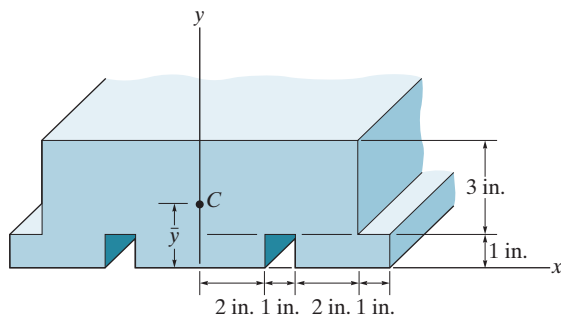
**Prob. 9-64**

**9-65.** Determine the location  $(\bar{x}, \bar{y})$  of the centroid  $C$  of the area.



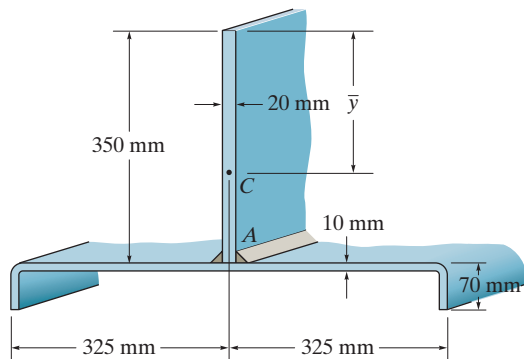
**Prob. 9-65**

**9-66.** Determine the location  $\bar{y}$  of the centroid  $C$  for a beam having the cross-sectional area shown. The beam is symmetric with respect to the  $y$  axis.



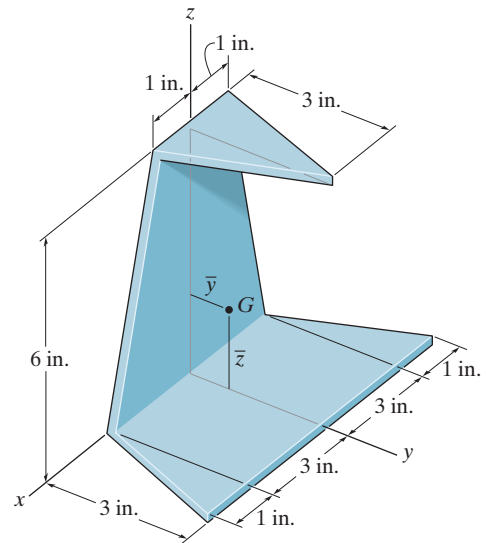
**Prob. 9-66**

**9-67.** Locate the centroid  $\bar{y}$  of the cross-sectional area of the beam constructed from a channel and a plate. Assume all corners are square and neglect the size of the weld at  $A$ .



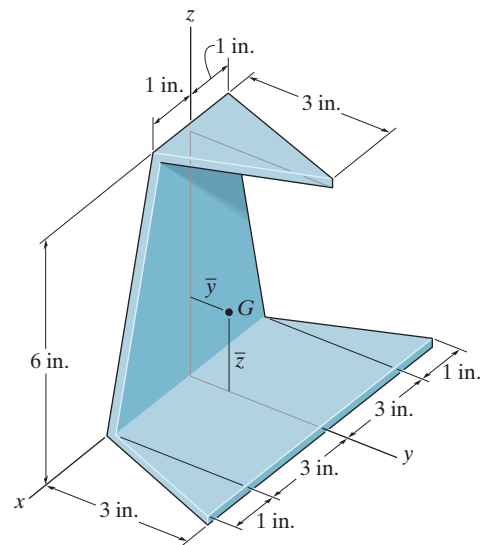
**Prob. 9-67**

**\*9-68.** A triangular plate made of homogeneous material has a constant thickness that is very small. If it is folded over as shown, determine the location  $\bar{y}$  of the plate's center of gravity  $G$ .



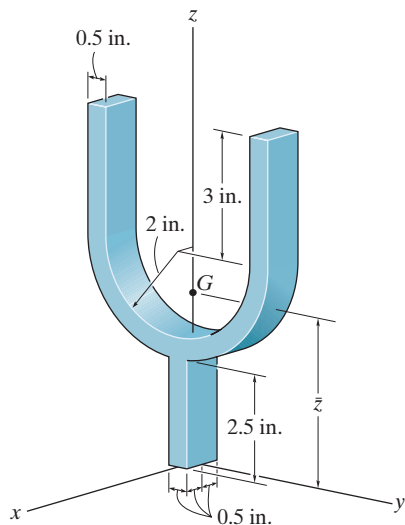
**Prob. 9-68**

**9-69.** A triangular plate made of homogeneous material has a constant thickness that is very small. If it is folded over as shown, determine the location  $\bar{z}$  of the plate's center of gravity  $G$ .



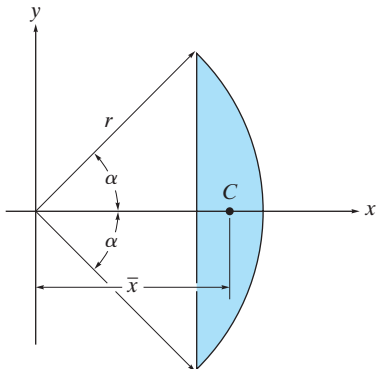
**Prob. 9-69**

**9-70.** Locate the center of mass  $\bar{z}$  of the forked level which is made from a homogeneous material and has the dimensions shown.



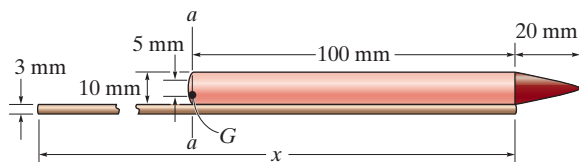
**Prob. 9-70**

**9-71.** Determine the location  $\bar{x}$  of the centroid  $C$  of the shaded area that is part of a circle having a radius  $r$ .



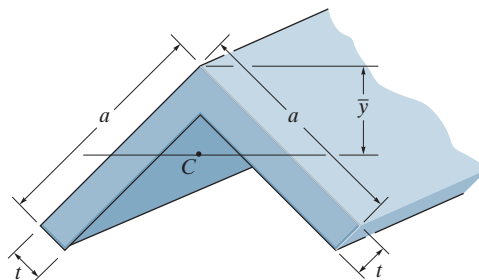
**Prob. 9-71**

**\*9-72.** A toy skyrocket consists of a solid conical top,  $\rho_i = 600 \text{ kg/m}^3$ , a hollow cylinder,  $\rho_c = 400 \text{ kg/m}^3$ , and a stick having a circular cross section,  $\rho_s = 300 \text{ kg/m}^3$ . Determine the length of the stick,  $x$ , so that the center of gravity  $G$  of the skyrocket is located along line  $aa$ .



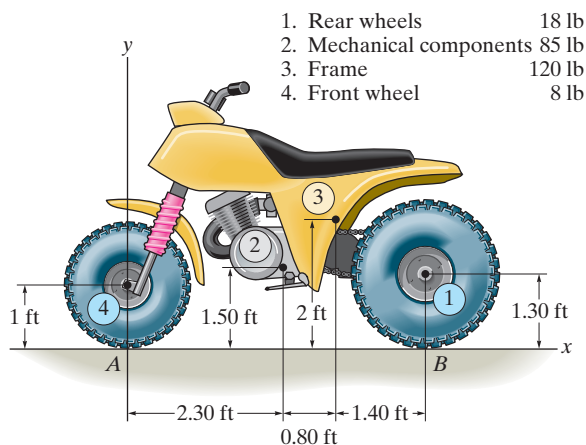
**Prob. 9-72**

**9-73.** Locate the centroid  $\bar{y}$  for the cross-sectional area of the angle.



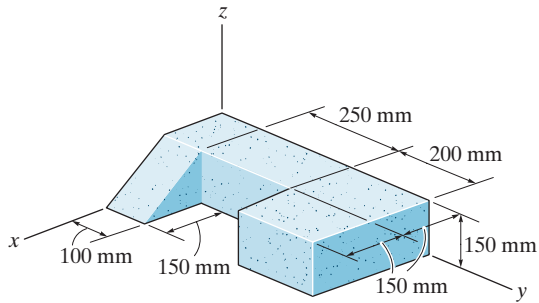
**Prob. 9-73**

**9-74.** Determine the location  $(\bar{x}, \bar{y})$  of the center of gravity of the three-wheeler. The location of the center of gravity of each component and its weight are tabulated in the figure. If the three-wheeler is symmetrical with respect to the  $x$ - $y$  plane, determine the normal reaction each of its wheels exerts on the ground.



**Prob. 9-74**

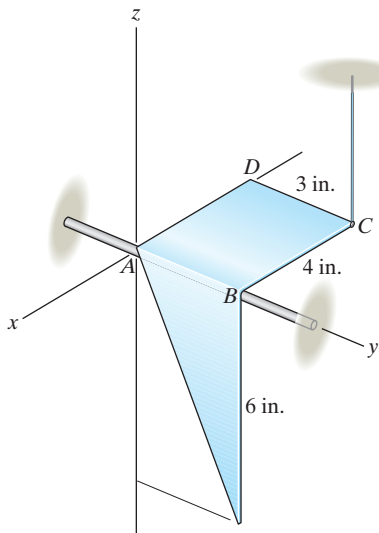
**9-75.** Locate the center of mass ( $\bar{x}$ ,  $\bar{y}$ ,  $\bar{z}$ ) of the homogeneous block assembly.



**Prob. 9-75**

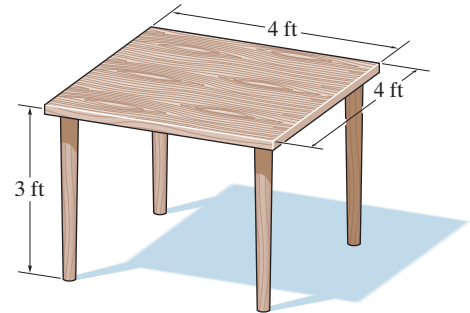
**\*9-76.** The sheet metal part has the dimensions shown. Determine the location ( $\bar{x}$ ,  $\bar{y}$ ,  $\bar{z}$ ) of its centroid.

**9-77.** The sheet metal part has a weight per unit area of 2 lb/ft<sup>2</sup> and is supported by the smooth rod and the cord at  $C$ . If the cord is cut, the part will rotate about the  $y$  axis until it reaches equilibrium. Determine the equilibrium angle of tilt, measured downward from the negative  $x$  axis, that  $AD$  makes with the  $-x$  axis.



**Probs. 9-76/77**

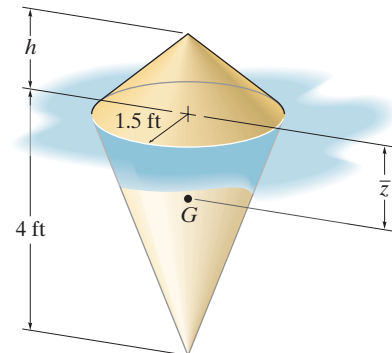
**9-78.** The wooden table is made from a square board having a weight of 15 lb. Each of the legs weighs 2 lb and is 3 ft long. Determine how high its center of gravity is from the floor. Also, what is the angle, measured from the horizontal, through which its top surface can be tilted on two of its legs before it begins to overturn? Neglect the thickness of each leg.



**Prob. 9-78**

**9-79.** The buoy is made from two homogeneous cones each having a radius of 1.5 ft. If  $h = 1.2$  ft, find the distance  $\bar{z}$  to the buoy's center of gravity  $G$ .

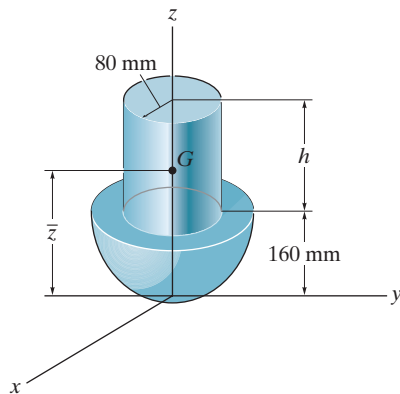
**\*9-80.** The buoy is made from two homogeneous cones each having a radius of 1.5 ft. If it is required that the buoy's center of gravity  $G$  be located at  $\bar{z} = 0.5$  ft, determine the height  $h$  of the top cone.



**Probs. 9-79/80**

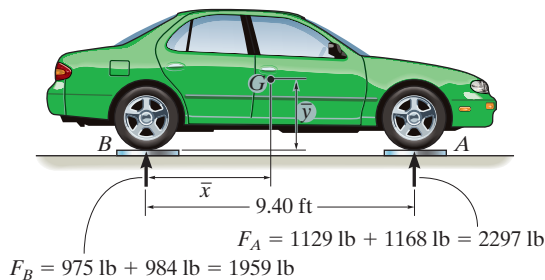
**9-81.** The assembly is made from a steel hemisphere,  $\rho_{st} = 7.80 \text{ Mg/m}^3$ , and an aluminum cylinder,  $\rho_{al} = 2.70 \text{ Mg/m}^3$ . Determine the mass center of the assembly if the height of the cylinder is  $h = 200 \text{ mm}$ .

**9-82.** The assembly is made from a steel hemisphere,  $\rho_{st} = 7.80 \text{ Mg/m}^3$ , and an aluminum cylinder,  $\rho_{al} = 2.70 \text{ Mg/m}^3$ . Determine the height  $h$  of the cylinder so that the mass center of the assembly is located at  $\bar{z} = 160 \text{ mm}$ .



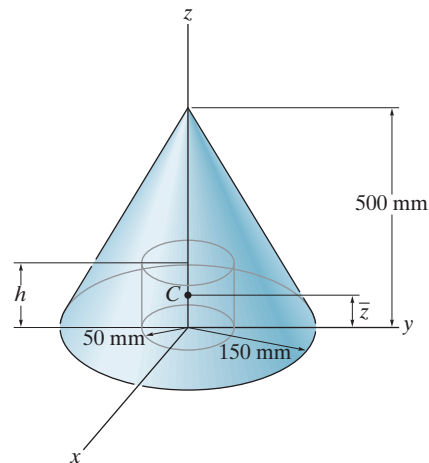
**Probs. 9-81/82**

**9-83.** The car rests on four scales and in this position the scale readings of both the front and rear tires are shown by  $F_A$  and  $F_B$ . When the rear wheels are elevated to a height of 3 ft above the front scales, the new readings of the front wheels are also recorded. Use this data to compute the location  $\bar{x}$  and  $\bar{y}$  to the center of gravity  $G$  of the car. The tires each have a diameter of 1.98 ft.



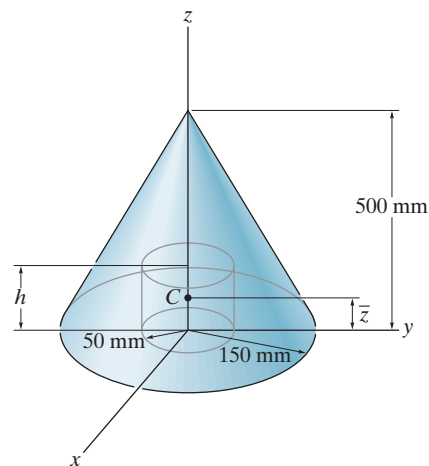
**Prob. 9-83**

**\*9-84.** Determine the distance  $h$  to which a 100-mm-diameter hole must be bored into the base of the cone so that the center of mass of the resulting shape is located at  $\bar{z} = 115 \text{ mm}$ . The material has a density of  $8 \text{ Mg/m}^3$ .



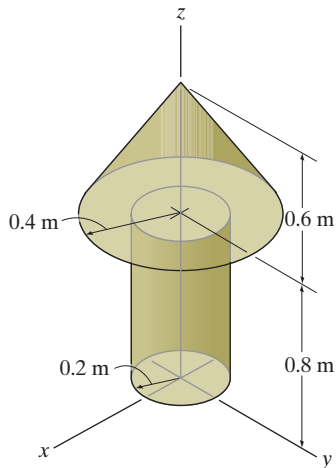
**Prob. 9-84**

**9-85.** Determine the distance  $\bar{z}$  to the centroid of the shape that consists of a cone with a hole of height  $h = 50 \text{ mm}$  bored into its base.



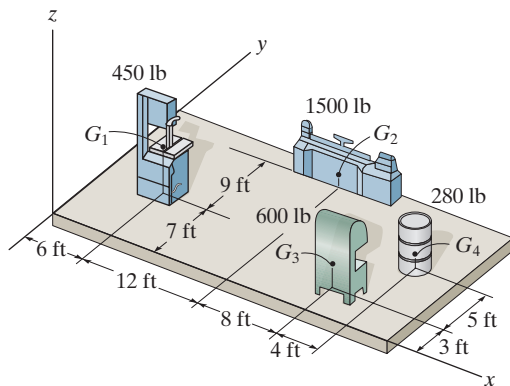
**Prob. 9-85**

**9-86.** Locate the center of mass  $\bar{z}$  of the assembly. The cylinder and the cone are made from materials having densities of  $5 \text{ Mg/m}^3$  and  $9 \text{ Mg/m}^3$ , respectively.



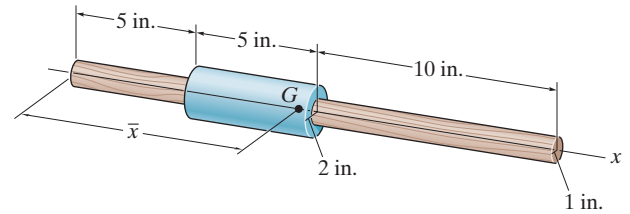
**Prob. 9-86**

**9-87.** Major floor loadings in a shop are caused by the weights of the objects shown. Each force acts through its respective center of gravity  $G$ . Locate the center of gravity  $(\bar{x}, \bar{y})$  of all these components.



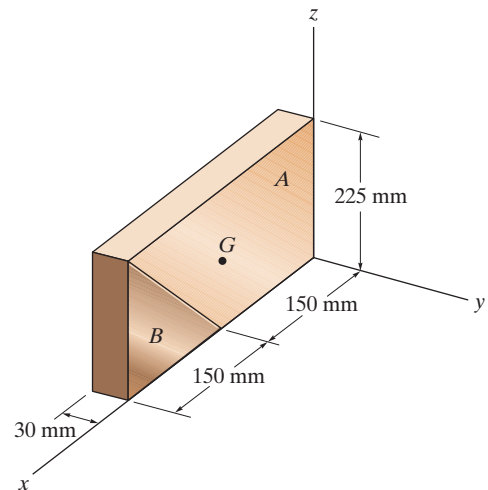
**Prob. 9-87**

**\*9-88.** The assembly consists of a 20-in. wooden dowel rod and a tight-fitting steel collar. Determine the distance  $\bar{x}$  to its center of gravity if the specific weights of the materials are  $\gamma_w = 150 \text{ lb/ft}^3$  and  $\gamma_{st} = 490 \text{ lb/ft}^3$ . The radii of the dowel and collar are shown.



**Prob. 9-88**

**9-89.** The composite plate is made from both steel ( $A$ ) and brass ( $B$ ) segments. Determine the mass and location  $(\bar{x}, \bar{y}, \bar{z})$  of its mass center  $G$ . Take  $\rho_{st} = 7.85 \text{ Mg/m}^3$  and  $\rho_{br} = 8.74 \text{ Mg/m}^3$ .



**Prob. 9-89**



### \*9.3 Theorems of Pappus and Guldinus

The two *theorems of Pappus and Guldinus* are used to find the surface area and volume of any body of revolution. They were first developed by Pappus of Alexandria during the fourth century A.D. and then restated at a later time by the Swiss mathematician Paul Guldin or Guldinus (1577–1643).

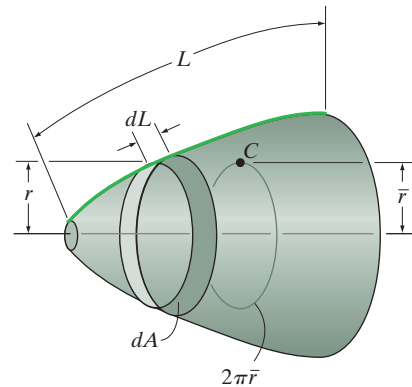


Fig. 9–19

**Surface Area.** If we revolve a *plane curve* about an axis that does not intersect the curve we will generate a *surface area of revolution*. For example, the surface area in Fig. 9–19 is formed by revolving the curve of length  $L$  about the horizontal axis. To determine this surface area, we will first consider the differential line element of length  $dL$ . If this element is revolved  $2\pi$  radians about the axis, a ring having a surface area of  $dA = 2\pi r dL$  will be generated. Thus, the surface area of the entire body is  $A = 2\pi \int r dL$ . Since  $\int r dL = \bar{r}L$  (Eq. 9–5), then  $A = 2\pi \bar{r}L$ . If the curve is revolved only through an angle  $\theta$  (radians), then

$$A = \theta \bar{r}L \quad (9-7)$$

where

$A$  = surface area of revolution

$\theta$  = angle of revolution measured in radians,  $\theta \leq 2\pi$

$\bar{r}$  = perpendicular distance from the axis of revolution to the centroid of the generating curve

$L$  = length of the generating curve



The amount of material used on this storage building can be estimated by using the first theorem of Pappus and Guldinus to determine its surface area. (© Russell C. Hibbeler)

Therefore the first theorem of Pappus and Guldinus states that *the area of a surface of revolution equals the product of the length of the generating curve and the distance traveled by the centroid of the curve in generating the surface area.*

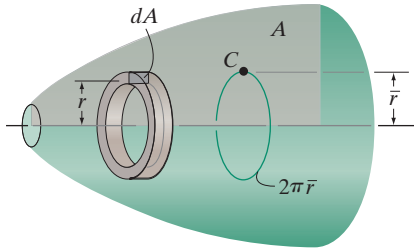


Fig. 9-20

**Volume.** A volume can be generated by revolving a *plane area* about an axis that does not intersect the area. For example, if we revolve the shaded area  $A$  in Fig. 9-20 about the horizontal axis, it generates the volume shown. This volume can be determined by first revolving the differential element of area  $dA$   $2\pi$  radians about the axis, so that a ring having the volume  $dV = 2\pi r dA$  is generated. The entire volume is then  $V = 2\pi \int r dA$ . However,  $\int r dA = \bar{r} A$ , Eq. 9-4, so that  $V = 2\pi \bar{r} A$ . If the area is only revolved through an angle  $\theta$  (radians), then

$$V = \theta \bar{r} A \quad (9-8)$$

where

$V$  = volume of revolution

$\theta$  = angle of revolution measured in radians,  $\theta \leq 2\pi$

$\bar{r}$  = perpendicular distance from the axis of revolution to the centroid of the generating area

$A$  = generating area

Therefore the second theorem of Pappus and Guldinus states that *the volume of a body of revolution equals the product of the generating area and the distance traveled by the centroid of the area in generating the volume.*

**Composite Shapes.** We may also apply the above two theorems to lines or areas that are composed of a series of composite parts. In this case the total surface area or volume generated is the addition of the surface areas or volumes generated by each of the composite parts. If the perpendicular distance from the axis of revolution to the centroid of each composite part is  $\tilde{r}$ , then

$$A = \theta \Sigma(\tilde{r} L) \quad (9-9)$$

and

$$V = \theta \Sigma(\tilde{r} A) \quad (9-10)$$

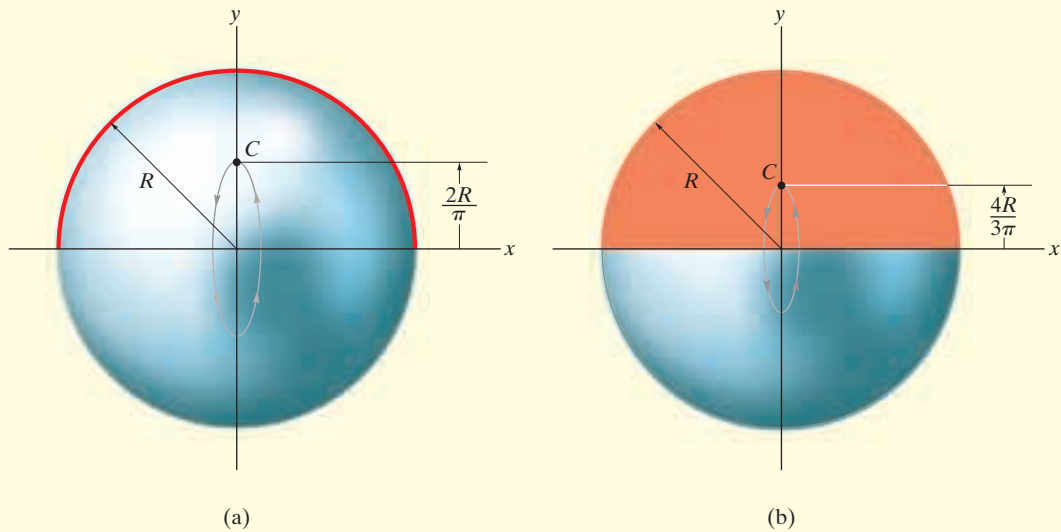
Application of the above theorems is illustrated numerically in the following examples.



The volume of fertilizer contained within this silo can be determined using the second theorem of Pappus and Guldinus. (© Russell C. Hibbeler)

**EXAMPLE 9.12**

Show that the surface area of a sphere is  $A = 4\pi R^2$  and its volume is  $V = \frac{4}{3}\pi R^3$ .



**Fig. 9-21**

**SOLUTION**

**Surface Area.** The surface area of the sphere in Fig. 9-21a is generated by revolving a semicircular *arc* about the  $x$  axis. Using the table on the inside back cover, it is seen that the centroid of this arc is located at a distance  $\bar{r} = 2R/\pi$  from the axis of revolution ( $x$  axis). Since the centroid moves through an angle of  $\theta = 2\pi$  rad to generate the sphere, then applying Eq. 9-7 we have

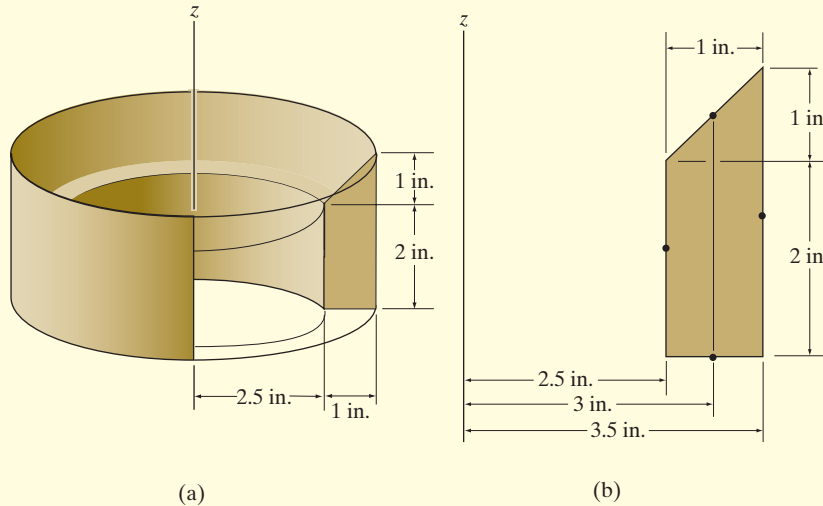
$$A = \theta \bar{r} L; \quad A = 2\pi \left( \frac{2R}{\pi} \right) \pi R = 4\pi R^2 \quad \text{Ans.}$$

**Volume.** The volume of the sphere is generated by revolving the semicircular *area* in Fig. 9-21b about the  $x$  axis. Using the table on the inside back cover to locate the centroid of the area, i.e.,  $\bar{r} = 4R/3\pi$ , and applying Eq. 9-8, we have

$$V = \theta \bar{r} A; \quad V = 2\pi \left( \frac{4R}{3\pi} \right) \left( \frac{1}{2} \pi R^2 \right) = \frac{4}{3} \pi R^3 \quad \text{Ans.}$$

**EXAMPLE 9.13**

Determine the surface area and volume of the full solid in Fig. 9-22a.



**Fig. 9-22**

**SOLUTION**

**Surface Area.** The surface area is generated by revolving the four line segments shown in Fig. 9-22b  $2\pi$  radians about the  $z$  axis. The distances from the centroid of each segment to the  $z$  axis are also shown in the figure. Applying Eq. 9-7 yields

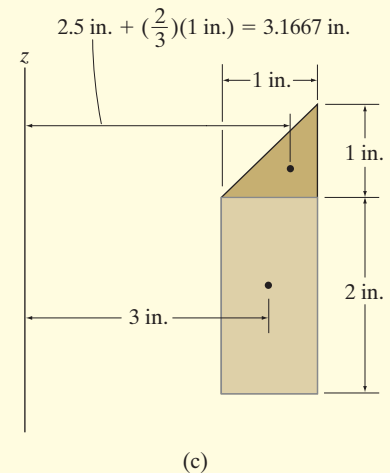
$$\begin{aligned}
 A &= 2\pi \sum \bar{r}L = 2\pi[(2.5 \text{ in.})(2 \text{ in.}) + (3 \text{ in.})\left(\sqrt{(1 \text{ in.})^2 + (1 \text{ in.})^2}\right) \\
 &\quad + (3.5 \text{ in.})(3 \text{ in.}) + (3 \text{ in.})(1 \text{ in.})] \\
 &= 143 \text{ in}^2
 \end{aligned}$$

*Ans.*

**Volume.** The volume of the solid is generated by revolving the two area segments shown in Fig. 9-22c  $2\pi$  radians about the  $z$  axis. The distances from the centroid of each segment to the  $z$  axis are also shown in the figure. Applying Eq. 9-10, we have

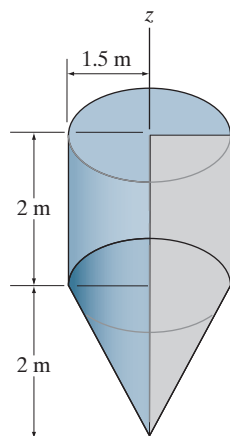
$$\begin{aligned}
 V &= 2\pi \sum \bar{r}A = 2\pi \left\{ (3.1667 \text{ in.}) \left[ \frac{1}{2} (1 \text{ in.})(1 \text{ in.}) \right] + (3 \text{ in.})[(2 \text{ in.})(1 \text{ in.})] \right\} \\
 &= 47.6 \text{ in}^3
 \end{aligned}$$

*Ans.*



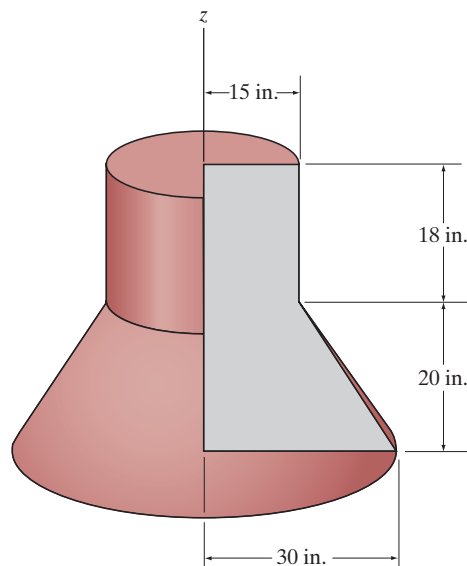
## FUNDAMENTAL PROBLEMS

**F9-13.** Determine the surface area and volume of the solid formed by revolving the shaded area  $360^\circ$  about the  $z$  axis.



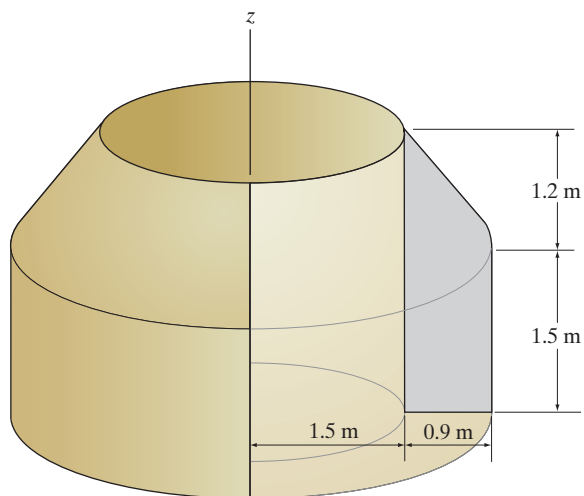
**Prob. F9-13**

**F9-15.** Determine the surface area and volume of the solid formed by revolving the shaded area  $360^\circ$  about the  $z$  axis.



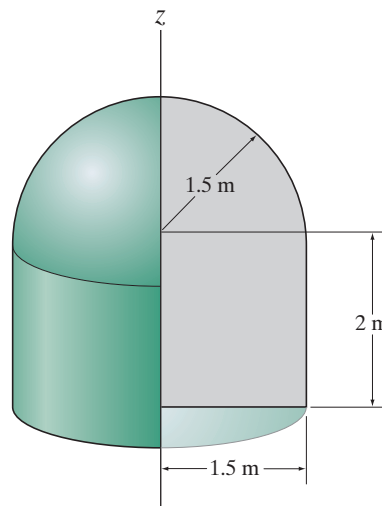
**Prob. F9-15**

**F9-14.** Determine the surface area and volume of the solid formed by revolving the shaded area  $360^\circ$  about the  $z$  axis.



**Prob. F9-14**

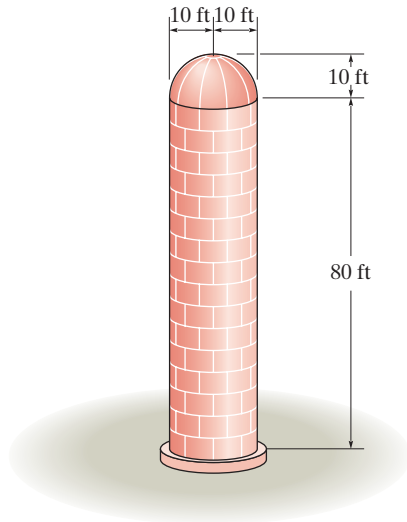
**F9-16.** Determine the surface area and volume of the solid formed by revolving the shaded area  $360^\circ$  about the  $z$  axis.



**Prob. F9-16**

## PROBLEMS

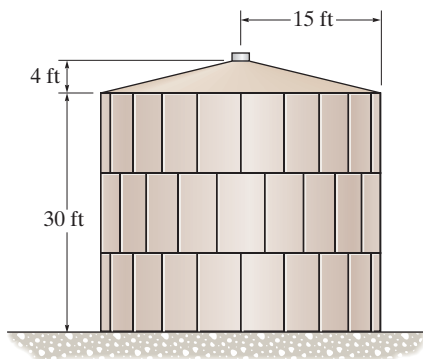
**9-90.** Determine the volume of the silo which consists of a cylinder and hemispherical cap. Neglect the thickness of the plates.



**Prob. 9-90**

**9-91.** Determine the outside surface area of the storage tank.

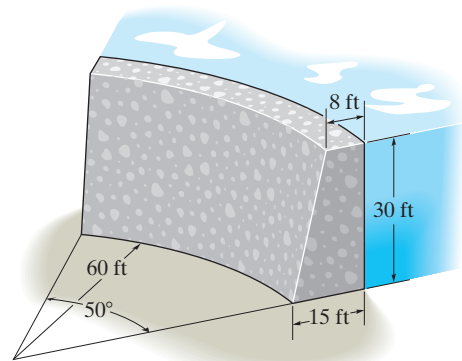
**\*9-92.** Determine the volume of the storage tank.



**Probs. 9-91/92**

**9-93.** Determine the surface area of the concrete seawall, excluding its bottom.

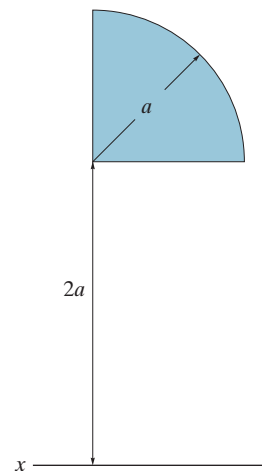
**9-94.** A circular seawall is made of concrete. Determine the total weight of the wall if the concrete has a specific weight of  $\gamma_c = 150 \text{ lb/ft}^3$ .



**Probs. 9-93/94**

**9-95.** A ring is generated by rotating the quarter circular area about the  $x$  axis. Determine its volume.

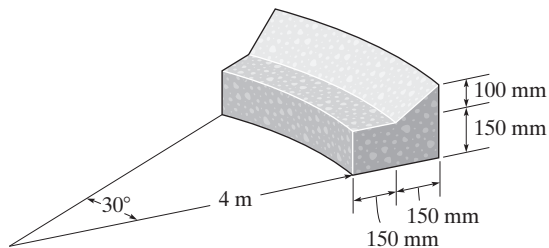
**\*9-96.** A ring is generated by rotating the quarter circular area about the  $x$  axis. Determine its surface area.



**Probs. 9-95/96**

**9-97.** Determine the volume of concrete needed to construct the curb.

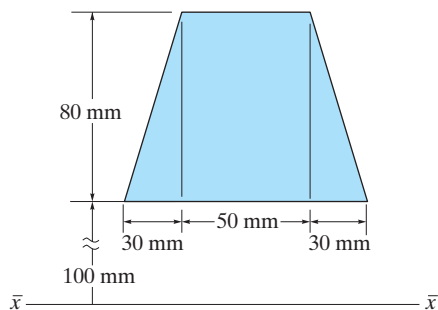
**9-98.** Determine the surface area of the curb. Do not include the area of the ends in the calculation.



**Probs. 9-97/98**

**9-99.** A ring is formed by rotating the area  $360^\circ$  about the  $\bar{x}-\bar{x}$  axes. Determine its surface area.

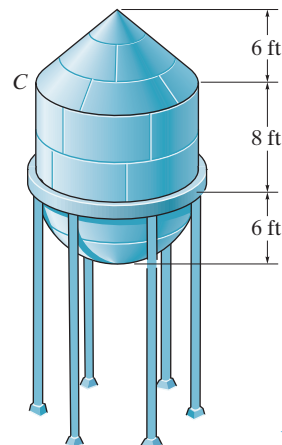
**\*9-100.** A ring is formed by rotating the area  $360^\circ$  about the  $\bar{x}-\bar{x}$  axes. Determine its volume.



**Probs. 9-99/100**

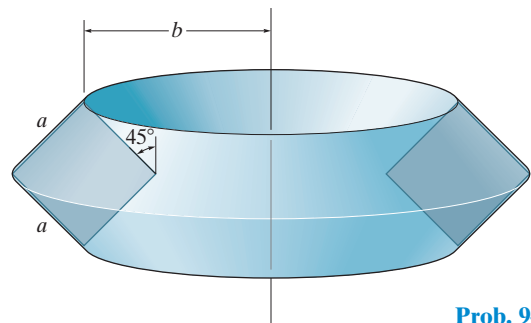
**9-101.** The water-supply tank has a hemispherical bottom and cylindrical sides. Determine the weight of water in the tank when it is filled to the top at C. Take  $\gamma_w = 62.4 \text{ lb/ft}^3$ .

**9-102.** Determine the number of gallons of paint needed to paint the outside surface of the water-supply tank, which consists of a hemispherical bottom, cylindrical sides, and conical top. Each gallon of paint can cover  $250 \text{ ft}^2$ .



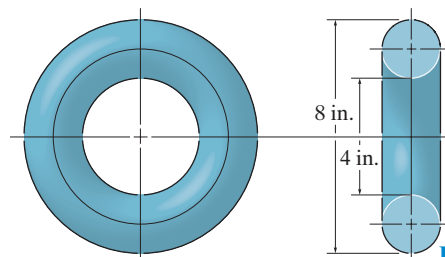
**Probs. 9-101/102**

**9-103.** Determine the surface area and the volume of the ring formed by rotating the square about the vertical axis.



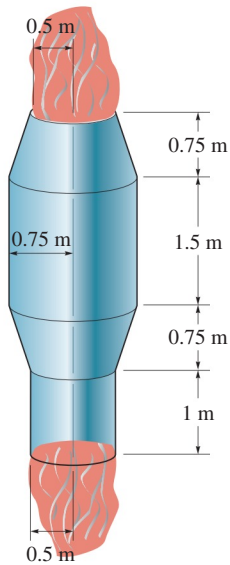
**Prob. 9-103**

**\*9-104.** Determine the surface area of the ring. The cross section is circular as shown.



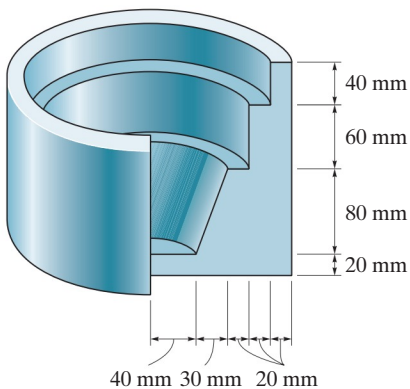
**Prob. 9-104**

**9–105.** The heat exchanger radiates thermal energy at the rate of 2500 kJ/h for each square meter of its surface area. Determine how many joules (J) are radiated within a 5-hour period.



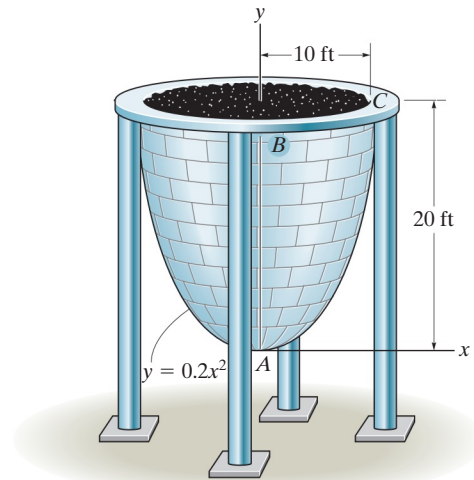
**Prob. 9–105**

**9–106.** Determine the interior surface area of the brake piston. It consists of a full circular part. Its cross section is shown in the figure.



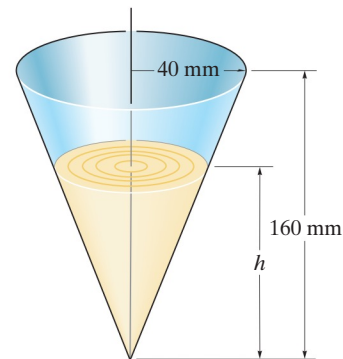
**Prob. 9–106**

**9–107.** The suspension bunker is made from plates which are curved to the natural shape which a completely flexible membrane would take if subjected to a full load of coal. This curve may be approximated by a parabola,  $y = 0.2x^2$ . Determine the weight of coal which the bunker would contain when completely filled. Coal has a specific weight of  $\gamma = 50 \text{ lb/ft}^3$ , and assume there is a 20% loss in volume due to air voids. Solve the problem by integration to determine the cross-sectional area of  $ABC$ ; then use the second theorem of Pappus–Guldinus to find the volume.



**Prob. 9–107**

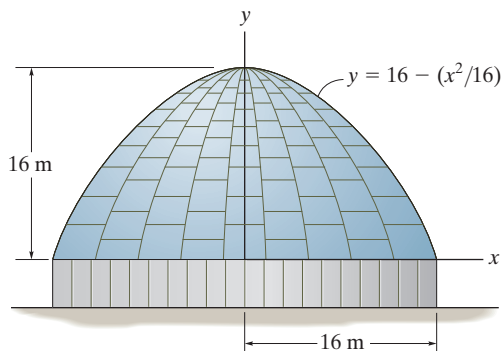
**\*9–108.** Determine the height  $h$  to which liquid should be poured into the cup so that it contacts three-fourths the surface area on the inside of the cup. Neglect the cup's thickness for the calculation.



**Prob. 9–108**

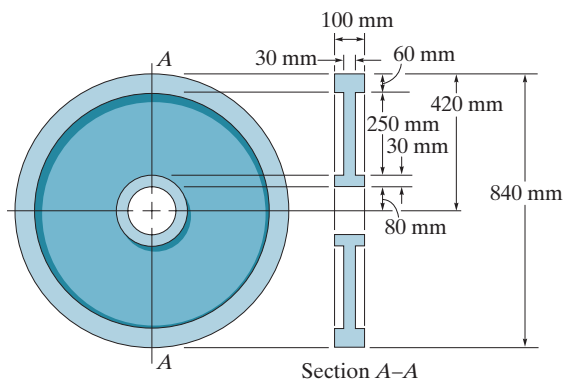


**9-109.** Determine the surface area of the roof of the structure if it is formed by rotating the parabola about the  $y$  axis.



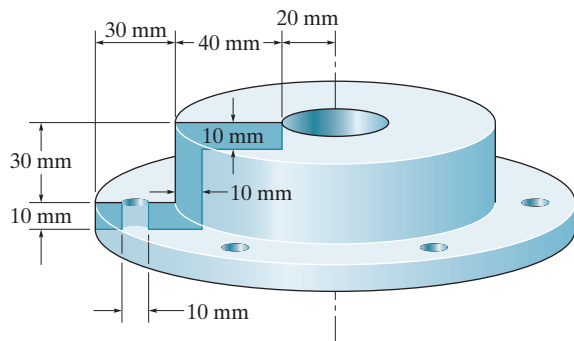
**Prob. 9-109**

**9-110.** A steel wheel has a diameter of 840 mm and a cross section as shown in the figure. Determine the total mass of the wheel if  $\rho = 5 \text{ Mg/m}^3$ .



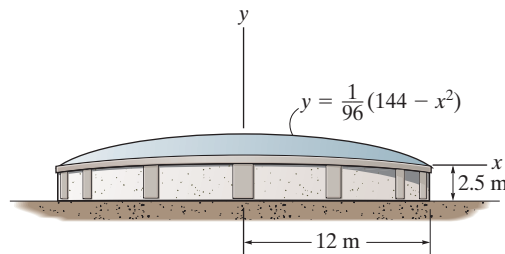
**Prob. 9-110**

**9-111.** Half the cross section of the steel housing is shown in the figure. There are six 10-mm-diameter bolt holes around its rim. Determine its mass. The density of steel is  $785 \text{ Mg/m}^3$ . The housing is a full circular part.



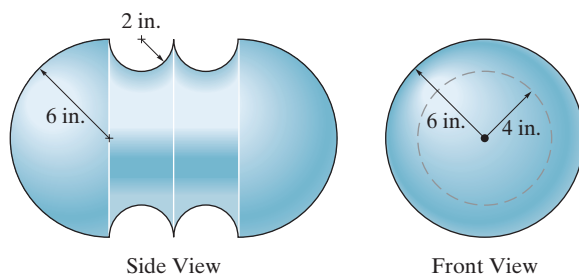
**Prob. 9-111**

**\*9-112.** The water tank has a paraboloid-shaped roof. If one liter of paint can cover  $3 \text{ m}^2$  of the tank, determine the number of liters required to coat the roof.



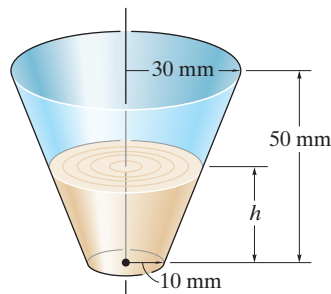
**Prob. 9-112**

**9-113.** Determine the volume of material needed to make the casting.



**Prob. 9-113**

**9-114.** Determine the height  $h$  to which liquid should be poured into the cup so that it contacts half the surface area on the inside of the cup. Neglect the cup's thickness for the calculation.



**Prob. 9-114**

## \*9.4 Resultant of a General Distributed Loading

In Sec. 4.9, we discussed the method used to simplify a two-dimensional distributed loading to a single resultant force acting at a specific point. In this section we will generalize this method to include flat surfaces that have an arbitrary shape and are subjected to a variable load distribution. Consider, for example, the flat plate shown in Fig. 9–23a, which is subjected to the loading defined by  $p = p(x, y)$  Pa, where 1 Pa (pascal) = 1 N/m<sup>2</sup>. Knowing this function, we can determine the resultant force  $\mathbf{F}_R$  acting on the plate and its location  $(\bar{x}, \bar{y})$ , Fig. 9–23b.

**Magnitude of Resultant Force.** The force  $d\mathbf{F}$  acting on the differential area  $dA$  m<sup>2</sup> of the plate, located at the arbitrary point  $(x, y)$ , has a magnitude of  $dF = [p(x, y) \text{ N/m}^2](dA \text{ m}^2) = [p(x, y) dA] \text{ N}$ . Notice that  $p(x, y) dA = dV$ , the colored differential *volume element* shown in Fig. 9–23a. The *magnitude* of  $\mathbf{F}_R$  is the sum of the differential forces acting over the plate's *entire surface area*  $A$ . Thus:

$$F_R = \Sigma F; \quad F_R = \int_A p(x, y) dA = \int_V dV = V \quad (9-11)$$

This result indicates that the *magnitude of the resultant force is equal to the total volume under the distributed-loading diagram*.

**Location of Resultant Force.** The location  $(\bar{x}, \bar{y})$  of  $\mathbf{F}_R$  is determined by setting the moments of  $\mathbf{F}_R$  equal to the moments of all the differential forces  $d\mathbf{F}$  about the respective  $y$  and  $x$  axes: From Figs. 9–23a and 9–23b, using Eq. 9–11, this results in

$$\bar{x} = \frac{\int_A xp(x, y) dA}{\int_A p(x, y) dA} = \frac{\int_V x dV}{\int_V dV} \quad \bar{y} = \frac{\int_A yp(x, y) dA}{\int_A p(x, y) dA} = \frac{\int_V y dV}{\int_V dV} \quad (9-12)$$

Hence, the *line of action of the resultant force passes through the geometric center or centroid of the volume under the distributed-loading diagram*.

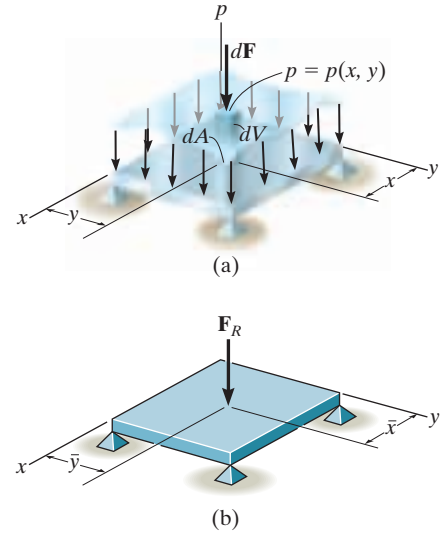


Fig. 9–23



The resultant of a wind loading that is distributed on the front or side walls of this building must be calculated using integration in order to design the framework that holds the building together. (© Russell C. Hibbeler)

## \*9.5 Fluid Pressure

According to Pascal's law, a fluid at rest creates a pressure  $p$  at a point that is the *same* in *all* directions. The magnitude of  $p$ , measured as a force per unit area, depends on the specific weight  $\gamma$  or mass density  $\rho$  of the fluid and the depth  $z$  of the point from the fluid surface.\* The relationship can be expressed mathematically as

$$p = \gamma z = \rho g z \quad (9-13)$$

where  $g$  is the acceleration due to gravity. This equation is valid only for fluids that are assumed *incompressible*, as in the case of most liquids. Gases are compressible fluids, and since their density changes significantly with both pressure and temperature, Eq. 9-13 cannot be used.

To illustrate how Eq. 9-13 is applied, consider the submerged plate shown in Fig. 9-24. Three points on the plate have been specified. Since point  $B$  is at depth  $z_1$  from the liquid surface, the *pressure* at this point has a magnitude  $p_1 = \gamma z_1$ . Likewise, points  $C$  and  $D$  are both at depth  $z_2$ ; hence,  $p_2 = \gamma z_2$ . In all cases, the pressure acts *normal* to the surface area  $dA$  located at the specified point.

Using Eq. 9-13 and the results of Sec. 9.4, it is possible to determine the resultant force caused by a liquid and specify its location on the surface of a submerged plate. Three different shapes of plates will now be considered.

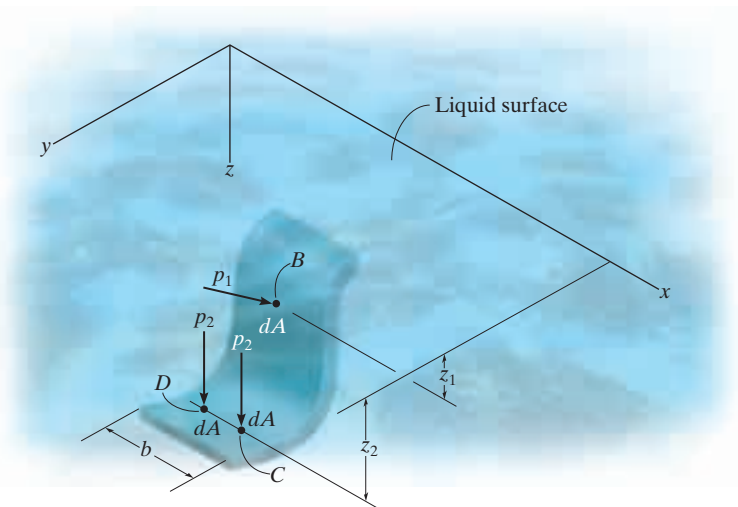


Fig. 9-24

\*In particular, for water  $\gamma = 62.4 \text{ lb/ft}^3$ , or  $\gamma = \rho g = 9810 \text{ N/m}^3$  since  $\rho = 1000 \text{ kg/m}^3$  and  $g = 9.81 \text{ m/s}^2$ .

**Flat Plate of Constant Width.** A flat rectangular plate of constant width, which is submerged in a liquid having a specific weight  $\gamma$ , is shown in Fig. 9–25a. Since pressure varies linearly with depth, Eq. 9–13, the distribution of pressure over the plate's surface is represented by a trapezoidal volume having an intensity of  $p_1 = \gamma z_1$  at depth  $z_1$  and  $p_2 = \gamma z_2$  at depth  $z_2$ . As noted in Sec. 9.4, the magnitude of the *resultant force*  $\mathbf{F}_R$  is equal to the *volume* of this loading diagram and  $\mathbf{F}_R$  has a *line of action* that passes through the volume's centroid  $C$ . Hence,  $\mathbf{F}_R$  does *not* act at the centroid of the plate; rather, it acts at point  $P$ , called the *center of pressure*.

Since the plate has a *constant width*, the loading distribution may also be viewed in two dimensions, Fig. 9–25b. Here the loading intensity is measured as force/length and varies linearly from  $w_1 = bp_1 = b\gamma z_1$  to  $w_2 = bp_2 = b\gamma z_2$ . The magnitude of  $\mathbf{F}_R$  in this case equals the trapezoidal *area*, and  $\mathbf{F}_R$  has a *line of action* that passes through the area's centroid  $C$ . For numerical applications, the area and location of the centroid for a trapezoid are tabulated on the inside back cover.



The walls of the tank must be designed to support the pressure loading of the liquid that is contained within it. (© Russell C. Hibbeler)

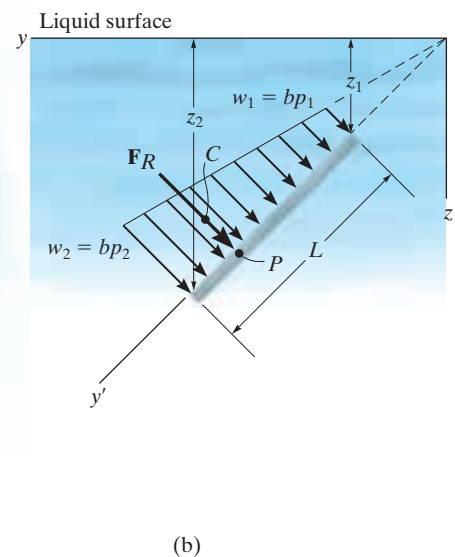
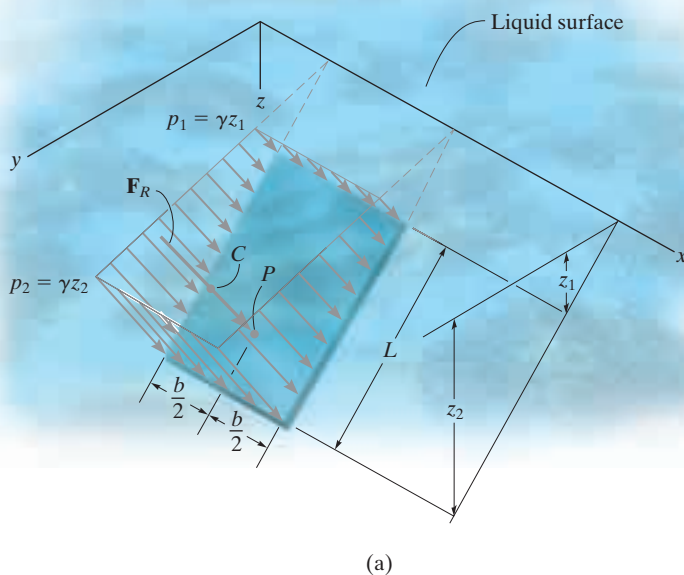


Fig. 9–25

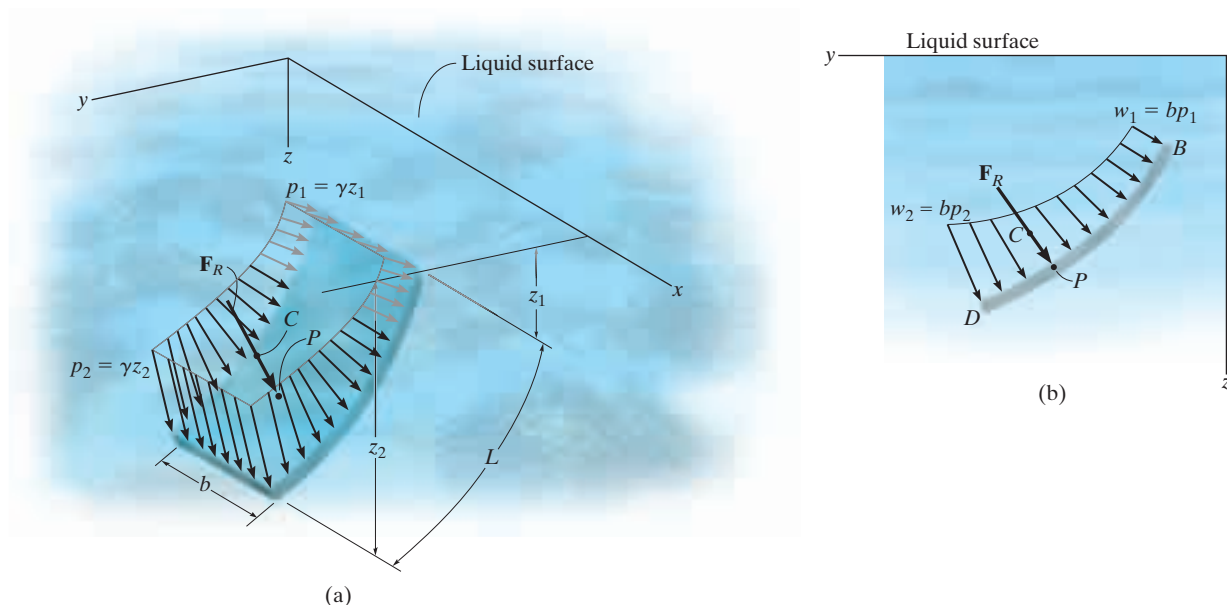
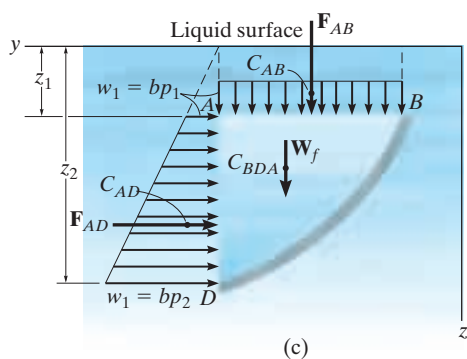


Fig. 9-26



**Curved Plate of Constant Width.** When a submerged plate of constant width is curved, the pressure acting normal to the plate continually changes both its magnitude and direction, and therefore calculation of the magnitude of  $\mathbf{F}_R$  and its location  $P$  is more difficult than for a flat plate. Three- and two-dimensional views of the loading distribution are shown in Figs. 9-26a and 9-26b, respectively. Although integration can be used to solve this problem, a simpler method exists. This method requires separate calculations for the horizontal and vertical components of  $\mathbf{F}_R$ .

For example, the distributed loading acting on the plate can be represented by the *equivalent loading* shown in Fig. 9-26c. Here the plate supports the weight of liquid  $W_f$  contained within the block  $BDA$ . This force has a magnitude  $W_f = (\gamma b)(\text{area}_{BDA})$  and acts through the centroid of  $BDA$ . In addition, there are the pressure distributions caused by the liquid acting along the vertical and horizontal sides of the block. Along the vertical side  $AD$ , the force  $\mathbf{F}_{AD}$  has a magnitude equal to the area of the trapezoid. It acts through the centroid  $C_{AD}$  of this area. The distributed loading along the horizontal side  $AB$  is *constant* since all points lying in this plane are at the same depth from the surface of the liquid. The magnitude of  $\mathbf{F}_{AB}$  is simply the area of the rectangle. This force acts through the centroid  $C_{AB}$  or at the midpoint of  $AB$ . Summing these three forces yields  $\mathbf{F}_R = \Sigma \mathbf{F} = \mathbf{F}_{AD} + \mathbf{F}_{AB} + \mathbf{W}_f$ . Finally, the location of the center of pressure  $P$  on the plate is determined by applying  $M_R = \Sigma M$ , which states that the moment of the resultant force about a convenient reference point such as  $D$  or  $B$ , in Fig. 9-26b, is equal to the sum of the moments of the three forces in Fig. 9-26c about this same point.

**Flat Plate of Variable Width.** The pressure distribution acting on the surface of a submerged plate having a variable width is shown in Fig. 9–27. If we consider the force  $d\mathbf{F}$  acting on the differential area strip  $dA$ , parallel to the  $x$  axis, then its magnitude is  $dF = p \, dA$ . Since the depth of  $dA$  is  $z$ , the pressure on the element is  $p = \gamma z$ . Therefore,  $dF = (\gamma z) dA$  and so the resultant force becomes

$$F_R = \int dF = \gamma \int z \, dA$$

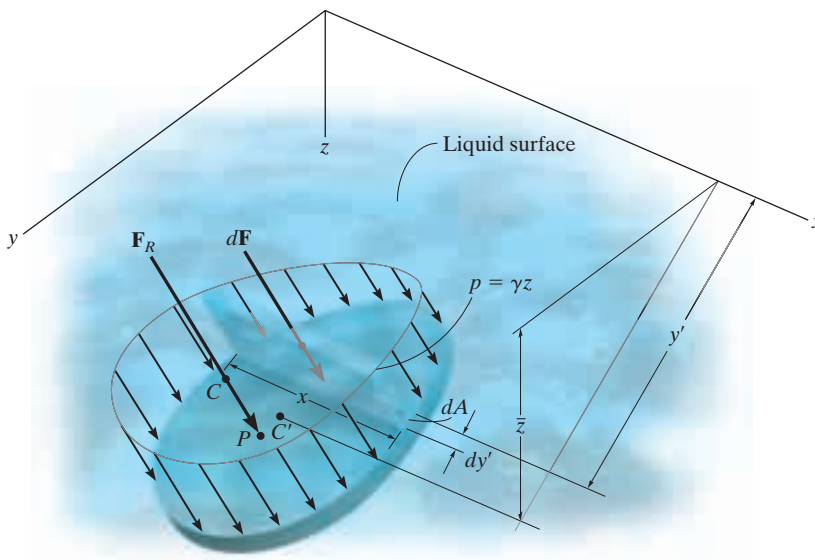
If the depth to the centroid  $C'$  of the area is  $\bar{z}$ , Fig. 9–27, then,  $\int z \, dA = \bar{z}A$ . Substituting, we have

$$F_R = \gamma \bar{z} A \quad (9-14)$$

In other words, *the magnitude of the resultant force acting on any flat plate is equal to the product of the area  $A$  of the plate and the pressure  $p = \gamma \bar{z}$  at the depth of the area's centroid  $C'$ .* As discussed in Sec. 9.4, this force is also equivalent to the volume under the pressure distribution. Realize that its line of action passes through the centroid  $C$  of this *volume* and intersects the plate at the center of pressure  $P$ , Fig. 9–27. Notice that the location of  $C'$  does not coincide with the location of  $P$ .

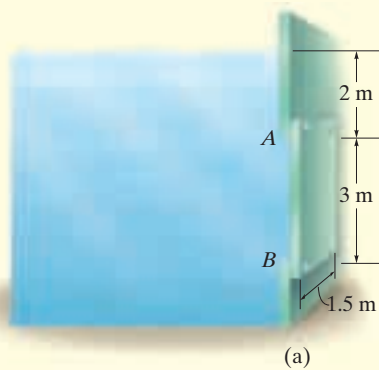


The resultant force of the water pressure and its location on the elliptical back plate of this tank truck must be determined by integration. (© Russell C. Hibbeler)



**Fig. 9–27**

## EXAMPLE 9.14



Determine the magnitude and location of the resultant hydrostatic force acting on the submerged rectangular plate  $AB$  shown in Fig. 9–28a. The plate has a width of 1.5 m;  $\rho_w = 1000 \text{ kg/m}^3$ .

## SOLUTION I

The water pressures at depths  $A$  and  $B$  are

$$p_A = \rho_w g z_A = (1000 \text{ kg/m}^3)(9.81 \text{ m/s}^2)(2 \text{ m}) = 19.62 \text{ kPa}$$

$$p_B = \rho_w g z_B = (1000 \text{ kg/m}^3)(9.81 \text{ m/s}^2)(5 \text{ m}) = 49.05 \text{ kPa}$$

Since the plate has a constant width, the pressure loading can be viewed in two dimensions, as shown in Fig. 9–28b. The intensities of the load at  $A$  and  $B$  are

$$w_A = bp_A = (1.5 \text{ m})(19.62 \text{ kPa}) = 29.43 \text{ kN/m}$$

$$w_B = bp_B = (1.5 \text{ m})(49.05 \text{ kPa}) = 73.58 \text{ kN/m}$$

From the table on the inside back cover, the magnitude of the resultant force  $\mathbf{F}_R$  created by this distributed load is

$$F_R = \text{area of a trapezoid} = \frac{1}{2}(3)(29.4 + 73.6) = 154.5 \text{ kN} \quad \text{Ans.}$$

This force acts through the centroid of this area,

$$h = \frac{1}{3} \left( \frac{2(29.43) + 73.58}{29.43 + 73.58} \right) (3) = 1.29 \text{ m} \quad \text{Ans.}$$

measured upward from  $B$ , Fig. 9–31b.

## SOLUTION II

The same results can be obtained by considering two components of  $\mathbf{F}_R$ , defined by the triangle and rectangle shown in Fig. 9–28c. Each force acts through its associated centroid and has a magnitude of

$$F_{Re} = (29.43 \text{ kN/m})(3 \text{ m}) = 88.3 \text{ kN}$$

$$F_t = \frac{1}{2}(44.15 \text{ kN/m})(3 \text{ m}) = 66.2 \text{ kN}$$

Hence,

$$F_R = F_{Re} + F_t = 88.3 + 66.2 = 154.5 \text{ kN} \quad \text{Ans.}$$

The location of  $\mathbf{F}_R$  is determined by summing moments about  $B$ , Figs. 9–28b and c, i.e.,

$$\zeta + (M_R)_B = \Sigma M_B; (154.5)h = 88.3(1.5) + 66.2(1)$$

$$h = 1.29 \text{ m} \quad \text{Ans.}$$

**NOTE:** Using Eq. 9–14, the resultant force can be calculated as  $F_R = \gamma \bar{z} A = (9810 \text{ N/m}^3)(3.5 \text{ m})(3 \text{ m})(1.5 \text{ m}) = 154.5 \text{ kN}$ .

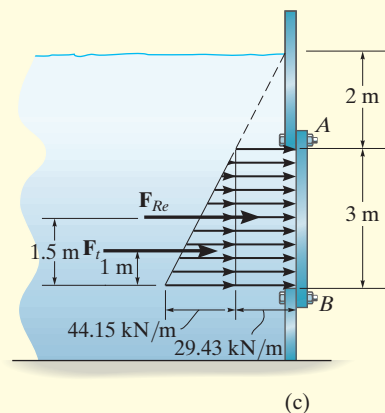
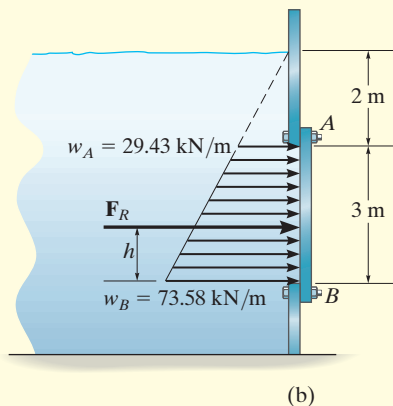
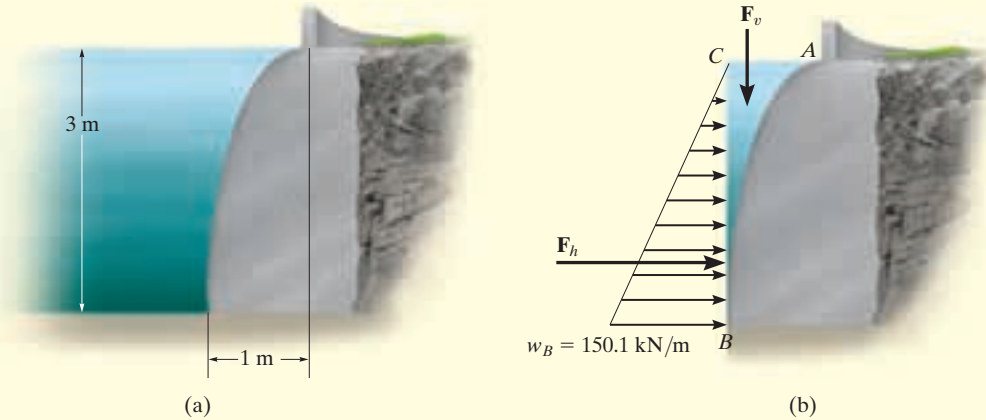


Fig. 9–28



**EXAMPLE 9.15**

Determine the magnitude of the resultant hydrostatic force acting on the surface of a seawall shaped in the form of a parabola, as shown in Fig. 9–29a. The wall is 5 m long;  $\rho_w = 1020 \text{ kg/m}^3$ .

**Fig. 9–29****SOLUTION**

The horizontal and vertical components of the resultant force will be calculated, Fig. 9–29b. Since

$$p_B = \rho_w g z_B = (1020 \text{ kg/m}^3)(9.81 \text{ m/s}^2)(3 \text{ m}) = 30.02 \text{ kPa}$$

then

$$w_B = b p_B = 5 \text{ m}(30.02 \text{ kPa}) = 150.1 \text{ kN/m}$$

Thus,

$$F_h = \frac{1}{2}(3 \text{ m})(150.1 \text{ kN/m}) = 225.1 \text{ kN}$$

The area of the parabolic section  $ABC$  can be determined using the formula for a parabolic area  $A = \frac{1}{3}ab$ . Hence, the weight of water within this 5-m-long region is

$$\begin{aligned} F_v &= (\rho_w g b)(\text{area}_{ABC}) \\ &= (1020 \text{ kg/m}^3)(9.81 \text{ m/s}^2)(5 \text{ m})\left[\frac{1}{3}(1 \text{ m})(3 \text{ m})\right] = 50.0 \text{ kN} \end{aligned}$$

The resultant force is therefore

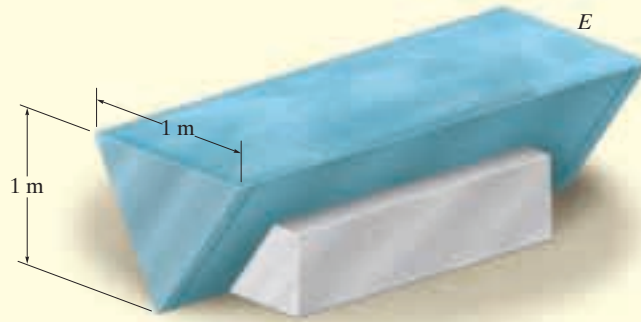
$$\begin{aligned} F_R &= \sqrt{F_h^2 + F_v^2} = \sqrt{(225.1 \text{ kN})^2 + (50.0 \text{ kN})^2} \\ &= 231 \text{ kN} \end{aligned}$$

*Ans.*

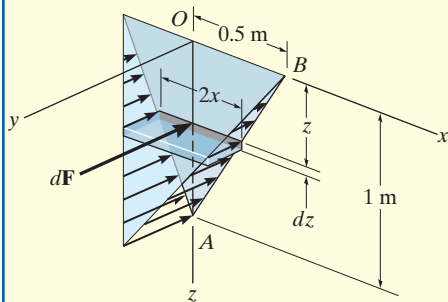


**EXAMPLE 9.16**

Determine the magnitude and location of the resultant force acting on the triangular end plates of the water trough shown in Fig. 9–30a;  $\rho_w = 1000 \text{ kg/m}^3$ .



(a)



(b)

**Fig. 9–30****SOLUTION**

The pressure distribution acting on the end plate *E* is shown in Fig. 9–30b. The magnitude of the resultant force is equal to the volume of this loading distribution. We will solve the problem by integration. Choosing the differential volume element shown in the figure, we have

$$dF = dV = p \, dA = \rho_w g z (2x \, dz) = 19\,620 z x \, dz$$

The equation of line *AB* is

$$x = 0.5(1 - z)$$

Hence, substituting and integrating with respect to *z* from *z* = 0 to *z* = 1 m yields

$$\begin{aligned} F = V &= \int_V dV = \int_0^{1 \text{ m}} (19\,620) z [0.5(1 - z)] \, dz \\ &= 9810 \int_0^{1 \text{ m}} (z - z^2) \, dz = 1635 \text{ N} = 1.64 \text{ kN} \quad \text{Ans.} \end{aligned}$$

This resultant passes through the *centroid of the volume*. Because of symmetry,

$$\bar{x} = 0 \quad \text{Ans.}$$

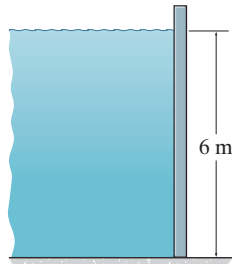
Since  $\tilde{z} = z$  for the volume element, then

$$\begin{aligned} \bar{z} &= \frac{\int_V \tilde{z} \, dV}{\int_V dV} = \frac{\int_0^{1 \text{ m}} z (19\,620) z [0.5(1 - z)] \, dz}{1635} = \frac{9810 \int_0^{1 \text{ m}} (z^2 - z^3) \, dz}{1635} \\ &= 0.5 \text{ m} \quad \text{Ans.} \end{aligned}$$

**NOTE:** We can also determine the resultant force by applying Eq. 9–14,  $F_R = \gamma \bar{z} A = (9810 \text{ N/m}^3) \left(\frac{1}{3}\right) (1 \text{ m}) \left[\frac{1}{2} (1 \text{ m}) (1 \text{ m})\right] = 1.64 \text{ kN}$ .

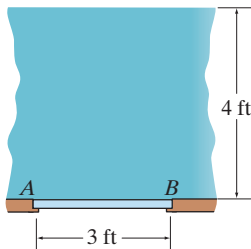
## FUNDAMENTAL PROBLEMS

**F9-17.** Determine the magnitude of the hydrostatic force acting per meter length of the wall. Water has a density of  $\rho = 1 \text{ Mg/m}^3$ .



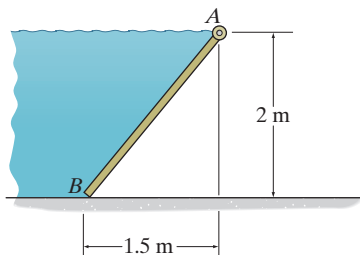
**Prob. F9-17**

**F9-18.** Determine the magnitude of the hydrostatic force acting on gate  $AB$ , which has a width of 4 ft. The specific weight of water is  $\gamma = 62.4 \text{ lb/ft}^3$ .



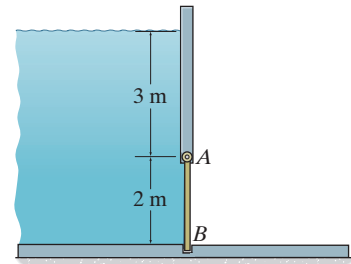
**Prob. F9-18**

**F9-19.** Determine the magnitude of the hydrostatic force acting on gate  $AB$ , which has a width of 1.5 m. Water has a density of  $\rho = 1 \text{ Mg/m}^3$ .



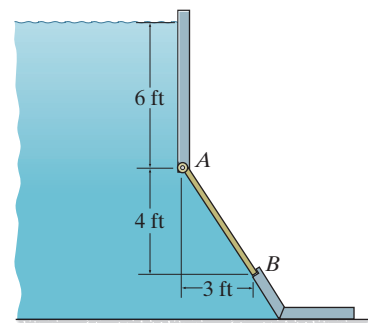
**Prob. F9-19**

**F9-20.** Determine the magnitude of the hydrostatic force acting on gate  $AB$ , which has a width of 2 m. Water has a density of  $\rho = 1 \text{ Mg/m}^3$ .



**Prob. F9-20**

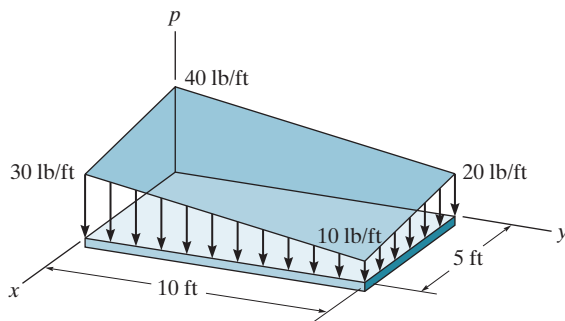
**F9-21.** Determine the magnitude of the hydrostatic force acting on gate  $AB$ , which has a width of 2 ft. The specific weight of water is  $\gamma = 62.4 \text{ lb/ft}^3$ .



**Prob. F9-21**

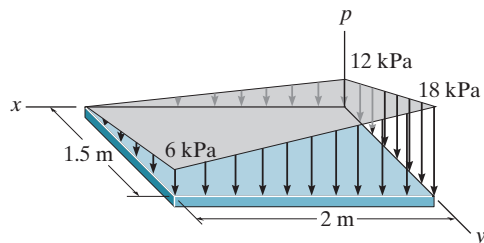
## PROBLEMS

**9-115.** The pressure loading on the plate varies uniformly along each of its edges. Determine the magnitude of the resultant force and the coordinates  $(\bar{x}, \bar{y})$  of the point where the line of action of the force intersects the plate. *Hint:* The equation defining the boundary of the load has the form  $p = ax + by + c$ , where the constants  $a$ ,  $b$ , and  $c$  have to be determined.



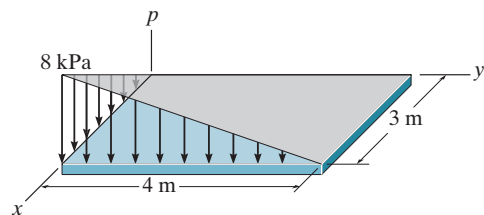
Prob. 9-115

**\*9-116.** The load over the plate varies linearly along the sides of the plate such that  $p = (12 - 6x + 4y)$  kPa. Determine the magnitude of the resultant force and the coordinates  $(\bar{x}, \bar{y})$  of the point where the line of action of the force intersects the plate.



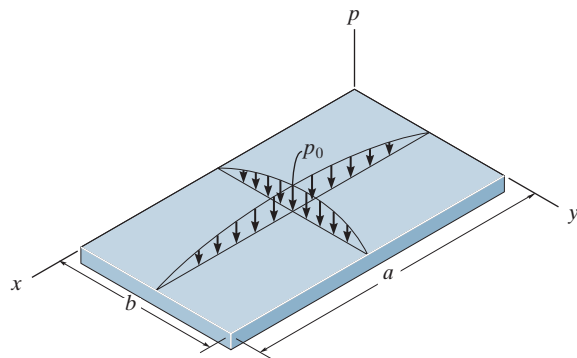
Prob. 9-116

**9-117.** The load over the plate varies linearly along the sides of the plate such that  $p = \frac{2}{3} [x(4 - y)]$  kPa. Determine the resultant force and its position  $(\bar{x}, \bar{y})$  on the plate.



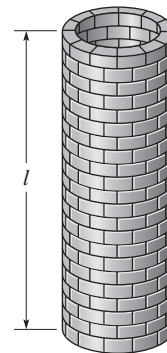
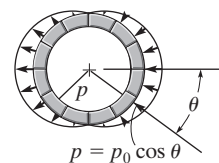
Prob. 9-117

**9-118.** The rectangular plate is subjected to a distributed load over its *entire surface*. The load is defined by the expression  $p = p_0 \sin(\pi x/a) \sin(\pi y/b)$ , where  $p_0$  represents the pressure acting at the center of the plate. Determine the magnitude and location of the resultant force acting on the plate.



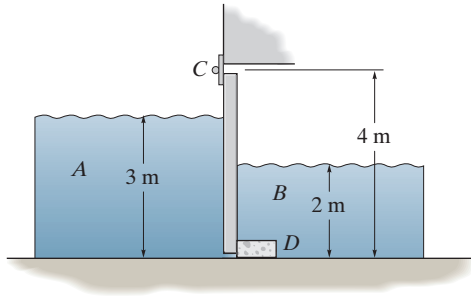
Prob. 9-118

**9-119.** A wind loading creates a positive pressure on one side of the chimney and a negative (suction) pressure on the other side, as shown. If this pressure loading acts uniformly along the chimney's length, determine the magnitude of the resultant force created by the wind.



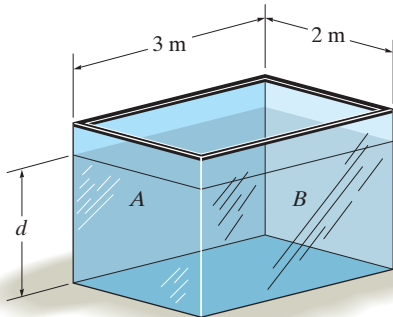
Prob. 9-119

**\*9-120.** When the tide water *A* subsides, the tide gate automatically swings open to drain the marsh *B*. For the condition of high tide shown, determine the horizontal reactions developed at the hinge *C* and stop block *D*. The length of the gate is 6 m and its height is 4 m.  $\rho_w = 1.0 \text{ Mg/m}^3$ .



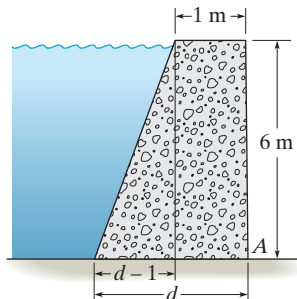
**Prob. 9-120**

**9-121.** The tank is filled with water to a depth of  $d = 4 \text{ m}$ . Determine the resultant force the water exerts on side *A* and side *B* of the tank. If oil instead of water is placed in the tank, to what depth  $d$  should it reach so that it creates the same resultant forces?  $\rho_o = 900 \text{ kg/m}^3$  and  $\rho_w = 1000 \text{ kg/m}^3$ .



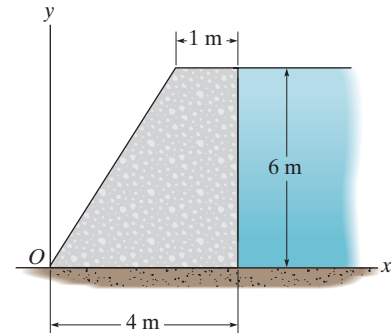
**Prob. 9-121**

**9-122.** The concrete “gravity” dam is held in place by its own weight. If the density of concrete is  $\rho_c = 2.5 \text{ Mg/m}^3$ , and water has a density of  $\rho_w = 1.0 \text{ Mg/m}^3$ , determine the smallest dimension  $d$  that will prevent the dam from overturning about its end *A*.



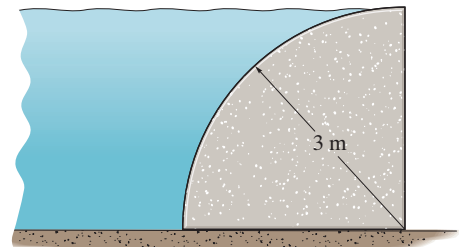
**Prob. 9-122**

**9-123.** The factor of safety for tipping of the concrete dam is defined as the ratio of the stabilizing moment due to the dam’s weight divided by the overturning moment about *O* due to the water pressure. Determine this factor if the concrete has a density of  $\rho_{\text{conc}} = 2.5 \text{ Mg/m}^3$  and for water  $\rho_w = 1 \text{ Mg/m}^3$ .



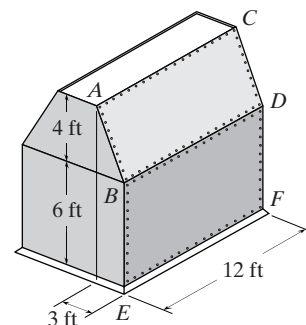
**Prob. 9-123**

**\*9-124.** The concrete dam in the shape of a quarter circle. Determine the magnitude of the resultant hydrostatic force that acts on the dam per meter of length. The density of water is  $\rho_w = 1 \text{ Mg/m}^3$ .



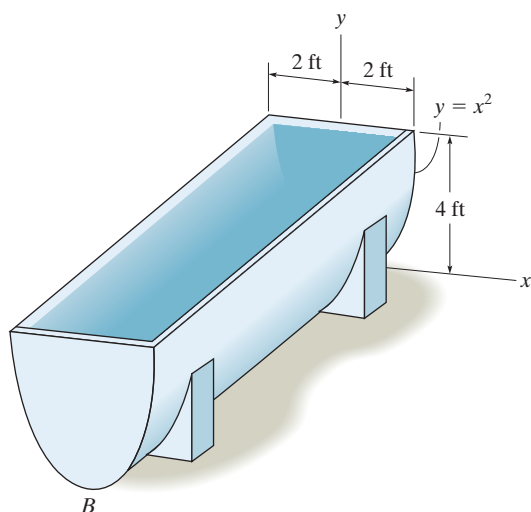
**Prob. 9-124**

**9-125.** The tank is used to store a liquid having a density of  $80 \text{ lb/ft}^3$ . If it is filled to the top, determine the magnitude of force the liquid exerts on each of its two sides *ABDC* and *BDFE*.



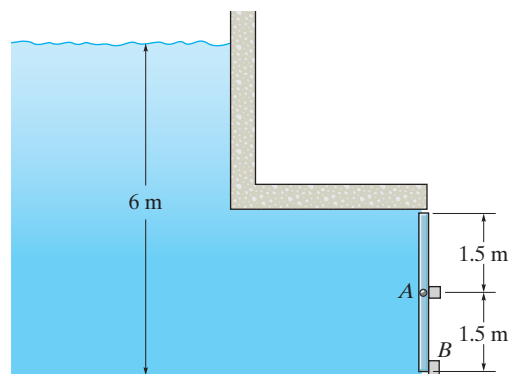
**Prob. 9-125**

**9-126.** The parabolic plate is subjected to a fluid pressure that varies linearly from 0 at its top to 100 lb/ft at its bottom  $B$ . Determine the magnitude of the resultant force and its location on the plate.



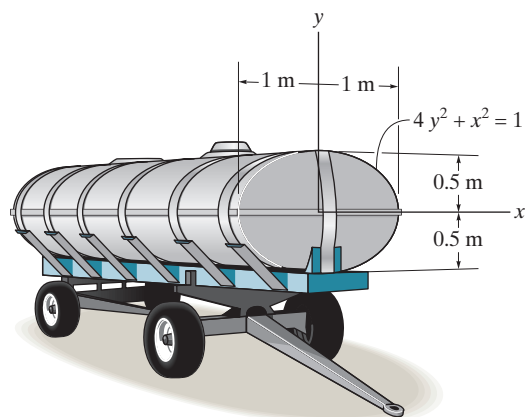
**Prob. 9-126**

**9-127.** The 2-m-wide rectangular gate is pinned at its center  $A$  and is prevented from rotating by the block at  $B$ . Determine the reactions at these supports due to hydrostatic pressure.  $\rho_w = 1.0 \text{ Mg/m}^3$ .



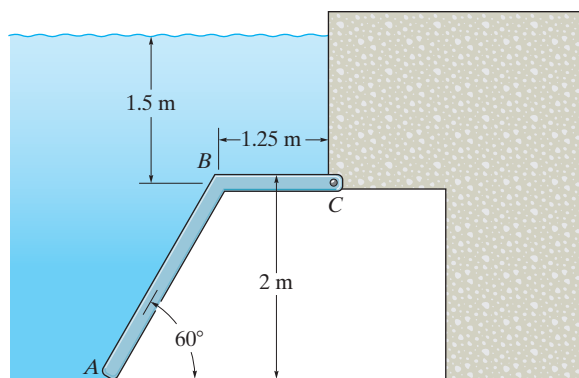
**Prob. 9-127**

**\*9-128.** The tank is filled with a liquid that has a density of  $900 \text{ kg/m}^3$ . Determine the resultant force that it exerts on the elliptical end plate, and the location of the center of pressure, measured from the  $x$  axis.



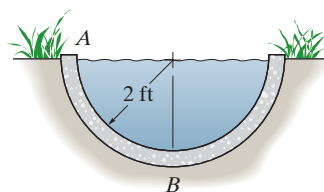
**Prob. 9-128**

**9-129.** Determine the magnitude of the resultant force acting on the gate  $ABC$  due to hydrostatic pressure. The gate has a width of 1.5 m.  $\rho_w = 1.0 \text{ Mg/m}^3$ .



**Prob. 9-129**

**9-130.** The semicircular drainage pipe is filled with water. Determine the resultant horizontal and vertical force components that the water exerts on the side  $AB$  of the pipe per foot of pipe length;  $\gamma_w = 62.4 \text{ lb/ft}^3$ .



**Prob. 9-130**

## CHAPTER REVIEW

### Center of Gravity and Centroid

The *center of gravity*  $G$  represents a point where the weight of the body can be considered concentrated. The distance from an axis to this point can be determined from a balance of moments, which requires that the moment of the weight of all the particles of the body about this axis must equal the moment of the entire weight of the body about the axis.

The center of mass will coincide with the center of gravity provided the acceleration of gravity is constant.

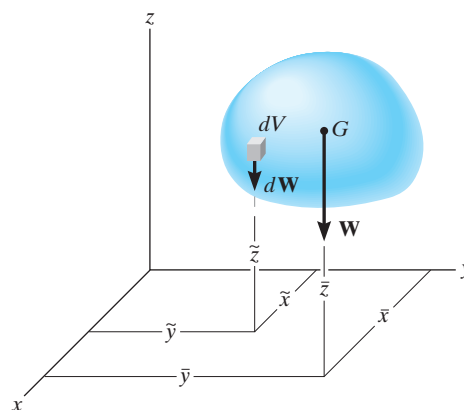
The *centroid* is the location of the geometric center for the body. It is determined in a similar manner, using a moment balance of geometric elements such as line, area, or volume segments. For bodies having a continuous shape, moments are summed (integrated) using differential elements.

The center of mass will coincide with the centroid provided the material is homogeneous, i.e., the density of the material is the same throughout. The centroid will always lie on an axis of symmetry.

$$\bar{x} = \frac{\int \tilde{x} dW}{\int dW}$$

$$\bar{y} = \frac{\int \tilde{y} dW}{\int dW}$$

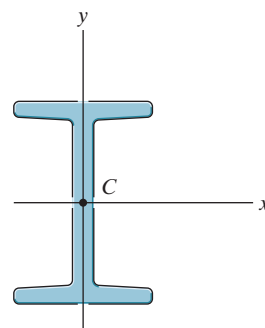
$$\bar{z} = \frac{\int \tilde{z} dW}{\int dW}$$



$$\bar{x} = \frac{\int_L \tilde{x} dL}{\int_L dL} \quad \bar{y} = \frac{\int_L \tilde{y} dL}{\int_L dL} \quad \bar{z} = \frac{\int_L \tilde{z} dL}{\int_L dL}$$

$$\bar{x} = \frac{\int_A \tilde{x} dA}{\int_A dA} \quad \bar{y} = \frac{\int_A \tilde{y} dA}{\int_A dA} \quad \bar{z} = \frac{\int_A \tilde{z} dA}{\int_A dA}$$

$$\bar{x} = \frac{\int_V \tilde{x} dV}{\int_V dV} \quad \bar{y} = \frac{\int_V \tilde{y} dV}{\int_V dV} \quad \bar{z} = \frac{\int_V \tilde{z} dV}{\int_V dV}$$



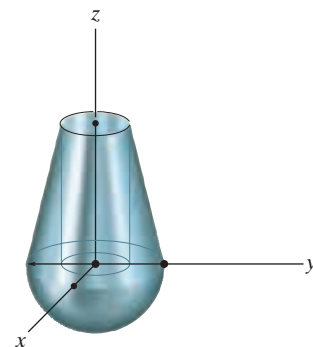
**Composite Body**

If the body is a composite of several shapes, each having a known location for its center of gravity or centroid, then the location of the center of gravity or centroid of the body can be determined from a discrete summation using its composite parts.

$$\bar{x} = \frac{\sum \tilde{x}W}{\sum W}$$

$$\bar{y} = \frac{\sum \tilde{y}W}{\sum W}$$

$$\bar{z} = \frac{\sum \tilde{z}W}{\sum W}$$

**Theorems of Pappus and Guldinus**

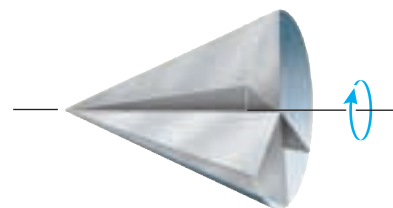
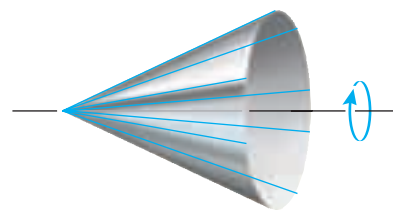
The theorems of Pappus and Guldinus can be used to determine the surface area and volume of a body of revolution.

The *surface area* equals the product of the length of the generating curve and the distance traveled by the centroid of the curve needed to generate the area.

$$A = \theta \bar{r} L$$

The *volume* of the body equals the product of the generating area and the distance traveled by the centroid of this area needed to generate the volume.

$$V = \theta \bar{r} A$$



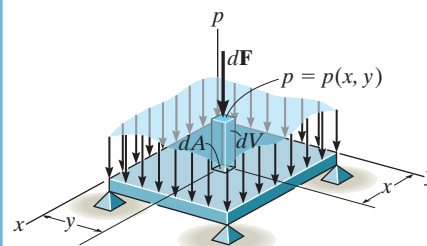
### General Distributed Loading

The magnitude of the resultant force is equal to the total volume under the distributed-loading diagram. The line of action of the resultant force passes through the geometric center or centroid of this volume.

$$F_R = \int_A p(x, y) dA = \int_V dV$$

$$\bar{x} = \frac{\int_V x dV}{\int_V dV}$$

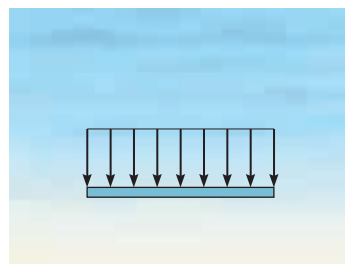
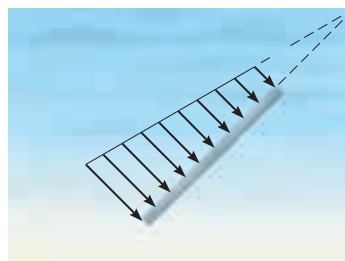
$$\bar{y} = \frac{\int_V y dV}{\int_V dV}$$



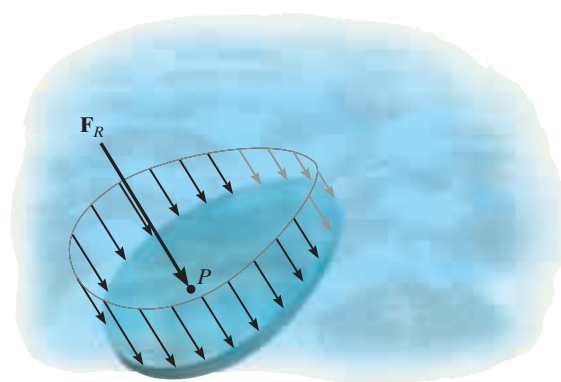
### Fluid Pressure

The pressure developed by a liquid at a point on a submerged surface depends upon the depth of the point and the density of the liquid in accordance with Pascal's law,  $p = \rho gh = \gamma h$ . This pressure will create a *linear distribution* of loading on a flat vertical or inclined surface.

If the surface is horizontal, then the loading will be *uniform*.



In any case, the resultants of these loadings can be determined by finding the volume under the loading curve or using  $F_R = \gamma \bar{z} A$ , where  $\bar{z}$  is the depth to the centroid of the plate's area. The line of action of the resultant force passes through the centroid of the volume of the loading diagram and acts at a point  $P$  on the plate called the center of pressure.

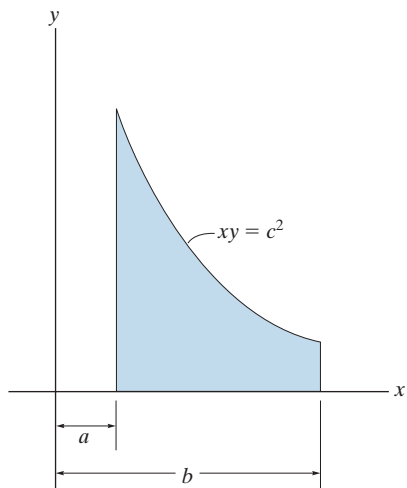




## REVIEW PROBLEMS

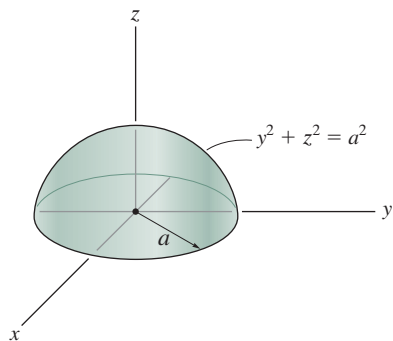
**R9-1.** Locate the centroid  $\bar{x}$  of the area.

**R9-2.** Locate the centroid  $\bar{y}$  of the area.



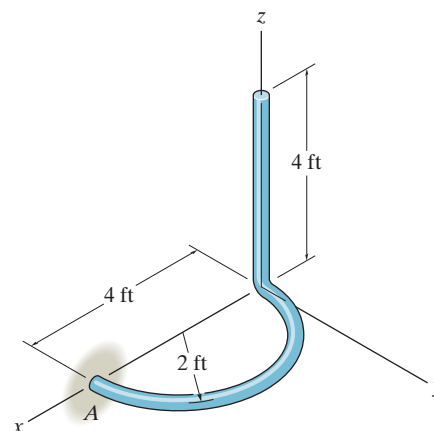
**Probs. R9-1/2**

**R9-3.** Locate the centroid  $\bar{z}$  of the hemisphere.



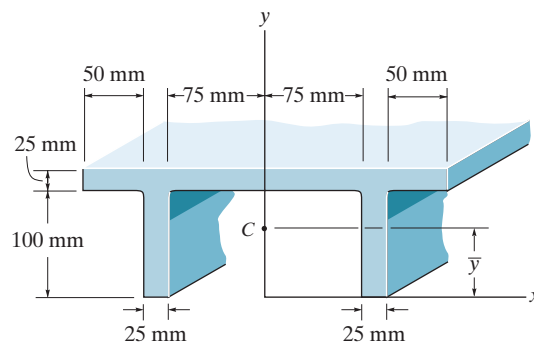
**Prob. R9-3**

**R9-4.** Locate the centroid of the rod.



**Prob. R9-4**

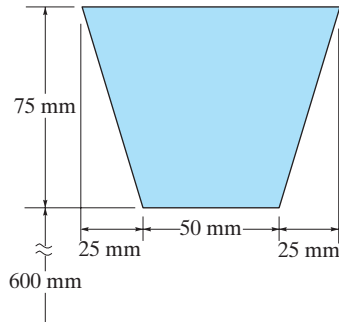
**R9-5.** Locate the centroid  $\bar{y}$  of the beam's cross-sectional area.



**Prob. R9-5**

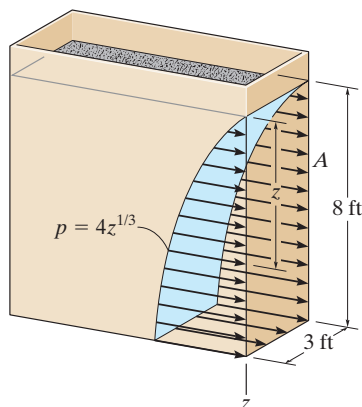
**R9-6.** A circular V-belt has an inner radius of 600 mm and a cross-sectional area as shown. Determine the surface area of the belt.

**R9-7.** A circular V-belt has an inner radius of 600 mm and a cross-sectional area as shown. Determine the volume of material required to make the belt.



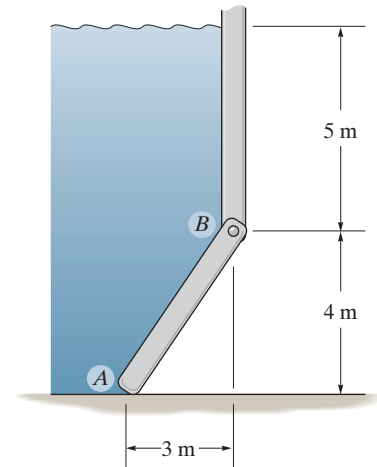
**Probs. R9-6/7**

**R9-8.** The rectangular bin is filled with coal, which creates a pressure distribution along wall  $A$  that varies as shown, i.e.,  $p = 4z^{1/3}$  lb/ft<sup>2</sup>, where  $z$  is in feet. Determine the resultant force created by the coal, and its location, measured from the top surface of the coal.



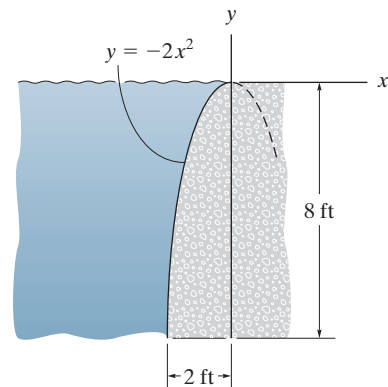
**Prob. R9-8**

**R9-9.** The gate  $AB$  is 8 m wide. Determine the horizontal and vertical components of force acting on the pin at  $B$  and the vertical reaction at the smooth support  $A$ ;  $\rho_w = 1.0 \text{ Mg/m}^3$ .



**Prob. R9-9**

**R9-10.** Determine the magnitude of the resultant hydrostatic force acting per foot of length on the seawall;  $\gamma_w = 62.4 \text{ lb/ft}^3$ .



**Prob. R9-10**

# Chapter 10



(© Michael N. Paras/AGE Fotostock/Alamy)

The design of these structural members requires calculation of their cross-sectional moment of inertia. In this chapter we will discuss how this is done.

# Moments of Inertia

## CHAPTER OBJECTIVES

- To develop a method for determining the moment of inertia for an area.
- To introduce the product of inertia and show how to determine the maximum and minimum moments of inertia for an area.
- To discuss the mass moment of inertia.

## 10.1 Definition of Moments of Inertia for Areas

Whenever a distributed load acts perpendicular to an area and its intensity varies linearly, the calculation of the moment of the loading about an axis will involve an integral of the form  $\int y^2 dA$ . For example, consider the plate in Fig. 10–1, which is submerged in a fluid and subjected to the pressure  $p$ . As discussed in Sec. 9.5, this pressure varies linearly with depth, such that  $p = \gamma y$ , where  $\gamma$  is the specific weight of the fluid. Thus, the force acting on the differential area  $dA$  of the plate is  $dF = p dA = (\gamma y)dA$ . The *moment* of this force about the  $x$  axis is therefore  $dM = y dF = \gamma y^2 dA$ , and so integrating  $dM$  over the entire area of the plate yields  $M = \gamma \int y^2 dA$ . The integral  $\int y^2 dA$  is sometimes referred to as the “second moment” of the area about an axis (the  $x$  axis), but more often it is called the **moment of inertia of the area**. The word “inertia” is used here since the formulation is similar to the mass moment of inertia,  $\int y^2 dm$ , which is a dynamical property described in Sec. 10.8. Although for an area this integral has no physical meaning, it often arises in formulas used in fluid mechanics, mechanics of materials, structural mechanics, and mechanical design, and so the engineer needs to be familiar with the methods used to determine the moment of inertia.

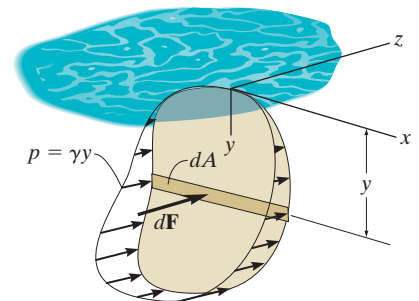


Fig. 10–1

**Moment of Inertia.** By definition, the moments of inertia of a differential area  $dA$  about the  $x$  and  $y$  axes are  $dI_x = y^2 dA$  and  $dI_y = x^2 dA$ , respectively, Fig. 10-2. For the entire area  $A$  the **moments of inertia** are determined by integration; i.e.,

$$\begin{aligned} I_x &= \int_A y^2 dA \\ I_y &= \int_A x^2 dA \end{aligned} \quad (10-1)$$

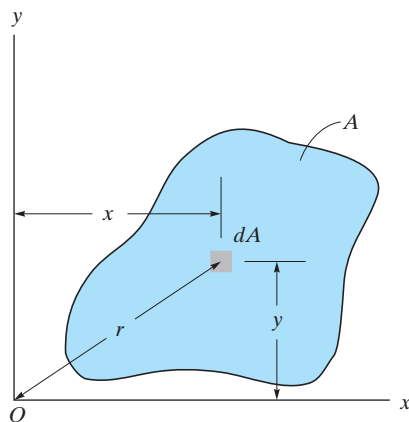


Fig. 10-2

We can also formulate this quantity for  $dA$  about the “pole”  $O$  or  $z$  axis, Fig. 10-2. This is referred to as the **polar moment of inertia**. It is defined as  $dJ_O = r^2 dA$ , where  $r$  is the perpendicular distance from the pole ( $z$  axis) to the element  $dA$ . For the entire area the **polar moment of inertia** is

$$J_O = \int_A r^2 dA = I_x + I_y \quad (10-2)$$

This relation between  $J_O$  and  $I_x$ ,  $I_y$  is possible since  $r^2 = x^2 + y^2$ , Fig. 10-2.

From the above formulations it is seen that  $I_x$ ,  $I_y$ , and  $J_O$  will *always* be *positive* since they involve the product of distance squared and area. Furthermore, the units for moment of inertia involve length raised to the fourth power, e.g.,  $\text{m}^4$ ,  $\text{mm}^4$ , or  $\text{ft}^4$ ,  $\text{in}^4$ .

## 10.2 Parallel-Axis Theorem for an Area

The **parallel-axis theorem** can be used to find the moment of inertia of an area about *any* axis that is parallel to an axis passing through the centroid and about which the moment of inertia is known. To develop this theorem, we will consider finding the moment of inertia of the shaded area shown in Fig. 10-3 about the  $x$  axis. To start, we choose a differential element  $dA$  located at an arbitrary distance  $y'$  from the *centroidal*  $x'$  axis. If the distance between the parallel  $x$  and  $x'$  axis is  $d_y$ , then the moment of inertia of  $dA$  about the  $x$  axis is  $dI_x = (y' + d_y)^2 dA$ . For the entire area,

$$\begin{aligned} I_x &= \int_A (y' + d_y)^2 dA \\ &= \int_A y'^2 dA + 2d_y \int_A y' dA + d_y^2 \int_A dA \end{aligned}$$

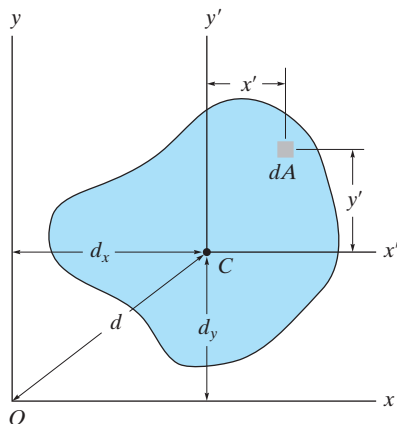


Fig. 10-3

The first integral represents the moment of inertia of the area about the centroidal axis,  $\bar{I}_{x'}$ . The second integral is zero since the  $x'$  axis passes through the area's centroid  $C$ ; i.e.,  $\int y' dA = \bar{y}' \int dA = 0$  since  $\bar{y}' = 0$ . Since the third integral represents the total area  $A$ , the final result is therefore

$$I_x = \bar{I}_{x'} + Ad_y^2 \quad (10-3)$$

A similar expression can be written for  $I_y$ ; i.e.,

$$I_y = \bar{I}_{y'} + Ad_x^2 \quad (10-4)$$

And finally, for the polar moment of inertia, since  $\bar{J}_C = \bar{I}_{x'} + \bar{I}_{y'}$  and  $d^2 = d_x^2 + d_y^2$ , we have

$$J_O = \bar{J}_C + Ad^2 \quad (10-5)$$

The form of each of these three equations states that *the moment of inertia for an area about an axis is equal to its moment of inertia about a parallel axis passing through the area's centroid plus the product of the area and the square of the perpendicular distance between the axes.*



In order to predict the strength and deflection of this beam, it is necessary to calculate the moment of inertia of the beam's cross-sectional area. (© Russell C. Hibbeler)

## 10.3 Radius of Gyration of an Area

The **radius of gyration** of an area about an axis has units of length and is a quantity that is often used for the design of columns in structural mechanics. Provided the areas and moments of inertia are *known*, the radii of gyration are determined from the formulas

$$\begin{aligned} k_x &= \sqrt{\frac{I_x}{A}} \\ k_y &= \sqrt{\frac{I_y}{A}} \\ k_O &= \sqrt{\frac{J_O}{A}} \end{aligned} \quad (10-6)$$

The form of these equations is easily remembered since it is similar to that for finding the moment of inertia for a differential area about an axis. For example,  $I_x = k_x^2 A$ ; whereas for a differential area,  $dI_x = y^2 dA$ .

## Important Points

- The moment of inertia is a geometric property of an area that is used to determine the strength of a structural member or the location of a resultant pressure force acting on a plate submerged in a fluid. It is sometimes referred to as the second moment of the area about an axis, because the distance from the axis to each area element is squared.
- If the moment of inertia of an area is known about its centroidal axis, then the moment of inertia about a corresponding parallel axis can be determined using the parallel-axis theorem.

## Procedure for Analysis

In most cases the moment of inertia can be determined using a single integration. The following procedure shows two ways in which this can be done.

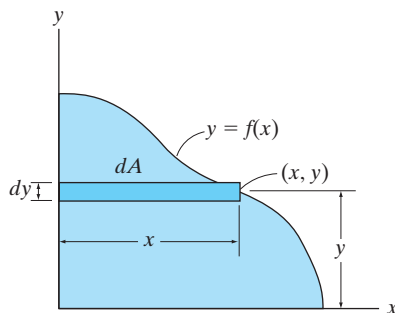
- If the curve defining the boundary of the area is expressed as  $y = f(x)$ , then select a rectangular differential element such that it has a finite length and differential width.
- The element should be located so that it intersects the curve at the *arbitrary point*  $(x, y)$ .

### Case 1.

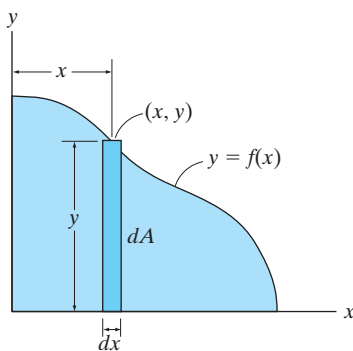
- Orient the element so that its length is *parallel* to the axis about which the moment of inertia is computed. This situation occurs when the rectangular element shown in Fig. 10–4a is used to determine  $I_x$  for the area. Here the entire element is at a distance  $y$  from the  $x$  axis since it has a thickness  $dy$ . Thus  $I_x = \int y^2 dA$ . To find  $I_y$ , the element is oriented as shown in Fig. 10–4b. This element lies at the *same* distance  $x$  from the  $y$  axis so that  $I_y = \int x^2 dA$ .

### Case 2.

- The length of the element can be oriented *perpendicular* to the axis about which the moment of inertia is computed; however, Eq. 10–1 *does not apply* since all points on the element will *not* lie at the same moment-arm distance from the axis. For example, if the rectangular element in Fig. 10–4a is used to determine  $I_y$ , it will first be necessary to calculate the moment of inertia of the *element* about an axis parallel to the  $y$  axis that passes through the element's centroid, and then determine the moment of inertia of the *element* about the  $y$  axis using the parallel-axis theorem. Integration of this result will yield  $I_y$ . See Examples 10.2 and 10.3.



(a)



(b)

Fig. 10–4

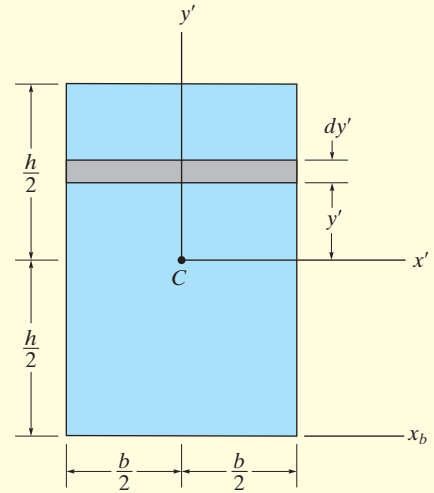
**EXAMPLE 10.1**

Determine the moment of inertia for the rectangular area shown in Fig. 10–5 with respect to (a) the centroidal  $x'$  axis, (b) the axis  $x_b$  passing through the base of the rectangle, and (c) the pole or  $z'$  axis perpendicular to the  $x'-y'$  plane and passing through the centroid  $C$ .

**SOLUTION (CASE 1)**

**Part (a).** The differential element shown in Fig. 10–5 is chosen for integration. Because of its location and orientation, the *entire element* is at a distance  $y'$  from the  $x'$  axis. Here it is necessary to integrate from  $y' = -h/2$  to  $y' = h/2$ . Since  $dA = b \, dy'$ , then

$$\begin{aligned}\bar{I}_{x'} &= \int_A y'^2 \, dA = \int_{-h/2}^{h/2} y'^2 (b \, dy') = b \int_{-h/2}^{h/2} y'^2 \, dy' \\ \bar{I}_{x'} &= \frac{1}{12} b h^3\end{aligned}\quad \text{Ans.}$$

**Fig. 10–5**

**Part (b).** The moment of inertia about an axis passing through the base of the rectangle can be obtained by using the above result of part (a) and applying the parallel-axis theorem, Eq. 10–3.

$$\begin{aligned}I_{x_b} &= \bar{I}_{x'} + A d_y^2 \\ &= \frac{1}{12} b h^3 + b h \left( \frac{h}{2} \right)^2 = \frac{1}{3} b h^3\end{aligned}\quad \text{Ans.}$$

**Part (c).** To obtain the polar moment of inertia about point  $C$ , we must first obtain  $\bar{I}_{y'}$ , which may be found by interchanging the dimensions  $b$  and  $h$  in the result of part (a), i.e.,

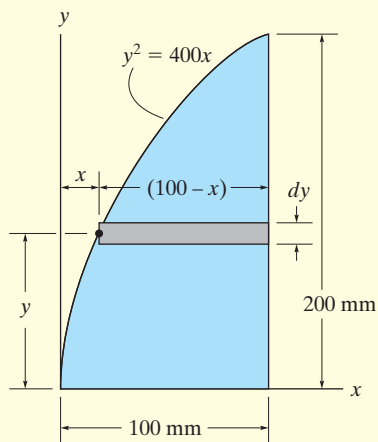
$$\bar{I}_{y'} = \frac{1}{12} h b^3$$

Using Eq. 10–2, the polar moment of inertia about  $C$  is therefore

$$\bar{J}_C = \bar{I}_{x'} + \bar{I}_{y'} = \frac{1}{12} b h (h^2 + b^2) \quad \text{Ans.}$$



## EXAMPLE 10.2



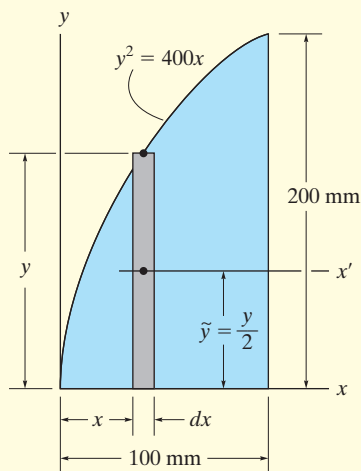
(a)

Determine the moment of inertia for the shaded area shown in Fig. 10–6a about the  $x$  axis.

## SOLUTION I (CASE 1)

A differential element of area that is *parallel* to the  $x$  axis, as shown in Fig. 10–6a, is chosen for integration. Since this element has a thickness  $dy$  and intersects the curve at the *arbitrary point*  $(x, y)$ , its area is  $dA = (100 - x) dy$ . Furthermore, the element lies at the same distance  $y$  from the  $x$  axis. Hence, integrating with respect to  $y$ , from  $y = 0$  to  $y = 200$  mm, yields

$$\begin{aligned} I_x &= \int_A y^2 dA = \int_0^{200 \text{ mm}} y^2 (100 - x) dy \\ &= \int_0^{200 \text{ mm}} y^2 \left( 100 - \frac{y^2}{400} \right) dy = \int_0^{200 \text{ mm}} \left( 100y^2 - \frac{y^4}{400} \right) dy \\ &= 107(10^6) \text{ mm}^4 \end{aligned} \quad \text{Ans.}$$



(b)

## SOLUTION II (CASE 2)

A differential element *parallel* to the  $y$  axis, as shown in Fig. 10–6b, is chosen for integration. It intersects the curve at the *arbitrary point*  $(x, y)$ . In this case, all points of the element do *not* lie at the same distance from the  $x$  axis, and therefore the parallel-axis theorem must be used to determine the *moment of inertia of the element* with respect to this axis. For a rectangle having a base  $b$  and height  $h$ , the moment of inertia about its centroidal axis has been determined in part (a) of Example 10.1. There it was found that  $\bar{I}_{x'} = \frac{1}{12}bh^3$ . For the differential element shown in Fig. 10–6b,  $b = dx$  and  $h = y$ , and thus  $d\bar{I}_{x'} = \frac{1}{12}dx y^3$ . Since the centroid of the element is  $\tilde{y} = y/2$  from the  $x$  axis, the moment of inertia of the element about this axis is

$$dI_x = d\bar{I}_{x'} + dA \tilde{y}^2 = \frac{1}{12}dx y^3 + y dx \left( \frac{y}{2} \right)^2 = \frac{1}{3}y^3 dx$$

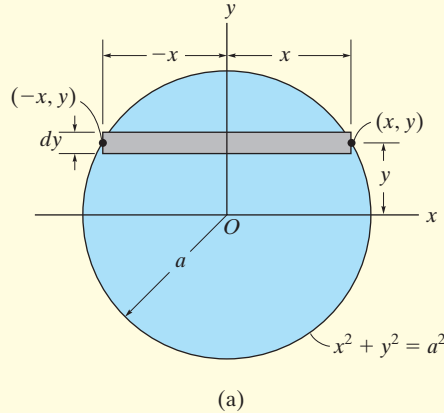
(This result can also be concluded from part (b) of Example 10.1.) Integrating with respect to  $x$ , from  $x = 0$  to  $x = 100$  mm, yields

$$\begin{aligned} I_x &= \int dI_x = \int_0^{100 \text{ mm}} \frac{1}{3}y^3 dx = \int_0^{100 \text{ mm}} \frac{1}{3}(400x)^{3/2} dx \\ &= 107(10^6) \text{ mm}^4 \end{aligned} \quad \text{Ans.}$$

Fig. 10–6

**EXAMPLE 10.3**

Determine the moment of inertia with respect to the  $x$  axis for the circular area shown in Fig. 10-7a.

**SOLUTION I (CASE 1)**

Using the differential element shown in Fig. 10-7a, since  $dA = 2x \, dy$ , we have

$$\begin{aligned} I_x &= \int_A y^2 \, dA = \int_A y^2 (2x) \, dy \\ &= \int_{-a}^a y^2 (2\sqrt{a^2 - y^2}) \, dy = \frac{\pi a^4}{4} \end{aligned} \quad \text{Ans.}$$

**SOLUTION II (CASE 2)**

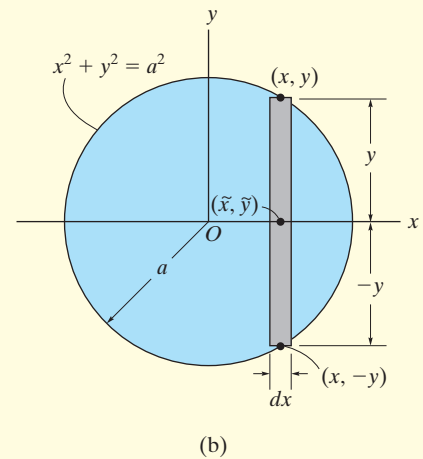
When the differential element shown in Fig. 10-7b is chosen, the centroid for the element happens to lie on the  $x$  axis, and since  $\bar{I}_{x'} = \frac{1}{12}bh^3$  for a rectangle, we have

$$\begin{aligned} dI_x &= \frac{1}{12} dx (2y)^3 \\ &= \frac{2}{3} y^3 \, dx \end{aligned}$$

Integrating with respect to  $x$  yields

$$I_x = \int_{-a}^a \frac{2}{3} (a^2 - x^2)^{3/2} \, dx = \frac{\pi a^4}{4} \quad \text{Ans.}$$

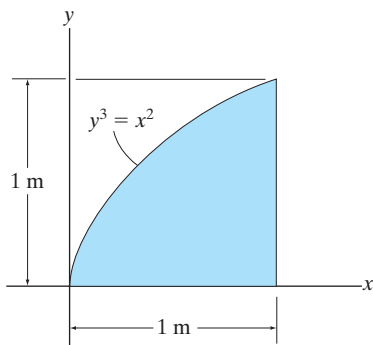
**NOTE:** By comparison, Solution I requires much less computation. Therefore, if an integral using a particular element appears difficult to evaluate, try solving the problem using an element oriented in the other direction.



**Fig. 10-7**

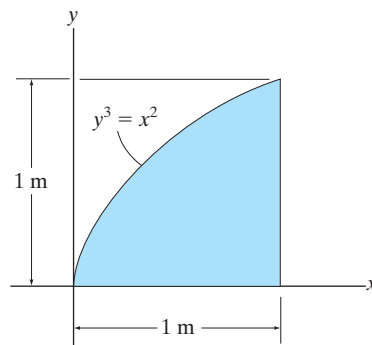
## FUNDAMENTAL PROBLEMS

**F10-1.** Determine the moment of inertia of the shaded area about the  $x$  axis.



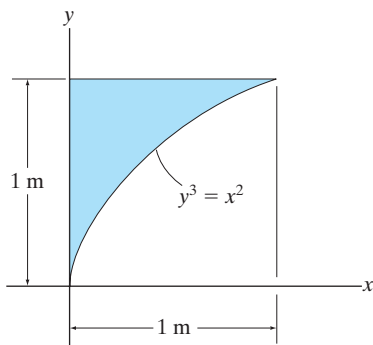
**Prob. F10-1**

**F10-3.** Determine the moment of inertia of the shaded area about the  $y$  axis.



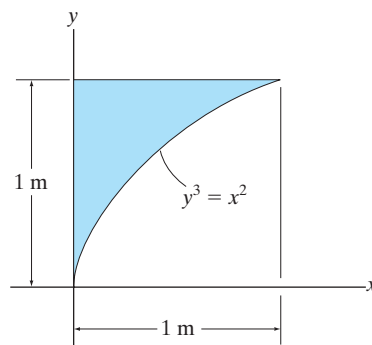
**Prob. F10-3**

**F10-2.** Determine the moment of inertia of the shaded area about the  $x$  axis.



**Prob. F10-2**

**F10-4.** Determine the moment of inertia of the shaded area about the  $y$  axis.

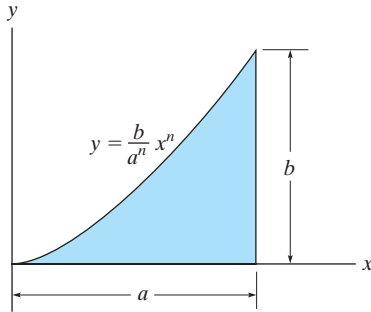


**Prob. F10-4**

## PROBLEMS

**10-1.** Determine the moment of inertia about the  $x$  axis.

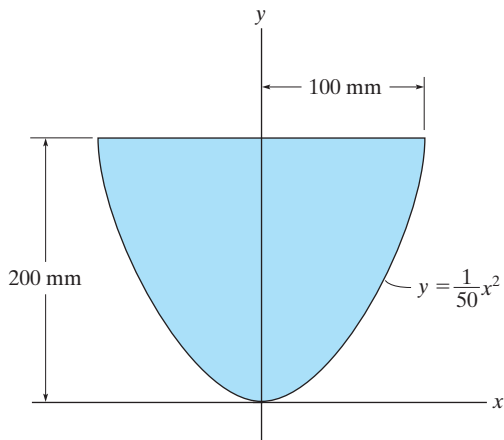
**10-2.** Determine the moment of inertia about the  $y$  axis.



**Probs. 10-1/2**

**10-3.** Determine the moment of inertia for the shaded area about the  $x$  axis.

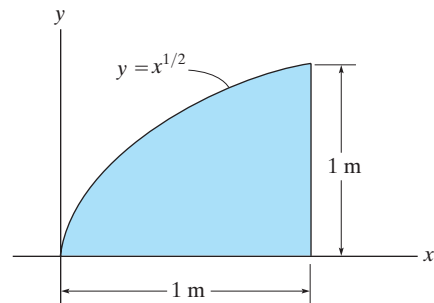
**\*10-4.** Determine the moment of inertia for the shaded area about the  $y$  axis.



**Probs. 10-3/4**

**10-5.** Determine the moment of inertia for the shaded area about the  $x$  axis.

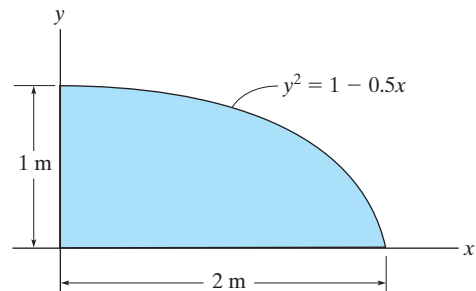
**10-6.** Determine the moment of inertia for the shaded area about the  $y$  axis.



**Probs. 10-5/6**

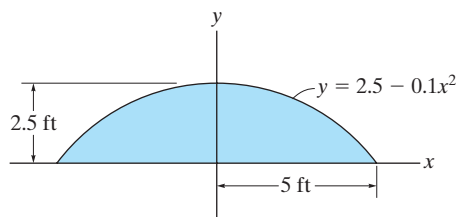
**10-7.** Determine the moment of inertia for the shaded area about the  $x$  axis.

**\*10-8.** Determine the moment of inertia for the shaded area about the  $y$  axis.



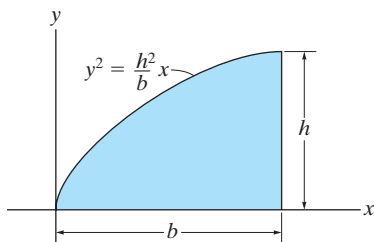
**Probs. 10-7/8**

**10–9.** Determine the moment of inertia of the area about the  $x$  axis. Solve the problem in two ways, using rectangular differential elements: (a) having a thickness  $dx$  and (b) having a thickness of  $dy$ .



**Prob. 10–9**

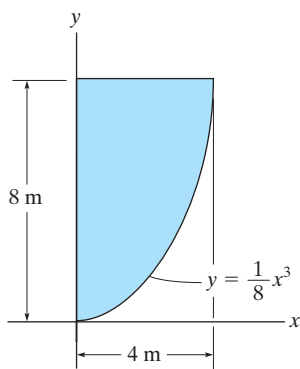
**10–10.** Determine the moment of inertia of the area about the  $x$  axis.



**Prob. 10–10**

**10–11.** Determine the moment of inertia for the shaded area about the  $x$  axis.

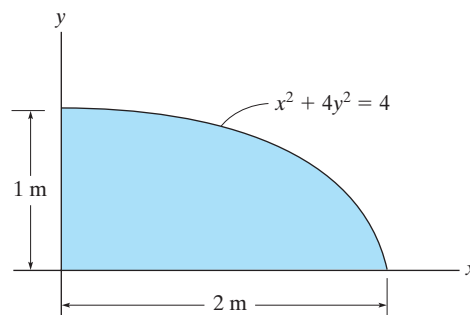
**\*10–12.** Determine the moment of inertia for the shaded area about the  $y$  axis.



**Probs. 10–11/12**

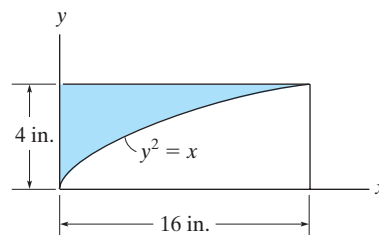
**10–13.** Determine the moment of inertia about the  $x$  axis.

**10–14.** Determine the moment of inertia about the  $y$  axis.



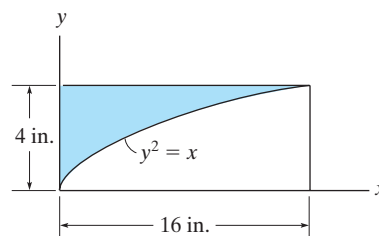
**Probs. 10–13/14**

**10–15.** Determine the moment of inertia for the shaded area about the  $x$  axis.



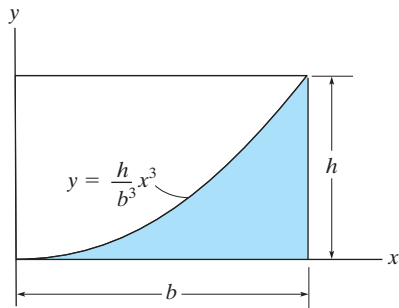
**Prob. 10–15**

**\*10–16.** Determine the moment of inertia for the shaded area about the  $y$  axis.



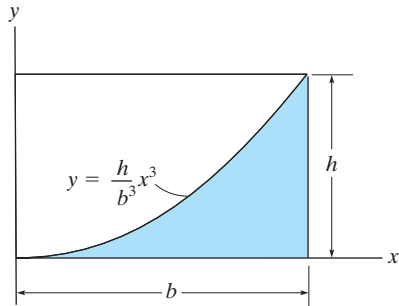
**Prob. 10–16**

**10–17.** Determine the moment of inertia for the shaded area about the  $x$  axis.



**Prob. 10–17**

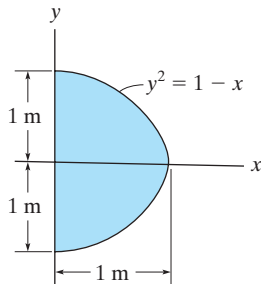
**10–18.** Determine the moment of inertia for the shaded area about the  $y$  axis.



**Prob. 10–18**

**10–19.** Determine the moment of inertia for the shaded area about the  $x$  axis.

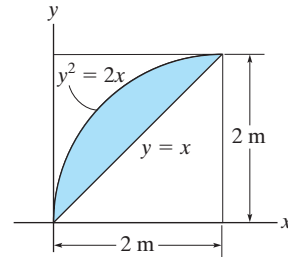
**\*10–20.** Determine the moment of inertia for the shaded area about the  $y$  axis.



**Probs. 10–19/20**

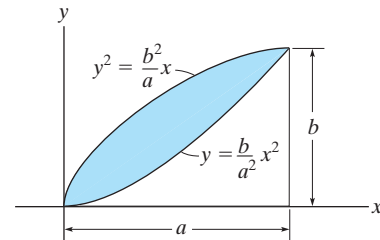
**10–21.** Determine the moment of inertia for the shaded area about the  $x$  axis.

**10–22.** Determine the moment of inertia for the shaded area about the  $y$  axis.



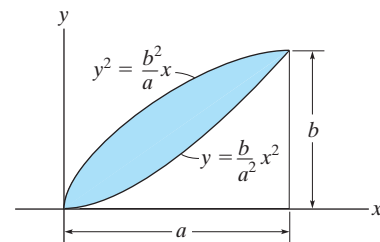
**Probs. 10–21/22**

**10–23.** Determine the moment of inertia for the shaded area about the  $x$  axis.



**Prob. 10–23**

**\*10–24.** Determine the moment of inertia for the shaded area about the  $y$  axis.



**Prob. 10–24**

## 10.4 Moments of Inertia for Composite Areas

A composite area consists of a series of connected “simpler” parts or shapes, such as rectangles, triangles, and circles. Provided the moment of inertia of each of these parts is known or can be determined about a common axis, then the moment of inertia for the composite area about this axis equals the *algebraic sum* of the moments of inertia of all its parts.

### Procedure for Analysis

The moment of inertia for a composite area about a reference axis can be determined using the following procedure.

#### Composite Parts.

- Using a sketch, divide the area into its composite parts and indicate the perpendicular distance from the centroid of each part to the reference axis.

#### Parallel-Axis Theorem.

- If the centroidal axis for each part does not coincide with the reference axis, the parallel-axis theorem,  $I = \bar{I} + Ad^2$ , should be used to determine the moment of inertia of the part about the reference axis. For the calculation of  $\bar{I}$ , use the table on the inside back cover.

#### Summation.

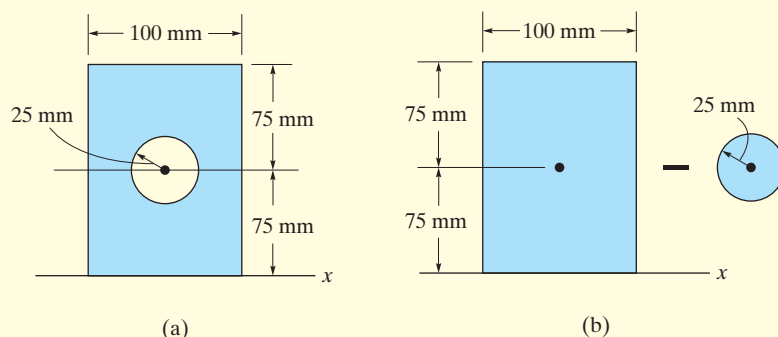
- The moment of inertia of the entire area about the reference axis is determined by summing the results of its composite parts about this axis.
- If a composite part has an empty region (hole), its moment of inertia is found by subtracting the moment of inertia of this region from the moment of inertia of the entire part including the region.

For design or analysis of this T-beam, engineers must be able to locate the centroid of its cross-sectional area, and then find the moment of inertia of this area about the centroidal axis.  
(© Russell C. Hibbeler)



**EXAMPLE 10.4**

Determine the moment of inertia of the area shown in Fig. 10–8a about the  $x$  axis.

**Fig. 10–8****SOLUTION**

**Composite Parts.** The area can be obtained by *subtracting* the circle from the rectangle shown in Fig. 10–8b. The centroid of each area is located in the figure.

**Parallel-Axis Theorem.** The moments of inertia about the  $x$  axis are determined using the parallel-axis theorem and the geometric properties formulae for circular and rectangular areas  $I_x = \frac{1}{4}\pi r^4$ ;  $I_x = \frac{1}{12}bh^3$ , found on the inside back cover.

*Circle*

$$\begin{aligned} I_x &= \bar{I}_{x'} + Ad_y^2 \\ &= \frac{1}{4}\pi(25)^4 + \pi(25)^2(75)^2 = 11.4(10^6) \text{ mm}^4 \end{aligned}$$

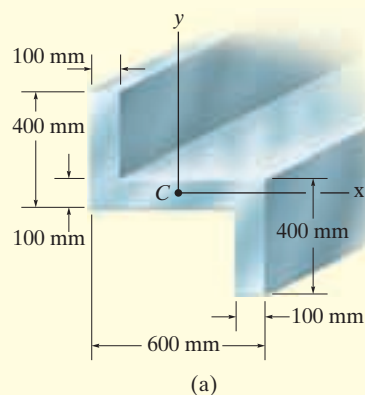
*Rectangle*

$$\begin{aligned} I_x &= \bar{I}_{x'} + Ad_y^2 \\ &= \frac{1}{12}(100)(150)^3 + (100)(150)(75)^2 = 112.5(10^6) \text{ mm}^4 \end{aligned}$$

**Summation.** The moment of inertia for the area is therefore

$$\begin{aligned} I_x &= -11.4(10^6) + 112.5(10^6) \\ &= 101(10^6) \text{ mm}^4 \end{aligned} \quad \text{Ans.}$$



**EXAMPLE 10.5**

Determine the moments of inertia for the cross-sectional area of the member shown in Fig. 10–9a about the  $x$  and  $y$  centroidal axes.

**SOLUTION**

**Composite Parts.** The cross section can be subdivided into the three rectangular areas  $A$ ,  $B$ , and  $D$  shown in Fig. 10–9b. For the calculation, the centroid of each of these rectangles is located in the figure.

**Parallel-Axis Theorem.** From the table on the inside back cover, or Example 10.1, the moment of inertia of a rectangle about its centroidal axis is  $\bar{I} = \frac{1}{12}bh^3$ . Hence, using the parallel-axis theorem for rectangles  $A$  and  $D$ , the calculations are as follows:

*Rectangles A and D*

$$I_x = \bar{I}_{x'} + A d_y^2 = \frac{1}{12}(100)(300)^3 + (100)(300)(200)^2 = 1.425(10^9) \text{ mm}^4$$

$$I_y = \bar{I}_{y'} + A d_x^2 = \frac{1}{12}(300)(100)^3 + (100)(300)(250)^2 = 1.90(10^9) \text{ mm}^4$$

*Rectangle B*

$$I_x = \frac{1}{12}(600)(100)^3 = 0.05(10^9) \text{ mm}^4$$

$$I_y = \frac{1}{12}(100)(600)^3 = 1.80(10^9) \text{ mm}^4$$

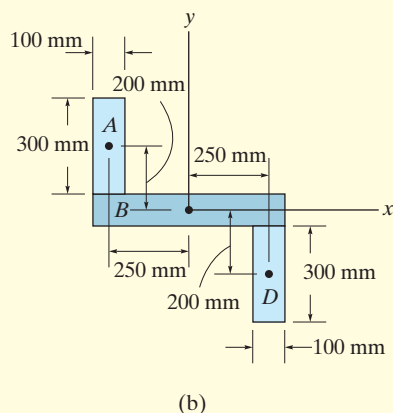
**Summation.** The moments of inertia for the entire cross section are thus

$$I_x = 2[1.425(10^9)] + 0.05(10^9) = 2.90(10^9) \text{ mm}^4$$

*Ans.*

$$I_y = 2[1.90(10^9)] + 1.80(10^9) = 5.60(10^9) \text{ mm}^4$$

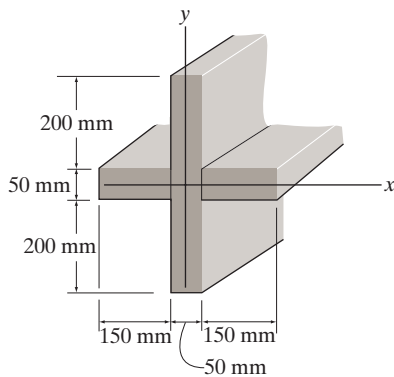
*Ans.*



**Fig. 10–9**

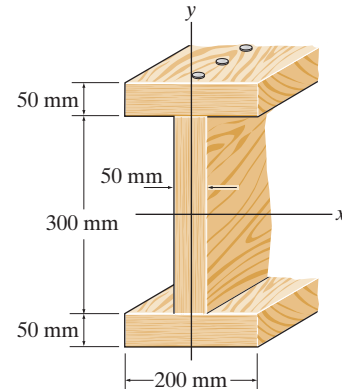
## FUNDAMENTAL PROBLEMS

**F10–5.** Determine the moment of inertia of the beam's cross-sectional area about the centroidal  $x$  and  $y$  axes.



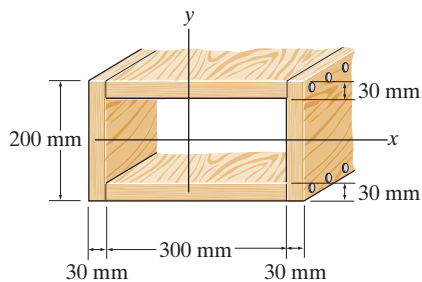
**Prob. F10–5**

**F10–7.** Determine the moment of inertia of the cross-sectional area of the channel with respect to the  $y$  axis.



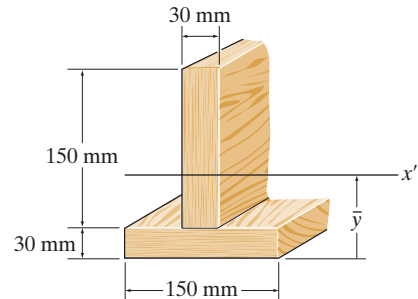
**Prob. F10–7**

**F10–6.** Determine the moment of inertia of the beam's cross-sectional area about the centroidal  $x$  and  $y$  axes.



**Prob. F10–6**

**F10–8.** Determine the moment of inertia of the cross-sectional area of the T-beam with respect to the  $x'$  axis passing through the centroid of the cross section.

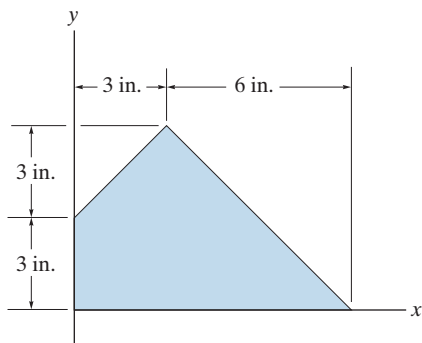


**Prob. F10–8**

## PROBLEMS

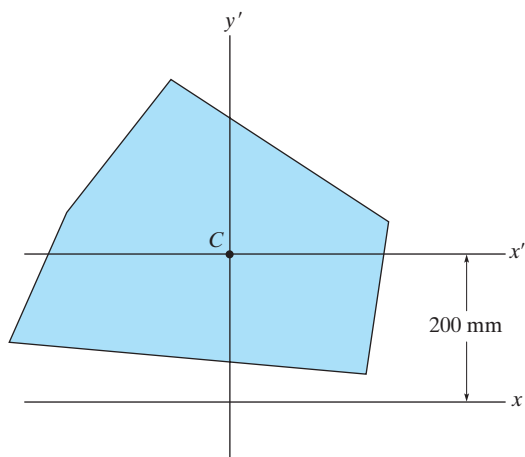
**10–25.** Determine the moment of inertia of the composite area about the  $x$  axis.

**10–26.** Determine the moment of inertia of the composite area about the  $y$  axis.



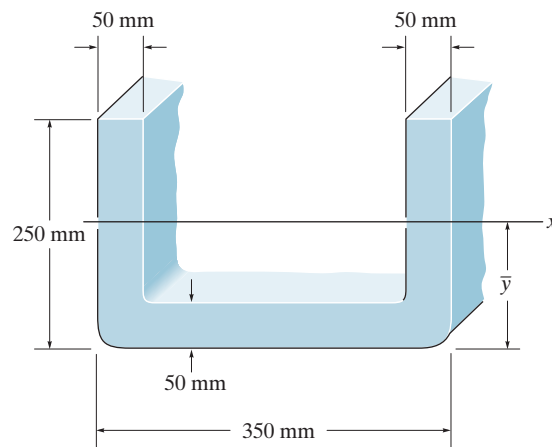
**Probs. 10–25/26**

**10–27.** The polar moment of inertia for the area is  $J_C = 642 (10^6) \text{ mm}^4$ , about the  $z'$  axis passing through the centroid  $C$ . The moment of inertia about the  $y'$  axis is  $264 (10^6) \text{ mm}^4$ , and the moment of inertia about the  $x$  axis is  $938 (10^6) \text{ mm}^4$ . Determine the area  $A$ .



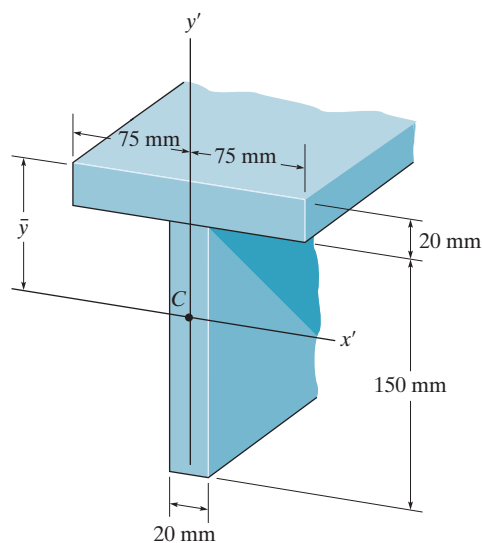
**Prob. 10–27**

**\*10–28.** Determine the location  $\bar{y}$  of the centroid of the channel's cross-sectional area and then calculate the moment of inertia of the area about this axis.



**Prob. 10–28**

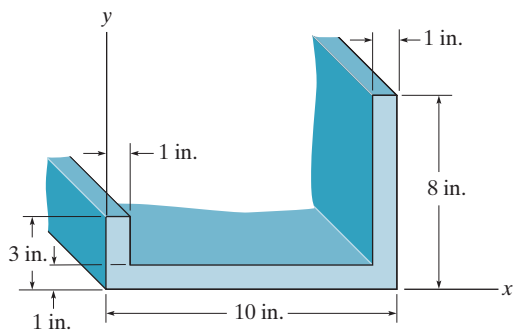
**10–29.** Determine  $\bar{y}$ , which locates the centroidal axis  $x'$  for the cross-sectional area of the T-beam, and then find the moments of inertia  $I_{x'}$  and  $I_{y'}$ .



**Prob. 10–29**

**10–30.** Determine the moment of inertia for the beam's cross-sectional area about the  $x$  axis.

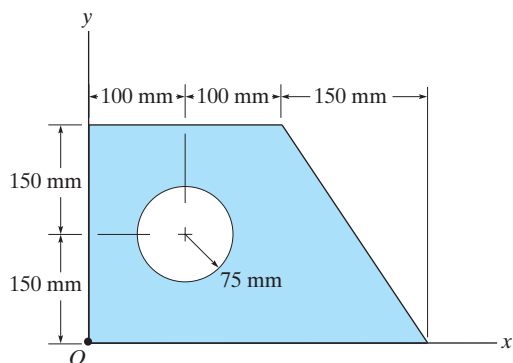
**10–31.** Determine the moment of inertia for the beam's cross-sectional area about the  $y$  axis.



**Probs. 10–30/31**

**\*10–32.** Determine the moment of inertia  $I_x$  of the shaded area about the  $x$  axis.

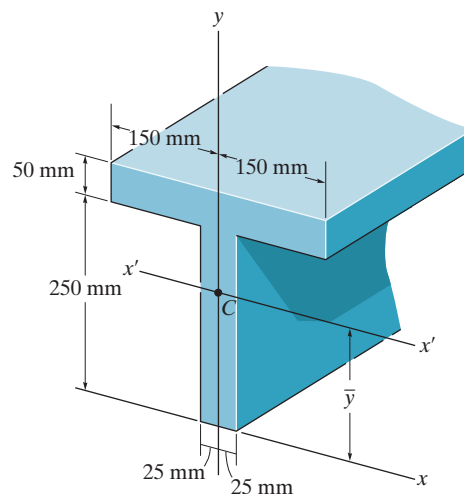
**10–33.** Determine the moment of inertia  $I_x$  of the shaded area about the  $y$  axis.



**Probs. 10–32/33**

**10–34.** Determine the moment of inertia of the beam's cross-sectional area about the  $y$  axis.

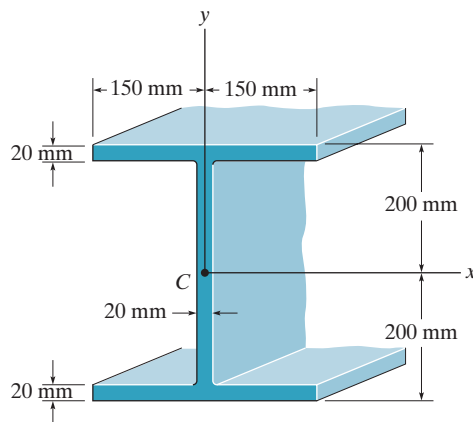
**10–35.** Determine  $\bar{y}$ , which locates the centroidal axis  $x'$  for the cross-sectional area of the T-beam, and then find the moment of inertia about the  $x'$  axis.



**Probs. 10–34/35**

**\*10–36.** Determine the moment of inertia about the  $x$  axis.

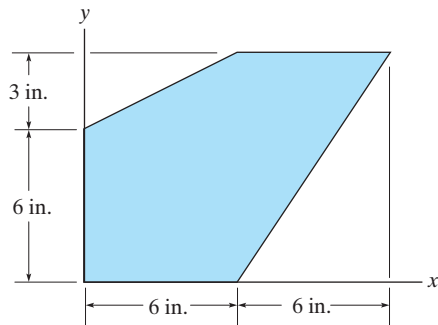
**10–37.** Determine the moment of inertia about the  $y$  axis.



**Probs. 10–36/37**

**10–38.** Determine the moment of inertia of the shaded area about the  $x$  axis.

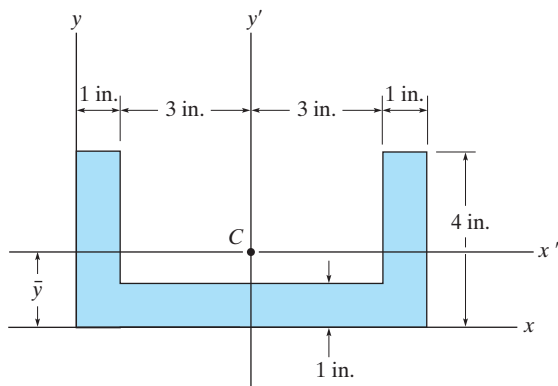
**10–39.** Determine the moment of inertia of the shaded area about the  $y$  axis.



**Probs. 10–38/39**

**\*10–40.** Determine the distance  $\bar{y}$  to the centroid of the beam's cross-sectional area; then find the moment of inertia about the centroidal  $x'$  axis.

**10–41.** Determine the moment of inertia for the beam's cross-sectional area about the  $y$  axis.



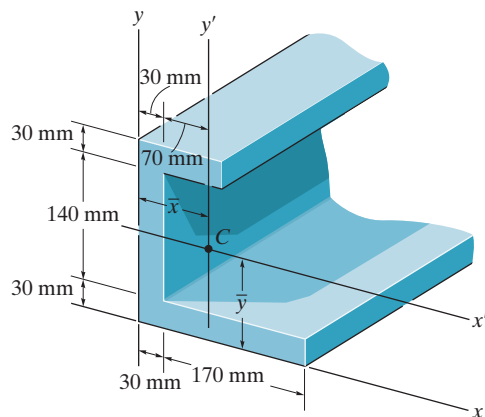
**Probs. 10–40/41**

**10–42.** Determine the moment of inertia of the beam's cross-sectional area about the  $x$  axis.

**10–43.** Determine the moment of inertia of the beam's cross-sectional area about the  $y$  axis.

**\*10–44.** Determine the distance  $\bar{y}$  to the centroid  $C$  of the beam's cross-sectional area and then compute the moment of inertia  $I_{x'}$  about the  $x'$  axis.

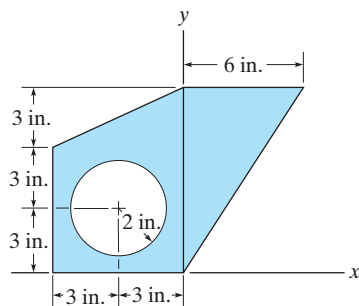
**10–45.** Determine the distance  $\bar{x}$  to the centroid  $C$  of the beam's cross-sectional area and then compute the moment of inertia  $I_{y'}$  about the  $y'$  axis.



**Probs. 10–42/43/44/45**

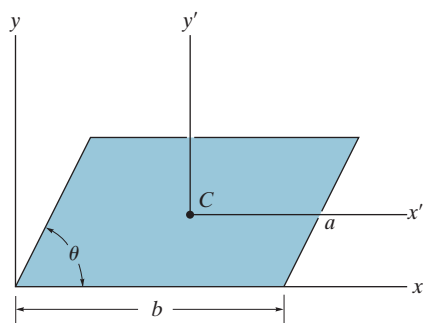
**10–46.** Determine the moment of inertia for the shaded area about the  $x$  axis.

**10–47.** Determine the moment of inertia for the shaded area about the  $y$  axis.



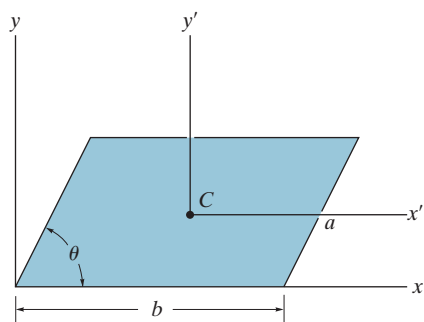
**Probs. 10–46/47**

**\*10–48.** Determine the moment of inertia of the parallelogram about the  $x'$  axis, which passes through the centroid  $C$  of the area.



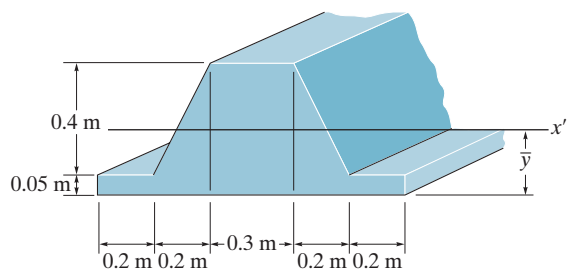
**Prob. 10–48**

**10–49.** Determine the moment of inertia of the parallelogram about the  $y'$  axis, which passes through the centroid  $C$  of the area.



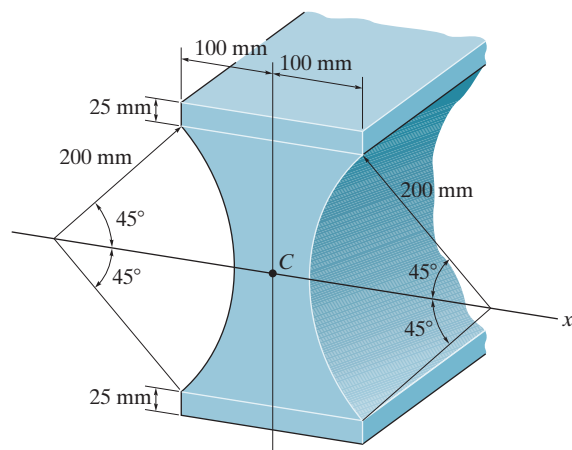
**Prob. 10–49**

**10–50.** Locate the centroid  $\bar{y}$  of the cross section and determine the moment of inertia of the section about the  $x'$  axis.



**Prob. 10–50**

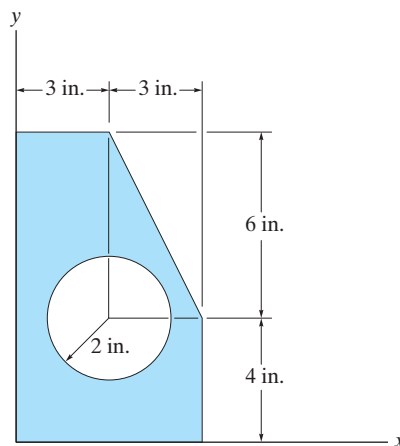
**10–51.** Determine the moment of inertia for the beam's cross-sectional area about the  $x'$  axis passing through the centroid  $C$  of the cross section.



**Prob. 10–51**

**\*10–52.** Determine the moment of inertia of the area about the  $x$  axis.

**10–53.** Determine the moment of inertia of the area about the  $y$  axis.



**Probs. 10–52/53**

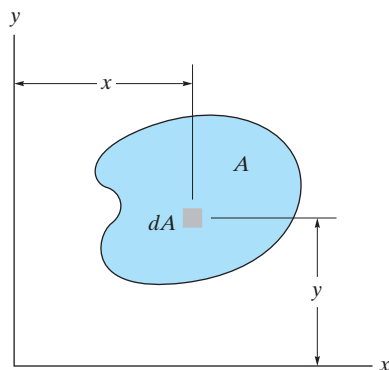


Fig. 10–10



The effectiveness of this beam to resist bending can be determined once its moments of inertia and its product of inertia are known. (© Russell C. Hibbeler)

## \*10.5 Product of Inertia for an Area

It will be shown in the next section that the property of an area, called the product of inertia, is required in order to determine the *maximum* and *minimum* moments of inertia for the area. These maximum and minimum values are important properties needed for designing structural and mechanical members such as beams, columns, and shafts.

The **product of inertia** of the area in Fig. 10–10 with respect to the  $x$  and  $y$  axes is defined as

$$I_{xy} = \int_A xy \, dA \quad (10-7)$$

If the element of area chosen has a differential size in two directions, as shown in Fig. 10–10, a double integration must be performed to evaluate  $I_{xy}$ . Most often, however, it is easier to choose an element having a differential size or thickness in only one direction in which case the evaluation requires only a single integration (see Example 10.6).

Like the moment of inertia, the product of inertia has units of length raised to the fourth power, e.g.,  $\text{m}^4$ ,  $\text{mm}^4$  or  $\text{ft}^4$ ,  $\text{in}^4$ . However, since  $x$  or  $y$  may be negative, the product of inertia may either be positive, negative, or zero, depending on the location and orientation of the coordinate axes. For example, the product of inertia  $I_{xy}$  for an area will be *zero* if either the  $x$  or  $y$  axis is an axis of *symmetry* for the area, as in Fig. 10–11. Here every element  $dA$  located at point  $(x, y)$  has a corresponding element  $dA$  located at  $(x, -y)$ . Since the products of inertia for these elements are, respectively,  $xy \, dA$  and  $-xy \, dA$ , the algebraic sum or integration of all the elements that are chosen in this way will cancel each other. Consequently, the product of inertia for the total area becomes zero. It also follows from the definition of  $I_{xy}$  that the “sign” of this quantity depends on the quadrant where the area is located. As shown in Fig. 10–12, if the area is rotated from one quadrant to another, the sign of  $I_{xy}$  will change.

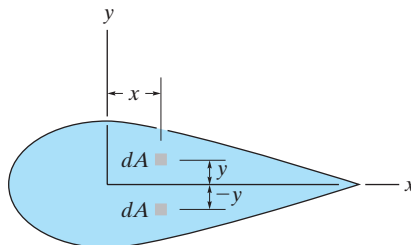


Fig. 10–11

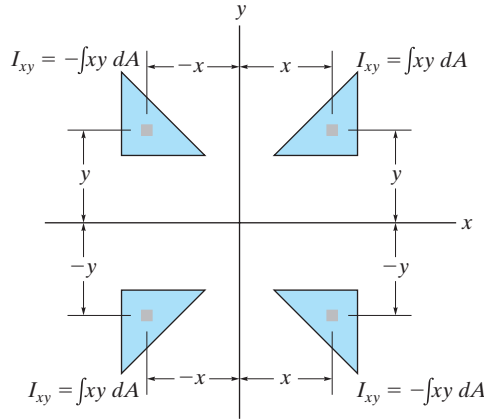


Fig. 10-12

**Parallel-Axis Theorem.** Consider the shaded area shown in Fig. 10-13, where  $x'$  and  $y'$  represent a set of axes passing through the *centroid* of the area, and  $x$  and  $y$  represent a corresponding set of parallel axes. Since the product of inertia of  $dA$  with respect to the  $x$  and  $y$  axes is  $dI_{xy} = (x' + d_x)(y' + d_y) dA$ , then for the entire area,

$$\begin{aligned} I_{xy} &= \int_A (x' + d_x)(y' + d_y) dA \\ &= \int_A x'y' dA + d_x \int_A y' dA + d_y \int_A x' dA + d_x d_y \int_A dA \end{aligned}$$

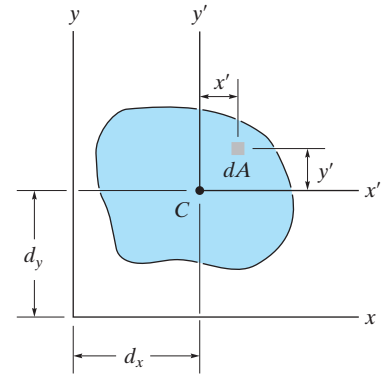


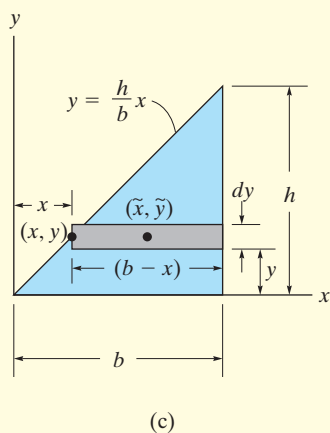
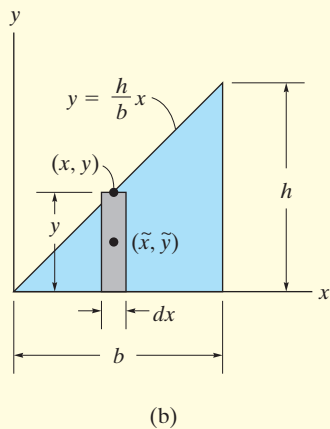
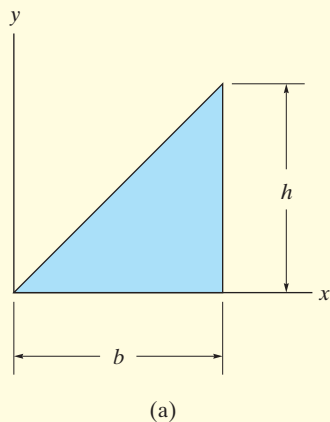
Fig. 10-13

The first term on the right represents the product of inertia for the area with respect to the centroidal axes,  $\bar{I}_{x'y'}$ . The integrals in the second and third terms are zero since the moments of the area are taken about the centroidal axis. Realizing that the fourth integral represents the entire area  $A$ , the parallel-axis theorem for the product of inertia becomes

$$I_{xy} = \bar{I}_{x'y'} + A d_x d_y \quad (10-8)$$

It is important that the *algebraic signs* for  $d_x$  and  $d_y$  be maintained when applying this equation.



**EXAMPLE 10.6****Fig. 10-14**

Determine the product of inertia  $I_{xy}$  for the triangle shown in Fig. 10-14a.

**SOLUTION I**

A differential element that has a thickness  $dx$ , as shown in Fig. 10-14b, has an area  $dA = y dx$ . The product of inertia of this element with respect to the  $x$  and  $y$  axes is determined using the parallel-axis theorem.

$$dI_{xy} = d\bar{I}_{x'y'} + dA \tilde{x} \tilde{y}$$

where  $\tilde{x}$  and  $\tilde{y}$  locate the *centroid* of the element or the origin of the  $x'$ ,  $y'$  axes. (See Fig. 10-13.) Since  $d\bar{I}_{x'y'} = 0$ , due to symmetry, and  $\tilde{x} = x$ ,  $\tilde{y} = y/2$ , then

$$\begin{aligned} dI_{xy} &= 0 + (y dx)x\left(\frac{y}{2}\right) = \left(\frac{h}{b}x dx\right)x\left(\frac{h}{2b}x\right) \\ &= \frac{h^2}{2b^2}x^3 dx \end{aligned}$$

Integrating with respect to  $x$  from  $x = 0$  to  $x = b$  yields

$$I_{xy} = \frac{h^2}{2b^2} \int_0^b x^3 dx = \frac{b^2 h^2}{8} \quad \text{Ans.}$$

**SOLUTION II**

The differential element that has a thickness  $dy$ , as shown in Fig. 10-14c, can also be used. Its area is  $dA = (b - x) dy$ . The *centroid* is located at point  $\tilde{x} = x + (b - x)/2 = (b + x)/2$ ,  $\tilde{y} = y$ , so the product of inertia of the element becomes

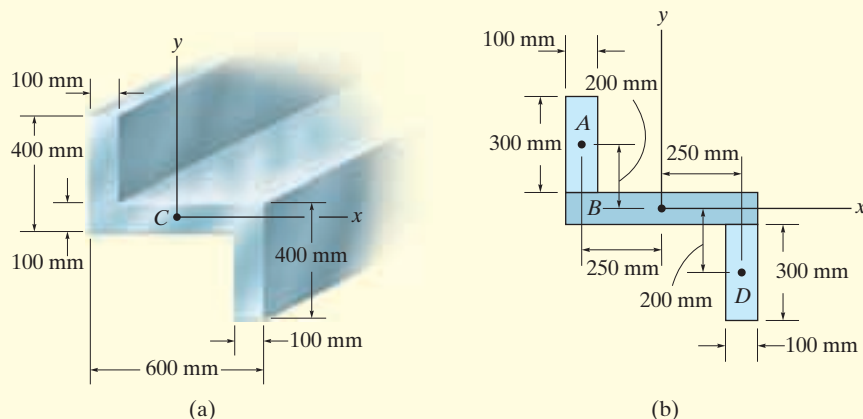
$$\begin{aligned} dI_{xy} &= d\bar{I}_{x'y'} + dA \tilde{x} \tilde{y} \\ &= 0 + (b - x) dy \left(\frac{b + x}{2}\right) y \\ &= \left(b - \frac{b}{h}y\right) dy \left[\frac{b + (b/h)y}{2}\right] y = \frac{1}{2}y \left(b^2 - \frac{b^2}{h^2}y^2\right) dy \end{aligned}$$

Integrating with respect to  $y$  from  $y = 0$  to  $y = h$  yields

$$I_{xy} = \frac{1}{2} \int_0^h y \left(b^2 - \frac{b^2}{h^2}y^2\right) dy = \frac{b^2 h^2}{8} \quad \text{Ans.}$$

**EXAMPLE 10.7**

Determine the product of inertia for the cross-sectional area of the member shown in Fig. 10–15*a*, about the  $x$  and  $y$  centroidal axes.

**Fig. 10–15****SOLUTION**

As in Example 10.5, the cross section can be subdivided into three composite rectangular areas  $A$ ,  $B$ , and  $D$ , Fig. 10–15*b*. The coordinates for the centroid of each of these rectangles are shown in the figure. Due to symmetry, the product of inertia of *each rectangle* is zero about a set of  $x'$ ,  $y'$  axes that passes through the centroid of each rectangle. Using the parallel-axis theorem, we have

Rectangle  $A$

$$\begin{aligned} I_{xy} &= \bar{I}_{x'y'} + A d_x d_y \\ &= 0 + (300)(100)(-250)(200) = -1.50(10^9) \text{ mm}^4 \end{aligned}$$

Rectangle  $B$

$$\begin{aligned} I_{xy} &= \bar{I}_{x'y'} + A d_x d_y \\ &= 0 + 0 = 0 \end{aligned}$$

Rectangle  $D$

$$\begin{aligned} I_{xy} &= \bar{I}_{x'y'} + A d_x d_y \\ &= 0 + (300)(100)(250)(-200) = -1.50(10^9) \text{ mm}^4 \end{aligned}$$

The product of inertia for the entire cross section is therefore

$$I_{xy} = -1.50(10^9) + 0 - 1.50(10^9) = -3.00(10^9) \text{ mm}^4 \quad \text{Ans.}$$

**NOTE:** This negative result is due to the fact that rectangles  $A$  and  $D$  have centroids located with negative  $x$  and negative  $y$  coordinates, respectively.

## \*10.6 Moments of Inertia for an Area about Inclined Axes

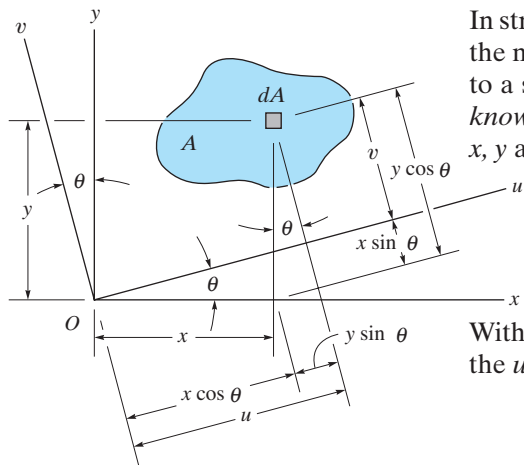


Fig. 10-16

In structural and mechanical design, it is sometimes necessary to calculate the moments and product of inertia  $I_u$ ,  $I_v$ , and  $I_{uv}$  for an area with respect to a set of inclined  $u$  and  $v$  axes when the values for  $\theta$ ,  $I_x$ ,  $I_y$ , and  $I_{xy}$  are known. To do this we will use *transformation equations* which relate the  $x$ ,  $y$  and  $u$ ,  $v$  coordinates. From Fig. 10-16, these equations are

$$u = x \cos \theta + y \sin \theta$$

$$v = y \cos \theta - x \sin \theta$$

With these equations, the moments and product of inertia of  $dA$  about the  $u$  and  $v$  axes become

$$dI_u = v^2 dA = (y \cos \theta - x \sin \theta)^2 dA$$

$$dI_v = u^2 dA = (x \cos \theta + y \sin \theta)^2 dA$$

$$dI_{uv} = uv dA = (x \cos \theta + y \sin \theta)(y \cos \theta - x \sin \theta) dA$$

Expanding each expression and integrating, realizing that  $I_x = \int y^2 dA$ ,  $I_y = \int x^2 dA$ , and  $I_{xy} = \int xy dA$ , we obtain

$$I_u = I_x \cos^2 \theta + I_y \sin^2 \theta - 2I_{xy} \sin \theta \cos \theta$$

$$I_v = I_x \sin^2 \theta + I_y \cos^2 \theta + 2I_{xy} \sin \theta \cos \theta$$

$$I_{uv} = I_x \sin \theta \cos \theta - I_y \sin \theta \cos \theta + I_{xy}(\cos^2 \theta - \sin^2 \theta)$$

Using the trigonometric identities  $\sin 2\theta = 2 \sin \theta \cos \theta$  and  $\cos 2\theta = \cos^2 \theta - \sin^2 \theta$  we can simplify the above expressions, in which case

$$\begin{aligned} I_u &= \frac{I_x + I_y}{2} + \frac{I_x - I_y}{2} \cos 2\theta - I_{xy} \sin 2\theta \\ I_v &= \frac{I_x + I_y}{2} - \frac{I_x - I_y}{2} \cos 2\theta + I_{xy} \sin 2\theta \\ I_{uv} &= \frac{I_x - I_y}{2} \sin 2\theta + I_{xy} \cos 2\theta \end{aligned} \quad (10-9)$$

Notice that if the first and second equations are added together, we can show that the polar moment of inertia about the  $z$  axis passing through point  $O$  is, as expected, *independent* of the orientation of the  $u$  and  $v$  axes; i.e.,

$$J_O = I_u + I_v = I_x + I_y$$

**Principal Moments of Inertia.** Equations 10–9 show that  $I_u$ ,  $I_v$ , and  $I_{uv}$  depend on the angle of inclination,  $\theta$ , of the  $u$ ,  $v$  axes. We will now determine the orientation of these axes about which the moments of inertia for the area are maximum and minimum. This particular set of axes is called the *principal axes* of the area, and the corresponding moments of inertia with respect to these axes are called the **principal moments of inertia**. In general, there is a set of principal axes for every chosen origin  $O$ . However, for structural and mechanical design, the origin  $O$  is located at the centroid of the area.

The angle which defines the orientation of the principal axes can be found by differentiating the first of Eqs. 10–9 with respect to  $\theta$  and setting the result equal to zero. Thus,

$$\frac{dI_u}{d\theta} = -2\left(\frac{I_x - I_y}{2}\right) \sin 2\theta - 2I_{xy} \cos 2\theta = 0$$

Therefore, at  $\theta = \theta_p$ ,

$$\tan 2\theta_p = \frac{-I_{xy}}{(I_x - I_y)/2} \quad (10-10)$$

The two roots  $\theta_{p_1}$  and  $\theta_{p_2}$  of this equation are  $90^\circ$  apart, and so they each specify the inclination of one of the principal axes. In order to substitute them into Eq. 10–9, we must first find the sine and cosine of  $2\theta_{p_1}$  and  $2\theta_{p_2}$ . This can be done using these ratios from the triangles shown in Fig. 10–17, which are based on Eq. 10–10.

Substituting each of the sine and cosine ratios into the first or second of Eqs. 10–9 and simplifying, we obtain

$$I_{\max/\min} = \frac{I_x + I_y}{2} \pm \sqrt{\left(\frac{I_x - I_y}{2}\right)^2 + I_{xy}^2} \quad (10-11)$$

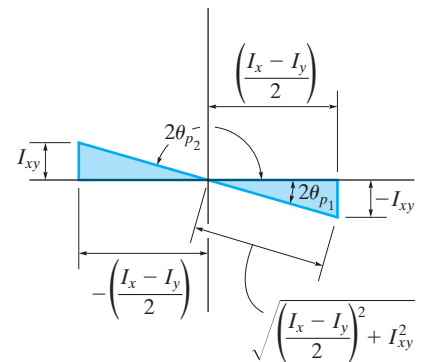
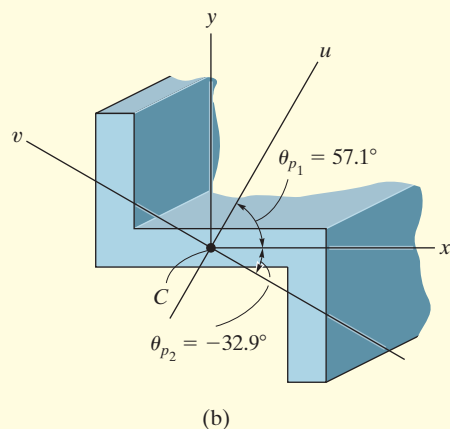
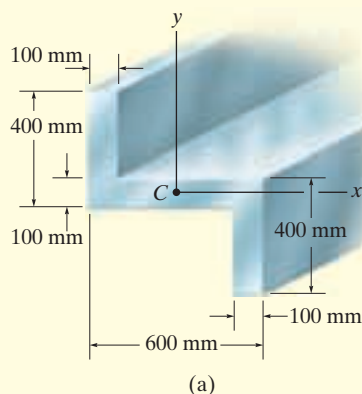


Fig. 10–17

Depending on the sign chosen, this result gives the maximum or minimum moment of inertia for the area. Furthermore, if the above trigonometric relations for  $\theta_{p_1}$  and  $\theta_{p_2}$  are substituted into the third of Eqs. 10–9, it can be shown that  $I_{uv} = 0$ ; that is, the *product of inertia with respect to the principal axes is zero*. Since it was indicated in Sec. 10.6 that the product of inertia is zero with respect to any symmetrical axis, it therefore follows that *any symmetrical axis represents a principal axis of inertia for the area*.

**EXAMPLE 10.8****Fig. 10-18**

Determine the principal moments of inertia and the orientation of the principal axes for the cross-sectional area of the member shown in Fig. 10-18a with respect to an axis passing through the centroid.

**SOLUTION**

The moments and product of inertia of the cross section with respect to the  $x$ ,  $y$  axes have been determined in Examples 10.5 and 10.7. The results are

$$I_x = 2.90(10^9) \text{ mm}^4 \quad I_y = 5.60(10^9) \text{ mm}^4 \quad I_{xy} = -3.00(10^9) \text{ mm}^4$$

Using Eq. 10-10, the angles of inclination of the principal axes  $u$  and  $v$  are

$$\tan 2\theta_p = \frac{-I_{xy}}{(I_x - I_y)/2} = \frac{-[-3.00(10^9)]}{[2.90(10^9) - 5.60(10^9)]/2} = -2.22$$

$$2\theta_p = -65.8^\circ \text{ and } 114.2^\circ$$

Thus, by inspection of Fig. 10-18b,

$$\theta_{p2} = -32.9^\circ \quad \text{and} \quad \theta_{p1} = 57.1^\circ$$

*Ans.*

The principal moments of inertia with respect to these axes are determined from Eq. 10-11. Hence,

$$\begin{aligned} I_{\max/\min} &= \frac{I_x + I_y}{2} \pm \sqrt{\left(\frac{I_x - I_y}{2}\right)^2 + I_{xy}^2} \\ &= \frac{2.90(10^9) + 5.60(10^9)}{2} \\ &\quad \pm \sqrt{\left[\frac{2.90(10^9) - 5.60(10^9)}{2}\right]^2 + [-3.00(10^9)]^2} \\ I_{\max/\min} &= 4.25(10^9) \pm 3.29(10^9) \end{aligned}$$

or

$$I_{\max} = 7.54(10^9) \text{ mm}^4 \quad I_{\min} = 0.960(10^9) \text{ mm}^4 \quad \text{Ans.}$$

**NOTE:** The maximum moment of inertia,  $I_{\max} = 7.54(10^9) \text{ mm}^4$ , occurs with respect to the  $u$  axis since *by inspection* most of the cross-sectional area is farthest away from this axis. Or, stated in another manner,  $I_{\max}$  occurs about the  $u$  axis since this axis is located within  $\pm 45^\circ$  of the  $y$  axis, which has the larger value of  $I$  ( $I_y > I_x$ ). Also, this can be concluded by substituting the data with  $\theta = 57.1^\circ$  into the first of Eqs. 10-9 and solving for  $I_u$ .

## \*10.7 Mohr's Circle for Moments of Inertia

Equations 10–9 to 10–11 have a graphical solution that is convenient to use and generally easy to remember. Squaring the first and third of Eqs. 10–9 and adding, it is found that

$$\left(I_u - \frac{I_x + I_y}{2}\right)^2 + I_{uv}^2 = \left(\frac{I_x - I_y}{2}\right)^2 + I_{xy}^2$$

Here  $I_x$ ,  $I_y$ , and  $I_{xy}$  are *known constants*. Thus, the above equation may be written in compact form as

$$(I_u - a)^2 + I_{uv}^2 = R^2$$

When this equation is plotted on a set of axes that represent the respective moment of inertia and the product of inertia, as shown in Fig. 10–19, the resulting graph represents a *circle* of radius

$$R = \sqrt{\left(\frac{I_x - I_y}{2}\right)^2 + I_{xy}^2}$$

and having its center located at point  $(a, 0)$ , where  $a = (I_x + I_y)/2$ . The circle so constructed is called **Mohr's circle**, named after the German engineer Otto Mohr (1835–1918).

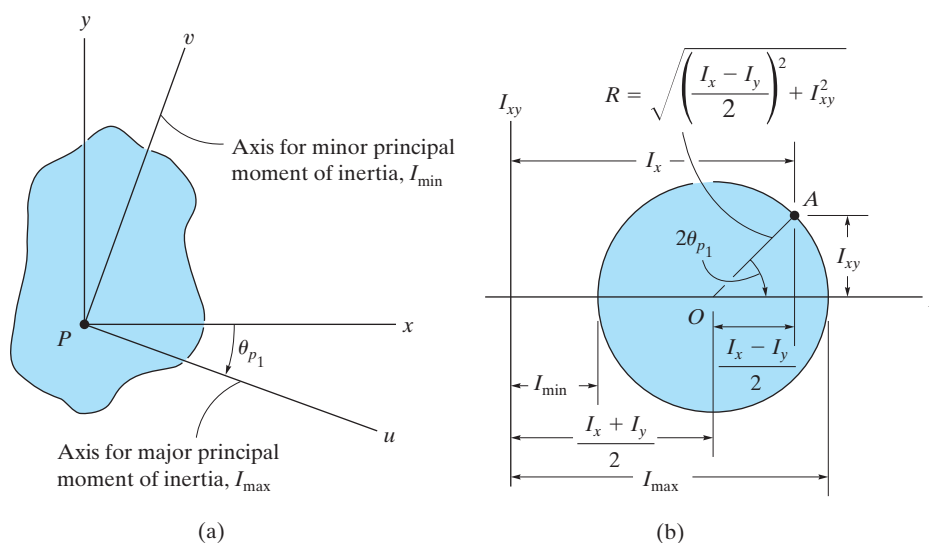


Fig. 10–19

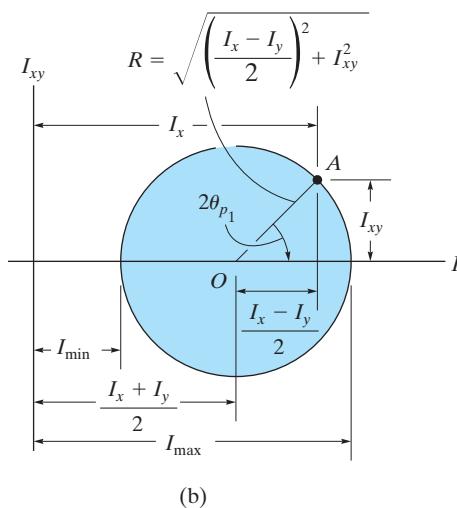
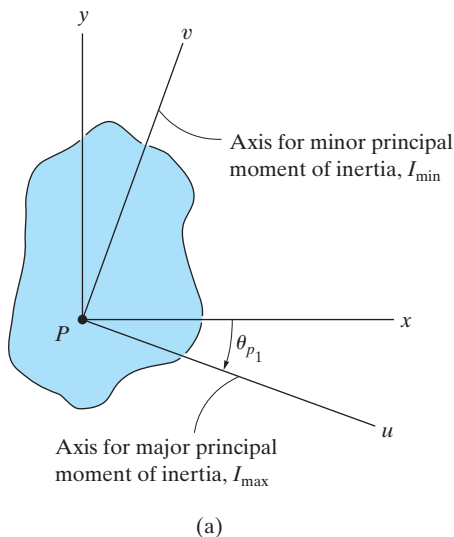


Fig. 10-19 (Repeated)

## Procedure for Analysis

The main purpose in using Mohr's circle here is to have a convenient means for finding the principal moments of inertia for an area. The following procedure provides a method for doing this.

### Determine $I_x$ , $I_y$ , and $I_{xy}$ .

- Establish the  $x, y$  axes and determine  $I_x$ ,  $I_y$ , and  $I_{xy}$ , Fig. 10-19a.

### Construct the Circle.

- Construct a rectangular coordinate system such that the horizontal axis represents the moment of inertia  $I$ , and the vertical axis represents the product of inertia  $I_{xy}$ , Fig. 10-19b.
- Determine the center of the circle,  $O$ , which is located at a distance  $(I_x + I_y)/2$  from the origin, and plot the reference point  $A$  having coordinates  $(I_x, I_{xy})$ . Remember,  $I_x$  is always positive, whereas  $I_{xy}$  can be either positive or negative.
- Connect the reference point  $A$  with the center of the circle and determine the distance  $OA$  by trigonometry. This distance represents the radius of the circle, Fig. 10-19b. Finally, draw the circle.

### Principal Moments of Inertia.

- The points where the circle intersects the  $I$  axis give the values of the principal moments of inertia  $I_{\min}$  and  $I_{\max}$ . Notice that, as expected, the *product of inertia will be zero at these points*, Fig. 10-19b.

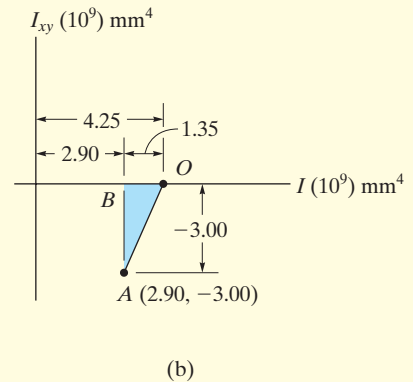
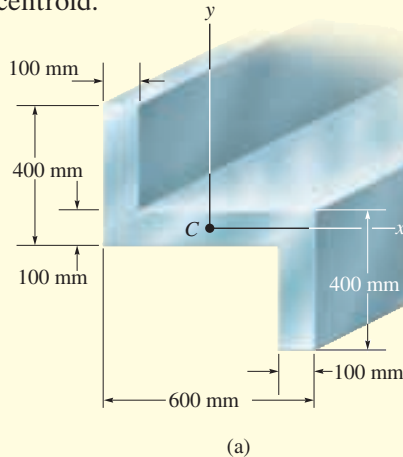
### Principal Axes.

- To find the orientation of the major principal axis, use trigonometry to find the angle  $2\theta_{p1}$ , *measured from the radius  $OA$  to the positive  $I$  axis*, Fig. 10-19b. This angle represents *twice* the angle from the  $x$  axis to the axis of maximum moment of inertia  $I_{\max}$ , Fig. 10-19a. Both the angle on the circle,  $2\theta_{p1}$ , and the angle  $\theta_{p1}$  *must be measured in the same sense*, as shown in Fig. 10-19. The axis for minimum moment of inertia  $I_{\min}$  is perpendicular to the axis for  $I_{\max}$ .

Using trigonometry, the above procedure can be verified to be in accordance with the equations developed in Sec. 10.6.

**EXAMPLE 10.9**

Using Mohr's circle, determine the principal moments of inertia and the orientation of the major principal axes for the cross-sectional area of the member shown in Fig. 10–20*a*, with respect to an axis passing through the centroid.

**SOLUTION**

**Determine  $I_x$ ,  $I_y$ ,  $I_{xy}$ .** The moments and product of inertia have been determined in Examples 10.5 and 10.7 with respect to the  $x$ ,  $y$  axes shown in Fig. 10–20*a*. The results are  $I_x = 2.90(10^9) \text{ mm}^4$ ,  $I_y = 5.60(10^9) \text{ mm}^4$ , and  $I_{xy} = -3.00(10^9) \text{ mm}^4$ .

**Construct the Circle.** The  $I$  and  $I_{xy}$  axes are shown in Fig. 10–20*b*. The center of the circle,  $O$ , lies at a distance  $(I_x + I_y)/2 = (2.90 + 5.60)/2 = 4.25$  from the origin. When the reference point  $A(I_x, I_{xy})$  or  $A(2.90, -3.00)$  is connected to point  $O$ , the radius  $OA$  is determined from the triangle  $OBA$  using the Pythagorean theorem.

$$OA = \sqrt{(1.35)^2 + (-3.00)^2} = 3.29$$

The circle is constructed in Fig. 10–20*c*.

**Principal Moments of Inertia.** The circle intersects the  $I$  axis at points  $(7.54, 0)$  and  $(0.960, 0)$ . Hence,

$$I_{\max} = (4.25 + 3.29)10^9 = 7.54(10^9) \text{ mm}^4$$

Ans.

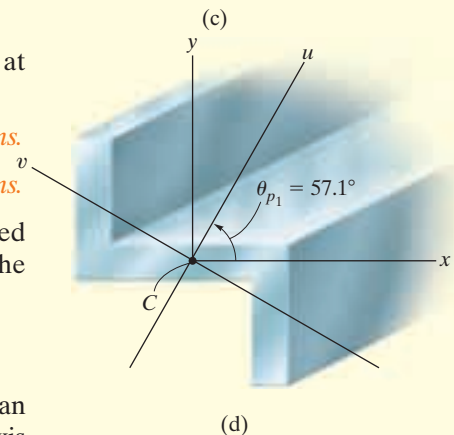
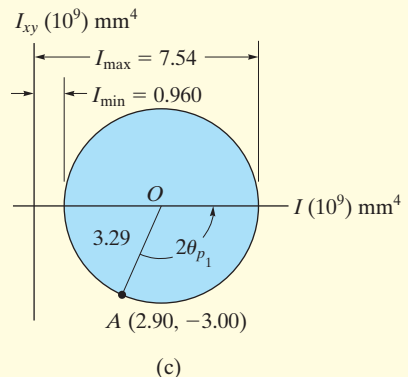
$$I_{\min} = (4.25 - 3.29)10^9 = 0.960(10^9) \text{ mm}^4$$

Ans.

**Principal Axes.** As shown in Fig. 10–20*c*, the angle  $2\theta_{p_1}$  is determined from the circle by measuring counterclockwise from  $OA$  to the direction of the *positive*  $I$  axis. Hence,

$$2\theta_{p_1} = 180^\circ - \sin^{-1}\left(\frac{|BA|}{|OA|}\right) = 180^\circ - \sin^{-1}\left(\frac{3.00}{3.29}\right) = 114.2^\circ$$

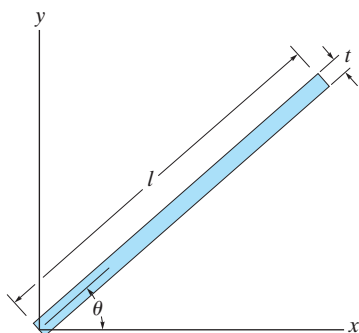
The principal axis for  $I_{\max} = 7.54(10^9) \text{ mm}^4$  is therefore oriented at an angle  $\theta_{p_1} = 57.1^\circ$ , measured *counterclockwise*, from the *positive*  $x$  axis to the *positive*  $u$  axis. The  $v$  axis is perpendicular to this axis. The results are shown in Fig. 10–20*d*.

**Fig. 10–20**



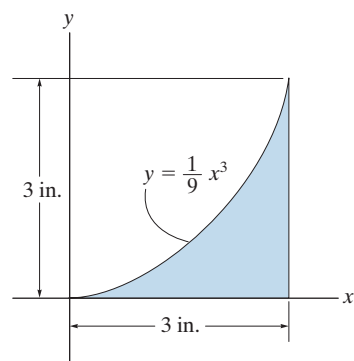
## PROBLEMS

**10–54.** Determine the product of inertia of the thin strip of area with respect to the  $x$  and  $y$  axes. The strip is oriented at an angle  $\theta$  from the  $x$  axis. Assume that  $t \ll l$ .



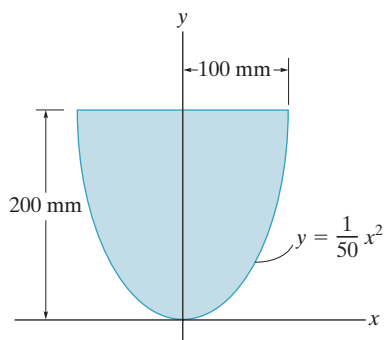
Prob. 10–54

**10–55.** Determine the product of inertia of the shaded area with respect to the  $x$  and  $y$  axes.



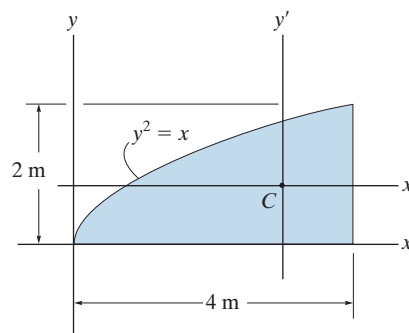
Prob. 10–55

**\*10–56.** Determine the product of inertia for the shaded portion of the parabola with respect to the  $x$  and  $y$  axes.



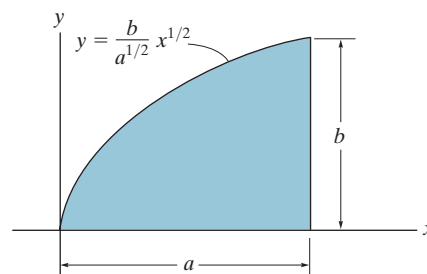
Prob. 10–56

**10–57.** Determine the product of inertia of the shaded area with respect to the  $x$  and  $y$  axes, and then use the parallel-axis theorem to find the product of inertia of the area with respect to the centroidal  $x'$  and  $y'$  axes.



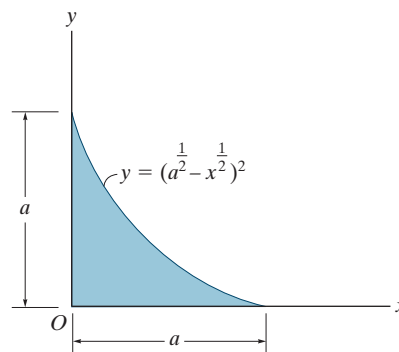
Prob. 10–57

**10–58.** Determine the product of inertia for the parabolic area with respect to the  $x$  and  $y$  axes.



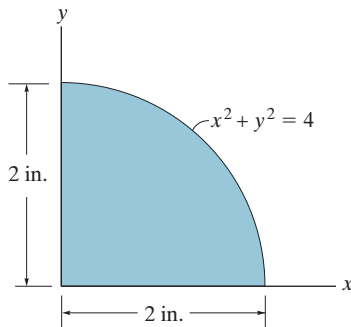
Prob. 10–58

**10–59.** Determine the product of inertia of the shaded area with respect to the  $x$  and  $y$  axes.



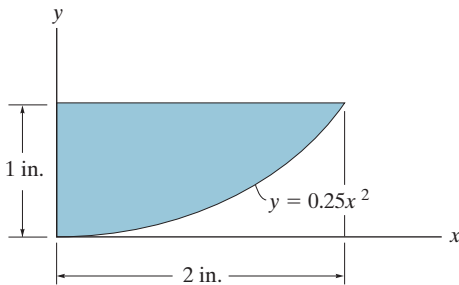
Prob. 10–59

**\*10–60.** Determine the product of inertia of the shaded area with respect to the  $x$  and  $y$  axes.



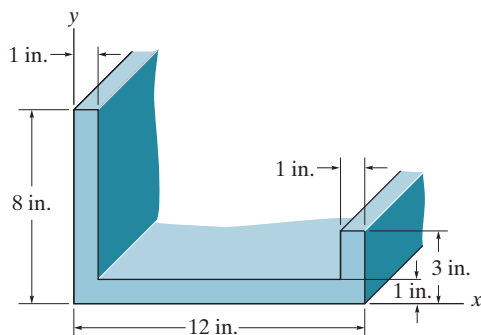
**Prob. 10–60**

**10–61.** Determine the product of inertia of the shaded area with respect to the  $x$  and  $y$  axes.



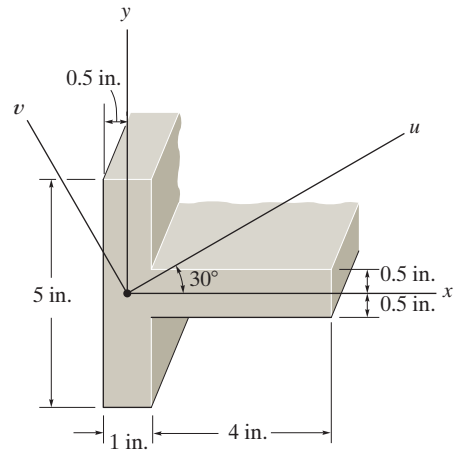
**Prob. 10–61**

**10–62.** Determine the product of inertia for the beam's cross-sectional area with respect to the  $x$  and  $y$  axes.



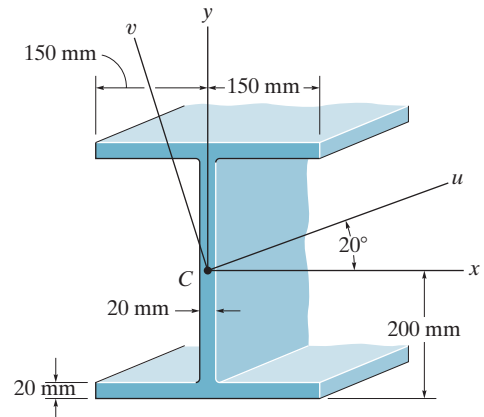
**Prob. 10–62**

**10–63.** Determine the moments of inertia of the shaded area with respect to the  $u$  and  $v$  axes.



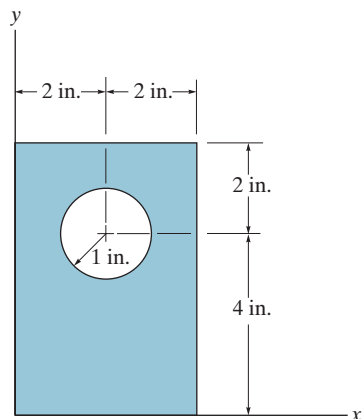
**Prob. 10–63**

**\*10–64.** Determine the product of inertia for the beam's cross-sectional area with respect to the  $u$  and  $v$  axes.



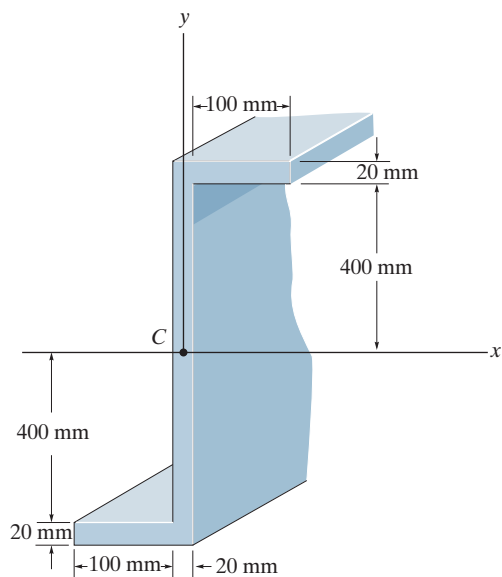
**Prob. 10–64**

**10–65.** Determine the product of inertia for the shaded area with respect to the  $x$  and  $y$  axes.



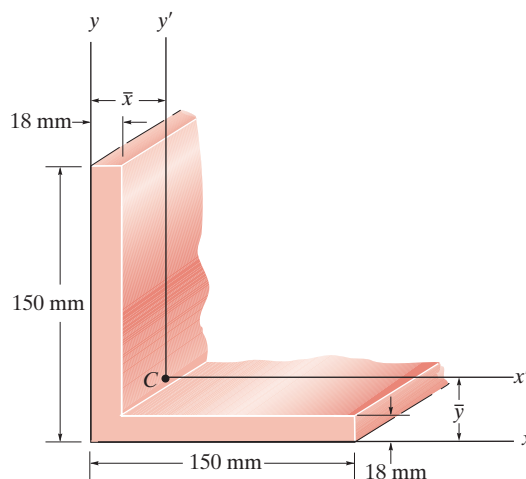
**Prob. 10–65**

**10–66.** Determine the product of inertia of the cross-sectional area with respect to the  $x$  and  $y$  axes.



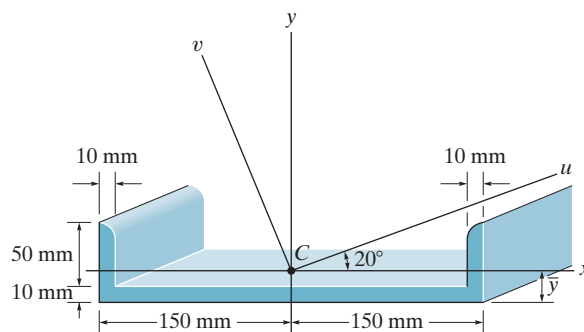
**Prob. 10–66**

**10–67.** Determine the location  $(\bar{x}, \bar{y})$  to the centroid  $C$  of the angle's cross-sectional area, and then compute the product of inertia with respect to the  $x'$  and  $y'$  axes.



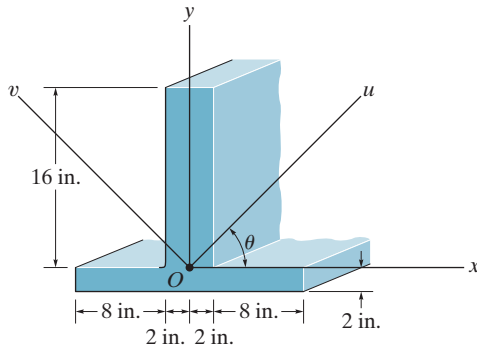
**Prob. 10–67**

**\*10–68.** Determine the distance  $\bar{y}$  to the centroid of the area and then calculate the moments of inertia  $I_u$  and  $I_v$  of the channel's cross-sectional area. The  $u$  and  $v$  axes have their origin at the centroid  $C$ . For the calculation, assume all corners to be square.



**Prob. 10–68**

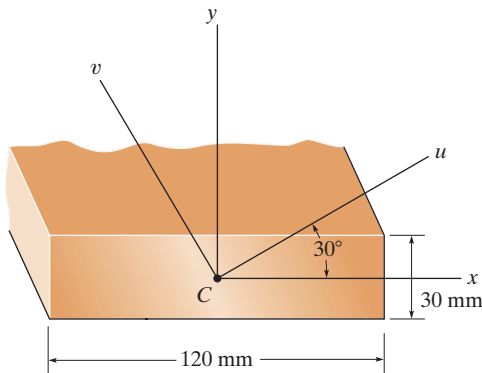
**10–69.** Determine the moments of inertia  $I_u$ ,  $I_v$  and the product of inertia  $I_{uv}$  for the beam's cross-sectional area. Take  $\theta = 45^\circ$ .



**Prob. 10–69**

**10–70.** Determine the moments of inertia  $I_u$ ,  $I_v$  and the product of inertia  $I_{uv}$  for the rectangular area. The  $u$  and  $v$  axes pass through the centroid  $C$ .

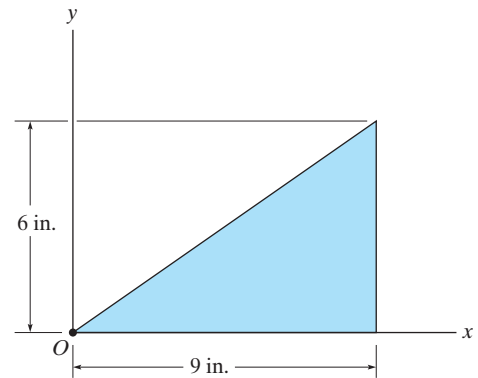
**10–71.** Solve Prob. 10–70 using Mohr's circle. *Hint:* To solve, find the coordinates of the point  $P(I_u, I_{uv})$  on the circle, measured counterclockwise from the radial line  $OA$ . (See Fig. 10–19.) The point  $Q(I_v, -I_{uv})$  is on the opposite side of the circle.



**Probs. 10–70/71**

**\*10–72.** Determine the directions of the principal axes having an origin at point  $O$ , and the principal moments of inertia for the triangular area about the axes.

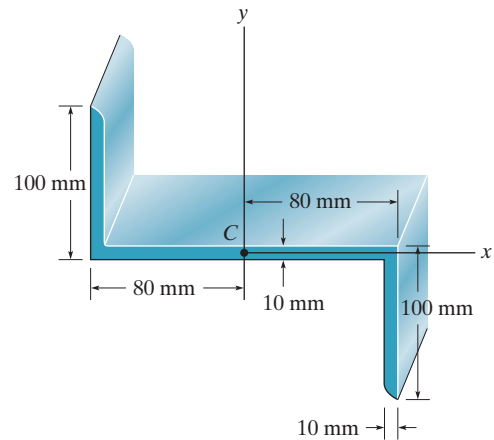
**10–73.** Solve Prob. 10–72 using Mohr's circle.



**Probs. 10–72/73**

**10–74.** Determine the orientation of the principal axes having an origin at point  $C$ , and the principal moments of inertia of the cross section about these axes.

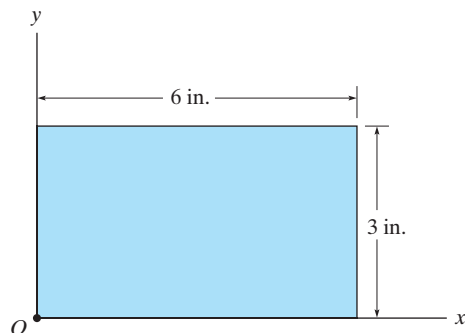
**10–75.** Solve Prob. 10–74 using Mohr's circle.



**Probs. 10–74/75**

**\*10–76.** Determine the orientation of the principal axes having an origin at point  $O$ , and the principal moments of inertia for the rectangular area about these axes.

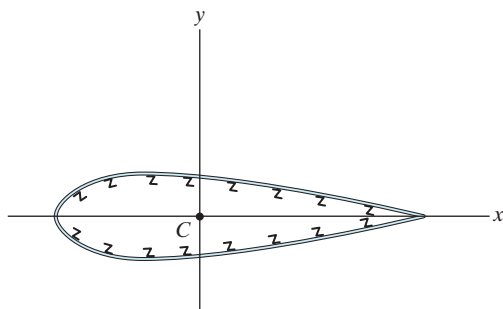
**10–77.** Solve Prob. 10–76 using Mohr's circle.



**Probs. 10–76/77**

**10–78.** The area of the cross section of an airplane wing has the following properties about the  $x$  and  $y$  axes passing through the centroid  $C$ :  $\bar{I}_x = 450 \text{ in}^4$ ,  $\bar{I}_y = 1730 \text{ in}^4$ ,  $\bar{I}_{xy} = 138 \text{ in}^4$ . Determine the orientation of the principal axes and the principal moments of inertia.

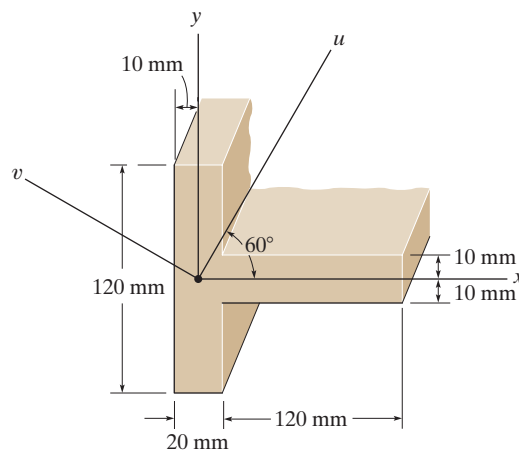
**10–79.** Solve Prob. 10–78 using Mohr's circle.



**Probs. 10–78/79**

**\*10–80.** Determine the moments and product of inertia for the shaded area with respect to the  $u$  and  $v$  axes.

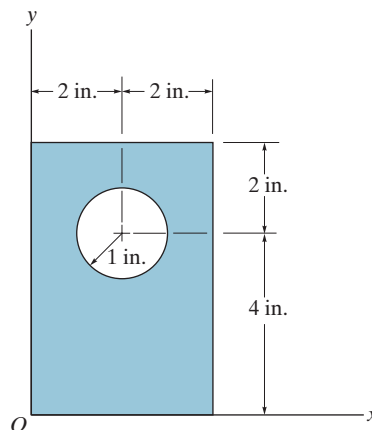
**10–81.** Solve Prob. 10–80 using Mohr's circle.



**Probs. 10–80/81**

**10–82.** Determine the directions of the principal axes with origin located at point  $O$ , and the principal moments of inertia for the area about these axes.

**10–83.** Solve Prob. 10–82 using Mohr's circle.



**Probs. 10–82/83**

## 10.8 Mass Moment of Inertia

The mass moment of inertia of a body is a measure of the body's resistance to angular acceleration. Since it is used in dynamics to study rotational motion, methods for its calculation will now be discussed.\*

Consider the rigid body shown in Fig. 10–21. We define the *mass moment of inertia* of the body about the  $z$  axis as

$$I = \int_m r^2 dm \quad (10-12)$$

Here  $r$  is the perpendicular distance from the axis to the arbitrary element  $dm$ . Since the formulation involves  $r$ , the value of  $I$  is *unique* for each axis about which it is computed. The axis which is generally chosen, however, passes through the body's mass center  $G$ . Common units used for its measurement are  $\text{kg} \cdot \text{m}^2$  or  $\text{slug} \cdot \text{ft}^2$ .

If the body consists of material having a density  $\rho$ , then  $dm = \rho dV$ , Fig. 10–22a. Substituting this into Eq. 10–12, the body's moment of inertia is then computed using *volume elements* for integration; i.e.,

$$I = \int_V r^2 \rho dV \quad (10-13)$$

For most applications,  $\rho$  will be a *constant*, and so this term may be factored out of the integral, and the integration is then purely a function of geometry.

$$I = \rho \int_V r^2 dV \quad (10-14)$$

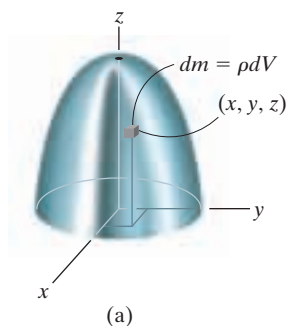


Fig. 10–22

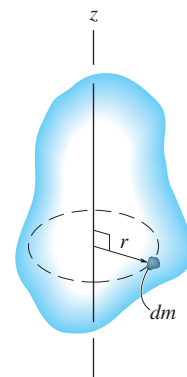


Fig. 10–21

\*Another property of the body, which measures the symmetry of the body's mass with respect to a coordinate system, is the mass product of inertia. This property most often applies to the three-dimensional motion of a body and is discussed in *Engineering Mechanics: Dynamics* (Chapter 21).

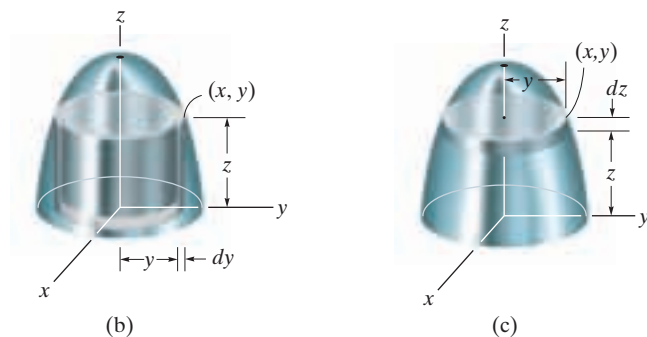


Fig. 10-22 (cont'd)

## Procedure for Analysis

If a body is symmetrical with respect to an axis, as in Fig. 10-22, then its mass moment of inertia about the axis can be determined by using a single integration. Shell and disk elements are used for this purpose.

### Shell Element.

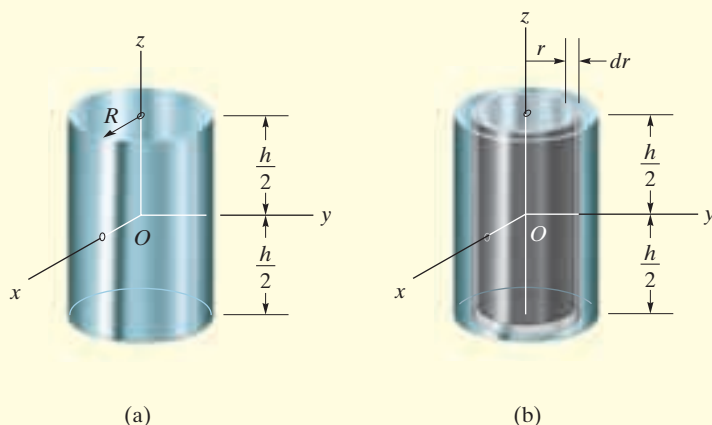
- If a *shell element* having a height  $z$ , radius  $y$ , and thickness  $dy$  is chosen for integration, Fig. 10-22b, then its volume is  $dV = (2\pi y)(z) dy$ .
- This element can be used in Eq. 10-13 or 10-14 for determining the moment of inertia  $I_z$  of the body about the  $z$  axis since the *entire element*, due to its “thinness,” lies at the *same* perpendicular distance  $r = y$  from the  $z$  axis (see Example 10.10).

### Disk Element.

- If a disk element having a radius  $y$  and a thickness  $dz$  is chosen for integration, Fig. 10-22c, then its volume is  $dV = (\pi y^2) dz$ .
- In this case the element is *finite* in the radial direction, and consequently its points *do not* all lie at the *same radial distance*  $r$  from the  $z$  axis. As a result, Eqs. 10-13 or 10-14 *cannot* be used to determine  $I_z$ . Instead, to perform the integration using this element, it is first necessary to determine the moment of inertia *of the element* about the  $z$  axis and then integrate this result (see Example 10.11).

**EXAMPLE 10.10**

Determine the mass moment of inertia of the cylinder shown in Fig. 10–23a about the  $z$  axis. The density of the material,  $\rho$ , is constant.

**Fig. 10–23****SOLUTION**

**Shell Element.** This problem will be solved using the *shell element* in Fig. 10–23b and thus only a single integration is required. The volume of the element is  $dV = (2\pi r)(h) dr$ , and so its mass is  $dm = \rho dV = \rho(2\pi hr dr)$ . Since the *entire element* lies at the same distance  $r$  from the  $z$  axis, the moment of inertia of the *element* is

$$dI_z = r^2 dm = \rho 2\pi h r^3 dr$$

Integrating over the entire cylinder yields

$$I_z = \int_m r^2 dm = \rho 2\pi h \int_0^R r^3 dr = \frac{\rho \pi}{2} R^4 h$$

Since the mass of the cylinder is

$$m = \int_m dm = \rho 2\pi h \int_0^R r dr = \rho \pi h R^2$$

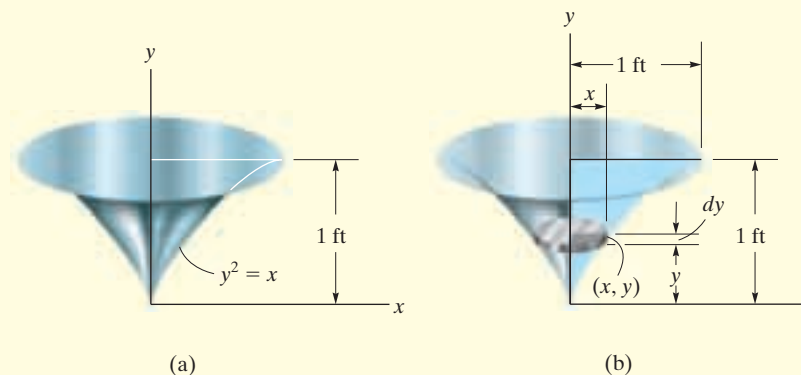
then

$$I_z = \frac{1}{2} m R^2 \quad \text{Ans.}$$



**EXAMPLE 10.11**

If the density of the solid in Fig. 10–24a is 5 slug/ft<sup>3</sup>, determine the mass moment of inertia about the  $y$  axis.

**Fig. 10–24****SOLUTION**

**Disk Element.** The moment of inertia will be determined using this *disk element*, as shown in Fig. 10–24b. Here the element intersects the curve at the arbitrary point  $(x, y)$  and has a mass

$$dm = \rho dV = \rho(\pi x^2) dy$$

Although all points on the element are *not* located at the same distance from the  $y$  axis, it is still possible to determine the moment of inertia  $dI_y$  of the element about the  $y$  axis. In the previous example it was shown that the moment of inertia of a homogeneous cylinder about its longitudinal axis is  $I = \frac{1}{2}mR^2$ , where  $m$  and  $R$  are the mass and radius of the cylinder. Since the height of the cylinder is not involved in this formula, we can also use this result for a disk. Thus, for the disk element in Fig. 10–24b, we have

$$dI_y = \frac{1}{2}(dm)x^2 = \frac{1}{2}[\rho(\pi x^2) dy]x^2$$

Substituting  $x = y^2$ ,  $\rho = 5$  slug/ft<sup>3</sup>, and integrating with respect to  $y$ , from  $y = 0$  to  $y = 1$  ft, yields the moment of inertia for the entire solid.

$$I_y = \frac{5\pi}{2} \int_0^{1 \text{ ft}} x^4 dy = \frac{5\pi}{2} \int_0^{1 \text{ ft}} y^8 dy = 0.873 \text{ slug} \cdot \text{ft}^2 \quad \text{Ans.}$$

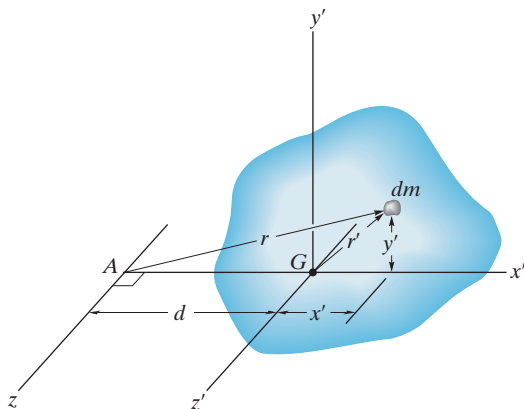


Fig. 10-25

**Parallel-Axis Theorem.** If the moment of inertia of the body about an axis passing through the body's mass center is known, then the moment of inertia about any other *parallel axis* can be determined by using the *parallel-axis theorem*. To derive this theorem, consider the body shown in Fig. 10-25. The  $z'$  axis passes through the mass center  $G$ , whereas the corresponding *parallel  $z$  axis* lies at a constant distance  $d$  away. Selecting the differential element of mass  $dm$ , which is located at point  $(x', y')$ , and using the Pythagorean theorem,  $r^2 = (d + x')^2 + y'^2$ , the moment of inertia of the body about the  $z$  axis is

$$\begin{aligned} I &= \int_m r^2 dm = \int_m [(d + x')^2 + y'^2] dm \\ &= \int_m (x'^2 + y'^2) dm + 2d \int_m x' dm + d^2 \int_m dm \end{aligned}$$

Since  $r'^2 = x'^2 + y'^2$ , the first integral represents  $I_G$ . The second integral is equal to *zero*, since the  $z'$  axis passes through the body's mass center, i.e.,  $\int x' dm = \bar{x} \int dm = 0$  since  $\bar{x} = 0$ . Finally, the third integral is the total mass  $m$  of the body. Hence, the moment of inertia about the  $z$  axis becomes

$$I = I_G + md^2 \quad (10-15)$$

where

$I_G$  = moment of inertia about the  $z'$  axis passing through the mass center  $G$

$m$  = mass of the body

$d$  = distance between the parallel axes

**Radius of Gyration.** Occasionally, the moment of inertia of a body about a specified axis is reported in handbooks using the *radius of gyration*,  $k$ . This value has units of length, and when it and the body's mass  $m$  are known, the moment of inertia can be determined from the equation

$$I = mk^2 \quad \text{or} \quad k = \sqrt{\frac{I}{m}} \quad (10-16)$$

Note the *similarity* between the definition of  $k$  in this formula and  $r$  in the equation  $dI = r^2 dm$ , which defines the moment of inertia of a differential element of mass  $dm$  of the body about an axis.

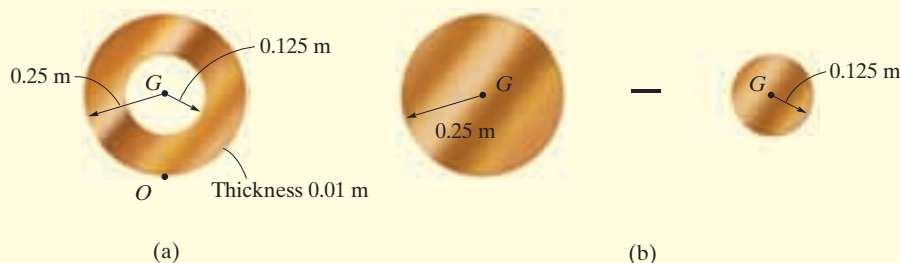
**Composite Bodies.** If a body is constructed from a number of simple shapes such as disks, spheres, and rods, the moment of inertia of the body about any axis  $z$  can be determined by adding algebraically the moments of inertia of all the composite shapes calculated about the same axis. Algebraic addition is necessary since a composite part must be considered as a negative quantity if it has already been included within another part—as in the case of a “hole” subtracted from a solid plate. Also, the parallel-axis theorem is needed for the calculations if the center of mass of each composite part does not lie on the  $z$  axis. For calculations, a table of some simple shapes is given on the inside back cover.



This flywheel, which operates a metal cutter, has a large moment of inertia about its center. Once it begins rotating it is difficult to stop it and therefore a uniform motion can be effectively transferred to the cutting blade. (© Russell C. Hibbeler)

**EXAMPLE 10.12**

If the plate shown in Fig. 10–26*a* has a density of  $8000 \text{ kg/m}^3$  and a thickness of 10 mm, determine its mass moment of inertia about an axis perpendicular to the page and passing through the pin at  $O$ .

**Fig. 10–26****SOLUTION**

The plate consists of two composite parts, the 250-mm-radius disk *minus* a 125-mm-radius disk, Fig. 10–26*b*. The moment of inertia about  $O$  can be determined by finding the moment of inertia of each of these parts about  $O$  and then *algebraically* adding the results. The calculations are performed by using the parallel-axis theorem in conjunction with the mass moment of inertia formula for a circular disk,  $I_G = \frac{1}{2}mr^2$ , as found on the inside back cover.

**Disk.** The moment of inertia of a disk about an axis perpendicular to the plane of the disk and passing through  $G$  is  $I_G = \frac{1}{2}mr^2$ . The mass center of both disks is 0.25 m from point  $O$ . Thus,

$$\begin{aligned} m_d &= \rho_d V_d = 8000 \text{ kg/m}^3 [\pi(0.25 \text{ m})^2(0.01 \text{ m})] = 15.71 \text{ kg} \\ (I_O)_d &= \frac{1}{2}m_d r_d^2 + m_d d^2 \\ &= \frac{1}{2}(15.71 \text{ kg})(0.25 \text{ m})^2 + (15.71 \text{ kg})(0.25 \text{ m})^2 \\ &= 1.473 \text{ kg} \cdot \text{m}^2 \end{aligned}$$

**Hole.** For the smaller disk (hole), we have

$$\begin{aligned} m_h &= \rho_h V_h = 8000 \text{ kg/m}^3 [\pi(0.125 \text{ m})^2(0.01 \text{ m})] = 3.93 \text{ kg} \\ (I_O)_h &= \frac{1}{2}m_h r_h^2 + m_h d^2 \\ &= \frac{1}{2}(3.93 \text{ kg})(0.125 \text{ m})^2 + (3.93 \text{ kg})(0.25 \text{ m})^2 \\ &= 0.276 \text{ kg} \cdot \text{m}^2 \end{aligned}$$

The moment of inertia of the plate about the pin is therefore

$$\begin{aligned} I_O &= (I_O)_d - (I_O)_h \\ &= 1.473 \text{ kg} \cdot \text{m}^2 - 0.276 \text{ kg} \cdot \text{m}^2 \\ &= 1.20 \text{ kg} \cdot \text{m}^2 \end{aligned}$$

*Ans.*

## EXAMPLE 10.13

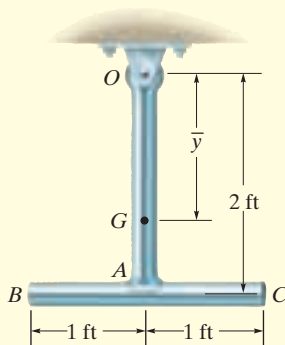


Fig. 10-27

The pendulum in Fig. 10-27 consists of two thin rods each having a weight of 10 lb. Determine the pendulum's mass moment of inertia about an axis passing through (a) the pin at  $O$ , and (b) the mass center  $G$  of the pendulum.

## SOLUTION

**Part (a).** Using the table on the inside back cover, the moment of inertia of rod  $OA$  about an axis perpendicular to the page and passing through the end point  $O$  of the rod is  $I_O = \frac{1}{3}ml^2$ . Hence,

$$(I_{OA})_O = \frac{1}{3}ml^2 = \frac{1}{3}\left(\frac{10 \text{ lb}}{32.2 \text{ ft/s}^2}\right)(2 \text{ ft})^2 = 0.414 \text{ slug} \cdot \text{ft}^2$$

Realize that this same value may be determined using  $I_G = \frac{1}{12}ml^2$  and the parallel-axis theorem; i.e.,

$$\begin{aligned}(I_{OA})_O &= \frac{1}{12}ml^2 + md^2 = \frac{1}{12}\left(\frac{10 \text{ lb}}{32.2 \text{ ft/s}^2}\right)(2 \text{ ft})^2 + \frac{10 \text{ lb}}{32.2 \text{ ft/s}^2}(1 \text{ ft})^2 \\ &= 0.414 \text{ slug} \cdot \text{ft}^2\end{aligned}$$

For rod  $BC$  we have

$$\begin{aligned}(I_{BC})_O &= \frac{1}{12}ml^2 + md^2 = \frac{1}{12}\left(\frac{10 \text{ lb}}{32.2 \text{ ft/s}^2}\right)(2 \text{ ft})^2 + \frac{10 \text{ lb}}{32.2 \text{ ft/s}^2}(2 \text{ ft})^2 \\ &= 1.346 \text{ slug} \cdot \text{ft}^2\end{aligned}$$

The moment of inertia of the pendulum about  $O$  is therefore

$$I_O = 0.414 + 1.346 = 1.76 \text{ slug} \cdot \text{ft}^2 \quad \text{Ans.}$$

**Part (b).** The mass center  $G$  will be located relative to the pin at  $O$ . Assuming this distance to be  $\bar{y}$ , Fig. 10-27, and using the formula for determining the mass center, we have

$$\bar{y} = \frac{\sum \tilde{y}m}{\sum m} = \frac{1(10/32.2) + 2(10/32.2)}{(10/32.2) + (10/32.2)} = 1.50 \text{ ft}$$

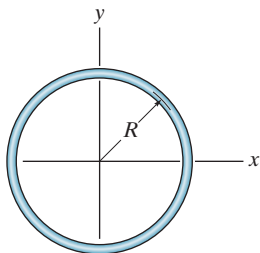
The moment of inertia  $I_G$  may be computed in the same manner as  $I_O$ , which requires successive applications of the parallel-axis theorem in order to transfer the moments of inertia of rods  $OA$  and  $BC$  to  $G$ . A more direct solution, however, involves applying the parallel-axis theorem using the result for  $I_O$  determined above; i.e.,

$$I_O = I_G + md^2; \quad 1.76 \text{ slug} \cdot \text{ft}^2 = I_G + \left(\frac{20 \text{ lb}}{32.2 \text{ ft/s}^2}\right)(1.50 \text{ ft})^2$$

$$I_G = 0.362 \text{ slug} \cdot \text{ft}^2 \quad \text{Ans.}$$

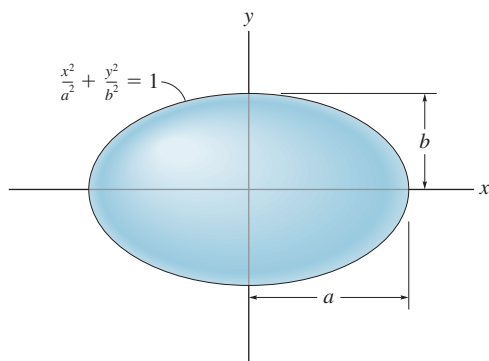
## PROBLEMS

**\*10–84.** Determine the moment of inertia of the thin ring about the  $z$  axis. The ring has a mass  $m$ .



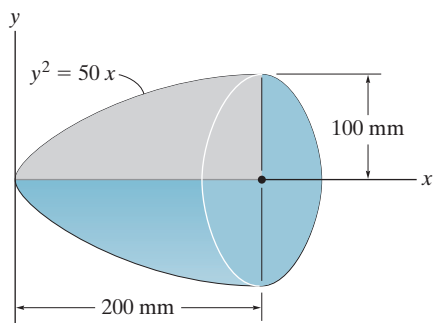
**Prob. 10–84**

**10–85.** Determine the moment of inertia of the ellipsoid with respect to the  $x$  axis and express the result in terms of the mass  $m$  of the ellipsoid. The material has a constant density  $\rho$ .



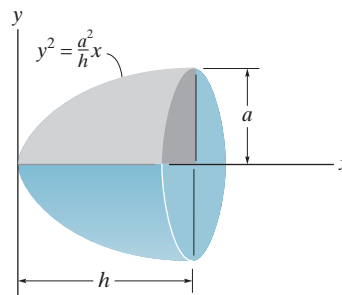
**Prob. 10–85**

**10–86.** Determine the radius of gyration  $k_x$  of the paraboloid. The density of the material is  $\rho = 5 \text{ Mg/m}^3$ .



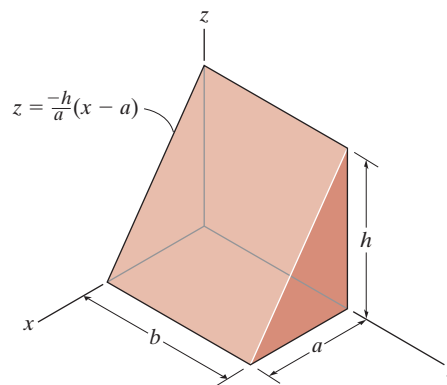
**Prob. 10–86**

**10–87.** The paraboloid is formed by revolving the shaded area around the  $x$  axis. Determine the moment of inertia about the  $x$  axis and express the result in terms of the total mass  $m$  of the paraboloid. The material has a constant density  $\rho$ .



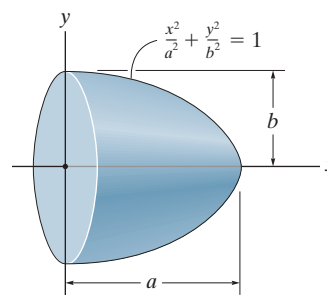
**Prob. 10–87**

**\*10–88.** Determine the moment of inertia of the homogenous triangular prism with respect to the  $y$  axis. Express the result in terms of the mass  $m$  of the prism. *Hint:* For integration, use thin plate elements parallel to the  $x$ – $y$  plane having a thickness of  $dz$ .



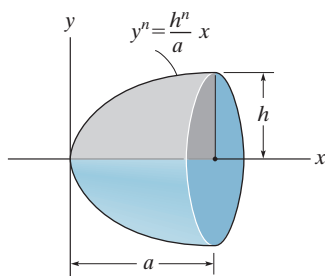
**Prob. 10–88**

**10–89.** Determine the moment of inertia of the semiellipsoid with respect to the  $x$  axis and express the result in terms of the mass  $m$  of the semiellipsoid. The material has a constant density  $\rho$ .



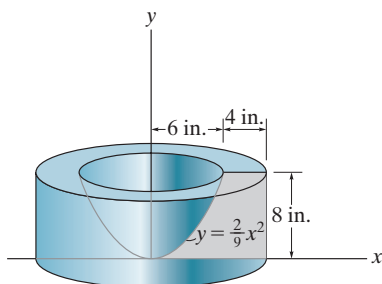
**Prob. 10–89**

**10–90.** Determine the radius of gyration  $k_x$  of the solid formed by revolving the shaded area about  $x$  axis. The density of the material is  $\rho$ .



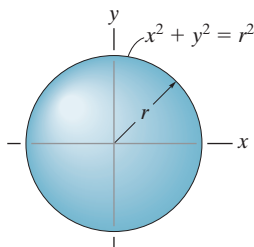
**Prob. 10–90**

**10–91.** The concrete shape is formed by rotating the shaded area about the  $y$  axis. Determine the moment of inertia  $I_y$ . The specific weight of concrete is  $\gamma = 150 \text{ lb/ft}^3$ .



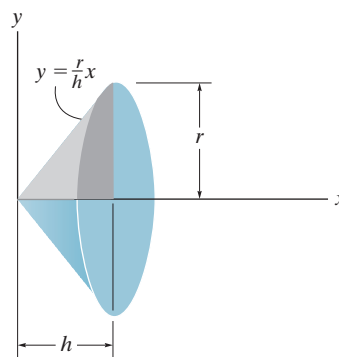
**Prob. 10–91**

**\*10–92.** Determine the moment of inertia  $I_x$  of the sphere and express the result in terms of the total mass  $m$  of the sphere. The sphere has a constant density  $\rho$ .



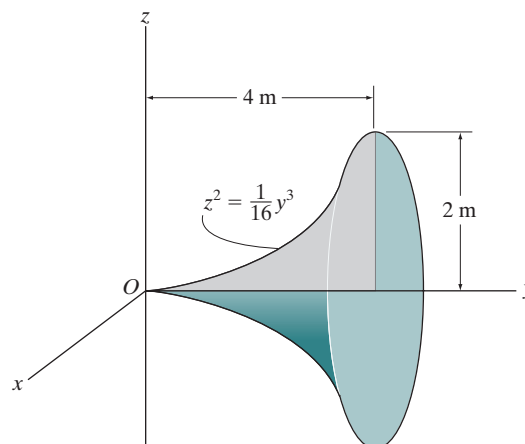
**Prob. 10–92**

**10–93.** The right circular cone is formed by revolving the shaded area around the  $x$  axis. Determine the moment of inertia  $I_x$  and express the result in terms of the total mass  $m$  of the cone. The cone has a constant density  $\rho$ .



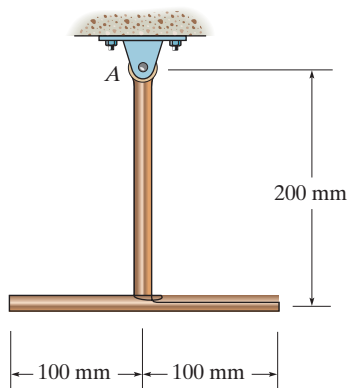
**Prob. 10–93**

**10–94.** Determine the mass moment of inertia  $I_y$  of the solid formed by revolving the shaded area around the  $y$  axis. The total mass of the solid is 1500 kg.



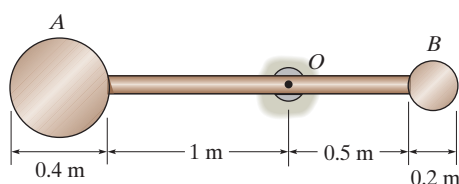
**Prob. 10–94**

**10–95.** The slender rods have a mass of 4 kg/m. Determine the moment of inertia of the assembly about an axis perpendicular to the page and passing through point  $A$ .



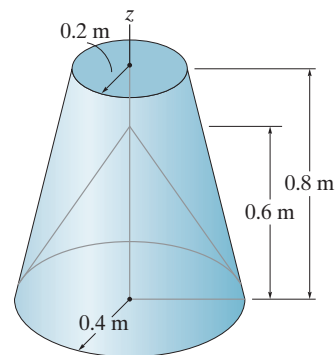
**Prob. 10–95**

**\*10–96.** The pendulum consists of a 8-kg circular disk  $A$ , a 2-kg circular disk  $B$ , and a 4-kg slender rod. Determine the radius of gyration of the pendulum about an axis perpendicular to the page and passing through point  $O$ .



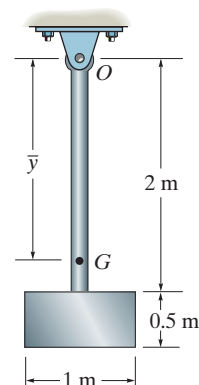
**Prob. 10–96**

**10–97.** Determine the moment of inertia  $I_z$  of the frustum of the cone which has a conical depression. The material has a density of  $200 \text{ kg/m}^3$ .



**Prob. 10–97**

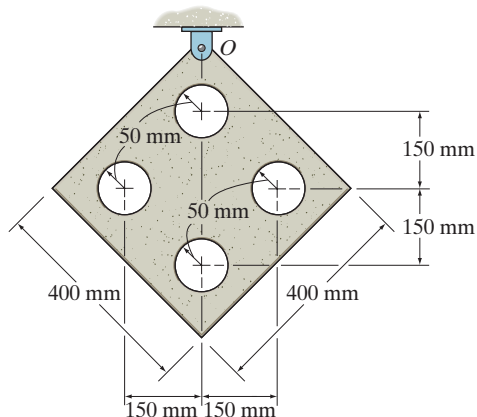
**10–98.** The pendulum consists of the 3-kg slender rod and the 5-kg thin plate. Determine the location  $\bar{y}$  of the center of mass  $G$  of the pendulum; then find the mass moment of inertia of the pendulum about an axis perpendicular to the page and passing through  $G$ .



**Prob. 10–98**

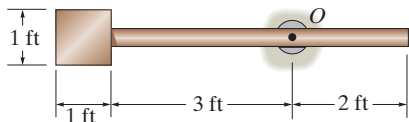


**10–99.** Determine the mass moment of inertia of the thin plate about an axis perpendicular to the page and passing through point  $O$ . The material has a mass per unit area of  $20 \text{ kg/m}^2$ .



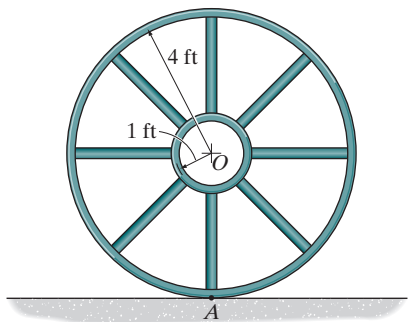
**Prob. 10–99**

**\*10–100.** The pendulum consists of a plate having a weight of 12 lb and a slender rod having a weight of 4 lb. Determine the radius of gyration of the pendulum about an axis perpendicular to the page and passing through point  $O$ .



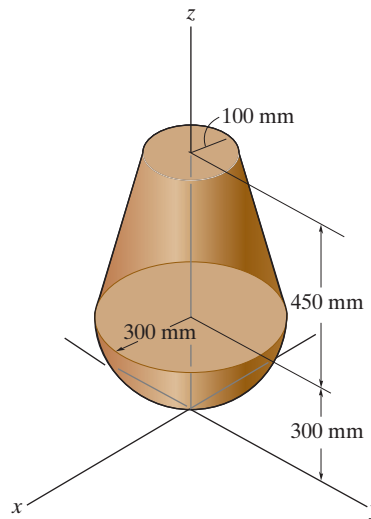
**Prob. 10–100**

**10–101.** If the large ring, small ring and each of the spokes weigh 100 lb, 15 lb, and 20 lb, respectively, determine the mass moment of inertia of the wheel about an axis perpendicular to the page and passing through point  $A$ .



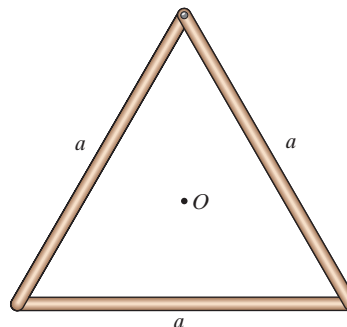
**Prob. 10–101**

**10–102.** Determine the mass moment of inertia of the assembly about the  $z$  axis. The density of the material is  $7.85 \text{ Mg/m}^3$ .



**Prob. 10–102**

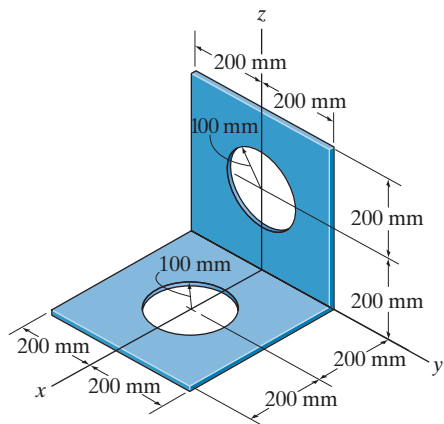
**10–103.** Each of the three slender rods has a mass  $m$ . Determine the moment of inertia of the assembly about an axis that is perpendicular to the page and passes through the center point  $O$ .



**Prob. 10–103**

**\*10–104.** The thin plate has a mass per unit area of  $10 \text{ kg/m}^2$ . Determine its mass moment of inertia about the  $y$  axis.

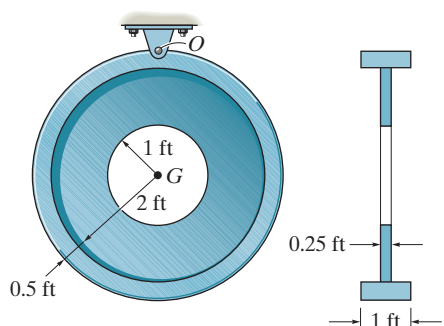
**10–105.** The thin plate has a mass per unit area of  $10 \text{ kg/m}^2$ . Determine its mass moment of inertia about the  $z$  axis.



**Probs. 10–104/105**

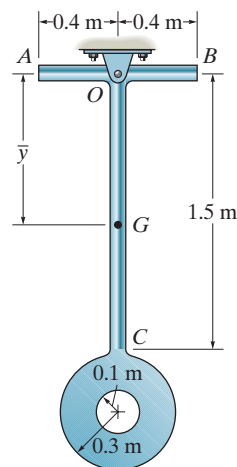
**10–106.** Determine the moment of inertia of the assembly about an axis that is perpendicular to the page and passes through the center of mass  $G$ . The material has a specific weight of  $\gamma = 90 \text{ lb/ft}^3$ .

**10–107.** Determine the moment of inertia of the assembly about an axis that is perpendicular to the page and passes through point  $O$ . The material has a specific weight of  $\gamma = 90 \text{ lb/ft}^3$ .



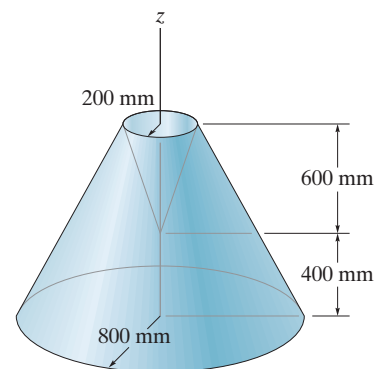
**Probs. 10–106/107**

**\*10–108.** The pendulum consists of two slender rods  $AB$  and  $OC$  which have a mass of  $3 \text{ kg/m}$ . The thin plate has a mass of  $12 \text{ kg/m}^2$ . Determine the location  $\bar{y}$  of the center of mass  $G$  of the pendulum, then calculate the moment of inertia of the pendulum about an axis perpendicular to the page and passing through  $G$ .



**Prob. 10–108**

**10–109.** Determine the moment of inertia  $I_z$  of the frustum of the cone which has a conical depression. The material has a density of  $200 \text{ kg/m}^3$ .



**Prob. 10–109**

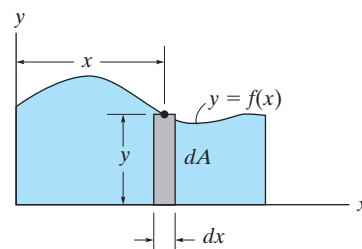
## CHAPTER REVIEW

**Area Moment of Inertia**

The *area moment of inertia* represents the second moment of the area about an axis. It is frequently used in formulas related to the strength and stability of structural members or mechanical elements.

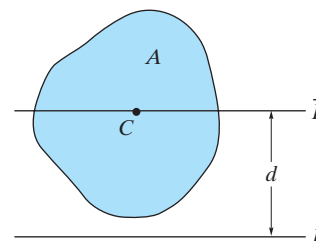
If the area shape is irregular but can be described mathematically, then a differential element must be selected and integration over the entire area must be performed to determine the moment of inertia.

$$I_y = \int_A x^2 dA$$

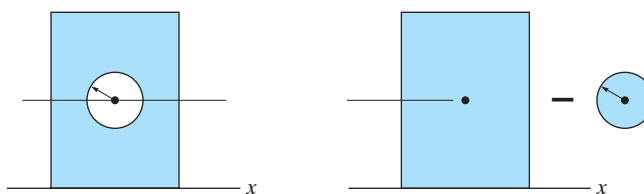
**Parallel-Axis Theorem**

If the moment of inertia for an area is known about a centroidal axis, then its moment of inertia about a parallel axis can be determined using the parallel-axis theorem.

$$I = \bar{I} + Ad^2$$

**Composite Area**

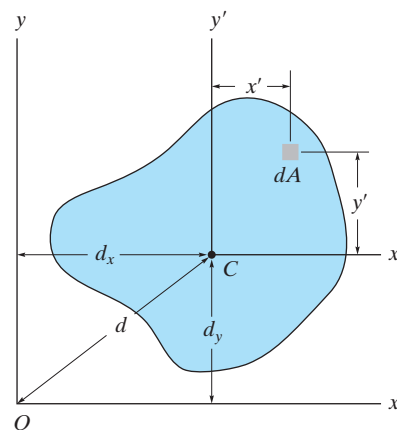
If an area is a composite of common shapes, as found on the inside back cover, then its moment of inertia is equal to the algebraic sum of the moments of inertia of each of its parts.

**Product of Inertia**

The *product of inertia* of an area is used in formulas to determine the orientation of an axis about which the moment of inertia for the area is a maximum or minimum.

$$I_{xy} = \int_A xy dA$$

$$I_{xy} = \bar{I}_{x'y'} + Ad_x d_y$$



### Principal Moments of Inertia

Provided the moments of inertia,  $I_x$  and  $I_y$ , and the product of inertia,  $I_{xy}$ , are known, then the transformation formulas, or Mohr's circle, can be used to determine the maximum and minimum or *principal moments of inertia* for the area, as well as finding the orientation of the principal axes of inertia.

$$I_{\max/\min} = \frac{I_x + I_y}{2} \pm \sqrt{\left(\frac{I_x - I_y}{2}\right)^2 + I_{xy}^2}$$

$$\tan 2\theta_p = \frac{-I_{xy}}{(I_x - I_y)/2}$$

### Mass Moment of Inertia

The *mass moment of inertia* is a property of a body that measures its resistance to a change in its rotation. It is defined as the “second moment” of the mass elements of the body about an axis.

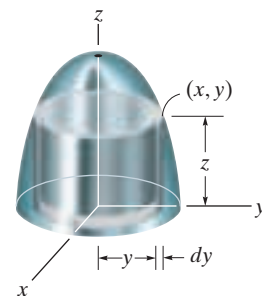
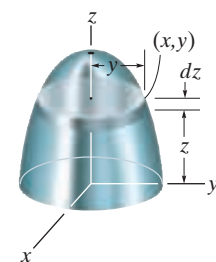
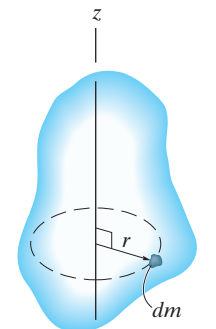
For homogeneous bodies having axial symmetry, the mass moment of inertia can be determined by a single integration, using a disk or shell element.

The mass moment of inertia of a composite body is determined by using tabular values of its composite shapes, found on the inside back cover, along with the parallel-axis theorem.

$$I = \int_m r^2 dm$$

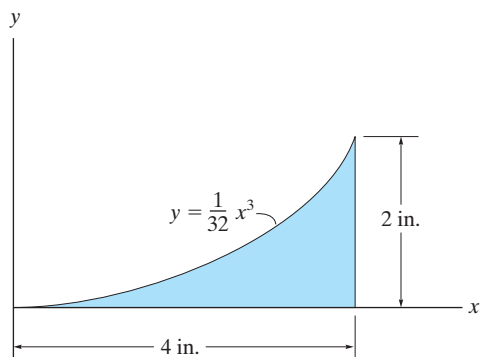
$$I = \rho \int_V r^2 dV$$

$$I = I_G + md^2$$



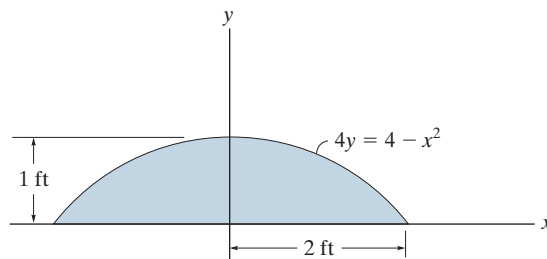
## REVIEW PROBLEMS

**R10-1.** Determine the moment of inertia for the shaded area about the  $x$  axis.



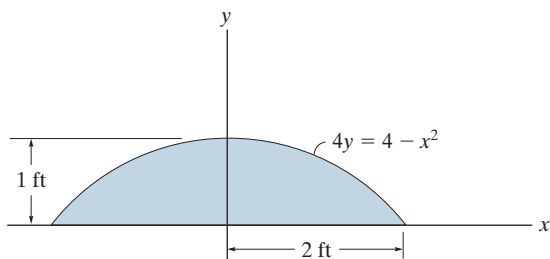
**Prob. R10-1**

**R10-3.** Determine the area moment of inertia of the shaded area about the  $y$  axis.



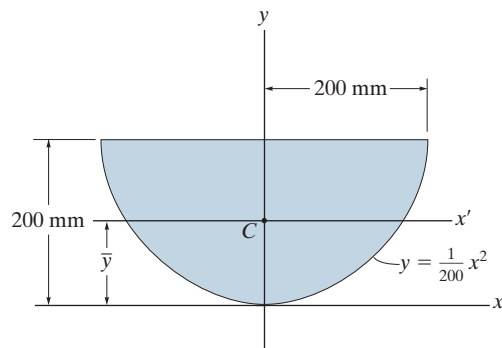
**Prob. R10-3**

**R10-2.** Determine the moment of inertia for the shaded area about the  $x$  axis.



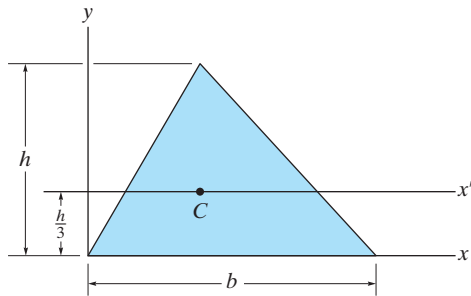
**Prob. R10-2**

**R10-4.** Determine the area moment of inertia of the area about the  $x$  axis. Then, using the parallel-axis theorem, find the area moment of inertia about the  $x'$  axis that passes through the centroid  $C$  of the area.  $\bar{y} = 120$  mm.



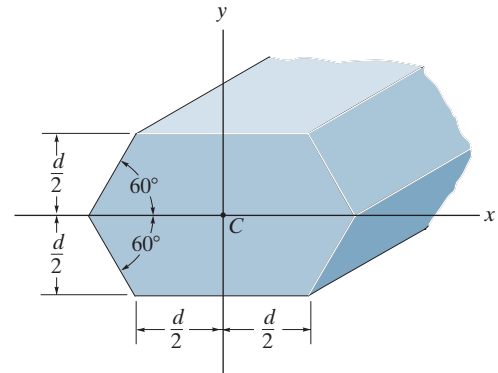
**Prob. R10-4**

**R10-5.** Determine the area moment of inertia of the triangular area about (a) the  $x$  axis, and (b) the centroidal  $x'$  axis.



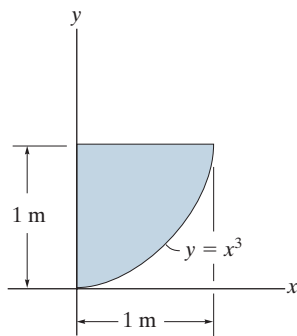
**Prob. R10-5**

**R10-7.** Determine the area moment of inertia of the beam's cross-sectional area about the  $x$  axis which passes through the centroid  $C$ .



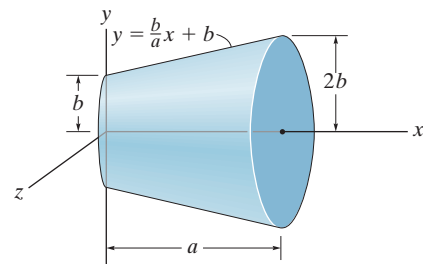
**Prob. R10-7**

**R10-6.** Determine the product of inertia of the shaded area with respect to the  $x$  and  $y$  axes.



**Prob. R10-6**

**R10-8.** Determine the mass moment of inertia  $I_x$  of the body and express the result in terms of the total mass  $m$  of the body. The density is constant.



**Prob. R10-8**

# Chapter 11



(© John Kershaw/Alamy)

Equilibrium and stability of this scissors lift as a function of its position can be determined using the methods of work and energy, which are explained in this chapter.

# Virtual Work

## CHAPTER OBJECTIVES

- To introduce the principle of virtual work and show how it applies to finding the equilibrium configuration of a system of pin-connected members.
- To establish the potential-energy function and use the potential-energy method to investigate the type of equilibrium or stability of a rigid body or system of pin-connected members.

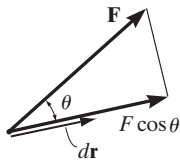
---

## 11.1 Definition of Work

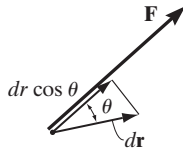
The principle of virtual work was proposed by the Swiss mathematician Jean Bernoulli in the eighteenth century. It provides an alternative method for solving problems involving the equilibrium of a particle, a rigid body, or a system of connected rigid bodies. Before we discuss this principle, however, we must first define the work produced by a force and by a couple moment.

**Work of a Force.** A force does work when it undergoes a displacement in the direction of its line of action. Consider, for example, the force  $\mathbf{F}$  in Fig. 11-1a that undergoes a differential displacement  $d\mathbf{r}$ . If  $\theta$  is the angle between the force and the displacement, then the component of  $\mathbf{F}$  in





(a)



(b)

Fig. 11-1

the direction of the displacement is  $F \cos \theta$ . And so the work produced by  $\mathbf{F}$  is

$$dU = F dr \cos \theta$$

Notice that this expression is also the product of the force  $F$  and the component of displacement in the direction of the force,  $dr \cos \theta$ , Fig. 11-1b. If we use the definition of the dot product (Eq. 2-11) the work can also be written as

$$dU = \mathbf{F} \cdot d\mathbf{r}$$

As the above equations indicate, work is a **scalar**, and like other scalar quantities, it has a magnitude that can either be *positive* or *negative*.

In the SI system, the unit of work is a **joule** (J), which is the work produced by a 1-N force that displaces through a distance of 1 m in the direction of the force ( $1 \text{ J} = 1 \text{ N} \cdot \text{m}$ ). The unit of work in the FPS system is the foot-pound ( $\text{ft} \cdot \text{lb}$ ), which is the work produced by a 1-lb force that displaces through a distance of 1 ft in the direction of the force.

The moment of a force has this same combination of units; however, the concepts of moment and work are in no way related. A moment is a vector quantity, whereas work is a scalar.

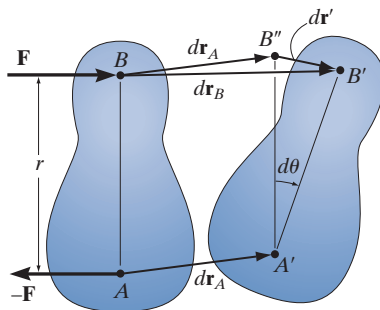


Fig. 11-2

**Work of a Couple Moment.** The rotation of a couple moment also produces work. Consider the rigid body in Fig. 11-2, which is acted upon by the couple forces  $\mathbf{F}$  and  $-\mathbf{F}$  that produce a couple moment  $\mathbf{M}$  having a magnitude  $M = Fr$ . When the body undergoes the differential displacement shown, points  $A$  and  $B$  move  $d\mathbf{r}_A$  and  $d\mathbf{r}_B$  to their final positions  $A'$  and  $B'$ , respectively. Since  $d\mathbf{r}_B = d\mathbf{r}_A + d\mathbf{r}'$ , this movement can be thought of as a *translation*  $d\mathbf{r}_A$ , where  $A$  and  $B$  move to  $A'$  and  $B''$ , and a *rotation* about  $A'$ , where the body rotates through the angle  $d\theta$  about  $A$ . The couple forces do no work during the translation  $d\mathbf{r}_A$  because each force undergoes the same amount of displacement in opposite directions, thus canceling out the work. During rotation, however,  $\mathbf{F}$  is displaced  $dr' = r d\theta$ , and so it does work  $dU = F dr' = F r d\theta$ . Since  $M = Fr$ , the work of the couple moment  $\mathbf{M}$  is therefore

$$dU = M d\theta$$

If  $\mathbf{M}$  and  $d\theta$  have the same sense, the work is *positive*; however, if they have the opposite sense, the work will be *negative*.

**Virtual Work.** The definitions of the work of a force and a couple have been presented in terms of *actual movements* expressed by differential displacements having magnitudes of  $dr$  and  $d\theta$ . Consider now an *imaginary* or **virtual movement** of a body in static equilibrium, which indicates a displacement or rotation that is *assumed* and *does not actually exist*. These movements are first-order differential quantities and will be denoted by the symbols  $\delta r$  and  $\delta\theta$  (delta  $r$  and delta  $\theta$ ), respectively. The *virtual work* done by a force having a virtual displacement  $\delta r$  is

$$\delta U = F \cos \theta \delta r \quad (11-1)$$

Similarly, when a couple undergoes a virtual rotation  $\delta\theta$  in the plane of the couple forces, the *virtual work* is

$$\delta U = M \delta\theta \quad (11-2)$$

## 11.2 Principle of Virtual Work

The **principle of virtual work** states that if a body is in equilibrium, then the algebraic sum of the virtual work done by all the forces and couple moments acting on the body is zero for any virtual displacement of the body. Thus,

$$\delta U = 0 \quad (11-3)$$

For example, consider the free-body diagram of the particle (ball) that rests on the floor, Fig. 11-3. If we “imagine” the ball to be displaced downwards a virtual amount  $\delta y$ , then the weight does positive virtual work,  $W \delta y$ , and the normal force does negative virtual work,  $-N \delta y$ . For equilibrium the total virtual work must be zero, so that  $\delta U = W \delta y - N \delta y = (W - N) \delta y = 0$ . Since  $\delta y \neq 0$ , then  $N = W$  as required by applying  $\Sigma F_y = 0$ .

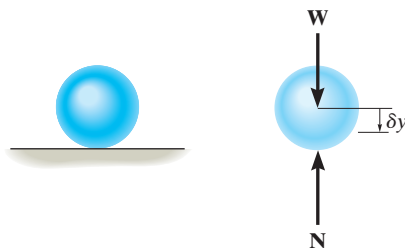


Fig. 11-3

In a similar manner, we can also apply the virtual-work equation  $\delta U = 0$  to a rigid body subjected to a coplanar force system. Here, separate virtual translations in the  $x$  and  $y$  directions, and a virtual rotation about an axis perpendicular to the  $x$ - $y$  plane that passes through an arbitrary point  $O$ , will correspond to the three equilibrium equations,  $\Sigma F_x = 0$ ,  $\Sigma F_y = 0$ , and  $\Sigma M_O = 0$ . When writing these equations, it is *not necessary* to include the work done by the *internal forces* acting within the body since a rigid body *does not deform* when subjected to an external loading, and furthermore, when the body moves through a virtual displacement, the internal forces occur in equal but opposite collinear pairs, so that the corresponding work done by each pair of forces will cancel.

To demonstrate an application, consider the simply supported beam in Fig. 11-4a. When the beam is given a virtual rotation  $\delta\theta$  about point  $B$ , Fig. 11-4b, the only forces that do work are  $\mathbf{P}$  and  $\mathbf{A}_y$ . Since  $\delta y = l \delta\theta$  and  $\delta y' = (l/2) \delta\theta$ , the virtual work equation for this case is  $\delta U = A_y(l \delta\theta) - P(l/2) \delta\theta = (A_y l - Pl/2) \delta\theta = 0$ . Since  $\delta\theta \neq 0$ , then  $A_y = P/2$ . Excluding  $\delta\theta$ , notice that the terms in parentheses actually represent the application of  $\Sigma M_B = 0$ .

As seen from the above two examples, no added advantage is gained by solving particle and rigid-body equilibrium problems using the principle of virtual work. This is because for each application of the virtual-work equation, the virtual displacement, common to every term, factors out, leaving an equation that could have been obtained in a more *direct manner* by simply applying an equation of equilibrium.

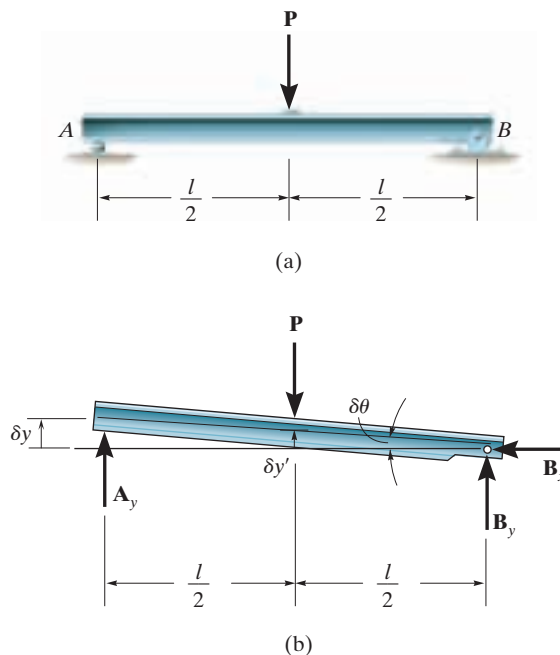


Fig. 11-4

## 11.3 Principle of Virtual Work for a System of Connected Rigid Bodies

The method of virtual work is particularly effective for solving equilibrium problems that involve a system of several *connected* rigid bodies, such as the ones shown in Fig. 11–5.

Each of these systems is said to have only one degree of freedom since the arrangement of the links can be completely specified using only one coordinate  $\theta$ . In other words, with this single coordinate and the length of the members, we can locate the position of the forces  $\mathbf{F}$  and  $\mathbf{P}$ .

In this text, we will only consider the application of the principle of virtual work to systems containing one degree of freedom.\* Because they are less complicated, they will serve as a way to approach the solution of more complex problems involving systems with many degrees of freedom. The procedure for solving problems involving a system of frictionless connected rigid bodies follows.

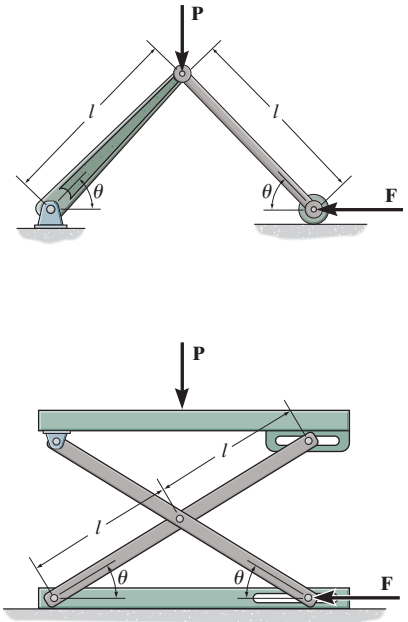


Fig. 11–5

### Important Points

- A force does work when it moves through a displacement in the direction of the force. A couple moment does work when it moves through a collinear rotation. Specifically, positive work is done when the force or couple moment and its displacement have the same sense of direction.
- The principle of virtual work is generally used to determine the equilibrium configuration for a system of multiple connected members.
- A virtual displacement is imaginary; i.e., it does not really happen. It is a differential displacement that is given in the positive direction of a position coordinate.
- Forces or couple moments that do not virtually displace do no virtual work.

\*This method of applying the principle of virtual work is sometimes called the *method of virtual displacements* because a virtual displacement is applied, resulting in the calculation of a real force. Although it is not used here, we can also apply the principle of virtual work as a *method of virtual forces*. This method is often used to apply a virtual force and then determine the displacements of points on deformable bodies. See R. C. Hibbeler, *Mechanics of Materials*, 8th edition, Pearson/Prentice Hall, 2011.



This scissors lift has one degree of freedom. Without the need for dismembering the mechanism, the force in the hydraulic cylinder  $AB$  required to provide the lift can be determined *directly* by using the principle of virtual work. (© Russell C. Hibbeler)

## Procedure for Analysis

### Free-Body Diagram.

- Draw the free-body diagram of the entire system of connected bodies and define the *coordinate*  $q$ .
- Sketch the “deflected position” of the system on the free-body diagram when the system undergoes a *positive virtual* displacement  $\delta q$ .

### Virtual Displacements.

- Indicate *position coordinates*  $s$ , each measured from a *fixed point* on the free-body diagram. These coordinates are directed to the forces that do work.
- Each of these coordinate axes should be *parallel* to the line of action of the force to which it is directed, so that the virtual work along the coordinate axis can be calculated.
- Relate each of the position coordinates  $s$  to the coordinate  $q$ ; then *differentiate* these expressions in order to express each virtual displacement  $\delta s$  in terms of  $\delta q$ .

### Virtual-Work Equation.

- Write the *virtual-work equation* for the system assuming that, whether possible or not, each position coordinate  $s$  undergoes a *positive* virtual displacement  $\delta s$ . If a force or couple moment is in the same direction as the positive virtual displacement, the work is positive. Otherwise, it is negative.
- Express the work of *each* force and couple moment in the equation in terms of  $\delta q$ .
- Factor out this common displacement from all the terms, and solve for the unknown force, couple moment, or equilibrium position  $q$ .

**EXAMPLE 11.1**

Determine the angle  $\theta$  for equilibrium of the two-member linkage shown in Fig. 11–6a. Each member has a mass of 10 kg.

**SOLUTION**

**Free-Body Diagram.** The system has only one degree of freedom since the location of both links can be specified by the single coordinate, ( $q =$ )  $\theta$ . As shown on the free-body diagram in Fig. 11–6b, when  $\theta$  has a *positive* (clockwise) virtual rotation  $\delta\theta$ , only the force  $\mathbf{F}$  and the two 98.1-N weights do work. (The reactive forces  $\mathbf{D}_x$  and  $\mathbf{D}_y$  are fixed, and  $\mathbf{B}_y$  does not displace along its line of action.)

**Virtual Displacements.** If the origin of coordinates is established at the *fixed* pin support  $D$ , then the position of  $\mathbf{F}$  and  $\mathbf{W}$  can be specified by the *position coordinates*  $x_B$  and  $y_w$ . In order to determine the work, note that, as required, these coordinates are parallel to the lines of action of their associated forces. Expressing these position coordinates in terms of  $\theta$  and taking the derivatives yields

$$x_B = 2(1 \cos \theta) \text{ m} \quad \delta x_B = -2 \sin \theta \delta\theta \text{ m} \quad (1)$$

$$y_w = \frac{1}{2}(1 \sin \theta) \text{ m} \quad \delta y_w = 0.5 \cos \theta \delta\theta \text{ m} \quad (2)$$

It is seen by the *signs* of these equations, and indicated in Fig. 11–6b, that an *increase* in  $\theta$  (i.e.,  $\delta\theta$ ) causes a *decrease* in  $x_B$  and an *increase* in  $y_w$ .

**Virtual-Work Equation.** If the virtual displacements  $\delta x_B$  and  $\delta y_w$  were *both positive*, then the forces  $\mathbf{W}$  and  $\mathbf{F}$  would do positive work since the forces and their corresponding displacements would have the same sense. Hence, the virtual-work equation for the displacement  $\delta\theta$  is

$$\delta U = 0; \quad W \delta y_w + W \delta y_w + F \delta x_B = 0 \quad (3)$$

Substituting Eqs. 1 and 2 into Eq. 3 in order to relate the virtual displacements to the common virtual displacement  $\delta\theta$  yields

$$98.1(0.5 \cos \theta \delta\theta) + 98.1(0.5 \cos \theta \delta\theta) + 25(-2 \sin \theta \delta\theta) = 0$$

Notice that the “negative work” done by  $\mathbf{F}$  (force in the opposite sense to displacement) has actually been *accounted for* in the above equation by the “negative sign” of Eq. 1. Factoring out the *common displacement*  $\delta\theta$  and solving for  $\theta$ , noting that  $\delta\theta \neq 0$ , yields

$$(98.1 \cos \theta - 50 \sin \theta) \delta\theta = 0$$

$$\theta = \tan^{-1} \frac{98.1}{50} = 63.0^\circ \quad \text{Ans.}$$

**NOTE:** If this problem had been solved using the equations of equilibrium, it would be necessary to dismember the links and apply three scalar equations to *each* link. The principle of virtual work, by means of calculus, has eliminated this task so that the answer is obtained directly.

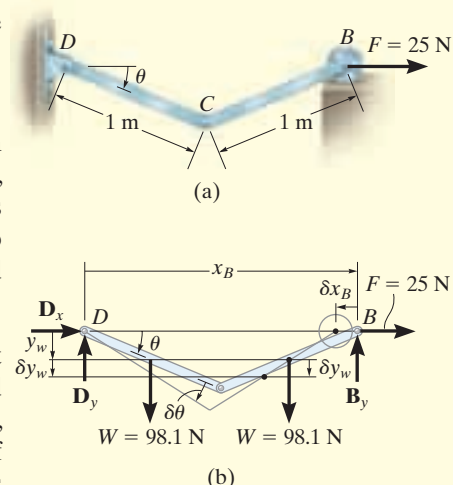
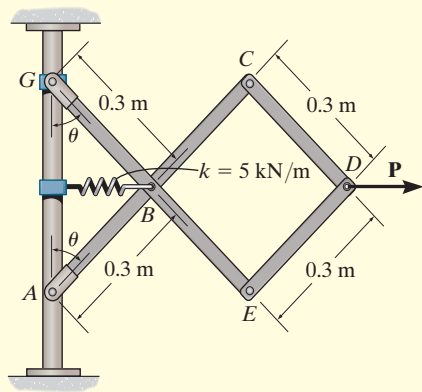
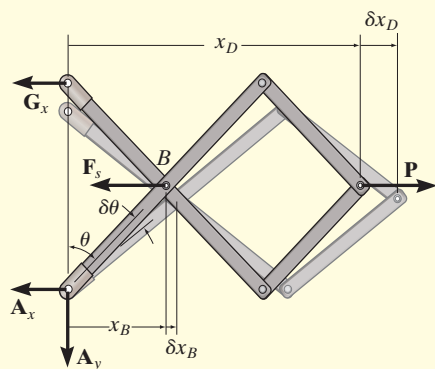


Fig. 11–6

## EXAMPLE 11.2



(a)



(b)

Fig. 11-7

Determine the required force  $P$  in Fig. 11-7a needed to maintain equilibrium of the scissors linkage when  $\theta = 60^\circ$ . The spring is unstretched when  $\theta = 30^\circ$ . Neglect the mass of the links.

## SOLUTION

**Free-Body Diagram.** Only  $\mathbf{F}_s$  and  $\mathbf{P}$  do work when  $\theta$  undergoes a positive virtual displacement  $\delta\theta$ , Fig. 11-7b. For the arbitrary position  $\theta$ , the spring is stretched  $(0.3 \text{ m}) \sin \theta - (0.3 \text{ m}) \sin 30^\circ$ , so that

$$\begin{aligned} F_s &= ks = 5000 \text{ N/m} [(0.3 \text{ m}) \sin \theta - (0.3 \text{ m}) \sin 30^\circ] \\ &= (1500 \sin \theta - 750) \text{ N} \end{aligned}$$

**Virtual Displacements.** The position coordinates,  $x_B$  and  $x_D$ , measured from the fixed point  $A$ , are used to locate  $\mathbf{F}_s$  and  $\mathbf{P}$ . These coordinates are parallel to the line of action of their corresponding forces. Expressing  $x_B$  and  $x_D$  in terms of the angle  $\theta$  using trigonometry,

$$x_B = (0.3 \text{ m}) \sin \theta$$

$$x_D = 3[(0.3 \text{ m}) \sin \theta] = (0.9 \text{ m}) \sin \theta$$

Differentiating, we obtain the virtual displacements of points  $B$  and  $D$ .

$$\delta x_B = 0.3 \cos \theta \delta\theta \quad (1)$$

$$\delta x_D = 0.9 \cos \theta \delta\theta \quad (2)$$

**Virtual-Work Equation.** Force  $\mathbf{P}$  does positive work since it acts in the positive sense of its virtual displacement. The spring force  $\mathbf{F}_s$  does negative work since it acts opposite to its positive virtual displacement. Thus, the virtual-work equation becomes

$$\begin{aligned} \delta U &= 0; & -F_s \delta x_B + P \delta x_D &= 0 \\ & & -[1500 \sin \theta - 750] (0.3 \cos \theta \delta\theta) + P (0.9 \cos \theta \delta\theta) &= 0 \\ & & [0.9P + 225 - 450 \sin \theta] \cos \theta \delta\theta &= 0 \end{aligned}$$

Since  $\cos \theta \delta\theta \neq 0$ , then this equation requires

$$P = 500 \sin \theta - 250$$

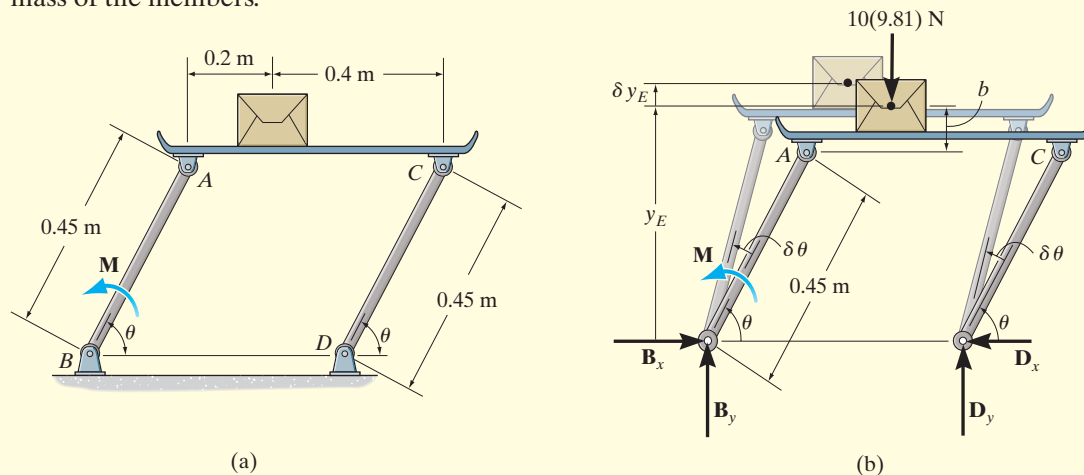
When  $\theta = 60^\circ$ ,

$$P = 500 \sin 60^\circ - 250 = 183 \text{ N}$$

*Ans.*

**EXAMPLE 11.3**

If the box in Fig. 11–8a has a mass of 10 kg, determine the couple moment  $M$  needed to maintain equilibrium when  $\theta = 60^\circ$ . Neglect the mass of the members.

**Fig. 11–8****SOLUTION**

**Free-Body Diagram.** When  $\theta$  undergoes a positive virtual displacement  $\delta\theta$ , only the couple moment  $M$  and the weight of the box do work, Fig. 11–8b.

**Virtual Displacements.** The position coordinate  $y_E$ , measured from the fixed point  $B$ , locates the weight,  $10(9.81)$  N. Here,

$$y_E = (0.45 \text{ m}) \sin \theta + b$$

where  $b$  is a constant distance. Differentiating this equation, we obtain

$$\delta y_E = 0.45 \text{ m} \cos \theta \delta\theta \quad (1)$$

**Virtual-Work Equation.** The virtual-work equation becomes

$$\delta U = 0; \quad M \delta\theta - [10(9.81) \text{ N}] \delta y_E = 0$$

Substituting Eq. 1 into this equation

$$\begin{aligned} M \delta\theta - 10(9.81) \text{ N}(0.45 \text{ m} \cos \theta \delta\theta) &= 0 \\ \delta\theta(M - 44.145 \cos \theta) &= 0 \end{aligned}$$

Since  $\delta\theta \neq 0$ , then

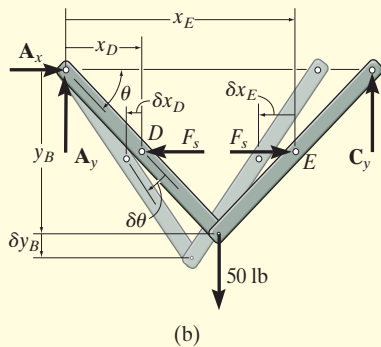
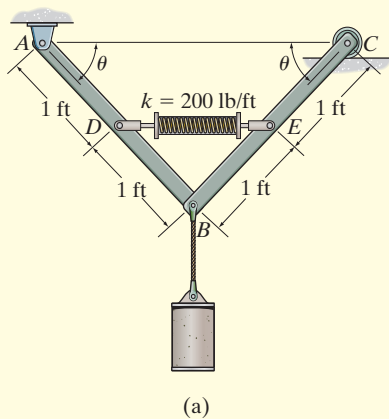
$$M - 44.145 \cos \theta = 0$$

Since it is required that  $\theta = 60^\circ$ , then

$$M = 44.145 \cos 60^\circ = 22.1 \text{ N} \cdot \text{m}$$

*Ans.*



**EXAMPLE 11.4****Fig. 11-9**

The mechanism in Fig. 11-9a supports the 50-lb cylinder. Determine the angle  $\theta$  for equilibrium if the spring has an unstretched length of 2 ft when  $\theta = 0^\circ$ . Neglect the mass of the members.

**SOLUTION**

**Free-Body Diagram.** When the mechanism undergoes a positive virtual displacement  $\delta\theta$ , Fig. 11-9b, only  $\mathbf{F}_s$  and the 50-lb force do work. Since the final length of the spring is  $2(1 \text{ ft} \cos \theta)$ , then

$$F_s = ks = (200 \text{ lb/ft})(2 \text{ ft} - 2 \text{ ft} \cos \theta) = (400 - 400 \cos \theta) \text{ lb}$$

**Virtual Displacements.** The position coordinates  $x_D$  and  $x_E$  are established from the *fixed point* A to locate  $\mathbf{F}_s$  at D and at E. The coordinate  $y_B$ , also measured from A, specifies the position of the 50-lb force at B. The coordinates can be expressed in terms of  $\theta$  using trigonometry.

$$x_D = (1 \text{ ft}) \cos \theta$$

$$x_E = 3[(1 \text{ ft}) \cos \theta] = (3 \text{ ft}) \cos \theta$$

$$y_B = (2 \text{ ft}) \sin \theta$$

Differentiating, we obtain the virtual displacements of points D, E, and B as

$$\delta x_D = -1 \sin \theta \delta\theta \quad (1)$$

$$\delta x_E = -3 \sin \theta \delta\theta \quad (2)$$

$$\delta y_B = 2 \cos \theta \delta\theta \quad (3)$$

**Virtual-Work Equation.** The virtual-work equation is written as if all virtual displacements are positive, thus

$$\delta U = 0; \quad F_s \delta x_E + 50 \delta y_B - F_s \delta x_D = 0$$

$$(400 - 400 \cos \theta)(-3 \sin \theta \delta\theta) + 50(2 \cos \theta \delta\theta)$$

$$- (400 - 400 \cos \theta)(-1 \sin \theta \delta\theta) = 0$$

$$\delta\theta(800 \sin \theta \cos \theta - 800 \sin \theta + 100 \cos \theta) = 0$$

Since  $\delta\theta \neq 0$ , then

$$800 \sin \theta \cos \theta - 800 \sin \theta + 100 \cos \theta = 0$$

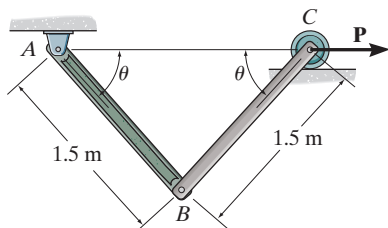
Solving by trial and error,

$$\theta = 34.9^\circ$$

*Ans.*

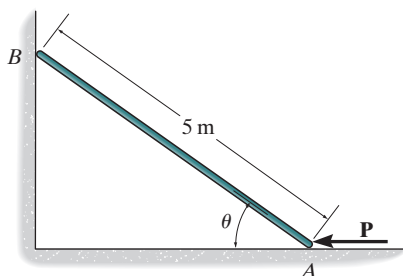
## FUNDAMENTAL PROBLEMS

**F11-1.** Determine the required magnitude of force  $\mathbf{P}$  to maintain equilibrium of the linkage at  $\theta = 60^\circ$ . Each link has a mass of 20 kg.



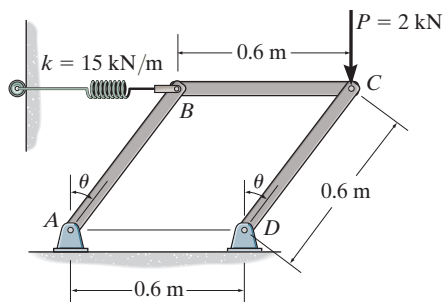
Prob. F11-1

**F11-2.** Determine the magnitude of force  $\mathbf{P}$  required to hold the 50-kg smooth rod in equilibrium at  $\theta = 60^\circ$ .



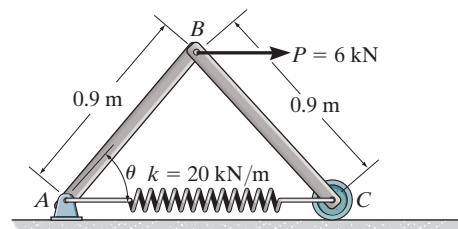
Prob. F11-2

**F11-3.** The linkage is subjected to a force of  $P = 2$  kN. Determine the angle  $\theta$  for equilibrium. The spring is unstretched when  $\theta = 0^\circ$ . Neglect the mass of the links.



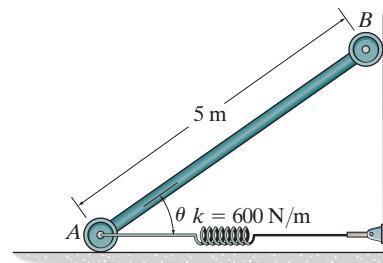
Prob. F11-3

**F11-4.** The linkage is subjected to a force of  $P = 6$  kN. Determine the angle  $\theta$  for equilibrium. The spring is unstretched at  $\theta = 60^\circ$ . Neglect the mass of the links.



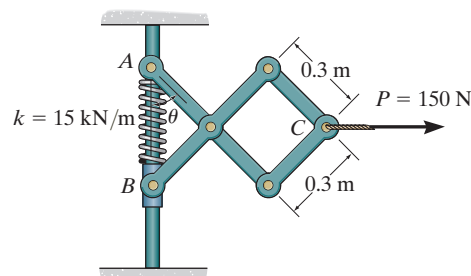
Prob. F11-4

**F11-5.** Determine the angle  $\theta$  where the 50-kg bar is in equilibrium. The spring is unstretched at  $\theta = 60^\circ$ .



Prob. F11-5

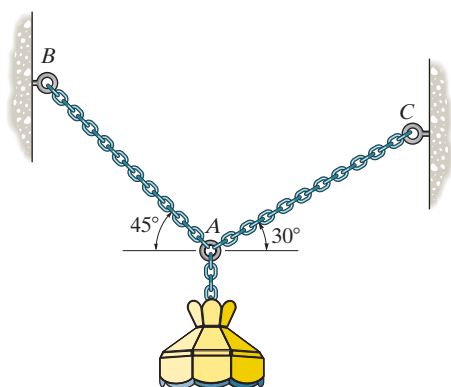
**F11-6.** The scissors linkage is subjected to a force of  $P = 150$  N. Determine the angle  $\theta$  for equilibrium. The spring is unstretched at  $\theta = 0^\circ$ . Neglect the mass of the links.



Prob. F11-6

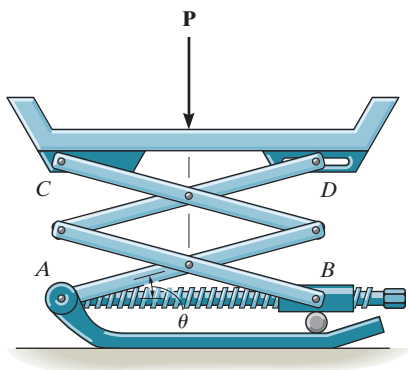
## PROBLEMS

**11-1.** Use the method of virtual work to determine the tension in cable  $AC$ . The lamp weighs 10 lb.



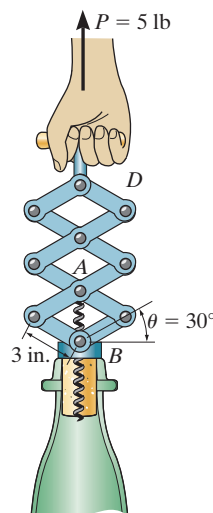
**Prob. 11-1**

**11-2.** The scissors jack supports a load  $P$ . Determine the axial force in the screw necessary for equilibrium when the jack is in the position  $\theta$ . Each of the four links has a length  $L$  and is pin connected at its center. Points  $B$  and  $D$  can move horizontally.



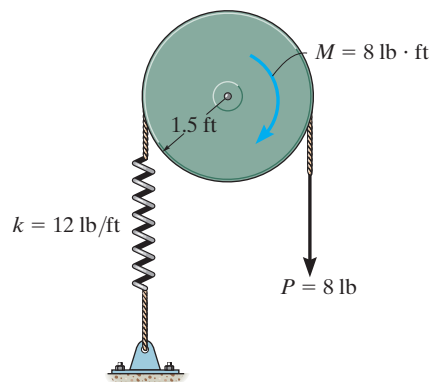
**Prob. 11-2**

**11-3.** If a force of  $P = 5$  lb is applied to the handle of the mechanism, determine the force the screw exerts on the cork of the bottle. The screw is attached to the pin at  $A$  and passes through the collar that is attached to the bottle neck at  $B$ .



**Prob. 11-3**

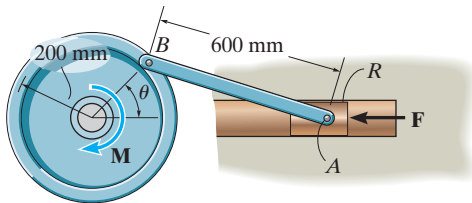
**\*11-4.** The disk has a weight of 10 lb and is subjected to a vertical force  $P = 8$  lb and a couple moment  $M = 8$  lb  $\cdot$  ft. Determine the disk's rotation  $\theta$  if the end of the spring wraps around the periphery of the disk as the disk turns. The spring is originally unstretched.



**Prob. 11-4**

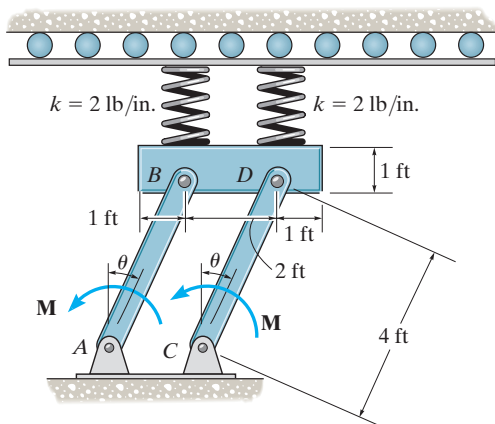
**11-5.** The punch press consists of the ram  $R$ , connecting rod  $AB$ , and a flywheel. If a torque of  $M = 75 \text{ N} \cdot \text{m}$  is applied to the flywheel, determine the force  $\mathbf{F}$  applied at the ram to hold the rod in the position  $\theta = 60^\circ$ .

**11-6.** The flywheel is subjected to a torque of  $M = 75 \text{ N} \cdot \text{m}$ . Determine the horizontal compressive force  $F$  and plot the result of  $F$  (ordinate) versus the equilibrium position  $\theta$  (abscissa) for  $0^\circ \leq \theta \leq 180^\circ$ .



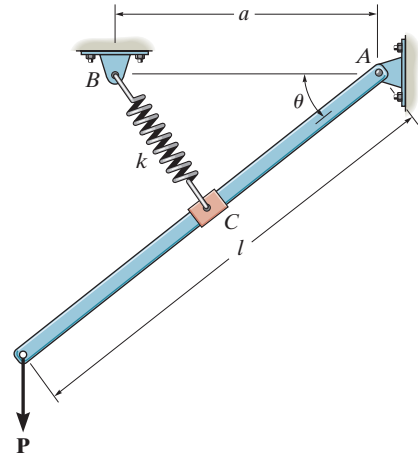
**Probs. 11-5/6**

**11-7.** When  $\theta = 20^\circ$ , the 50-lb uniform block compresses the two vertical springs 4 in. If the uniform links  $AB$  and  $CD$  each weigh 10 lb, determine the magnitude of the applied couple moments  $\mathbf{M}$  needed to maintain equilibrium when  $\theta = 20^\circ$ .



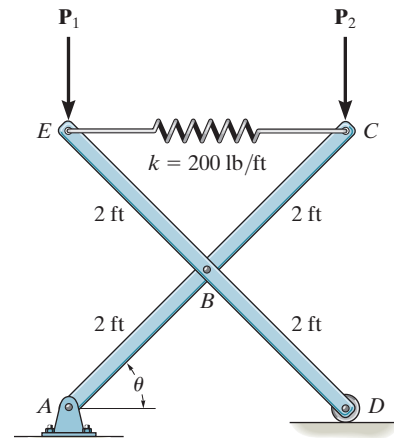
**Prob. 11-7**

**\*11-8.** The bar is supported by the spring and smooth collar that allows the spring to be always perpendicular to the bar for any angle  $\theta$ . If the unstretched length of the spring is  $l_0$ , determine the force  $P$  needed to hold the bar in the equilibrium position  $\theta$ . Neglect the weight of the bar.



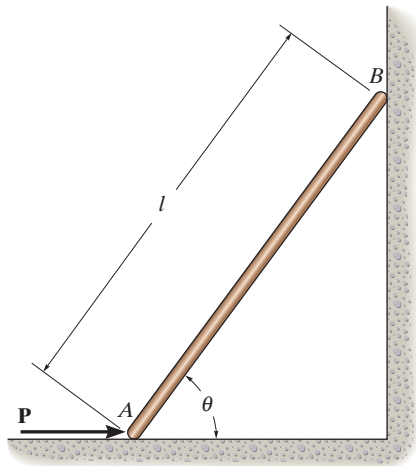
**Prob. 11-8**

**11-9.** The 4-ft members of the mechanism are pin connected at their centers. If vertical forces  $P_1 = P_2 = 30 \text{ lb}$  act at  $C$  and  $E$  as shown, determine the angle  $\theta$  for equilibrium. The spring is unstretched when  $\theta = 45^\circ$ . Neglect the weight of the members.



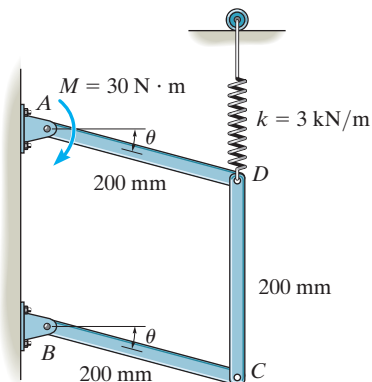
**Prob. 11-9**

**11–10.** The thin rod of weight  $W$  rests against the smooth wall and floor. Determine the magnitude of force  $\mathbf{P}$  needed to hold it in equilibrium for a given angle  $\theta$ .



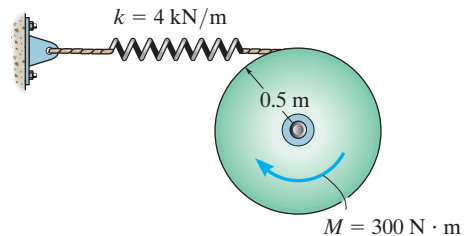
**Prob. 11–10**

**11–11.** If each of the three links of the mechanism have a mass of 4 kg, determine the angle  $\theta$  for equilibrium. The spring, which always remains vertical, is unstretched when  $\theta = 0^\circ$ .



**Prob. 11–11**

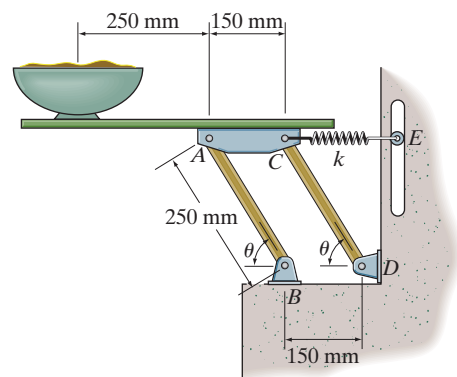
**\*11–12.** The disk is subjected to a couple moment  $M$ . Determine the disk's rotation  $\theta$  required for equilibrium. The end of the spring wraps around the periphery of the disk as the disk turns. The spring is originally unstretched.



**Prob. 11–12**

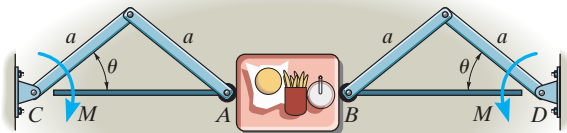
**11–13.** A 5-kg uniform serving table is supported on each side by pairs of two identical links,  $AB$  and  $CD$ , and springs  $CE$ . If the bowl has a mass of 1 kg, determine the angle  $\theta$  where the table is in equilibrium. The springs each have a stiffness of  $k = 200 \text{ N/m}$  and are unstretched when  $\theta = 90^\circ$ . Neglect the mass of the links.

**11–14.** A 5-kg uniform serving table is supported on each side by two pairs of identical links,  $AB$  and  $CD$ , and springs  $CE$ . If the bowl has a mass of 1 kg and is in equilibrium when  $\theta = 45^\circ$ , determine the stiffness  $k$  of each spring. The springs are unstretched when  $\theta = 90^\circ$ . Neglect the mass of the links.



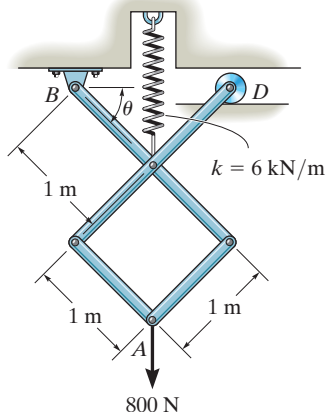
**Probs. 11–13/14**

**11–15.** The service window at a fast-food restaurant consists of glass doors that open and close automatically using a motor which supplies a torque  $\mathbf{M}$  to each door. The far ends,  $A$  and  $B$ , move along the horizontal guides. If a food tray becomes stuck between the doors as shown, determine the horizontal force the doors exert on the tray at the position  $\theta$ .



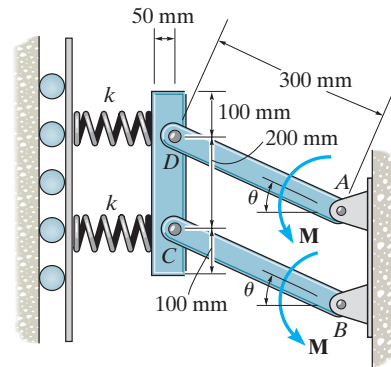
**Prob. 11–15**

**\*11–16.** The members of the mechanism are pin connected. If a vertical force of 800 N acts at  $A$ , determine the angle  $\theta$  for equilibrium. The spring is unstretched when  $\theta = 0^\circ$ . Neglect the mass of the links.



**Prob. 11–16**

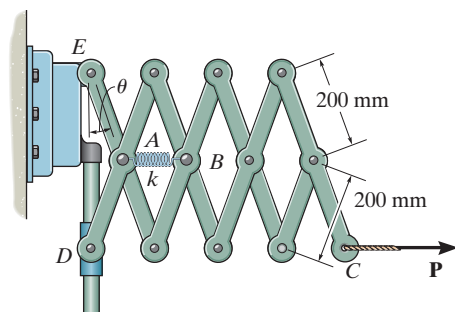
**11–17.** When  $\theta = 30^\circ$ , the 25-kg uniform block compresses the two horizontal springs 100 mm. Determine the magnitude of the applied couple moments  $\mathbf{M}$  needed to maintain equilibrium. Take  $k = 3 \text{ kN/m}$  and neglect the mass of the links.



**Prob. 11–17**

**11–18.** The “Nuremberg scissors” is subjected to a horizontal force of  $P = 600 \text{ N}$ . Determine the angle  $\theta$  for equilibrium. The spring has a stiffness of  $k = 15 \text{ kN/m}$  and is unstretched when  $\theta = 15^\circ$ .

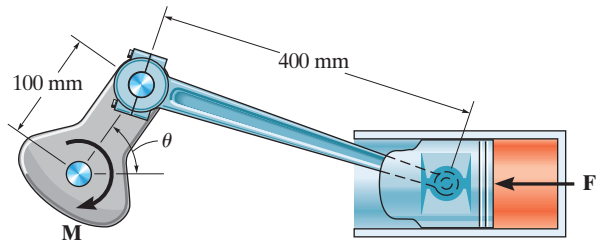
**11–19.** The “Nuremberg scissors” is subjected to a horizontal force of  $P = 600 \text{ N}$ . Determine the stiffness  $k$  of the spring for equilibrium when  $\theta = 60^\circ$ . The spring is unstretched when  $\theta = 15^\circ$ .



**Probs. 11–18/19**

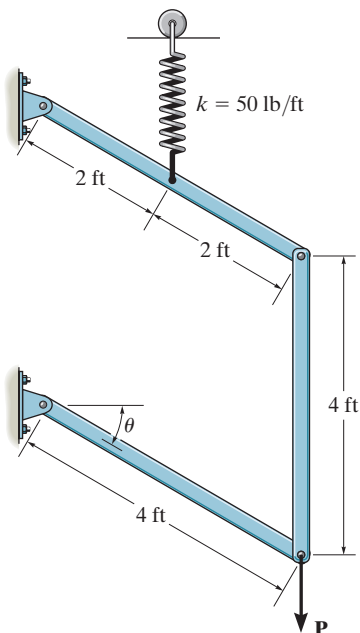
**\*11-20.** The crankshaft is subjected to a torque of  $M = 50 \text{ N} \cdot \text{m}$ . Determine the horizontal compressive force  $F$  applied to the piston for equilibrium when  $\theta = 60^\circ$ .

**11-21.** The crankshaft is subjected to a torque of  $M = 50 \text{ N} \cdot \text{m}$ . Determine the horizontal compressive force  $F$  and plot the result of  $F$  (ordinate) versus  $\theta$  (abscissa) for  $0^\circ \leq \theta \leq 90^\circ$ .



**Probs. 11-20/21**

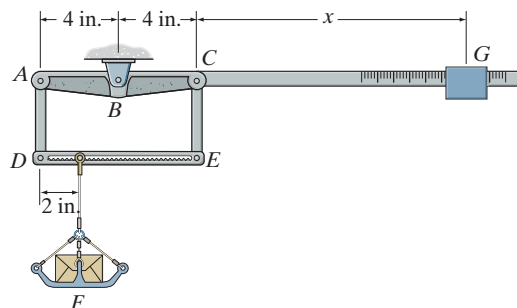
**11-22.** The spring is unstretched when  $\theta = 0^\circ$ . If  $P = 8 \text{ lb}$ , determine the angle  $\theta$  for equilibrium. Due to the roller guide, the spring always remains vertical. Neglect the weight of the links.



**Prob. 11-22**

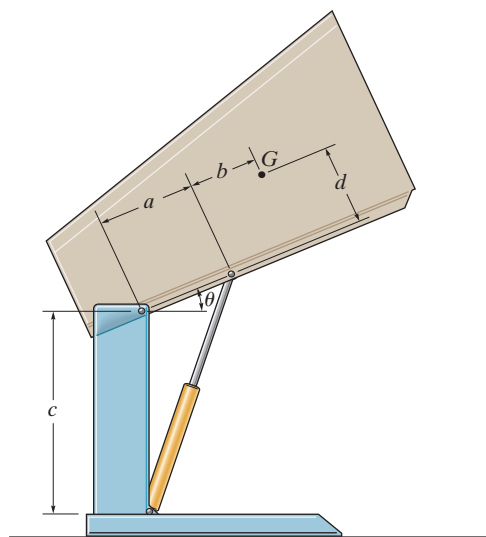
**11-23.** Determine the weight of block  $G$  required to balance the differential lever when the 20-lb load  $F$  is placed on the pan. The lever is in balance when the load and block are not on the lever. Take  $x = 12 \text{ in.}$

**\*11-24.** If the load  $F$  weighs 20 lb and the block  $G$  weighs 2 lb, determine its position  $x$  for equilibrium of the differential lever. The lever is in balance when the load and block are not on the lever.



**Probs. 11-23/24**

**11-25.** The dumpster has a weight  $W$  and a center of gravity at  $G$ . Determine the force in the hydraulic cylinder needed to hold it in the general position  $\theta$ .



**Prob. 11-25**

## \*11.4 Conservative Forces

When a force does work that depends only upon the initial and final positions of the force, and it is *independent* of the path it travels, then the force is referred to as a **conservative force**. The weight of a body and the force of a spring are two examples of conservative forces.

**Weight.** Consider a block of weight  $\mathbf{W}$  that travels along the path in Fig. 11–10a. When it is displaced up the path by an amount  $d\mathbf{r}$ , then the work is  $dU = \mathbf{W} \cdot d\mathbf{r}$  or  $dU = -W(dr \cos \theta) = -Wdy$ , as shown in Fig. 11–10b. In this case, the work is *negative* since  $\mathbf{W}$  acts in the opposite sense of  $dy$ . Thus, if the block moves from  $A$  to  $B$ , through the vertical displacement  $h$ , the work is

$$U = - \int_0^h W dy = -Wh$$

The weight of a body is therefore a conservative force, since the work done by the weight depends only on the *vertical displacement* of the body, and is independent of the path along which the body travels.

**Spring Force.** Now consider the linearly elastic spring in Fig. 11–11, which undergoes a displacement  $ds$ . The work done by the spring force on the block is  $dU = -F_s ds = -ks ds$ . The work is *negative* because  $\mathbf{F}_s$  acts in the opposite sense to that of  $ds$ . Thus, the work of  $\mathbf{F}_s$  when the block is displaced from  $s = s_1$  to  $s = s_2$  is

$$U = - \int_{s_1}^{s_2} ks ds = - \left( \frac{1}{2} ks_2^2 - \frac{1}{2} ks_1^2 \right)$$

Here the work depends only on the spring's initial and final positions,  $s_1$  and  $s_2$ , measured from the spring's unstretched position. Since this result is independent of the path taken by the block as it moves, then a spring force is also a *conservative force*.

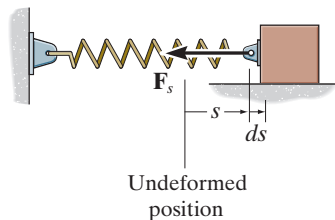


Fig. 11–11

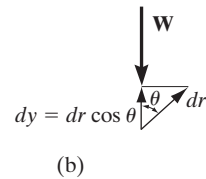
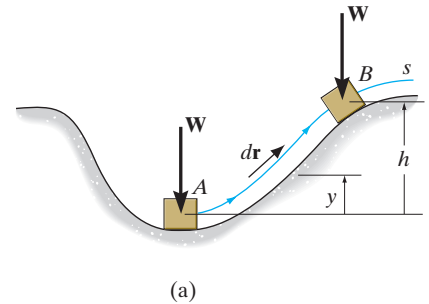


Fig. 11–10



**Friction.** In contrast to a conservative force, consider the force of *friction* exerted on a sliding body by a fixed surface. The work done by the frictional force depends on the path; the longer the path, the greater the work. Consequently, frictional forces are *nonconservative*, and most of the work done by them is dissipated from the body in the form of heat.

## \*11.5 Potential Energy

A conservative force can give the body the capacity to do work. This capacity, measured as **potential energy**, depends on the location or “position” of the body measured relative to a fixed reference position or datum.

**Gravitational Potential Energy.** If a body is located a distance  $y$  *above* a fixed horizontal reference or datum as in Fig. 11–12, the weight of the body has *positive* gravitational potential energy  $V_g$  since  $\mathbf{W}$  has the capacity of doing positive work when the body is moved back down to the datum. Likewise, if the body is located a distance  $y$  *below* the datum,  $V_g$  is *negative* since the weight does negative work when the body is moved back up to the datum. At the datum,  $V_g = 0$ .

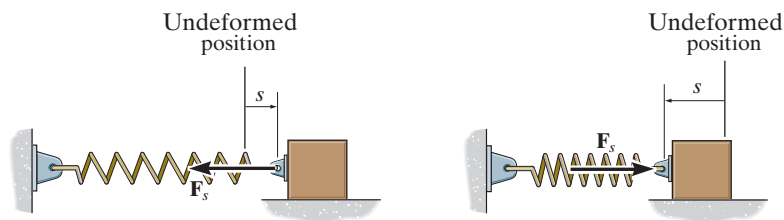
Measuring  $y$  as *positive upward*, the gravitational potential energy of the body’s weight  $\mathbf{W}$  is therefore

$$V_g = Wy \quad (11-4)$$

**Elastic Potential Energy.** When a spring is either elongated or compressed by an amount  $s$  from its unstretched position (the datum), the energy stored in the spring is called *elastic potential energy*. It is determined from

$$V_e = \frac{1}{2} ks^2 \quad (11-5)$$

This energy is always a positive quantity since the spring force acting on the attached body does *positive* work on the body as the force returns the body to the spring’s unstretched position, Fig. 11–13.



$$V_e = +\frac{1}{2} ks^2$$

Fig. 11–13

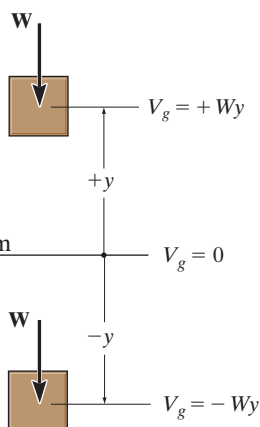


Fig. 11–12

**Potential Function.** In the general case, if a body is subjected to *both* gravitational and elastic forces, the *potential energy or potential function*  $V$  of the body can be expressed as the algebraic sum

$$V = V_g + V_e \quad (11-6)$$

where measurement of  $V$  depends on the location of the body with respect to a selected datum in accordance with Eqs. 11-4 and 11-5.

In particular, if a *system* of frictionless connected rigid bodies has a *single degree of freedom*, such that its vertical distance from the datum is defined by the coordinate  $q$ , then the potential function for the system can be expressed as  $V = V(q)$ . The work done by all the weight and spring forces acting on the system in moving it from  $q_1$  to  $q_2$ , is measured by the *difference* in  $V$ ; i.e.,

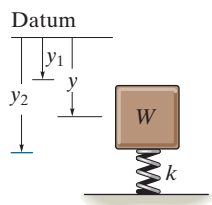
$$U_{1-2} = V(q_1) - V(q_2) \quad (11-7)$$

For example, the potential function for a system consisting of a block of weight  $\mathbf{W}$  supported by a spring, as in Fig. 11-14, can be expressed in terms of the coordinate ( $q =$ )  $y$ , measured from a fixed datum located at the unstretched length of the spring. Here

$$\begin{aligned} V &= V_g + V_e \\ &= -Wy + \frac{1}{2}ky^2 \end{aligned} \quad (11-8)$$

If the block moves from  $y_1$  to  $y_2$ , then applying Eq. 11-7 the work of  $\mathbf{W}$  and  $\mathbf{F}_s$  is

$$U_{1-2} = V(y_1) - V(y_2) = -W(y_1 - y_2) + \frac{1}{2}ky_1^2 - \frac{1}{2}ky_2^2$$



(a)

**Fig. 11-14**

## \*11.6 Potential-Energy Criterion for Equilibrium

If a frictionless connected system has one degree of freedom, and its position is defined by the coordinate  $q$ , then if it displaces from  $q$  to  $q + dq$ , Eq. 11-7 becomes

$$dU = V(q) - V(q + dq)$$

or

$$dU = -dV$$

If the system is in equilibrium and undergoes a *virtual displacement*  $\delta q$ , rather than an actual displacement  $dq$ , then the above equation becomes  $\delta U = -\delta V$ . However, the principle of virtual work requires that  $\delta U = 0$ , and therefore,  $\delta V = 0$ , and so we can write  $\delta V = (dV/dq)\delta q = 0$ . Since  $\delta q \neq 0$ , this expression becomes

$$\frac{dV}{dq} = 0 \quad (11-9)$$

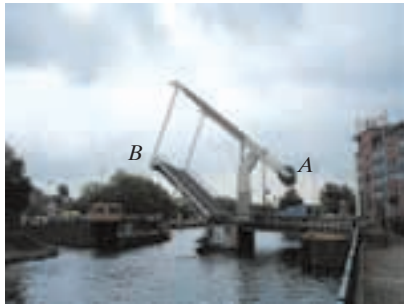
Hence, when a frictionless connected system of rigid bodies is in equilibrium, the first derivative of its potential function is zero. For example, using Eq. 11-8 we can determine the equilibrium position for the spring and block in Fig. 11-14a. We have

$$\frac{dV}{dy} = -W + ky = 0$$

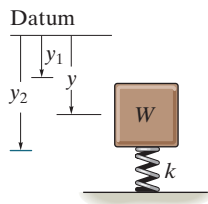
Hence, the equilibrium position  $y = y_{eq}$  is

$$y_{eq} = \frac{W}{k}$$

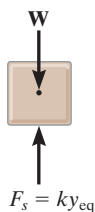
Of course, this *same result* can be obtained by applying  $\Sigma F_y = 0$  to the forces acting on the free-body diagram of the block, Fig. 11-14b.



The counterweight at A balances the weight of the deck B of this simple lift bridge. By applying the method of potential energy we can analyze the equilibrium state of the bridge. (© Russell C. Hibbeler)



(a)



(b)

Fig. 11-14 (cont'd)

## \*11.7 Stability of Equilibrium Configuration

The potential function  $V$  of a system can also be used to investigate the stability of the equilibrium configuration, which is classified as *stable*, *neutral*, or *unstable*.

**Stable Equilibrium.** A system is said to be in *stable equilibrium* if a system has a tendency to return to its original position when a small displacement is given to the system. The potential energy of the system in this case is at its *minimum*. A simple example is shown in Fig. 11–15a. When the disk is given a small displacement, its center of gravity  $G$  will always move (rotate) back to its equilibrium position, which is at the *lowest point* of its path. This is where the potential energy of the disk is at its *minimum*.

**Neutral Equilibrium.** A system is said to be in *neutral equilibrium* if the system still remains in equilibrium when the system is given a small displacement away from its original position. In this case, the potential energy of the system is *constant*. Neutral equilibrium is shown in Fig. 11–15b, where a disk is pinned at  $G$ . Each time the disk is rotated, a new equilibrium position is established and the potential energy remains unchanged.

**Unstable Equilibrium.** A system is said to be in *unstable equilibrium* if it has a tendency to be *displaced farther away* from its original equilibrium position when it is given a small displacement. The potential energy of the system in this case is a *maximum*. An unstable equilibrium position of the disk is shown in Fig. 11–15c. Here the disk will rotate away from its equilibrium position when its center of gravity is slightly displaced. At this *highest point*, its potential energy is at a *maximum*.

**One-Degree-of-Freedom System.** If a system has only one degree of freedom, and its position is defined by the coordinate  $q$ , then the potential function  $V$  for the system in terms of  $q$  can be plotted, Fig. 11–16.



During high winds and when going around a curve, these sugar-cane trucks can become unstable and tip over since their center of gravity is high off the road when they are fully loaded. (© Russell C. Hibbeler)

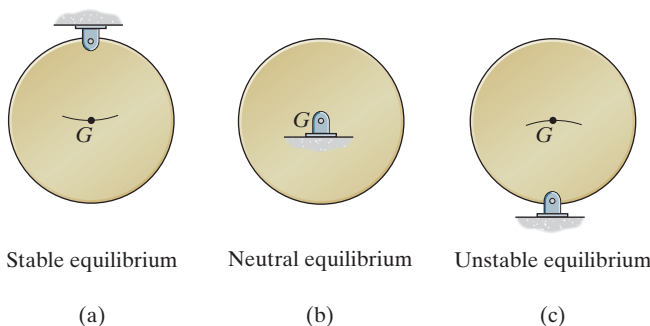
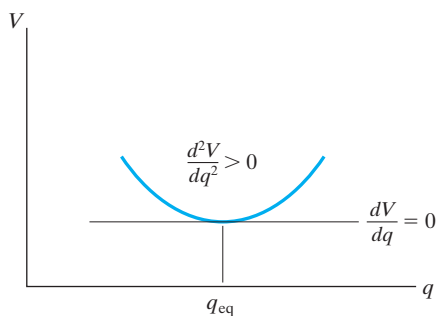
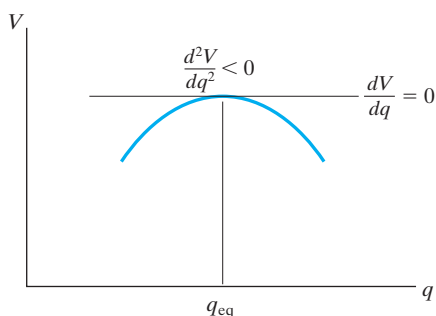


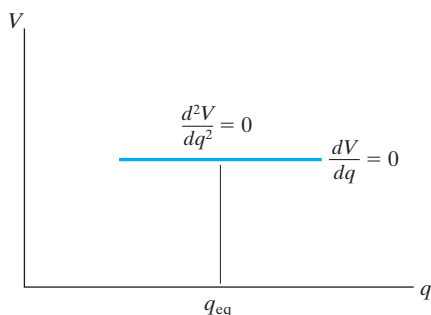
Fig. 11–15



Stable equilibrium  
(a)



Unstable equilibrium  
(b)



Neutral equilibrium  
(c)

Fig. 11-16

Provided the system is in *equilibrium*, then  $dV/dq$ , which represents the slope of this function, must be equal to zero. An investigation of stability at the equilibrium configuration therefore requires that the second derivative of the potential function be evaluated.

If  $d^2V/dq^2$  is greater than zero, Fig. 11-16a, the potential energy of the system will be a *minimum*. This indicates that the equilibrium configuration is *stable*. Thus,

$$\frac{dV}{dq} = 0, \quad \frac{d^2V}{dq^2} > 0 \quad \text{stable equilibrium} \quad (11-10)$$

If  $d^2V/dq^2$  is less than zero, Fig. 11-16b, the potential energy of the system will be a *maximum*. This indicates an *unstable* equilibrium configuration. Thus,

$$\frac{dV}{dq} = 0, \quad \frac{d^2V}{dq^2} < 0 \quad \text{unstable equilibrium} \quad (11-11)$$

Finally, if  $d^2V/dq^2$  is equal to zero, it will be necessary to investigate the higher order derivatives to determine the stability. The equilibrium configuration will be *stable* if the first non-zero derivative is of an *even* order and it is *positive*. Likewise, the equilibrium will be *unstable* if this first non-zero derivative is odd or if it is even and negative. If all the higher order derivatives are *zero*, the system is said to be in *neutral equilibrium*, Fig. 11-16c. Thus,

$$\frac{dV}{dq} = \frac{d^2V}{dq^2} = \frac{d^3V}{dq^3} = \cdots = 0 \quad \text{neutral equilibrium} \quad (11-12)$$

This condition occurs only if the potential-energy function for the system is constant at or around the neighborhood of  $q_{eq}$ .

### Important Points

- A conservative force does work that is independent of the path through which the force moves. Examples include the weight and the spring force.
- Potential energy provides the body with the capacity to do work when the body moves relative to a fixed position or datum. Gravitational potential energy can be positive when the body is above a datum, and negative when the body is below the datum. Spring or elastic potential energy is always positive. It depends upon the stretch or compression of the spring.
- The sum of these two forms of potential energy represents the potential function. Equilibrium requires that the first derivative of the potential function be equal to zero. Stability at the equilibrium position is determined from the second derivative of the potential function.

## Procedure for Analysis

Using potential-energy methods, the equilibrium positions and the stability of a body or a system of connected bodies having a single degree of freedom can be obtained by applying the following procedure.

### Potential Function.

- Sketch the system so that it is in the *arbitrary position* specified by the coordinate  $q$ .
- Establish a horizontal *datum* through a *fixed point*\* and express the gravitational potential energy  $V_g$  in terms of the weight  $W$  of each member and its vertical distance  $y$  from the datum,  $V_g = Wy$ .
- Express the elastic potential energy  $V_e$  of the system in terms of the stretch or compression,  $s$ , of any connecting spring,  $V_e = \frac{1}{2}ks^2$ .
- Formulate the potential function  $V = V_g + V_e$  and express the *position coordinates*  $y$  and  $s$  in terms of the single coordinate  $q$ .

### Equilibrium Position.

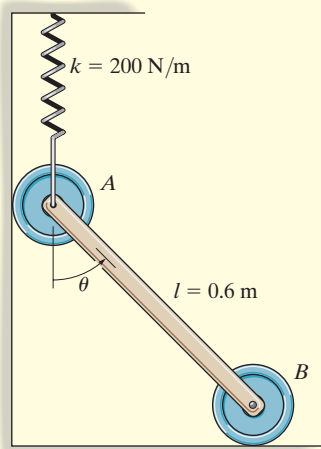
- The equilibrium position of the system is determined by taking the first derivative of  $V$  and setting it equal to zero,  $dV/dq = 0$ .

### Stability.

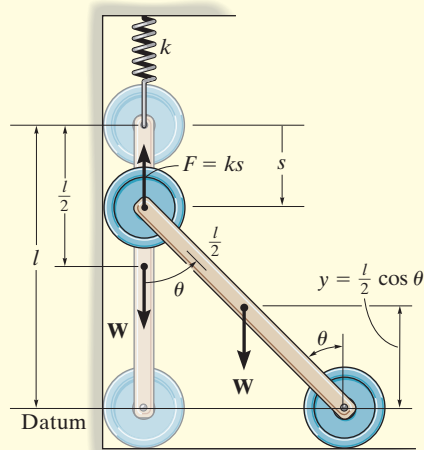
- Stability at the equilibrium position is determined by evaluating the second or higher-order derivatives of  $V$ .
- If the second derivative is greater than zero, the system is stable; if all derivatives are equal to zero, the system is in neutral equilibrium; and if the second derivative is less than zero, the system is unstable.

\*The location of the datum is *arbitrary*, since only the *changes* or differentials of  $V$  are required for investigation of the equilibrium position and its stability.

## EXAMPLE 11.5



(a)



(b)

Fig. 11-17

The uniform link shown in Fig. 11-17a has a mass of 10 kg. If the spring is unstretched when  $\theta = 0^\circ$ , determine the angle  $\theta$  for equilibrium and investigate the stability at the equilibrium position.

## SOLUTION

**Potential Function.** The datum is established at the bottom of the link, Fig. 11-17b. When the link is located in the arbitrary position  $\theta$ , the spring increases its potential energy by stretching and the weight decreases its potential energy. Hence,

$$V = V_e + V_g = \frac{1}{2}ks^2 + W_y$$

Since  $l = s + l \cos \theta$  or  $s = l(1 - \cos \theta)$ , and  $y = (l/2) \cos \theta$ , then

$$V = \frac{1}{2}kl^2(1 - \cos \theta)^2 + W\left(\frac{l}{2} \cos \theta\right)$$

**Equilibrium Position.** The first derivative of  $V$  is

$$\frac{dV}{d\theta} = kl^2(1 - \cos \theta) \sin \theta - \frac{Wl}{2} \sin \theta = 0$$

or

$$l \left[ kl(1 - \cos \theta) - \frac{W}{2} \right] \sin \theta = 0$$

This equation is satisfied provided

$$\sin \theta = 0 \quad \theta = 0^\circ \quad \text{Ans.}$$

or

$$\theta = \cos^{-1} \left( 1 - \frac{W}{2kl} \right) = \cos^{-1} \left[ 1 - \frac{10(9.81)}{2(200)(0.6)} \right] = 53.8^\circ \quad \text{Ans.}$$

**Stability.** The second derivative of  $V$  is

$$\begin{aligned} \frac{d^2V}{d\theta^2} &= kl^2(1 - \cos \theta) \cos \theta + kl^2 \sin \theta \sin \theta - \frac{Wl}{2} \cos \theta \\ &= kl^2(\cos \theta - \cos 2\theta) - \frac{Wl}{2} \cos \theta \end{aligned}$$

Substituting values for the constants, with  $\theta = 0^\circ$  and  $\theta = 53.8^\circ$ , yields

$$\begin{aligned} \left. \frac{d^2V}{d\theta^2} \right|_{\theta=0^\circ} &= 200(0.6)^2(\cos 0^\circ - \cos 0^\circ) - \frac{10(9.81)(0.6)}{2} \cos 0^\circ \\ &= -29.4 < 0 \quad (\text{unstable equilibrium at } \theta = 0^\circ) \quad \text{Ans.} \end{aligned}$$

$$\begin{aligned} \left. \frac{d^2V}{d\theta^2} \right|_{\theta=53.8^\circ} &= 200(0.6)^2(\cos 53.8^\circ - \cos 107.6^\circ) - \frac{10(9.81)(0.6)}{2} \cos 53.8^\circ \\ &= 46.9 > 0 \quad (\text{stable equilibrium at } \theta = 53.8^\circ) \quad \text{Ans.} \end{aligned}$$

**EXAMPLE 11.6**

If the spring  $AD$  in Fig. 11–18a has a stiffness of  $18 \text{ kN/m}$  and is unstretched when  $\theta = 60^\circ$ , determine the angle  $\theta$  for equilibrium. The load has a mass of  $1.5 \text{ Mg}$ . Investigate the stability at the equilibrium position.

**SOLUTION**

**Potential Energy.** The gravitational potential energy for the load with respect to the fixed datum, shown in Fig. 11–18b, is

$$V_g = mgy = 1500(9.81) \text{ N}[(4 \text{ m}) \sin \theta + h] = 58\,860 \sin \theta + 14\,715h$$

where  $h$  is a constant distance. From the geometry of the system, the elongation of the spring when the load is on the platform is  $s = (4 \text{ m}) \cos \theta - (4 \text{ m}) \cos 60^\circ = (4 \text{ m}) \cos \theta - 2 \text{ m}$ .

Thus, the elastic potential energy of the system is

$$V_e = \frac{1}{2}ks^2 = \frac{1}{2}(18\,000 \text{ N/m})(4 \text{ m} \cos \theta - 2 \text{ m})^2 = 9000(4 \cos \theta - 2)^2$$

The potential energy function for the system is therefore

$$V = V_g + V_e = 58\,860 \sin \theta + 14\,715h + 9000(4 \cos \theta - 2)^2 \quad (1)$$

**Equilibrium.** When the system is in equilibrium,

$$\begin{aligned} \frac{dV}{d\theta} &= 58\,860 \cos \theta + 18\,000(4 \cos \theta - 2)(-4 \sin \theta) = 0 \\ 58\,860 \cos \theta - 288\,000 \sin \theta \cos \theta + 144\,000 \sin \theta &= 0 \end{aligned}$$

Since  $\sin 2\theta = 2 \sin \theta \cos \theta$ ,

$$58\,860 \cos \theta - 144\,000 \sin 2\theta + 144\,000 \sin \theta = 0$$

Solving by trial and error,

$$\theta = 28.18^\circ \text{ and } \theta = 45.51^\circ$$

**Stability.** Taking the second derivative of Eq. 1,

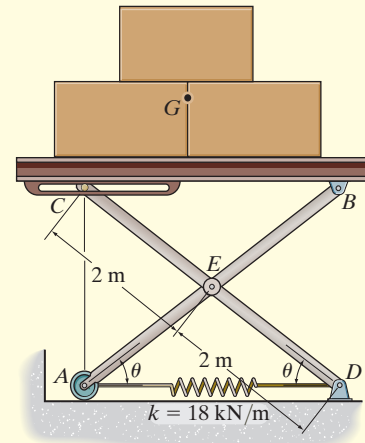
$$\frac{d^2V}{d\theta^2} = -58\,860 \sin \theta - 288\,000 \cos 2\theta + 144\,000 \cos \theta$$

Substituting  $\theta = 28.18^\circ$  yields

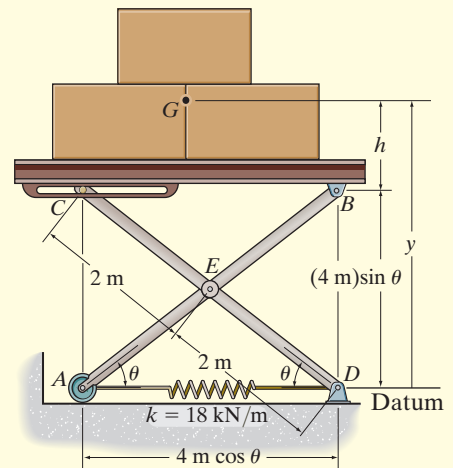
$$\frac{d^2V}{d\theta^2} = -60\,402 < 0 \quad \text{Unstable} \quad \text{Ans.}$$

And for  $\theta = 45.51^\circ$ ,

$$\frac{d^2V}{d\theta^2} = 64\,073 > 0 \quad \text{Stable} \quad \text{Ans.}$$



(a)

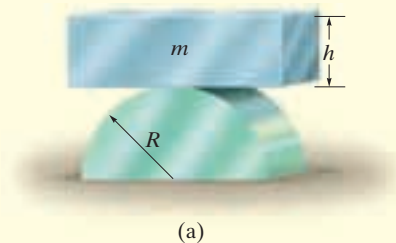


(b)

**Fig. 11–18**



## EXAMPLE 11.7



The uniform block having a mass  $m$  rests on the top surface of the half cylinder, Fig. 11-19a. Show that this is a condition of unstable equilibrium if  $h > 2R$ .

## SOLUTION

**Potential Function.** The datum is established at the base of the cylinder, Fig. 11-19b. If the block is displaced by an amount  $\theta$  from the equilibrium position, the potential function is

$$\begin{aligned} V &= V_e + V_g \\ &= 0 + mgy \end{aligned}$$

From Fig. 11-19b,

$$y = \left(R + \frac{h}{2}\right) \cos \theta + R\theta \sin \theta$$

Thus,

$$V = mg \left[ \left(R + \frac{h}{2}\right) \cos \theta + R\theta \sin \theta \right]$$

**Equilibrium Position.**

$$\begin{aligned} \frac{dV}{d\theta} &= mg \left[ -\left(R + \frac{h}{2}\right) \sin \theta + R \sin \theta + R\theta \cos \theta \right] = 0 \\ &= mg \left( -\frac{h}{2} \sin \theta + R\theta \cos \theta \right) = 0 \end{aligned}$$

Note that  $\theta = 0^\circ$  satisfies this equation.

**Stability.** Taking the second derivative of  $V$  yields

$$\frac{d^2V}{d\theta^2} = mg \left( -\frac{h}{2} \cos \theta + R \cos \theta - R\theta \sin \theta \right)$$

At  $\theta = 0^\circ$ ,

$$\left. \frac{d^2V}{d\theta^2} \right|_{\theta=0^\circ} = -mg \left( \frac{h}{2} - R \right)$$

Since all the constants are positive, the block is in unstable equilibrium provided  $h > 2R$ , because then  $d^2V/d\theta^2 < 0$ .

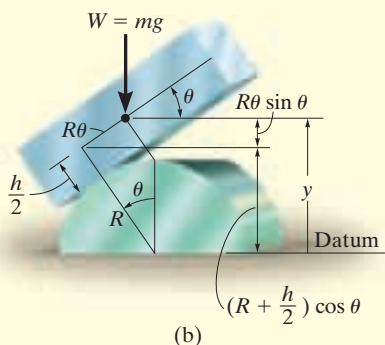


Fig. 11-19

## PROBLEMS

**11-26.** The potential energy of a one-degree-of-freedom system is defined by  $V = (20x^3 - 10x^2 - 25x - 10) \text{ ft} \cdot \text{lb}$ , where  $x$  is in ft. Determine the equilibrium positions and investigate the stability for each position.

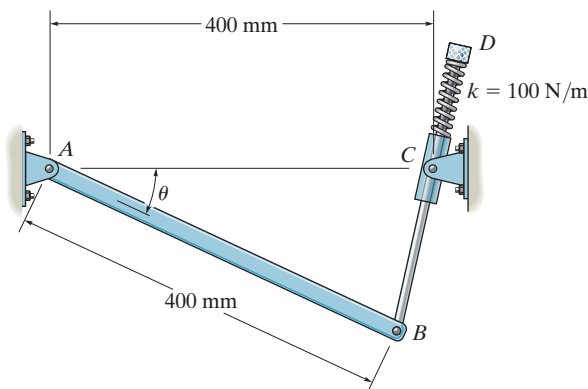
**11-27.** If the potential function for a conservative one-degree-of-freedom system is  $V = (12 \sin 2\theta + 15 \cos \theta) \text{ J}$ , where  $0^\circ < \theta < 180^\circ$ , determine the positions for equilibrium and investigate the stability at each of these positions.

**\*11-28.** If the potential function for a conservative one-degree-of-freedom system is  $V = (8x^3 - 2x^2 - 10) \text{ J}$ , where  $x$  is given in meters, determine the positions for equilibrium and investigate the stability at each of these positions.

**11-29.** If the potential function for a conservative one-degree-of-freedom system is  $V = (10 \cos 2\theta + 25 \sin \theta) \text{ J}$ , where  $0^\circ < \theta < 180^\circ$ , determine the positions for equilibrium and investigate the stability at each of these positions.

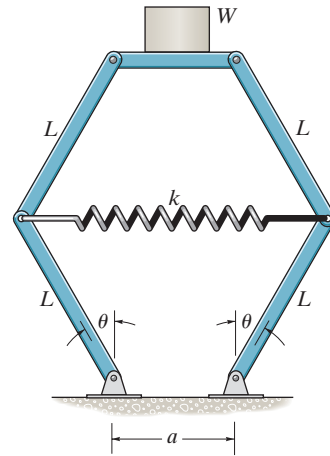
**11-30.** If the potential energy for a conservative one-degree-of-freedom system is expressed by the relation  $V = (4x^3 - x^2 - 3x + 10) \text{ ft} \cdot \text{lb}$ , where  $x$  is given in feet, determine the equilibrium positions and investigate the stability at each position.

**11-31.** The uniform link  $AB$ , has a mass of 3 kg and is pin connected at both of its ends. The rod  $BD$ , having negligible weight, passes through a swivel block at  $C$ . If the spring has a stiffness of  $k = 100 \text{ N/m}$  and is unstretched when  $\theta = 0^\circ$ , determine the angle  $\theta$  for equilibrium and investigate the stability at the equilibrium position. Neglect the size of the swivel block.



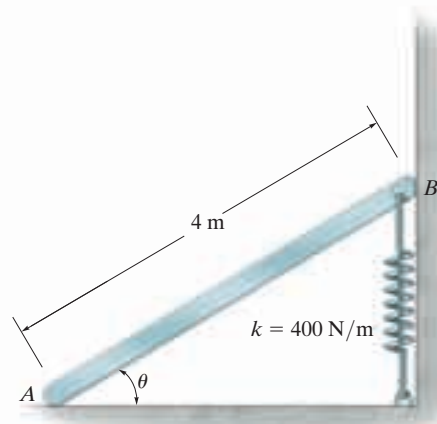
**Prob. 11-31**

**\*11-32.** The spring of the scale has an unstretched length of  $a$ . Determine the angle  $\theta$  for equilibrium when a weight  $W$  is supported on the platform. Neglect the weight of the members. What value  $W$  would be required to keep the scale in neutral equilibrium when  $\theta = 0^\circ$ ?



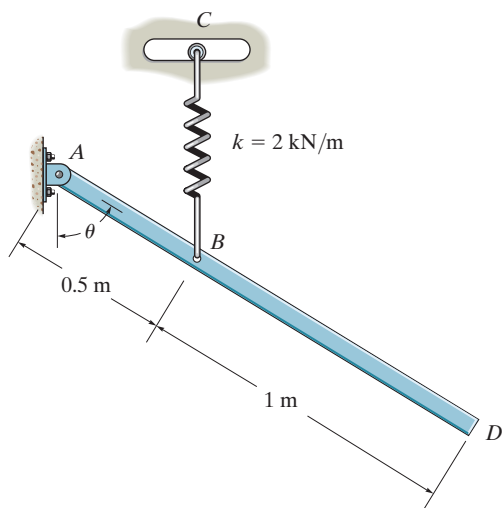
**Prob. 11-32**

**11-33.** The uniform bar has a mass of 80 kg. Determine the angle  $\theta$  for equilibrium and investigate the stability of the bar when it is in this position. The spring has an unstretched length when  $\theta = 90^\circ$ .



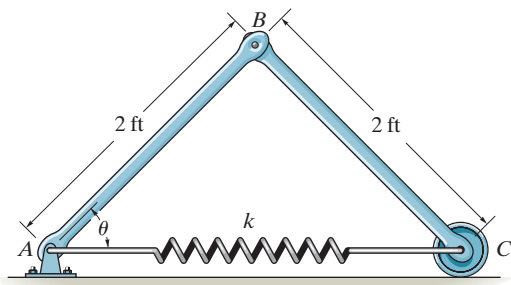
**Prob. 11-33**

**11-34.** The uniform bar  $AD$  has a mass of 20 kg. If the attached spring is unstretched when  $\theta = 90^\circ$ , determine the angle  $\theta$  for equilibrium. Note that the spring always remains in the vertical position due to the roller guide. Investigate the stability of the bar when it is in the equilibrium position.



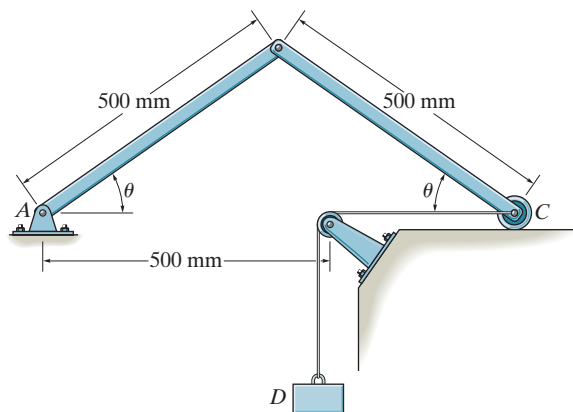
**Prob. 11-34**

**11-35.** The two bars each have a weight of 8 lb. Determine the required stiffness  $k$  of the spring so that the two bars are in equilibrium when  $\theta = 30^\circ$ . The spring has an unstretched length of 1 ft.



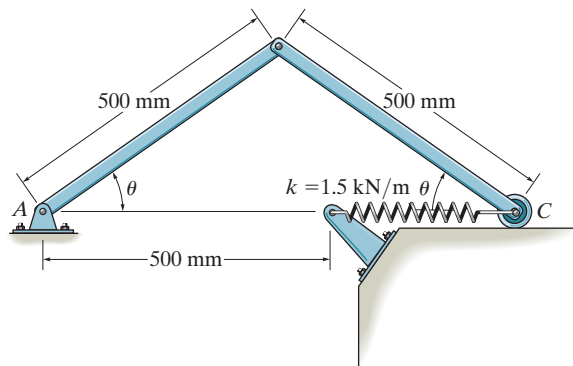
**Prob. 11-35**

**\*11-36.** Determine the angle  $\theta$  for equilibrium and investigate the stability at this position. The bars each have a mass of 3 kg and the suspended block  $D$  has a mass of 7 kg. Cord  $DC$  has a total length of 1 m.



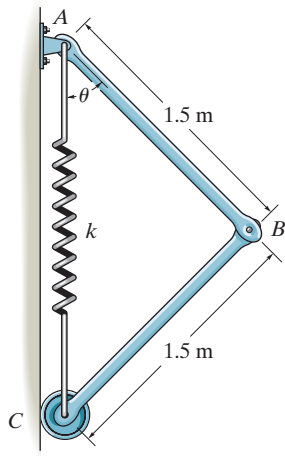
**Prob. 11-36**

**11-37.** Determine the angle  $\theta$  for equilibrium and investigate the stability at this position. The bars each have a mass of 10 kg and the spring has an unstretched length of 100 mm.



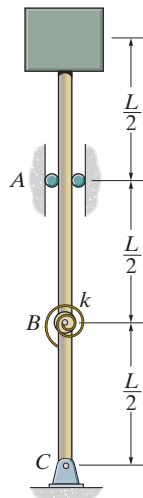
**Prob. 11-37**

**11–38.** The two bars each have a mass of 8 kg. Determine the required stiffness  $k$  of the spring so that the two bars are in equilibrium when  $\theta = 60^\circ$ . The spring has an unstretched length of 1 m. Investigate the stability of the system at the equilibrium position.



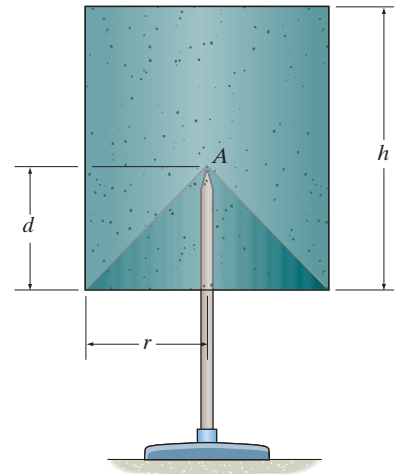
**Prob. 11–38**

**11–39.** A spring with a torsional stiffness  $k$  is attached to the hinge at  $B$ . It is unstretched when the rod assembly is in the vertical position. Determine the weight  $W$  of the block that results in neutral equilibrium. *Hint:* Establish the potential energy function for a small angle  $\theta$ , i.e., approximate  $\sin \theta \approx \theta$ , and  $\cos \theta \approx 1 - \theta^2/2$ .



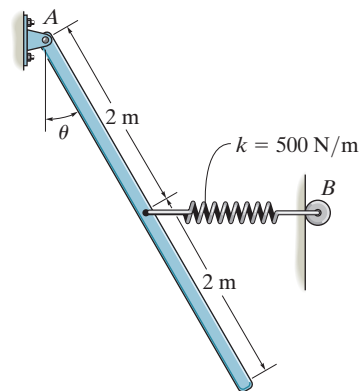
**Prob. 11–39**

**\*11–40.** A conical hole is drilled into the bottom of the cylinder, which is supported on the fulcrum at  $A$ . Determine the minimum distance  $d$  in order for it to remain in stable equilibrium.



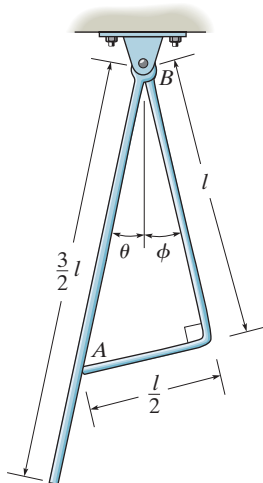
**Prob. 11–40**

**11–41.** The uniform rod has a mass of 100 kg. If the spring is unstretched when  $\theta = 60^\circ$ , determine the angle  $\theta$  for equilibrium and investigate the stability at the equilibrium position. The spring is always in the horizontal position due to the roller guide at  $B$ .



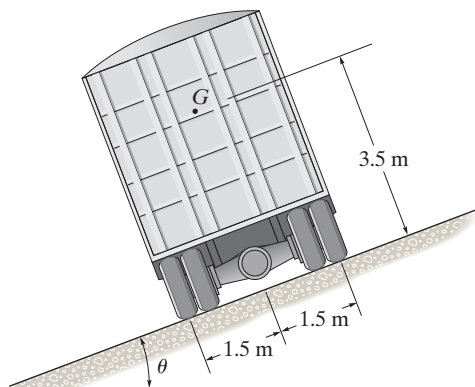
**Prob. 11–41**

**11–42.** Each bar has a mass per length of  $m_0$ . Determine the angles  $\theta$  and  $\phi$  at which they are suspended in equilibrium. The contact at  $A$  is smooth, and both are pin connected at  $B$ .



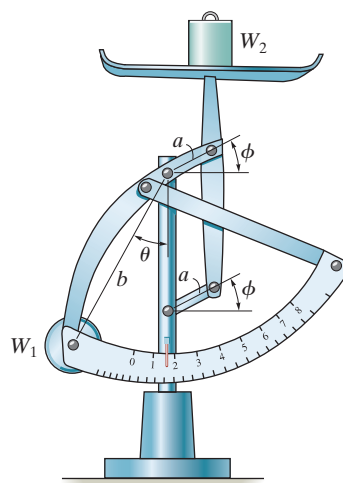
**Prob. 11–42**

**11–43.** The truck has a mass of  $20\text{ Mg}$  and a mass center at  $G$ . Determine the steepest grade  $\theta$  along which it can park without overturning and investigate the stability in this position.



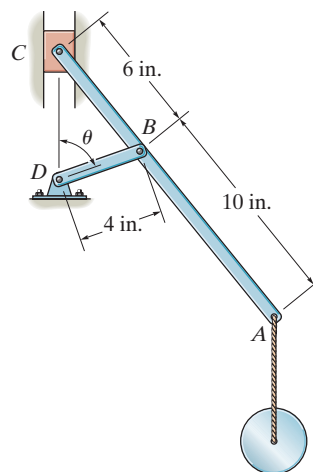
**Prob. 11–43**

**\*11–44.** The small postal scale consists of a counterweight  $W_1$ , connected to the members having negligible weight. Determine the weight  $W_2$  that is on the pan in terms of the angles  $\theta$  and  $\phi$  and the dimensions shown. All members are pin connected.



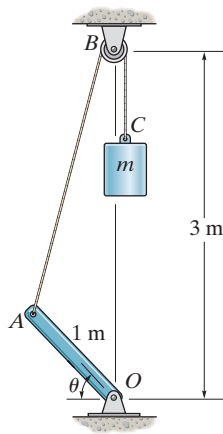
**Prob. 11–44**

**11–45.** A  $3\text{-lb}$  weight is attached to the end of rod  $ABC$ . If the rod is supported by a smooth slider block at  $C$  and rod  $BD$ , determine the angle  $\theta$  for equilibrium. Neglect the weight of the rods and the slider.



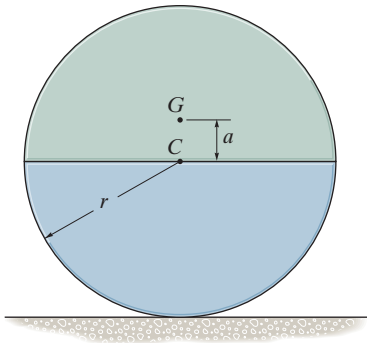
**Prob. 11–45**

**11-46.** If the uniform rod  $OA$  has a mass of 12 kg, determine the mass  $m$  that will hold the rod in equilibrium when  $\theta = 30^\circ$ . Point  $C$  is coincident with  $B$  when  $OA$  is horizontal. Neglect the size of the pulley at  $B$ .



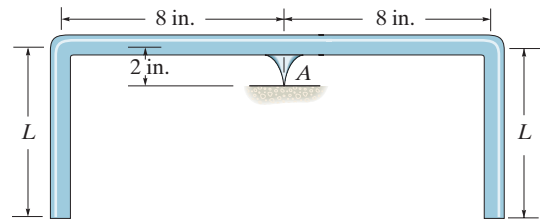
**Prob. 11-46**

**11-47.** The cylinder is made of two materials such that it has a mass of  $m$  and a center of gravity at point  $G$ . Show that when  $G$  lies above the centroid  $C$  of the cylinder, the equilibrium is unstable.



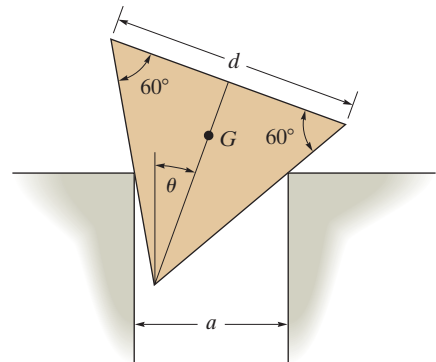
**Prob. 11-47**

**\*11-48.** The bent rod has a weight of 5 lb/ft. A pivot is attached at its center  $A$  and the rod is balanced as shown. Determine the length  $L$  of its vertical segments so that it remains in neutral equilibrium. Neglect the thickness of the rod.



**Prob. 11-48**

**11-49.** The triangular block of weight  $W$  rests on the smooth corners which are a distance  $a$  apart. If the block has three equal sides of length  $d$ , determine the angle  $\theta$  for equilibrium.



**Prob. 11-49**

## CHAPTER REVIEW

**Principle of Virtual Work**

The forces on a body will do *virtual work* when the body undergoes an *imaginary* differential displacement or rotation.

For equilibrium, the sum of the virtual work done by all the forces acting on the body must be equal to zero for any virtual displacement. This is referred to as the *principle of virtual work*, and it is useful for finding the equilibrium configuration for a mechanism or a reactive force acting on a series of connected members.

If the system of connected members has one degree of freedom, then its position can be specified by one independent coordinate, such as  $\theta$ .

To apply the principle of virtual work, it is first necessary to use *position coordinates* to locate all the forces and moments on the mechanism that will do work when the mechanism undergoes a virtual movement  $\delta\theta$ .

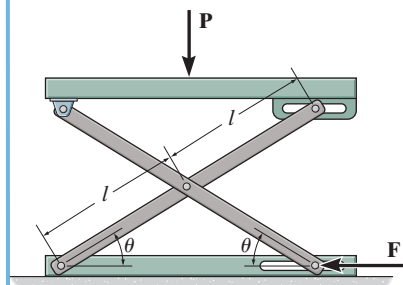
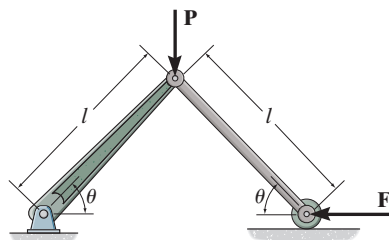
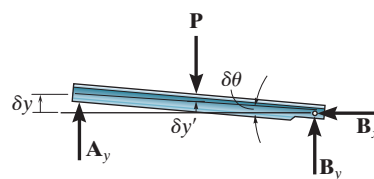
The coordinates are related to the independent coordinate  $\theta$  and then these expressions are differentiated in order to relate the *virtual* coordinate displacements to the virtual displacement  $\delta\theta$ .

Finally, the equation of virtual work is written for the mechanism in terms of the common virtual displacement  $\delta\theta$ , and then it is set equal to zero. By factoring  $\delta\theta$  out of the equation, it is then possible to determine either the unknown force or couple moment, or the equilibrium position  $\theta$ .

$\delta y, \delta y'$ —virtual displacements

$\delta\theta$ —virtual rotation

$$\delta U = 0$$



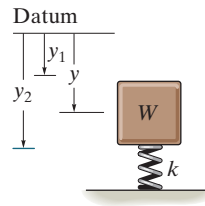
### Potential-Energy Criterion for Equilibrium

When a system is subjected only to conservative forces, such as weight and spring forces, then the equilibrium configuration can be determined using the *potential-energy function*  $V$  for the system.

The potential-energy function is established by expressing the weight and spring potential energy for the system in terms of the independent coordinate  $q$ .

Once the potential-energy function is formulated, its first derivative is set equal to zero. The solution yields the equilibrium position  $q_{eq}$  for the system.

The stability of the system can be investigated by taking the second derivative of  $V$ .



$$V = V_g + V_e = -Wy + \frac{1}{2}ky^2$$

$$\frac{dV}{dq} = 0$$

$$\frac{dV}{dq} = 0, \quad \frac{d^2V}{dq^2} > 0 \quad \text{stable equilibrium}$$

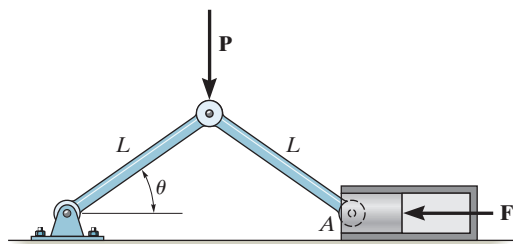
$$\frac{dV}{dq} = 0, \quad \frac{d^2V}{dq^2} < 0 \quad \text{unstable equilibrium}$$

$$\frac{dV}{dq} = \frac{d^2V}{dq^2} = \frac{d^3V}{dq^3} = \dots = 0 \quad \text{neutral equilibrium}$$



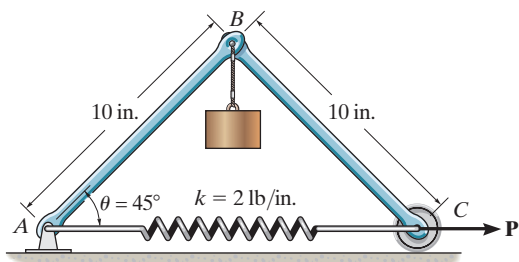
## REVIEW PROBLEMS

**R11-1.** The toggle joint is subjected to the load  $P$ . Determine the compressive force  $F$  it creates on the cylinder at  $A$  as a function of  $\theta$ .



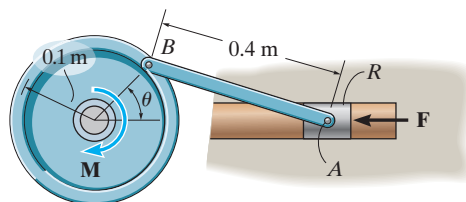
**Prob. R11-1**

**R11-2.** The uniform links  $AB$  and  $BC$  each weigh 2 lb and the cylinder weighs 20 lb. Determine the horizontal force  $P$  required to hold the mechanism in the position when  $\theta = 45^\circ$ . The spring has an unstretched length of 6 in.



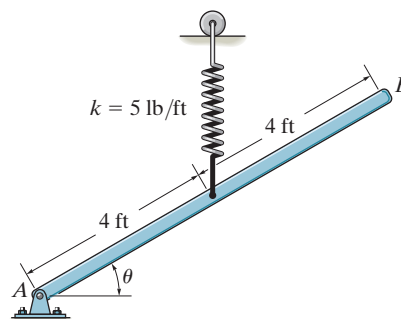
**Prob. R11-2**

**R11-3.** The punch press consists of the ram  $R$ , connecting rod  $AB$ , and a flywheel. If a torque of  $M = 50 \text{ N} \cdot \text{m}$  is applied to the flywheel, determine the force  $F$  applied at the ram to hold the rod in the position  $\theta = 60^\circ$ .



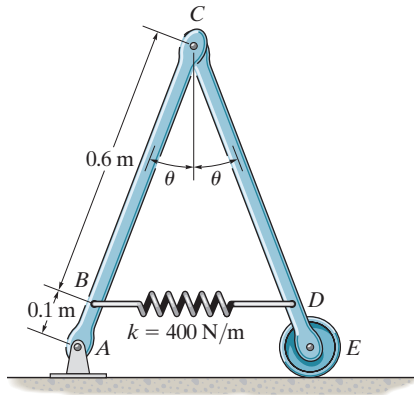
**Prob. R11-3**

**R11-4.** The uniform bar  $AB$  weighs 10 lb. If the attached spring is unstretched when  $\theta = 90^\circ$ , use the method of virtual work and determine the angle  $\theta$  for equilibrium. Note that the spring always remains in the vertical position due to the roller guide.



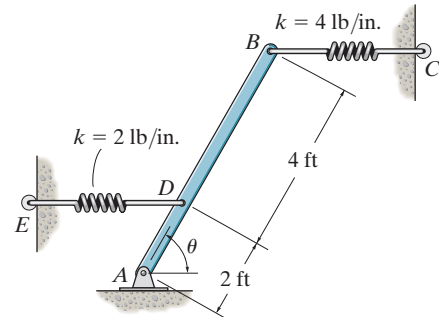
**Prob. R11-4**

**R11-5.** The spring has an unstretched length of 0.3 m. Determine the angle  $\theta$  for equilibrium if the uniform links each have a mass of 5 kg.



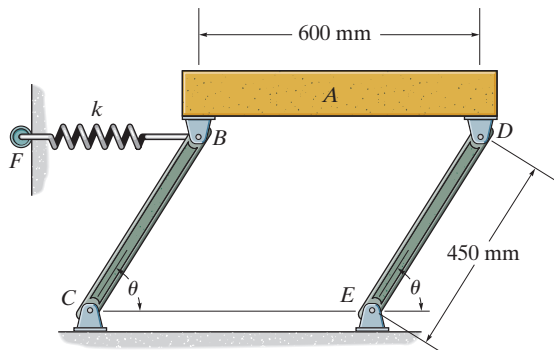
**Prob. R11-5**

**R11-7.** The uniform bar  $AB$  weighs 100 lb. If both springs  $DE$  and  $BC$  are unstretched when  $\theta = 90^\circ$ , determine the angle  $\theta$  for equilibrium using the principle of potential energy. Investigate the stability at the equilibrium position. Both springs always act in the horizontal position because of the roller guides at  $C$  and  $E$ .



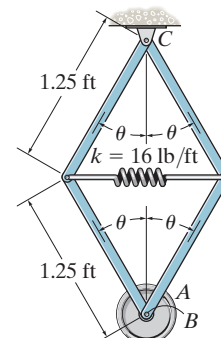
**Prob. R11-7**

**R11-6.** Determine the angle  $\theta$  for equilibrium and investigate the stability of the mechanism in this position. The spring has a stiffness of  $k = 1.5 \text{ kN/m}$  and is unstretched when  $\theta = 90^\circ$ . The block  $A$  has a mass of 40 kg. Neglect the mass of the links.



**Prob. R11-6**

**R11-8.** The spring attached to the mechanism has an unstretched length when  $\theta = 90^\circ$ . Determine the position  $\theta$  for equilibrium and investigate the stability of the mechanism at this position. Disk  $A$  is pin connected to the frame at  $B$  and has a weight of 20 lb. Neglect the weight of the bars.



**Prob. R11-8**

## A

# Mathematical Review and Expressions

## Geometry and Trigonometry Review

The angles  $\theta$  in Fig. A-1 are equal between the transverse and two parallel lines.

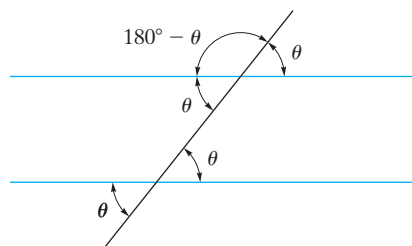


Fig. A-1

For a line and its normal, the angles  $\theta$  in Fig. A-2 are equal.

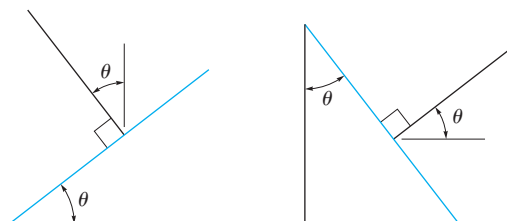


Fig. A-2

For the circle in Fig. A-3,  $s = \theta r$ , so that when  $\theta = 360^\circ = 2\pi$  rad then the circumference is  $s = 2\pi r$ . Also, since  $180^\circ = \pi$  rad, then  $\theta$  (rad)  $= (\pi/180^\circ)\theta^\circ$ . The area of the circle is  $A = \pi r^2$ .

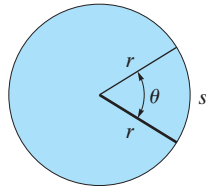


Fig. A-3

The sides of a similar triangle can be obtained by proportion as in Fig. A-4, where  $\frac{a}{A} = \frac{b}{B} = \frac{c}{C}$ .

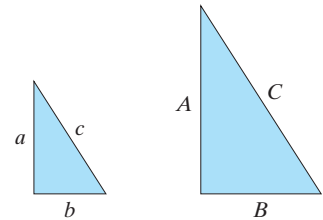


Fig. A-4

For the right triangle in Fig. A-5, the Pythagorean theorem is

$$h = \sqrt{o^2 + a^2}$$

The trigonometric functions are

$$\sin \theta = \frac{o}{h}$$

$$\cos \theta = \frac{a}{h}$$

$$\tan \theta = \frac{o}{a}$$

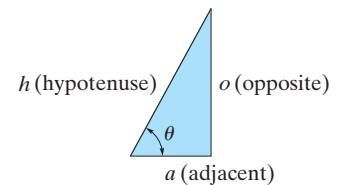


Fig. A-5

This is easily remembered as “soh, cah, toa” i.e., the sine is the opposite over the hypotenuse, etc. The other trigonometric functions follow from this.

$$\csc \theta = \frac{1}{\sin \theta} = \frac{h}{o}$$

$$\sec \theta = \frac{1}{\cos \theta} = \frac{h}{a}$$

$$\cot \theta = \frac{1}{\tan \theta} = \frac{a}{o}$$

## Trigonometric Identities

$$\sin^2 \theta + \cos^2 \theta = 1$$

$$\sin(\theta \pm \phi) = \sin \theta \cos \phi \pm \cos \theta \sin \phi$$

$$\sin 2\theta = 2 \sin \theta \cos \theta$$

$$\cos(\theta \pm \phi) = \cos \theta \cos \phi \mp \sin \theta \sin \phi$$

$$\cos 2\theta = \cos^2 \theta - \sin^2 \theta$$

$$\cos \theta = \pm \sqrt{\frac{1 + \cos 2\theta}{2}}, \sin \theta = \pm \sqrt{\frac{1 - \cos 2\theta}{2}}$$

$$\tan \theta = \frac{\sin \theta}{\cos \theta}$$

$$1 + \tan^2 \theta = \sec^2 \theta \quad 1 + \cot^2 \theta = \csc^2 \theta$$

## Quadratic Formula

$$\text{If } ax^2 + bx + c = 0, \text{ then } x = \frac{-b \pm \sqrt{b^2 - 4ac}}{2a}$$

## Hyperbolic Functions

$$\sinh x = \frac{e^x - e^{-x}}{2},$$

$$\cosh x = \frac{e^x + e^{-x}}{2},$$

$$\tanh x = \frac{\sinh x}{\cosh x}$$

## Power-Series Expansions

$$\sin x = x - \frac{x^3}{3!} + \cdots, \cos x = 1 - \frac{x^2}{2!} + \cdots$$

$$\sinh x = x + \frac{x^3}{3!} + \cdots, \cosh x = 1 + \frac{x^2}{2!} + \cdots$$

## Derivatives

$$\frac{d}{dx}(u^n) = nu^{n-1} \frac{du}{dx} \quad \frac{d}{dx}(\sin u) = \cos u \frac{du}{dx}$$

$$\frac{d}{dx}(uv) = u \frac{dv}{dx} + v \frac{du}{dx} \quad \frac{d}{dx}(\cos u) = -\sin u \frac{du}{dx}$$

$$\frac{d}{dx}\left(\frac{u}{v}\right) = \frac{v \frac{du}{dx} - u \frac{dv}{dx}}{v^2} \quad \frac{d}{dx}(\tan u) = \sec^2 u \frac{du}{dx}$$

$$\frac{d}{dx}(\cot u) = -\csc^2 u \frac{du}{dx} \quad \frac{d}{dx}(\sinh u) = \cosh u \frac{du}{dx}$$

$$\frac{d}{dx}(\sec u) = \tan u \sec u \frac{du}{dx} \quad \frac{d}{dx}(\cosh u) = \sinh u \frac{du}{dx}$$

$$\frac{d}{dx}(\csc u) = -\csc u \cot u \frac{du}{dx}$$

## Integrals

$$\int x^n dx = \frac{x^{n+1}}{n+1} + C, n \neq -1$$

$$\int \frac{dx}{a+bx} = \frac{1}{b} \ln(a+bx) + C$$

$$\int \frac{dx}{a+bx^2} = \frac{1}{2\sqrt{-ab}} \ln \left[ \frac{a+x\sqrt{-ab}}{a-x\sqrt{-ab}} \right] + C,$$

$$ab < 0$$

$$\int \frac{x dx}{a+bx^2} = \frac{1}{2b} \ln(bx^2+a) + C$$

$$\int \frac{x^2 dx}{a+bx^2} = \frac{x}{b} - \frac{a}{b\sqrt{ab}} \tan^{-1} \frac{x\sqrt{ab}}{a} + C, ab > 0$$

$$\int \sqrt{a+bx} dx = \frac{2}{3b} \sqrt{(a+bx)^3} + C$$

$$\int x\sqrt{a+bx} dx = \frac{-2(2a-3bx)\sqrt{(a+bx)^3}}{15b^2} + C$$

$$\int x^2\sqrt{a+bx} dx =$$

$$\frac{2(8a^2-12abx+15b^2x^2)\sqrt{(a+bx)^3}}{105b^3} + C$$

$$\int \sqrt{a^2-x^2} dx = \frac{1}{2} \left[ x\sqrt{a^2-x^2} + a^2 \sin^{-1} \frac{x}{a} \right] + C,$$

$$a > 0$$

$$\int x\sqrt{a^2-x^2} dx = -\frac{1}{3} \sqrt{(a^2-x^2)^3} + C$$

$$\int x^2\sqrt{a^2-x^2} dx = -\frac{x}{4} \sqrt{(a^2-x^2)^3}$$

$$+ \frac{a^2}{8} \left( x\sqrt{a^2-x^2} + a^2 \sin^{-1} \frac{x}{a} \right) + C, a > 0$$

$$\int \sqrt{x^2 \pm a^2} dx =$$

$$\frac{1}{2} \left[ x\sqrt{x^2 \pm a^2} \pm a^2 \ln(x + \sqrt{x^2 \pm a^2}) \right] + C$$

$$\int x\sqrt{x^2 \pm a^2} dx = \frac{1}{3} \sqrt{(x^2 \pm a^2)^3} + C$$

$$\int x^2\sqrt{x^2 \pm a^2} dx = \frac{x}{4} \sqrt{(x^2 \pm a^2)^3}$$

$$\mp \frac{a^2}{8} x\sqrt{x^2 \pm a^2} - \frac{a^4}{8} \ln(x + \sqrt{x^2 \pm a^2}) + C$$

$$\int \frac{dx}{\sqrt{a+bx}} = \frac{2\sqrt{a+bx}}{b} + C$$

$$\int \frac{x dx}{\sqrt{x^2 \pm a^2}} = \sqrt{x^2 \pm a^2} + C$$

$$\int \frac{dx}{\sqrt{a+bx+cx^2}} = \frac{1}{\sqrt{c}} \ln \left[ \sqrt{a+bx+cx^2} + \right.$$

$$\left. x\sqrt{c} + \frac{b}{2\sqrt{c}} \right] + C, c > 0$$

$$= \frac{1}{\sqrt{-c}} \sin^{-1} \left( \frac{-2cx-b}{\sqrt{b^2-4ac}} \right) + C, c < 0$$

$$\int \sin x dx = -\cos x + C$$

$$\int \cos x dx = \sin x + C$$

$$\int x \cos(ax) dx = \frac{1}{a^2} \cos(ax) + \frac{x}{a} \sin(ax) + C$$

$$\int x^2 \cos(ax) dx = \frac{2x}{a^2} \cos(ax) + \frac{a^2 x^2 - 2}{a^3} \sin(ax) + C$$

$$\int e^{ax} dx = \frac{1}{a} e^{ax} + C$$

$$\int x e^{ax} dx = \frac{e^{ax}}{a^2} (ax-1) + C$$

$$\int \sinh x dx = \cosh x + C$$

$$\int \cosh x dx = \sinh x + C$$

# Fundamental Problems Partial Solutions And Answers

## Chapter 2

### F2-1.

$$F_R = \sqrt{(2 \text{ kN})^2 + (6 \text{ kN})^2 - 2(2 \text{ kN})(6 \text{ kN}) \cos 105^\circ}$$

$$= 6.798 \text{ kN} = 6.80 \text{ kN}$$

Ans.

$$\frac{\sin \phi}{6 \text{ kN}} = \frac{\sin 105^\circ}{6.798 \text{ kN}}, \quad \phi = 58.49^\circ$$

$$\theta = 45^\circ + \phi = 45^\circ + 58.49^\circ = 103^\circ \quad \text{Ans.}$$

### F2-2.

$$F_R = \sqrt{200^2 + 500^2 - 2(200)(500) \cos 140^\circ}$$

$$= 666 \text{ N}$$

Ans.

### F2-3.

$$F_R = \sqrt{600^2 + 800^2 - 2(600)(800) \cos 60^\circ}$$

$$= 721.11 \text{ N} = 721 \text{ N}$$

Ans.

$$\frac{\sin \alpha}{800} = \frac{\sin 60^\circ}{721.11}; \quad \alpha = 73.90^\circ$$

$$\phi = \alpha - 30^\circ = 73.90^\circ - 30^\circ = 43.9^\circ \quad \text{Ans.}$$

### F2-4.

$$\frac{F_u}{\sin 45^\circ} = \frac{30}{\sin 105^\circ}; \quad F_u = 22.0 \text{ lb}$$

Ans.

$$\frac{F_v}{\sin 30^\circ} = \frac{30}{\sin 105^\circ}; \quad F_v = 15.5 \text{ lb}$$

Ans.

### F2-5.

$$\frac{F_{AB}}{\sin 105^\circ} = \frac{450}{\sin 30^\circ}$$

$$F_{AB} = 869 \text{ lb}$$

Ans.

$$\frac{F_{AC}}{\sin 45^\circ} = \frac{450}{\sin 30^\circ}$$

$$F_{AC} = 636 \text{ lb}$$

Ans.

### F2-6.

$$\frac{F}{\sin 30^\circ} = \frac{6}{\sin 105^\circ} \quad F = 3.11 \text{ kN}$$

Ans.

$$\frac{F_v}{\sin 45^\circ} = \frac{6}{\sin 105^\circ} \quad F_v = 4.39 \text{ kN}$$

Ans.

### F2-7.

$$(F_1)_x = 0 \quad (F_1)_y = 300 \text{ N}$$

Ans.

$$(F_2)_x = -(450 \text{ N}) \cos 45^\circ = -318 \text{ N}$$

Ans.

$$(F_2)_y = (450 \text{ N}) \sin 45^\circ = 318 \text{ N}$$

Ans.

$$(F_3)_x = \left(\frac{3}{5}\right)600 \text{ N} = 360 \text{ N}$$

Ans.

$$(F_3)_y = \left(\frac{4}{5}\right)600 \text{ N} = 480 \text{ N}$$

Ans.

### F2-8.

$$F_{Rx} = 300 + 400 \cos 30^\circ - 250\left(\frac{4}{5}\right) = 446.4 \text{ N}$$

$$F_{Ry} = 400 \sin 30^\circ + 250\left(\frac{3}{5}\right) = 350 \text{ N}$$

$$F_R = \sqrt{(446.4)^2 + 350^2} = 567 \text{ N} \quad \text{Ans.}$$

$$\theta = \tan^{-1} \frac{350}{446.4} = 38.1^\circ \quad \text{Ans.}$$

### F2-9.

$$\rightarrow (F_R)_x = \Sigma F_x;$$

$$(F_R)_x = -(700 \text{ lb}) \cos 30^\circ + 0 + \left(\frac{3}{5}\right)(600 \text{ lb})$$

$$= -246.22 \text{ lb}$$

$$+\uparrow (F_R)_y = \Sigma F_y;$$

$$(F_R)_y = -(700 \text{ lb}) \sin 30^\circ - 400 \text{ lb} - \left(\frac{4}{5}\right)(600 \text{ lb})$$

$$= -1230 \text{ lb}$$

$$F_R = \sqrt{(246.22 \text{ lb})^2 + (1230 \text{ lb})^2} = 1254 \text{ lb} \quad \text{Ans.}$$

$$\phi = \tan^{-1} \left( \frac{1230 \text{ lb}}{246.22 \text{ lb}} \right) = 78.68^\circ$$

$$\theta = 180^\circ + \phi = 180^\circ + 78.68^\circ = 259^\circ \quad \text{Ans.}$$

### F2-10.

$$\rightarrow (F_R)_x = \Sigma F_x;$$

$$750 \text{ N} = F \cos \theta + \left(\frac{5}{13}\right)(325 \text{ N}) + (600 \text{ N}) \cos 45^\circ$$

$$+\uparrow (F_R)_y = \Sigma F_y;$$

$$0 = F \sin \theta + \left(\frac{12}{13}\right)(325 \text{ N}) - (600 \text{ N}) \sin 45^\circ$$

$$\tan \theta = 0.6190 \quad \theta = 31.76^\circ = 31.8^\circ \quad \text{Ans.}$$

$$F = 236 \text{ N} \quad \text{Ans.}$$

### F2-11.

$$\rightarrow (F_R)_x = \Sigma F_x;$$

$$(80 \text{ lb}) \cos 45^\circ = F \cos \theta + 50 \text{ lb} - \left(\frac{3}{5}\right)90 \text{ lb}$$

$$+\uparrow (F_R)_y = \Sigma F_y;$$

$$-(80 \text{ lb}) \sin 45^\circ = F \sin \theta - \left(\frac{4}{5}\right)90 \text{ lb}$$

$$\tan \theta = 0.2547 \quad \theta = 14.29^\circ = 14.3^\circ \quad \text{Ans.}$$

$$F = 62.5 \text{ lb} \quad \text{Ans.}$$

### F2-12.

$$(F_R)_x = 15\left(\frac{4}{5}\right) + 0 + 15\left(\frac{4}{5}\right) = 24 \text{ kN} \rightarrow$$

$$(F_R)_y = 15\left(\frac{3}{5}\right) + 20 - 15\left(\frac{3}{5}\right) = 20 \text{ kN} \uparrow$$

$$F_R = 31.2 \text{ kN} \quad \text{Ans.}$$

$$\theta = 39.8^\circ \quad \text{Ans.}$$

### F2-13.

$$F_x = 75 \cos 30^\circ \sin 45^\circ = 45.93 \text{ lb}$$

$$F_y = 75 \cos 30^\circ \cos 45^\circ = 45.93 \text{ lb}$$

$$F_z = -75 \sin 30^\circ = -37.5 \text{ lb}$$

$$\alpha = \cos^{-1} \left( \frac{45.93}{75} \right) = 52.2^\circ \quad \text{Ans.}$$

$$\beta = \cos^{-1} \left( \frac{45.93}{75} \right) = 52.2^\circ \quad \text{Ans.}$$

$$\gamma = \cos^{-1} \left( \frac{-37.5}{75} \right) = 120^\circ \quad \text{Ans.}$$

**F2-14.**  $\cos \beta = \sqrt{1 - \cos^2 120^\circ - \cos^2 60^\circ} = \pm 0.7071$   
Require  $\beta = 135^\circ$ .

$\mathbf{F} = F\mathbf{u}_F = (500 \text{ N})(-0.5\mathbf{i} - 0.7071\mathbf{j} + 0.5\mathbf{k})$   
 $= \{-250\mathbf{i} - 354\mathbf{j} + 250\mathbf{k}\} \text{ N}$  *Ans.*

**F2-15.**  $\cos^2 \alpha + \cos^2 135^\circ + \cos^2 120^\circ = 1$   
 $\alpha = 60^\circ$   
 $\mathbf{F} = F\mathbf{u}_F = (500 \text{ N})(0.5\mathbf{i} - 0.7071\mathbf{j} - 0.5\mathbf{k})$   
 $= \{250\mathbf{i} - 354\mathbf{j} - 250\mathbf{k}\} \text{ N}$  *Ans.*

**F2-16.**  $F_z = (50 \text{ lb}) \sin 45^\circ = 35.36 \text{ lb}$   
 $F' = (50 \text{ lb}) \cos 45^\circ = 35.36 \text{ lb}$   
 $F_x = \left(\frac{3}{5}\right)(35.36 \text{ lb}) = 21.21 \text{ lb}$   
 $F_y = \left(\frac{4}{5}\right)(35.36 \text{ lb}) = 28.28 \text{ lb}$   
 $\mathbf{F} = \{-21.2\mathbf{i} + 28.3\mathbf{j} + 35.4\mathbf{k}\} \text{ lb}$  *Ans.*

**F2-17.**  $F_z = (750 \text{ N}) \sin 45^\circ = 530.33 \text{ N}$   
 $F' = (750 \text{ N}) \cos 45^\circ = 530.33 \text{ N}$   
 $F_x = (530.33 \text{ N}) \cos 60^\circ = 265.2 \text{ N}$   
 $F_y = (530.33 \text{ N}) \sin 60^\circ = 459.3 \text{ N}$   
 $\mathbf{F}_2 = \{265\mathbf{i} - 459\mathbf{j} + 530\mathbf{k}\} \text{ N}$  *Ans.*

**F2-18.**  $\mathbf{F}_1 = \left(\frac{4}{5}\right)(500 \text{ lb})\mathbf{j} + \left(\frac{3}{5}\right)(500 \text{ lb})\mathbf{k}$   
 $= \{400\mathbf{j} + 300\mathbf{k}\} \text{ lb}$   
 $\mathbf{F}_2 = [(800 \text{ lb}) \cos 45^\circ] \cos 30^\circ \mathbf{i}$   
 $+ [(800 \text{ lb}) \cos 45^\circ] \sin 30^\circ \mathbf{j}$   
 $+ (800 \text{ lb}) \sin 45^\circ (-\mathbf{k})$   
 $= \{489.90\mathbf{i} + 282.84\mathbf{j} - 565.69\mathbf{k}\} \text{ lb}$   
 $\mathbf{F}_R = \mathbf{F}_1 + \mathbf{F}_2 = \{490\mathbf{i} + 683\mathbf{j} - 266\mathbf{k}\} \text{ lb}$  *Ans.*

**F2-19.**  $\mathbf{r}_{AB} = \{-6\mathbf{i} + 6\mathbf{j} + 3\mathbf{k}\} \text{ m}$  *Ans.*  
 $r_{AB} = \sqrt{(-6 \text{ m})^2 + (6 \text{ m})^2 + (3 \text{ m})^2} = 9 \text{ m}$  *Ans.*  
 $\alpha = 132^\circ, \quad \beta = 48.2^\circ, \quad \gamma = 70.5^\circ$  *Ans.*

**F2-20.**  $\mathbf{r}_{AB} = \{-4\mathbf{i} + 2\mathbf{j} + 4\mathbf{k}\} \text{ ft}$  *Ans.*  
 $r_{AB} = \sqrt{(-4 \text{ ft})^2 + (2 \text{ ft})^2 + (4 \text{ ft})^2} = 6 \text{ ft}$  *Ans.*  
 $\alpha = \cos^{-1}\left(\frac{-4 \text{ ft}}{6 \text{ ft}}\right) = 131.8^\circ$   
 $\theta = 180^\circ - 131.8^\circ = 48.2^\circ$  *Ans.*

**F2-21.**  $\mathbf{r}_{AB} = \{2\mathbf{i} + 3\mathbf{j} - 6\mathbf{k}\} \text{ m}$   
 $\mathbf{F}_{AB} = F_{AB}\mathbf{u}_{AB}$   
 $= (630 \text{ N})\left(\frac{2}{7}\mathbf{i} + \frac{3}{7}\mathbf{j} - \frac{6}{7}\mathbf{k}\right)$   
 $= \{180\mathbf{i} + 270\mathbf{j} - 540\mathbf{k}\} \text{ N}$  *Ans.*

**F2-22.**  $\mathbf{F} = F\mathbf{u}_{AB} = 900 \text{ N}\left(-\frac{4}{9}\mathbf{i} + \frac{7}{9}\mathbf{j} - \frac{4}{9}\mathbf{k}\right)$   
 $= \{-400\mathbf{i} + 700\mathbf{j} - 400\mathbf{k}\} \text{ N}$  *Ans.*

**F2-23.**  $\mathbf{F}_B = F_B\mathbf{u}_B$   
 $= (840 \text{ N})\left(\frac{3}{7}\mathbf{i} - \frac{2}{7}\mathbf{j} - \frac{6}{7}\mathbf{k}\right)$   
 $= \{360\mathbf{i} - 240\mathbf{j} - 720\mathbf{k}\} \text{ N}$   
 $\mathbf{F}_C = F_C\mathbf{u}_C$   
 $= (420 \text{ N})\left(\frac{2}{7}\mathbf{i} + \frac{3}{7}\mathbf{j} - \frac{6}{7}\mathbf{k}\right)$   
 $= \{120\mathbf{i} + 180\mathbf{j} - 360\mathbf{k}\} \text{ N}$   
 $F_R = \sqrt{(480 \text{ N})^2 + (-60 \text{ N})^2 + (-1080 \text{ N})^2}$   
 $= 1.18 \text{ kN}$  *Ans.*

**F2-24.**  $\mathbf{F}_B = F_B\mathbf{u}_B$   
 $= (600 \text{ lb})\left(-\frac{1}{3}\mathbf{i} + \frac{2}{3}\mathbf{j} - \frac{2}{3}\mathbf{k}\right)$   
 $= \{-200\mathbf{i} + 400\mathbf{j} - 400\mathbf{k}\} \text{ lb}$   
 $\mathbf{F}_C = F_C\mathbf{u}_C$   
 $= (490 \text{ lb})\left(-\frac{6}{7}\mathbf{i} + \frac{3}{7}\mathbf{j} - \frac{2}{7}\mathbf{k}\right)$   
 $= \{-420\mathbf{i} + 210\mathbf{j} - 140\mathbf{k}\} \text{ lb}$   
 $\mathbf{F}_R = \mathbf{F}_B + \mathbf{F}_C = \{-620\mathbf{i} + 610\mathbf{j} - 540\mathbf{k}\} \text{ lb}$  *Ans.*

**F2-25.**  $\mathbf{u}_{AO} = -\frac{1}{3}\mathbf{i} + \frac{2}{3}\mathbf{j} - \frac{2}{3}\mathbf{k}$   
 $\mathbf{u}_F = -0.5345\mathbf{i} + 0.8018\mathbf{j} + 0.2673\mathbf{k}$   
 $\theta = \cos^{-1}(\mathbf{u}_{AO} \cdot \mathbf{u}_F) = 57.7^\circ$  *Ans.*

**F2-26.**  $\mathbf{u}_{AB} = -\frac{3}{5}\mathbf{j} + \frac{4}{5}\mathbf{k}$   
 $\mathbf{u}_F = \frac{4}{5}\mathbf{i} - \frac{3}{5}\mathbf{j}$   
 $\theta = \cos^{-1}(\mathbf{u}_{AB} \cdot \mathbf{u}_F) = 68.9^\circ$  *Ans.*

**F2-27.**  $\mathbf{u}_{OA} = \frac{12}{13}\mathbf{i} + \frac{5}{13}\mathbf{j}$   
 $\mathbf{u}_{OA} \cdot \mathbf{j} = u_{OA}(1) \cos \theta$   
 $\cos \theta = \frac{5}{13}; \quad \theta = 67.4^\circ$  *Ans.*

**F2-28.**  $\mathbf{u}_{OA} = \frac{12}{13}\mathbf{i} + \frac{5}{13}\mathbf{j}$   
 $\mathbf{F} = F\mathbf{u}_F = [650\mathbf{j}] \text{ N}$   
 $F_{OA} = \mathbf{F} \cdot \mathbf{u}_{OA} = 250 \text{ N}$   
 $\mathbf{F}_{OA} = F_{OA} \mathbf{u}_{OA} = \{231\mathbf{i} + 96.2\mathbf{j}\} \text{ N}$  *Ans.*



$$\begin{aligned}
 \mathbf{F} &= (400 \text{ N}) \frac{\{4 \mathbf{i} + 1 \mathbf{j} - 6 \mathbf{k}\} \text{ m}}{\sqrt{(4 \text{ m})^2 + (1 \text{ m})^2 + (-6 \text{ m})^2}} \\
 &= \{219.78 \mathbf{i} + 54.94 \mathbf{j} - 329.67 \mathbf{k}\} \text{ N} \\
 \mathbf{u}_{AO} &= \frac{\{-4 \mathbf{j} - 6 \mathbf{k}\} \text{ m}}{\sqrt{(-4 \text{ m})^2 + (-6 \text{ m})^2}} \\
 &= -0.5547 \mathbf{j} - 0.8321 \mathbf{k} \\
 (F_{AO})_{\text{proj}} &= \mathbf{F} \cdot \mathbf{u}_{AO} = 244 \text{ N} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F} &= [(-600 \text{ lb}) \cos 60^\circ] \sin 30^\circ \mathbf{i} \\
 &\quad + [(600 \text{ lb}) \cos 60^\circ] \cos 30^\circ \mathbf{j} \\
 &\quad + [(600 \text{ lb}) \sin 60^\circ] \mathbf{k} \\
 &= \{-150 \mathbf{i} + 259.81 \mathbf{j} + 519.62 \mathbf{k}\} \text{ lb} \\
 \mathbf{u}_A &= -\frac{2}{3} \mathbf{i} + \frac{2}{3} \mathbf{j} + \frac{1}{3} \mathbf{k} \\
 (F_A)_{\text{par}} &= \mathbf{F} \cdot \mathbf{u}_A = 446.41 \text{ lb} = 446 \text{ lb} \quad \text{Ans.} \\
 (F_A)_{\text{per}} &= \sqrt{(600 \text{ lb})^2 - (446.41 \text{ lb})^2} \\
 &= 401 \text{ lb} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F} &= 56 \text{ N} \left( \frac{3}{5} \mathbf{i} - \frac{6}{5} \mathbf{j} + \frac{2}{5} \mathbf{k} \right) \\
 &= \{24 \mathbf{i} - 48 \mathbf{j} + 16 \mathbf{k}\} \text{ N} \\
 (F_{AO})_{\parallel} &= \mathbf{F} \cdot \mathbf{u}_{AO} = (24 \mathbf{i} - 48 \mathbf{j} + 16 \mathbf{k}) \cdot \left( \frac{3}{5} \mathbf{i} - \frac{6}{5} \mathbf{j} - \frac{2}{5} \mathbf{k} \right) \\
 &= 46.86 \text{ N} = 46.9 \text{ N} \quad \text{Ans.} \\
 (F_{AO})_{\perp} &= \sqrt{F^2 - (F_{AO})_{\parallel}^2} = \sqrt{(56)^2 - (46.86)^2} \\
 &= 30.7 \text{ N} \quad \text{Ans.}
 \end{aligned}$$

## Chapter 3

$$\begin{aligned}
 \mathbf{F3-1.} \quad & \rightarrow \Sigma F_x = 0; \quad \frac{4}{5} F_{AC} - F_{AB} \cos 30^\circ = 0 \\
 & + \uparrow \Sigma F_y = 0; \quad \frac{3}{5} F_{AC} + F_{AB} \sin 30^\circ - 550 = 0 \\
 & F_{AB} = 478 \text{ lb} \quad \text{Ans.} \\
 & F_{AC} = 518 \text{ lb} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F3-2.} \quad & + \uparrow \Sigma F_y = 0; \quad -2(1500) \sin \theta + 700 = 0 \\
 & \theta = 13.5^\circ \\
 & L_{ABC} = 2 \left( \frac{5 \text{ ft}}{\cos 13.5^\circ} \right) = 10.3 \text{ ft} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F3-3.} \quad & \rightarrow \Sigma F_x = 0; \quad T \cos \theta - T \cos \phi = 0 \\
 & \phi = \theta \\
 & + \uparrow \Sigma F_y = 0; \quad 2T \sin \theta - 49.05 \text{ N} = 0 \\
 & \theta = \tan^{-1} \left( \frac{0.15 \text{ m}}{0.2 \text{ m}} \right) = 36.87^\circ \\
 & T = 40.9 \text{ N} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F3-4.} \quad & + \nearrow \Sigma F_x = 0; \quad \frac{4}{5} (F_{sp}) - 5(9.81) \sin 45^\circ = 0 \\
 & F_{sp} = 43.35 \text{ N} \\
 & F_{sp} = k(l - l_0); \quad 43.35 = 200(0.5 - l_0) \\
 & l_0 = 0.283 \text{ m} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F3-5.} \quad & + \uparrow \Sigma F_y = 0; \quad (392.4 \text{ N}) \sin 30^\circ - m_A(9.81) = 0 \\
 & m_A = 20 \text{ kg} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F3-6.} \quad & + \uparrow \Sigma F_y = 0; \quad T_{AB} \sin 15^\circ - 10(9.81) \text{ N} = 0 \\
 & T_{AB} = 379.03 \text{ N} = 379 \text{ N} \quad \text{Ans.} \\
 & \rightarrow \Sigma F_x = 0; \quad T_{BC} - 379.03 \text{ N} \cos 15^\circ = 0 \\
 & T_{BC} = 366.11 \text{ N} = 366 \text{ N} \quad \text{Ans.} \\
 & \rightarrow \Sigma F_x = 0; \quad T_{CD} \cos \theta - 366.11 \text{ N} = 0 \\
 & + \uparrow \Sigma F_y = 0; \quad T_{CD} \sin \theta - 15(9.81) \text{ N} = 0 \\
 & T_{CD} = 395 \text{ N} \quad \text{Ans.} \\
 & \theta = 21.9^\circ \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F3-7.} \quad & \Sigma F_x = 0; \quad \left[ \left( \frac{3}{5} \right) F_3 \right] \left( \frac{3}{5} \right) + 600 \text{ N} - F_2 = 0 \quad (1) \\
 & \Sigma F_y = 0; \quad \left( \frac{4}{5} \right) F_1 - \left[ \left( \frac{3}{5} \right) F_3 \right] \left( \frac{4}{5} \right) = 0 \quad (2) \\
 & \Sigma F_z = 0; \quad \left( \frac{4}{5} \right) F_3 + \left( \frac{3}{5} \right) F_1 - 900 \text{ N} = 0 \quad (3) \\
 & F_3 = 776 \text{ N} \quad \text{Ans.} \\
 & F_1 = 466 \text{ N} \quad \text{Ans.} \\
 & F_2 = 879 \text{ N} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F3-8.} \quad & \Sigma F_z = 0; \quad F_{AD} \left( \frac{4}{5} \right) - 900 = 0 \\
 & F_{AD} = 1125 \text{ N} = 1.125 \text{ kN} \quad \text{Ans.} \\
 & \Sigma F_y = 0; \quad F_{AC} \left( \frac{4}{5} \right) - 1125 \left( \frac{3}{5} \right) = 0 \\
 & F_{AC} = 843.75 \text{ N} = 844 \text{ N} \quad \text{Ans.} \\
 & \Sigma F_x = 0; \quad F_{AB} - 843.75 \left( \frac{3}{5} \right) = 0 \\
 & F_{AB} = 506.25 \text{ N} = 506 \text{ N} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F3-9.} \quad & \mathbf{F}_{AD} = F_{AD} \left( \frac{\mathbf{r}_{AD}}{r_{AD}} \right) = \frac{1}{3} F_{AD} \mathbf{i} - \frac{2}{3} F_{AD} \mathbf{j} + \frac{2}{3} F_{AD} \mathbf{k} \\
 & \Sigma F_z = 0; \quad \frac{2}{3} F_{AD} - 600 = 0 \\
 & F_{AD} = 900 \text{ N} \quad \text{Ans.} \\
 & \Sigma F_y = 0; \quad F_{AB} \cos 30^\circ - \frac{2}{3} (900) = 0 \\
 & F_{AB} = 692.82 \text{ N} = 693 \text{ N} \quad \text{Ans.} \\
 & \Sigma F_x = 0; \quad \frac{1}{3} (900) + 692.82 \sin 30^\circ - F_{AC} = 0 \\
 & F_{AC} = 646.41 \text{ N} = 646 \text{ N} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F3-10.} \quad & \mathbf{F}_{AC} = F_{AC} \{ -\cos 60^\circ \sin 30^\circ \mathbf{i} \\
 & \quad + \cos 60^\circ \cos 30^\circ \mathbf{j} + \sin 60^\circ \mathbf{k} \} \\
 & = -0.25 F_{AC} \mathbf{i} + 0.4330 F_{AC} \mathbf{j} + 0.8660 F_{AC} \mathbf{k} \\
 & \mathbf{F}_{AD} = F_{AD} \{ \cos 120^\circ \mathbf{i} + \cos 120^\circ \mathbf{j} + \cos 45^\circ \mathbf{k} \} \\
 & = -0.5 F_{AD} \mathbf{i} - 0.5 F_{AD} \mathbf{j} + 0.7071 F_{AD} \mathbf{k} \\
 & \Sigma F_y = 0; \quad 0.4330 F_{AC} - 0.5 F_{AD} = 0 \\
 & \Sigma F_z = 0; \quad 0.8660 F_{AC} + 0.7071 F_{AD} - 300 = 0 \\
 & F_{AD} = 175.74 \text{ lb} = 176 \text{ lb} \quad \text{Ans.} \\
 & F_{AC} = 202.92 \text{ lb} = 203 \text{ lb} \quad \text{Ans.} \\
 & \Sigma F_x = 0; \quad F_{AB} - 0.25(202.92) - 0.5(175.74) = 0 \\
 & F_{AB} = 138.60 \text{ lb} = 139 \text{ lb} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F}_B &= F_B \left( \frac{\mathbf{r}_{AB}}{r_{AB}} \right) \\
 &= F_B \left[ \frac{\{-6\mathbf{i} + 3\mathbf{j} + 2\mathbf{k}\} \text{ ft}}{\sqrt{(-6 \text{ ft})^2 + (3 \text{ ft})^2 + (2 \text{ ft})^2}} \right] \\
 &= -\frac{6}{7}F_B\mathbf{i} + \frac{3}{7}F_B\mathbf{j} + \frac{2}{7}F_B\mathbf{k} \\
 \mathbf{F}_C &= F_C \left( \frac{\mathbf{r}_{AC}}{r_{AC}} \right) \\
 &= F_C \left[ \frac{\{-6\mathbf{i} - 2\mathbf{j} + 3\mathbf{k}\} \text{ ft}}{\sqrt{(-6 \text{ ft})^2 + (-2 \text{ ft})^2 + (3 \text{ ft})^2}} \right] \\
 &= -\frac{6}{7}F_C\mathbf{i} - \frac{2}{7}F_C\mathbf{j} + \frac{3}{7}F_C\mathbf{k} \\
 \mathbf{F}_D &= F_D\mathbf{i} \\
 \mathbf{W} &= \{-150\mathbf{k}\} \text{ lb} \\
 \Sigma F_x &= 0; \quad -\frac{6}{7}F_B - \frac{6}{7}F_C + F_D = 0 \quad (1) \\
 \Sigma F_y &= 0; \quad \frac{3}{7}F_B - \frac{2}{7}F_C = 0 \quad (2) \\
 \Sigma F_z &= 0; \quad \frac{2}{7}F_B + \frac{3}{7}F_C - 150 = 0 \quad (3) \\
 F_B &= 162 \text{ lb} \quad \text{Ans.} \\
 F_C &= 1.5(162 \text{ lb}) = 242 \text{ lb} \quad \text{Ans.} \\
 F_D &= 346.15 \text{ lb} = 346 \text{ lb} \quad \text{Ans.}
 \end{aligned}$$

## Chapter 4

$$\begin{aligned}
 \mathbf{F4-1.} \quad \zeta + M_O &= -\left(\frac{4}{5}\right)(100 \text{ N})(2 \text{ m}) - \left(\frac{3}{5}\right)(100 \text{ N})(5 \text{ m}) \\
 &= -460 \text{ N} \cdot \text{m} = 460 \text{ N} \cdot \text{m} \curvearrowright \quad \text{Ans.} \\
 \mathbf{F4-2.} \quad \zeta + M_O &= [(300 \text{ N}) \sin 30^\circ][0.4 \text{ m} + (0.3 \text{ m}) \cos 45^\circ] \\
 &\quad - [(300 \text{ N}) \cos 30^\circ][(0.3 \text{ m}) \sin 45^\circ] \\
 &= 36.7 \text{ N} \cdot \text{m} \quad \text{Ans.} \\
 \mathbf{F4-3.} \quad \zeta + M_O &= (600 \text{ lb})(4 \text{ ft} + (3 \text{ ft}) \cos 45^\circ) - 1 \text{ ft} \\
 &= 3.07 \text{ kip} \cdot \text{ft} \quad \text{Ans.} \\
 \mathbf{F4-4.} \quad \zeta + M_O &= 50 \sin 60^\circ (0.1 + 0.2 \cos 45^\circ + 0.1) \\
 &\quad - 50 \cos 60^\circ (0.2 \sin 45^\circ) \\
 &= 11.2 \text{ N} \cdot \text{m} \quad \text{Ans.} \\
 \mathbf{F4-5.} \quad \zeta + M_O &= 600 \sin 50^\circ (5) + 600 \cos 50^\circ (0.5) \\
 &= 2.49 \text{ kip} \cdot \text{ft} \quad \text{Ans.} \\
 \mathbf{F4-6.} \quad \zeta + M_O &= 500 \sin 45^\circ (3 + 3 \cos 45^\circ) \\
 &\quad - 500 \cos 45^\circ (3 \sin 45^\circ) \\
 &= 1.06 \text{ kN} \cdot \text{m} \quad \text{Ans.} \\
 \mathbf{F4-7.} \quad \zeta + (M_R)_O &= \Sigma Fd; \\
 (M_R)_O &= -(600 \text{ N})(1 \text{ m}) \\
 &\quad + (500 \text{ N})[3 \text{ m} + (2.5 \text{ m}) \cos 45^\circ] \\
 &\quad - (300 \text{ N})[(2.5 \text{ m}) \sin 45^\circ] \\
 &= 1254 \text{ N} \cdot \text{m} = 1.25 \text{ kN} \cdot \text{m} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F4-8.} \quad \zeta + (M_R)_O &= \Sigma Fd; \\
 (M_R)_O &= \left[\left(\frac{3}{5}\right)500 \text{ N}\right](0.425 \text{ m}) \\
 &\quad - \left[\left(\frac{4}{5}\right)500 \text{ N}\right](0.25 \text{ m}) \\
 &\quad - [(600 \text{ N}) \cos 60^\circ](0.25 \text{ m}) \\
 &\quad - [(600 \text{ N}) \sin 60^\circ](0.425 \text{ m}) \\
 &= -268 \text{ N} \cdot \text{m} = 268 \text{ N} \cdot \text{m} \curvearrowright \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F4-9.} \quad \zeta + (M_R)_O &= \Sigma Fd; \\
 (M_R)_O &= (300 \cos 30^\circ \text{ lb})(6 \text{ ft} + 6 \sin 30^\circ \text{ ft}) \\
 &\quad - (300 \sin 30^\circ \text{ lb})(6 \cos 30^\circ \text{ ft}) \\
 &\quad + (200 \text{ lb})(6 \cos 30^\circ \text{ ft}) \\
 &= 2.60 \text{ kip} \cdot \text{ft} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F4-10.} \quad \mathbf{F} &= F\mathbf{u}_{AB} = 500 \text{ N} \left( \frac{4}{5}\mathbf{i} - \frac{3}{5}\mathbf{j} \right) = \{400\mathbf{i} - 300\mathbf{j}\} \text{ N} \\
 \mathbf{M}_O &= \mathbf{r}_{OA} \times \mathbf{F} = \{3\mathbf{j}\} \text{ m} \times \{400\mathbf{i} - 300\mathbf{j}\} \text{ N} \\
 &= \{-1200\mathbf{k}\} \text{ N} \cdot \text{m} \quad \text{Ans.} \\
 \text{or} \\
 \mathbf{M}_O &= \mathbf{r}_{OB} \times \mathbf{F} = \{4\mathbf{i}\} \text{ m} \times \{400\mathbf{i} - 300\mathbf{j}\} \text{ N} \\
 &= \{-1200\mathbf{k}\} \text{ N} \cdot \text{m} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F4-11.} \quad \mathbf{F} &= F\mathbf{u}_{BC} \\
 &= 120 \text{ lb} \left[ \frac{\{4\mathbf{i} - 4\mathbf{j} - 2\mathbf{k}\} \text{ ft}}{\sqrt{(4 \text{ ft})^2 + (-4 \text{ ft})^2 + (-2 \text{ ft})^2}} \right] \\
 &= \{80\mathbf{i} - 80\mathbf{j} - 40\mathbf{k}\} \text{ lb} \\
 \mathbf{M}_O &= \mathbf{r}_C \times \mathbf{F} = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 5 & 0 & 0 \\ 80 & -80 & -40 \end{vmatrix} \\
 &= \{200\mathbf{j} - 400\mathbf{k}\} \text{ lb} \cdot \text{ft} \quad \text{Ans.} \\
 \text{or} \\
 \mathbf{M}_O &= \mathbf{r}_B \times \mathbf{F} = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 1 & 4 & 2 \\ 80 & -80 & -40 \end{vmatrix} \\
 &= \{200\mathbf{j} - 400\mathbf{k}\} \text{ lb} \cdot \text{ft} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F4-12.} \quad \mathbf{F}_R &= \mathbf{F}_1 + \mathbf{F}_2 \\
 &= \{(100 - 200)\mathbf{i} + (-120 + 250)\mathbf{j} \\
 &\quad + (75 + 100)\mathbf{k}\} \text{ lb} \\
 &= \{-100\mathbf{i} + 130\mathbf{j} + 175\mathbf{k}\} \text{ lb} \\
 (\mathbf{M}_R)_O &= \mathbf{r}_A \times \mathbf{F}_R = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 4 & 5 & 3 \\ -100 & 130 & 175 \end{vmatrix} \\
 &= \{485\mathbf{i} - 1000\mathbf{j} + 1020\mathbf{k}\} \text{ lb} \cdot \text{ft} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned} \mathbf{F4-13.} \quad M_x &= \mathbf{i} \cdot (\mathbf{r}_{OB} \times \mathbf{F}) = \begin{vmatrix} 1 & 0 & 0 \\ 0.3 & 0.4 & -0.2 \\ 300 & -200 & 150 \end{vmatrix} \\ &= 20 \text{ N} \cdot \text{m} \quad \text{Ans.} \end{aligned}$$

$$\begin{aligned} \mathbf{F4-14.} \quad \mathbf{u}_{OA} &= \frac{\mathbf{r}_A}{r_A} = \frac{\{0.3\mathbf{i} + 0.4\mathbf{j}\} \text{ m}}{\sqrt{(0.3 \text{ m})^2 + (0.4 \text{ m})^2}} = 0.6\mathbf{i} + 0.8\mathbf{j} \\ M_{OA} &= \mathbf{u}_{OA} \cdot (\mathbf{r}_{AB} \times \mathbf{F}) = \begin{vmatrix} 0.6 & 0.8 & 0 \\ 0 & 0 & -0.2 \\ 300 & -200 & 150 \end{vmatrix} \\ &= -72 \text{ N} \cdot \text{m} \quad \text{Ans.} \end{aligned}$$

$$|M_{OA}| = 72 \text{ N} \cdot \text{m}$$

$$\begin{aligned} \mathbf{F4-15.} \quad &\text{Scalar Analysis} \\ &\text{The magnitudes of the force components are} \\ F_x &= |200 \cos 120^\circ| = 100 \text{ N} \\ F_y &= 200 \cos 60^\circ = 100 \text{ N} \\ F_z &= 200 \cos 45^\circ = 141.42 \text{ N} \\ M_x &= -F_y(z) + F_z(y) \\ &= -(100 \text{ N})(0.25 \text{ m}) + (141.42 \text{ N})(0.3 \text{ m}) \\ &= 17.4 \text{ N} \cdot \text{m} \quad \text{Ans.} \end{aligned}$$

Vector Analysis

$$M_x = \begin{vmatrix} 1 & 0 & 0 \\ 0 & 0.3 & 0.25 \\ -100 & 100 & 141.42 \end{vmatrix} = 17.4 \text{ N} \cdot \text{m} \quad \text{Ans.}$$

$$\begin{aligned} \mathbf{F4-16.} \quad M_y &= \mathbf{j} \cdot (\mathbf{r}_A \times \mathbf{F}) = \begin{vmatrix} 0 & 1 & 0 \\ -3 & -4 & 2 \\ 30 & -20 & 50 \end{vmatrix} \\ &= 210 \text{ N} \cdot \text{m} \quad \text{Ans.} \end{aligned}$$

$$\begin{aligned} \mathbf{F4-17.} \quad \mathbf{u}_{AB} &= \frac{\mathbf{r}_{AB}}{r_{AB}} = \frac{\{-4\mathbf{i} + 3\mathbf{j}\} \text{ ft}}{\sqrt{(-4 \text{ ft})^2 + (3 \text{ ft})^2}} = -0.8\mathbf{i} + 0.6\mathbf{j} \\ M_{AB} &= \mathbf{u}_{AB} \cdot (\mathbf{r}_{AC} \times \mathbf{F}) \\ &= \begin{vmatrix} -0.8 & 0.6 & 0 \\ 0 & 0 & 2 \\ 50 & -40 & 20 \end{vmatrix} = -4 \text{ lb} \cdot \text{ft} \\ M_{AB} &= M_{AB}\mathbf{u}_{AB} = \{3.2\mathbf{i} - 2.4\mathbf{j}\} \text{ lb} \cdot \text{ft} \quad \text{Ans.} \end{aligned}$$

$$\begin{aligned} \mathbf{F4-18.} \quad &\text{Scalar Analysis} \\ &\text{The magnitudes of the force components are} \\ F_x &= \left(\frac{3}{5}\right)\left[\frac{4}{5}(500)\right] = 240 \text{ N} \\ F_y &= \frac{4}{5}\left[\frac{4}{5}(500)\right] = 320 \text{ N} \end{aligned}$$

$$\begin{aligned} F_z &= \frac{3}{5}(500) = 300 \text{ N} \\ M_x &= -320(3) + 300(2) = -360 \text{ N} \cdot \text{m} \quad \text{Ans.} \\ M_y &= -240(3) - 300(-2) = -120 \text{ N} \cdot \text{m} \quad \text{Ans.} \\ M_z &= 240(2) - 320(2) = -160 \text{ N} \cdot \text{m} \quad \text{Ans.} \end{aligned}$$

Vector Analysis

$$\mathbf{F} = \{-240\mathbf{i} + 320\mathbf{j} + 300\mathbf{k}\} \text{ N}$$

$$\mathbf{r}_{OA} = \{-2\mathbf{i} + 2\mathbf{j} + 3\mathbf{k}\} \text{ m}$$

$$M_x = \mathbf{i} \cdot (\mathbf{r}_{OA} \times \mathbf{F}) = -360 \text{ N} \cdot \text{m}$$

$$M_y = \mathbf{j} \cdot (\mathbf{r}_{OA} \times \mathbf{F}) = -120 \text{ N} \cdot \text{m}$$

$$M_z = \mathbf{k} \cdot (\mathbf{r}_{OA} \times \mathbf{F}) = -160 \text{ N} \cdot \text{m}$$

$$\begin{aligned} \mathbf{F4-19.} \quad \zeta + M_{C_R} &= \Sigma M_A = 400(3) - 400(5) + 300(5) \\ &\quad + 200(0.2) = 740 \text{ N} \cdot \text{m} \quad \text{Ans.} \end{aligned}$$

Also,

$$\begin{aligned} \zeta + M_{C_R} &= 300(5) - 400(2) + 200(0.2) \\ &= 740 \text{ N} \cdot \text{m} \quad \text{Ans.} \end{aligned}$$

$$\begin{aligned} \mathbf{F4-20.} \quad \zeta + M_{C_R} &= 300(4) + 200(4) + 150(4) \\ &= 2600 \text{ lb} \cdot \text{ft} \quad \text{Ans.} \end{aligned}$$

$$\begin{aligned} \mathbf{F4-21.} \quad \zeta + (M_B)_R &= \Sigma M_B \\ -1.5 \text{ kN} \cdot \text{m} &= (2 \text{ kN})(0.3 \text{ m}) - F(0.9 \text{ m}) \end{aligned}$$

$$F = 2.33 \text{ kN} \quad \text{Ans.}$$

$$\begin{aligned} \mathbf{F4-22.} \quad \zeta + M_C &= 10\left(\frac{3}{5}\right)(2) - 10\left(\frac{4}{5}\right)(4) = -20 \text{ kN} \cdot \text{m} \\ &= 20 \text{ kN} \cdot \text{m} \quad \text{Ans.} \end{aligned}$$

$$\begin{aligned} \mathbf{F4-23.} \quad \mathbf{u}_1 &= \frac{\mathbf{r}_1}{r_1} = \frac{\{-2\mathbf{i} + 2\mathbf{j} + 3.5\mathbf{k}\} \text{ ft}}{\sqrt{(-2 \text{ ft})^2 + (2 \text{ ft})^2 + (3.5 \text{ ft})^2}} \\ &= -\frac{2}{4.5}\mathbf{i} + \frac{2}{4.5}\mathbf{j} + \frac{3.5}{4.5}\mathbf{k} \end{aligned}$$

$$\mathbf{u}_2 = -\mathbf{k}$$

$$\mathbf{u}_3 = \frac{1.5}{2.5}\mathbf{i} - \frac{2}{2.5}\mathbf{j}$$

$$\begin{aligned} (\mathbf{M}_c)_1 &= (M_c)_1\mathbf{u}_1 \\ &= (450 \text{ lb} \cdot \text{ft})\left(-\frac{2}{4.5}\mathbf{i} + \frac{2}{4.5}\mathbf{j} + \frac{3.5}{4.5}\mathbf{k}\right) \\ &= \{-200\mathbf{i} + 200\mathbf{j} + 350\mathbf{k}\} \text{ lb} \cdot \text{ft} \end{aligned}$$

$$\begin{aligned} (\mathbf{M}_c)_2 &= (M_c)_2\mathbf{u}_2 = (250 \text{ lb} \cdot \text{ft})(-\mathbf{k}) \\ &= \{-250\mathbf{k}\} \text{ lb} \cdot \text{ft} \end{aligned}$$

$$\begin{aligned} (\mathbf{M}_c)_3 &= (M_c)_3\mathbf{u}_3 = (300 \text{ lb} \cdot \text{ft})\left(\frac{1.5}{2.5}\mathbf{i} - \frac{2}{2.5}\mathbf{j}\right) \\ &= \{180\mathbf{i} - 240\mathbf{j}\} \text{ lb} \cdot \text{ft} \end{aligned}$$

$$(\mathbf{M}_c)_R = \Sigma \mathbf{M}_c;$$

$$(\mathbf{M}_c)_R = \{-20\mathbf{i} - 40\mathbf{j} + 100\mathbf{k}\} \text{ lb} \cdot \text{ft} \quad \text{Ans.}$$

**F4-24.**  $\mathbf{F}_B = \left(\frac{4}{5}\right)(450 \text{ N})\mathbf{j} - \left(\frac{3}{5}\right)(450 \text{ N})\mathbf{k}$   
 $= \{360\mathbf{j} - 270\mathbf{k}\} \text{ N}$

$$\mathbf{M}_c = \mathbf{r}_{AB} \times \mathbf{F}_B = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 0.4 & 0 & 0 \\ 0 & 360 & -270 \end{vmatrix}$$

$$= \{108\mathbf{j} + 144\mathbf{k}\} \text{ N} \cdot \text{m}$$

Also,

$$\mathbf{M}_c = (\mathbf{r}_A \times \mathbf{F}_A) + (\mathbf{r}_B \times \mathbf{F}_B)$$

$$= \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 0 & 0 & 0.3 \\ 0 & -360 & 270 \end{vmatrix} + \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 0.4 & 0 & 0.3 \\ 0 & 360 & -270 \end{vmatrix}$$

$$= \{108\mathbf{j} + 144\mathbf{k}\} \text{ N} \cdot \text{m}$$

*Ans.*

**F4-25.**  $+\leftarrow F_{Rx} = \Sigma F_x; F_{Rx} = 200 - \frac{3}{5}(100) = 140 \text{ lb}$   
 $+\downarrow F_{Ry} = \Sigma F_y; F_{Ry} = 150 - \frac{4}{5}(100) = 70 \text{ lb}$

$$F_R = \sqrt{140^2 + 70^2} = 157 \text{ lb}$$

*Ans.*

$$\theta = \tan^{-1}\left(\frac{70}{140}\right) = 26.6^\circ \nearrow$$

*Ans.*

$$\zeta + M_{A_R} = \Sigma M_A;$$

$$M_{A_R} = \frac{3}{5}(100)(4) - \frac{4}{5}(100)(6) + 150(3)$$

$$M_{A_R} = 210 \text{ lb} \cdot \text{ft}$$

*Ans.*

**F4-26.**  $+\rightarrow F_{Rx} = \Sigma F_x; F_{Rx} = \frac{4}{5}(50) = 40 \text{ N}$   
 $+\downarrow F_{Ry} = \Sigma F_y; F_{Ry} = 40 + 30 + \frac{3}{5}(50)$   
 $= 100 \text{ N}$

$$F_R = \sqrt{(40)^2 + (100)^2} = 108 \text{ N}$$

*Ans.*

$$\theta = \tan^{-1}\left(\frac{100}{40}\right) = 68.2^\circ \searrow$$

*Ans.*

$$\zeta + M_{A_R} = \Sigma M_A;$$

$$M_{A_R} = 30(3) + \frac{3}{5}(50)(6) + 200$$

$$= 470 \text{ N} \cdot \text{m}$$

*Ans.*

**F4-27.**  $+\rightarrow (F_R)_x = \Sigma F_x;$   
 $(F_R)_x = 900 \sin 30^\circ = 450 \text{ N} \rightarrow$

$$+\uparrow (F_R)_y = \Sigma F_y;$$

$$(F_R)_y = -900 \cos 30^\circ - 300$$

$$= -1079.42 \text{ N} = 1079.42 \text{ N} \downarrow$$

$$F_R = \sqrt{450^2 + 1079.42^2}$$

$$= 1169.47 \text{ N} = 1.17 \text{ kN}$$

*Ans.*

$$\theta = \tan^{-1}\left(\frac{1079.42}{450}\right) = 67.4^\circ \searrow$$

*Ans.*

$$\zeta + (M_R)_A = \Sigma M_A;$$

$$(M_R)_A = 300 - 900 \cos 30^\circ (0.75) - 300(2.25)$$

$$= -959.57 \text{ N} \cdot \text{m}$$

$$= 960 \text{ N} \cdot \text{m} \curvearrowright$$

*Ans.*

**F4-28.**  $+\rightarrow (F_R)_x = \Sigma F_x;$

$$(F_R)_x = 150\left(\frac{3}{5}\right) + 50 - 100\left(\frac{4}{5}\right) = 60 \text{ lb} \rightarrow$$

$$+\uparrow (F_R)_y = \Sigma F_y;$$

$$(F_R)_y = -150\left(\frac{4}{5}\right) - 100\left(\frac{3}{5}\right)$$

$$= -180 \text{ lb} = 180 \text{ lb} \downarrow$$

$$F_R = \sqrt{60^2 + 180^2} = 189.74 \text{ lb} = 190 \text{ lb}$$

*Ans.*

$$\theta = \tan^{-1}\left(\frac{180}{60}\right) = 71.6^\circ \swarrow$$

*Ans.*

$$\zeta + (M_R)_A = \Sigma M_A;$$

$$(M_R)_A = 100\left(\frac{4}{5}\right)(1) - 100\left(\frac{3}{5}\right)(6) - 150\left(\frac{4}{5}\right)(3)$$

$$= -640 = 640 \text{ lb} \cdot \text{ft} \curvearrowright$$

*Ans.*

**F4-29.**  $\mathbf{F}_R = \Sigma \mathbf{F};$

$$\mathbf{F}_R = \mathbf{F}_1 + \mathbf{F}_2$$

$$= (-300\mathbf{i} + 150\mathbf{j} + 200\mathbf{k}) + (-450\mathbf{k})$$

$$= \{-300\mathbf{i} + 150\mathbf{j} - 250\mathbf{k}\} \text{ N}$$

*Ans.*

$$\mathbf{r}_{OA} = (2 - 0)\mathbf{j} = \{2\mathbf{j}\} \text{ m}$$

$$\mathbf{r}_{OB} = (-1.5 - 0)\mathbf{i} + (2 - 0)\mathbf{j} + (1 - 0)\mathbf{k}$$

$$= \{-1.5\mathbf{i} + 2\mathbf{j} + 1\mathbf{k}\} \text{ m}$$

$$(\mathbf{M}_R)_O = \Sigma \mathbf{M};$$

$$(\mathbf{M}_R)_O = \mathbf{r}_{OB} \times \mathbf{F}_1 + \mathbf{r}_{OA} \times \mathbf{F}_2$$

$$= \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ -1.5 & 2 & 1 \\ -300 & 150 & 200 \end{vmatrix} + \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 0 & 2 & 0 \\ 0 & 0 & -450 \end{vmatrix}$$

$$= \{-650\mathbf{i} + 375\mathbf{k}\} \text{ N} \cdot \text{m}$$

*Ans.*

**F4-30.**  $\mathbf{F}_1 = \{-100\mathbf{j}\} \text{ N}$

$$\mathbf{F}_2 = (200 \text{ N}) \left[ \frac{\{-0.4\mathbf{i} - 0.3\mathbf{k}\} \text{ m}}{\sqrt{(-0.4 \text{ m})^2 + (-0.3 \text{ m})^2}} \right]$$

$$= \{-160\mathbf{i} - 120\mathbf{k}\} \text{ N}$$

$$\mathbf{M}_c = \{-75\mathbf{i}\} \text{ N} \cdot \text{m}$$

$$\mathbf{F}_R = \{-160\mathbf{i} - 100\mathbf{j} - 120\mathbf{k}\} \text{ N}$$

*Ans.*

$$(\mathbf{M}_R)_O = (0.3\mathbf{k}) \times (-100\mathbf{j})$$

$$+ \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 0 & 0.5 & 0.3 \\ -160 & 0 & -120 \end{vmatrix} + (-75\mathbf{i})$$

$$= \{-105\mathbf{i} - 48\mathbf{j} + 80\mathbf{k}\} \text{ N} \cdot \text{m}$$

*Ans.*

**F4-31.**  $+\downarrow F_R = \Sigma F_y; F_R = 500 + 250 + 500$

$$= 1250 \text{ lb}$$

*Ans.*

$$\zeta + F_{Rx} = \Sigma M_O;$$

$$1250(x) = 500(3) + 250(6) + 500(9)$$

$$x = 6 \text{ ft}$$

*Ans.*

**F4-32.**  $\rightarrow (F_R)_x = \Sigma F_x;$   
 $(F_R)_x = 100\left(\frac{3}{5}\right) + 50 \sin 30^\circ = 85 \text{ lb} \rightarrow$   
 $+\uparrow (F_R)_y = \Sigma F_y;$   
 $(F_R)_y = 200 + 50 \cos 30^\circ - 100\left(\frac{4}{5}\right)$   
 $= 163.30 \text{ lb} \uparrow$   
 $F_R = \sqrt{85^2 + 163.30^2} = 184 \text{ lb}$   
 $\theta = \tan^{-1}\left(\frac{163.30}{85}\right) = 62.5^\circ \swarrow$  *Ans.*  
 $\zeta + (M_R)_A = \Sigma M_A;$   
 $163.30(d) = 200(3) - 100\left(\frac{4}{5}\right)(6) + 50 \cos 30^\circ(9)$   
 $d = 3.12 \text{ ft}$  *Ans.*

**F4-33.**  $\rightarrow (F_R)_x = \Sigma F_x;$   
 $(F_R)_x = 15\left(\frac{4}{5}\right) = 12 \text{ kN} \rightarrow$   
 $+\uparrow (F_R)_y = \Sigma F_y;$   
 $(F_R)_y = -20 + 15\left(\frac{3}{5}\right) = -11 \text{ kN} = 11 \text{ kN} \downarrow$   
 $F_R = \sqrt{12^2 + 11^2} = 16.3 \text{ kN}$  *Ans.*  
 $\theta = \tan^{-1}\left(\frac{11}{12}\right) = 42.5^\circ \swarrow$  *Ans.*  
 $\zeta + (M_R)_A = \Sigma M_A;$   
 $-11(d) = -20(2) - 15\left(\frac{4}{5}\right)(2) + 15\left(\frac{3}{5}\right)(6)$   
 $d = 0.909 \text{ m}$  *Ans.*

**F4-34.**  $\rightarrow (F_R)_x = \Sigma F_x;$   
 $(F_R)_x = \left(\frac{3}{5}\right) 5 \text{ kN} - 8 \text{ kN}$   
 $= -5 \text{ kN} = 5 \text{ kN} \leftarrow$   
 $+\uparrow (F_R)_y = \Sigma F_y;$   
 $(F_R)_y = -6 \text{ kN} - \left(\frac{4}{5}\right) 5 \text{ kN}$   
 $= -10 \text{ kN} = 10 \text{ kN} \downarrow$   
 $F_R = \sqrt{5^2 + 10^2} = 11.2 \text{ kN}$  *Ans.*  
 $\theta = \tan^{-1}\left(\frac{10 \text{ kN}}{5 \text{ kN}}\right) = 63.4^\circ \searrow$  *Ans.*  
 $\zeta + (M_R)_A = \Sigma M_A;$   
 $5 \text{ kN}(d) = 8 \text{ kN}(3 \text{ m}) - 6 \text{ kN}(0.5 \text{ m})$   
 $- \left[\left(\frac{4}{5}\right) 5 \text{ kN}\right](2 \text{ m})$   
 $- \left[\left(\frac{3}{5}\right) 5 \text{ kN}\right](4 \text{ m})$   
 $d = 0.2 \text{ m}$  *Ans.*

**F4-35.**  $+\downarrow F_R = \Sigma F_z;$   $F_R = 400 + 500 - 100$   
 $= 800 \text{ N}$  *Ans.*  
 $M_{Rx} = \Sigma M_x; -800y = -400(4) - 500(4)$   
 $y = 4.50 \text{ m}$  *Ans.*  
 $M_{Ry} = \Sigma M_y; 800x = 500(4) - 100(3)$   
 $x = 2.125 \text{ m}$  *Ans.*

**F4-36.**  $+\downarrow F_R = \Sigma F_z;$   
 $F_R = 200 + 200 + 100 + 100$   
 $= 600 \text{ N}$  *Ans.*  
 $\zeta + M_{Rx} = \Sigma M_x;$   
 $-600y = 200(1) + 200(1) + 100(3) - 100(3)$   
 $y = -0.667 \text{ m}$  *Ans.*  
 $\zeta + M_{Ry} = \Sigma M_y;$   
 $600x = 100(3) + 100(3) + 200(2) - 200(3)$   
 $x = 0.667 \text{ m}$  *Ans.*

**F4-37.**  $+\uparrow F_R = \Sigma F_y;$   
 $-F_R = -6(1.5) - 9(3) - 3(1.5)$   
 $F_R = 40.5 \text{ kN} \downarrow$  *Ans.*  
 $\zeta + (M_R)_A = \Sigma M_A;$   
 $-40.5(d) = 6(1.5)(0.75)$   
 $-9(3)(1.5) - 3(1.5)(3.75)$   
 $d = 1.25 \text{ m}$  *Ans.*

**F4-38.**  $F_R = \frac{1}{2}(6)(150) + 8(150) = 1650 \text{ lb}$  *Ans.*  
 $\zeta + M_{Ar} = \Sigma M_A;$   
 $1650d = \left[\frac{1}{2}(6)(150)\right](4) + [8(150)](10)$   
 $d = 8.36 \text{ ft}$  *Ans.*

**F4-39.**  $+\uparrow F_R = \Sigma F_y;$   
 $-F_R = -\frac{1}{2}(6)(3) - \frac{1}{2}(6)(6)$   
 $F_R = 27 \text{ kN} \downarrow$  *Ans.*  
 $\zeta + (M_R)_A = \Sigma M_A;$   
 $-27(d) = \frac{1}{2}(6)(3)(1) - \frac{1}{2}(6)(6)(2)$   
 $d = 1 \text{ m}$  *Ans.*

**F4-40.**  $+\downarrow F_R = \Sigma F_y;$   
 $F_R = \frac{1}{2}(50)(6) + 150(6) + 500$   
 $= 1550 \text{ lb}$  *Ans.*  
 $\zeta + M_{Ar} = \Sigma M_A;$   
 $1550d = \left[\frac{1}{2}(50)(6)\right](4) + [150(6)](3) + 500(9)$   
 $d = 5.03 \text{ ft}$  *Ans.*

**F4-41.**  $+\uparrow F_R = \Sigma F_y;$   
 $-F_R = -\frac{1}{2}(3)(4.5) - 3(6)$   
 $F_R = 24.75 \text{ kN} \downarrow$  *Ans.*  
 $\zeta + (M_R)_A = \Sigma M_A;$   
 $-24.75(d) = -\frac{1}{2}(3)(4.5)(1.5) - 3(6)(3)$   
 $d = 2.59 \text{ m}$  *Ans.*

$$\begin{aligned}\mathbf{F4-42.} \quad F_R &= \int w(x) dx = \int_0^4 2.5x^3 dx = 160 \text{ N} \\ \zeta + M_{A_R} &= \Sigma M_A; \\ x &= \frac{\int xw(x) dx}{\int w(x) dx} = \frac{\int_0^4 2.5x^4 dx}{160} = 3.20 \text{ m} \quad \text{Ans.}\end{aligned}$$

## Chapter 5

$$\begin{aligned}\mathbf{F5-1.} \quad +\Sigma F_x &= 0; \quad -A_x + 500\left(\frac{3}{5}\right) = 0 \\ A_x &= 300 \text{ lb} \quad \text{Ans.} \\ \zeta + \Sigma M_A &= 0; \quad B_y(10) - 500\left(\frac{4}{5}\right)(5) - 600 = 0 \\ B_y &= 260 \text{ lb} \quad \text{Ans.} \\ +\uparrow \Sigma F_y &= 0; \quad A_y + 260 - 500\left(\frac{4}{5}\right) = 0 \\ A_y &= 140 \text{ lb} \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}\mathbf{F5-2.} \quad \zeta + \Sigma M_A &= 0; \\ F_{CD} \sin 45^\circ(1.5 \text{ m}) - 4 \text{ kN}(3 \text{ m}) &= 0 \\ F_{CD} &= 11.31 \text{ kN} = 11.3 \text{ kN} \quad \text{Ans.} \\ +\Sigma F_x &= 0; \quad A_x + (11.31 \text{ kN}) \cos 45^\circ = 0 \\ A_x &= -8 \text{ kN} = 8 \text{ kN} \leftarrow \quad \text{Ans.} \\ +\uparrow \Sigma F_y &= 0; \\ A_y + (11.31 \text{ kN}) \sin 45^\circ - 4 \text{ kN} &= 0 \\ A_y &= -4 \text{ kN} = 4 \text{ kN} \downarrow \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}\mathbf{F5-3.} \quad \zeta + \Sigma M_A &= 0; \\ N_B[6 \text{ m} + (6 \text{ m}) \cos 45^\circ] & \\ - 10 \text{ kN}[2 \text{ m} + (6 \text{ m}) \cos 45^\circ] & \\ - 5 \text{ kN}(4 \text{ m}) &= 0 \\ N_B &= 8.047 \text{ kN} = 8.05 \text{ kN} \quad \text{Ans.} \\ +\Sigma F_x &= 0; \\ (5 \text{ kN}) \cos 45^\circ - A_x &= 0 \\ A_x &= 3.54 \text{ kN} \quad \text{Ans.} \\ +\uparrow \Sigma F_y &= 0; \\ A_y + 8.047 \text{ kN} - (5 \text{ kN}) \sin 45^\circ - 10 \text{ kN} &= 0 \\ A_y &= 5.49 \text{ kN} \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}\mathbf{F5-4.} \quad +\Sigma F_x &= 0; \quad -A_x + 400 \cos 30^\circ = 0 \\ A_x &= 346 \text{ N} \quad \text{Ans.} \\ +\uparrow \Sigma F_y &= 0; \\ A_y - 200 - 200 - 200 - 400 \sin 30^\circ &= 0 \\ A_y &= 800 \text{ N} \quad \text{Ans.} \\ \zeta + \Sigma M_A &= 0; \\ M_A - 200(2.5) - 200(3.5) - 200(4.5) & \\ - 400 \sin 30^\circ(4.5) - 400 \cos 30^\circ(3 \sin 60^\circ) &= 0 \\ M_A &= 3.90 \text{ kN} \cdot \text{m} \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}\mathbf{F5-5.} \quad \zeta + \Sigma M_A &= 0; \\ N_C(0.7 \text{ m}) - [25(9.81) \text{ N}](0.5 \text{ m}) \cos 30^\circ &= 0 \\ N_C &= 151.71 \text{ N} = 152 \text{ N} \quad \text{Ans.} \\ +\Sigma F_x &= 0; \\ T_{AB} \cos 15^\circ - (151.71 \text{ N}) \cos 60^\circ &= 0 \\ T_{AB} &= 78.53 \text{ N} = 78.5 \text{ N} \quad \text{Ans.} \\ +\uparrow \Sigma F_y &= 0; \\ F_A + (78.53 \text{ N}) \sin 15^\circ & \\ + (151.71 \text{ N}) \sin 60^\circ - 25(9.81) \text{ N} &= 0 \\ F_A &= 93.5 \text{ N} \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}\mathbf{F5-6.} \quad +\Sigma F_x &= 0; \\ N_C \sin 30^\circ - (250 \text{ N}) \sin 60^\circ &= 0 \\ N_C &= 433.0 \text{ N} = 433 \text{ N} \quad \text{Ans.} \\ \zeta + \Sigma M_B &= 0; \\ -N_A \sin 30^\circ(0.15 \text{ m}) - 433.0 \text{ N}(0.2 \text{ m}) & \\ + [(250 \text{ N}) \cos 30^\circ](0.6 \text{ m}) &= 0 \\ N_A &= 577.4 \text{ N} = 577 \text{ N} \quad \text{Ans.} \\ +\uparrow \Sigma F_y &= 0; \\ N_B - 577.4 \text{ N} + (433.0 \text{ N}) \cos 30^\circ & \\ - (250 \text{ N}) \cos 60^\circ &= 0 \\ N_B &= 327 \text{ N} \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}\mathbf{F5-7.} \quad \Sigma F_z &= 0; \\ T_A + T_B + T_C - 200 - 500 &= 0 \\ \Sigma M_x &= 0; \\ T_A(3) + T_C(3) - 500(1.5) - 200(3) &= 0 \\ \Sigma M_y &= 0; \\ -T_B(4) - T_C(4) + 500(2) + 200(2) &= 0 \\ T_A = 350 \text{ lb}, T_B = 250 \text{ lb}, T_C = 100 \text{ lb} & \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}\mathbf{F5-8.} \quad \Sigma M_y &= 0; \\ 600 \text{ N}(0.2 \text{ m}) + 900 \text{ N}(0.6 \text{ m}) - F_A(1 \text{ m}) &= 0 \\ F_A &= 660 \text{ N} \quad \text{Ans.} \\ \Sigma M_x &= 0; \\ D_z(0.8 \text{ m}) - 600 \text{ N}(0.5 \text{ m}) - 900 \text{ N}(0.1 \text{ m}) &= 0 \\ D_z &= 487.5 \text{ N} \quad \text{Ans.} \\ \Sigma F_x &= 0; \quad D_x = 0 \quad \text{Ans.} \\ \Sigma F_y &= 0; \quad D_y = 0 \quad \text{Ans.} \\ \Sigma F_z &= 0; \\ T_{BC} + 660 \text{ N} + 487.5 \text{ N} - 900 \text{ N} - 600 \text{ N} &= 0 \\ T_{BC} &= 352.5 \text{ N} \quad \text{Ans.}\end{aligned}$$

**F5-9.**  $\Sigma F_y = 0$ ;  $400 \text{ N} + C_y = 0$ ; *Ans.*  
 $C_y = -400 \text{ N}$   
 $\Sigma M_y = 0$ ;  $-C_x(0.4 \text{ m}) - 600 \text{ N}(0.6 \text{ m}) = 0$   
 $C_x = -900 \text{ N}$  *Ans.*  
 $\Sigma M_x = 0$ ;  $B_z(0.6 \text{ m}) + 600 \text{ N}(1.2 \text{ m})$   
 $+ (-400 \text{ N})(0.4 \text{ m}) = 0$   
 $B_z = -933.3 \text{ N}$  *Ans.*  
 $\Sigma M_z = 0$ ;  
 $-B_x(0.6 \text{ m}) - (-900 \text{ N})(1.2 \text{ m})$   
 $+ (-400 \text{ N})(0.6 \text{ m}) = 0$   
 $B_x = 1400 \text{ N}$  *Ans.*  
 $\Sigma F_x = 0$ ;  $1400 \text{ N} + (-900 \text{ N}) + A_x = 0$   
 $A_x = -500 \text{ N}$  *Ans.*  
 $\Sigma F_z = 0$ ;  $A_z - 933.3 \text{ N} + 600 \text{ N} = 0$   
 $A_z = 333.3 \text{ N}$  *Ans.*

**F5-10.**  $\Sigma F_x = 0$ ;  $B_x = 0$  *Ans.*  
 $\Sigma M_z = 0$ ;  
 $C_y(0.4 \text{ m} + 0.6 \text{ m}) = 0$   $C_y = 0$  *Ans.*  
 $\Sigma F_y = 0$ ;  $A_y + 0 = 0$   $A_y = 0$  *Ans.*  
 $\Sigma M_x = 0$ ;  $C_z(0.6 \text{ m} + 0.6 \text{ m}) + B_z(0.6 \text{ m})$   
 $- 450 \text{ N}(0.6 \text{ m} + 0.6 \text{ m}) = 0$   
 $1.2C_z + 0.6B_z - 540 = 0$   
 $\Sigma M_y = 0$ ;  $-C_z(0.6 \text{ m} + 0.4 \text{ m})$   
 $- B_z(0.6 \text{ m}) + 450 \text{ N}(0.6 \text{ m}) = 0$   
 $-C_z - 0.6B_z + 270 = 0$   
 $C_z = 1350 \text{ N}$   $B_z = -1800 \text{ N}$  *Ans.*  
 $\Sigma F_z = 0$ ;  
 $A_z + 1350 \text{ N} + (-1800 \text{ N}) - 450 \text{ N} = 0$   
 $A_z = 900 \text{ N}$  *Ans.*

**F5-11.**  $\Sigma F_y = 0$ ;  $A_y = 0$  *Ans.*  
 $\Sigma M_x = 0$ ;  $-9(3) + F_{CE}(3) = 0$  *Ans.*  
 $F_{CE} = 9 \text{ kN}$   
 $\Sigma M_z = 0$ ;  $F_{CF}(3) - 6(3) = 0$  *Ans.*  
 $F_{CF} = 6 \text{ kN}$   
 $\Sigma M_y = 0$ ;  $9(4) - A_z(4) - 6(1.5) = 0$  *Ans.*  
 $A_z = 6.75 \text{ kN}$  *Ans.*  
 $\Sigma F_x = 0$ ;  $A_x + 6 - 6 = 0$   $A_x = 0$  *Ans.*  
 $\Sigma F_z = 0$ ;  $F_{DB} + 9 - 9 + 6.75 = 0$  *Ans.*  
 $F_{DB} = -6.75 \text{ kN}$

**F5-12.**  $\Sigma F_x = 0$ ;  $A_x = 0$  *Ans.*  
 $\Sigma F_y = 0$ ;  $A_y = 0$  *Ans.*  
 $\Sigma F_z = 0$ ;  $A_z + F_{BC} - 80 = 0$   
 $\Sigma M_x = 0$ ;  $(M_A)_x + 6F_{BC} - 80(6) = 0$   
 $\Sigma M_y = 0$ ;  $3F_{BC} - 80(1.5) = 0$   $F_{BC} = 40 \text{ lb}$  *Ans.*  
 $\Sigma M_z = 0$ ;  $(M_A)_z = 0$  *Ans.*  
 $A_z = 40 \text{ lb}$   $(M_A)_x = 240 \text{ lb} \cdot \text{ft}$  *Ans.*

## Chapter 6

**F6-1. Joint A.**  
 $+ \uparrow \Sigma F_y = 0$ ;  $225 \text{ lb} - F_{AD} \sin 45^\circ = 0$   
 $F_{AD} = 318.20 \text{ lb} = 318 \text{ lb (C)}$  *Ans.*  
 $\rightarrow \Sigma F_x = 0$ ;  $F_{AB} - (318.20 \text{ lb}) \cos 45^\circ = 0$   
 $F_{AB} = 225 \text{ lb (T)}$  *Ans.*

**Joint B.**  
 $\rightarrow \Sigma F_x = 0$ ;  $F_{BC} - 225 \text{ lb} = 0$   
 $F_{BC} = 225 \text{ lb (T)}$  *Ans.*  
 $+ \uparrow \Sigma F_y = 0$ ;  $F_{BD} = 0$  *Ans.*

**Joint D.**  
 $\rightarrow \Sigma F_x = 0$ ;  
 $F_{CD} \cos 45^\circ + (318.20 \text{ lb}) \cos 45^\circ - 450 \text{ lb} = 0$   
 $F_{CD} = 318.20 \text{ lb} = 318 \text{ lb (T)}$  *Ans.*

**F6-2. Joint D.**  
 $+ \uparrow \Sigma F_y = 0$ ;  $\frac{3}{5} F_{CD} - 300 = 0$ ;  
 $F_{CD} = 500 \text{ lb (T)}$  *Ans.*  
 $\rightarrow \Sigma F_x = 0$ ;  $-F_{AD} + \frac{4}{5}(500) = 0$   
 $F_{AD} = 400 \text{ lb (C)}$  *Ans.*  
 $F_{BC} = 500 \text{ lb (T)}, F_{AC} = F_{AB} = 0$  *Ans.*

**F6-3.**  $D_x = 200 \text{ lb}, D_y = 650 \text{ lb}, B_y = 150 \text{ lb}$   
**Joint B.**  
 $\rightarrow \Sigma F_x = 0$ ;  $F_{BA} = 0$  *Ans.*  
 $+ \uparrow \Sigma F_y = 0$ ;  $150 - F_{BC} = 0$ ;  $F_{BC} = 150 \text{ lb (C)}$  *Ans.*

**Joint A.**  
 $\rightarrow \Sigma F_x = 0$ ;  $F_{AC}(\frac{4}{5}) = 0$ ;  $F_{AC} = 0$  *Ans.*  
 $+ \uparrow \Sigma F_y = 0$ ;  $F_{AD} - 800 = 0$ ;  $F_{AD} = 800 \text{ lb (T)}$  *Ans.*

**Joint C.**  
 $\rightarrow \Sigma F_x = 0$ ;  $-F_{CD} + 200 = 0$ ;  $F_{CD} = 200 \text{ lb (T)}$  *Ans.*

**F6-4.** Joint C.

$$+\uparrow \Sigma F_y = 0; \quad 2F \cos 30^\circ - P = 0$$

$$F_{AC} = F_{BC} = F = \frac{P}{2 \cos 30^\circ} = 0.5774P \text{ (C)}$$

## Joint B.

$$\rightarrow \Sigma F_x = 0; 0.5774P \cos 60^\circ - F_{AB} = 0$$

$$F_{AB} = 0.2887P \text{ (T)}$$

$$F_{AB} = 0.2887P = 2 \text{ kN}$$

$$P = 6.928 \text{ kN}$$

$$F_{AC} = F_{BC} = 0.5774P = 1.5 \text{ kN}$$

$$P = 2.598 \text{ kN}$$

The smaller value of  $P$  is chosen,

$$P = 2.598 \text{ kN} = 2.60 \text{ kN} \quad \text{Ans.}$$

**F6-5.**  $F_{CB} = 0$ 

Ans.

$$F_{CD} = 0$$

Ans.

$$F_{AE} = 0$$

Ans.

$$F_{DE} = 0$$

Ans.

**F6-6.** Joint C.

$$+\uparrow \Sigma F_y = 0; \quad 259.81 \text{ lb} - F_{CD} \sin 30^\circ = 0$$

$$F_{CD} = 519.62 \text{ lb} = 520 \text{ lb (C)} \quad \text{Ans.}$$

$$\rightarrow \Sigma F_x = 0; (519.62 \text{ lb}) \cos 30^\circ - F_{BC} = 0$$

$$F_{BC} = 450 \text{ lb (T)} \quad \text{Ans.}$$

## Joint D.

$$+\nearrow \Sigma F_{y'} = 0; \quad F_{BD} \cos 30^\circ = 0 \quad F_{BD} = 0 \quad \text{Ans.}$$

$$+\searrow \Sigma F_{x'} = 0; \quad F_{DE} - 519.62 \text{ lb} = 0$$

$$F_{DE} = 519.62 \text{ lb} = 520 \text{ lb (C)} \quad \text{Ans.}$$

## Joint B.

$$+\uparrow \Sigma F_y = 0; \quad F_{BE} \sin \phi = 0 \quad F_{BE} = 0 \quad \text{Ans.}$$

$$\rightarrow \Sigma F_x = 0; \quad 450 \text{ lb} - F_{AB} = 0$$

$$F_{AB} = 450 \text{ lb (T)} \quad \text{Ans.}$$

## Joint A.

$$+\uparrow \Sigma F_y = 0; \quad 340.19 \text{ lb} - F_{AE} = 0$$

$$F_{AE} = 340 \text{ lb (C)} \quad \text{Ans.}$$

**F6-7.**  $+\uparrow \Sigma F_y = 0; F_{CF} \sin 45^\circ - 600 - 800 = 0$ 

$$F_{CF} = 1980 \text{ lb (T)} \quad \text{Ans.}$$

$$\zeta + \Sigma M_C = 0; F_{FE}(4) - 800(4) = 0$$

$$F_{FE} = 800 \text{ lb (T)} \quad \text{Ans.}$$

$$\zeta + \Sigma M_F = 0; F_{BC}(4) - 600(4) - 800(8) = 0$$

$$F_{BC} = 2200 \text{ lb (C)} \quad \text{Ans.}$$

$$\text{F6-8.} \quad \zeta + \Sigma M_A = 0; \quad G_y(12 \text{ m}) - 20 \text{ kN}(2 \text{ m}) - 30 \text{ kN}(4 \text{ m}) - 40 \text{ kN}(6 \text{ m}) = 0$$

$$G_y = 33.33 \text{ kN}$$

$$+\uparrow \Sigma F_y = 0; F_{KC} + 33.33 \text{ kN} - 40 \text{ kN} = 0$$

$$F_{KC} = 6.67 \text{ kN (C)} \quad \text{Ans.}$$

$$\zeta + \Sigma M_K = 0;$$

$$33.33 \text{ kN}(8 \text{ m}) - 40 \text{ kN}(2 \text{ m}) - F_{CD}(3 \text{ m}) = 0$$

$$F_{CD} = 62.22 \text{ kN} = 62.2 \text{ kN (T)} \quad \text{Ans.}$$

$$\rightarrow \Sigma F_x = 0; \quad F_{LK} - 62.22 \text{ kN} = 0$$

$$F_{LK} = 62.2 \text{ kN (C)} \quad \text{Ans.}$$

**F6-9.** From the geometry of the truss,

$$\phi = \tan^{-1}(3 \text{ m}/2 \text{ m}) = 56.31^\circ.$$

$$\zeta + \Sigma M_K = 0;$$

$$33.33 \text{ kN}(8 \text{ m}) - 40 \text{ kN}(2 \text{ m}) - F_{CD}(3 \text{ m}) = 0$$

$$F_{CD} = 62.2 \text{ kN (T)} \quad \text{Ans.}$$

$$\zeta + \Sigma M_D = 0; \quad 33.33 \text{ kN}(6 \text{ m}) - F_{KJ}(3 \text{ m}) = 0$$

$$F_{KJ} = 66.7 \text{ kN (C)} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0;$$

$$33.33 \text{ kN} - 40 \text{ kN} + F_{KD} \sin 56.31^\circ = 0$$

$$F_{KD} = 8.01 \text{ kN (T)} \quad \text{Ans.}$$

**F6-10.** From the geometry of the truss,

$$\tan \phi = \frac{(9 \text{ ft}) \tan 30^\circ}{3 \text{ ft}} = 1.732 \quad \phi = 60^\circ$$

$$\zeta + \Sigma M_C = 0;$$

$$F_{EF} \sin 30^\circ(6 \text{ ft}) + 300 \text{ lb}(6 \text{ ft}) = 0$$

$$F_{EF} = -600 \text{ lb} = 600 \text{ lb (C)} \quad \text{Ans.}$$

$$\zeta + \Sigma M_D = 0;$$

$$300 \text{ lb}(6 \text{ ft}) - F_{CF} \sin 60^\circ(6 \text{ ft}) = 0$$

$$F_{CF} = 346.41 \text{ lb} = 346 \text{ lb (T)} \quad \text{Ans.}$$

$$\zeta + \Sigma M_F = 0;$$

$$300 \text{ lb}(9 \text{ ft}) - 300 \text{ lb}(3 \text{ ft}) - F_{BC}(9 \text{ ft}) \tan 30^\circ = 0$$

$$F_{BC} = 346.41 \text{ lb} = 346 \text{ lb (T)} \quad \text{Ans.}$$

**F6-11.** From the geometry of the truss,

$$\theta = \tan^{-1}(1 \text{ m}/2 \text{ m}) = 26.57^\circ$$

$$\phi = \tan^{-1}(3 \text{ m}/2 \text{ m}) = 56.31^\circ.$$

The location of  $O$  can be found using similar triangles.

$$\frac{1 \text{ m}}{2 \text{ m}} = \frac{2 \text{ m}}{2 \text{ m} + x}$$

$$4 \text{ m} = 2 \text{ m} + x$$

$$x = 2 \text{ m}$$



$$\zeta + \Sigma M_G = 0;$$

$$26.25 \text{ kN}(4 \text{ m}) - 15 \text{ kN}(2 \text{ m}) - F_{CD}(3 \text{ m}) = 0$$

$$F_{CD} = 25 \text{ kN (T)} \quad \text{Ans.}$$

$$\zeta + \Sigma M_D = 0;$$

$$26.25 \text{ kN}(2 \text{ m}) - F_{GF} \cos 26.57^\circ(2 \text{ m}) = 0$$

$$F_{GF} = 29.3 \text{ kN (C)} \quad \text{Ans.}$$

$$\zeta + \Sigma M_O = 0; 15 \text{ kN}(4 \text{ m}) - 26.25 \text{ kN}(2 \text{ m})$$

$$- F_{GD} \sin 56.31^\circ(4 \text{ m}) = 0$$

$$F_{GD} = 2.253 \text{ kN} = 2.25 \text{ kN (T)} \quad \text{Ans.}$$

**F6-12.**  $\zeta + \Sigma M_H = 0;$

$$F_{DC}(12 \text{ ft}) + 1200 \text{ lb}(9 \text{ ft}) - 1600 \text{ lb}(21 \text{ ft}) = 0$$

$$F_{DC} = 1900 \text{ lb (C)} \quad \text{Ans.}$$

$$\zeta + \Sigma M_D = 0;$$

$$1200 \text{ lb}(21 \text{ ft}) - 1600 \text{ lb}(9 \text{ ft}) - F_{HI}(12 \text{ ft}) = 0$$

$$F_{HI} = 900 \text{ lb (C)} \quad \text{Ans.}$$

$$\zeta + \Sigma M_C = 0; F_{JI} \cos 45^\circ(12 \text{ ft}) + 1200 \text{ lb}(21 \text{ ft})$$

$$- 900 \text{ lb}(12 \text{ ft}) - 1600 \text{ lb}(9 \text{ ft}) = 0$$

$$F_{JI} = 0 \quad \text{Ans.}$$

**F6-13.**  $+\uparrow \Sigma F_y = 0; 3P - 60 = 0$

$$P = 20 \text{ lb} \quad \text{Ans.}$$

**F6-14.**  $\zeta + \Sigma M_C = 0;$

$$-\left(\frac{4}{5}\right)(F_{AB})(9) + 400(6) + 500(3) = 0$$

$$F_{AB} = 541.67 \text{ lb}$$

$$+\rightarrow \Sigma F_x = 0; -C_x + \frac{3}{5}(541.67) = 0$$

$$C_x = 325 \text{ lb} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; C_y + \frac{4}{5}(541.67) - 400 - 500 = 0$$

$$C_y = 467 \text{ lb} \quad \text{Ans.}$$

**F6-15.**  $\zeta + \Sigma M_A = 0; 100 \text{ N}(250 \text{ mm}) - N_B(50 \text{ mm}) = 0$

$$N_B = 500 \text{ N} \quad \text{Ans.}$$

$$+\rightarrow \Sigma F_x = 0; (500 \text{ N}) \sin 45^\circ - A_x = 0$$

$$A_x = 353.55 \text{ N}$$

$$+\uparrow \Sigma F_y = 0; A_y - 100 \text{ N} - (500 \text{ N}) \cos 45^\circ = 0$$

$$A_y = 453.55 \text{ N}$$

$$F_A = \sqrt{(353.55 \text{ N})^2 + (453.55 \text{ N})^2}$$

$$= 575 \text{ N} \quad \text{Ans.}$$

**F6-16.**  $\zeta + \Sigma M_C = 0;$

$$400(2) + 800 - F_{BA} \left(\frac{3}{\sqrt{10}}\right)(1) - F_{BA} \left(\frac{1}{\sqrt{10}}\right)(3) = 0$$

$$F_{BA} = 843.27 \text{ N}$$

$$+\rightarrow \Sigma F_x = 0; C_x - 843.27 \left(\frac{3}{\sqrt{10}}\right) = 0$$

$$C_x = 800 \text{ N} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; C_y + 843.27 \left(\frac{1}{\sqrt{10}}\right) - 400 = 0$$

$$C_y = 133 \text{ N} \quad \text{Ans.}$$

**F6-17.** Plate A:

$$+\uparrow \Sigma F_y = 0; 2T + N_{AB} - 100 = 0$$

Plate B:

$$+\uparrow \Sigma F_y = 0; 2T - N_{AB} - 30 = 0$$

$$T = 32.5 \text{ lb}, N_{AB} = 35 \text{ lb} \quad \text{Ans.}$$

**F6-18.** Pulley C:

$$+\uparrow \Sigma F_y = 0; T - 2P = 0; T = 2P$$

Beam:

$$+\uparrow \Sigma F_y = 0; 2P + P - 6 = 0$$

$$P = 2 \text{ kN} \quad \text{Ans.}$$

$$\zeta + \Sigma M_A = 0; 2(1) - 6(x) = 0$$

$$x = 0.333 \text{ m} \quad \text{Ans.}$$

**F6-19.** Member CD

$$\zeta + \Sigma M_D = 0; 600(1.5) - N_C(3) = 0$$

$$N_C = 300 \text{ N}$$

Member ABC

$$\zeta + \Sigma M_A = 0; -800 + B_y(2) - (300 \sin 45^\circ) 4 = 0$$

$$B_y = 824.26 = 824 \text{ N} \quad \text{Ans.}$$

$$+\rightarrow \Sigma F_x = 0; A_x - 300 \cos 45^\circ = 0;$$

$$A_x = 212 \text{ N} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; -A_y + 824.26 - 300 \sin 45^\circ = 0;$$

$$A_y = 612 \text{ N} \quad \text{Ans.}$$

**F6-20.** AB is a two-force member.

Member BC

$$\zeta + \Sigma M_c = 0; 15(3) + 10(6) - F_{BC} \left(\frac{4}{5}\right)(9) = 0$$

$$F_{BC} = 14.58 \text{ kN}$$

$$+\rightarrow \Sigma F_x = 0; (14.58) \left(\frac{3}{5}\right) - C_x = 0;$$

$$C_x = 8.75 \text{ kN}$$

$$+\uparrow \Sigma F_y = 0; (14.58) \left(\frac{4}{5}\right) - 10 - 15 + C_y = 0;$$

$$C_y = 13.3 \text{ kN}$$

Member CD

$$+\rightarrow \Sigma F_x = 0; 8.75 - D_x = 0; D_x = 8.75 \text{ kN} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; -13.3 + D_y = 0; D_y = 13.3 \text{ kN} \quad \text{Ans.}$$

$$\zeta + \Sigma M_D = 0; -8.75(4) + M_D = 0; M_D = 35 \text{ kN} \cdot \text{m} \quad \text{Ans.}$$

**F6-21.** Entire frame

$$\zeta + \Sigma M_A = 0; \quad -600(3) - [400(3)](1.5) + C_y(3) = 0$$

$$C_y = 1200 \text{ N} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad A_y - 400(3) + 1200 = 0$$

$$A_y = 0 \quad \text{Ans.}$$

$$\rightarrow \Sigma F_x = 0; \quad 600 - A_x - C_x = 0$$

Member *AB*

$$\zeta + \Sigma M_B = 0; \quad 400(1.5)(0.75) - A_x(3) = 0$$

$$A_x = 150 \text{ N} \quad \text{Ans.}$$

$$C_x = 450 \text{ N} \quad \text{Ans.}$$

These same results can be obtained by considering members *AB* and *BC*.

**F6-22.** Entire frame

$$\zeta + \Sigma M_E = 0; \quad 250(6) - A_y(6) = 0$$

$$A_y = 250 \text{ N}$$

$$\rightarrow \Sigma F_x = 0; \quad E_x = 0$$

$$+\uparrow \Sigma F_y = 0; \quad 250 - 250 + E_y = 0; \quad E_y = 0$$

Member *BD*

$$\zeta + \Sigma M_D = 0; \quad 250(4.5) - B_y(3) = 0;$$

$$B_y = 375 \text{ N}$$

Member *ABC*

$$\zeta + \Sigma M_C = 0; \quad -250(3) + 375(1.5) + B_x(2) = 0$$

$$B_x = 93.75 \text{ N}$$

$$\rightarrow \Sigma F_x = 0; \quad C_x - B_x = 0; \quad C_x = 93.75 \text{ N} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad 250 - 375 + C_y = 0; \quad C_y = 125 \text{ N} \quad \text{Ans.}$$

**F6-23.** *AD*, *CB* are two-force members.Member *AB*

$$\zeta + \Sigma M_A = 0; \quad -\left[\frac{1}{2}(3)(4)\right](1.5) + B_y(3) = 0$$

$$B_y = 3 \text{ kN}$$

Since *BC* is a two-force member  $C_y = B_y = 3 \text{ kN}$  and  $C_x = 0$  ( $\Sigma M_B = 0$ ).

Member *EDC*

$$\zeta + \Sigma M_E = 0; \quad F_{DA}\left(\frac{4}{3}\right)(1.5) - 5(3) - 3(3) = 0$$

$$F_{DA} = 20 \text{ kN}$$

$$\rightarrow \Sigma F_x = 0; \quad E_x - 20\left(\frac{3}{5}\right) = 0; \quad E_x = 12 \text{ kN} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad -E_y + 20\left(\frac{4}{5}\right) - 5 - 3 = 0;$$

$$E_y = 8 \text{ kN} \quad \text{Ans.}$$

**F6-24.** *AC* and *DC* are two-force members.Member *BC*

$$\zeta + \Sigma M_C = 0; \quad \left[\frac{1}{2}(3)(8)\right](1) - B_y(3) = 0$$

$$B_y = 4 \text{ kN}$$

Member *BA*

$$\zeta + \Sigma M_B = 0; \quad 6(2) - A_x(4) = 0$$

$$A_x = 3 \text{ kN} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad -4 \text{ kN} + A_y = 0; \quad A_y = 4 \text{ kN} \quad \text{Ans.}$$

Entire Frame

$$\zeta + \Sigma M_A = 0; \quad -6(2) - \left[\frac{1}{2}(3)(8)\right](2) + D_y(3) = 0$$

$$D_y = 12 \text{ kN} \quad \text{Ans.}$$

Since *DC* is a two-force member ( $\Sigma M_C = 0$ ) then

$$D_x = 0 \quad \text{Ans.}$$

**Chapter 7****F7-1.**  $\zeta + \Sigma M_A = 0; \quad B_y(6) - 10(1.5) - 15(4.5) = 0$ 

$$B_y = 13.75 \text{ kN}$$

$$\rightarrow \Sigma F_x = 0; \quad N_C = 0 \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad V_C + 13.75 - 15 = 0$$

$$V_C = 1.25 \text{ kN} \quad \text{Ans.}$$

$$\zeta + \Sigma M_C = 0; \quad 13.75(3) - 15(1.5) - M_C = 0$$

$$M_C = 18.75 \text{ kN} \cdot \text{m} \quad \text{Ans.}$$

**F7-2.**  $\zeta + \Sigma M_B = 0; \quad 30 - 10(1.5) - A_y(3) = 0$ 

$$A_y = 5 \text{ kN}$$

$$\rightarrow \Sigma F_x = 0; \quad N_C = 0 \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad 5 - V_C = 0$$

$$V_C = 5 \text{ kN} \quad \text{Ans.}$$

$$\zeta + \Sigma M_C = 0; \quad M_C + 30 - 5(1.5) = 0$$

$$M_C = -22.5 \text{ kN} \cdot \text{m} \quad \text{Ans.}$$

**F7-3.**  $\rightarrow \Sigma F_x = 0; \quad B_x = 0$ 

$$\zeta + \Sigma M_A = 0; \quad 3(6)(3) - B_y(9) = 0$$

$$B_y = 6 \text{ kip}$$

$$\rightarrow \Sigma F_x = 0; \quad N_C = 0 \quad \text{Ans.}$$

$$\zeta + \uparrow \Sigma F_y = 0; \quad V_C - 6 = 0$$

$$V_C = 6 \text{ kip} \quad \text{Ans.}$$

$$\zeta + \Sigma M_C = 0; \quad -M_C - 6(4.5) = 0$$

$$M_C = -27 \text{ kip} \cdot \text{ft} \quad \text{Ans.}$$

**F7-4.**  $\zeta + \Sigma M_A = 0; \quad B_y(6) - 12(1.5) - 9(3)(4.5) = 0$ 

$$B_y = 23.25 \text{ kN}$$

$$\rightarrow \Sigma F_x = 0; \quad N_C = 0 \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad V_C + 23.25 - 9(1.5) = 0$$

$$V_C = -9.75 \text{ kN} \quad \text{Ans.}$$

$$\zeta + \Sigma M_C = 0;$$

$$23.25(1.5) - 9(1.5)(0.75) - M_C = 0$$

$$M_C = 24.75 \text{ kN} \cdot \text{m} \quad \text{Ans.}$$

**F7-5.**  $\zeta + \Sigma M_A = 0; B_y(6) - \frac{1}{2}(9)(6)(3) = 0$   
 $B_y = 13.5 \text{ kN}$   
 $+\Sigma F_x = 0; N_C = 0$  *Ans.*  
 $+\uparrow \Sigma F_y = 0; V_C + 13.5 - \frac{1}{2}(9)(3) = 0$   
 $V_C = 0$  *Ans.*  
 $\zeta + \Sigma M_C = 0; 13.5(3) - \frac{1}{2}(9)(3)(1) - M_C = 0$   
 $M_C = 27 \text{ kN} \cdot \text{m}$  *Ans.*

**F7-6.**  $\zeta + \Sigma M_A = 0;$   
 $B_y(6) - \frac{1}{2}(6)(3)(2) - 6(3)(4.5) = 0$   
 $B_y = 16.5 \text{ kN}$   
 $+\Sigma F_x = 0; N_C = 0$  *Ans.*  
 $+\uparrow \Sigma F_y = 0; V_C + 16.5 - 6(3) = 0$   
 $V_C = 1.50 \text{ kN}$  *Ans.*  
 $\zeta + \Sigma M_C = 0; 16.5(3) - 6(3)(1.5) - M_C = 0$   
 $M_C = 22.5 \text{ kN} \cdot \text{m}$  *Ans.*

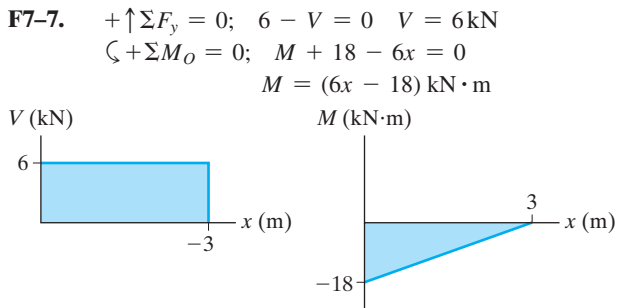


Fig. F7-7

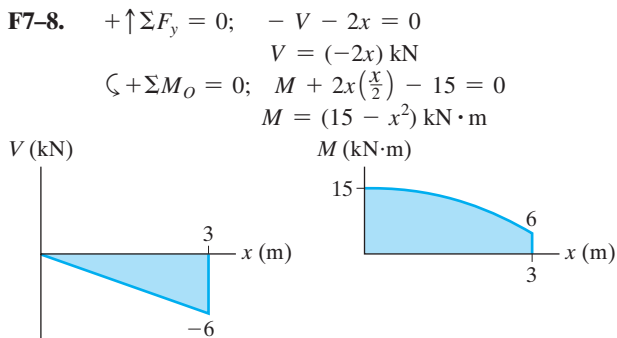


Fig. F7-8

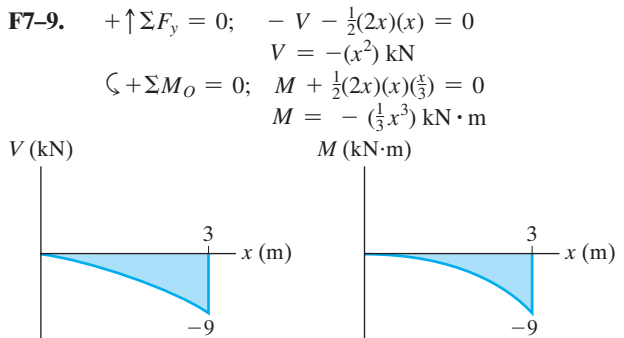


Fig. F7-9

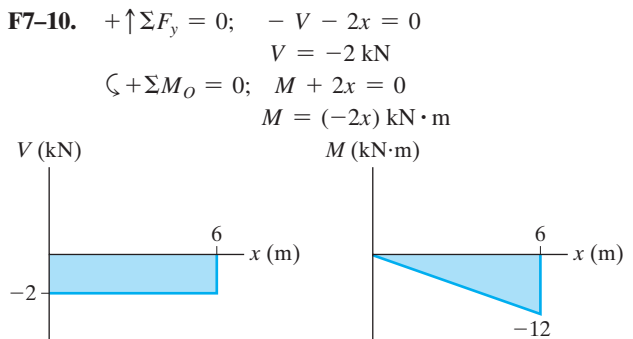


Fig. F7-10

**F7-11.** Region  $3 \text{ m} \leq x < 3 \text{ m}$   
 $+\uparrow \Sigma F_y = 0; -V - 5 = 0 \quad V = -5 \text{ kN}$   
 $\zeta + \Sigma M_O = 0; M + 5x = 0$   
 $M = (-5x) \text{ kN} \cdot \text{m}$

Region  $0 < x \leq 6 \text{ m}$

$+\uparrow \Sigma F_y = 0; V + 5 = 0 \quad V = -5 \text{ kN}$   
 $\zeta + \Sigma M_O = 0; 5(6 - x) - M = 0$   
 $M = (5(6 - x)) \text{ kN} \cdot \text{m}$

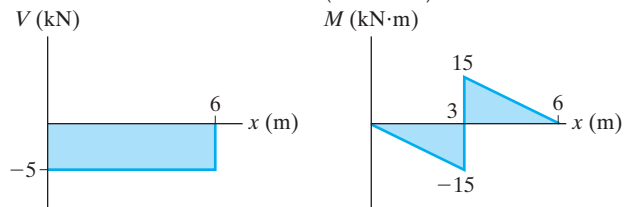


Fig. F7-11

**F7-12.** Region  $0 \leq x < 3$  m

$$+\uparrow \Sigma F_y = 0; \quad V = 0$$

$$\zeta + \Sigma M_O = 0; \quad M - 12 = 0$$

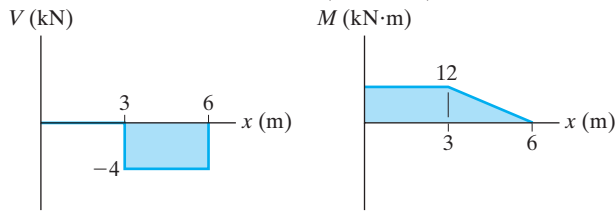
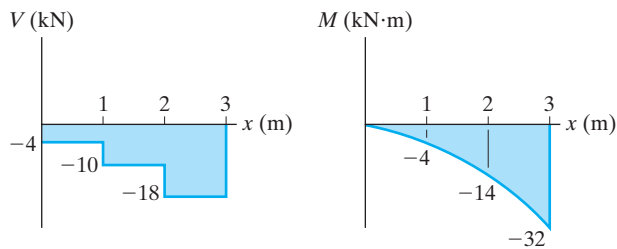
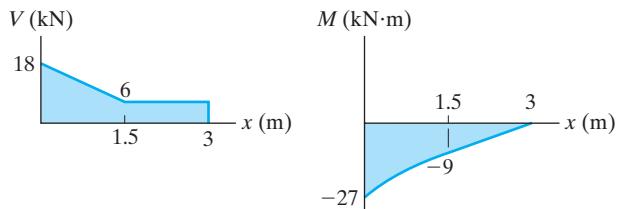
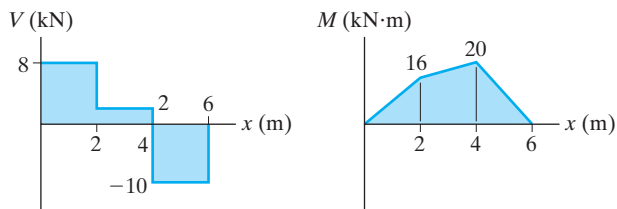
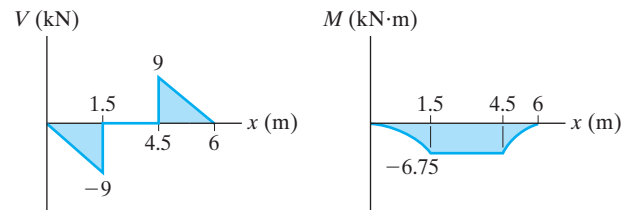
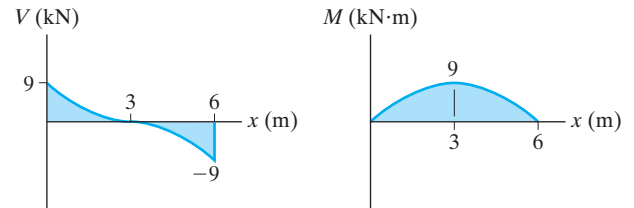
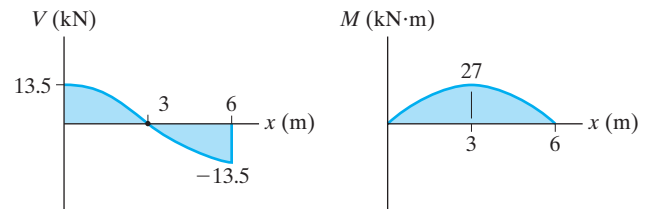
$$M = 12 \text{ kN} \cdot \text{m}$$

Region  $3 \text{ m} < x \leq 6$  m

$$+\uparrow \Sigma F_y = 0; \quad V + 4 = 0 \quad V = -4 \text{ kN}$$

$$\zeta + \Sigma M_O = 0; \quad 4(6 - x) - M = 0$$

$$M = (4(6 - x)) \text{ kN} \cdot \text{m}$$

**Fig. F7-12****F7-13.****Fig. F7-13****F7-14.****Fig. F7-14****F7-15.****Fig. F7-15****F7-16.****Fig. F7-16****F7-17.****Fig. F7-17****F7-18.****Fig. F7-18**

## Chapter 8

**F8-1.** a)  $+\uparrow \Sigma F_y = 0; \quad N - 50(9.81) - 200\left(\frac{3}{5}\right) = 0$

$$N = 610.5 \text{ N}$$

$$\rightarrow \Sigma F_x = 0; \quad F - 200\left(\frac{4}{5}\right) = 0$$

$$F = 160 \text{ N}$$

$$F < F_{\max} = \mu_s N = 0.3(610.5) = 183.15 \text{ N},$$

therefore  $F = 160 \text{ N}$ *Ans.*

b)  $+\uparrow \Sigma F_y = 0; \quad N - 50(9.81) - 400\left(\frac{3}{5}\right) = 0$

$$N = 730.5 \text{ N}$$

$$\rightarrow \Sigma F_x = 0; \quad F - 400\left(\frac{4}{5}\right) = 0$$

$$F = 320 \text{ N}$$

$$F > F_{\max} = \mu_s N = 0.3(730.5) = 219.15 \text{ N}$$

Block slips

$$F = \mu_s N = 0.2(730.5) = 146 \text{ N}$$

*Ans.*

**F8-2.**  $\zeta + \Sigma M_B = 0;$

$$N_A(3) + 0.2N_A(4) - 30(9.81)(2) = 0$$

$$N_A = 154.89 \text{ N}$$

$$\rightarrow \Sigma F_x = 0; \quad P - 154.89 = 0$$

$$P = 154.89 \text{ N} = 155 \text{ N}$$

*Ans.*

**F8-3.** Crate *A*

$$+\uparrow \Sigma F_y = 0; \quad N_A - 50(9.81) = 0$$

$$N_A = 490.5 \text{ N}$$

$$\rightarrow \Sigma F_x = 0; \quad T - 0.25(490.5) = 0$$

$$T = 122.62 \text{ N}$$

Crate *B*

$$+\uparrow \Sigma F_y = 0; \quad N_B + P \sin 30^\circ - 50(9.81) = 0$$

$$N_B = 490.5 - 0.5P$$

$$\rightarrow \Sigma F_x = 0;$$

$$P \cos 30^\circ - 0.25(490.5 - 0.5P) - 122.62 = 0$$

$$P = 247 \text{ N}$$

*Ans.*

**F8-4.**  $\rightarrow \Sigma F_x = 0; \quad N_A - 0.3N_B = 0$

$$+\uparrow \Sigma F_y = 0;$$

$$N_B + 0.3N_A + P - 100(9.81) = 0$$

$$\zeta + \Sigma M_O = 0;$$

$$P(0.6) - 0.3N_B(0.9) - 0.3N_A(0.9) = 0$$

$$N_A = 175.70 \text{ N} \quad N_B = 585.67 \text{ N}$$

$$P = 343 \text{ N}$$

*Ans.*

**F8-5.** If slipping occurs:

$$+\uparrow \Sigma F_y = 0; \quad N_c - 250 \text{ lb} = 0; \quad N_c = 250 \text{ lb}$$

$$\rightarrow \Sigma F_x = 0; \quad P - 0.4(250) = 0; \quad P = 100 \text{ lb}$$

If tipping occurs:

$$\zeta + \Sigma M_A = 0; \quad -P(4.5) + 250(1.5) = 0$$

$$P = 83.3 \text{ lb}$$

*Ans.*

**F8-6.**

$$\zeta + \Sigma M_A = 0; \quad 490.5(0.6) - T \cos 60^\circ (0.3 \cos 60^\circ + 0.6) \\ - T \sin 60^\circ (0.3 \sin 60^\circ) = 0$$

$$T = 490.5 \text{ N}$$

$$\rightarrow \Sigma F_x = 0; \quad 490.5 \sin 60^\circ - N_A = 0; \quad N_A = 424.8 \text{ N}$$

$$+\uparrow \Sigma F_y = 0; \quad \mu_s(424.8) + 490.5 \cos 60^\circ - 490.5 = 0$$

$$\mu_s = 0.577$$

*Ans.*

**F8-7.** *A* will not move. Assume *B* is about to slip on *C* and *A*, and *C* is stationary.

$$\rightarrow \Sigma F_x = 0; \quad P - 0.3(50) - 0.4(75); \quad P = 45 \text{ N}$$

Assume *C* is about to slip and *B* does not slip on *C*, but is about to slip at *A*.

$$\rightarrow \Sigma F_x = 0; \quad P - 0.3(50) - 0.35(90) = 0$$

$$P = 46.5 \text{ N} > 45 \text{ N}$$

$$P = 45 \text{ N}$$

*Ans.*

**F8-8.** *A* is about to move down the plane and *B* moves upward.

Block *A*

$$+\nearrow \Sigma F_y = 0; \quad N = W \cos \theta$$

$$+\nearrow \Sigma F_x = 0; \quad T + \mu_s(W \cos \theta) - W \sin \theta = 0$$

$$T = W \sin \theta - \mu_s W \cos \theta \quad (1)$$

Block *B*

$$+\nearrow \Sigma F_y = 0; \quad N' = 2W \cos \theta$$

$$+\nearrow \Sigma F_x = 0; \quad 2T - \mu_s W \cos \theta - \mu_s(2W \cos \theta) \\ - W \sin \theta = 0$$

Using Eq.(1);

$$\theta = \tan^{-1} 5\mu_s$$

*Ans.*

**F8-9.** Assume *B* is about to slip on *A*,  $F_B = 0.3 N_B$ .

$$\rightarrow \Sigma F_x = 0; \quad P - 0.3(10)(9.81) = 0$$

$$P = 29.4 \text{ N}$$

Assume *B* is about to tip on *A*,  $x = 0$ .

$$\zeta + \Sigma M_O = 0; \quad 10(9.81)(0.15) - P(0.4) = 0$$

$$P = 36.8 \text{ N}$$

Assume *A* is about to slip,  $F_A = 0.1 N_A$ .

$$\rightarrow \Sigma F_x = 0 \quad P - 0.1[7(9.81) + 10(9.81)] = 0$$

$$P = 16.7 \text{ N}$$

Choose the smallest result.  $P = 16.7 \text{ N}$

*Ans.*

## Chapter 9

**F9-1.**  $\bar{x} = \frac{\int_A \tilde{x} dA}{\int_A dA} = \frac{\frac{1}{2} \int_0^{1 \text{ m}} y^{2/3} dy}{\int_0^{1 \text{ m}} y^{1/3} dy} = 0.4 \text{ m}$  *Ans.*

$\bar{y} = \frac{\int_A \tilde{y} dA}{\int_A dA} = \frac{\int_0^{1 \text{ m}} y^{4/3} dy}{\int_0^{1 \text{ m}} y^{1/3} dy} = 0.571 \text{ m}$  *Ans.*

$$\text{F9-2.} \quad \bar{x} = \frac{\int_A \tilde{x} dA}{\int_A dA} = \frac{\int_0^{1\text{ m}} x(x^3 dx)}{\int_0^{1\text{ m}} x^3 dx}$$

$$= 0.8\text{ m}$$

$$\bar{y} = \frac{\int_A \tilde{y} dA}{\int_A dA} = \frac{\int_0^{1\text{ m}} \frac{1}{2} x^3 (x^3 dx)}{\int_0^{1\text{ m}} x^3 dx}$$

$$= 0.286\text{ m}$$

$$\text{F9-3.} \quad \bar{y} = \frac{\int_A \tilde{y} dA}{\int_A dA} = \frac{\int_0^{2\text{ m}} y \left( 2 \left( \frac{y^{1/2}}{\sqrt{2}} \right) \right) dy}{\int_0^{2\text{ m}} 2 \left( \frac{y^{1/2}}{\sqrt{2}} \right) dy}$$

$$= 1.2\text{ m}$$

$$\text{F9-4.} \quad \bar{x} = \frac{\int_m \tilde{x} dm}{\int_m dm} = \frac{\int_0^L x \left[ m_0 \left( 1 + \frac{x^2}{L^2} \right) dx \right]}{\int_0^L m_0 \left( 1 + \frac{x^2}{L^2} \right) dx}$$

$$= \frac{9}{16} L$$

$$\text{F9-5.} \quad \bar{y} = \frac{\int_V \tilde{y} dV}{\int_V dV} = \frac{\int_0^{1\text{ m}} y \left( \frac{\pi}{4} y dy \right)}{\int_0^{1\text{ m}} \frac{\pi}{4} y dy}$$

$$= 0.667\text{ m}$$

$$\text{F9-6.} \quad \bar{z} = \frac{\int_V \tilde{z} dV}{\int_V dV} = \frac{\int_0^{2\text{ ft}} z \left[ \frac{9\pi}{64} (4 - z)^2 dz \right]}{\int_0^{2\text{ ft}} \frac{9\pi}{64} (4 - z)^2 dz}$$

$$= 0.786\text{ ft}$$

$$\text{F9-7.} \quad \bar{x} = \frac{\Sigma \tilde{x} L}{\Sigma L} = \frac{150(300) + 300(600) + 300(400)}{300 + 600 + 400}$$

$$= 265\text{ mm}$$

$$\bar{y} = \frac{\Sigma \tilde{y} L}{\Sigma L} = \frac{0(300) + 300(600) + 600(400)}{300 + 600 + 400}$$

$$= 323\text{ mm}$$

$$\bar{z} = \frac{\Sigma \tilde{z} L}{\Sigma L} = \frac{0(300) + 0(600) + (-200)(400)}{300 + 600 + 400}$$

$$= -61.5\text{ mm}$$

$$\text{F9-8.} \quad \bar{y} = \frac{\Sigma \tilde{y} A}{\Sigma A} = \frac{150[300(50)] + 325[50(300)]}{300(50) + 50(300)}$$

$$= 237.5\text{ mm}$$

$$\text{F9-9.} \quad \bar{y} = \frac{\Sigma \tilde{y} A}{\Sigma A} = \frac{100[2(200)(50)] + 225[50(400)]}{2(200)(50) + 50(400)}$$

$$= 162.5\text{ mm}$$

$$\text{F9-10.} \quad \bar{x} = \frac{\Sigma \tilde{x} A}{\Sigma A} = \frac{0.25[4(0.5)] + 1.75[0.5(2.5)]}{4(0.5) + 0.5(2.5)}$$

$$= 0.827\text{ in.}$$

$$\bar{y} = \frac{\Sigma \tilde{y} A}{\Sigma A} = \frac{2[4(0.5)] + 0.25[(0.5)(2.5)]}{4(0.5) + (0.5)(2.5)}$$

$$= 1.33\text{ in.}$$

$$\text{F9-11.} \quad \bar{x} = \frac{\Sigma \tilde{x} V}{\Sigma V} = \frac{1[2(7)(6)] + 4[4(2)(3)]}{2(7)(6) + 4(2)(3)}$$

$$= 1.67\text{ ft}$$

$$\bar{y} = \frac{\Sigma \tilde{y} V}{\Sigma V} = \frac{3.5[2(7)(6)] + 1[4(2)(3)]}{2(7)(6) + 4(2)(3)}$$

$$= 2.94\text{ ft}$$

$$\bar{z} = \frac{\Sigma \tilde{z} V}{\Sigma V} = \frac{3[2(7)(6)] + 1.5[4(2)(3)]}{2(7)(6) + 4(2)(3)}$$

$$= 2.67\text{ ft}$$

$$\text{F9-12.} \quad \bar{x} = \frac{\Sigma \tilde{x} V}{\Sigma V}$$

$$= \frac{0.25[0.5(2.5)(1.8)] + 0.25 \left[ \frac{1}{2}(1.5)(1.8)(0.5) \right] + (1.0) \left[ \frac{1}{2}(1.5)(1.8)(0.5) \right]}{0.5(2.5)(1.8) + \frac{1}{2}(1.5)(1.8)(0.5) + \frac{1}{2}(1.5)(1.8)(0.5)}$$

$$= 0.391\text{ m}$$

$$\bar{y} = \frac{\Sigma \tilde{y} V}{\Sigma V} = \frac{5.00625}{3.6} = 1.39\text{ m}$$

$$\bar{z} = \frac{\Sigma \tilde{z} V}{\Sigma V} = \frac{2.835}{3.6} = 0.7875\text{ m}$$

$$\text{F9-13.} \quad A = 2\pi \Sigma \tilde{r} L$$

$$= 2\pi [0.75(1.5) + 1.5(2) + 0.75\sqrt{(1.5)^2 + (2)^2}]$$

$$= 37.7\text{ m}^2$$

$$V = 2\pi \Sigma \tilde{r} A$$

$$= 2\pi [0.75(1.5)(2) + 0.5 \left( \frac{1}{2} \right) (1.5)(2)]$$

$$= 18.8\text{ m}^3$$

Ans.

Ans.

Ans.

Ans.

Ans.

Ans.

Ans.

Ans.

Ans.

Ans.

Ans.

Ans.

Ans.

Ans.

Ans.

Ans.

Ans.

Ans.

Ans.

Ans.

Ans.

$$\begin{aligned}
 \mathbf{F9-14.} \quad A &= 2\pi\Sigma\tilde{r}L \\
 &= 2\pi[1.95\sqrt{(0.9)^2 + (1.2)^2} + 2.4(1.5) + 1.95(0.9) + 1.5(2.7)] \\
 &= 77.5 \text{ m}^2 \quad \text{Ans.} \\
 V &= 2\pi\Sigma\tilde{r}A \\
 &= 2\pi[1.8(\tfrac{1}{2})(0.9)(1.2) + 1.95(0.9)(1.5)] \\
 &= 22.6 \text{ m}^3 \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F9-15.} \quad A &= 2\pi\Sigma\tilde{r}L \\
 &= 2\pi[7.5(15) + 15(18) + 22.5\sqrt{15^2 + 20^2} + 15(30)] \\
 &= 8765 \text{ in.}^2 \quad \text{Ans.} \\
 V &= 2\pi\Sigma\tilde{r}A \\
 &= 2\pi[7.5(15)(38) + 20(\tfrac{1}{2})(15)(20)] \\
 &= 45\,710 \text{ in.}^3 \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F9-16.} \quad A &= 2\pi\Sigma\tilde{r}L \\
 &= 2\pi[\tfrac{2(1.5)}{\pi}(\tfrac{\pi(1.5)}{2}) + 1.5(2) + 0.75(1.5)] \\
 &= 40.1 \text{ m}^2 \quad \text{Ans.} \\
 V &= 2\pi\Sigma\tilde{r}A \\
 &= 2\pi[\tfrac{4(1.5)}{3\pi}(\tfrac{\pi(1.5^2)}{4}) + 0.75(1.5)(2)] \\
 &= 21.2 \text{ m}^3 \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F9-17.} \quad w_b &= \rho_w g h b = 1000(9.81)(6)(1) \\
 &= 58.86 \text{ kN/m} \\
 F_R &= \tfrac{1}{2}(58.76)(6) = 176.58 \text{ kN} = 177 \text{ kN} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F9-18.} \quad w_b &= \gamma_w h b = 62.4(4)(4) = 998.4 \text{ lb/ft} \\
 F_R &= 998.4(3) = 3.00 \text{ kip} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F9-19.} \quad w_b &= \rho_w g h b = 1000(9.81)(2)(1.5) \\
 &= 29.43 \text{ kN/m} \\
 F_R &= \tfrac{1}{2}(29.43)(\sqrt{(1.5)^2 + (2)^2}) \\
 &= 36.8 \text{ kN} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F9-20.} \quad w_A &= \rho_w g h_A b = 1000(9.81)(3)(2) \\
 &= 58.86 \text{ kN/m} \\
 w_B &= \rho_w g h_B b = 1000(9.81)(5)(2) \\
 &= 98.1 \text{ kN/m} \\
 F_R &= \tfrac{1}{2}(58.86 + 98.1)(2) = 157 \text{ kN} \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F9-21.} \quad w_A &= \gamma_w h_A b = 62.4(6)(2) = 748.8 \text{ lb/ft} \\
 w_B &= \gamma_w h_B b = 62.4(10)(2) = 1248 \text{ lb/ft} \\
 F_R &= \tfrac{1}{2}(748.8 + 1248)(\sqrt{(3)^2 + (4)^2}) \\
 &= 4.99 \text{ kip} \quad \text{Ans.}
 \end{aligned}$$

## Chapter 10

### F10-1.

$$I_x = \int_A y^2 dA = \int_0^{1 \text{ m}} y^2 (1 - y^{3/2}) dy = 0.111 \text{ m}^4 \quad \text{Ans.}$$

### F10-2.

$$I_x = \int_A y^2 dA = \int_0^{1 \text{ m}} y^2 (y^{3/2} dy) = 0.222 \text{ m}^4 \quad \text{Ans.}$$

### F10-3.

$$I_y = \int_A x^2 dA = \int_0^{1 \text{ m}} x^2 (x^{2/3}) dx = 0.273 \text{ m}^4 \quad \text{Ans.}$$

### F10-4.

$$I_y = \int_A x^2 dA = \int_0^{1 \text{ m}} x^2 (1 - x^{2/3}) dx = 0.0606 \text{ m}^4 \quad \text{Ans.}$$

$$\begin{aligned}
 \mathbf{F10-5.} \quad I_x &= [\tfrac{1}{12}(50)(450^3) + 0] + [\tfrac{1}{12}(300)(50^3) + 0] \\
 &= 383(10^6) \text{ mm}^4 \quad \text{Ans.} \\
 I_y &= [\tfrac{1}{12}(450)(50^3) + 0] \\
 &\quad + 2[\tfrac{1}{12}(50)(150^3) + (150)(50)(100)^2] \\
 &= 183(10^6) \text{ mm}^4 \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F10-6.} \quad I_x &= \tfrac{1}{12}(360)(200^3) - \tfrac{1}{12}(300)(140^3) \\
 &= 171(10^6) \text{ mm}^4 \quad \text{Ans.} \\
 I_y &= \tfrac{1}{12}(200)(360^3) - \tfrac{1}{12}(140)(300^3) \\
 &= 463(10^6) \text{ mm}^4 \quad \text{Ans.}
 \end{aligned}$$

$$\begin{aligned}
 \mathbf{F10-7.} \quad I_y &= 2[\tfrac{1}{12}(50)(200^3) + 0] \\
 &\quad + [\tfrac{1}{12}(300)(50^3) + 0] \\
 &= 69.8(10^6) \text{ mm}^4 \quad \text{Ans.}
 \end{aligned}$$

### F10-8.

$$\begin{aligned}
 \bar{y} &= \frac{\Sigma \tilde{y} A}{\Sigma A} = \frac{15(150)(30) + 105(30)(150)}{150(30) + 30(150)} = 60 \text{ mm} \\
 \bar{I}_{x'} &= \Sigma(\bar{I} + A d^2) \\
 &= [\tfrac{1}{12}(150)(30)^3 + (150)(30)(60 - 15)^2] \\
 &\quad + [\tfrac{1}{12}(30)(150)^3 + 30(150)(105 - 60)^2] \\
 &= 27.0(10^6) \text{ mm}^4 \quad \text{Ans.}
 \end{aligned}$$

## Chapter 11

**F11-1.**  $y_G = 0.75 \sin \theta$        $\delta y_G = 0.75 \cos \theta \delta \theta$   
 $x_C = 2(1.5) \cos \theta$        $\delta x_C = -3 \sin \theta \delta \theta$   
 $\delta U = 0; \quad 2W\delta y_G + P\delta x_C = 0$   
 $(294.3 \cos \theta - 3P \sin \theta)\delta \theta = 0$   
 $P = 98.1 \cot \theta|_{\theta=60^\circ} = 56.6 \text{ N}$

Ans.

**F11-2.**  $x_A = 5 \cos \theta$        $\delta x_A = -5 \sin \theta \delta \theta$   
 $y_G = 2.5 \sin \theta$        $\delta y_G = 2.5 \cos \theta \delta \theta$   
 $\delta U = 0; \quad -P\delta x_A + (-W\delta y_G) = 0$   
 $(5P \sin \theta - 1226.25 \cos \theta)\delta \theta = 0$   
 $P = 245.25 \cot \theta|_{\theta=60^\circ} = 142 \text{ N}$

Ans.

**F11-3.**  $x_B = 0.6 \sin \theta$        $\delta x_B = 0.6 \cos \theta \delta \theta$   
 $y_C = 0.6 \cos \theta$        $\delta y_C = -0.6 \sin \theta \delta \theta$   
 $\delta U = 0; \quad -F_{sp}\delta x_B + (-P\delta y_C) = 0$   
 $-9(10^3) \sin \theta (0.6 \cos \theta \delta \theta)$   
 $-2000(-0.6 \sin \theta \delta \theta) = 0$   
 $\sin \theta = 0 \quad \theta = 0^\circ$   
 $-5400 \cos \theta + 1200 = 0$   
 $\theta = 77.16^\circ = 77.2^\circ$

Ans.

**F11-4.**  $x_B = 0.9 \cos \theta$        $\delta x_B = -0.9 \sin \theta \delta \theta$   
 $x_C = 2(0.9 \cos \theta)$        $\delta x_C = -1.8 \sin \theta \delta \theta$   
 $\delta U = 0; \quad P\delta x_B + (-F_{sp} \delta x_C) = 0$

$$6(10^3)(-0.9 \sin \theta \delta \theta)$$

$$-36(10^3)(\cos \theta - 0.5)(-1.8 \sin \theta \delta \theta) = 0$$

$$\sin \theta (64\,800 \cos \theta - 37\,800)\delta \theta = 0$$

$$\sin \theta = 0 \quad \theta = 0^\circ$$

$$64\,800 \cos \theta - 37\,800 = 0$$

$$\theta = 54.31^\circ = 54.3^\circ$$

Ans.

Ans.

**F11-5.**  $y_G = 2.5 \sin \theta$        $\delta y_G = 2.5 \cos \theta \delta \theta$   
 $x_A = 5 \cos \theta$        $\delta x_C = -5 \sin \theta \delta \theta$   
 $\delta U = 0; \quad (-F_{sp}\delta x_A) - W\delta y_G = 0$   
 $(15\,000 \sin \theta \cos \theta - 7500 \sin \theta$   
 $- 1226.25 \cos \theta)\delta \theta = 0$   
 $\theta = 56.33^\circ = 56.3^\circ$   
or  $\theta = 9.545^\circ = 9.55^\circ$

Ans.

Ans.

**F11-6.**  $F_{sp} = 15\,000(0.6 - 0.6 \cos \theta)$   
 $x_C = 3[0.3 \sin \theta]$        $\delta x_C = 0.9 \cos \theta \delta \theta$   
 $y_B = 2[0.3 \cos \theta]$        $\delta y_B = -0.6 \sin \theta \delta \theta$   
 $\delta U = 0; \quad P\delta x_C + F_{sp}\delta y_B = 0$   
 $(135 \cos \theta - 5400 \sin \theta + 5400 \sin \theta \cos \theta)\delta \theta = 0$   
 $\theta = 20.9^\circ$

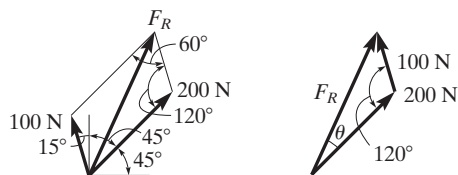
Ans.



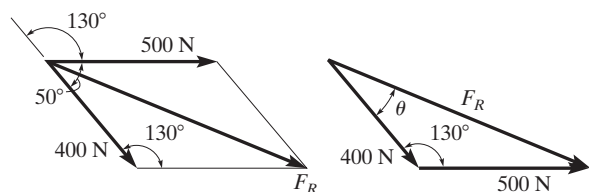
# Preliminary Problems Statics Solutions

## Chapter 2

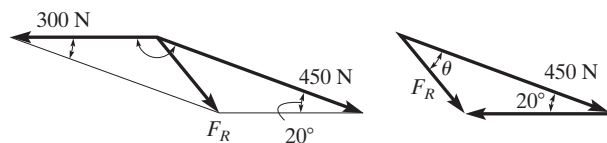
2-1.



(a)

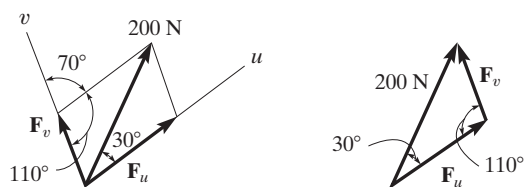


(b)

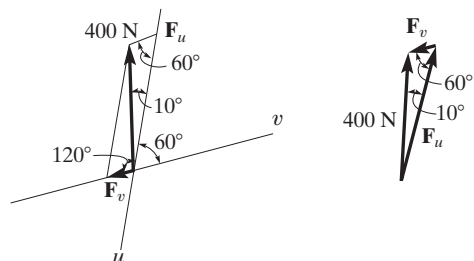


(c)

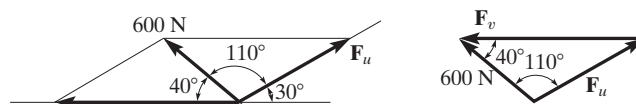
2-2.



(a)

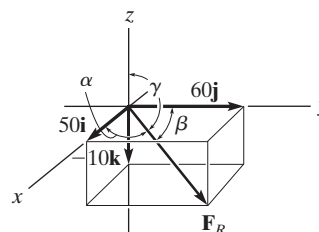


(b)

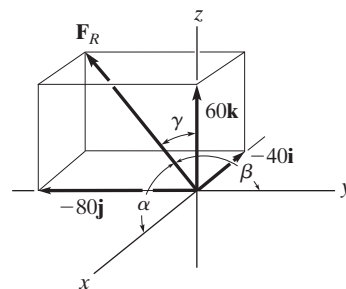


(c)

2-3.



(a)

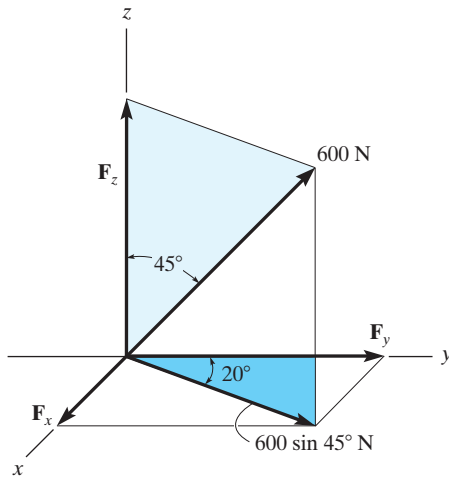


(b)

2-4. a)  $\mathbf{F} = \{-4\mathbf{i} - 4\mathbf{j} + 2\mathbf{k}\} \text{ kN}$   
 $F = \sqrt{(4)^2 + (-4)^2 + (2)^2} = 6 \text{ kN}$   
 $\cos \beta = \frac{-2}{3}$

b)  $\mathbf{F} = \{20\mathbf{i} + 20\mathbf{j} - 10\mathbf{k}\} \text{ N}$   
 $F = \sqrt{(20)^2 + (20)^2 + (-10)^2} = 30 \text{ N}$   
 $\cos \beta = \frac{2}{3}$

2-5.

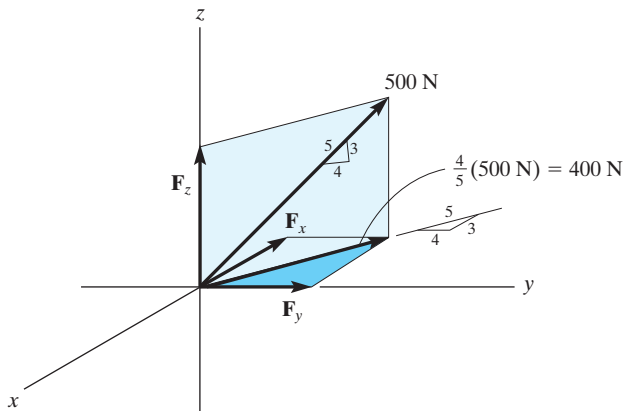


(a)

$$F_x = (600 \sin 45^\circ) \sin 20^\circ \text{ N}$$

$$F_y = (600 \sin 45^\circ) \cos 20^\circ \text{ N}$$

$$F_z = 600 \cos 45^\circ \text{ N}$$

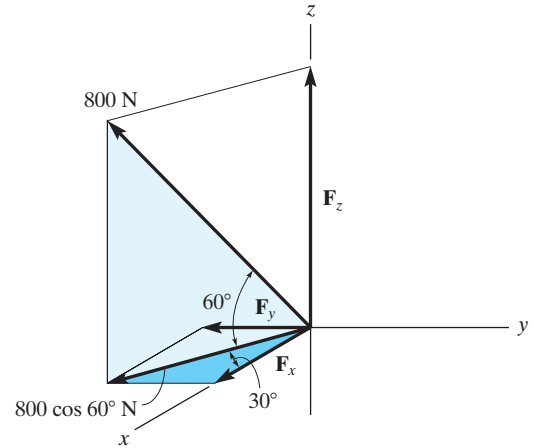


(b)

$$F_x = -\frac{3}{5}(400) \text{ N}$$

$$F_y = \frac{4}{5}(400) \text{ N}$$

$$F_z = \frac{3}{5}(500) \text{ N}$$



(c)

$$F_x = 800 \cos 60^\circ \cos 30^\circ \text{ N}$$

$$F_y = -800 \cos 60^\circ \sin 30^\circ \text{ N}$$

$$F_z = 800 \sin 60^\circ \text{ N}$$

2-6. a)  $\mathbf{r}_{AB} = \{-5\mathbf{i} + 3\mathbf{j} - 2\mathbf{k}\} \text{ m}$

b)  $\mathbf{r}_{AB} = \{4\mathbf{i} + 8\mathbf{j} - 3\mathbf{k}\} \text{ m}$

c)  $\mathbf{r}_{AB} = \{6\mathbf{i} - 3\mathbf{j} - 4\mathbf{k}\} \text{ m}$

2-7. a)  $\mathbf{F} = 15 \text{ kN} \left( \frac{-3}{5}\mathbf{i} + \frac{4}{5}\mathbf{j} \right) = \{-9\mathbf{i} + 12\mathbf{j}\} \text{ kN}$

b)  $\mathbf{F} = 600 \text{ N} \left( \frac{2}{3}\mathbf{i} + \frac{2}{3}\mathbf{j} - \frac{1}{3}\mathbf{k} \right)$   
 $= \{400\mathbf{i} + 400\mathbf{j} - 200\mathbf{k}\} \text{ N}$

c)  $\mathbf{F} = 300 \text{ N} \left( -\frac{2}{3}\mathbf{i} + \frac{2}{3}\mathbf{j} - \frac{1}{3}\mathbf{k} \right)$   
 $= \{-200\mathbf{i} + 200\mathbf{j} - 100\mathbf{k}\} \text{ N}$

2-8. a)  $\mathbf{r}_A = \{3\mathbf{k}\} \text{ m}, \quad r_A = 3 \text{ m}$

$$\mathbf{r}_B = \{2\mathbf{i} + 2\mathbf{j} - 1\mathbf{k}\} \text{ m}, \quad r_B = 3 \text{ m}$$

$$\mathbf{r}_A \cdot \mathbf{r}_B = 0(2) + 0(2) + (3)(-1) = -3 \text{ m}^2$$

$$\mathbf{r}_A \cdot \mathbf{r}_B = r_A r_B \cos \theta$$

$$-3 = 3(3) \cos \theta$$

b)  $\mathbf{r}_A = \{-2\mathbf{i} + 2\mathbf{j} + 1\mathbf{k}\} \text{ m}, \quad r_A = 3 \text{ m}$

$$\mathbf{r}_B = \{1.5\mathbf{i} - 2\mathbf{k}\} \text{ m}, \quad r_B = 2.5 \text{ m}$$

$$\mathbf{r}_A \cdot \mathbf{r}_B = (-2)(1.5) + 2(0) + (1)(-2) = -5 \text{ m}^2$$

$$\mathbf{r}_A \cdot \mathbf{r}_B = r_A r_B \cos \theta$$

$$-5 = 3(2.5) \cos \theta$$

2-9. a)

$$\mathbf{F} = 300 \text{ N} \left( \frac{2}{3} \mathbf{i} + \frac{2}{3} \mathbf{j} - \frac{1}{3} \mathbf{k} \right) = \{200\mathbf{i} + 200\mathbf{j} - 100\mathbf{k}\} \text{ N}$$

$$\mathbf{u}_a = -\frac{3}{5} \mathbf{i} + \frac{4}{5} \mathbf{j}$$

$$F_a = \mathbf{F} \cdot \mathbf{u}_a = (200) \left( -\frac{3}{5} \right) + (200) \left( \frac{4}{5} \right) + (-100) (0)$$

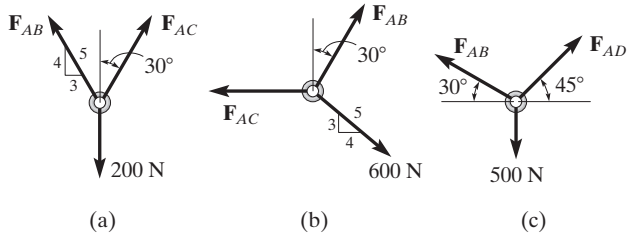
$$\text{b) } \mathbf{F} = 500 \text{ N} \left( -\frac{4}{5} \mathbf{j} + \frac{3}{5} \mathbf{k} \right) = \{-400\mathbf{j} + 300\mathbf{k}\} \text{ N}$$

$$\mathbf{u}_a = -\frac{1}{3} \mathbf{i} + \frac{2}{3} \mathbf{j} + \frac{2}{3} \mathbf{k}$$

$$F_a = \mathbf{F} \cdot \mathbf{u}_a = (0) \left( -\frac{1}{3} \right) + (-400) \left( \frac{2}{3} \right) + (300) \left( \frac{2}{3} \right)$$

## Chapter 3

3-1.



$$\text{3-2. a) } \Sigma F_x = 0; \quad F \cos 60^\circ - P \left( \frac{1}{\sqrt{2}} \right) - 600 \left( \frac{4}{5} \right) = 0$$

$$\Sigma F_y = 0; \quad -F \sin 60^\circ - P \left( \frac{1}{\sqrt{2}} \right) + 600 \left( \frac{3}{5} \right) = 0$$

$$\text{b) } \Sigma F_x = 0; \quad P \left( \frac{4}{5} \right) - F \sin 60^\circ - 200 \sin 15^\circ = 0$$

$$\Sigma F_y = 0; \quad -P \left( \frac{3}{5} \right) - F \cos 60^\circ + 200 \cos 15^\circ = 0$$

$$\text{c) } \Sigma F_x = 0;$$

$$300 \cos 40^\circ + 450 \cos 30^\circ - P \cos 30^\circ + F \sin 10^\circ = 0$$

$$\Sigma F_y = 0;$$

$$-300 \sin 40^\circ + 450 \sin 30^\circ - P \sin 30^\circ - F \cos 10^\circ = 0$$

## Chapter 4

$$\text{4-1. a) } M_O = 100 \text{ N}(2 \text{ m}) = 200 \text{ N} \cdot \text{m} \curvearrowright$$

$$\text{b) } M_O = -100 \text{ N}(1 \text{ m}) = 100 \text{ N} \cdot \text{m} \curvearrowright$$

$$\text{c) } M_O = -\left( \frac{3}{5} \right) (500 \text{ N})(2 \text{ m}) = 600 \text{ N} \cdot \text{m} \curvearrowright$$

$$\text{d) } M_O = \left( \frac{4}{5} \right) (500 \text{ N})(3 \text{ m}) = 1200 \text{ N} \cdot \text{m} \curvearrowright$$

$$\text{e) } M_O = -\left( \frac{3}{5} \right) (100 \text{ N})(5 \text{ m}) = 300 \text{ N} \cdot \text{m} \curvearrowright$$

$$\text{f) } M_O = 100 \text{ N}(0) = 0$$

$$\text{g) } M_O = -\left( \frac{3}{5} \right) (500 \text{ N})(2 \text{ m}) + \left( \frac{4}{5} \right) (500 \text{ N})(1 \text{ m}) = 200 \text{ N} \cdot \text{m} \curvearrowright$$

$$\text{h) } M_O = -\left( \frac{3}{5} \right) (500 \text{ N})(3 \text{ m} - 1 \text{ m}) + \left( \frac{4}{5} \right) (500 \text{ N})(1 \text{ m}) = 200 \text{ N} \cdot \text{m} \curvearrowright$$

$$\text{i) } M_O = \left( \frac{3}{5} \right) (500 \text{ N})(1 \text{ m}) - \left( \frac{4}{5} \right) (500 \text{ N})(3 \text{ m}) = 900 \text{ N} \cdot \text{m} \curvearrowright$$

4-2.

$$\mathbf{M}_P = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 2 & -3 & 0 \\ -3 & 2 & 5 \end{vmatrix} \quad \mathbf{M}_P = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 2 & 5 & -1 \\ 2 & -4 & -3 \end{vmatrix}$$

$$\mathbf{M}_P = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 5 & -4 & -1 \\ -2 & 3 & 4 \end{vmatrix}$$

$$\text{4-3. a) } M_x = -(100 \text{ N})(3 \text{ m}) = -300 \text{ N} \cdot \text{m}$$

$$M_y = -(200 \text{ N})(2 \text{ m}) = -400 \text{ N} \cdot \text{m}$$

$$M_z = -(300 \text{ N})(2 \text{ m}) = -600 \text{ N} \cdot \text{m}$$

$$\text{b) } M_x = (50 \text{ N})(0.5 \text{ m}) = 25 \text{ N} \cdot \text{m}$$

$$M_y = (400 \text{ N})(0.5 \text{ m}) - (300 \text{ N})(3 \text{ m}) = -700 \text{ N} \cdot \text{m}$$

$$M_z = (100 \text{ N})(3 \text{ m}) = 300 \text{ N} \cdot \text{m}$$

$$\text{c) } M_x = (300 \text{ N})(2 \text{ m}) - (100 \text{ N})(2 \text{ m}) = 400 \text{ N} \cdot \text{m}$$

$$M_y = -(300 \text{ N})(1 \text{ m}) + (50 \text{ N})(1 \text{ m})$$

$$+ (400 \text{ N})(0.5 \text{ m}) = 250 \text{ N} \cdot \text{m}$$

$$M_z = -(200 \text{ N})(1 \text{ m}) = -200 \text{ N} \cdot \text{m}$$

4-4. a)

$$M_a = \begin{vmatrix} -\frac{4}{5} & -\frac{3}{5} & 0 \\ -5 & 2 & 0 \\ 6 & 2 & 3 \end{vmatrix} = \begin{vmatrix} -\frac{4}{5} & -\frac{3}{5} & 0 \\ -1 & 5 & 0 \\ 6 & 2 & 3 \end{vmatrix}$$

b)

$$M_a = \begin{vmatrix} -\frac{1}{\sqrt{2}} & \frac{1}{\sqrt{2}} & 0 \\ 3 & 4 & -2 \\ 2 & -4 & 3 \end{vmatrix} = \begin{vmatrix} -\frac{1}{\sqrt{2}} & \frac{1}{\sqrt{2}} & 0 \\ 5 & 2 & -2 \\ 2 & -4 & 3 \end{vmatrix}$$

$$\text{c) } M_a = \begin{vmatrix} \frac{2}{3} & -\frac{1}{3} & \frac{2}{3} \\ -5 & -4 & 0 \\ 2 & -4 & 3 \end{vmatrix} = \begin{vmatrix} \frac{2}{3} & -\frac{1}{3} & \frac{2}{3} \\ -3 & -5 & 2 \\ 2 & -4 & 3 \end{vmatrix}$$

$$\mathbf{4-5.} \quad \text{a) } \overset{+}{\rightarrow}(F_R)_x = \Sigma F_x;$$

$$(F_R)_x = -\left(\frac{4}{5}\right)(500 \text{ N}) + 200 \text{ N} = -200 \text{ N}$$

$$+\uparrow(F_R)_y = \Sigma F_y;$$

$$(F_R)_y = -\frac{3}{5}(500 \text{ N}) - 400 \text{ N} = -700 \text{ N}$$

$$\zeta + (M_R)_O = \Sigma M_O;$$

$$(M_R)_O = -\left(\frac{3}{5}\right)(500 \text{ N})(2 \text{ m}) - 400 \text{ N}(4 \text{ m}) \\ = -2200 \text{ N} \cdot \text{m}$$

$$\text{b) } \overset{+}{\rightarrow}(F_R)_x = \Sigma F_x;$$

$$(F_R)_x = \left(\frac{4}{5}\right)(500 \text{ N}) = 400 \text{ N}$$

$$+\uparrow(F_R)_y = \Sigma F_y;$$

$$(F_R)_y = -(300 \text{ N}) - \left(\frac{3}{5}\right)(500 \text{ N}) = -600 \text{ N}$$

$$\zeta + (M_R)_O = \Sigma M_O;$$

$$(M_R)_O = -(300 \text{ N})(2 \text{ m}) - \left(\frac{3}{5}\right)(500 \text{ N})(4 \text{ m}) \\ - 200 \text{ N} \cdot \text{m} = -2000 \text{ N} \cdot \text{m}$$

$$\text{c) } \overset{+}{\rightarrow}(F_R)_x = \Sigma F_x;$$

$$(F_R)_x = \left(\frac{3}{5}\right)(500 \text{ N}) + 100 \text{ N} = 400$$

$$+\uparrow(F_R)_y = \Sigma F_y;$$

$$(F_R)_y = -(500 \text{ N}) - \left(\frac{4}{5}\right)(500 \text{ N}) = -900 \text{ N}$$

$$\zeta + (M_R)_O = \Sigma M_O;$$

$$(M_R)_O = -(500 \text{ N})(2 \text{ m}) - \left(\frac{4}{5}\right)(500 \text{ N})(4 \text{ m}) \\ + \left(\frac{3}{5}\right)(500 \text{ N})(2 \text{ m}) = -2000 \text{ N} \cdot \text{m}$$

$$\text{d) } \overset{+}{\rightarrow}(F_R)_x = \Sigma F_x;$$

$$(F_R)_x = -\left(\frac{4}{5}\right)(500 \text{ N}) + \left(\frac{3}{5}\right)(500 \text{ N}) = -100 \text{ N}$$

$$+\uparrow(F_R)_y = \Sigma F_y;$$

$$(F_R)_y = -\left(\frac{3}{5}\right)(500 \text{ N}) - \left(\frac{4}{5}\right)(500 \text{ N}) = -700 \text{ N}$$

$$\zeta + (M_R)_O = \Sigma M_O;$$

$$(M_R)_O = \left(\frac{4}{5}\right)(500 \text{ N})(4 \text{ m}) + \left(\frac{3}{5}\right)(500 \text{ N})(2 \text{ m}) \\ - \left(\frac{3}{5}\right)(500 \text{ N})(4 \text{ m}) + 200 \text{ N} \cdot \text{m} = 1200 \text{ N} \cdot \text{m}$$

$$\mathbf{4-6.} \quad \text{a) } \overset{+}{\rightarrow}(F_R)_x = \Sigma F_x; \quad (F_R)_x = 0$$

$$+\uparrow(F_R)_y = \Sigma F_y;$$

$$(F_R)_y = -200 \text{ N} - 260 \text{ N} = -460 \text{ N}$$

$$\zeta + (F_R)_y d = \Sigma M_O;$$

$$-(460 \text{ N})d = -(200 \text{ N})(2 \text{ m}) - (260 \text{ N})(4 \text{ m}) \\ d = 3.13 \text{ m}$$

Note: Although 460 N acts downward, this is *not* why  $-(460 \text{ N})d$  is negative. It is because the *moment* of 460 N about *O* is negative.

$$\text{b) } \overset{+}{\rightarrow}(F_R)_x = \Sigma F_x;$$

$$(F_R)_x = -\left(\frac{3}{5}\right)(500 \text{ N}) = -300 \text{ N}$$

$$+\uparrow(F_R)_y = \Sigma F_y;$$

$$(F_R)_y = -400 \text{ N} - \left(\frac{4}{5}\right)(500 \text{ N}) = -800 \text{ N}$$

$$\zeta + (F_R)_y d = \Sigma M_O;$$

$$-(800 \text{ N})d = -(400 \text{ N})(2 \text{ m}) - \left(\frac{4}{5}\right)(500 \text{ N})(4 \text{ m}) \\ d = 3 \text{ m}$$

$$\text{c) } \overset{+}{\rightarrow}(F_R)_x = \Sigma F_x;$$

$$(F_R)_x = \left(\frac{4}{5}\right)(500 \text{ N}) - \left(\frac{4}{5}\right)(500 \text{ N}) = 0$$

$$+\uparrow(F_R)_y = \Sigma F_y;$$

$$(F_R)_y = -\left(\frac{3}{5}\right)(500 \text{ N}) - \left(\frac{3}{5}\right)(500 \text{ N}) = -600 \text{ N}$$

$$\zeta + (F_R)_y d = \Sigma M_O;$$

$$-(600 \text{ N})d = -\left(\frac{3}{5}\right)(500 \text{ N})(2 \text{ m}) - \left(\frac{3}{5}\right)(500 \text{ N})(4 \text{ m}) \\ - 600 \text{ N} \cdot \text{m} \\ d = 4 \text{ m}$$

$$\mathbf{4-7.} \quad \text{a) } +\downarrow F_R = \Sigma F_z;$$

$$F_R = 200 \text{ N} + 100 \text{ N} + 200 \text{ N} = 500 \text{ N}$$

$$(M_R)_x = \Sigma M_x;$$

$$-(500 \text{ N})y = -(100 \text{ N})(2 \text{ m}) - (200 \text{ N})(2 \text{ m}) \\ y = 1.20 \text{ m}$$

$$(M_R)_y = \Sigma M_y;$$

$$(500 \text{ N})x = (100 \text{ N})(2 \text{ m}) + (200 \text{ N})(1 \text{ m}) \\ x = 0.80 \text{ m}$$

$$b) +\downarrow F_R = \Sigma F_z;$$

$$F_R = 100 \text{ N} - 100 \text{ N} + 200 \text{ N} = 200 \text{ N}$$

$$(M_R)_x = \Sigma M_x;$$

$$-(200 \text{ N})y = (100 \text{ N})(1 \text{ m}) + (100 \text{ N})(2 \text{ m})$$

$$- (200 \text{ N})(2 \text{ m})$$

$$y = 0.5 \text{ m}$$

$$(M_R)_y = \Sigma M_y;$$

$$(200 \text{ N})x = -(100 \text{ N})(2 \text{ m}) + (100 \text{ N})(2 \text{ m})$$

$$x = 0$$

$$c) +\downarrow F_R = \Sigma F_z;$$

$$F_R = 400 \text{ N} + 300 \text{ N} + 200 \text{ N} + 100 \text{ N} = 1000 \text{ N}$$

$$(M_R)_x = \Sigma M_x;$$

$$-(1000 \text{ N})y = -(300 \text{ N})(4 \text{ m}) - (100 \text{ N})(4 \text{ m})$$

$$y = 1.6 \text{ m}$$

$$(M_R)_y = \Sigma M_y;$$

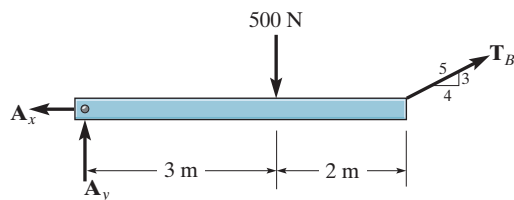
$$(1000 \text{ N})x = (400 \text{ N})(2 \text{ m}) + (300 \text{ N})(2 \text{ m})$$

$$- (200 \text{ N})(2 \text{ m}) - (100 \text{ N})(2 \text{ m})$$

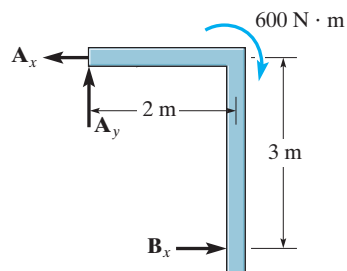
$$x = 0.8 \text{ m}$$

## Chapter 5

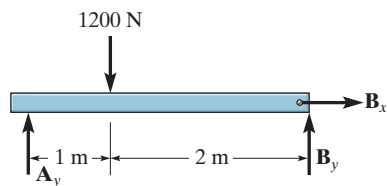
5-1.



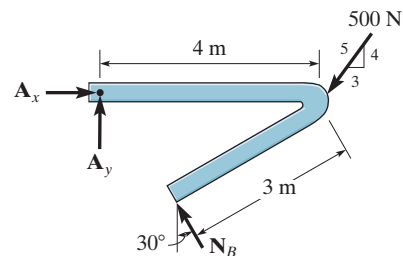
(a)



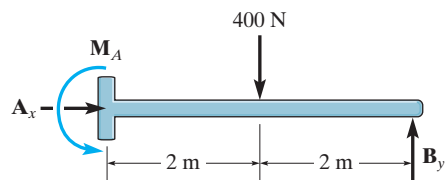
(b)



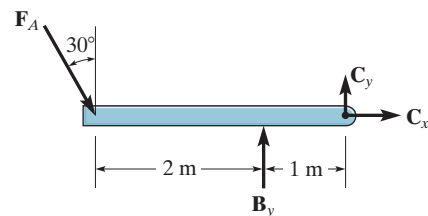
(c)



(d)

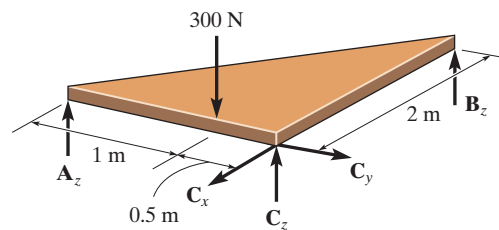


(e)

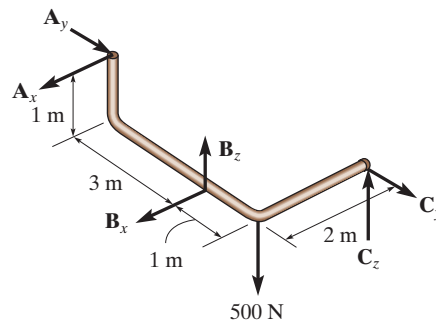


(f)

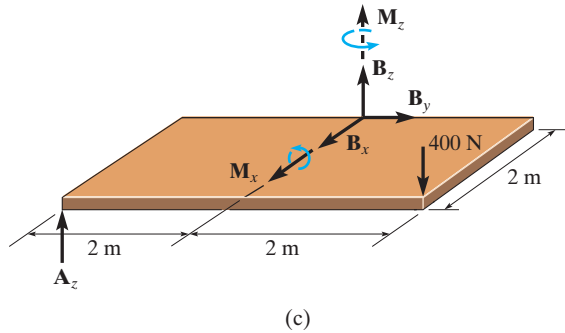
5-2.



(a)

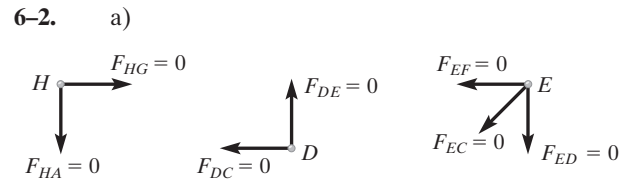
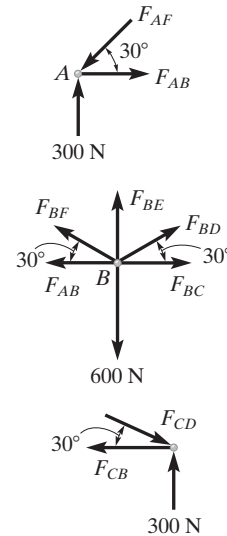


(b)



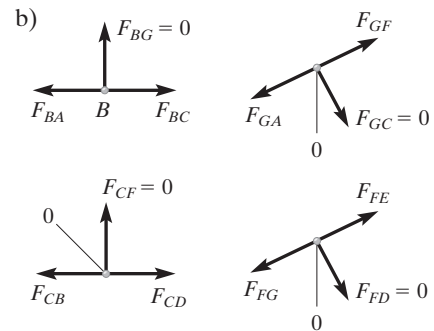
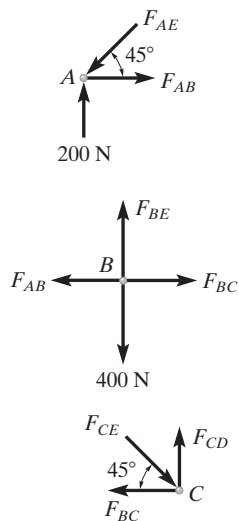
- 5-3. a)  $\Sigma M_x = 0$ ;  
 $-(400 \text{ N})(2 \text{ m}) - (600 \text{ N})(5 \text{ m}) + B_z(5 \text{ m}) = 0$   
 $\Sigma M_y = 0$ ;  $-A_z(4 \text{ m}) - B_z(4 \text{ m}) = 0$   
 $\Sigma M_z = 0$ ;  $B_y(4 \text{ m}) - B_x(5 \text{ m}) + (300 \text{ N})(5 \text{ m}) = 0$
- b)  $\Sigma M_x = 0$ ;  $A_z(4 \text{ m}) + C_z(6 \text{ m}) = 0$   
 $\Sigma M_y = 0$ ;  $B_z(1 \text{ m}) - C_z(1 \text{ m}) = 0$   
 $\Sigma M_z = 0$ ;  $-B_y(1 \text{ m}) + (300 \text{ N})(2 \text{ m}) - A_x(4 \text{ m}) + C_y(1 \text{ m}) = 0$
- c)  $\Sigma M_x = 0$ ;  $B_z(2 \text{ m}) + C_z(3 \text{ m}) - 800 \text{ N} \cdot \text{m} = 0$   
 $\Sigma M_y = 0$ ;  $-C_z(1.5 \text{ m}) = 0$   
 $\Sigma M_z = 0$ ;  $-B_x(2 \text{ m}) + C_y(1.5 \text{ m}) = 0$

b)  $A_y = 300 \text{ N}$ ,  $C_x = 0$ ,  $C_y = 300 \text{ N}$

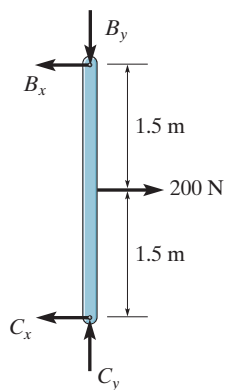
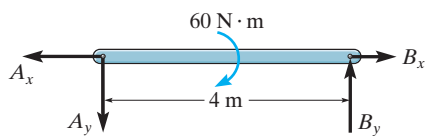
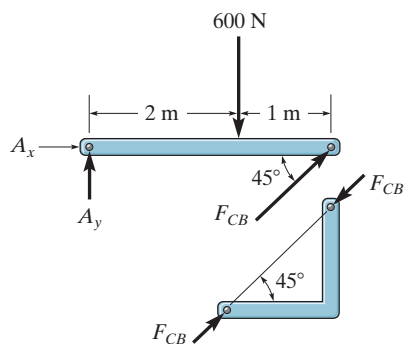
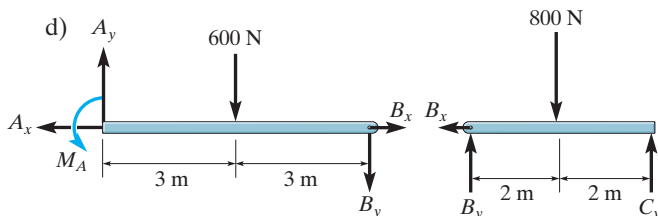
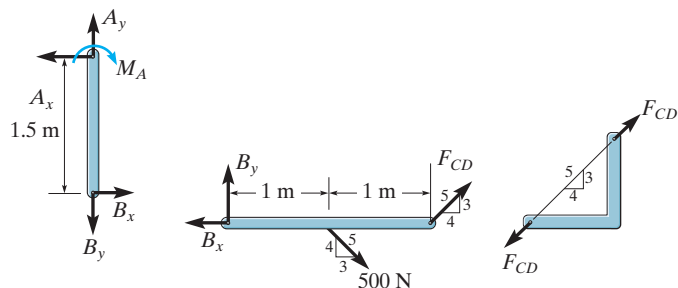
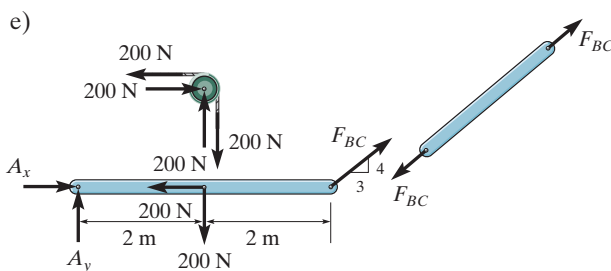
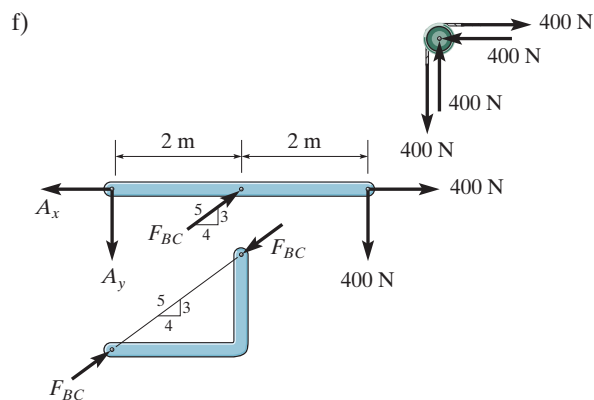


## Chapter 6

6-1. a)  $A_y = 200 \text{ N}$ ,  $D_x = 0$ ,  $D_y = 200 \text{ N}$

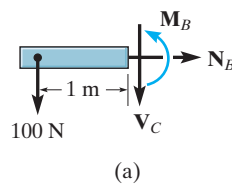


6-3. a)

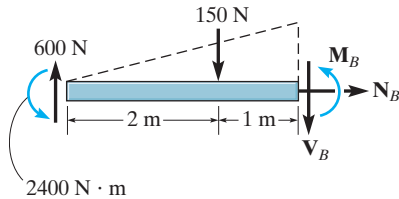
b)  $CB$  is a two-force member.c)  $CD$  is a two-force member. $BC$  is a two-force member. $BC$  is a two-force member.

## Chapter 7

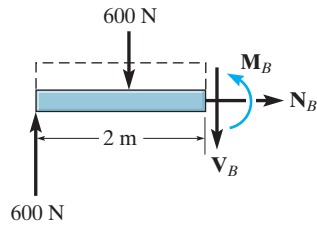
7-1.



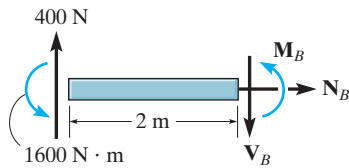
(a)



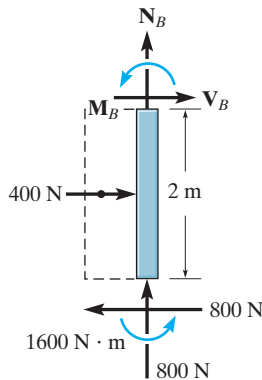
(b)



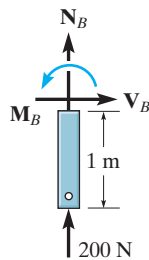
(c)



(d)



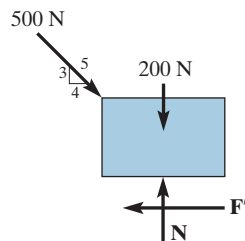
(e)



(f)

## Chapter 8

8-1. a)



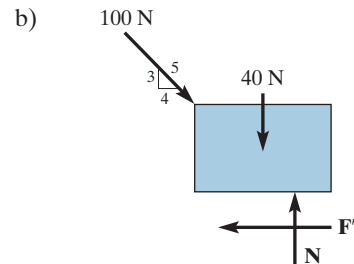
$$\begin{aligned} \rightarrow \Sigma F_x &= 0; \\ \left(\frac{4}{5}\right)(500 \text{ N}) - F' &= 0, F' = 400 \text{ N} \end{aligned}$$

$$+\uparrow \Sigma F_y = 0;$$

$$N - 200 \text{ N} - \left(\frac{3}{5}\right)(500 \text{ N}) = 0, N = 500 \text{ N}$$

$$F_{\max} = 0.3(500 \text{ N}) = 150 \text{ N} < 400 \text{ N}$$

$$\text{Slipping } F = \mu_k N = 0.2(500 \text{ N}) = 100 \text{ N} \quad \text{Ans.}$$



$$\rightarrow \Sigma F_x = 0;$$

$$\frac{4}{5}(100 \text{ N}) - F' = 0; F' = 80 \text{ N}$$

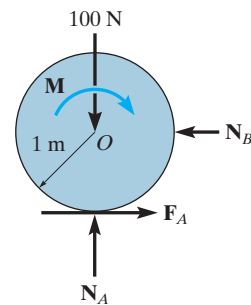
$$+\uparrow \Sigma F_y = 0;$$

$$N - 40 \text{ N} - \left(\frac{3}{5}\right)(100 \text{ N}) = 0; N = 100 \text{ N}$$

$$F_{\max} = 0.9(100 \text{ N}) = 90 \text{ N} > 80 \text{ N}$$

$$F = F' = 80 \text{ N} \quad \text{Ans.}$$

8-2.



Require

$$+\uparrow \Sigma F_y = 0;$$

$$F_A = 0.1 N_A$$

$$N_A - 100 \text{ N} = 0$$

$$N_A = 100 \text{ N}$$

$$F_A = 0.1(100 \text{ N}) = 10 \text{ N}$$

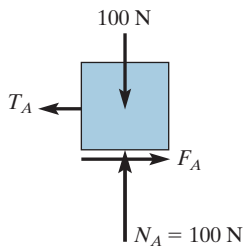
$$\zeta + \Sigma M_O = 0;$$

$$-M + (10 \text{ N})(1 \text{ m}) = 0$$

$$M = 10 \text{ N} \cdot \text{m}$$

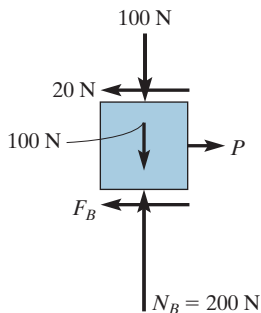


- 8-3. a) Slipping must occur between  $A$  and  $B$ .



$$F_A = 0.2(100 \text{ N}) = 20 \text{ N}$$

- b) Assume  $B$  slips on  $C$  and  $C$  does not slip.

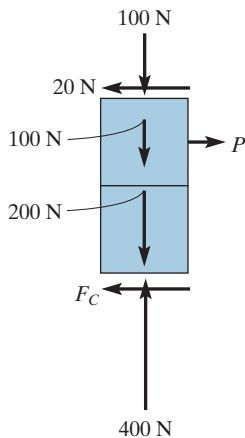


$$F_B = 0.2(200 \text{ N}) = 40 \text{ N}$$

$$\rightarrow \Sigma F_x = 0; \quad P - 20 \text{ N} - 40 \text{ N} = 0$$

$$P = 60 \text{ N}$$

- c) Assume  $C$  slips and  $B$  does not slip on  $C$ .



$$F_C = 0.1(400 \text{ N}) = 40 \text{ N}$$

$$\rightarrow \Sigma F_x = 0; \quad P - 20 \text{ N} - 40 \text{ N} = 0$$

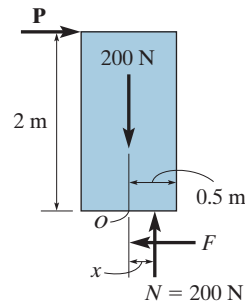
$$P = 60 \text{ N}$$

$$\text{Therefore,} \quad P = 60 \text{ N}$$

*Ans.*

#### 8-4.

- a)



Assume slipping,

$$F = 0.3(200 \text{ N}) = 60 \text{ N}$$

$$\rightarrow \Sigma F_x = 0; \quad P - 60 \text{ N} = 0; \quad P = 60 \text{ N}$$

$$\zeta + \Sigma M_O = 0; \quad 200 \text{ N}(x) - (60 \text{ N})(2 \text{ m}) = 0$$

$$x = 0.6 \text{ m} > 0.5 \text{ m}$$

Block tips,

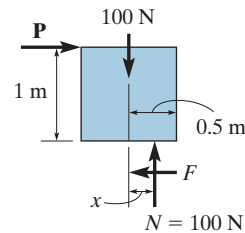
$$x = 0.5 \text{ m}$$

$$\zeta + \Sigma M_O = 0 \quad (200 \text{ N})(0.5 \text{ m}) - P(2 \text{ m}) = 0$$

$$P = 50 \text{ N}$$

*Ans.*

- b)



Assume slipping,

$$F = 0.4(100 \text{ N}) = 40 \text{ N}$$

$$\rightarrow \Sigma F_x = 0; \quad P - 40 \text{ N} = 0; \quad P = 40 \text{ N}$$

$$\zeta + \Sigma M_O = 0; \quad (100 \text{ N})(x) - (40 \text{ N})(1 \text{ m}) = 0$$

$$x = 0.4 \text{ m} < 0.5 \text{ m}$$

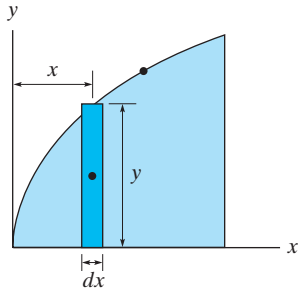
No tipping

$$P = 40 \text{ N}$$

*Ans.*

## Chapter 9

9-1. a)

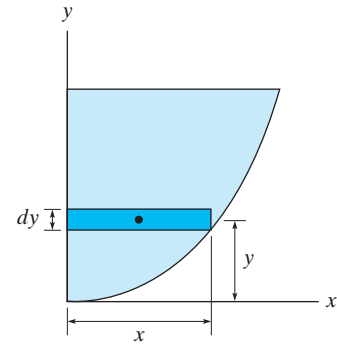


$$\tilde{x} = x$$

$$\tilde{y} = \frac{y}{2} = \frac{\sqrt{x}}{2}$$

$$dA = ydx = \sqrt{x} dx$$

c)

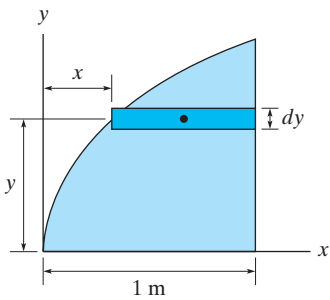


$$\tilde{x} = \frac{x}{2} = \frac{\sqrt{y}}{2}$$

$$\tilde{y} = y$$

$$dA = xdy = \sqrt{y} dy$$

b)

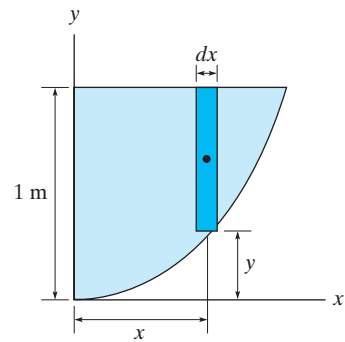


$$\tilde{x} = x + \left( \frac{1-x}{2} \right) = \frac{1+x}{2} = \frac{1+y^2}{2}$$

$$\tilde{y} = y$$

$$dA = (1-x)dy = (1-y^2)dy$$

d)



$$\tilde{x} = x$$

$$\tilde{y} = y + \left( \frac{1-y}{2} \right) = \frac{1+y}{2} = \frac{1+x^2}{2}$$

$$dA = (1-y)dx = (1-x^2)dx$$

# Review Problem Solutions

## Chapter 2

$$\mathbf{R2-1.} \quad F_R = \sqrt{(300)^2 + (500)^2 - 2(300)(500) \cos 95^\circ}$$

$$= 605.1 = 605 \text{ N} \quad \text{Ans.}$$

$$\frac{605.1}{\sin 95^\circ} = \frac{500}{\sin \theta}$$

$$\theta = 55.40^\circ$$

$$\phi = 55.40^\circ + 30^\circ = 85.4^\circ \quad \text{Ans.}$$

$$\mathbf{R2-2.} \quad \frac{F_{1v}}{\sin 30^\circ} = \frac{250}{\sin 105^\circ} \quad F_{1v} = 129 \text{ N} \quad \text{Ans.}$$

$$\frac{F_{1u}}{\sin 45^\circ} = \frac{250}{\sin 105^\circ} \quad F_{1u} = 183 \text{ N} \quad \text{Ans.}$$

$$\mathbf{R2-3.} \quad F_{Rx} = F_{1x} + F_{2x} + F_{3x} + F_{4x}$$

$$F_{Rx} = -200 + 320 + 180 - 300 = 0$$

$$F_{Ry} = F_{1y} + F_{2y} + F_{3y} + F_{4y}$$

$$F_{Ry} = 0 - 240 + 240 + 0 = 0$$

Thus,  $F_R = 0$  Ans.

$$\mathbf{R2-4.} \quad \cos^2 30^\circ + \cos^2 70^\circ + \cos^2 \gamma = 1$$

$$\cos \gamma = \pm 0.3647$$

$$\gamma = 68.61^\circ \text{ or } 111.39^\circ$$

By inspection,  $\gamma = 111.39^\circ$ .

$$\mathbf{F} = 250 \{ \cos 30^\circ \mathbf{i} + \cos 70^\circ \mathbf{j} + \cos 111.39^\circ \}$$

$$= \{ 217 \mathbf{i} + 85.5 \mathbf{j} - 91.2 \mathbf{k} \} \text{ lb} \quad \text{Ans.}$$

$$\mathbf{R2-5.} \quad \mathbf{r} = \{ 50 \sin 20^\circ \mathbf{i} + 50 \cos 20^\circ \mathbf{j} - 35 \mathbf{k} \} \text{ ft}$$

$$r = \sqrt{(17.10)^2 + (46.98)^2 + (-35)^2} = 61.03 \text{ ft}$$

$$\mathbf{u} = \frac{\mathbf{r}}{r} = \{ 0.280 \mathbf{i} + 0.770 \mathbf{j} - 0.573 \mathbf{k} \}$$

$$\mathbf{F} = F\mathbf{u} = \{ 98.1 \mathbf{i} + 269 \mathbf{j} - 201 \mathbf{k} \} \text{ lb} \quad \text{Ans.}$$

$$\mathbf{R2-6.} \quad \mathbf{F}_1 = 600 \left( \frac{4}{5} \right) \cos 30^\circ (+\mathbf{i}) + 600 \left( \frac{4}{5} \right) \sin 30^\circ (-\mathbf{j})$$

$$+ 600 \left( \frac{3}{5} \right) (+\mathbf{k})$$

$$= \{ 415.69 \mathbf{i} - 240 \mathbf{j} + 360 \mathbf{k} \} \text{ N} \quad \text{Ans.}$$

$$\mathbf{F}_2 = 0 \mathbf{i} + 450 \cos 45^\circ (+\mathbf{j}) + 450 \sin 45^\circ (+\mathbf{k})$$

$$= \{ 318.20 \mathbf{j} + 318.20 \mathbf{k} \} \text{ N} \quad \text{Ans.}$$

$$\mathbf{R2-7.} \quad \mathbf{r}_1 = \{ 400 \mathbf{i} + 250 \mathbf{k} \} \text{ mm}; \quad r_1 = 471.70 \text{ mm}$$

$$\mathbf{r}_2 = \{ 50 \mathbf{i} + 300 \mathbf{j} \} \text{ mm}; \quad r_2 = 304.14 \text{ mm}$$

$$\mathbf{r}_1 \cdot \mathbf{r}_2 = (400)(50) + 0(300) + 250(0) = 20\,000$$

$$\theta = \cos^{-1} \left( \frac{\mathbf{r}_1 \cdot \mathbf{r}_2}{r_1 r_2} \right) = \cos^{-1} \left( \frac{20\,000}{(471.70)(304.14)} \right)$$

$$= 82.0^\circ \quad \text{Ans.}$$

$$\mathbf{R2-8.} \quad F_{\text{Proj}} = \mathbf{F} \cdot \mathbf{u}_v = (2\mathbf{i} + 4\mathbf{j} + 10\mathbf{k}) \cdot \left( \frac{2}{3}\mathbf{i} + \frac{2}{3}\mathbf{j} - \frac{1}{3}\mathbf{k} \right)$$

$$F_{\text{Proj}} = 0.667 \text{ kN}$$

## Chapter 3

$$\mathbf{R3-1.} \quad \rightarrow \Sigma F_x = 0; \quad F_B - F_A \cos 60^\circ - 50 \left( \frac{4}{5} \right) = 0$$

$$+ \uparrow \Sigma F_y = 0; \quad -F_A \sin 60^\circ + 50 \left( \frac{3}{5} \right) = 0$$

$$F_A = 34.6 \text{ lb} \quad F_B = 57.3 \text{ lb} \quad \text{Ans.}$$

$$\mathbf{R3-2.} \quad \rightarrow \Sigma F_x = 0; \quad F_{AC} \cos 30^\circ - F_{AB} = 0 \quad (1)$$

$$+ \uparrow \Sigma F_y = 0; \quad F_{AC} \sin 30^\circ - W = 0 \quad (2)$$

Assuming cable  $AB$  reaches the maximum tension  $F_{AB} = 450 \text{ lb}$ .

$$\text{From Eq. (1)} \quad F_{AC} \cos 30^\circ - 450 = 0$$

$$F_{AC} = 519.6 \text{ lb} > 480 \text{ lb} \quad (\text{No Good})$$

Assuming cable  $AC$  reaches the maximum tension  $F_{AC} = 480 \text{ lb}$ .

$$\text{From Eq. (1)} \quad 480 \cos 30^\circ - F_{AB} = 0$$

$$F_{AB} = 415.7 \text{ lb} < 450 \text{ lb} \quad (\text{OK})$$

$$\text{From Eq. (2)} \quad 480 \sin 30^\circ - W = 0 \quad W = 240 \text{ lb} \quad \text{Ans.}$$

$$\mathbf{R3-3.} \quad \rightarrow \Sigma F_x = 0; \quad F_{AC} \sin 30^\circ - F_{AB} \left( \frac{3}{5} \right) = 0$$

$$F_{AC} = 1.20 F_{AB} \quad (1)$$

$$+ \uparrow \Sigma F_y = 0; \quad F_{AC} \cos 30^\circ + F_{AB} \left( \frac{4}{5} \right) - W = 0$$

$$0.8660 F_{AC} + 0.8 F_{AB} = W \quad (2)$$

Since  $F_{AC} > F_{AB}$ , failure will occur first at cable  $AC$  with  $F_{AC} = 50 \text{ lb}$ . Then solving Eqs. (1) and (2) yields

$$F_{AB} = 41.67 \text{ lb}$$

$$W = 76.6 \text{ lb} \quad \text{Ans.}$$

$$\mathbf{R3-4.} \quad s_1 = \frac{60}{40} = 1.5 \text{ ft}$$

$$+\uparrow \Sigma F_y = 0; \quad F - 2\left(\frac{1}{2}T\right) = 0; \quad F = T$$

$$\pm \Sigma F_x = 0; \quad -F_s + 2\left(\frac{\sqrt{3}}{2}\right)F = 0$$

$$F_s = 1.732F$$

$$\text{Final stretch is } 1.5 + 0.268 = 1.768 \text{ ft}$$

$$40(1.768) = 1.732F$$

$$F = 40.8 \text{ lb} \quad \text{Ans.}$$

$$\mathbf{R3-5.} \quad \Sigma F_x = 0; \quad -F_1 \sin 45^\circ = 0 \quad F_1 = 0 \quad \text{Ans.}$$

$$\Sigma F_z = 0; \quad F_2 \sin 40^\circ - 200 = 0$$

$$F_2 = 311.14 \text{ lb} = 311 \text{ lb} \quad \text{Ans.}$$

Using the results  $F_1 = 0$  and  $F_2 = 311.14 \text{ lb}$  and then summing forces along the  $y$  axis, we have

$$\Sigma F_y = 0; \quad F_3 - 311.14 \cos 40^\circ = 0$$

$$F_3 = 238 \text{ lb} \quad \text{Ans.}$$

$$\mathbf{R3-6.} \quad \mathbf{F}_1 = F_1 \{\cos 60^\circ \mathbf{i} + \sin 60^\circ \mathbf{k}\}$$

$$= \{0.5F_1 \mathbf{i} + 0.8660F_1 \mathbf{k}\} \text{ N}$$

$$\mathbf{F}_2 = F_2 \left\{ \frac{3}{5} \mathbf{i} - \frac{4}{5} \mathbf{j} \right\}$$

$$= \{0.6 F_2 \mathbf{i} - 0.8 F_2 \mathbf{j}\} \text{ N}$$

$$\mathbf{F}_3 = F_3 \{-\cos 30^\circ \mathbf{i} - \sin 30^\circ \mathbf{j}\}$$

$$= \{-0.8660F_3 \mathbf{i} - 0.5F_3 \mathbf{j}\} \text{ N}$$

$$\Sigma F_x = 0; \quad 0.5F_1 + 0.6F_2 - 0.8660F_3 = 0$$

$$\Sigma F_y = 0; \quad -0.8F_2 - 0.5F_3 + 800 \sin 30^\circ = 0$$

$$\Sigma F_z = 0; \quad 0.8660F_1 - 800 \cos 30^\circ = 0$$

$$F_1 = 800 \text{ N} \quad F_2 = 147 \text{ N} \quad F_3 = 564 \text{ N} \quad \text{Ans.}$$

$$\mathbf{R3-7.} \quad \Sigma F_x = 0; \quad F_{CA} \left( \frac{1}{\sqrt{10}} \right) - F_{CB} \left( \frac{1}{\sqrt{10}} \right) = 0$$

$$\Sigma F_y = 0; \quad -F_{CA} \left( \frac{3}{\sqrt{10}} \right) - F_{CB} \left( \frac{3}{\sqrt{10}} \right) + F_{CD} \left( \frac{3}{5} \right) = 0$$

$$\Sigma F_z = 0; \quad -500 + F_{CD} \left( \frac{4}{5} \right) = 0$$

Solving:

$$F_{CD} = 625 \text{ lb} \quad F_{CA} = F_{CB} = 198 \text{ lb}$$

$$\mathbf{R3-8.} \quad \mathbf{F}_{AB} = 700 \left( \frac{2\mathbf{i} + 3\mathbf{j} - 6\mathbf{k}}{\sqrt{2^2 + 3^2 + (-6)^2}} \right)$$

$$= \{200\mathbf{i} + 300\mathbf{j} - 600\mathbf{k}\} \text{ N}$$

$$\mathbf{F}_{AC} = F_{AC} \left( \frac{-1.5\mathbf{i} + 2\mathbf{j} - 6\mathbf{k}}{\sqrt{(-1.5)^2 + 2^2 + (-6)^2}} \right)$$

$$= -0.2308F_{AC}\mathbf{i} + 0.3077F_{AC}\mathbf{j} - 0.9231F_{AC}\mathbf{k}$$

$$\mathbf{F}_{AD} = F_{AD} \left( \frac{-3\mathbf{i} - 6\mathbf{j} - 6\mathbf{k}}{\sqrt{(-3)^2 + (-6)^2 + (-6)^2}} \right)$$

$$= -0.3333F_{AD}\mathbf{i} - 0.6667F_{AD}\mathbf{j} - 0.6667F_{AD}\mathbf{k}$$

$$\mathbf{F} = F\mathbf{k}$$

$$\Sigma \mathbf{F} = 0; \quad \mathbf{F}_{AB} + \mathbf{F}_{AC} + \mathbf{F}_{AD} + \mathbf{F} = \mathbf{0}$$

$$(200 - 0.2308F_{AC} - 0.3333F_{AD})\mathbf{i}$$

$$+ (300 + 0.3077F_{AC} - 0.6667F_{AD})\mathbf{j}$$

$$+ (-600 - 0.9231F_{AC} - 0.6667F_{AD} + F)\mathbf{k} = \mathbf{0}$$

$$200 - 0.2308F_{AC} - 0.3333F_{AD} = 0$$

$$300 + 0.3077F_{AC} - 0.6667F_{AD} = 0$$

$$-600 - 0.9231F_{AC} - 0.6667F_{AD} + F = 0$$

$$F_{AC} = 130 \text{ N} \quad F_{AD} = 510 \text{ N}$$

$$F = 1060 \text{ N} = 1.06 \text{ kN} \quad \text{Ans.}$$

## Chapter 4

$$\mathbf{R4-1.} \quad 20(10^3) = 800(16 \cos 30^\circ) + W(30 \cos 30^\circ + 2)$$

$$W = 319 \text{ lb} \quad \text{Ans.}$$

$$\mathbf{R4-2.} \quad \mathbf{F}_R = 50 \text{ lb} \left[ \frac{(10\mathbf{i} + 15\mathbf{j} - 30\mathbf{k})}{\sqrt{(10)^2 + (15)^2 + (-30)^2}} \right]$$

$$\mathbf{F}_R = \{14.3\mathbf{i} + 21.4\mathbf{j} - 42.9\mathbf{k}\} \text{ lb} \quad \text{Ans.}$$

$$(\mathbf{M}_R)_C = \mathbf{r}_{CB} \times \mathbf{F} = \begin{vmatrix} \mathbf{i} & \mathbf{j} & \mathbf{k} \\ 10 & 45 & 0 \\ 14.29 & 21.43 & -42.86 \end{vmatrix}$$

$$= \{-1929\mathbf{i} + 428.6\mathbf{j} - 428.6\mathbf{k}\} \text{ lb} \cdot \text{ft} \quad \text{Ans.}$$

$$\mathbf{R4-3.} \quad \mathbf{r} = \{4\mathbf{i}\} \text{ ft}$$

$$\mathbf{F} = 24 \text{ lb} \left( \frac{-2\mathbf{i} + 2\mathbf{j} + 4\mathbf{k}}{\sqrt{(-2)^2 + (2)^2 + (4)^2}} \right)$$

$$= \{-9.80\mathbf{i} + 9.80\mathbf{j} + 19.60\mathbf{k}\} \text{ lb}$$

$$M_y = \begin{vmatrix} 0 & 1 & 0 \\ 4 & 0 & 0 \\ -9.80 & 9.80 & 19.60 \end{vmatrix} = -78.4 \text{ lb} \cdot \text{ft}$$

$$\mathbf{M}_y = \{-78.4\mathbf{j}\} \text{ lb} \cdot \text{ft} \quad \text{Ans.}$$

$$\mathbf{R4-4.} \quad (M_c)_R = \Sigma M_z; \quad 0 = 100 - 0.75F$$

$$F = 133 \text{ N} \quad \text{Ans.}$$

$$\begin{aligned}\mathbf{R4-5.} \quad \rightarrow \Sigma F_{Rx} &= \Sigma F_x; & F_{Rx} &= 6\left(\frac{5}{13}\right) - 4 \cos 60^\circ \\ & & &= 0.30769 \text{ kN} \\ +\uparrow \Sigma F_{Ry} &= \Sigma F_y; & F_{Ry} &= 6\left(\frac{12}{13}\right) - 4 \sin 60^\circ \\ & & &= 2.0744 \text{ kN}\end{aligned}$$

$$F_R = \sqrt{(0.30769)^2 + (2.0744)^2} = 2.10 \text{ kN} \quad \text{Ans.}$$

$$\theta = \tan^{-1}\left[\frac{2.0744}{0.30769}\right] = 81.6^\circ \swarrow \quad \text{Ans.}$$

$$\begin{aligned}\zeta + M_P &= \Sigma M_P; & M_P &= 8 - 6\left(\frac{12}{13}\right)(7) + 6\left(\frac{5}{13}\right)(5) \\ & & &- 4 \cos 60^\circ(4) + 4 \sin 60^\circ(3) \\ M_P &= -16.8 \text{ kN} \cdot \text{m} \\ &= 16.8 \text{ kN} \cdot \text{m} \searrow \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}\mathbf{R4-6.} \quad \rightarrow \Sigma (F_R)_x &= \Sigma F_x; & (F_R)_x &= 200 \cos 45^\circ - 250\left(\frac{4}{5}\right) \\ & & &- 300 = -358.58 \text{ lb} = 358.58 \text{ lb} \leftarrow \\ +\uparrow (F_R)_y &= \Sigma F_y; & (F_R)_y &= -200 \sin 45^\circ - 250\left(\frac{3}{5}\right) \\ & & &= -291.42 \text{ lb} = 291.42 \text{ lb} \downarrow\end{aligned}$$

$$\begin{aligned}F_R &= \sqrt{(F_R)_x^2 + (F_R)_y^2} = \sqrt{358.58^2 + 291.42^2} \\ &= 462.07 \text{ lb} = 462 \text{ lb} \quad \text{Ans.}\end{aligned}$$

$$\theta = \tan^{-1}\left[\frac{(F_R)_y}{(F_R)_x}\right] = \tan^{-1}\left[\frac{291.42}{358.58}\right] = 39.1^\circ \searrow \quad \text{Ans.}$$

$$\begin{aligned}\zeta + (M_R)_A &= \Sigma M_A; & 358.58(d) &= 250\left(\frac{3}{5}\right)(2.5) + 250\left(\frac{4}{5}\right)(4) \\ & & &+ 300(4) - 200 \cos 45^\circ(6) - 200 \sin 45^\circ(3) \\ d &= 3.07 \text{ ft} \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}\mathbf{R4-7.} \quad +\uparrow F_R &= \Sigma F_z; & F_R &= -20 - 50 - 30 - 40 \\ & & &= -140 \text{ kN} = 140 \text{ kN} \downarrow \quad \text{Ans.} \\ (M_R)_x &= \Sigma M_x; & -140y &= -50(3) - 30(11) - 40(13) \\ & & &y = 7.14 \text{ m} \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}(M_R)_y &= \Sigma M_y; & 140x &= 50(4) + 20(10) + 40(10) \\ & & &x = 5.71 \text{ m} \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}\mathbf{R4-8.} \quad +\downarrow F_R &= \Sigma F; & F_R &= 12000 + 6000 = 18000 \text{ lb} \\ & & &F_R = 18.0 \text{ kip} \quad \text{Ans.} \\ \zeta + M_{RC} &= \Sigma M_C; & 18000x &= 12000(7.5) + 6000(20) \\ & & &x = 11.7 \text{ ft} \quad \text{Ans.}\end{aligned}$$

## Chapter 5

$$\begin{aligned}\mathbf{R5-1.} \quad \zeta + \Sigma M_A &= 0; & F(6) + F(4) + F(2) - 3 \cos 45^\circ(2) &= 0 \\ & & &F = 0.3536 \text{ kN} = 354 \text{ N} \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}\mathbf{R5-2.} \quad \zeta + \Sigma M_A &= 0; & N_B(7) - 1400(3.5) - 300(6) &= 0 \\ & & &N_B = 957.14 \text{ N} = 957 \text{ N} \quad \text{Ans.} \\ +\uparrow \Sigma F_y &= 0; & A_y - 1400 - 300 + 957 &= 0 \quad A_y = 743 \text{ N}\end{aligned}$$

$$\rightarrow \Sigma F_x = 0; \quad A_x = 0 \quad \text{Ans.}$$

$$\begin{aligned}\mathbf{R5-3.} \quad \zeta + \Sigma M_A &= 0; & 10(0.6 + 1.2 \cos 60^\circ) + 6(0.4) \\ & & &- N_A(1.2 + 1.2 \cos 60^\circ) = 0 \\ N_A &= 8.00 \text{ kN} \quad \text{Ans.}\end{aligned}$$

$$\rightarrow \Sigma F_x = 0; \quad B_x - 6 \cos 30^\circ = 0; \quad B_x = 5.20 \text{ kN} \quad \text{Ans.}$$

$$\begin{aligned}+\uparrow \Sigma F_y &= 0; & B_y + 8.00 - 6 \sin 30^\circ - 10 &= 0 \\ B_y &= 5.00 \text{ kN} \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}\mathbf{R5-4.} \quad \zeta + \Sigma M_A &= 0; & 50 \cos 30^\circ(20) + 50 \sin 30^\circ(14) \\ & & &- F_B(18) = 0 \\ F_B &= 67.56 \text{ lb} = 67.6 \text{ lb} \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}\rightarrow \Sigma F_x &= 0; & A_x - 50 \sin 30^\circ &= 0 \\ A_x &= 25 \text{ lb} \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}+\uparrow \Sigma F_y &= 0; & A_y - 50 \cos 30^\circ - 67.56 &= 0 \\ A_y &= 110.86 \text{ lb} = 111 \text{ lb} \quad \text{Ans.}\end{aligned}$$

$$\mathbf{R5-5.} \quad \Sigma F_x = 0; \quad A_x = 0 \quad \text{Ans.}$$

$$\begin{aligned}\Sigma F_y &= 0; & A_y + 200 &= 0 \\ A_y &= -200 \text{ N} \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}\Sigma F_z &= 0; & A_z - 150 &= 0 \\ A_z &= 150 \text{ N} \quad \text{Ans.}\end{aligned}$$

$$\begin{aligned}\Sigma M_x &= 0; & -150(2) + 200(2) - (M_A)_x &= 0 \\ (M_A)_x &= 100 \text{ N} \cdot \text{m} \quad \text{Ans.}\end{aligned}$$

$$\Sigma M_y = 0; \quad (M_A)_y = 0 \quad \text{Ans.}$$

$$\begin{aligned}\Sigma M_z &= 0; & 200(2.5) - (M_A)_z &= 0 \\ (M_A)_z &= 500 \text{ N} \cdot \text{m} \quad \text{Ans.}\end{aligned}$$

### R5-6.

$$\Sigma M_y = 0; \quad P(8) - 80(10) = 0 \quad P = 100 \text{ lb} \quad \text{Ans.}$$

$$\Sigma M_x = 0; \quad B_z(28) - 80(14) = 0 \quad B_z = 40 \text{ lb} \quad \text{Ans.}$$

$$\Sigma M_z = 0; \quad -B_x(28) - 100(10) = 0 \quad B_x = -35.7 \text{ lb} \quad \text{Ans.}$$

$$\Sigma F_x = 0; \quad A_x + (-35.7) - 100 = 0 \quad A_x = 136 \text{ lb} \quad \text{Ans.}$$

$$\Sigma F_y = 0; \quad B_y = 0 \quad \text{Ans.}$$

$$\Sigma F_z = 0; \quad A_z + 40 - 80 = 0 \quad A_z = 40 \text{ lb} \quad \text{Ans.}$$

**R5-7.**  $W = (4 \text{ ft})(2 \text{ ft})(2 \text{ lb/ft}^2) = 16 \text{ lb}$

$$\Sigma F_x = 0; \quad A_x = 0 \quad \text{Ans.}$$

$$\Sigma F_y = 0; \quad A_y = 0 \quad \text{Ans.}$$

$$\Sigma F_z = 0; \quad A_z + B_z + C_z - 16 = 0$$

$$\Sigma M_x = 0; \quad 2B_z - 16(1) + C_z(1) = 0$$

$$\Sigma M_y = 0; \quad -B_z(2) + 16(2) - C_z(4) = 0$$

$$A_z + B_z + C_z = 5.33 \text{ lb} \quad \text{Ans.}$$

**R5-8.**

$$\Sigma F_x = 0; \quad A_x = 0 \quad \text{Ans.}$$

$$\Sigma F_y = 0; \quad 350 - 0.6F_{BC} + 0.6F_{BD} = 0$$

$$\Sigma F_z = 0; \quad A_z - 800 + 0.8F_{BC} + 0.8F_{BD} = 0$$

$$\Sigma M_x = 0; \quad (M_A)_x + 0.8F_{BD}(6) + 0.8F_{BC}(6) - 800(6) = 0$$

$$\Sigma M_y = 0; \quad 800(2) - 0.8F_{BC}(2) - 0.8F_{BD}(2) = 0$$

$$\Sigma M_z = 0; \quad (M_A)_z - 0.6F_{BC}(2) + 0.6F_{BD}(2) = 0$$

$$F_{BD} = 208 \text{ N} \quad \text{Ans.}$$

$$F_{BC} = 792 \text{ N} \quad \text{Ans.}$$

$$A_z = 0 \quad \text{Ans.}$$

$$(M_A)_x = 0 \quad \text{Ans.}$$

$$(M_A)_z = 700 \text{ N} \cdot \text{m} \quad \text{Ans.}$$

## Chapter 6

**R6-1.** Joint B:

$$\rightarrow \Sigma F_x = 0; \quad F_{BC} = 3 \text{ kN (C)} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad F_{BA} = 8 \text{ kN (C)} \quad \text{Ans.}$$

Joint A:

$$+\uparrow \Sigma F_y = 0; \quad 8.875 - 8 - \frac{3}{5}F_{AC} = 0$$

$$F_{AC} = 1.458 = 1.46 \text{ kN (C)} \quad \text{Ans.}$$

$$\rightarrow \Sigma F_x = 0; \quad F_{AF} - 3 - \frac{4}{5}(1.458) = 0$$

$$F_{AF} = 4.17 \text{ kN (T)} \quad \text{Ans.}$$

Joint C:

$$\rightarrow \Sigma F_x = 0; \quad 3 + \frac{4}{5}(1.458) - F_{CD} = 0$$

$$F_{CD} = 4.167 = 4.17 \text{ kN (C)} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad F_{CF} - 4 + \frac{3}{5}(1.458) = 0$$

$$F_{CF} = 3.125 = 3.12 \text{ kN (C)} \quad \text{Ans.}$$

Joint E:

$$\rightarrow \Sigma F_x = 0; \quad F_{EF} = 0 \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad F_{ED} = 13.125 = 13.1 \text{ kN (C)} \quad \text{Ans.}$$

Joint D:

$$+\uparrow \Sigma F_y = 0; \quad 13.125 - 10 - \frac{3}{5}F_{DF} = 0$$

$$F_{DF} = 5.21 \text{ kN (T)} \quad \text{Ans.}$$

**R6-2.** Joint A:

$$\rightarrow \Sigma F_x = 0; \quad F_{AB} - F_{AG} \cos 45^\circ = 0$$

$$+\uparrow \Sigma F_y = 0; \quad 333.3 - F_{AG} \sin 45^\circ = 0$$

$$F_{AG} = 471 \text{ lb (C)} \quad \text{Ans.}$$

$$F_{AB} = 333.3 = 333 \text{ lb (T)} \quad \text{Ans.}$$

Joint B:

$$\rightarrow \Sigma F_x = 0; \quad F_{BC} = 333.3 = 333 \text{ lb (T)} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad F_{GB} = 0 \quad \text{Ans.}$$

Joint D:

$$\rightarrow \Sigma F_x = 0; \quad -F_{DC} + F_{DE} \cos 45^\circ = 0 \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad 666.7 - F_{DE} \sin 45^\circ = 0$$

$$F_{DE} = 942.9 \text{ lb} = 943 \text{ lb (C)} \quad \text{Ans.}$$

$$F_{DC} = 666.7 \text{ lb} = 667 \text{ lb (T)} \quad \text{Ans.}$$

Joint E:

$$\rightarrow \Sigma F_x = 0; \quad -942.9 \sin 45^\circ + F_{EG} = 0$$

$$+\uparrow \Sigma F_y = 0; \quad -F_{EC} + 942.9 \cos 45^\circ = 0$$

$$F_{EC} = 666.7 \text{ lb} = 667 \text{ lb (T)} \quad \text{Ans.}$$

$$F_{EG} = 666.7 \text{ lb} = 667 \text{ lb (C)} \quad \text{Ans.}$$

Joint C:

$$+\uparrow \Sigma F_y = 0; \quad F_{GC} \cos 45^\circ + 666.7 - 1000 = 0$$

$$F_{GC} = 471 \text{ lb (T)} \quad \text{Ans.}$$

**R6-3.**  $\zeta + \Sigma M_C = 0; \quad -1000(10) + 1500(20)$

$$- F_{GJ} \cos 30^\circ (20 \tan 30^\circ) = 0$$

$$F_{GJ} = 2.00 \text{ kip (C)} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad -1000 + 2(2000 \cos 60^\circ) - F_{GC} = 0$$

$$F_{GC} = 1.00 \text{ kip (T)} \quad \text{Ans.}$$

**R6-4.**

$$+\uparrow \Sigma F_y = 0; \quad 2A_y - 800 - 600 - 800 = 0 \quad A_y = 1100 \text{ lb}$$

$$\rightarrow \Sigma F_x = 0; \quad A_x = 0$$

$$\zeta + \Sigma M_B = 0; \quad F_{GF} \sin 30^\circ (10) + 800(10 - 10 \cos^2 30^\circ) - 1100(10) = 0$$

$$F_{GF} = 1800 \text{ lb (C)} = 1.80 \text{ kip (C)} \quad \text{Ans.}$$

$$\zeta + \Sigma M_A = 0; \quad F_{FB} \sin 60^\circ (10) - 800(10 \cos^2 30^\circ) = 0$$

$$F_{FB} = 692.82 \text{ lb (T)} = 693 \text{ lb (T)} \quad \text{Ans.}$$

$$\zeta + \Sigma M_F = 0; \quad F_{BC}(15 \tan 30^\circ) + 800(15 - 10 \cos^2 30^\circ) - 1100(15) = 0$$

$$F_{BC} = 1212.43 \text{ lb (T)} = 1.21 \text{ kip (T)} \quad \text{Ans.}$$

**R6-5. Joint A:**

$$\Sigma F_z = 0; \quad F_{AD} \left( \frac{2}{\sqrt{68}} \right) - 600 = 0$$

$$F_{AD} = 2473.86 \text{ lb (T)} = 2.47 \text{ kip (T)} \quad \text{Ans.}$$

$$\Sigma F_x = 0; \quad F_{AC} \left( \frac{1.5}{\sqrt{66.25}} \right) - F_{AB} \left( \frac{1.5}{\sqrt{66.25}} \right) = 0$$

$$F_{AC} = F_{AB}$$

$$\Sigma F_y = 0; \quad F_{AC} \left( \frac{8}{\sqrt{66.25}} \right) + F_{AB} \left( \frac{8}{\sqrt{66.25}} \right)$$

$$- 2473.86 \left( \frac{8}{\sqrt{68}} \right) = 0$$

$$0.9829 F_{AC} + 0.9829 F_{AB} = 2400$$

$$F_{AC} = F_{AB} = 1220.91 \text{ lb (C)} = 1.22 \text{ kip (C)} \quad \text{Ans.}$$

**R6-6. CB is a two force member.****Member AC:**

$$\zeta + \Sigma M_A = 0; \quad -600(0.75) + 1.5(F_{CB} \sin 75^\circ) = 0$$

$$F_{CB} = 310.6$$

$$B_x = B_y = 310.6 \left( \frac{1}{\sqrt{2}} \right) = 220 \text{ N} \quad \text{Ans.}$$

$$\pm \Sigma F_x = 0; \quad -A_x + 600 \sin 60^\circ - 310.6 \cos 45^\circ = 0$$

$$A_x = 300 \text{ N} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad A_y - 600 \cos 60^\circ + 310.6 \sin 45^\circ = 0$$

$$A_y = 80.4 \text{ N} \quad \text{Ans.}$$

**R6-7. Member AB:**

$$\zeta + \Sigma M_A = 0; \quad -750(2) + B_y(3) = 0$$

$$B_y = 500 \text{ N}$$

**Member BC:**

$$\zeta + \Sigma M_C = 0; \quad -1200(1.5) - 900(1) + B_x(3) - 500(3) = 0$$

$$B_x = 1400 \text{ N}$$

$$+\uparrow \Sigma F_y = 0; \quad A_y - 750 + 500 = 0$$

$$A_y = 250 \text{ N} \quad \text{Ans.}$$

**Member AB:**

$$\pm \Sigma F_x = 0; \quad -A_x + 1400 = 0$$

$$A_x = 1400 \text{ N} = 1.40 \text{ kN} \quad \text{Ans.}$$

**Member BC:**

$$\pm \Sigma F_x = 0; \quad C_x + 900 - 1400 = 0$$

$$C_x = 500 \text{ N} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad -500 - 1200 + C_y = 0$$

$$C_y = 1700 \text{ N} = 1.70 \text{ kN} \quad \text{Ans.}$$

$$\mathbf{R6-8.} \quad \zeta + \Sigma M_B = 0; \quad F_{CD}(7) - \frac{4}{5}F_{BE}(2) = 0$$

$$\zeta + \Sigma M_A = 0; \quad -150(7)(3.5) + \frac{4}{5}F_{BE}(5) - F_{CD}(7) = 0$$

$$F_{BE} = 1531 \text{ lb} = 1.53 \text{ kip} \quad \text{Ans.}$$

$$F_{CD} = 350 \text{ lb} \quad \text{Ans.}$$

**Chapter 7**

$$\mathbf{R7-1.} \quad \zeta + \Sigma M_A = 0; \quad F_{CD}(8) - 150(8 \tan 30^\circ) = 0$$

$$F_{CD} = 86.60 \text{ lb}$$

Since member *CF* is a two-force member,

$$V_D = M_D = 0 \quad \text{Ans.}$$

$$N_D = F_{CD} = 86.6 \text{ lb} \quad \text{Ans.}$$

$$\zeta + \Sigma M_A = 0; \quad B_y(12) - 150(8 \tan 30^\circ) = 0$$

$$B_y = 57.735 \text{ lb}$$

$$\pm \Sigma F_x = 0; \quad N_E = 0 \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad V_E + 57.735 - 86.60 = 0$$

$$V_E = 28.9 \text{ lb} \quad \text{Ans.}$$

$$\zeta + \Sigma M_E = 0; \quad 57.735(9) - 86.60(5) - M_E = 0$$

$$M_E = 86.6 \text{ lb} \cdot \text{ft} \quad \text{Ans.}$$

**R7-2. Segment DC**

$$\pm \Sigma F_x = 0; \quad N_C = 0 \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad V_C - 3.00 - 6 = 0 \quad V_C = 9.00 \text{ kN} \quad \text{Ans.}$$

$$\zeta + \Sigma M_C = 0; \quad -M_C - 3.00(1.5) - 6(3) - 40 = 0$$

$$M_C = -62.5 \text{ kN} \cdot \text{m} \quad \text{Ans.}$$

**Segment DB**

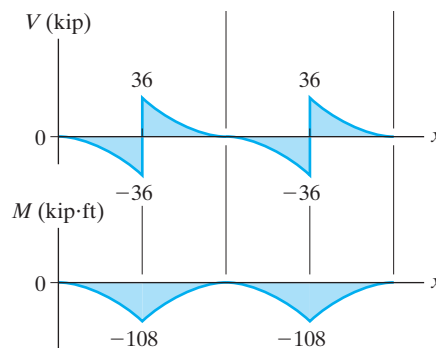
$$\pm \Sigma F_x = 0; \quad N_B = 0 \quad \text{Ans.}$$

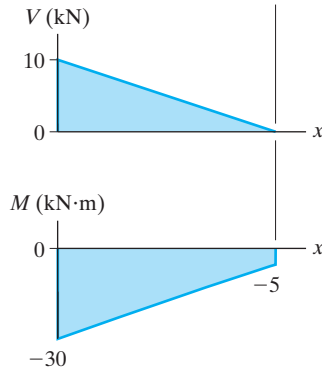
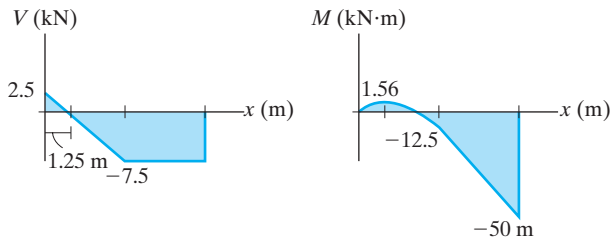
$$+\uparrow \Sigma F_y = 0; \quad V_B - 10.0 - 7.5 - 4.00 - 6 = 0$$

$$V_B = 27.5 \text{ kN} \quad \text{Ans.}$$

$$\zeta + \Sigma M_B = 0; \quad -M_B - 10.0(2.5) - 7.5(5) - 4.00(7) - 6(9) - 40 = 0$$

$$M_B = -184.5 \text{ kN} \cdot \text{m} \quad \text{Ans.}$$

**R7-3.**

**R7-4.****R7-5.****R7-6.**

$$\text{At } x = 30 \text{ ft; } y = 3 \text{ ft; } 3 = \frac{F_H}{0.5} \left[ \cosh \left( \frac{0.5}{F_H} (30) \right) - 1 \right]$$

$$F_H = 75.25 \text{ lb}$$

$$\tan \theta_{\max} = \frac{dy}{dx} \Big|_{x=30 \text{ ft}} = \sinh \left( \frac{0.5(30)}{75.25} \right) \quad \theta_{\max} = 11.346^\circ$$

$$T_{\max} = \frac{F_H}{\cos \theta_{\max}} = \frac{75.25}{\cos 11.346^\circ} = 76.7 \text{ lb} \quad \text{Ans.}$$

**Chapter 8****R8-1.** Assume that the ladder slips at A:

$$F_A = 0.4 N_A$$

$$+\uparrow \Sigma F_y = 0; \quad N_A - 20 = 0$$

$$N_A = 20 \text{ lb}$$

$$F_A = 0.4(20) = 8 \text{ lb}$$

$$\zeta + \Sigma M_B = 0; \quad P(4) - 20(3) + 20(6) - 8(8) = 0$$

$$P = 1 \text{ lb}$$

*Ans.*

$$\pm \Sigma F_x = 0; \quad N_B + 1 - 8 = 0$$

$$N_B = 7 \text{ lb} > 0$$

**OK**

The ladder will remain in contact with the wall.

**R8-2.** Crate

$$+\uparrow \Sigma F_y = 0; \quad N_d - 588.6 = 0 \quad N_d = 588.6 \text{ N}$$

$$\pm \Sigma F_x = 0; \quad P - F_d = 0 \quad (1)$$

$$\zeta + \Sigma M_A = 0; \quad 588.6(x) - P(0.8) = 0 \quad (2)$$

Crate and dolly

$$+\uparrow \Sigma F_y = 0; \quad N_B + N_A - 588.6 - 98.1 = 0 \quad (3)$$

$$\pm \Sigma F_x = 0; \quad P - F_A = 0 \quad (4)$$

$$\zeta + \Sigma M_B = 0; \quad N_A(1.5) - P(1.05) - 588.6(0.95) - 98.1(0.75) = 0 \quad (5)$$

**Friction:** Assuming the crate slips on dolly, then  $F_d = \mu_{sd} N_d = 0.5(588.6) = 294.3 \text{ N}$ . Solving Eqs. (1) and (2)

$$P = 294.3 \text{ N} \quad x = 0.400 \text{ m}$$

Since  $x > 0.3 \text{ m}$ , the crate tips on the dolly. If this is the case  $x = 0.3 \text{ m}$ . Solving Eqs. (1) and (2) with  $x = 0.3 \text{ m}$  yields

$$P = 220.725 \text{ N}$$

$$F_d = 220.725 \text{ N}$$

Assuming the dolly slips at A, then  $F_A = \mu_{sf} N_A = 0.35 N_A$ . Substituting this value into Eqs. (3), (4), and (5) and solving, we have

$$N_A = 559 \text{ N} \quad N_B = 128 \text{ N}$$

$$P = 195.6 \text{ N} = 196 \text{ N (Controls)} \quad \text{Ans.}$$

**R8-3.** Bar

$$\zeta + \Sigma M_B = 0; \quad P(600) - A_y(900) = 0 \quad A_y = 0.6667P$$

Disk

$$+\uparrow \Sigma F_y = 0; \quad N_C \sin 60^\circ - F_C \sin 30^\circ - 0.6667P - 343.35 = 0 \quad (1)$$

$$\zeta + \Sigma M_O = 0; \quad F_C(200) - 0.6667P(200) = 0 \quad (2)$$

**Friction:** If the disk is on the verge of moving, slipping would have to occur at point C. Hence,  $F_C = \mu_s N_C = 0.2 N_C$ . Substituting this into Eqs. (1) and (2) and solving, we have

$$P = 182 \text{ N} \quad \text{Ans.}$$

$$N_C = 606.60 \text{ N}$$

**R8-4.** Cam:

$$\zeta + \Sigma M_O = 0; \quad 5 - 0.4 N_B(0.06) - 0.01(N_B) = 0$$

$$N_B = 147.06 \text{ N}$$

Follower:

$$+\uparrow \Sigma F_y = 0; \quad 147.06 - P = 0$$

$$P = 147 \text{ N} \quad \text{Ans.}$$



$$\mathbf{R8-5.} \quad \rightarrow \Sigma F_x = 0; \quad -P + 0.5(1250) = 0$$

$$P = 625 \text{ lb}$$

Assume block  $B$  slips up and block  $A$  does not move.

Block  $A$ :

$$\rightarrow \Sigma F_x = 0; \quad F_A - N'' = 0$$

$$+\uparrow \Sigma F_y = 0; \quad N_A - 600 - 0.3N'' = 0$$

Block  $B$ :

$$\rightarrow \Sigma F_x = 0; \quad N'' - N' \cos 45^\circ - 0.3 N' \sin 45^\circ = 0$$

$$+\uparrow \Sigma F_y = 0; \quad N' \sin 45^\circ - 0.3 N' \cos 45^\circ - 150 - 0.3 N'' = 0$$

Block  $C$ :

$$\rightarrow \Sigma F_x = 0; \quad 0.3 N' \cos 45^\circ - N' \cos 45^\circ - 0.5 N_C - P = 0$$

$$+\uparrow \Sigma F_y = 0; \quad N_C - N' \sin 45^\circ - 0.3 N' \sin 45^\circ - 500 = 0$$

Solving

$$N'' = 629.0 \text{ lb}, \quad N' = 684.3 \text{ lb}, \quad N_C = 838.7 \text{ lb}, \quad P = 1048 \text{ lb},$$

$$N_A = 411.3 \text{ lb}$$

$$F_A = 629.0 \text{ lb} > 0.5(411.3) = 205.6 \text{ lb} \quad \text{No good}$$

$$\text{All blocks slip at the same time:} \quad P = 625 \text{ lb} \quad \text{Ans.}$$

$$\mathbf{R8-6.} \quad \alpha = \tan^{-1}\left(\frac{10}{25}\right) = 21.80^\circ$$

$$\zeta + \Sigma M_A = 0; \quad -6000(35) + F_{BD} \cos 21.80^\circ(10) + F_{BD} \sin 21.80^\circ(20) = 0$$

$$F_{BD} = 12\,565 \text{ lb}$$

$$\phi_s = \tan^{-1}(0.4) = 21.80^\circ$$

$$\theta = \tan^{-1}\left(\frac{0.2}{2\pi(0.25)}\right) = 7.256^\circ$$

$$M = W r \tan(\theta + \phi)$$

$$M = 12\,565(0.25) \tan(7.256^\circ + 21.80^\circ)$$

$$M = 1745 \text{ lb} \cdot \text{in} = 145 \text{ lb} \cdot \text{ft} \quad \text{Ans.}$$

**R8-7.** Block:

$$+\uparrow \Sigma F_y = 0; \quad N - 100 = 0$$

$$N = 100 \text{ lb}$$

$$\rightarrow \Sigma F_x = 0; \quad T_1 = 0.4(100) = 0$$

$$T_1 = 40 \text{ lb}$$

$$T_2 = T_1 e^{u^{\beta}}; \quad T_2 = 40e^{0.4(\frac{\pi}{2})} = 74.978 \text{ lb}$$

System:

$$\zeta + \Sigma M_A = 0; \quad -100(d) - 40(1) - 50(5) + 74.978(10) = 0$$

$$d = 4.60 \text{ ft} \quad \text{Ans.}$$

$$\mathbf{R8-8.} \quad P \approx \frac{Wa}{r}$$

$$= 500(9.81)\left(\frac{2}{40}\right)$$

$$P = 245 \text{ N} \quad \text{Ans.}$$

## Chapter 9

**R9-1.** Using an element of thickness  $dx$ ,

$$\bar{x} = \frac{\int_A \tilde{x} dA}{\int_A dA} = \frac{\int_a^b x \left(\frac{c^2}{x} dx\right)}{c^2 \ln \frac{b}{a}} = \frac{\int_a^b c^2 dx}{c^2 \ln \frac{b}{a}} = \frac{c^2 x \Big|_a^b}{c^2 \ln \frac{b}{a}} = \frac{b-a}{\ln \frac{b}{a}} \quad \text{Ans.}$$

**R9-2.** Using an element of thickness  $dx$ ,

$$\bar{y} = \frac{\int_A y dA}{\int_A dA} = \frac{\int_a^b \left(\frac{c^2}{2x}\right) \left(\frac{c^2}{x} dx\right)}{c^2 \ln \frac{b}{a}} = \frac{\int_a^b \frac{c^4}{2x^2} dx}{c^2 \ln \frac{b}{a}}$$

$$= \frac{-\frac{c^4}{2x} \Big|_a^b}{c^2 \ln \frac{b}{a}} = \frac{c^2(b-a)}{2ab \ln \frac{b}{a}} \quad \text{Ans.}$$

$$\mathbf{R9-3.} \quad \bar{z} = \frac{\int_v \tilde{z} dV}{\int_v dV} = \frac{\int_0^a z [\pi(a^2 - z^2) dz]}{\int_0^a \pi(a^2 - z^2) dz}$$

$$= \frac{\pi \left( \frac{a^2 z^2}{2} - \frac{z^4}{4} \right) \Big|_0^a}{\pi \left( a^2 z - \frac{z^3}{3} \right) \Big|_0^a} = \frac{3}{8} a \quad \text{Ans.}$$

$$\mathbf{R9-4.} \quad \Sigma \tilde{x} L = 0(4) + 2(\pi)(2) = 12.5664 \text{ ft}^2$$

$$\Sigma \tilde{y} L = 0(4) + \frac{2(2)}{\pi}(\pi)(2) = 8 \text{ ft}^2$$

$$\Sigma \tilde{z} L = 2(4) + 0(\pi)(2) = 8 \text{ ft}^2$$

$$\Sigma L = 4 + \pi(2) = 10.2832 \text{ ft}$$

$$\bar{x} = \frac{\Sigma \tilde{x} L}{\Sigma L} = \frac{12.5664}{10.2832} = 1.22 \text{ ft} \quad \text{Ans.}$$

$$\bar{y} = \frac{\Sigma \tilde{y} L}{\Sigma L} = \frac{8}{10.2832} = 0.778 \text{ ft} \quad \text{Ans.}$$

$$\bar{z} = \frac{\Sigma \tilde{z} L}{\Sigma L} = \frac{8}{10.2832} = 0.778 \text{ ft} \quad \text{Ans.}$$

**R9-5.**

Segment	$A (\text{mm}^2)$	$\tilde{y} (\text{mm})$	$\tilde{y} A (\text{mm}^3)$
1	300(25)	112.5	843 750
2	100(50)	50	250 000
$\Sigma$	12 500		1 093 750

Thus,

$$\bar{y} = \frac{\Sigma \tilde{y}A}{\Sigma A} = \frac{1\,093\,750}{12\,500} = 87.5 \text{ mm}$$

### R9-6.

$$\begin{aligned} A &= \Sigma \theta \tilde{r} L \\ &= 2\pi \left[ 0.6(0.05) + 2(0.6375)\sqrt{(0.025)^2 + (0.075)^2} \right. \\ &\quad \left. + 0.675(0.1) \right] \\ &= 1.25 \text{ m}^2 \end{aligned}$$

### R9-7.

$$\begin{aligned} V &= \Sigma \theta \tilde{r} A \\ &= 2\pi \left[ 2(0.65) \left( \frac{1}{2} (0.025)(0.075) \right) + 0.6375(0.05)(0.075) \right] \\ &= 0.0227 \text{ m}^3 \end{aligned}$$

$$\begin{aligned} \text{R9-8. } dF &= \int dA = 4z^{\frac{1}{3}}(3)dz \\ F &= 12 \int_0^x z^{\frac{1}{3}} dz = 12 \left[ \frac{3}{4} z^{\frac{4}{3}} \right]_0^8 = 144 \text{ lb} \\ \int_A z dF &= 12 \int_0^8 z^{\frac{4}{3}} dz = 12 \left[ \frac{3}{7} z^{\frac{7}{3}} \right]_0^8 = 658.29 \text{ lb} \cdot \text{ft} \\ \bar{z} &= \frac{658.29}{144} = 4.57 \text{ ft} \end{aligned}$$

### R9-9.

$$p_a = 1.0(10^3)(9.81)(9) = 88\,290 \text{ N/m}^2 = 88.29 \text{ kN/m}^2$$

$$p_b = 1.0(10^3)(9.81)(5) = 49\,050 \text{ N/m}^2 = 49.05 \text{ kN/m}^2$$

Thus,

$$w_A = 88.29(8) = 706.32 \text{ kN/m}$$

$$w_B = 49.05(8) = 392.40 \text{ kN/m}$$

$$F_{R_1} = 392.4(5) = 1962.0 \text{ kN}$$

$$F_{R_2} = \frac{1}{2}(706.32 - 392.4)(5) = 784.8 \text{ kN}$$

$$\zeta + \Sigma M_B = 0; \quad 1962.0(2.5) + 784.8(3.333) - A_y(3) = 0$$

$$A_y = 2507 \text{ kN} = 2.51 \text{ MN} \quad \text{Ans.}$$

$$\pm \Sigma F_x = 0; \quad 784.8\left(\frac{4}{5}\right) + 1962\left(\frac{4}{5}\right) - B_x = 0$$

$$B_x = 2197 \text{ kN} = 2.20 \text{ MN} \quad \text{Ans.}$$

$$+\uparrow \Sigma F_y = 0; \quad 2507 - 784.8\left(\frac{3}{5}\right) - 1962\left(\frac{3}{5}\right) - B_y = 0$$

$$B_y = 859 \text{ kN} \quad \text{Ans.}$$

### R9-10.

$$\text{Ans. } A = \int_A dA = \int_{-2}^a -y dx = \int_{-2}^a 2x^2 dx = \frac{2}{3} x^3 \Big|_{-2}^0 = 5.333 \text{ ft}^2$$

$$w = b \gamma h = 1(62.4)(8) = 499.2 \text{ lb} \cdot \text{ft}$$

$$F_y = 5.333(1)(62.4) = 332.8 \text{ lb}$$

$$F_x = \frac{1}{2}(499.2)(8) = 1997 \text{ lb}$$

$$\text{Ans. } F_N = \sqrt{(332.8)^2 + (1997)^2} = 2024 \text{ lb} = 2.02 \text{ kip}$$

## Chapter 10

### R10-1.

$$\begin{aligned} I_x &= \int_A y^2 dA = \int_0^2 y^2(4-x) dy = \int_0^2 y^2(4 - (32)^{\frac{1}{3}} y^{\frac{1}{3}}) dy \\ &= 1.07 \text{ in}^4 \end{aligned} \quad \text{Ans.}$$

### R10-2.

$$\begin{aligned} I_x &= \int_A y^2 dA = \int_0^1 y^2(2x dy) = \int_0^1 y^2(4(1-y)^{\frac{1}{2}}) dy \\ &= 0.610 \text{ ft}^4 \end{aligned} \quad \text{Ans.}$$

### R10-3.

$$\begin{aligned} I_y &= \int_A x^2 dA = 2 \int_0^2 x^2(y dx) = 2 \int_0^2 x^2(1 - 0.25 x^2) dx \\ &= 2.13 \text{ ft}^4 \end{aligned} \quad \text{Ans.}$$

$$\begin{aligned} \text{R10-4. } dI_{xy} &= d\bar{I}_{x^2y^2} + dA\bar{x}\bar{y} = 0 + (y^{\frac{1}{3}} dy) \left( \frac{1}{2} y^{\frac{1}{3}} \right) (y) \\ &= \frac{1}{2} y^{\frac{5}{3}} dy \end{aligned}$$

$$I_{xy} = \int dI_{xy} = \int_0^{1 \text{ m}} \frac{1}{2} y^{\frac{5}{3}} dy = \frac{3}{16} y^{\frac{8}{3}} \Big|_0^{1 \text{ m}} = 0.1875 \text{ m}^4 \quad \text{Ans.}$$

$$\text{R10-5. } \frac{s}{h-y} = \frac{b}{h}, \quad s = \frac{b}{h}(h-y)$$

$$(a) dA = s dy = \left[ \frac{b}{h}(h-y) \right] dy$$

$$I_x = \int y^2 dA = \int_0^h y^2 \left[ \frac{b}{h}(h-y) \right] dy = \frac{bh^3}{12} \quad \text{Ans.}$$

$$(b) I_x = \bar{I}_{x'} + A d^2 = \bar{I}_{x'} + \frac{1}{2} bh \left( \frac{h}{3} \right)^2 \quad I_x = \frac{bh^3}{36} \quad \text{Ans.}$$

$$\begin{aligned} \text{R10-6. } dI_{xy} &= dI_{x^2y^2} + dA \bar{x} \bar{y} \\ &= 0 + (y^{\frac{1}{3}} dy) \left( \frac{1}{2} y^{\frac{1}{3}} \right) (y) \\ &= \frac{1}{2} y^{\frac{5}{3}} dy \end{aligned}$$

$$I_{xy} = \int dI_{xy} = \int_0^{1\text{ m}} \frac{1}{2} y^{\frac{5}{3}} dy = \frac{3}{16} y^{\frac{8}{3}} \Big|_0^1 = 0.1875 \text{ m}^4 \quad \text{Ans.}$$

$$\begin{aligned} \mathbf{R10-7.} \quad I_y &= \left[ \frac{1}{12}(d)(d^3) + 0 \right] + 4 \left[ \frac{1}{36}(0.2887d) \left( \frac{d}{2} \right)^3 \right. \\ &\quad \left. + \frac{1}{2}(0.2887d) \left( \frac{d}{2} \right) \left( \frac{d}{6} \right)^2 \right] \\ &= 0.0954d^4 \quad \text{Ans.} \end{aligned}$$

$$\begin{aligned} \mathbf{R10-8.} \quad dI_x &= \frac{1}{2}\rho\pi y^4 dx = \frac{1}{2}\rho\pi \left( \frac{b^4}{a^4}x^4 + \frac{4b^4}{a^3}x^3 + \frac{6b^4}{a^2}x^2 \right. \\ &\quad \left. + \frac{4b^4}{a}x + b^4 \right) dx \\ I_x &= \int dI_x = \frac{1}{2}\rho\pi \int_0^a \left( \frac{b^4}{a^4}x^4 + \frac{4b^4}{a^3}x^3 + \frac{6b^4}{a^2}x^2 \right. \\ &\quad \left. + \frac{4b^4}{a}x + b^4 \right) dx \\ &= \frac{31}{10}\rho\pi ab^4 \\ m &= \int_m dm = \int_0^a \rho\pi y^2 dx \\ &= \rho\pi \int_0^a \left( \frac{b^2}{a^2}x^2 + \frac{2b^2}{a}x + b^2 \right) dx \\ &= \frac{7}{3}\rho\pi ab^2 \\ I_x &= \frac{93}{70}mb^2 \quad \text{Ans.} \end{aligned}$$

## Chapter 11

$$\begin{aligned} \mathbf{R11-1.} \quad x &= 2L \cos \theta \\ \delta x &= -2L \sin \theta \delta \theta \\ y &= L \sin \theta \\ \delta y &= L \cos \theta \delta \theta \\ \delta U &= 0; \quad -P\delta y - F\delta x = 0 \\ -PL \cos \theta \delta \theta - F(-2L \sin \theta) \delta \theta &= 0 \\ -P \cos \theta + 2F \sin \theta &= 0 \\ F &= \frac{P}{2 \tan \theta} \quad \text{Ans.} \end{aligned}$$

$$\begin{aligned} \mathbf{R11-2.} \quad y_B &= 10 \sin \theta & \delta y_B &= 10 \cos \theta \delta \theta \\ y_D &= 5 \sin \theta & \delta y_D &= 5 \cos \theta \delta \theta \\ x_C &= 2(10 \cos \theta) & \delta x_C &= -20 \sin \theta \delta \theta \end{aligned}$$

$$\begin{aligned} \delta U &= 0; \quad -F_{sp}\delta x_C - 2(2\delta y_D - 20\delta y_B + P\delta x_C) = 0 \\ (20F_{sp} \sin \theta - 20P \sin \theta - 220 \cos \theta) \delta \theta &= 0 \end{aligned}$$

However, from the spring formula,  
 $F_{sp} = kx = 2[2(10 \cos \theta) - 6] = 40 \cos \theta - 12$ .

Substituting

$$(800 \sin \theta \cos \theta - 240 \sin \theta - 220 \cos \theta - 20P \sin \theta) \delta \theta = 0$$

Since  $\delta \theta \neq 0$ , then

$$\begin{aligned} 800 \sin \theta \cos \theta - 240 \sin \theta - 220 \cos \theta - 20P \sin \theta &= 0 \\ P &= 40 \cos \theta - 11 \cot \theta - 12 \end{aligned}$$

At the equilibrium position,  $\theta = 45^\circ$ . Then

$$P = 40 \cos 45^\circ - 11 \cot 45^\circ - 12 = 5.28 \text{ lb} \quad \text{Ans.}$$

**R11-3.** Using the law of cosines,

$$0.4^2 = x_A^2 + 0.1^2 - 2(x_A)(0.1)\cos \theta$$

Differentiating,

$$0 = 2x_A \delta x_A - 0.2 \delta x_A \cos \theta + 0.2x_A \sin \theta \delta \theta$$

$$\delta x_A = \frac{0.2x_A \sin \theta}{0.2 \cos \theta - 2x_A} \delta \theta$$

$$\delta U = 0; \quad -F\delta x_A - 50\delta \theta = 0$$

$$\left( \frac{0.2x_A \sin \theta}{0.2 \cos \theta - 2x_A} F - 50 \right) \delta \theta = 0$$

Since  $\delta \theta \neq 0$ , then

$$\frac{0.2x_A \sin \theta}{0.2 \cos \theta - 2x_A} F - 50 = 0$$

$$F = \frac{50(0.2 \cos \theta - 2x_A)}{0.2x_A \sin \theta}$$

At the equilibrium position,  $\theta = 60^\circ$ ,

$$0.4^2 = x_A^2 + 0.1^2 - 2(x_A)(0.1) \cos 60^\circ$$

$$x_A = 0.4405 \text{ m}$$

$$F = -\frac{50[0.2 \cos 60^\circ - 2(0.4405)]}{0.2(0.4405) \sin 60^\circ} = 512 \text{ N} \quad \text{Ans.}$$

**R11-4.**  $y = 4 \sin \theta$

$$\delta y = 4 \cos \theta \delta \theta$$

$$F_s = 5(4 - 4 \sin \theta)$$

$$\delta U = 0; \quad -10\delta y + F_s\delta y = 0$$

$$[-10 + 20(1 - \sin \theta)](4 \cos \theta \delta \theta) = 0$$

$$\cos \theta = 0 \quad \text{and} \quad 10 - 20 \sin \theta = 0$$

$$\theta = 90^\circ \quad \theta = 30^\circ \quad \text{Ans.}$$

**R11-5.**  $x_B = 0.1 \sin \theta$        $\delta x_B = 0.1 \cos \theta \delta \theta$

$$x_D = 2(0.7 \sin \theta) - 0.1 \sin \theta = 1.3 \sin \theta \quad \delta x_D = 1.3 \cos \theta \delta \theta$$

$$y_G = 0.35 \cos \theta \quad \delta y_G = -0.35 \sin \theta \delta \theta$$

$$\delta U = 0; \quad 2(-49.05\delta y_G) + F_{sp}(\delta x_B - \delta x_D) = 0$$

$$(34.335 \sin \theta - 1.2F_{sp} \cos \theta)\delta\theta = 0$$

However, from the spring formula,

$$F_{sp} = kx = 400[2(0.6 \sin \theta) - 0.3] = 480 \sin \theta - 120.$$

Substituting,

$$(34.335 \sin \theta - 576 \sin \theta \cos \theta + 144 \cos \theta)\delta\theta = 0$$

Since  $\delta\theta \neq 0$ , then

$$34.335 \sin \theta - 576 \sin \theta \cos \theta + 144 \cos \theta = 0$$

$$\theta = 15.5^\circ \quad \text{Ans.}$$

$$\text{and } \theta = 85.4^\circ \quad \text{Ans.}$$

### R11-6.

$$V_g = mgy = 40(9.81)(0.45 \sin \theta + b) = 176.58 \sin \theta + 392.4 b$$

$$V_e = \frac{1}{2}(1500)(0.45 \cos \theta)^2 = 151.875 \cos^2 \theta$$

$$V = V_g + V_e = 176.58 \sin \theta + 151.875 \cos^2 \theta + 392.4 b$$

$$\frac{dV}{d\theta} = 176.58 \cos \theta - 303.75 \cos \theta \sin \theta = 0$$

$$\cos \theta(176.58 - 303.75 \sin \theta) = 0$$

$$\cos \theta = 0 \quad \theta = 90^\circ \quad \text{Ans.}$$

$$\theta = 35.54^\circ = 35.5^\circ \quad \text{Ans.}$$

$$\frac{d^2V}{d^2\theta} = -176.58 \sin \theta - 303.75 \cos 2\theta$$

$$\text{At } \theta = 90^\circ, \frac{d^2V}{d^2\theta} \Big|_{\theta=90^\circ} = -176.58 \sin 90^\circ - 303.75 \cos 180^\circ$$

$$= 127.17 > 0$$

$$= 127.17 > 0 \quad \text{Stable} \quad \text{Ans.}$$

$$\text{At } \theta = 35.54^\circ, \frac{d^2V}{d^2\theta} \Big|_{\theta=35.54^\circ} = -176.58 \sin 35.54^\circ$$

$$- 303.75 \cos 71.09^\circ$$

$$= -201.10 < 0 \quad \text{Unstable} \quad \text{Ans.}$$

$$\begin{aligned} \text{R11-7. } V &= V_e + V_g \\ &= \frac{1}{2}(24)(2 \cos \theta)^2 + \frac{1}{2}(48)(6 \cos \theta)^2 \\ &\quad + 100(3 \sin \theta) \\ &= 912 \cos^2 \theta + 300 \sin \theta \end{aligned}$$

$$\frac{dV}{d\theta} = -1824 \sin \theta \cos \theta + 300 \cos \theta = 0$$

$$\frac{dV}{d\theta} = -912 \sin 2\theta + 300 \cos \theta = 0$$

$$\theta = 90^\circ \quad \text{or} \quad \theta = 9.467^\circ$$

$$\frac{d^2V}{d\theta^2} = -1824 \cos 2\theta - 300 \sin \theta$$

$$\begin{aligned} \frac{d^2V}{d\theta^2} \Big|_{\theta=90^\circ} &= -1824 \cos 180^\circ - 300 \sin 90^\circ \\ &= 1524 > 0 \end{aligned} \quad \text{Ans.}$$

$$\begin{aligned} \frac{d^2V}{d\theta^2} \Big|_{\theta=9.467^\circ} &= -1824 \cos 18.933^\circ - 300 \sin 9.467^\circ \\ &= 1774.7 < 0 \end{aligned}$$

Thus, the system is in unstable equilibrium at  $\theta = 9.47^\circ$ . *Ans.*

### R11-8. $V = V_e + V_g$

$$= \frac{1}{2}kx^2 - Wy$$

$$= \frac{1}{2}(16)(2.5 - 2.5 \sin \theta)^2 - 20(2.5 \cos \theta)$$

$$= 50 \sin^2 \theta - 100 \sin \theta - 50 \cos \theta + 50$$

$$\frac{dV}{d\theta} = 100 \sin \theta \cos \theta - 100 \cos \theta + 50 \sin \theta = 0$$

$$\frac{dV}{d\theta} = 50 \sin 2\theta - 100 \cos \theta + 50 \sin \theta = 0$$

$$\theta = 37.77^\circ = 37.8^\circ$$

$$\frac{d^2V}{d\theta^2} = 100 \cos 2\theta + 100 \sin \theta + 50 \cos \theta$$

$$\begin{aligned} \frac{d^2V}{d\theta^2} \Big|_{\theta=37.77^\circ} &= 100 \cos 75.55^\circ + 100 \sin 37.77^\circ + 50 \cos 37.77^\circ \\ &= 125.7 > 0 \end{aligned}$$

Thus, the system is in stable equilibrium at  $\theta = 37.8^\circ$ . *Ans.*

# Answers to Selected Problems

## Chapter 1

- 1-1. a. 78.5 N  
b. 0.392 mN  
c. 746 MN
- 1-2. a. GN/s  
b. Gg/N  
c. GN/(kg · s)
- 1-3. a. Gg/s  
b. kN/m  
c. kN/(kg · s)
- 1-5. a. 45.3 MN  
b. 56.8 km  
c. 5.63  $\mu$ g
- 1-6. a. 58.3 km  
b. 68.5 s  
c. 2.55 kN  
d. 7.56 mg
- 1-7. a. 0.431 g  
b. 35.3 kN  
c. 5.32 m
- 1-9. a. km/s  
b. mm  
c. Gs/kg  
d. mm · N
- 1-10. a. kN · m  
b. Gg/m  
c.  $\mu$ N/s<sup>2</sup>  
d. GN/s
- 1-11. a. 8.653 s  
b. 8.368 kN  
c. 893 g
- 1-13. 2.71 Mg/m<sup>3</sup>
- 1-14. a. 44.9(10)<sup>-3</sup> N<sup>2</sup>  
b. 2.79(10<sup>3</sup>) s<sup>2</sup>  
c. 23.4 s
- 1-15. 7.41  $\mu$ N
- 1-17. 1.00 Mg/m<sup>3</sup>
- 1-18. a. 0.447 kg · m/N  
b. 0.911 kg · s  
c. 18.8 GN/m
- 1-19. 1.04 kip
- 1-21.  $F = 10.0$  nN,  $W_1 = 78.5$  N,  $W_2 = 118$  N

## Chapter 2

- 2-1.  $F_R = 497$  N,  $\phi = 155^\circ$   
2-2.  $F = 960$  N,  $\theta = 45.2^\circ$   
2-3.  $F_R = 393$  lb,  $\phi = 353^\circ$

- 2-5.  $F_{AB} = 314$  lb,  $F_{AC} = 256$  lb  
2-6.  $\phi = 1.22^\circ$   
2-7.  $(F_1)_v = 2.93$  kN,  $(F_1)_u = 2.07$  kN  
2-9.  $F = 616$  lb,  $\theta = 46.9^\circ$   
2-10.  $F_R = 980$  lb,  $\phi = 19.4^\circ$   
2-11.  $F_R = 10.8$  kN,  $\phi = 3.16^\circ$   
2-13.  $F_a = 30.6$  lb,  $F_b = 26.9$  lb  
2-14.  $F = 19.6$  lb,  $F_b = 26.4$  lb  
2-15.  $F = 917$  lb,  $\theta = 31.8^\circ$   
2-17.  $F_R = 19.2$  N,  $\theta = 2.37^\circ \swarrow$   
2-18.  $F_R = 19.2$  N,  $\theta = 2.37^\circ \swarrow$   
2-19.  $\theta = 53.5^\circ$ ,  $F_{AB} = 621$  lb  
2-21.  $F_R = 257$  N,  $\phi = 163^\circ$   
2-22.  $F_R = 257$  N,  $\phi = 163^\circ$   
2-23.  $\theta = 75.5^\circ$   
2-25.  $\theta = 36.3^\circ$ ,  $\phi = 26.4^\circ$   
2-26.  $\theta = 54.3^\circ$ ,  $F_A = 686$  N  
2-27.  $F_R = 1.23$  kN,  $\theta = 6.08^\circ$   
2-29.  $F_B = 1.61$  kN,  $\theta = 38.3^\circ$   
2-30.  $F_R = 4.01$  kN,  $\phi = 16.2^\circ$   
2-31.  $\theta = 90^\circ$ ,  $F_B = 1$  kN,  $F_R = 1.73$  kN  
2-33.  $F_R = 983$  N,  $\theta = 21.8^\circ$   
2-34.  $\mathbf{F}_1 = \{200\mathbf{i} + 346\mathbf{j}\}$  N,  $\mathbf{F}_2 = \{177\mathbf{i} - 177\mathbf{j}\}$  N  
2-35.  $F_R = 413$  N,  $\theta = 24.2^\circ$   
2-37.  $F_R = 1.96$  kN,  $\theta = 4.12^\circ$   
2-38.  $\mathbf{F}_1 = \{30\mathbf{i} + 40\mathbf{j}\}$  N,  $\mathbf{F}_2 = \{-20.7\mathbf{i} - 77.3\mathbf{j}\}$  N,  
 $\mathbf{F}_3 = \{30\mathbf{i}\}$ ,  $F_R = 54.2$  N,  $\theta = 43.5^\circ$   
2-39.  $F_{1x} = 141$  N,  $F_{1y} = 141$  N,  $F_{2x} = -130$  N,  
 $F_{2y} = 75$  N  
2-41.  $F_R = 12.5$  kN,  $\theta = 64.1^\circ$   
2-42.  $\mathbf{F}_1 = \{680\mathbf{i} - 510\mathbf{j}\}$  N,  $\mathbf{F}_2 = \{-312\mathbf{i} - 541\mathbf{j}\}$  N,  
 $\mathbf{F}_3 = \{-530\mathbf{i} + 530\mathbf{j}\}$  N  
2-43.  $F_R = 546$  N,  $\theta = 253^\circ$   
2-45.  $F_R = \sqrt{F_1^2 + F_2^2 + 2F_1F_2 \cos \phi}$ ,  
 $\theta = \tan^{-1}\left(\frac{F_1 \sin \phi}{F_2 + F_1 \cos \phi}\right)$   
2-46.  $\theta = 68.6^\circ$ ,  $F_B = 960$  N  
2-47.  $F_R = 839$  N,  $\theta = 14.8^\circ$   
2-49.  $F_R = 389$  N,  $\phi' = 42.7^\circ$   
2-50.  $\mathbf{F}_1 = \{9.64\mathbf{i} + 11.5\mathbf{j}\}$  kN,  $\mathbf{F}_2 = \{-24\mathbf{i} + 10\mathbf{j}\}$  kN,  
 $\mathbf{F}_3 = \{31.2\mathbf{i} - 18\mathbf{j}\}$  kN  
2-51.  $F_R = 17.2$  kN,  $\theta = 11.7^\circ$   
2-53.  $\mathbf{F}_1 = \{-15.0\mathbf{i} - 26.0\mathbf{j}\}$  kN,  
 $\mathbf{F}_2 = \{-10.0\mathbf{i} + 24.0\mathbf{j}\}$  kN  
2-54.  $F_R = 25.1$  kN,  $\theta = 185^\circ$   
2-55.  $F = 2.03$  kN,  $F_R = 7.87$  kN  
2-57.  $F_R = 380$  N,  $F_1 = 57.8$  N

- 2-58.  $\theta = 86.0^\circ$ ,  $F = 1.97$  kN  
 2-59.  $F_R = 11.1$  kN,  $\theta = 47.7^\circ$   
 2-61.  $F_x = 40$  N,  $F_y = 40$  N,  $F_z = 56.6$  N  
 2-62.  $\alpha = 48.4^\circ$ ,  $\beta = 124^\circ$ ,  $\gamma = 60^\circ$ ,  $F = 8.08$  kN  
 2-63.  $F_R = 114$  lb,  $\alpha = 62.1^\circ$ ,  $\beta = 113^\circ$ ,  $\gamma = 142^\circ$   
 2-65.  $\mathbf{F}_1 = \{-106\mathbf{i} + 106\mathbf{j} + 260\mathbf{k}\}$  N,  
 $\mathbf{F}_2 = \{250\mathbf{i} + 354\mathbf{j} - 250\mathbf{k}\}$  N,  
 $\mathbf{F}_R = \{144\mathbf{i} + 460\mathbf{j} + 9.81\mathbf{k}\}$  N,  $F_R = 482$  N,  
 $\alpha = 72.6^\circ$ ,  $\beta = 17.4^\circ$ ,  $\gamma = 88.8^\circ$   
 2-66.  $\alpha_1 = 111^\circ$ ,  $\beta_1 = 69.3^\circ$ ,  $\gamma_1 = 30.0^\circ$   
 2-67.  $F_3 = 428$  lb,  $\alpha = 88.3^\circ$ ,  $\beta = 20.6^\circ$ ,  $\gamma = 69.5^\circ$   
 2-69.  $F_R = 430$  N,  $\alpha = 28.9^\circ$ ,  $\beta = 67.3^\circ$ ,  $\gamma = 107^\circ$   
 2-70.  $F_R = 384$  N,  $\cos \alpha = 14.8^\circ$ ,  $\cos \beta = 88.9^\circ$ ,  
 $\cos \gamma = 105^\circ$   
 2-71.  $F_1 = 429$  lb,  $\alpha_1 = 62.2^\circ$ ,  $\beta_1 = 110^\circ$ ,  $\gamma_1 = 145^\circ$   
 2-73.  $\mathbf{F}_1 = \{72.0\mathbf{i} + 54.0\mathbf{k}\}$  N,  
 $\mathbf{F}_2 = \{53.0\mathbf{i} + 53.0\mathbf{j} + 130\mathbf{k}\}$  N,  $\mathbf{F}_3 = \{200\mathbf{k}\}$   
 2-74.  $F_R = 407$  N,  $\alpha = 72.1^\circ$ ,  $\beta = 82.5^\circ$ ,  $\gamma = 19.5^\circ$   
 2-75.  $\mathbf{F}_1 = \{14.0\mathbf{j} - 48.0\mathbf{k}\}$  lb,  
 $\mathbf{F}_2 = \{90\mathbf{i} - 127\mathbf{j} + 90\mathbf{k}\}$  lb  
 2-77.  $F_R = 610$  N,  $\alpha = 19.4^\circ$ ,  $\beta = 77.5^\circ$ ,  $\gamma = 105^\circ$   
 2-78.  $F_2 = 66.4$  lb,  $\alpha = 59.8^\circ$ ,  $\beta = 107^\circ$ ,  $\gamma = 144^\circ$   
 2-79.  $\alpha = 124^\circ$ ,  $\beta = 71.3^\circ$ ,  $\gamma = 140^\circ$   
 2-81.  $F_R = 1.55$  kip,  $\alpha = 82.4^\circ$ ,  $\beta = 37.6^\circ$ ,  $\gamma = 53.4^\circ$   
 2-82.  $F_R = 1.60$  kN,  $\alpha = 82.6^\circ$ ,  $\beta = 29.4^\circ$ ,  $\gamma = 61.7^\circ$   
 2-83.  $\alpha_3 = 139^\circ$ ,  
 $\beta_3 = 128^\circ$ ,  $\gamma_3 = 102^\circ$ ,  $F_{R1} = 387$  N,  
 $\beta_3 = 60.7^\circ$ ,  $\gamma_3 = 64.4^\circ$ ,  $F_{R2} = 1.41$  kN  
 2-85.  $F = 2.02$  kN,  $F_y = 0.523$  kN  
 2-86.  $r_{AB} = 397$  mm  
 2-87.  $\mathbf{F} = \{59.4\mathbf{i} - 88.2\mathbf{j} - 83.2\mathbf{k}\}$  lb,  $\alpha = 63.9^\circ$ ,  
 $\beta = 131^\circ$ ,  $\gamma = 128^\circ$   
 2-89.  $x = 5.06$  m,  $y = 3.61$  m,  $z = 6.51$  m  
 2-90.  $z = 6.63$  m  
 2-91.  $x = y = 4.42$  m  
 2-93.  $F_R = 1.17$  kN,  $\alpha = 66.9^\circ$ ,  $\beta = 92.0^\circ$ ,  $\gamma = 157^\circ$   
 2-94.  $F_R = 1.17$  kN,  $\alpha = 68.0^\circ$ ,  $\beta = 96.8^\circ$ ,  $\gamma = 157^\circ$   
 2-95.  $\mathbf{F}_{BA} = \{-109\mathbf{i} + 131\mathbf{j} + 306\mathbf{k}\}$  lb,  
 $\mathbf{F}_{CA} = \{103\mathbf{i} + 103\mathbf{j} + 479\mathbf{k}\}$  lb,  
 $\mathbf{F}_{DA} = \{-52.1\mathbf{i} - 156\mathbf{j} + 365\mathbf{k}\}$  lb  
 2-97.  $F_R = 757$  N,  $\alpha = 149^\circ$ ,  $\beta = 90.0^\circ$ ,  $\gamma = 59.0^\circ$   
 2-98.  $F = \{-34.3\mathbf{i} + 22.9\mathbf{j} - 68.6\mathbf{k}\}$  lb  
 2-99.  $\mathbf{F} = \{13.4\mathbf{i} + 23.2\mathbf{j} + 53.7\mathbf{k}\}$  lb  
 2-101.  $\mathbf{F}_A = \{169\mathbf{i} + 33.8\mathbf{j} - 101\mathbf{k}\}$  lb,  
 $\mathbf{F}_B = \{97.6\mathbf{i} + 97.6\mathbf{j} - 58.6\mathbf{k}\}$  lb,  
 $F_R = 338$  lb,  $\alpha = 37.8^\circ$ ,  
 $\beta = 67.1^\circ$ ,  $\gamma = 118^\circ$   
 2-102.  $\mathbf{F}_1 = \{389\mathbf{i} - 64.9\mathbf{j} + 64.9\mathbf{k}\}$  lb,  
 $\mathbf{F}_2 = \{-584\mathbf{i} + 97.3\mathbf{j} - 97.3\mathbf{k}\}$  lb  
 2-103.  $F_R = 52.2$  lb,  $\alpha = 87.8^\circ$ ,  $\beta = 63.7^\circ$ ,  $\gamma = 154^\circ$   
 2-105.  $F = 105$  lb

- 2-106.  $\mathbf{F} = \{466\mathbf{i} + 339\mathbf{j} - 169\mathbf{k}\}$  N  
 2-107.  $\mathbf{F} = \{476\mathbf{i} + 329\mathbf{j} - 159\mathbf{k}\}$  N  
 2-109.  $F = 52.1$  lb  
 2-110.  $r_{AB} = 10.0$  ft,  
 $\mathbf{F} = \{-19.1\mathbf{i} - 14.9\mathbf{j} + 43.7\mathbf{k}\}$  lb,  $\alpha = 112^\circ$ ,  
 $\beta = 107^\circ$ ,  $\gamma = 29.0^\circ$   
 2-111.  $r_{AB} = 592$  mm,  $\mathbf{F} = \{-13.2\mathbf{i} - 17.7\mathbf{j} + 20.3\mathbf{k}\}$  N  
 2-113.  $(F_{ED})_{||} = 334$  N,  $(F_{ED})_{\perp} = 498$  N  
 2-114.  $\theta = 36.4^\circ$   
 2-115.  $(F_1)_{AC} = 56.3$  N  
 2-117.  $|\text{Proj } F_{AB}| = 70.5$  N,  $|\text{Proj } F_{AC}| = 65.1$  N  
 2-118.  $\theta = 31.0^\circ$   
 2-119.  $F_1 = 18.3$  lb,  $F_2 = 35.6$  lb  
 2-121.  $\theta = 100^\circ$   
 2-122.  $\theta = 19.2^\circ$   
 2-123.  $F_{BA} = 187$  N  
 2-125.  $F_u = 246$  N  
 2-126.  $F_{||} = 10.5$  lb  
 2-127.  $\theta = 142^\circ$   
 2-129.  $F_{||} = 0.182$  kN  
 2-130.  $\theta = 74.4^\circ$ ,  $\phi = 55.4^\circ$   
 2-131.  $(F_{BC})_{||} = 28.3$  lb,  $(F_{BC})_{\perp} = 68.0$  lb  
 2-133.  $\theta = 132^\circ$   
 2-134.  $\theta = 23.4^\circ$   
 2-135.  $[(F)_{AB}]_{||} = 63.2$  lb,  $[(F)_{AB}]_{\perp} = 64.1$  lb  
 2-137.  $F_{OA} = 242$  N  
 2-138.  $\theta = 82.9^\circ$   
 2-139.  $\text{Proj } \mathbf{F}_{AB} = \{0.229\mathbf{i} - 0.916\mathbf{j} + 1.15\mathbf{k}\}$  lb

## Chapter 3

- 3-1.  $F_2 = 9.60$  kN,  $F_1 = 1.83$  kN  
 3-2.  $\theta = 4.69^\circ$ ,  $F_1 = 4.31$  kN  
 3-3.  $\theta = 82.2^\circ$ ,  $F = 3.96$  kN  
 3-5.  $T = 7.20$  kN,  $F = 5.40$  kN  
 3-6.  $T = 7.66$  kN,  $\theta = 70.1^\circ$   
 3-7.  $\theta = 20^\circ$ ,  $T = 30.5$  lb  
 3-9.  $F = 960$  lb  
 3-10.  $\theta = 40^\circ$ ,  $T_{AB} = 37.6$  lb  
 3-11.  $\theta = 40^\circ$ ,  $W = 42.6$  lb  
 3-13.  $F_{CA} = 500(10^3)$  lb,  $F_{AB} = 433(10^3)$  lb,  
 $F_{AD} = 250(10^3)$  lb  
 3-14.  $x_{AD} = 0.4905$  m,  $x_{AC} = 0.793$  m,  $x_{AB} = 0.467$  m  
 3-15.  $m = 8.56$  kg  
 3-17.  $\frac{1}{k_T} = \frac{1}{k_1} + \frac{1}{k_2}$   
 3-18.  $k = 176$  N/m  
 3-19.  $l_0 = 2.03$  m  
 3-21.  $l = 2.66$  ft  
 3-22.  $F = 158$  N  
 3-23.  $d = 1.56$  m  
 3-25.  $y = 2$  m,  $F_1 = 833$  N

- 3-26.  $T_{HA} = 294 \text{ N}$ ,  $T_{AB} = 340 \text{ N}$ ,  $T_{AE} = 170 \text{ N}$ ,  
 $T_{BD} = 490 \text{ N}$ ,  $T_{BC} = 562 \text{ N}$
- 3-27.  $m = 26.7 \text{ kg}$
- 3-29.  $F_{DE} = 392 \text{ N}$ ,  $F_{CD} = 340 \text{ N}$ ,  $F_{CB} = 275 \text{ N}$ ,  
 $F_{CA} = 243 \text{ N}$
- 3-30.  $m = 20.4 \text{ kg}$
- 3-31.  $s = 3.38 \text{ m}$ ,  $F = 76.0 \text{ N}$
- 3-33.  $T_{AB} = 11.0 \text{ lb}$ ,  $T_{AC} = 7.76 \text{ lb}$ ,  $T_{BC} = 11.0 \text{ lb}$ ,  
 $T_{BE} = 19.0 \text{ lb}$ ,  $T_{CD} = 17.4 \text{ lb}$ ,  $\theta = 18.4^\circ$
- 3-34.  $\theta = 18.4^\circ$ ,  $W = 15.8 \text{ lb}$
- 3-35.  $F_{AB} = 175 \text{ lb}$ ,  $l = 2.34 \text{ ft}$ , or  
 $F_{AB} = 82.4 \text{ lb}$ ,  $l = 1.40 \text{ ft}$
- 3-37.  $m_B = 3.58 \text{ kg}$ ,  $N = 19.7 \text{ N}$
- 3-38.  $F_{AB} = 98.6 \text{ N}$ ,  $F_{AC} = 267 \text{ N}$
- 3-39.  $d = 2.42 \text{ m}$
- 3-41.  $T = 30.6 \text{ lb}$ ,  $x = 1.92 \text{ ft}$
- 3-42.  $W_B = 18.3 \text{ lb}$
- 3-43.  $F_{AD} = 763 \text{ N}$ ,  $F_{AC} = 392 \text{ N}$ ,  $F_{AB} = 523 \text{ N}$
- 3-45.  $F_{DA} = 10.0 \text{ lb}$ ,  $F_{DB} = 1.11 \text{ lb}$ ,  $F_{DC} = 15.6 \text{ lb}$
- 3-46.  $s_{OB} = 327 \text{ mm}$ ,  $s_{OA} = 218 \text{ mm}$
- 3-47.  $F_{AB} = 219 \text{ N}$ ,  $F_{AC} = F_{AD} = 54.8 \text{ N}$
- 3-49.  $m = 102 \text{ kg}$
- 3-50.  $F_{AC} = 113 \text{ lb}$ ,  $F_{AB} = 257 \text{ lb}$ ,  $F_{AD} = 210 \text{ lb}$
- 3-51.  $F = 1558 \text{ lb}$
- 3-53.  $F_{AD} = 557 \text{ lb}$ ,  $W = 407 \text{ lb}$
- 3-54.  $F_{AB} = 79.2 \text{ lb}$ ,  $F_{AC} = 119 \text{ lb}$ ,  $F_{AD} = 283 \text{ lb}$
- 3-55.  $W_C = 265 \text{ lb}$
- 3-57.  $W = 55.8 \text{ N}$
- 3-58.  $F_{AB} = 441 \text{ N}$ ,  $F_{AC} = 515 \text{ N}$ ,  $F_{AD} = 221 \text{ N}$
- 3-59.  $F_{AB} = 348 \text{ N}$ ,  $F_{AC} = 413 \text{ N}$ ,  $F_{AD} = 174 \text{ N}$
- 3-61.  $F_{AC} = 85.8 \text{ N}$ ,  $F_{AB} = 578 \text{ N}$ ,  $F_{AD} = 565 \text{ N}$
- 3-62.  $m = 88.5 \text{ kg}$
- 3-63.  $F_{AD} = 1.56 \text{ kN}$ ,  $F_{BD} = 521 \text{ N}$ ,  $F_{CD} = 1.28 \text{ kN}$
- 3-65.  $F_{AE} = 2.91 \text{ kip}$ ,  $F = 1.61 \text{ kip}$
- 3-66.  $F_{AB} = 360 \text{ lb}$ ,  $F_{AC} = 180 \text{ lb}$ ,  $F_{AD} = 360 \text{ lb}$
- 3-67.  $W = 375 \text{ lb}$

## Chapter 4

- 4-5.  $(M_{F_1})_B = 4.125 \text{ kip} \cdot \text{ft} \curvearrowright$ ,  
 $(M_{F_2})_B = 2.00 \text{ kip} \cdot \text{ft} \curvearrowright$ ,  
 $(M_{F_3})_B = 40.0 \text{ lb} \cdot \text{ft} \curvearrowright$
- 4-6.  $M_P = 341 \text{ in} \cdot \text{lb} \curvearrowright$   
 $M_F = 403 \text{ in} \cdot \text{lb} \curvearrowright$   
 Not sufficient
- 4-7.  $(M_{F_1})_A = 433 \text{ N} \cdot \text{m} \curvearrowright$   
 $(M_{F_2})_A = 1.30 \text{ kN} \cdot \text{m} \curvearrowright$   
 $(M_{F_3})_A = 800 \text{ N} \cdot \text{m} \curvearrowright$
- 4-9.  $M_B = 90.6 \text{ lb} \cdot \text{ft} \curvearrowright$ ,  $M_C = 141 \text{ lb} \cdot \text{ft} \curvearrowright$
- 4-10.  $M_A = 195 \text{ lb} \cdot \text{ft} \curvearrowright$
- 4-11.  $(M_O)_{\max} = 48.0 \text{ kN} \cdot \text{m} \curvearrowright$ ,  $x = 9.81 \text{ m}$
- 4-13.  $\mathbf{M}_B = \{-3.36\mathbf{k}\} \text{ N} \cdot \text{m}$ ,  $\alpha = 90^\circ$ ,  $\beta = 90^\circ$ ,  
 $\gamma = 180^\circ$
- 4-14.  $\mathbf{M}_O = \{0.5\mathbf{i} + 0.866\mathbf{j} - 3.36\mathbf{k}\} \text{ N} \cdot \text{m}$ ,  
 $\alpha = 81.8^\circ$ ,  $\beta = 75.7^\circ$ ,  $\gamma = 163^\circ$
- 4-15.  $(M_A)_C = 768 \text{ lb} \cdot \text{ft} \curvearrowright$   
 $(M_A)_B = 636 \text{ lb} \cdot \text{ft} \curvearrowright$   
 Clockwise
- 4-17.  $m = \left(\frac{l}{d+l}\right)M$
- 4-18.  $M_P = (537.5 \cos \theta + 75 \sin \theta) \text{ lb} \cdot \text{ft}$
- 4-19.  $F = 239 \text{ lb}$
- 4-21.  $F = 27.6 \text{ lb}$
- 4-22.  $r = 13.3 \text{ mm}$
- 4-23.  $(M_R)_A = (M_R)_B = 76.0 \text{ kN} \cdot \text{m} \curvearrowright$
- 4-25.  $(M_{AB})_A = 3.88 \text{ kip} \cdot \text{ft} \curvearrowright$ ,  
 $(M_{BCD})_A = 2.05 \text{ kip} \cdot \text{ft} \curvearrowright$ ,  
 $(M_{\text{man}})_A = 2.10 \text{ kip} \cdot \text{ft} \curvearrowright$
- 4-26.  $(M_R)_A = 8.04 \text{ kip} \cdot \text{ft} \curvearrowright$
- 4-27.  $\mathbf{M}_O = \{-40\mathbf{i} - 44\mathbf{j} - 8\mathbf{k}\} \text{ kN} \cdot \text{m}$
- 4-29.  $\mathbf{M}_O = \{-25\mathbf{i} + 6200\mathbf{j} - 900\mathbf{k}\} \text{ lb} \cdot \text{ft}$
- 4-30.  $\mathbf{M}_A = \{-175\mathbf{i} + 5600\mathbf{j} - 900\mathbf{k}\} \text{ lb} \cdot \text{ft}$
- 4-31.  $\mathbf{M}_P = \{-24\mathbf{i} + 24\mathbf{j} + 8\mathbf{k}\} \text{ kN} \cdot \text{m}$
- 4-33.  $\mathbf{M}_B = \{-110\mathbf{i} - 180\mathbf{j} - 420\mathbf{k}\} \text{ N} \cdot \text{m}$
- 4-34.  $\mathbf{M}_A = \{574\mathbf{i} + 350\mathbf{j} + 1385\mathbf{k}\} \text{ N} \cdot \text{m}$
- 4-35.  $F = 585 \text{ N}$
- 4-37.  $\mathbf{M}_O = \{163\mathbf{i} - 346\mathbf{j} - 360\mathbf{k}\} \text{ N} \cdot \text{m}$
- 4-38.  $\mathbf{M}_A = \{-82.9\mathbf{i} + 41.5\mathbf{j} + 232\mathbf{k}\} \text{ lb} \cdot \text{ft}$
- 4-39.  $\mathbf{M}_B = \{-82.9\mathbf{i} - 96.8\mathbf{j} - 52.8\mathbf{k}\} \text{ lb} \cdot \text{ft}$
- 4-41.  $F = 18.6 \text{ lb}$
- 4-42.  $M_O = 4.27 \text{ N} \cdot \text{m}$ ,  $\alpha = 95.2^\circ$ ,  $\beta = 110^\circ$ ,  $\gamma = 20.6^\circ$
- 4-43.  $\mathbf{M}_A = \{-5.39\mathbf{i} + 13.1\mathbf{j} + 11.4\mathbf{k}\} \text{ N} \cdot \text{m}$
- 4-45.  $y = 2 \text{ m}$ ,  $z = 1 \text{ m}$
- 4-46.  $y = 1 \text{ m}$ ,  $z = 3 \text{ m}$ ,  $d = 1.15 \text{ m}$
- 4-47.  $\mathbf{M}_A = \{-16.0\mathbf{i} - 32.1\mathbf{k}\} \text{ N} \cdot \text{m}$
- 4-49.  $\mathbf{M}_B = \{1.00\mathbf{i} + 0.750\mathbf{j} - 1.56\mathbf{k}\} \text{ kN} \cdot \text{m}$
- 4-50.  $\mathbf{M}_O = \{373\mathbf{i} - 99.9\mathbf{j} + 173\mathbf{k}\} \text{ N} \cdot \text{m}$
- 4-51.  $\theta_{\max} = 90^\circ$ ,  $\theta_{\min} = 0, 180^\circ$
- 4-53. Yes, yes
- 4-54.  $M_{y'} = 464 \text{ lb} \cdot \text{ft}$
- 4-55.  $M_x = 440 \text{ lb} \cdot \text{ft}$
- 4-57.  $\mathbf{M}_{AC} = \{11.5\mathbf{i} + 8.64\mathbf{j}\} \text{ lb} \cdot \text{ft}$
- 4-58.  $M_x = 21.7 \text{ N} \cdot \text{m}$
- 4-59.  $F = 139 \text{ N}$
- 4-61.  $M_{AB} = 136 \text{ N} \cdot \text{m}$
- 4-62.  $M_{BC} = 165 \text{ N} \cdot \text{m}$
- 4-63.  $M_{CA} = 226 \text{ N} \cdot \text{m}$
- 4-65.  $F = 5.66 \text{ N}$
- 4-66.  $M_a = 4.37 \text{ N} \cdot \text{m}$ ,  $\alpha = 33.7^\circ$ ,  $\beta = 90^\circ$ ,  $\gamma = 56.3^\circ$ ,  
 $M = 5.41 \text{ N} \cdot \text{m}$
- 4-67.  $R = 28.9 \text{ N}$
- 4-69.  $F = 75 \text{ N}$ ,  $P = 100 \text{ N}$
- 4-70.  $(M_R)_C = 435 \text{ lb} \cdot \text{ft} \curvearrowright$
- 4-71.  $F = 139 \text{ lb}$
- 4-73.  $F = 830 \text{ N}$

- 4-74.  $M_C = 22.5 \text{ N} \cdot \text{m} \curvearrowright$   
 4-75.  $F = 83.3 \text{ N}$   
 4-77.  $(M_R)_C = 240 \text{ lb} \cdot \text{ft} \curvearrowright$   
 4-78.  $F = 167 \text{ lb}$ . Resultant couple can act anywhere.  
 4-79.  $d = 2.03 \text{ ft}$   
 4-81.  $M_C = 126 \text{ lb} \cdot \text{ft} \curvearrowright$   
 4-82.  $\mathbf{M}_C = \{-50\mathbf{i} + 60\mathbf{j}\} \text{ lb} \cdot \text{ft}$   
 4-83.  $M_R = 96.0 \text{ lb} \cdot \text{ft}$ ,  $\alpha = 47.4^\circ$ ,  $\beta = 74.9^\circ$ ,  $\gamma = 133^\circ$   
 4-85.  $M_R = 64.0 \text{ lb} \cdot \text{ft}$ ,  $\alpha = 94.7^\circ$ ,  $\beta = 13.2^\circ$ ,  $\gamma = 102^\circ$   
 4-86.  $M_2 = 424 \text{ N} \cdot \text{m}$ ,  $M_3 = 300 \text{ N} \cdot \text{m}$   
 4-87.  $M_R = 576 \text{ lb} \cdot \text{in.}$ ,  $\alpha = 37.0^\circ$ ,  $\beta = 111^\circ$ ,  $\gamma = 61.2^\circ$   
 4-89.  $F = 15.4 \text{ N}$   
 4-90.  $M_C = 45.1 \text{ N} \cdot \text{m}$   
 4-91.  $F = 832 \text{ N}$   
 4-93.  $F = 98.1 \text{ N}$   
 4-94.  $\mathbf{M}_C = \{-2\mathbf{i} + 20\mathbf{j} + 17\mathbf{k}\} \text{ kN} \cdot \text{m}$ ,  
 $M_C = 26.3 \text{ kN} \cdot \text{m}$   
 4-95.  $(M_C)_R = 71.9 \text{ N} \cdot \text{m}$ ,  $\alpha = 44.2^\circ$ ,  $\beta = 131^\circ$ ,  $\gamma = 103^\circ$   
 4-97.  $F_R = 365 \text{ N}$ ,  $\theta = 70.8^\circ \curvearrowright$ ,  $(M_R)_O = 2364 \text{ N} \cdot \text{m} \curvearrowright$   
 4-98.  $F_R = 365 \text{ N}$ ,  $\theta = 70.8^\circ \curvearrowright$ ,  $(M_R)_P = 2799 \text{ N} \cdot \text{m} \curvearrowright$   
 4-99.  $F_R = 5.93 \text{ kN}$ ,  $\theta = 77.8^\circ \curvearrowright$ ,  $M_{R_A} = 34.8 \text{ kN} \cdot \text{m} \curvearrowright$   
 4-101.  $F_R = 294 \text{ N}$ ,  $\theta = 40.1^\circ \curvearrowright$ ,  
 $M_{RO} = 39.6 \text{ N} \cdot \text{m} \curvearrowright$   
 4-102.  $F_R = 1.30 \text{ kN}$ ,  $\theta = 86.7^\circ \curvearrowleft$ ,  
 $(M_R)_A = 1.02 \text{ kN} \cdot \text{m} \curvearrowright$   
 4-103.  $F_R = 1.30 \text{ kN}$ ,  $\theta = 86.7^\circ \curvearrowleft$ ,  
 $(M_R)_B = 10.1 \text{ kN} \cdot \text{m} \curvearrowright$   
 4-105.  $F_R = 938 \text{ N}$ ,  $\theta = 35.9^\circ \curvearrowleft$ ,  $(M_R)_A = 680 \text{ N} \cdot \text{m} \curvearrowright$   
 4-106.  $\mathbf{M}_{RO} = \{0.650\mathbf{i} + 19.75\mathbf{j} - 9.05\mathbf{k}\} \text{ kN} \cdot \text{m}$   
 4-107.  $\mathbf{F}_R = \{270\mathbf{k}\} \text{ N}$ ,  $\mathbf{M}_{RO} = \{-2.22\mathbf{i}\} \text{ N} \cdot \text{m}$   
 4-109.  $\mathbf{F}_R = \{6\mathbf{i} + 5\mathbf{j} - 5\mathbf{k}\} \text{ kN}$ ,  
 $(\mathbf{M}_R)_O = \{2.5\mathbf{i} - 7\mathbf{j}\} \text{ kN} \cdot \text{m}$   
 4-110.  $\mathbf{F}_R = \{44.5\mathbf{i} + 53.1\mathbf{j} - 40.0\mathbf{k}\} \text{ N}$ ,  
 $\mathbf{M}_{RA} = \{-5.39\mathbf{i} + 13.1\mathbf{j} + 11.4\mathbf{k}\} \text{ N} \cdot \text{m}$   
 4-111.  $\mathbf{F}_R = \{-40\mathbf{j} - 40\mathbf{k}\} \text{ N}$ ,  
 $\mathbf{M}_{RA} = \{-12\mathbf{j} + 12\mathbf{k}\} \text{ N} \cdot \text{m}$   
 4-113.  $F_R = 10.75 \text{ kip} \downarrow$ ,  $d = 13.7 \text{ ft}$   
 4-114.  $F_R = 10.75 \text{ kip} \downarrow$ ,  $d = 9.26 \text{ ft}$   
 4-115.  $F = 798 \text{ lb}$ ,  $67.9^\circ \curvearrowright$ ,  $x = 7.43 \text{ ft}$   
 4-117.  $F = 1302 \text{ N}$ ,  $\theta = 84.5^\circ \curvearrowright$ ,  $x = 7.36 \text{ m}$   
 4-118.  $F = 1302 \text{ N}$ ,  $\theta = 84.5^\circ \curvearrowright$ ,  
 $x = 1.36 \text{ m}$  (to the right)  
 4-119.  $F_R = 1000 \text{ N}$ ,  $\theta = 53.1^\circ \curvearrowleft$ ,  $d = 2.17 \text{ m}$   
 4-121.  $F_R = 356 \text{ N}$ ,  $\theta = 51.8^\circ$ ,  $d = 1.75 \text{ m}$   
 4-122.  $F_R = 542 \text{ N}$ ,  $\theta = 10.6^\circ \curvearrowleft$ ,  $d = 0.827 \text{ m}$   
 4-123.  $F_R = 542 \text{ N}$ ,  $\theta = 10.6^\circ \curvearrowright$ ,  $d = 2.17 \text{ m}$   
 4-125.  $F_R = 197 \text{ lb}$ ,  $\theta = 42.6^\circ \curvearrowright$ ,  $d = 5.24 \text{ ft}$   
 4-126.  $F_R = 197 \text{ lb}$ ,  $\theta = 42.6^\circ \curvearrowright$ ,  $d = 0.824 \text{ ft}$   
 4-127.  $F_R = 26 \text{ kN}$ ,  $y = 82.7 \text{ mm}$ ,  $x = 3.85 \text{ mm}$   
 4-129.  $F_C = 600 \text{ N}$ ,  $F_D = 500 \text{ N}$   
 4-130.  $F_R = 35 \text{ kN}$ ,  $y = 11.3 \text{ m}$ ,  $x = 11.5 \text{ m}$   
 4-131.  $F_1 = 27.6 \text{ kN}$ ,  $F_2 = 24.0 \text{ kN}$

- 4-133.  $F_A = 30 \text{ kN}$ ,  $F_B = 20 \text{ kN}$ ,  $F_R = 190 \text{ kN}$   
 4-134.  $\mathbf{F}_R = \{141\mathbf{i} + 100\mathbf{j} + 159\mathbf{k}\} \text{ N}$ ,  
 $\mathbf{M}_{RO} = \{122\mathbf{i} - 183\mathbf{k}\} \text{ N} \cdot \text{m}$   
 4-135.  $F_R = 379 \text{ N}$ ,  $M_R = 590 \text{ N} \cdot \text{m}$ ,  $z = 2.68 \text{ m}$ ,  
 $x = -2.76 \text{ m}$   
 4-137.  $F_R = 539 \text{ N}$ ,  $M_R = 1.45 \text{ kN} \cdot \text{m}$ ,  $x = 1.21 \text{ m}$ ,  
 $y = 3.59 \text{ m}$   
 4-138.  $F_R = 0$ ,  $M_{RO} = 1.35 \text{ kip} \cdot \text{ft}$   
 4-139.  $F_R = 6.75 \text{ kN}$ ,  $\bar{x} = 2.5 \text{ m}$   
 4-141.  $F_R = 7 \text{ lb}$ ,  $\bar{x} = 0.268 \text{ ft}$   
 4-142.  $F_R = 15.0 \text{ kN}$ ,  $d = 3.40 \text{ m}$   
 4-143.  $F_R = 12.5 \text{ kN}$ ,  $d = 1.54 \text{ m}$   
 4-145.  $F_R = 15.4 \text{ kN}$ ,  $(M_R)_O = 18.5 \text{ kN} \cdot \text{m} \curvearrowright$   
 4-146.  $F_R = 27.0 \text{ kN}$ ,  $(M_R)_A = 81.0 \text{ kN} \cdot \text{m} \curvearrowright$   
 4-147.  $a = 1.54 \text{ m}$   
 4-149.  $w_2 = 17.2 \text{ kN/m}$ ,  $w_1 = 30.3 \text{ kN/m}$   
 4-150.  $F_R = 51.0 \text{ kN} \downarrow$ ,  $M_{R_O} = 914 \text{ kN} \cdot \text{m} \curvearrowright$   
 4-151.  $F_R = 51.0 \text{ kN} \downarrow$ ,  $d = 17.9 \text{ m}$   
 4-153.  $F_R = 1.80 \text{ kN}$ ,  $d = 2.33 \text{ m}$   
 4-154.  $F_R = 12.0 \text{ kN}$ ,  $\theta = 48.4^\circ \curvearrowright$ ,  $d = 3.28 \text{ m}$   
 4-155.  $F_R = 12.0 \text{ kN}$ ,  $\theta = 48.4^\circ \curvearrowright$ ,  $d = 3.69 \text{ m}$   
 4-157.  $F_R = 6.75 \text{ kN}$ ,  $(M_R)_O = 4.05 \text{ kN} \cdot \text{m} \curvearrowright$   
 4-158.  $F_R = 43.6 \text{ lb}$ ,  $x = 3.27 \text{ ft}$   
 4-159.  $d = 2.22 \text{ ft}$

$$4-161. F_R = \frac{2Lw_0}{\pi}, (M_R)_O = \left(\frac{2\pi - 4}{\pi^2}\right)w_0L^2 \curvearrowright$$

$$4-162. F_R = 107 \text{ kN}, h = 1.60 \text{ m}$$

## Chapter 5

- 5-10.  $A_x = 3.46 \text{ kN}$ ,  $A_y = 8 \text{ kN}$ ,  $M_A = 20.2 \text{ kN} \cdot \text{m}$   
 5-11.  $N_A = 750 \text{ N}$ ,  $B_y = 600 \text{ N}$ ,  $B_x = 450 \text{ N}$   
 5-13.  $N_A = 2.175 \text{ kN}$ ,  $B_y = 1.875 \text{ kN}$ ,  $B_x = 0$   
 5-14.  $N_A = 3.33 \text{ kN}$ ,  $B_x = 2.40 \text{ kN}$ ,  $B_y = 133 \text{ N}$   
 5-15.  $A_y = 5.00 \text{ kN}$ ,  $N_B = 9.00 \text{ kN}$ ,  $A_x = 5.00 \text{ kN}$   
 5-17.  $\theta = 41.4^\circ$   
 5-18.  $A_x = 0$ ,  $B_y = P$ ,  $M_A = \frac{PL}{2}$   
 5-19.  $T = \frac{W}{2} \sin \theta$   
 5-21.  $T_{BC} = 113 \text{ N}$   
 5-22.  $N_A = 3.71 \text{ kN}$ ,  $B_x = 1.86 \text{ kN}$ ,  $B_y = 8.78 \text{ kN}$   
 5-23.  $w = 2.67 \text{ kN/m}$   
 5-25.  $N_A = 39.7 \text{ lb}$ ,  $N_B = 82.5 \text{ lb}$ ,  $M_A = 106 \text{ lb} \cdot \text{ft}$   
 5-26.  $\theta = 70.3^\circ$ ,  $N'_A = (29.4 - 31.3 \sin \theta) \text{ kN}$ ,  
 $N'_B = (73.6 + 31.3 \sin \theta) \text{ kN}$   
 5-27.  $N_B = 98.1 \text{ N}$ ,  $A_x = 85.0 \text{ N}$ ,  $A_y = 147 \text{ N}$   
 5-29.  $P = 272 \text{ N}$   
 5-30.  $P_{\min} = 271 \text{ N}$   
 5-31.  $F_B = 86.6 \text{ N}$ ,  $B_x = 43.3 \text{ N}$ ,  $B_y = 110 \text{ N}$   
 5-33.  $A_x = 25.4 \text{ kN}$ ,  $B_y = 22.8 \text{ kN}$ ,  $B_x = 25.4 \text{ kN}$   
 5-34.  $F = 14.0 \text{ kN}$



- 5-35.  $N_A = 173 \text{ N}$ ,  $N_C = 416 \text{ N}$ ,  $N_B = 69.2 \text{ N}$   
 5-37.  $N_A = 975 \text{ lb}$ ,  $B_x = 975 \text{ lb}$ ,  $B_y = 780 \text{ lb}$   
 5-38.  $A_x = 1.46 \text{ kip}$ ,  $F_B = 1.66 \text{ kip}$   
 5-39.  $\theta = 17.5^\circ$   
 5-41.  $F = 311 \text{ kN}$ ,  $A_x = 460 \text{ kN}$ ,  $A_y = 7.85 \text{ kN}$   
 5-42.  $F_{CB} = 782 \text{ N}$ ,  $A_x = 625 \text{ N}$ ,  $A_y = 681 \text{ N}$   
 5-43.  $F_2 = 724 \text{ N}$ ,  $F_1 = 1.45 \text{ kN}$ ,  $F_A = 1.75 \text{ kN}$   
 5-45.  $P = 660 \text{ N}$ ,  $N_A = 442 \text{ N}$ ,  $\theta = 48.0^\circ$   $\swarrow$   
 5-46.  $d = \frac{3a}{4}$   
 5-47.  $F_{BC} = 80 \text{ kN}$ ,  $A_x = 54 \text{ kN}$ ,  $A_y = 16 \text{ kN}$   
 5-49.  $F_C = 10 \text{ mN}$   
 5-50.  $k = 250 \text{ N/m}$   
 5-51.  $w_B = 2.19 \text{ kip/ft}$ ,  $w_A = 10.7 \text{ kip/ft}$   
 5-53.  $\alpha = 10.4^\circ$   
 5-54.  $h = 0.645 \text{ m}$   
 5-55.  $h = \sqrt{\frac{s^2 - l^2}{3}}$   
 5-57.  $w_1 = 83.3 \text{ lb/ft}$ ,  $w_2 = 167 \text{ lb/ft}$   
 5-58.  $w_1 = \frac{2P}{L}$ ,  $w_2 = \frac{4P}{L}$   
 5-59.  $\theta = 23.2^\circ$ ,  $85.2^\circ$   
 5-61.  $N_A = 346 \text{ N}$ ,  $N_B = 693 \text{ N}$ ,  $a = 0.650 \text{ m}$   
 5-62.  $T = 1.84 \text{ kN}$ ,  $F = 6.18 \text{ kN}$   
 5-63.  $R_D = 22.6 \text{ kip}$ ,  $R_E = 22.6 \text{ kip}$ ,  $R_F = 13.7 \text{ kip}$   
 5-65.  $N_A = 28.6 \text{ lb}$ ,  $N_B = 10.7 \text{ lb}$ ,  $N_C = 10.7 \text{ lb}$   
 5-66.  $T_{BC} = 43.9 \text{ N}$ ,  $N_B = 58.9 \text{ N}$ ,  $A_x = 58.9 \text{ N}$ ,  
 $A_y = 39.2 \text{ N}$ ,  $A_z = 177 \text{ N}$   
 5-67.  $T_C = 14.8 \text{ kN}$ ,  $T_B = 16.5 \text{ kN}$ ,  $T_A = 7.27 \text{ kN}$   
 5-69.  $F_{AB} = 467 \text{ N}$ ,  $F_{AC} = 674 \text{ N}$ ,  $D_x = 1.04 \text{ kN}$ ,  
 $D_y = 0$ ,  $D_z = 0$   
 5-70.  $T_{BA} = 2.00 \text{ kN}$ ,  $T_{BC} = 1.35 \text{ kN}$ ,  $D_x = 0.327 \text{ kN}$ ,  
 $D_y = 1.31 \text{ kN}$ ,  $D_z = 4.58 \text{ kN}$   
 5-71.  $F_{BD} = F_{BC} = 350 \text{ N}$ ,  $A_x = 600 \text{ N}$ ,  
 $A_y = 0$ ,  $A_z = 300 \text{ N}$   
 5-73.  $C_y = 800 \text{ N}$ ,  $B_z = 107 \text{ N}$ ,  $B_y = 600 \text{ N}$ ,  
 $C_x = 53.6 \text{ N}$ ,  $A_x = 400 \text{ N}$ ,  $A_z = 800 \text{ N}$   
 5-74.  $F = 746 \text{ N}$   
 5-75.  $T_{BC} = 1.40 \text{ kN}$ ,  $A_y = 800 \text{ N}$ ,  $A_x = 1.20 \text{ kN}$ ,  
 $(M_A)_x = 600 \text{ N} \cdot \text{m}$ ,  $(M_A)_y = 1.20 \text{ kN} \cdot \text{m}$ ,  
 $(M_A)_z = 2.40 \text{ kN} \cdot \text{m}$   
 5-77.  $A_x = 300 \text{ N}$ ,  $A_y = 500 \text{ N}$ ,  $N_B = 400 \text{ N}$ ,  
 $(M_A)_x = 1.00 \text{ kN} \cdot \text{m}$ ,  $(M_A)_y = 200 \text{ N} \cdot \text{m}$ ,  
 $(M_A)_z = 1.50 \text{ kN} \cdot \text{m}$   
 5-78.  $A_x = 633 \text{ lb}$ ,  $A_y = -141 \text{ lb}$ ,  $B_x = -721 \text{ lb}$ ,  
 $B_z = 895 \text{ lb}$ ,  $C_y = 200 \text{ lb}$ ,  $C_z = -506 \text{ lb}$   
 5-79.  $F_2 = 674 \text{ lb}$   
 5-81.  $C_z = 10.6 \text{ lb}$ ,  $D_y = -0.230 \text{ lb}$ ,  
 $C_y = 0.230 \text{ lb}$ ,  $D_x = 5.17 \text{ lb}$ ,  
 $C_x = 5.44 \text{ lb}$ ,  $M = 0.459 \text{ lb} \cdot \text{ft}$   
 5-82.  $F_{BD} = 294 \text{ N}$ ,  $F_{BC} = 589 \text{ N}$ ,  $A_x = 0$ ,  
 $A_y = 589 \text{ N}$ ,  $A_z = 490.5 \text{ N}$

- 5-83.  $T = 58.0 \text{ N}$ ,  $C_z = 87.0 \text{ N}$ ,  $C_y = 28.8 \text{ N}$ ,  $D_x = 0$ ,  
 $D_y = 79.2 \text{ N}$ ,  $D_z = 58.0 \text{ N}$   
 5-85.  $F_{BC} = 0$ ,  $A_y = 0$ ,  $A_z = 800 \text{ lb}$ ,  
 $(M_A)_x = 4.80 \text{ kip} \cdot \text{ft}$ ,  $(M_A)_y = 0$ ,  $(M_A)_z = 0$

## Chapter 6

- 6-1.  $F_{CB} = 0$ ,  $F_{CD} = 20.0 \text{ kN (C)}$ ,  
 $F_{DB} = 33.3 \text{ kN (T)}$ ,  $F_{DA} = 36.7 \text{ kN (C)}$   
 6-2.  $F_{CB} = 0$ ,  $F_{CD} = 45.0 \text{ kN (C)}$ ,  
 $F_{DB} = 75.0 \text{ kN (T)}$ ,  $F_{DA} = 90.0 \text{ kN (C)}$   
 6-3.  $F_{AC} = 150 \text{ lb (C)}$ ,  $F_{AB} = 140 \text{ lb (T)}$ ,  
 $F_{BD} = 140 \text{ lb (T)}$ ,  $F_{BC} = 0$ ,  $F_{CD} = 150 \text{ lb (T)}$ ,  
 $F_{CE} = 180 \text{ lb (C)}$ ,  $F_{DE} = 120 \text{ lb (C)}$ ,  
 $F_{DF} = 230 \text{ lb (T)}$ ,  $F_{EF} = 300 \text{ lb (C)}$   
 6-5.  $F_{CD} = 5.21 \text{ kN (C)}$ ,  $F_{CB} = 4.17 \text{ kN (T)}$ ,  
 $F_{AD} = 1.46 \text{ kN (C)}$ ,  $F_{AB} = 4.17 \text{ kN (T)}$ ,  
 $F_{BD} = 4 \text{ kN (T)}$   
 6-6.  $F_{CD} = 5.21 \text{ kN (C)}$ ,  $F_{CB} = 2.36 \text{ kN (T)}$ ,  
 $F_{AD} = 1.46 \text{ kN (C)}$ ,  $F_{AB} = 2.36 \text{ kN (T)}$ ,  
 $F_{BD} = 4 \text{ kN (T)}$   
 6-7.  $F_{DE} = 16.3 \text{ kN (C)}$ ,  $F_{DC} = 8.40 \text{ kN (T)}$ ,  
 $F_{EA} = 8.85 \text{ kN (C)}$ ,  $F_{EC} = 6.20 \text{ kN (C)}$ ,  
 $F_{CF} = 8.77 \text{ kN (T)}$ ,  $F_{CB} = 2.20 \text{ kN (T)}$ ,  
 $F_{BA} = 3.11 \text{ kN (T)}$ ,  $F_{BF} = 6.20 \text{ kN (C)}$ ,  
 $F_{FA} = 6.20 \text{ kN (T)}$   
 6-9.  $F_{AE} = 5.66 \text{ kN (C)}$ ,  $F_{AB} = 4.00 \text{ kN (T)}$ ,  
 $F_{DE} = 7.07 \text{ kN (C)}$ ,  $F_{DC} = 5.00 \text{ kN (T)}$ ,  
 $F_{BE} = 3.16 \text{ kN (T)}$ ,  $F_{BC} = 3.00 \text{ kN (T)}$ ,  
 $F_{CE} = 6.32 \text{ kN (T)}$   
 6-10.  $F_{AE} = 9.90 \text{ kN (C)}$ ,  $F_{AB} = 7.00 \text{ kN (T)}$ ,  
 $F_{DE} = 11.3 \text{ kN (C)}$ ,  $F_{DC} = 8.00 \text{ kN (T)}$ ,  
 $F_{BE} = 6.32 \text{ kN (T)}$ ,  $F_{BC} = 5.00 \text{ kN (T)}$ ,  
 $F_{CE} = 9.49 \text{ kN (T)}$   
 6-11.  $F_{JD} = 33.3 \text{ kN (T)}$ ,  
 $F_{AL} = F_{GH} = F_{LK} = F_{HI} = 28.3 \text{ kN (C)}$ ,  
 $F_{AB} = F_{GF} = F_{BC} = F_{FE} = F_{CD} = F_{ED} =$   
 $20 \text{ kN (T)}$ ,  
 $F_{BL} = F_{FH} = F_{LC} = F_{HE} = 0$ ,  
 $F_{CK} = F_{EI} = 10 \text{ kN (T)}$ ,  $F_{KJ} = F_{IJ} = 23.6 \text{ kN (C)}$ ,  
 $F_{KD} = F_{ID} = 7.45 \text{ kN (C)}$   
 6-13.  $F_{CD} = F_{AD} = 0.687P \text{ (T)}$ ,  
 $F_{CB} = F_{AB} = 0.943P \text{ (C)}$ ,  
 $F_{DB} = 1.33P \text{ (T)}$   
 6-14.  $P_{\max} = 849 \text{ lb}$   
 6-15.  $P_{\max} = 849 \text{ lb}$   
 6-17.  $P = 5.20 \text{ kN}$   
 6-18.  $F_{DE} = 8.94 \text{ kN (T)}$ ,  $F_{DC} = 4.00 \text{ kN (C)}$ ,  
 $F_{CB} = 4.00 \text{ kN (C)}$ ,  $F_{CE} = 0$ ,  
 $F_{EB} = 11.3 \text{ kN (C)}$ ,  $F_{EF} = 12.0 \text{ kN (T)}$ ,  
 $F_{BA} = 12.0 \text{ kN (C)}$ ,  $F_{BF} = 18.0 \text{ kN (T)}$ ,  
 $F_{FA} = 20.1 \text{ kN (C)}$ ,  $F_{FG} = 21.0 \text{ kN (T)}$

- 6-19.  $F_{DE} = 13.4 \text{ kN (T)}$ ,  $F_{DC} = 6.00 \text{ kN (C)}$ ,  
 $F_{CB} = 6.00 \text{ kN (C)}$ ,  $F_{CE} = 0$ ,  $F_{EB} = 17.0 \text{ kN (C)}$ ,  
 $F_{EF} = 18.0 \text{ kN (T)}$ ,  $F_{BA} = 18.0 \text{ kN (C)}$ ,  
 $F_{BF} = 20.0 \text{ kN (T)}$ ,  $F_{FA} = 22.4 \text{ kN (C)}$ ,  
 $F_{FG} = 28.0 \text{ kN (T)}$
- 6-21.  $F_{DE} = F_{DC} = F_{FA} = 0$ ,  $F_{CE} = 34.4 \text{ kN (C)}$ ,  
 $F_{CB} = 20.6 \text{ kN (T)}$ ,  $F_{BA} = 20.6 \text{ kN (T)}$ ,  
 $F_{BE} = 15.0 \text{ kN (T)}$ ,  $F_{FE} = 30.0 \text{ kN (C)}$ ,  
 $F_{EA} = 15.6 \text{ kN (T)}$
- 6-22.  $F_{FE} = 0.667P \text{ (T)}$ ,  $F_{FD} = 1.67P \text{ (T)}$ ,  
 $F_{AB} = 0.471P \text{ (C)}$ ,  $F_{AE} = 1.67P \text{ (T)}$ ,  
 $F_{AC} = 1.49P \text{ (C)}$ ,  $F_{BF} = 1.41P \text{ (T)}$ ,  
 $F_{BD} = 1.49P \text{ (C)}$ ,  $F_{EC} = 1.41P \text{ (T)}$ ,  
 $F_{CD} = 0.471P \text{ (C)}$
- 6-23.  $F_{EC} = 1.20P \text{ (T)}$ ,  $F_{ED} = 0$ ,  
 $F_{AB} = F_{AD} = 0.373P \text{ (C)}$ ,  $F_{DC} = 0.373P \text{ (C)}$ ,  
 $F_{DB} = 0.333P \text{ (T)}$ ,  $F_{BC} = 0.373P \text{ (C)}$
- 6-25.  $F_{CB} = 2.31 \text{ kN (C)}$ ,  $F_{CD} = 1.15 \text{ kN (C)}$ ,  
 $F_{DB} = 4.00 \text{ kN (T)}$ ,  $F_{DA} = 4.62 \text{ kN (C)}$ ,  
 $F_{AB} = 2.31 \text{ kN (C)}$
- 6-26.  $P_{\max} = 1.30 \text{ kN}$
- 6-27.  $F_{HI} = 42.5 \text{ kN (T)}$ ,  $F_{HC} = 100 \text{ kN (T)}$ ,  
 $F_{DC} = 125 \text{ kN (C)}$
- 6-29.  $F_{HG} = 1125 \text{ lb (T)}$ ,  $F_{DE} = 3375 \text{ lb (C)}$ ,  
 $F_{EH} = 3750 \text{ lb (T)}$
- 6-30.  $F_{CD} = 3375 \text{ lb (C)}$ ,  $F_{HI} = 6750 \text{ lb (T)}$ ,  
 $F_{CH} = 5625 \text{ lb (C)}$
- 6-31.  $F_{KJ} = 11.25 \text{ kip (T)}$ ,  $F_{CD} = 9.375 \text{ kip (C)}$ ,  
 $F_{CJ} = 3.125 \text{ kip (C)}$ ,  $F_{DJ} = 0$
- 6-33.  $F_{GF} = 12.5 \text{ kN (C)}$ ,  $F_{CD} = 6.67 \text{ kN (T)}$ ,  $F_{GC} = 0$
- 6-34.  $F_{GH} = 12.5 \text{ kN (C)}$ ,  $F_{BG} = 6.01 \text{ kN (T)}$ ,  
 $F_{BC} = 6.67 \text{ kN (T)}$
- 6-35.  $F_{BC} = 5.33 \text{ kN (C)}$ ,  $F_{EF} = 5.33 \text{ kN (T)}$ ,  
 $F_{CF} = 4.00 \text{ kN (T)}$
- 6-37.  $F_{EF} = 14.0 \text{ kN (C)}$ ,  $F_{BC} = 13.0 \text{ kN (T)}$ ,  
 $F_{BE} = 1.41 \text{ kN (T)}$ ,  $F_{BF} = 8.00 \text{ kN (T)}$
- 6-38.  $F_{EF} = 15.0 \text{ kN (C)}$ ,  $F_{BC} = 12.0 \text{ kN (T)}$ ,  
 $F_{BE} = 4.24 \text{ kN (T)}$
- 6-39.  $F_{BC} = 10.4 \text{ kN (C)}$ ,  $F_{HG} = 9.16 \text{ kN (T)}$ ,  
 $F_{HC} = 2.24 \text{ kN (T)}$
- 6-41.  $F_{BC} = 18.0 \text{ kN (T)}$ ,  $F_{FE} = 15.0 \text{ kN (C)}$ ,  
 $F_{EB} = 5.00 \text{ kN (C)}$
- 6-42.  $F_{HG} = 17.6 \text{ kN (C)}$ ,  $F_{HC} = 5.41 \text{ kN (C)}$ ,  
 $F_{BC} = 19.1 \text{ kN (T)}$
- 6-43.  $F_{GJ} = 17.6 \text{ kN (C)}$ ,  $F_{CJ} = 8.11 \text{ kN (C)}$ ,  
 $F_{CD} = 21.4 \text{ kN (T)}$ ,  $F_{CG} = 7.50 \text{ kN (T)}$
- 6-45.  $F_{BF} = 0$ ,  $F_{BG} = 35.4 \text{ kN (C)}$ ,  $F_{AB} = 45 \text{ kN (T)}$
- 6-46.  $F_{BC} = 11.0 \text{ kN (T)}$ ,  $F_{GH} = 11.2 \text{ kN (C)}$ ,  
 $F_{CH} = 1.25 \text{ kN (C)}$ ,  $F_{CG} = 10.0 \text{ kN (T)}$
- 6-47.  $F_{CD} = 18.0 \text{ kN (T)}$ ,  $F_{CJ} = 10.8 \text{ kN (T)}$ ,  
 $F_{KJ} = 26.8 \text{ kN (T)}$
- 6-49.  $F_{EF} = 12.9 \text{ kN (T)}$ ,  $F_{FI} = 7.21 \text{ kN (T)}$ ,  
 $F_{HI} = 21.1 \text{ kN (C)}$
- 6-50.  $F_{CA} = F_{CB} = 122 \text{ lb (C)}$ ,  $F_{CD} = 173 \text{ lb (T)}$ ,  
 $F_{BD} = 86.6 \text{ lb (T)}$ ,  $F_{BA} = 0$ ,  $F_{DA} = 86.6 \text{ lb (T)}$
- 6-51.  $F_{AB} = 6.46 \text{ kN (T)}$ ,  $F_{AC} = F_{AD} = 1.50 \text{ kN (C)}$ ,  
 $F_{BC} = F_{BD} = 3.70 \text{ kN (C)}$ ,  $F_{BE} = 4.80 \text{ kN (T)}$
- 6-53.  $F_{CA} = 833 \text{ lb (T)}$ ,  $F_{CB} = 667 \text{ lb (C)}$ ,  
 $F_{CD} = 333 \text{ lb (T)}$ ,  $F_{AD} = F_{AB} = 354 \text{ lb (C)}$ ,  
 $F_{DB} = 50 \text{ lb (T)}$
- 6-54.  $F_{CA} = 1000 \text{ lb (C)}$ ,  $F_{CD} = 406 \text{ lb (T)}$ ,  
 $F_{CB} = 344 \text{ lb (C)}$ ,  $F_{AB} = F_{AD} = 424 \text{ lb (T)}$ ,  
 $F_{DB} = 544 \text{ lb (C)}$
- 6-55.  $F_{DF} = 5.31 \text{ kN (C)}$ ,  $F_{EF} = 2.00 \text{ kN (T)}$ ,  
 $F_{AF} = 0.691 \text{ kN (T)}$
- 6-57.  $F_{BF} = 0$ ,  $F_{BC} = 0$ ,  $F_{BE} = 500 \text{ lb (T)}$ ,  
 $F_{AB} = 300 \text{ lb (C)}$ ,  $F_{AC} = 583 \text{ lb (T)}$ ,  
 $F_{AD} = 333 \text{ lb (T)}$ ,  $F_{AE} = 667 \text{ lb (C)}$ ,  $F_{DE} = 0$ ,  
 $F_{EF} = 300 \text{ lb (C)}$ ,  $F_{CD} = 300 \text{ lb (C)}$ ,  
 $F_{CF} = 300 \text{ lb (C)}$ ,  $F_{DF} = 424 \text{ lb (T)}$
- 6-58.  $F_{BF} = 0$ ,  $F_{BC} = 0$ ,  $F_{BE} = 500 \text{ lb (T)}$ ,  
 $F_{AB} = 300 \text{ lb (C)}$ ,  $F_{AC} = 972 \text{ lb (T)}$ ,  $F_{AD} = 0$ ,  
 $F_{AE} = 367 \text{ lb (C)}$ ,  $F_{DE} = 0$ ,  $F_{EF} = 300 \text{ lb (C)}$ ,  
 $F_{CD} = 500 \text{ lb (C)}$ ,  $F_{CF} = 300 \text{ lb (C)}$ ,  
 $F_{DF} = 424 \text{ lb (T)}$
- 6-59.  $F_{AD} = 686 \text{ N (T)}$ ,  $F_{BD} = 0$ ,  $F_{CD} = 615 \text{ N (C)}$ ,  
 $F_{BC} = 229 \text{ N (T)}$ ,  $F_{AC} = 343 \text{ N (T)}$ ,  
 $F_{EC} = 457 \text{ N (C)}$
- 6-61.  $P = 12.5 \text{ lb}$
- 6-62. **a.**  $P = 25.0 \text{ lb}$ , **b.**  $P = 33.3 \text{ lb}$ , **c.**  $P = 11.1 \text{ lb}$
- 6-63.  $P = 18.9 \text{ N}$
- 6-65.  $B_x = 4.00 \text{ kN}$ ,  $B_y = 5.33 \text{ kN}$ ,  $A_x = 4.00 \text{ kN}$ ,  
 $A_y = 5.33 \text{ kN}$
- 6-66.  $A_x = 24.0 \text{ kN}$ ,  $A_y = 12.0 \text{ kN}$ ,  $D_x = 18.0 \text{ kN}$ ,  
 $D_y = 24.0 \text{ kN}$
- 6-67.  $A_x = 120 \text{ lb}$ ,  $A_y = 0$ ,  $N_C = 15.0 \text{ lb}$
- 6-69.  $B_x = 2.80 \text{ kip}$ ,  $B_y = 1.05 \text{ kip}$ ,  $A_x = 2.80 \text{ kip}$ ,  
 $A_y = 5.10 \text{ kip}$ ,  $M_A = 43.2 \text{ kip} \cdot \text{ft}$
- 6-70.  $C_y = 184 \text{ N}$ ,  $C_x = 490.5 \text{ N}$ ,  $B_x = 1.23 \text{ kN}$ ,  
 $B_y = 920 \text{ kN}$
- 6-71.  $N_E = 18.0 \text{ kN}$ ,  $N_C = 4.50 \text{ kN}$ ,  $A_x = 0$ ,  
 $A_y = 7.50 \text{ kN}$ ,  $M_A = 22.5 \text{ kN} \cdot \text{m}$
- 6-73.  $N_E = 3.60 \text{ kN}$ ,  $N_B = 900 \text{ N}$ ,  $A_x = 0$ ,  
 $A_y = 2.70 \text{ kN}$ ,  $M_A = 8.10 \text{ kN} \cdot \text{m}$
- 6-74.  $T = 350 \text{ lb}$ ,  $A_y = 700 \text{ lb}$ ,  $A_x = 1.88 \text{ kip}$ ,  
 $D_x = 1.70 \text{ kip}$ ,  $D_y = 1.70 \text{ kip}$
- 6-75.  $T = 350 \text{ lb}$ ,  $A_y = 700 \text{ lb}$ ,  $D_x = 1.82 \text{ kip}$ ,  
 $D_y = 1.84 \text{ kip}$ ,  $A_x = 2.00 \text{ kip}$
- 6-77.  $A_x = 96 \text{ lb}$ ,  $A_y = 72 \text{ lb}$ ,  $D_y = 2.18 \text{ kip}$ ,  
 $E_x = 96.0 \text{ lb}$ ,  $E_y = 1.61 \text{ kip}$
- 6-78.  $N_C = 3.00 \text{ kN}$ ,  $N_A = 3.00 \text{ kN}$ ,  
 $B_y = 18.0 \text{ kN}$ ,  $B_x = 0$

- 6-79.  $N_C = N_D = 2 \text{ lb}$   
 6-81.  $F_{FB} = 1.94 \text{ kN}$ ,  $F_{BD} = 2.60 \text{ kN}$   
 6-82.  $N_A = 36.0 \text{ lb}$   
 6-83.  $F_{FD} = 20.1 \text{ kN}$ ,  $F_{BD} = 25.5 \text{ kN}$ ,  
 Member  $EDC$ :  $C'_x = 18.0 \text{ kN}$ ,  $C'_y = 12.0 \text{ kN}$ ,  
 Member  $ABC$ :  $C''_y = 12.0 \text{ kN}$ ,  $C''_x = 18.0 \text{ kN}$   
 6-85.  $T_{AI} = 2.88 \text{ kip}$ ,  $F_H = 3.99 \text{ kip}$   
 6-86.  $M = 314 \text{ lb} \cdot \text{ft}$   
 6-87.  $F_C = 19.6 \text{ kN}$   
 6-89.  $C_x = 650 \text{ N}$ ,  $C_y = 0$   
 6-90.  $N_B = N_C = 49.5 \text{ N}$   
 6-91.  $F_{EF} = 8.18 \text{ kN (T)}$ ,  $F_{AD} = 158 \text{ kN (C)}$   
 6-93. 
$$P(\theta) = \frac{250 \sqrt{2.25^2 - \cos^2 \theta}}{\sin \theta \cos \theta + \sqrt{2.25^2 - \cos^2 \theta} \cdot \cos \theta}$$
  
 6-94.  $N_B = 0.1175 \text{ lb}$ ,  $N_A = 0.0705 \text{ lb}$   
 6-95.  $F_N = 5.25 \text{ lb}$   
 6-97. a.  $F = 205 \text{ lb}$ ,  $N_C = 380 \text{ lb}$ ,  
 b.  $F = 102 \text{ lb}$ ,  $N_C = 72.5 \text{ lb}$   
 6-98.  $E_y = 1.00 \text{ kN}$ ,  $E_x = 3.00 \text{ kN}$ ,  $B_x = 2.50 \text{ kN}$ ,  
 $B_y = 1.00 \text{ kN}$ ,  $A_x = 2.50 \text{ kN}$ ,  $A_y = 500 \text{ N}$   
 6-99.  $N_C = 12.7 \text{ kN}$ ,  $A_x = 12.7 \text{ kN}$ ,  $A_y = 2.94 \text{ kN}$ ,  
 $N_D = 1.05 \text{ kN}$   
 6-101.  $F = 370 \text{ N}$   
 6-102.  $N_A = 284 \text{ N}$   
 6-103.  $B_y = 2.67 \text{ kN}$ ,  $B_x = 4.25 \text{ kN}$ ,  
 $A_y = 3.33 \text{ kN}$ ,  $A_x = 7.25 \text{ kN}$   
 6-105.  $P = 198 \text{ N}$   
 6-106.  $F = 66.1 \text{ lb}$   
 6-107.  $d = 0.638 \text{ ft}$   
 6-109.  $P = 46.9 \text{ lb}$   
 6-110.  $\theta = 23.7^\circ$   
 6-111.  $m = 26.0 \text{ kg}$   
 6-113.  $m_S = 1.71 \text{ kg}$   
 6-114.  $m_L = 106 \text{ kg}$   
 6-115.  $P = 283 \text{ N}$ ,  $B_x = D_x = 42.5 \text{ N}$ ,  
 $B_y = D_y = 283 \text{ N}$ ,  $B_z = D_z = 283 \text{ N}$   
 6-117.  $M_{Ex} = 0.5 \text{ kN} \cdot \text{m}$ ,  $M_{Ey} = 0$ ,  $E_y = 0$ ,  $E_x = 0$   
 6-118.  $F_D = 20.8 \text{ lb}$ ,  $F_F = 14.7 \text{ lb}$ ,  $F_A = 24.5 \text{ lb}$

## Chapter 7

- 7-1.  $N_C = 0$ ,  $V_C = -386 \text{ lb}$ ,  $M_C = -857 \text{ lb} \cdot \text{ft}$ ,  
 $N_D = 0$ ,  $V_D = 300 \text{ lb}$ ,  $M_D = -600 \text{ lb} \cdot \text{ft}$   
 7-2.  $N_C = 0$ ,  $V_C = -1.00 \text{ kip}$ ,  $M_C = 56.0 \text{ kip} \cdot \text{ft}$ ,  
 $N_D = 0$ ,  $V_D = -1.00 \text{ kip}$ ,  $M_D = 48.0 \text{ kip} \cdot \text{ft}$   
 7-3.  $V_A = 0$ ,  $N_A = -39 \text{ kN}$ ,  $M_A = -2.425 \text{ kN} \cdot \text{m}$   
 7-5.  $V_C = -133 \text{ lb}$ ,  $M_C = 133 \text{ lb} \cdot \text{in.}$   
 7-6.  $a = \frac{L}{3}$   
 7-7.  $V_C = -4.00 \text{ kip}$ ,  $M_C = 24.0 \text{ kip} \cdot \text{ft}$   
 7-9.  $N_C = -30 \text{ kN}$ ,  $V_C = -8 \text{ kN}$ ,  $M_C = 6 \text{ kN} \cdot \text{m}$

- 7-10.  $P = 0.533 \text{ kN}$ ,  $N_C = -2 \text{ kN}$ ,  $V_C = -0.533 \text{ kN}$ ,  
 $M_C = 0.400 \text{ kN} \cdot \text{m}$   
 7-11.  $N_C = 265 \text{ lb}$ ,  $V_C = -649 \text{ lb}$ ,  $M_C = -4.23 \text{ kip} \cdot \text{ft}$ ,  
 $N_D = -265 \text{ lb}$ ,  $V_D = 637 \text{ lb}$ ,  $M_D = -3.18 \text{ kip} \cdot \text{ft}$   
 7-13.  $N_D = 0$ ,  $V_D = 3.00 \text{ kip}$ ,  $M_D = 12.0 \text{ kip} \cdot \text{ft}$ ,  
 $N_E = 0$ ,  $V_E = -8.00 \text{ kip}$ ,  $M_E = -20.0 \text{ kip} \cdot \text{ft}$   
 7-14.  $M_C = -15.0 \text{ kip} \cdot \text{ft}$ ,  $N_C = 0$ ,  $V_C = 2.01 \text{ kip}$ ,  
 $M_D = 3.77 \text{ kip} \cdot \text{ft}$ ,  $N_D = 0$ ,  $V_D = 1.11 \text{ kip}$   
 7-15.  $N_C = 0$ ,  $V_C = -1.50 \text{ kN}$ ,  $M_C = 13.5 \text{ kN} \cdot \text{m}$   
 7-17.  $N_A = 86.6 \text{ lb}$ ,  $V_A = 150 \text{ lb}$ ,  $M_A = 1.80 \text{ kip} \cdot \text{in.}$   
 7-18.  $V_C = 2.49 \text{ kN}$ ,  $N_C = 2.49 \text{ kN}$ ,  $M_C = 4.97 \text{ kN} \cdot \text{m}$ ,  
 $N_D = 0$ ,  $V_D = -2.49 \text{ kN}$ ,  $M_D = 16.5 \text{ kN} \cdot \text{m}$   
 7-19.  $N_C = -4.32 \text{ kip}$ ,  $V_C = 1.35 \text{ kip}$ ,  $M_C = 4.72 \text{ kip} \cdot \text{ft}$   
 7-21.  $N_E = 720 \text{ N}$ ,  $V_E = 1.12 \text{ kN}$ ,  $M_E = -320 \text{ N} \cdot \text{m}$ ,  
 $N_F = 0$ ,  $V_F = -1.24 \text{ kN}$ ,  $M_F = -1.41 \text{ kN} \cdot \text{m}$   
 7-22.  $N_D = 4 \text{ kN}$ ,  $V_D = -9 \text{ kN}$ ,  $M_D = -18 \text{ kN} \cdot \text{m}$ ,  
 $N_E = 4 \text{ kN}$ ,  $V_E = 3.75 \text{ kN}$ ,  $M_E = -4.875 \text{ kN} \cdot \text{m}$   
 7-23.  $N_C = 400 \text{ N}$ ,  $V_C = -96 \text{ N}$ ,  $M_C = -144 \text{ N} \cdot \text{m}$   
 7-25.  $N_D = 0$ ,  $V_D = 0.75 \text{ kip}$ ,  $M_D = 13.5 \text{ kip} \cdot \text{ft}$ ,  
 $N_E = 0$ ,  $V_E = -9 \text{ kip}$ ,  $M_E = -24.0 \text{ kip} \cdot \text{ft}$   
 7-26.  $N_C = -20.0 \text{ kN}$ ,  $V_C = 70.6 \text{ kN}$ ,  
 $M_C = -302 \text{ kN} \cdot \text{m}$   
 7-27.  $N_C = -1.60 \text{ kN}$ ,  $V_C = 200 \text{ N}$ ,  $M_C = 200 \text{ N} \cdot \text{m}$   
 7-29.  $N_C = -406 \text{ lb}$ ,  $V_C = 903 \text{ lb}$ ,  $M_C = 1.35 \text{ kip} \cdot \text{ft}$   
 7-30.  $N_D = -464 \text{ lb}$ ,  $V_D = -203 \text{ lb}$ ,  $M_D = 2.61 \text{ kip} \cdot \text{ft}$   
 7-31.  $N_E = 2.20 \text{ kN}$ ,  $V_E = 0$ ,  $M_E = 0$ ,  
 $N_D = -2.20 \text{ kN}$ ,  $V_D = 600 \text{ N}$ ,  $M_D = 1.20 \text{ kN} \cdot \text{m}$   
 7-33.  $N_D = -2.25 \text{ kN}$ ,  $V_D = 1.25 \text{ kN}$ ,  $-1.88 \text{ kN} \cdot \text{m}$   
 7-34.  $N_E = 1.25 \text{ kN}$ ,  $V_E = 0$ ,  $M_B = 1.69 \text{ kN} \cdot \text{m}$   
 7-35.  $d = 0.200 \text{ m}$   
 7-37.  $N_D = 1.26 \text{ kN}$ ,  $V_D = 0$ ,  $M_D = 500 \text{ N} \cdot \text{m}$   
 7-38.  $N_E = -1.48 \text{ kN}$ ,  $V_E = 500 \text{ N}$ ,  $M_E = 1000 \text{ N} \cdot \text{m}$   
 7-39.  $V = 0.278 w_0 r$ ,  $N = 0.0759 w_0 r$ ,  
 $M = 0.0759 w_0 r^2$   
 7-41.  $N_C = -350 \text{ lb}$ ,  $(V_C)_y = 700 \text{ lb}$ ,  $(V_C)_z = -150 \text{ lb}$ ,  
 $(M_C)_x = -1.20 \text{ kip} \cdot \text{ft}$ ,  $(M_C)_y = -750 \text{ lb} \cdot \text{ft}$ ,  
 $(M_C)_z = 1.40 \text{ kip} \cdot \text{ft}$   
 7-42.  $(V_C)_x = 104 \text{ lb}$ ,  $N_C = 0$ ,  $(V_C)_z = 10 \text{ lb}$ ,  
 $(M_C)_x = 20 \text{ lb} \cdot \text{ft}$ ,  $(M_C)_y = 72 \text{ lb} \cdot \text{ft}$ ,  
 $(M_C)_z = -178 \text{ lb} \cdot \text{ft}$   
 7-43.  $N_x = -500 \text{ N}$ ,  $V_y = 100 \text{ N}$ ,  $V_z = 900 \text{ N}$ ,  
 $M_x = 600 \text{ N} \cdot \text{m}$ ,  $M_y = -900 \text{ N} \cdot \text{m}$ ,  
 $M_z = 400 \text{ N} \cdot \text{m}$

7-45. a.  $0 \leq x < a$ :  $V = \left(1 - \frac{a}{L}\right)P$ ,

$$M = \left(1 - \frac{a}{L}\right)Px,$$

$$a < x \leq L$$

$$V = -\left(\frac{a}{L}\right)P,$$

$$M = P\left(a - \frac{a}{L}x\right)$$

- b.**  $0 \leq x < 2 \text{ m}$ :  $V = 6 \text{ kN}$ ,  $M = \{6x\} \text{ kN} \cdot \text{m}$   
 $2 \text{ m} < x \leq 6 \text{ m}$ :  $V = -3 \text{ kN}$ ,  
 $M = \{18 - 3x\} \text{ kN} \cdot \text{m}$
- 7-46. a.** For  $0 \leq x < a$ ,  $V = P$ ,  $M = Px$ ,  
 For  $a < x < L - a$ ,  $V = 0$ ,  $M = Pa$ ,  
 For  $L - a < x \leq L$ ,  $V = -P$ ,  
 $M = P(L - x)$
- b.** For  $0 \leq x < 5 \text{ ft}$ ,  $V = 800 \text{ lb}$ ,  
 $M = 800x \text{ lb} \cdot \text{ft}$ ,  
 For  $5 \text{ ft} < x < 7 \text{ ft}$ ,  $V = 0$ ,  
 $M = 4000 \text{ lb} \cdot \text{ft}$ ,  
 For  $7 \text{ ft} < x \leq 12 \text{ ft}$ ,  $V = -800 \text{ lb}$ ,  
 $M = (9600 - 800x) \text{ lb} \cdot \text{ft}$
- 7-47. a.** For  $0 \leq x < a$ ,  $V = \frac{Pb}{a+b}$ ,  $M = \frac{Pb}{a+b}x$ ,  
 For  $a < x \leq a+b$ ,  $V = -\frac{Pa}{a+b}$ ,  
 $M = Pa - \frac{Pa}{a+b}x$ ,
- b.** For  $0 \leq x < 5 \text{ ft}$ ,  $V = 350 \text{ lb}$ ,  
 $M = 350x \text{ lb} \cdot \text{ft}$ ,  
 For  $5 \text{ ft} < x \leq 12 \text{ ft}$ ,  $V = -250 \text{ lb}$ ,  
 $M = 3000 - 250x \text{ lb} \cdot \text{ft}$
- 7-49.**  $0 \leq x < \frac{L}{3}$ :  $V = 0$ ,  $M = 0$ ,  
 $\frac{L}{3} < x < \frac{2L}{3}$ :  $V = 0$ ,  $M = M_0$ ,  
 $\frac{2L}{3} < x \leq L$ :  $V = 0$ ,  $M = 0$ ,  
 $0 \leq x < \frac{8}{3} \text{ m}$ :  $V = 0$ ,  $M = 0$ ,  
 $\frac{8}{3} \text{ m} < x < \frac{16}{3} \text{ m}$ :  $V = 0$ ,  $M = 500 \text{ N} \cdot \text{m}$ ,  
 $\frac{16}{3} \text{ m} < x \leq 8 \text{ m}$ :  $V = 0$ ,  $M = 0$
- 7-50.**  $M_{\max} = 2 \text{ kN} \cdot \text{m}$
- 7-51.**  $0 \leq x < a$ :  $V = -wx$ ,  $M = -\frac{w}{2}x^2$   
 $a < x \leq 2a$ :  $V = w(2a - x)$ ,  
 $M = 2wax - wa^2 - \frac{w}{2}x^2$
- 7-53.** For  $0 \leq x < 20 \text{ ft}$ ,  $V = \{490 - 50.0x\} \text{ lb}$ ,  
 $M = \{490x - 25.0x^2\} \text{ lb} \cdot \text{ft}$ ,  
 For  $20 \text{ ft} < x \leq 30 \text{ ft}$ ,  $V = 0$ ,  $M = -200 \text{ lb} \cdot \text{ft}$
- 7-54. a.**  $V = \frac{w}{2}(L - 2x)$ ,  $M = \frac{w}{2}(Lx - x^2)$
- b.**  $V = (2500 - 500x) \text{ lb}$ ,  
 $M = (2500x - 250x^2) \text{ lb} \cdot \text{ft}$
- 7-55.** For  $0 \leq x < 8 \text{ m}$ ,  $V = (133.75 - 40x) \text{ kN}$ ,  
 $M = (133.75x - 20x^2) \text{ kN} \cdot \text{m}$ ,  
 For  $8 \text{ m} < x \leq 11 \text{ m}$ ,  $V = 20 \text{ kN}$ ,  
 $M = (20x - 370) \text{ kN} \cdot \text{m}$
- 7-57.** For  $0 \leq x < L$ ,  $V = \frac{w}{18}(7L - 18x)$ ,  
 $M = \frac{w}{18}(7Lx - 9x^2)$ ,  
 For  $L < x < 2L$ ,  
 $V = \frac{w}{2}(3L - 2x)$ ,  $M = \frac{w}{18}(27Lx - 20L^2 - 9x^2)$ ,  
 For  $2L < x \leq 3L$ ,  $V = \frac{w}{18}(47L - 18x)$ ,  
 $M = \frac{w}{18}(47Lx - 9x^2 - 60L^2)$
- 7-58.** Member  $AB$ : For  $0 \leq x < 12 \text{ ft}$ ,  
 $V = \{875 - 150x\} \text{ lb}$ ,  
 $M = \{875x - 75.0x^2\} \text{ lb} \cdot \text{ft}$ ,  
 For  $12 \text{ ft} < x \leq 14 \text{ ft}$ ,  $V = \{2100 - 150x\} \text{ lb}$ ,  
 $M = \{-75.0x^2 + 2100x - 14700\} \text{ lb} \cdot \text{ft}$ ,  
 Member  $CBD$ : For  $0 \leq x < 2 \text{ ft}$ ,  $V = 919 \text{ lb}$ ,  
 $M = \{919x\} \text{ lb} \cdot \text{ft}$ , For  $2 \text{ ft} < x \leq 8 \text{ ft}$ ,  
 $V = -306 \text{ lb}$ ,  $M = \{2450 - 306x\} \text{ lb} \cdot \text{ft}$
- 7-59.** For  $0 \leq x < 9 \text{ ft}$ ,  $V = 25 - 1.67x^2$ ,  
 $M = 25x - 0.556x^3$   
 For  $9 \text{ ft} < x \leq 13.5 \text{ ft}$ ,  $V = 0$ ,  $M = -180$
- 7-61.**  $x = 15^-$ ,  $V = -20$ ,  $M = -300$ ,  
 $x = 30^+$ ,  $V = 0$ ,  $M = 150$ ,  
 $x = 45^-$ ,  $V = -60$ ,  $M = -300$
- 7-62.**  $x = \frac{L}{2}$ ,  $P = \frac{4M_{\max}}{L}$
- 7-63.**  $0 \leq x \leq 12 \text{ ft}$ :  $V = \left\{48.0 - \frac{x^2}{6}\right\} \text{ kip}$ ,  
 $M = \left\{48.0x - \frac{x^3}{18} - 576\right\} \text{ kip} \cdot \text{ft}$ ,  
 $12 < x \leq 24 \text{ ft}$ :  $V = \left\{\frac{1}{6}(24 - x)^2\right\} \text{ kip}$ ,  
 $M = \left\{-\frac{1}{18}(24 - x)^3\right\} \text{ kip} \cdot \text{ft}$
- 7-65.** For  $0 \leq x < 3 \text{ m}$ ,  $V = \{21.0 - 2x^2\} \text{ kN}$ ,  
 $M = \left\{21.0x - \frac{2}{3}x^3\right\} \text{ kN} \cdot \text{m}$ ,  
 For  $3 \text{ m} < x \leq 6 \text{ m}$ ,  $V = \{39.0 - 12x\} \text{ kN}$ ,  
 $M = \{-6x^2 + 39x - 18\} \text{ kN} \cdot \text{m}$
- 7-66.**  $V = \frac{w}{12L}(4L^2 - 6Lx - 3x^2)$ ,  
 $M = \frac{w}{12L}(4L^2x - 3Lx^2 - x^3)$ ,  $M_{\max} = 0.0940 wL^2$
- 7-67.**  $N = P \sin(\theta + \phi)$ ,  $V = -P \cos(\theta + \phi)$ ,  
 $M = Pr[\sin(\theta + \phi) - \sin \phi]$
- 7-69.**  $V_x = 0$ ,  $V_z = \{24.0 - 4y\} \text{ lb}$ ,  
 $M_x = \{2y^2 - 24y + 64.0\} \text{ lb} \cdot \text{ft}$ ,  
 $M_y = 8.00 \text{ lb} \cdot \text{ft}$ ,  $M_z = 0$
- 7-70.**  $x = 1^-$ ,  $V = 450 \text{ N}$ ,  $M = 450 \text{ N} \cdot \text{m}$ ,  
 $x = 3^+$ ,  $V = -950 \text{ N}$ ,  $M = 950 \text{ N} \cdot \text{m}$

- 7-71.  $x = 1^-$ ,  $V = 600 \text{ N}$ ,  $M = 600 \text{ N} \cdot \text{m}$   
 7-74.  $x = 0.5^+$ ,  $V = 450 \text{ N}$ ,  $M = -150 \text{ N} \cdot \text{m}$ ,  
 $x = 1.5^-$ ,  $V = -750 \text{ N}$ ,  $M = -300 \text{ N} \cdot \text{m}$   
 7-75.  $x = 2^+$ ,  $V = -375 \text{ N}$ ,  $M = 750 \text{ N} \cdot \text{m}$   
 7-77.  $x = 10^+$ ,  $V = 20.0 \text{ kip}$ ,  $M = -50.0 \text{ kip} \cdot \text{ft}$   
 7-78.  $x = 2^+$ ,  $V = -14.3$ ,  $M = -8.6$   
 7-79.  $x = 1^+$ ,  $V = 175$ ,  $M = -200$ ,  
 $x = 5^-$ ,  $V = -225$ ,  $M = -300$   
 7-81.  $x = 4.5^-$ ,  $V = -31.5 \text{ kN}$ ,  $M = -45.0 \text{ kN} \cdot \text{m}$ ,  
 $x = 8.5^+$ ,  $V = 36.0 \text{ kN}$ ,  $M = -54.0 \text{ kN} \cdot \text{m}$   
 7-82.  $x = 2.75$ ,  $V = 0$ ,  $M = 1356 \text{ N} \cdot \text{m}$   
 7-83.  $x = 3$ ,  $V = -2.25 \text{ kN}$ ,  $M = 20.25 \text{ kN} \cdot \text{m}$   
 7-85.  $x = 3^+$ ,  $V = 1800 \text{ lb}$ ,  $M = -900 \text{ lb} \cdot \text{ft}$   
 $x = 6$ ,  $V = 0$ ,  $M = 1800 \text{ lb} \cdot \text{ft}$   
 7-86.  $x = 1.5$ ,  $V = 2.25 \text{ kN}$ ,  $M = -2.25 \text{ kN} \cdot \text{m}$   
 7-87.  $x = 3$ ,  $V = 3.00 \text{ kN}$ ,  $M = -1.50 \text{ kN} \cdot \text{m}$   
 7-89.  $x = \sqrt{15}$ ,  $V = 0$ ,  $M = 1291 \text{ lb} \cdot \text{ft}$   
 $x = 12^-$ ,  $V = -1900 \text{ lb}$ ,  $M = -6000 \text{ lb} \cdot \text{ft}$   
 7-90.  $x = 0$ ,  $V = 13.5 \text{ kN}$ ,  $M = -9.5 \text{ kN} \cdot \text{m}$   
 7-91.  $x = 3$ ,  $V = 0$ ,  $M = 18.0 \text{ kN} \cdot \text{m}$   
 $x = 6^-$ ,  $V = -27.0 \text{ kN}$ ,  $M = -18.0 \text{ kN} \cdot \text{m}$   
 7-93.  $x = 15$ ,  $V = 0$ ,  $M = 37.5 \text{ kip} \cdot \text{ft}$   
 7-94.  $y_B = 2.22 \text{ m}$ ,  $y_D = 1.55 \text{ m}$   
 7-95.  $P_1 = 320 \text{ N}$ ,  $y_D = 2.33 \text{ m}$   
 7-97.  $x_B = 5.39 \text{ m}$   
 7-98.  $P = 700 \text{ N}$   
 7-99.  $y_B = 8.67 \text{ ft}$ ,  $y_D = 7.04 \text{ ft}$   
 7-101.  $y_B = 3.53 \text{ m}$ ,  $P = 0.8 \text{ kN}$ ,  $T_{\max} = T_{DE} = 8.17 \text{ kN}$   
 7-102.  $w = 51.9 \text{ lb/ft}$   
 7-103.  $T_{\max} = 14.4 \text{ kip}$ ,  $T_{\min} = 13.0 \text{ kip}$   
 7-105.  $T_{AB} = T_{CD} = 212 \text{ lb (max)}$ ,  $y_B = 2 \text{ ft}$   
 7-106.  $x = 2.57 \text{ ft}$ ,  $W = 247 \text{ lb}$   
 7-107.  $T_A = 61.7 \text{ kip}$ ,  $T_B = 36.5 \text{ kip}$ ,  $T_C = 50.7 \text{ kip}$   
 7-109.  $T_{\max} = 594 \text{ kN}$   
 7-110.  $T_{\min} = 552 \text{ kN}$   
 7-111.  $T_{\max} = 3.63 \text{ kip}$   
 7-113.  $y = \frac{x^2}{7813} \left( 75 - \frac{x^2}{200} \right)$ ,  $T_{\max} = 9.28 \text{ kip}$   
 7-114.  $h = 4.44 \text{ ft}$   
 7-115.  $(F_h)_R = 6.25 \text{ kip}$ ,  $(F_v)_R = 2.51 \text{ kip}$   
 7-117.  $(F_v)_A = 165 \text{ N}$ ,  $(F_h)_A = 73.9 \text{ N}$   
 7-118.  $W = 4.00 \text{ kip}$ ,  $T_{\max} = 2.01 \text{ kip}$   
 7-121.  $l = 104 \text{ ft}$   
 7-122.  $h = 146 \text{ ft}$   
 7-123.  $L = 302 \text{ ft}$

## Chapter 8

- 8-1.  $P = 12.8 \text{ kN}$   
 8-2.  $N_B = 2.43 \text{ kip}$ ,  $N_C = 1.62 \text{ kip}$ ,  $F = 200 \text{ lb}$   
 8-3.  $N_A = 16.5 \text{ kN}$ ,  $N_B = 42.3 \text{ kN}$ ,  
 It does not move.  
 8-5.  $F = 2.76 \text{ kN}$

- 8-6.  $F = 5.79 \text{ kN}$   
 8-7. a. No  
 b. Yes  
 8-10.  $\phi = \theta$ ,  $P = W \sin(\alpha + \theta)$   
 8-11. a.  $W = 318 \text{ lb}$   
 b.  $W = 360 \text{ lb}$   
 8-13.  $F_{CD} = 3.05 \text{ kN}$   
 8-14.  $\theta = 21.8^\circ$   
 8-15.  $l = 26.7 \text{ ft}$   
 8-17.  $\mu_s = 0.231$   
 8-18.  $P = 1350 \text{ lb}$   
 8-19.  $N_A = 200 \text{ lb}$   
 8-21.  $n = 12$   
 8-22.  $\mu_s = 0.595$   
 8-23.  $\theta = 33.4^\circ$   
 8-25.  $d = 13.4 \text{ ft}$   
 8-26.  $P = 740 \text{ N}$   
 8-27.  $P = 860 \text{ N}$   
 8-29.  $\theta = 11.0^\circ$   
 8-30.  $\theta = 10.6^\circ$ ,  $x = 0.184 \text{ ft}$   
 8-31.  $\theta = 8.53^\circ$ ,  $F_A = 1.48 \text{ lb}$ ,  $F_B = 0.890 \text{ lb}$   
 8-33. No  
 8-34. If  $P = \frac{1}{2} W$ ,  $\mu_s = \frac{1}{3}$   
 If  $P \neq \frac{1}{2} W$ ,  

$$\mu_s = \frac{(P + W) - \sqrt{(W + 7P)(W - P)}}{2(2P - W)}$$
  
 for  $0 < P < W$   
 8-35.  $P = 8.18 \text{ lb}$   
 8-37.  $O_y = 400 \text{ N}$ ,  $O_x = 46.4 \text{ N}$   
 8-38.  $P = 350 \text{ N}$ ,  $O_y = 945 \text{ N}$ ,  $O_x = 280 \text{ N}$   
 8-39.  $\mu_s = 0.230$   
 8-41.  $\theta = 31.0^\circ$   
 8-42.  $P = 654 \text{ N}$   
 8-43. The block fails to be in equilibrium.  
 8-45.  $P = 355 \text{ N}$   
 8-46.  $\mu_C = 0.0734$ ,  $\mu_B = 0.0964$   
 8-47.  $\theta = 16.3^\circ$   
 8-49. Yes  
 8-50.  $m = 66.7 \text{ kg}$   
 8-51.  $P = 408 \text{ N}$   
 8-53.  $M = 55.2 \text{ lb} \cdot \text{ft}$   
 8-54.  $\theta = 33.4^\circ$   
 8-55.  $P = 13.3 \text{ lb}$   
 8-57.  $P = 100 \text{ N}$ ,  $d = 1.50 \text{ ft}$   
 8-58.  $\theta = 33.4^\circ$   
 8-59.  $P = 5.53 \text{ kN}$ , yes  
 8-61.  $P = 39.6 \text{ lb}$   
 8-62.  $x = 18.3 \text{ mm}$   
 8-63.  $P = 2.39 \text{ kN}$

- 8-65.  $P = 4.05 \text{ kip}$   
 8-66.  $P = 106 \text{ lb}$   
 8-67.  $F = 66.7 \text{ N}$   
 8-69.  $W = 7.19 \text{ kN}$   
 8-70. The screw is self-locking.  
 8-71.  $P = 617 \text{ lb}$   
 8-74.  $M = 40.6 \text{ N} \cdot \text{m}$   
 8-75.  $M = 48.3 \text{ N} \cdot \text{m}$   
 8-77.  $\mu_s = 0.0637$   
 8-78.  $M = 5.69 \text{ lb} \cdot \text{in.}$   
 8-79.  $F = 1.98 \text{ kN}$   
 8-81.  $F = 11.6 \text{ kN}$   
 8-82.  $P = 104 \text{ N}$   
 8-83. a.  $F = 1.31 \text{ kN}$   
 b.  $F = 372 \text{ N}$   
 8-85. He will successfully restrain the cow.  
 8-86. Yes, it is possible.  
 $F = 137 \text{ lb}$   
 8-87.  $T_1 = 57.7 \text{ lb}$   
 8-89.  $m_A = 2.22 \text{ kg}$   
 8-90.  $\theta = 99.2^\circ$   
 8-91.  $n = 3 \text{ half turns, } N_m = 6.74 \text{ lb}$   
 8-93.  $M = 458 \text{ N} \cdot \text{m}$   
 8-94.  $W = 9.17 \text{ lb}$   
 8-95.  $P = 78.7 \text{ lb}$   
 8-97.  $M = 75.4 \text{ N} \cdot \text{m, } V = 0.171 \text{ m}^3$   
 8-99.  $P = 53.6 \text{ N}$   
 8-101.  $x = 0.384 \text{ m}$   
 8-102.  $F_s = 85.4 \text{ N}$   
 8-103.  $W_D = 12.7 \text{ lb}$   
 8-105.  $\theta_{\max} = 38.8^\circ$   
 8-106.  $M = 50.0 \text{ N} \cdot \text{m, } x = 286 \text{ mm}$   
 8-107.  $M = 132 \text{ N} \cdot \text{m}$   
 8-109.  $F = 10.7 \text{ lb}$   
 8-110.  $M = 16.1 \text{ N} \cdot \text{m}$   
 8-111.  $M = 237 \text{ N} \cdot \text{m}$   
 8-113.  $M = \frac{2\mu_s PR}{3 \cos \theta}$   
 8-114.  $T = 905 \text{ lb} \cdot \text{in.}$   
 8-115.  $P = 118 \text{ N}$   
 8-117.  $P = 29.0 \text{ lb}$   
 8-118.  $M = \frac{8}{15} \mu_s PR$   
 8-119.  $F = 18.9 \text{ N}$   
 8-121.  $P = 20.5 \text{ lb}$   
 8-122.  $T = 289 \text{ lb, } N = 479 \text{ lb, } F = 101 \text{ lb}$   
 8-123.  $\mu_s = 0.0407$   
 8-125.  $r = 20.6 \text{ mm}$   
 8-126.  $P = 42.2 \text{ lb}$   
 8-127.  $\mu_s = 0.411$   
 8-129.  $P = 1333 \text{ lb}$   
 8-130.  $P = 25.3 \text{ lb}$

## Chapter 9

- 9-1.  $\bar{x} = 124 \text{ mm, } \bar{y} = 0$   
 9-2.  $\bar{x} = 0, \bar{y} = 1.82 \text{ ft}$   
 9-3.  $\bar{x} = 0.574 \text{ m, } B_x = 0, A_y = 63.1 \text{ N, } B_y = 84.8 \text{ N}$   
 9-5.  $\bar{y} = 0.857 \text{ m}$   
 9-6.  $\bar{y} = \frac{2}{5} \text{ m}$   
 9-7.  $\bar{x} = \frac{3}{8} a$   
 9-9.  $\bar{x} = \frac{3}{2} \text{ m}$   
 9-10.  $\bar{y} = \frac{12}{5} \text{ m}$   
 9-11.  $\bar{x} = \frac{3}{4} b$   
 9-13.  $\bar{x} = 6 \text{ m}$   
 9-14.  $\bar{y} = 2.8 \text{ m}$   
 9-15.  $\bar{x} = 0.398 \text{ m}$   
 9-17.  $\bar{y} = 1.43 \text{ in.}$   
 9-18.  $\bar{x} = \frac{a(1+n)}{2(2+n)}$   
 9-19.  $\bar{y} = \frac{hn}{2n+1}$   
 9-21.  $\bar{x} = 1\frac{3}{5} \text{ ft}$   
 9-22.  $\bar{y} = 4\frac{8}{55} \text{ ft}$   
 9-23.  $\bar{x} = \frac{3}{8} a$   
 9-25.  $\bar{x} = 3.20 \text{ ft, } \bar{y} = 3.20 \text{ ft, } T_A = 384 \text{ lb,}$   
 $T_C = 384 \text{ lb, } T_B = 1.15 \text{ kip}$   
 9-26.  $\bar{x} = 3 \text{ ft}$   
 9-27.  $\bar{y} = \frac{6}{5} \text{ ft}$   
 9-29.  $\bar{y} = 40.0 \text{ mm}$   
 9-30.  $\bar{x} = \frac{1}{3}(a+b)$   
 9-31.  $\bar{y} = \frac{h}{3}$   
 9-33.  $\bar{y} = \frac{\pi a}{8}$   
 9-34.  $\bar{x} = 1.26 \text{ m, } \bar{y} = 0.143 \text{ m, } N_B = 47.9 \text{ kN,}$   
 $A_x = 33.9 \text{ kN, } A_y = 73.9 \text{ kN}$   
 9-35.  $\bar{x} = \left[ \frac{2(n+1)}{3(n+2)} \right] a$   
 9-37.  $\bar{x} = \frac{2}{3} \left( \frac{r \sin \alpha}{\alpha} \right)$   
 9-38.  $\bar{x} = 0.785 a$   
 9-39.  $\bar{x} = \bar{y} = 0, \bar{z} = \frac{4}{3} \text{ m}$

- 9-41.  $\bar{z} = \frac{R^2 + 3r^2 + 2rR}{4(R^2 + r^2 + rR)}h$   
 9-42.  $\bar{y} = 2.61 \text{ ft}$   
 9-43.  $\bar{z} = \frac{h}{4}, \bar{x} = \bar{y} = \frac{a}{\pi}$   
 9-45.  $\bar{z} = \frac{4}{3} \text{ m}$   
 9-46.  $\bar{y} = \frac{3}{8}b, \bar{x} = \bar{z} = 0$   
 9-47.  $\bar{z} = 12.8 \text{ in.}$   
 9-49.  $\bar{z} = 0.675a$   
 9-50.  $\bar{z} = \frac{c}{4}$   
 9-51.  $d = 3 \text{ m}$   
 9-53.  $\bar{x} = 24.4 \text{ mm}, \bar{y} = 40.6 \text{ mm}$   
 9-54.  $\bar{x} = 0, \bar{y} = 58.3 \text{ mm}$   
 9-55.  $\bar{x} = 112 \text{ mm}, \bar{y} = 112 \text{ mm}, \bar{z} = 136 \text{ mm}$   
 9-57.  $\bar{x} = 0.200 \text{ m}, \bar{y} = 4.37 \text{ m}$   
 9-58.  $\bar{y} = 154 \text{ mm}$   
 9-59.  $\bar{x} = 0.571 \text{ in.}, \bar{y} = -0.571 \text{ in.}$   
 9-61.  $\bar{y} = 79.7 \text{ mm}$   
 9-62.  $\bar{x} = -1.00 \text{ in.}, \bar{y} = 4.625 \text{ in.}$   
 9-63.  $\bar{y} = 85.9 \text{ mm}$   
 9-65.  $\bar{x} = 1.57 \text{ in.}, \bar{y} = 1.57 \text{ in.}$   
 9-66.  $\bar{y} = 2 \text{ in.}$   
 9-67.  $\bar{y} = 272 \text{ mm}$   
 9-69.  $\bar{z} = 1.625 \text{ in.}$   
 9-70.  $\bar{z} = 4.32 \text{ in.}$   
 9-71.  $\bar{x} = \frac{\frac{2}{3}r \sin^3 \alpha}{\alpha - \frac{\sin 2\alpha}{2}}$   
 9-73.  $\bar{y} = \frac{\sqrt{2}(a^2 + at - t^2)}{2(2a - t)}$   
 9-74.  $\bar{x} = 2.81 \text{ ft}, \bar{y} = 1.73 \text{ ft}, N_B = 72.1 \text{ lb},$   
 $N_A = 86.9 \text{ lb}$   
 9-75.  $\bar{x} = 120 \text{ mm}, \bar{y} = 305 \text{ mm}, \bar{z} = 73.4 \text{ mm}$   
 9-77.  $\theta = 53.1^\circ$   
 9-78.  $\bar{z} = 2.48 \text{ ft}, \theta = 38.9^\circ$   
 9-79.  $\bar{z} = 0.70 \text{ ft}$   
 9-81.  $\bar{z} = 122 \text{ mm}$   
 9-82.  $h = 385 \text{ mm}$   
 9-83.  $\bar{x} = 5.07 \text{ ft}, \bar{y} = 3.80 \text{ ft}$   
 9-85.  $\bar{z} = 128 \text{ mm}$   
 9-86.  $\bar{z} = 754 \text{ mm}$   
 9-87.  $\bar{x} = 19.0 \text{ ft}, \bar{y} = 11.0 \text{ ft}$   
 9-89.  $\Sigma m = 16.4 \text{ kg}, \bar{x} = 153 \text{ mm},$   
 $\bar{y} = -15 \text{ mm}, \bar{z} = 111 \text{ mm}$   
 9-90.  $V = 27.2(10^3) \text{ ft}^3$   
 9-91.  $A = 3.56(10^3) \text{ ft}^2$   
 9-93.  $A = 4856 \text{ ft}^2$   
 9-94.  $W = 3.12(10^6) \text{ lb}$   
 9-95.  $V = \frac{\pi(6\pi + 4)}{6}a^3$   
 9-97.  $V = 0.114 \text{ m}^3$   
 9-98.  $A = 2.25 \text{ m}^2$   
 9-99.  $A = 276(10^3) \text{ mm}^2$   
 9-101.  $W = 84.7 \text{ kip}$   
 9-102. Number of gal. = 2.75 gal  
 9-103.  $A = 8\pi ba, V = 2\pi ba^2$   
 9-105.  $Q = 205 \text{ MJ}$   
 9-106.  $A = 119(10^3) \text{ mm}^2$   
 9-107.  $W = 126 \text{ kip}$   
 9-109.  $A = 1365 \text{ m}^2$   
 9-110.  $m = 138 \text{ kg}$   
 9-111.  $m = 2.68 \text{ kg}$   
 9-113.  $V = 1.40(10^3) \text{ in}^3$   
 9-114.  $h = 29.9 \text{ mm}$   
 9-115.  $F_R = 1250 \text{ lb}, \bar{x} = 2.33 \text{ ft}, \bar{y} = 4.33 \text{ ft}$   
 9-117.  $F_R = 24.0 \text{ kN},$   
 $\bar{x} = 2.00 \text{ m}, \bar{y} = 1.33 \text{ m}$   
 9-118.  $F_R = \frac{4ab}{\pi^2}p_0, \bar{x} = \frac{a}{2}, \bar{y} = \frac{b}{2}$   
 9-119.  $F_{Rx} = 2rp_0\left(\frac{\pi}{2}\right), F_R = \pi rp_0$   
 9-121. For water:  $F_{RA} = 157 \text{ kN}, F_{RB} = 235 \text{ kN}$   
 For oil:  $d = 4.22 \text{ m}$   
 9-122.  $d = 2.61 \text{ m}$   
 9-123. F.S. = 2.71  
 9-125.  $F_1 = 9.60 \text{ kip}, F_2 = 40.3 \text{ kip}$   
 9-126.  $F_R = 427 \text{ lb}, \bar{y} = 1.71 \text{ ft}, \bar{x} = 0$   
 9-127.  $F_B = 29.4 \text{ kN}, F_A = 235 \text{ kN}$   
 9-129.  $F = 102 \text{ kN}$   
 9-130.  $F_{R_v} = 196 \text{ lb}, F_{R_h} = 125 \text{ lb}$

## Chapter 10

- 10-1.  $I_x = \frac{ab^3}{3(3n + 1)}$   
 10-2.  $I_y = \frac{a^3b}{n + 3}$   
 10-3.  $I_x = 457(10^6) \text{ mm}^4$   
 10-5.  $I_x = 0.133 \text{ m}^4$   
 10-6.  $I_y = 0.286 \text{ m}^4$   
 10-7.  $I_x = 0.267 \text{ m}^4$   
 10-9.  $I_x = 23.8 \text{ ft}^4$   
 10-10.  $I_x = \frac{2}{15}bh^3$   
 10-11.  $I_x = 614 \text{ m}^4$   
 10-13.  $I_x = \frac{\pi}{8} \text{ m}^4$   
 10-14.  $I_y = \frac{\pi}{2} \text{ m}^4$



$$10-15. I_x = 205 \text{ in}^4$$

$$10-17. I_x = \frac{1}{30}bh^3$$

$$10-18. I_y = \frac{b^3h}{6}$$

$$10-19. I_x = 0.267 \text{ m}^4$$

$$10-21. I_x = 0.8 \text{ m}^4$$

$$10-22. I_y = 0.571 \text{ m}^4$$

$$10-23. I_x = \frac{3ab^3}{35}$$

$$10-25. I_x = 209 \text{ in}^4$$

$$10-26. I_y = 533 \text{ in}^4$$

$$10-27. A = 14.0(10^3) \text{ mm}^2$$

$$10-29. \bar{y} = 52.5 \text{ mm}, I_{x'} = 16.6(10^6) \text{ mm}^4, \\ I_{y'} = 5.725(10^6) \text{ mm}^4$$

$$10-30. I_x = 182 \text{ in}^4$$

$$10-31. I_y = 966 \text{ in}^4$$

$$10-33. I_y = 2.03(10^9) \text{ mm}^4$$

$$10-34. I_y = 115(10^6) \text{ mm}^4$$

$$10-35. \bar{y} = 207 \text{ mm}, \bar{I}_{x'} = 222(10^6) \text{ mm}^4$$

$$10-37. I_y = 90.2(10^6) \text{ mm}^4$$

$$10-38. I_x = 1971 \text{ in}^4$$

$$10-39. I_y = 2376 \text{ in}^4$$

$$10-41. I_y = 341 \text{ in}^4$$

$$10-42. I_x = 154(10^6) \text{ mm}^4$$

$$10-43. I_y = 91.3(10^6) \text{ mm}^4$$

$$10-45. \bar{x} = 61.6 \text{ mm}, \bar{I}_{y'} = 41.2(10^6) \text{ mm}^4$$

$$10-46. I_x = 1845 \text{ in}^4$$

$$10-47. I_y = 522 \text{ in}^4$$

$$10-49. I_{y'} = \frac{ab \sin \theta}{12}(b^2 + a^2 \cos^2 \theta)$$

$$10-50. \bar{y} = 0.181 \text{ m}, \bar{I}_{x'} = 4.23(10^{-3}) \text{ m}^4$$

$$10-51. \bar{I}_{x'} = 520(10^6) \text{ mm}^4$$

$$10-53. I_y = 365 \text{ in}^4$$

$$10-54. I_{xy} = \frac{1}{3}t^3 \sin 2\theta$$

$$10-55. I_{xy} = 5.06 \text{ in}^4$$

$$10-57. I_{xy} = 10.7 \text{ m}^4, \bar{I}_{x'y'} = 1.07 \text{ m}^4$$

$$10-58. I_{xy} = \frac{1}{6}a^2b^2$$

$$10-59. I_{xy} = \frac{a^4}{280}$$

$$10-61. I_{xy} = 0.667 \text{ in}^4$$

$$10-62. I_{xy} = 97.8 \text{ in}^4$$

$$10-63. I_u = 15.75 \text{ in}^4, I_v = 25.75 \text{ in}^4$$

$$10-65. I_{xy} = 119 \text{ in}^4$$

$$10-66. I_{xy} = 98.4(10^6) \text{ mm}^4$$

$$10-67. \bar{x} = \bar{y} = 44.1 \text{ mm}, I_{x'y'} = -6.26(10^6) \text{ mm}^4$$

$$10-69. I_u = 3.47(10^3) \text{ in}^4, I_v = 3.47(10^3) \text{ in}^4, \\ I_{uv} = 2.05(10^3) \text{ in}^4$$

$$10-70. I_u = 1.28(10^6) \text{ mm}^4, I_v = 3.31(10^6) \text{ mm}^4, \\ I_{uv} = -1.75(10^6) \text{ mm}^4$$

$$10-71. I_u = 1.28(10^6) \text{ mm}^4, I_{uv} = -1.75(10^6) \text{ mm}^4, \\ I_v = 3.31(10^6) \text{ mm}^4$$

$$10-73. I_{\max} = 1219 \text{ in}^4, I_{\min} = 36.3 \text{ in}^4, (\theta_p)_2 = 19.0^\circ \searrow, \\ (\theta_p)_1 = 71.0^\circ \swarrow$$

$$10-74. I_{\max} = 17.4(10^6) \text{ mm}^4, I_{\min} = 1.84(10^6) \text{ mm}^4 \\ (\theta_p)_1 = 60.0^\circ, (\theta_p)_2 = -30.0^\circ$$

$$10-75. I_{\max} = 17.4(10^6) \text{ mm}^4, I_{\min} = 1.84(10^6) \text{ mm}^4, \\ (\theta_p)_2 = 30.0^\circ \swarrow, (\theta_p)_1 = 60.0^\circ \searrow$$

$$10-77. I_{\max} = 250 \text{ in}^4, I_{\min} = 20.4 \text{ in}^4, (\theta_p)_2 = 22.5^\circ \searrow \\ (\theta_p)_1 = 67.5^\circ \swarrow$$

$$10-78. \theta = 6.08^\circ, I_{\max} = 1.74(10^3) \text{ in}^4, I_{\min} = 435 \text{ in}^4$$

$$10-79. \theta = 6.08^\circ, I_{\max} = 1.74(10^3) \text{ in}^4, I_{\min} = 435 \text{ in}^4$$

$$10-81. I_u = 11.8(10^6) \text{ mm}^4, I_{uv} = -5.09(10^6) \text{ mm}^4, \\ I_v = 5.90(10^6) \text{ mm}^4$$

$$10-82. \theta_{p1} = -31.4^\circ, \theta_{p2} = 58.6^\circ, I_{\max} = 309 \text{ in}^4, \\ I_{\min} = 42.1 \text{ in}^4$$

$$10-83. I_{\max} = 309 \text{ in}^4, I_{\min} = 42.1 \text{ in}^4, \\ \theta_{p1} = -31.4^\circ, \theta_{p2} = 58.6^\circ$$

$$10-85. I_x = \frac{2}{5}mb^2$$

$$10-86. k_x = 57.7 \text{ mm}$$

$$10-87. I_x = \frac{1}{3}ma^2$$

$$10-89. I_x = \frac{2}{5}mb^2$$

$$10-90. k_x = \sqrt{\frac{n+2}{2(n+4)}}h$$

$$10-91. I_y = 2.25 \text{ slug} \cdot \text{ft}^2$$

$$10-93. I_x = \frac{3}{10}mr^2$$

$$10-94. I_y = 1.71(10^3) \text{ kg} \cdot \text{m}^2$$

$$10-95. I_A = 0.0453 \text{ kg} \cdot \text{m}^2$$

$$10-97. I_z = 1.53 \text{ kg} \cdot \text{m}^2$$

$$10-98. \bar{y} = 1.78 \text{ m}, I_G = 4.45 \text{ kg} \cdot \text{m}^2$$

$$10-99. I_O = 0.276 \text{ kg} \cdot \text{m}^2$$

$$10-101. I_A = 222 \text{ slug} \cdot \text{ft}^2$$

$$10-102. I_z = 29.4 \text{ kg} \cdot \text{m}^2$$

$$10-103. I_O = \frac{1}{2}ma^2$$

$$10-105. I_z = 0.113 \text{ kg} \cdot \text{m}^2$$

$$10-106. I_G = 118 \text{ slug} \cdot \text{ft}^2$$

$$10-107. I_O = 282 \text{ slug} \cdot \text{ft}^2$$

$$10-109. I_z = 34.2 \text{ kg} \cdot \text{m}^2$$

## Chapter 11

$$11-1. F_{AC} = 7.32 \text{ lb}$$

$$11-2. F = 2P \cot \theta$$

$$11-3. F_S = 15 \text{ lb}$$

$$11-5. F = 369 \text{ N}$$



**11-7.**  $M = 52.0 \text{ lb} \cdot \text{ft}$

**11-9.**  $\theta = 16.6^\circ, \theta = 35.8^\circ$

**11-10.**  $P = \frac{W}{2} \cot \theta$

**11-11.**  $\theta = 23.8^\circ, \theta = 72.3^\circ$

**11-13.**  $\theta = 90^\circ, \theta = 36.1^\circ$

**11-14.**  $k = 166 \text{ N/m}$

**11-15.**  $F = \frac{M}{2a \sin \theta}$

**11-17.**  $M = 13.1 \text{ N} \cdot \text{m}$

**11-18.**  $\theta = 41.2^\circ$

**11-19.**  $k = 9.88 \text{ kN/m}$

**11-21.**  $F = \frac{500 \sqrt{0.04 \cos^2 \theta + 0.6}}{(0.2 \cos \theta + \sqrt{0.04 \cos^2 \theta + 0.6}) \sin \theta}$

**11-22.**  $\theta = 9.21^\circ$

**11-23.**  $W_G = 2.5 \text{ lb}$

**11-25.**  $F = \frac{W(a + b - d \tan \theta)}{ac} \sqrt{a^2 + c^2 + 2ac \sin \theta}$

**11-26.**  $x = -0.5 \text{ ft}$  unstable,  $x = 0.833 \text{ ft}$  stable

**11-27.** Unstable at  $\theta = 34.6^\circ$ , stable at  $\theta = 145^\circ$

**11-29.**  $\theta = 38.7^\circ$  unstable,  $\theta = 90^\circ$  stable,  
 $\theta = 141^\circ$  unstable

**11-30.**  $x = -0.424 \text{ ft}$  unstable,  $x = 0.590 \text{ ft}$  stable

**11-31.**  $\theta = 20.2^\circ$ , stable

**11-33.** Unstable equilibrium at  $\theta = 90^\circ$   
Stable equilibrium at  $\theta = 49.0^\circ$

**11-34.** Unstable equilibrium at  $\theta = 0^\circ$   
Stable equilibrium at  $\theta = 72.9^\circ$

**11-35.**  $k = 2.81 \text{ lb/ft}$

**11-37.** Stable equilibrium at  $\theta = 51.2^\circ$   
Unstable equilibrium at  $\theta = 4.71^\circ$

**11-38.**  $k = 157 \text{ N/m}$   
Stable equilibrium at  $\theta = 60^\circ$

**11-39.**  $W = \frac{8k}{3L}$

**11-41.** Stable equilibrium at  $\theta = 24.6^\circ$

**11-42.**  $\phi = 17.4^\circ, \theta = 9.18^\circ$

**11-43.** Unstable equilibrium at  $\theta = 23.2^\circ$

**11-45.**  $\theta = 0^\circ, \theta = 33.0^\circ$

**11-46.**  $m = 5.29 \text{ kg}$

**11-49.**  $\theta = 0^\circ, \theta = \cos^{-1}\left(\frac{d}{4a}\right)$

# Index

- Active force, 89
- Angles, 45–47, 69–73, 82–83, 403–405, 432
  - azimuth ( $\phi$ ), 46–47
  - Cartesian force vectors, 45–47
  - coordinate direction, 45–46, 82–83
  - dot product used for, 69–73, 83
  - dry friction and, 403–405, 432
  - formed between intersecting lines, 70
  - impending motion and, 403–405
  - kinetic friction ( $\theta_k$ ), 404–405
  - lead, 432
  - mathematical review of, 616–617
  - Pythagorean's theorem for, 70, 617
  - screws, 432
  - static friction ( $\theta_s$ ), 403, 405
  - transverse ( $\theta$ ), 46–47
  - vectors and, 45–47, 69–73, 82–81
- Applied force (**P**), 402–405, 459–460
- Area ( $A$ ), 468, 470, 502–505, 523–524,
  - 529–535, 540–542, 548–557, 576
  - axial symmetry and rotation, 502–505, 524, 548–549
  - centroid ( $C$ ) of an, 468, 470, 502–505, 523–524
  - centroidal axis of, 530–531
  - composite shapes, 503, 540–542, 576
  - inclined axis, about, 552–554
  - integration for, 468, 523, 529–532
  - Mohr's circle for, 555–557
  - moments of inertia ( $I$ ) for, 529–535, 540–542, 548–557, 576
  - Pappus and Guldinus, theorems of, 502–505, 524
  - parallel-axis theorem for, 530–531, 540, 549, 567, 576
  - polar moment of inertia, 530–531
  - principal moments of inertia, 553–554
  - procedures for analysis of, 470, 532, 540
  - product of inertia for, 548–551, 576
  - radius of gyration of, 531
  - surface of revolution, 502, 504–505, 524
  - transformation equations for, 552
  - volume of revolution, 503–505, 524
- Associative law, 126
- Axes, 145–149, 190, 202, 529–535, 540–542,
  - 552–557, 563–570, 576–577
  - area moments of inertia for, 529–535, 552–554
  - centroidal axis of, 530–531
  - composite bodies, 540–542, 568
  - distributed loads along single, 190
  - inclined, area about, 552–554
  - mass moments of inertia for, 563–570, 577
  - Mohr's circle for, 555–557
  - moment of a force about specified, 145–149, 202
  - moments of inertia ( $I$ ), 529–535, 540–542, 552–557, 563–570, 576–577
  - parallel-axis theorem for, 530–531, 540, 549, 567, 576
  - principal, 553–554, 556
  - procedures for analysis of, 532, 556, 564
  - product of inertia and, 548–551, 576
  - radius of gyration for, 531, 568
  - resultant forces and, 145–149, 190, 202
  - scalar analysis, 145
  - transformation equations for, 552
  - vector analysis, 146–147
- Axial loads, friction ( $F$ ) and, 447–449, 461
- Axial revolution, 502–505, 524
- Axial symmetry, 488–489, 502–505, 523–524
  - axial revolution and, 502–505, 524
  - center of gravity ( $G$ ) and, 488–489, 502–505, 523
  - centroid ( $C$ ) and, 488–489, 502–505, 523
  - composite bodies, 488–489, 503
  - Pappus and Guldinus, theorems of, 502–505, 524
  - rotation and, 502–505, 524
  - surface area and, 502, 504–505, 524
  - volume and, 503–505, 524
- Axis of symmetry, 467, 469, 488–489, 523, 548–551
  - area product of inertia, 548–551
  - centroid ( $C$ ) and, 467, 469, 488, 523
- Azimuth angles, 46
- Ball and socket connections, 245–246, 248
- Base units, 7
- Beams, 342–380, 396–398
  - bending moments (**M**) and, 344–345, 370–375, 396
  - cantilevered, 361
  - centroid ( $C$ ), 344
  - couple moment (**M**) and, 372
  - distributed loads, 370–375, 398
  - force equilibrium, 370–371
  - free-body diagrams, 343–350, 396
  - internal forces, 342–380, 396–398
  - internal loads of, 361–364, 370–375
  - method of sections for, 343–350
  - moments, 344–345, 370–375, 396
  - normal force (**N**) and, 344–345, 396
  - procedures for analysis of, 345, 362
  - resultant loadings, 344, 396
  - shear and moment diagrams, 361–364, 397
  - shear force (**V**) and, 344–345, 370–375, 396
  - sign convention for, 345, 397
  - simply supported, 361
  - torsional (twisting) moment, 344, 396
- Bearings, 246–248, 447–451, 461
  - collar, 447–449, 461
  - free-body diagrams, 246–248
  - frictional analysis of, 447–451, 461
  - journal, 246–247, 450–451, 461
  - pivot, 447–449, 461
  - rigid-body support reactions, 246–248
  - thrust, 247–248
- Belts (flat), frictional analysis of, 439–441, 460
- Bending moment diagrams, 361–364. *See also* Shear and moment diagrams
- Bending moments (**M**), 344–345, 370–375, 396, 398
  - distributed loads and, 370–375, 398
  - internal forces and, 344–345, 370–375, 396, 398
  - method of sections for, 344–345
  - shear (**V**) and, 371
- Body at rest (zero), 208
- By inspection, determination of forces, 282, 292
- Cables, 88, 117, 210, 246, 381–395, 398
  - concentrated loads, 381–383, 398
  - continuous, 88, 117
  - distributed loads, 384–387, 398
  - equilibrium of, 88, 117
  - flexibility of, 381
  - free-body diagram for, 88, 246
  - inextensible, 381
  - internal forces of, 381–395, 398
  - support reactions, 88, 246
  - weight of as force, 388–391, 398
- Calculations, engineering importance of, 10–11
- Cantilevered beam, 361
- Cartesian coordinate system, 44–49, 56–58, 69, 82–83, 125–131, 201
  - addition of vectors, 47
  - azimuth angles ( $\phi$ ), 46
  - concurrent force resultants, 47–49, 83
  - coordinate direction angles, 45–46, 82–83
  - coplanar force resultants, 34
  - cross product using, 125–127
  - direction and, 45–47, 125, 128
  - dot product in, 69
  - magnitude in, 45, 82, 125, 128
  - moment of a force, calculations by, 128–131, 201
  - position vectors (**r**), 56–58, 83

- Cartesian coordinate system (continued)
  - rectangular components, 44, 82
  - right-hand rule, 44, 56, 125–126, 128
  - three-dimensional systems, 44–49
  - transverse angles ( $\theta$ ), 46–47
  - two-dimensional systems, 34
  - unit vectors, 44, 82
  - vector formulation, 126–127, 129
  - vector representation, 45, 82–83
- Cartesian vector notation, 34
- Center of gravity ( $G$ ), 6, 212, 464–527
  - center of mass ( $C_m$ ) and, 467, 523
  - centroid ( $C$ ) and, 464–527
  - composite bodies, 488–492, 524
  - constant density and, 488
  - coplanar forces, 212
  - free-body diagrams of, 212
  - location of, 465–466, 470, 523
  - Newton's law and, 6
  - procedure for analysis of, 470, 489
  - rigid-body equilibrium and, 212
  - specific weight and, 488
  - weight ( $W$ ) and, 6, 212, 465–466, 488, 523
- Center of pressure ( $P$ ), 513, 525
- Centroid ( $C$ ), 191, 212, 344, 464–527
  - area in  $x$ - $y$  plane, 468, 523
  - axis of symmetry, 467, 469, 488, 523
  - axial symmetry, 488–489, 502–505, 523–524
  - beam cross-section location, 344
  - center of gravity ( $G$ ) and, 464–527
  - composite bodies, 488–492, 524
  - composite shapes, 503
  - coplanar forces, 212
  - distributed loads and, 511–518, 525
  - distributed loads, 191
  - flat surfaces, 511
  - fluid pressure and, 512–518, 525
  - free-body diagrams and, 212
  - integration for determination of, 467–477, 523
  - line in  $x$ - $y$  plane, 468–469, 523
  - line of action and, 191, 511, 513, 525
  - location of, 191, 467–477, 523
  - mass of a body ( $C_m$ ), 467, 478, 523
  - method of sections and, 344
  - Pappus and Guldinus, theorems of, 502–505, 524
  - plates, 497–518
  - procedure for analysis of, 470, 489
  - Pythagorean's theorem for, 469
  - resultant forces and, 191, 344, 511, 513–518, 525
  - rigid-body equilibrium and, 212
  - rotation of an axis, 502–505, 524
  - surface area and, 502, 504–505, 524
  - volume and, 467, 503–505, 523–524
- Centroidal axis, 530–531
- Coefficient of kinetic friction ( $\mu_k$ ), 404–405
- Coefficient of rolling resistance, 452–453
- Coefficient of static friction ( $\mu_s$ ), 403, 405
- Collar bearings, frictional analysis of, 447–449, 461
- Collinear vectors, 19, 81
- Communitative law, 18, 126
- Components of a force, 18, 20–22
- Composite bodies, 488–492, 503, 503, 524, 540–542, 568, 576–577
  - area of, 503, 540–542, 576
  - axial symmetry and, 488–489
  - center of gravity ( $G$ ), 488–492, 524
  - centroid ( $C$ ) of, 488–492, 503, 524
  - constant density and, 488
  - mass moments of inertia, 568, 577
  - moments of inertia ( $I$ ), 540–542, 568, 576
  - procedure for analysis of, 489, 540
  - theorem of Pappus and Guldinus for parts of, 503
  - specific weight and, 488
  - weight ( $W$ ) and, 488, 524
- Compressive forces ( $C$ ), 275–277, 291–292
  - method of joints and, 276–277
  - method of sections and, 291–292
  - truss members, 275
- Concentrated force, 5
- Concentrated loads, 370–371, 381–383, 397–398
  - cables subjected to, 381–383, 398
  - distributed loads, 370–371
  - shear and moment discontinuities from, 371, 397
- Concurrent forces, 47–49, 83, 106–110, 117, 177, 252
  - addition of vectors, 47–49
  - Cartesian coordinate system for, 47–49, 83
  - couple moments and, 177
  - equilibrium of, 106–110, 117, 252
  - statical determinacy and, 252
  - systems, simplification of, 177
- Conservative forces, 597–599
  - potential energy and, 598–599
  - potential function for, 599
  - spring force, 597
  - virtual work ( $U$ ) and, 597–599
  - weight, 597
- Constant density, center of gravity ( $G$ ) and, 488
- Constraints, 251–259
  - improper, 252–253
  - procedure of analysis of, 254
  - redundant, 251
  - statical determinacy and, 251–259
  - rigid-body equilibrium and, 251–259
- Conversion of units, 9
- Coordinate direction angles, 45–46, 82–83
- Coordinates, 44–49, 56–58, 82–83, 585–586, 600, 612. *See also* Cartesian coordinate system
  - Cartesian, 44–49, 56–58, 82–83
  - direction angles ( $\theta$ ), 45–46, 82
  - frictionless systems, 600
  - position, 585–586, 600, 612
  - potential energy and, 600
  - vector representation, 44–49, 56–58
  - virtual work for rigid-body connections, 585–586, 600, 612
  - $x$ ,  $y$ ,  $z$  positions, 56
- Coplanar distributed loads, 190–194
- Coplanar forces, 33–38, 82, 91–95, 117, 166–171, 177, 208–244, 268–269
  - addition of systems of, 33–38
  - Cartesian vector notation, 34
  - center of gravity, 212
  - centroid (geometric center), 212
  - couple moments and, 166–171, 177
  - direct solution for unknowns, 220–229, 269
  - direction of, 33, 34
  - equations of equilibrium, 91, 208, 220–229
  - equilibrium of, 91–95, 117, 208–244, 268–269
  - equivalent system, 166–171
  - free-body diagrams, 91–92, 209–218, 268
  - idealized models of, 212–213
  - internal forces and, 212
  - magnitude of, 33, 34, 91
  - particles subjected to, 91–95, 117
  - procedure for analysis of, 92, 214, 221
  - rectangular components, 33–38, 82
  - resultants, 34–38
  - rigid bodies, 208–244, 268–269
  - scalar notation, 33, 34
  - support reactions, 209–211, 268
  - system components, 33–38
  - systems, simplification of, 166–171, 177
  - three-force members, 230–231
  - two-force members, 230–231
  - vectors for, 33–38, 82
  - weight and, 212
- Cosine functions, 617

- Cosine law, 22, 81
- Coulomb friction, 401. *See also*
  - Dry friction
- Couple, 154
- Couple moments ( $M_0$ ), 154–159, 166–171, 177–183, 202–203, 372, 582–583
  - concurrent force systems and, 177
  - coplanar force systems and, 166–171, 177
  - distributed load relationships, 372
  - equivalent couples, 155
  - equivalent system, 166–171
  - force systems, 154–159
  - free vectors, 154
  - internal forces and, 372
  - parallel force systems and, 178
  - procedure for analysis of, 168
  - resultant, 155–156
  - right-hand rule for, 154
  - rotation of, 582
  - scalar formulation of, 154
  - shear load ( $\mathbf{V}$ ) relationships, 372
  - systems, simplification of, 166–171, 177–183
  - three-dimensional systems, 166–171, 177–183
  - translation of, 582
  - vector formulation of, 154
  - virtual work of, 583
  - work of, 582
  - wrench, reduction of forces to, 179
- Cross product, 125–127
  - Cartesian vector formulation, 126–127
  - direction and magnitude by, 125
  - laws of operation, 126
  - right-hand rule for, 125–126
  - vector multiplication using, 125–127
- Curved plates, fluid pressure and, 514
- Cylinders, rolling resistance of, 452–453, 461
- Derivatives, 618
- Derived units, 7–8
- Dimensional homogeneity, 10
- Direct solution for unknowns, 220–229, 269
- Direction, 17, 33, 34, 45–47, 70, 81, 122, 125, 128, 201, 405, 407
  - azimuth angles, 46
  - Cartesian coordinate vectors, 45–47
  - coordinate direction angles, 45–46
  - coplanar force systems, 33, 34
  - cross product and, 125
  - dot product applications, 70
  - frictional forces, 405, 407
  - moments, 122, 125, 128, 201
  - right-hand rule for, 125, 128, 201
  - three-dimensional systems, 45–47
  - transverse angles, 46–47
  - vector sense of, 17, 33, 34, 81
- Direction cosines, 45–46
- Disks, 447–449, 461, 564, 577
  - frictional analysis of, 447–449, 461
  - mass moments of inertia, 564, 577
- Displacement ( $\delta$ ), 583–590, 600, 612
  - frictionless systems, 600
  - potential energy and, 600
  - principle of virtual work and, 583–590, 612
  - procedure for analysis of, 586
  - rigid-bodies, connected systems of, 585–590
  - virtual work ( $U$ ) and, 583–590, 600, 612
  - virtual work equations for, 583
- Distributed loads, 190–194, 203, 370–375, 384–387, 397–398, 511–518, 525
  - axis representation, along single, 190
  - beams subjected to, 370–375, 397–398
  - bending moment ( $\mathbf{M}$ ) relationships, 370–375, 398
  - cables subjected to, 384–387, 398
  - center of pressure ( $P$ ), 513, 525
  - centroid ( $C$ ) of, 191, 511–518, 525
  - concentrated loads, 370–371
  - coplanar, 190
  - couple moment ( $M_0$ ) relationships, 372
  - fluid pressure and, 512–518, 525
  - force equilibrium, 370–371
  - force system resultants, 190–194, 203
  - incompressible fluids, 512
  - internal forces, 370–375, 384–387, 397–398
  - linearly, 513, 515, 525
  - line of action of, 191
  - magnitude and, 190, 511, 525
  - reduction of force and, 190–194, 203
  - resultant forces of, 190–194, 203, 511, 525
  - shear force ( $\mathbf{V}$ ) relationships, 370–375, 398
  - uniform, 370, 525
- Distributive law, 69, 132
- Dot notation, 10
- Dot product, 69–73, 83, 146
  - applications in mechanics, 70
  - Cartesian vector formulation, 69
  - laws of operation, 69
  - moment about a specified axis, 146
  - vector angles and direction from, 69–73, 83
- Dry friction, 400–463
  - angles ( $\theta$ ) of, 403–404
  - applied force ( $\mathbf{P}$ ) and, 402–405, 459–460
  - bearings, analysis of, 447–451, 447
  - belts (flat), analysis of, 439–441, 460
  - collar and pivot bearings, analysis of, 447–449, 461
  - characteristics of, 401–405, 459
  - coefficients of ( $\mu$ ), 403–405, 459
  - direction of force, 405, 407
  - disks, analysis of, 447–449
  - equations for friction versus equilibrium, 407–414
  - equilibrium and, 402, 407
  - impending motion, 403, 406–414, 432–433, 459–460
  - journal bearings, analysis of, 450–451, 461
  - kinetic force ( $F_k$ ), 404–405, 459
  - motion and, 403–405, 406–414, 432–434, 459–460
  - problems involving, 406–414
  - procedure for analysis of, 409
  - rolling resistance and, 452–453, 461
  - screws, forces on, 432–434, 460
  - slipping and, 403–405, 406–414, 459
  - static force ( $F_s$ ), 403, 405, 459
  - theory of, 402
  - tipping effect, balance of, 402, 459
  - wedges and, 430–431, 460
- Dynamics, study of, 3
- Elastic potential energy ( $V_e$ ), 598
- Engineering notation, 11
- Equations of equilibrium, 87, 91, 106, 208, 220–229, 250, 268–269, 407–414
  - alternative sets, 220–221
  - body at rest (zero), 208
  - coplanar force systems, 91, 220–229, 268–269
  - direct solution, 220–229, 269
  - frictional equations and, 407–414
  - particles, 87, 91, 106
  - procedure for analysis using, 221
  - rigid bodies, 208, 220–229, 268–269
  - scalar form, 250, 268–269
  - three-dimensional force systems, 106, 250, 269
  - three-force members, 230–231
  - two-force members, 230–231
  - vector form, 250, 269
- Equilibrium, 86–119, 206–271, 370–371, 402, 407–414, 600–606, 613
  - concurrent forces, 106–110, 117

- Equilibrium (continued)
- conditions for, 87, 207–208, 220
  - coplanar force systems, 91–95, 117, 208–244, 268–269
  - distributed loads, 370–371
  - free-body diagrams, 88–91, 106, 209–218, 245–248, 268–269
  - friction and, 402, 407
  - frictionless systems, 600
  - impending motion and, 407–414
  - improper constraints and, 252–253
  - neutral, 601–602
  - one degree-of-freedom system, 601
  - particles, 86–119
  - potential-energy ( $V$ ) criterion for, 600, 613
  - procedures for analysis of, 92, 106, 214, 221, 254, 603
  - redundant constraints and, 251
  - rigid bodies, 206–271
  - stability of systems, 601–606, 613
  - stable, 601–602
  - statical determinacy and, 251–259, 269
  - support reactions, 209–211, 245–249, 268–269
  - three-dimensional force systems, 106–110, 117, 245–259, 269
  - three-force members, 230–231
  - tipping effect, balance of, 402, 459
  - two-dimensional force systems, 91–95, 117
  - two-force members, 230–231
  - unstable, 601–602
  - virtual work ( $U$ ) and, 600–606, 613
  - zero condition, 87, 117, 208
- Equivalent couples, 155
- Equivalent systems, 166–171, 177–183
- concurrent force system, 177
  - coplanar systems, 166–171, 177
  - force and couple moment simplification, 166–171, 177–183
  - parallel force systems, 178
  - principle of transmissibility for, 166
  - procedures for analysis, 168, 178
  - wrench, reduction to, 179
  - three-dimensional systems, 166–171, 177
- Exponential notation, 10
- External effects, 166
- External forces, 207, 305
- Fixed supports, 209, 211, 247
- Flat plates, 511, 513, 515, 525
- constant width, 513
  - distributed loads on, 511, 525
  - fluid pressure and, 513, 515, 525
  - variable width, 515
- Floor beams, truss analysis and, 274
- Fluid pressure, 512–518, 525
- center of pressure ( $P$ ), 513
  - centroid ( $C$ ), 512–518, 525
  - curved plate of constant width, 514
  - flat plate of constant width, 513
  - flat plate of variable width, 515
  - incompressible fluids, 512
  - line of action, 513
  - Pascal's law, 512
  - plates, 512–518, 525
  - resultant forces and, 513–518, 525
- Force, 4, 5–9, 16–85, 86–119, 120–205, 212, 230–231, 275–277, 291, 305, 342–399, 402–405, 459–460, 511, 513–518, 525, 581–583, 585–590, 597–598
- active, 89
  - addition of vectors, 20–26, 33–38, 47–49
  - applied ( $\mathbf{P}$ ), 402–405, 459–460
  - axis, about a specified, 145–149, 190
  - basic quantity of mechanics, 4
  - by inspection, 282, 292
  - cables, 88, 381–395
  - Cartesian vector notation for, 34
  - components of, 20–22, 33–38
  - compressive ( $C$ ), 275–277, 291–292
  - concentrated, 5
  - concurrent, 47–49, 83, 166–171, 177
  - conservative, 597–598
  - coplanar, 33–38, 91–95, 117, 166–171, 177, 203
  - couple moments and, 154–159, 166–171, 177–183, 203
  - cross product, 125–127
  - directed along a line, 59–62
  - displacements from, 585–590
  - distributed loads, 190–194, 203, 511, 525
  - dot product, 69–73, 83
  - equilibrium and, 86–119, 230–231, 370–371
  - equivalent system, reduction to, 166–171, 177–183
  - external, 207, 305
  - free-body diagrams, 88–92, 117, 291–296, 305, 343–350
  - friction, 402–405, 459, 597
  - gravitational, 7
  - internal, 212, 291, 305, 342–399
  - kinetic frictional ( $F_k$ ), 404–405, 459
  - line of action, 17, 59–62, 83
  - method of sections for, 291–296, 343–350
  - moment of, 121–124, 128–131, 145–149, 154–159, 166–171, 201–202
  - motion and, 403–405
  - Newton's laws, 6–7
  - nonconservative, 597
  - normal ( $\mathbf{N}$ ), 344–345, 396, 402–403
  - parallel systems, 178
  - parallelogram law for, 18, 20–22, 81
  - particles subjected to, 86–119
  - position vectors and, 56–58, 83
  - principle of moments, 132–134
  - principle of transmissibility, 128, 166
  - procedures for analysis of, 22, 89, 92, 168, 178, 345
  - pulleys, 88
  - reactive, 89
  - rectangular components, 33–38, 44, 82
  - resultant, 18, 20–22, 34–38, 120–205, 511, 513–518, 525
  - scalar notation for, 33, 34
  - scalars and, 17, 18, 69, 81, 121–124, 201
  - shear ( $\mathbf{V}$ ), 344–345, 370–375, 396, 398
  - simplification of systems, 166–171, 203
  - spring ( $\mathbf{F}_s$ ), 597
  - springs, 88
  - static frictional ( $F_s$ ), 403, 405, 459
  - structural analysis and, 275–277, 291–292, 305, 343–350
  - structural members, 230–231, 274–275, 292–293, 343–380
  - systems of, 33–38, 120–205
  - tensile ( $T$ ), 275–277, 291–292
  - three-dimensional systems, 44–49, 56–58, 106–110, 117, 166–171
  - unbalanced, 6
  - units of, 8–9
  - unknown, 291–292
  - virtual work ( $U$ ) and, 581–583, 585–590, 597–598
  - weight, 7, 388–391, 398, 597
  - work ( $W$ ) of, 581–583
  - wrench, reduction to, 179
  - vectors and, 16–85, 86–119, 125–131, 201
- Frames, 305–320, 337
- free-body diagrams for, 305–311, 337
  - procedure for analysis of, 311
  - structural analysis of, 305–320, 337
- Free vector, 154
- Free-body diagrams, 88–92, 106, 117, 209–218, 245–249, 251, 268–269, 291–296, 305–311, 337, 343–350, 396
- beams, 343–350, 396
  - cables, 88
  - center of gravity, 212

- centroid (geometric center), 212
- concurrent forces, 106
- coplanar force systems, 91–92, 209–218, 221, 268
- equilibrium and, 88–92, 209–218, 221, 245–259, 251, 268–269
- external forces and, 305
- frames, 305–311, 337
- idealized models of, 212–213
- internal forces and, 212, 305, 343–350, 396
- machines, 305–311, 337
- method of sections using, 291–296, 343–350
- particle equilibrium, 88–92
- procedures for analysis using, 214, 221, 254, 311
- pulleys, 88
- rigid bodies, 209–218, 245–249, 251, 268–269
- smooth surfaces, 88
- springs, 88
- statical determinacy and, 251, 269
- structural analysis using, 291–296, 305–311, 337
- support reactions, 209–211, 245–248, 251, 268–269
- three-dimensional systems, 245–249, 251, 269
- weight and, 212
- Frictional circle, 450
- Friction ( $F$ ), 400–463, 597
  - angles ( $\theta$ ) of, 403–404
  - applied force ( $\mathbf{P}$ ), 402–405, 459–460
  - axial loads and, 447–449, 461
  - bearings, analysis of, 447–451, 461
  - belts (flat), forces on, 439–441, 460
  - characteristics of, 401–405, 459
  - coefficients of ( $\mu$ ), 403–405, 452–453, 459
  - collar bearings, analysis of, 447–449, 461
  - Coulomb, 401
  - disks, analysis of, 447–449, 461
  - dry, 400–463
  - equations for friction and equilibrium, 407–414
  - equilibrium and, 402, 407
  - force of, 402–405, 459
  - impending motion, 403, 406–414, 432–433, 459–460
  - journal bearings, analysis of, 450–451, 461
  - kinetic force ( $F_k$ ), 404–405, 459
  - lateral loads and, 450–451, 461
  - nonconservative force, as a, 597
  - point of contact, 401–402, 404
  - pivot bearings, analysis of, 447–449, 461
  - procedure for analysis of, 409
  - rolling resistance and, 452–453, 461
  - screws, forces of, 432–434, 460
  - shaft rotation and, 447–451, 461
  - slipping and, 404–405, 406–414, 459
  - static force ( $F_s$ ), 403, 405, 459
  - virtual work ( $U$ ) and, 597
  - wedges and, 430–431, 460
- Frictionless systems, 600
- Geometric center, 191, 212, 344. *See also* Centroid ( $C$ )
- Gravitational attraction, Newton's law of, 7
- Gravitational potential energy ( $V_g$ ), 598
- Gravity, *see* Center of gravity ( $G$ )
- Hinge connections, 209, 212, 245, 247
- Hyperbolic functions, 618
- Idealizations for mechanics, 5
- Impending motion, 403, 406–414, 432–433, 459–460
  - all points of contact, 406
  - angle of static friction for, 403
  - coefficient of static friction ( $\mu_s$ ) for, 403
  - downward, 433, 460
  - dry friction problems due to, 406–414
  - equilibrium and frictional equations for, 407–414
  - friction and, 403, 406–414, 432–433, 459–460
  - no apparent, 406
  - points of contact, 404
  - procedure for analysis of, 409
  - screws and, 432–434, 460
  - some points of contact, 407
  - upward, 432–433, 460
  - verge of slipping, 403
- Inclined axes, moment of inertia for area about, 552–554
- Incompressible fluids, 512
- Inertia, *see* Moments of inertia
- Integrals, 619
- Integration, 467–477, 511, 515, 525, 529–532, 563, 576–577
  - area ( $A$ ) integration, 468, 529–532
  - center of mass ( $C_m$ ), determination of using, 467–477
  - centroid ( $C$ ), determination of using, 467–477, 511, 515, 525
  - distributed loads, 511, 515, 525
  - line segment, 468–469
  - mass moments of inertia, determination of using, 563, 577
  - moments of inertia, determination of using, 529–532, 576
  - parallel-axis theorem for, 530–531
  - pressure distribution and, 515, 525
  - procedure for analysis using, 532
  - resultant force integration, 511, 525
  - volume ( $V$ ), 467
  - volume elements for, 563
- Internal forces, 212, 291, 305, 342–399
  - beams subjected to, 342–380, 396–398
  - bending moments ( $\mathbf{M}$ ) and, 344–345, 370–375, 396, 398
  - cables subjected to, 381–395, 398
  - compressive ( $C$ ), 291
  - concentrated loads, 370–371, 381–383, 397–398
  - couple moment ( $M_0$ ) and, 372
  - distributed loads, 370–375, 397–398
  - force equilibrium, 370–371
  - frames, 305
  - free-body diagrams, 305, 343–350, 396
  - machines, 305
  - method of sections and, 291, 343–350
  - moments and, 344–345, 370–375, 396–398
  - normal force ( $\mathbf{N}$ ) and, 344–345, 396
  - procedures for analysis of, 345, 362
  - resultant loadings, 344, 396
  - rigid-body equilibrium and, 212
  - shear and moment diagrams, 361–364, 397
  - shear force ( $\mathbf{V}$ ) and, 344–345, 370–375, 396, 398
  - sign convention for, 345, 397
  - structural members with, 343–350, 396
  - tensile ( $T$ ), 291
  - torsional (twisting) moment, 344, 396
  - weight, 388–391, 398
- International System (SI) of units, 8, 9–10
- Joints, truss analysis and, 273–274, 276–281. *See also* Method of joints
- Joules ( $J$ ), unit of, 582
- Journal bearings, 246–248, 450–451, 461
  - frictional analysis of, 450–451, 461
  - support connections, 246–248
- Kinetic frictional force ( $F_k$ ), 404–405, 459
- Lateral loads, friction ( $F$ ) and, 450–451, 460
- Laws of operation, 69
- Lead of a screw, 432



- Lead angle, 432
- Length, 4, 8–9, 468–470, 523
  - basic quantity of mechanics, 4
  - centroid ( $C$ ) of lines, 468–470, 523
  - procedure for analysis of, 470
  - Pythagorean theorem for, 469
  - units of, 8–9
- Line of action, 17, 59–62, 83, 191, 511, 513, 525
  - force vector directed along, 59–62, 83
  - resultant force, 191, 511
  - vector representation of, 17
- Linear elastic behavior, 88
- Lines, centroid ( $C$ ) of, 468–469. *See also* Length
- Loads, 190–194, 274, 370–375, 381–383, 396–398, 447–451, 461, 512–518, 525.
  - See also* Distributed loads
  - axial, 447–449, 461
  - beams, 370–375, 396–397
  - cables, 381–383, 398
  - concentrated, 370–371, 381–383, 397–398
  - distributed, 370–375, 398
  - fluid pressure, 512–518
  - friction ( $F$ ) and, 447–451, 461
  - lateral, 450–451, 461
  - linear distribution of, 513–514, 525
  - moment ( $M$ ) relations with, 370–375, 398
  - resultant forces, 190–192
  - reduction of distributed, 190–194
  - shaft rotation and, 447–451, 461
  - shear ( $V$ ), 370–375, 396, 398
  - single axis representation, 190
  - three-dimensional, 344, 396
  - truss joints, 274
  - uniform, 525
- Machines, 305–320, 337
  - free-body diagrams for, 305–311, 337
  - procedure for analysis of, 311
  - structural analysis and, 305–320, 337
- Magnitude, 17, 33, 34, 44, 88, 91, 122, 125, 128, 190, 201, 511, 525
  - Cartesian vectors, 45
  - coplanar force systems, 33, 34, 91
  - constant, 88
  - cross product and, 125
  - distributed load reduction and, 190, 511, 525
  - equilibrium and, 88, 91
  - integration for, 511, 525
  - moments and, 122, 125, 128, 201
  - resultant forces, 190, 511, 525
  - right-hand rule for, 128
  - vector representation of, 17, 33, 34, 45
  - units of, 122
- Mass, 4, 8–9, 467, 478, 523
  - basic quantity of mechanics, 4
  - center of ( $C_m$ ), 467, 478, 523
  - integration of, 467, 523
  - units of, 8–9
- Mass moments of inertia, 563–570, 577
  - axis systems, 563–570, 563, 577
  - composite bodies, 568, 577
  - disk elements, 564, 577
  - parallel-axis theorem for, 567
  - procedure for analysis of, 564
  - radius of gyration for, 568
  - shell elements, 564, 577
  - volume elements for integration, 563
- Mathematical expressions, 616–619
- Mechanics, study of, 3
- Members, 230–231, 274–275, 292–293, 343–350, 396
  - compressive force ( $C$ ), 275
  - equilibrium of forces, 230–231
  - internal loads in, 343–350, 396
  - joint connections, 274
  - tensile force ( $T$ ), 275
  - three-force, 230–231
  - truss analysis and, 274–275, 291–292
  - two-force, 230–231
  - unknown forces, 291–292
- Method of joints, 276–284, 301, 335
  - compressive forces, 276–277
  - procedures for analysis using, 277, 301
  - space truss analysis, 301
  - structural analysis using, 276–284, 301, 335
  - tensile forces, 276–277
  - truss analysis, 276–284, 301, 335
  - zero-force members, 282–284
- Method of sections, 291–296, 301, 336, 343–350
  - compressive forces, 291–292
  - internal forces from, 291, 343–350
  - free-body diagrams for, 291–296, 343–350
  - procedures for analysis using, 293, 301, 345
  - space truss analysis, 301
  - structural analysis using, 291–296, 301, 336, 343–350
  - tensile forces, 291–292
  - truss analysis, 291–296, 336
  - unknown member forces, 291–292, 336
- Models, idealized rigid bodies, 212–213
- Mohr's circle, 555–557
- Moment arm (perpendicular distance), 121–122
- Moment axis, 122, 145–149, 202
  - direction and, 122
  - force about a, 145–149, 202
  - scalar analysis of, 145
  - vector analysis of, 146–147
- Moments ( $M$ ), 120–205, 344–345, 370–375, 396, 398
  - bending ( $\mathbf{M}$ ), 344–345, 370–375, 396, 398
  - concentrated load discontinuities, 371
  - couple ( $M_0$ ), 154–159, 166–171, 177–183, 202–203, 372
  - cross product for, 125–127
  - direction and, 122, 125, 128, 201
  - distributed loads and, 190–194, 203, 370–375, 398
  - force, of, 120–205
  - free vector, 154
  - internal forces and, 344–345, 370–375, 396, 398
  - magnitude and, 122, 125, 128, 201
  - parallel force systems and, 178
  - perpendicular to force resultants, 177–183
  - principle of moments, 132–134
  - principle of transmissibility, 128, 166
  - procedures for analysis of, 168, 178
  - resultant, 122–124, 129, 155–156
  - scalar formulation of, 121–124, 154, 201
  - shear loads ( $V$ ) and, 370–375, 398
  - sign convention for, 122, 126
  - system simplification of, 166–171, 177–183, 203
  - torque, 121
  - torsional (twisting), 344, 396
  - Varignon's theorem, 132–134
  - vector formulation of, 126–131, 154, 201
  - wrench, reduction of force and couple to, 179
- Moments of inertia ( $I$ ), 528–579
  - algebraic sum of, 540
  - area ( $A$ ), 529–535, 540–542, 548–557, 576
  - axis systems, 529–535, 540–542, 548–554, 563–570
  - composite shapes, 540–542, 564, 576–577
  - disk elements, 564
  - inclined axis, area about, 552–554
  - integration and, 529–532

- mass, 563–570, 563, 577
- Mohr's circle for, 555–557
- parallel-axis theorem for, 530–531, 540, 549, 567, 576
- polar, 530–531
- principle, 552–554, 556, 577
- procedures for analysis of, 532, 540, 556, 564
- product of inertia and, 548–551, 576
- radius of gyration for, 531, 568
- shell elements, 564
- transformation equations for, 552
- Motion, 6, 403–414, 430–434, 439–441, 447–435, 459–461
  - belt drives, 439–441, 460
  - coefficients of friction ( $\mu$ ) and, 403–405, 452–453, 459
  - downward, 433, 460
  - equilibrium and frictional equations for, 407–414
  - friction and, 403–414, 430–434, 439–441, 447–435, 459–460
  - impending, 403, 406–414, 432–433, 459–460
  - kinetic frictional force ( $F_k$ ), 404–405, 459
  - Newton's laws of, 6
  - points of contact, 404
  - procedure for analysis of, 409
  - rolling resistance and, 452–453, 461
  - screws and, 432–434, 460
  - self-locking mechanisms, 430, 433
  - shaft rotation, 447–451, 461
  - slipping, 404–405, 406–414, 459
  - static frictional force ( $F_s$ ), 403, 405, 459
  - upward, 432–433, 460
  - verge of sliding, 403
  - wedges, 430–431, 460
- Multiforce members, 305. *See also* Frames; Machines
- Neutral equilibrium, 601–602
- Newton, unit of, 8
- Newton's laws, 6–7
  - gravitational attraction, 7
  - motion, 6
- Nonconservative force, friction as a, 597
- Normal force (**N**), 344–345, 396, 402–403
  - friction and, 402–403
  - internal forces as, 344–345
  - method of sections for, 344–345
- Numerical calculations, importance of, 10–11
- Pappus and Guldinus, theorems of, 502–505, 524
  - axial revolution and symmetry, 502–505
  - centroid (*C*) and, 502–505, 524
  - composite shapes, 503
  - surface area and, 502, 504–505, 524
  - volume and, 503–505, 524
- Parallel-axis theorem, 530–531, 540, 549, 567, 576
  - area moments of inertia determined by, 530–531
  - area product of inertia determined by, 549, 576
  - centroidal axis for, 530–531, 576
  - composite areas, 540
  - mass moments of inertia, 567
  - moments of inertia, 530–531, 540, 567, 576
  - product of inertia determined by, 549, 576
- Parallel force and couple moments, simplification of, 178
- Parallelogram law, 18, 20–22, 81
- Particles, 5–7, 86–119
  - coplanar force systems, 91–95, 117
  - defined, 5
  - equations of equilibrium, 87, 91, 106
  - equilibrium of, 86–119
  - free-body diagrams, 88–91
  - gravitational attraction, 7
  - Newton's laws applied to, 6–7
  - nonaccelerating reference of motion, 6
  - procedures for analysis of, 92, 106
  - three-dimensional force systems, 106–110, 117
  - two-dimensional force systems, 91–95, 117
  - zero condition, 87, 117
- Perpendicular distance (moment arm), 121–122
- Pin connections, 209–211, 213, 247–248, 274
  - coplanar systems, 209–211, 213
  - free-body diagrams of, 209–211, 247–248
  - three-dimensional systems, 247–248
  - truss member joints, 274
- Pivot bearings, frictional analysis of, 447–449
- Planar truss, 273
- Plates, 511–518, 525
  - flat of constant width, 513
  - distributed loads on, 511
  - flat of variable width, 515
  - centroid (*C*), 511–518, 525
  - curved of constant width, 514
  - flat of constant width, 499
  - flat of variable width, 501
  - fluid pressure and, 512–518, 525
  - resultant forces acting on, 511, 513–518, 525
- Point of contact, 401–402, 404
- Polar moments of inertia, 530–531
- Position coordinates, 585–586, 600, 612
- Position vectors (**r**), 56–58, 83
  - head-to-tail addition, 56–57
  - x*, *y*, *z* coordinates, 56, 83
- Potential energy (*V*), 598–606, 613
  - elastic ( $V_e$ ), 598
  - equilibrium, criterion for, 600, 613
  - equilibrium configurations, 601–606
  - frictionless systems, 600
  - gravitational ( $V_g$ ), 598
  - position coordinates for, 600
  - potential function equations, 599
  - procedure for analysis of, 603
  - single degree-of-freedom systems, 599, 601
  - stability of systems and, 601–606, 613
  - virtual work (*V*) and, 598–606, 613
- Power-series expansions, 618
- Pressure, *see* Fluid pressure
- Principal axes, 552–554, 556
- Principle moments of inertia, 553–554, 556, 563
- Principle of moments, 132–134
- Principle of transmissibility, 128, 166
- Principle of virtual work, 581, 583–590, 612
- Product of inertia, 548–551, 576
  - axis of symmetry for, 548–549
  - moments of inertia of an area and, 548–551, 576
  - parallel-axis theorem for, 549, 576
- Procedure for analysis, 12–14
- Projection, 70, 146
- Pulleys, free-body diagram of, 88
- Purlins, 273
- Pythagorean theorem, 70, 469, 617
- Quadratic formula, 618
- Radius of gyration, 531, 568
- Reactive force, 89
- Rectangular components, force vectors of, 33–38, 44
- Resultants, 18, 20–22, 34–38, 81, 120–205, 344, 396, 511, 513–518, 525
  - axis, moment of force about, 145–149, 190, 202



- Resultants (continued)
- Cartesian vector components, 44
  - Cartesian vector notation for, 34
  - centroid ( $C$ ) and, 191, 344, 511, 513–518, 525
  - concurrent forces, 47–49, 177
  - coplanar forces, 34–38, 177
  - couple moments, 154–159, 166–171, 177–183, 203
  - distributed loads, reduction of, 190–194, 203, 511, 525
  - fluid pressure and, 513–518, 525
  - force components of, 18, 20–22
  - force system, 120–205
  - integration for, 511, 525
  - internal forces, 344, 396
  - line of action, 191, 511, 513
  - magnitude of, 190, 511, 525
  - method of sections for, 344
  - moments of a force, 128–132
  - parallel forces, 178
  - parallelogram law for, 18, 20–22, 81
  - perpendicular to moments, 177–183
  - plates, 511–518, 525
  - principle moments, 132–134
  - procedure for analysis of, 178
  - scalar formulation of, 121–124, 145, 154, 201
  - scalar notation for, 33
  - system reduction for, 166–171, 177–183, 203
  - vector addition for, 18, 20–22
  - vector formulation of, 128–131, 154, 202
  - wrench, reduction to, 179
- Revolution, 502–505, 524
- axial symmetry and, 502–505
  - centroid ( $C$ ) and, 502–505, 524
  - composite shapes, 503
  - Pappus and Guldinus, theorems of, 502–505, 524
  - surface area, 502, 504–505, 524
  - volume, 503–505, 524
- Right-hand rule, 44, 56, 125–126, 128, 154
- cross product direction, 125–126
  - moment of a couple, 154
  - three-dimensional coordinate systems, 44, 56
  - vector formulation, 126, 128
- Rigid bodies, 3, 5, 206–271, 585–590, 612
- center of gravity, 212
  - centroid (geometric center), 212
  - conditions for, 207–208
  - connected systems of, 585–590, 612
  - constraints of, 251–259
  - coplanar force systems, 208–244, 268–269
  - defined, 5
  - displacement ( $\delta$ ) and, 585–590, 600, 612
  - equations of equilibrium for, 208, 220–229, 268–269
  - equilibrium of, 206–271
  - external forces and, 207
  - force and couple systems acting on, 207–208
  - free-body diagrams, 209–218, 245–249, 251, 268–269
  - frictionless systems, 600
  - idealized models of, 212–213
  - internal forces and, 212
  - improper constraints for, 252–253
  - mechanics, study of, 3
  - position coordinates for, 585–586, 600, 612
  - procedures for analysis of, 214, 221, 254, 586
  - redundant constraints for, 251
  - statical determinacy and, 251–259, 269
  - support reactions, 209–211, 245–248, 251–259, 268–269
  - three-dimensional systems, 245–259, 269
  - three-force members, 230–231
  - two-force members, 230–231
  - uniform, 212
  - virtual work ( $V$ ) for, 585–590, 600, 612
  - weight and, 212
- Rocker connections, 210
- Roller connections, 209–210, 213, 246
- Rolling resistance, frictional forces and, 452–453, 461
- Roof truss, 273–274, 335
- Rotation of couple moments, 582. *See also* Revolution; Shaft rotation
- Rounding off numbers, 11
- Scalar notation, 33, 34
- Scalar product, 69
- Scalar triple product, 146
- Scalars, 17, 18, 33, 69, 121–124, 145, 154, 201, 250, 268–269, 582
- couple moments, formulation by, 154
  - dot product and, 69
  - equations of equilibrium, 250, 268–269
  - moment of a force about an axis, 145
  - moment of a force, formulation by, 121–124, 201
  - multiplication and division of vectors by, 18
  - vectors and, 17, 69
  - negative, 33, 91
  - torque, 121
  - work as, 582
- Screw, reduction of force and couple to, 179
- Screws, frictional forces on, 430–434, 460
- Self-locking mechanisms, 430, 433
- Sense of direction, 17
- Shaft rotation, 447–451, 461
- axial loads, 447–449, 461
  - collar and pivot bearings for, 447–449
  - frictional analysis of, 447–451, 461
  - frictional circle, 450
  - journal bearings for, 450–451, 461
  - lateral loads, 450–451, 461
- Shear and moment diagrams, 361–364, 370–375, 397–398
- beam analysis using, 361–364, 370–375
  - couple moment ( $M_0$ ) and, 372
  - discontinuities in, 371
  - distributed load relations and, 370–375, 398
  - internal forces and, 361–364, 370–375, 397–398
  - moment ( $M$ ) relations in, 371–375, 398
  - procedure for analysis of, 362
  - shear force ( $V$ ) relations in, 370–375, 398
- Shear force ( $V$ ), 344–345, 370–375, 396, 398
- beams, 344–345, 370–375, 396, 398
- bending moments ( $M$ ) and, 344–345, 370–375, 396, 398
- concentrated load discontinuities, 371
  - couple moment ( $M_0$ ) and, 372
  - distributed load relations, 370–375, 398
  - internal forces, 344–345, 370–375, 396, 398
  - method of sections for, 344–345
- Shell elements, mass moments of inertia, 564, 577
- Significant figures, 11
- Simple truss, 275
- Simply supported beam, 361
- Sine functions, 617
- Sine law, 22, 81
- Single degree-of-freedom systems, 599, 601
- Sliding vector, 128, 166
- Slipping, 403–414, 459
- friction and, 403–414, 459
  - impending motion of, 403, 406–414, 459
  - kinetic frictional force ( $F_k$ ), 404–405, 459

- motion of, 404–414
  - points of contact, 404
  - problems involving, 406–414
  - static frictional force ( $F_s$ ), 403, 405, 459
  - verge of, 403, 459
- Slug, unit of, 8
- Smooth surface support, 88, 246
- Solving problems, procedure for, 12–14
- Space trusses, structural analysis of, 301–302, 337
- Specific weight, center of gravity ( $G$ ) and, 488
- Spring constant ( $k$ ), 88
- Spring force ( $F_s$ ), virtual work and, 597
- Springs, free-body diagram of, 88, 117
- Stability of a system, 252–253, 269, 601–606, 613. *See also* Equilibrium
  - equilibrium configurations for, 601–602, 613
  - potential energy and, 601–606
  - procedure for analysis of, 603
  - statical determinacy and, 252–253, 269
  - virtual work and, 601–606, 613
- Stable equilibrium, 601–602
- Static frictional force ( $F_s$ ), 403, 405, 459
- Statistical determinacy, 251–259, 269
  - procedure for analysis of, 254
  - improper constraints and, 252–253
  - indeterminacy, 251, 269
  - reactive parallel forces, 243
  - redundant constraints and, 251
  - rigid-body equilibrium and, 251–259, 269
  - stability and, 252–253, 269
- Statically indeterminate bodies, 251, 269
- Statics, 2–15
  - basic quantities, 4
  - concentrated force, 5
  - force, 4, 5–9
  - gravitational attraction, 7
  - historical development of, 4
  - idealizations, 5
  - length, 4, 8–9
  - mass, 4, 8–9
  - mechanics study of, 3
  - motion, 6
  - Newton's laws, 6–7
  - numerical calculations for, 10–11
  - particles, 5
  - procedure for analysis of, 12–14
  - rigid bodies, 5
  - study of, 2–15
  - time, 4, 8
  - units of measurement, 7–10
  - weight, 7
- Stiffness factor ( $k$ ), 88
- Stringers, 274
- Structural analysis, 272–341, 343–350
  - compressive forces ( $C$ ), 275–277, 291–292
  - frames, 305–320, 337
  - free-body diagrams, 291–296, 305–311, 337
  - internal forces and, 343–350
  - machines, 305–320, 337
  - method of joints, 276–284, 301, 335
  - method of sections, 291–296, 301, 336, 343–350
  - multiforce members, 305
  - procedures for analysis of, 277, 293, 301, 311, 345
  - space trusses, 301–302, 337
  - tensile forces ( $T$ ), 275–277, 291–292
  - trusses, 273–304, 335–337
  - zero-force members, 282–284
- Structural members, *see* Members
- Support reactions, 209–211, 245–248, 251–259, 268–269
  - coplanar force systems, 209–211, 268
  - improper constraints, 252–253
  - procedure for analysis of, 254
  - redundant constraints, 251
  - rigid-body equilibrium and, 209–211, 245–248, 268–269
  - statical determinacy and, 251–259, 269
  - three-dimensional force systems, 245–248, 251–259, 269
- Surface area, centroid ( $C$ ) and, 502, 504–505, 524
- Symmetry, *see* Axial symmetry; Axis of symmetry
- System simplification, 166–171, 177–183
  - concurrent force system, 177
  - coplanar force system, 177
  - coplanar systems, 166–171, 177
  - equivalent system, reduction to, 166–171, 177–183
  - force and couple moments, 167
  - parallel force systems, 178
  - procedures for analysis, 168, 178
  - reduction to a wrench, 179
  - three-dimensional systems, 166–171, 177
- Tangent functions, 617
- Tensile forces ( $T$ ), 275–277, 291–292
  - method of joints and, 276–277
  - method of sections and, 291–292
  - truss members, 275
- Tetrahedron form, 301
- Thread of a screw, 432
- Three-dimensional systems, 44–49, 56–58, 82–83, 106–110, 117, 166–171, 245–259, 269.
  - See also* Concurrent forces
  - addition of vectors, 47
  - azimuth angles, 46
  - Cartesian coordinate system for, 44–49, 82–83
  - Cartesian unit vectors, 44
  - Cartesian vector representation, 45
  - concurrent forces, 47–49, 83, 106–110, 117, 252
  - constraints for, 251–259, 269
  - coordinate direction angles, 45–46
  - direction and, 45–47
  - equations of equilibrium, 106, 250, 269
  - equilibrium of, 106–110, 117, 245–259, 269
  - equivalent system, 166–171
  - force and couple moments, 166–171
  - force vectors, 44–49
  - free-body diagrams, 106
  - magnitude in, 45
  - particles, 106–110, 117
  - position vectors, 56–58, 83
  - procedure for analysis of, 106
  - reactive parallel forces, 253
  - rectangular components, 44
  - resultants, 47–49
  - right-hand rule, 44, 56
  - rigid bodies, 245–259
  - statical determinacy and, 251–259, 269
  - support reactions for, 245–248, 251–259, 269
  - transverse angles ( $\theta$ ), 46–47
- Three-force member equilibrium, 230–231
- Thrust bearing connections, 247, 248
- Time, 4, 8
  - basic quantity of mechanics, 4
  - units of, 8
- Tipping effect, balance of, 402, 459
- Torque, 121. *See also* Moments ( $M$ )
- Torsional (twisting) moment, 344, 396
- Transformation equations, moments of inertia ( $I$ ) and, 552
- Translation of a couple moment, 582
- Transverse angles, 46–47
- Triangle rule, 18–19, 81
- Triangular truss, 275
- Trigonometric identities, 618

- Trusses, 273–304, 335–337  
 assumptions for design, 274–275, 301  
 compressive force ( $C$ ) and, 275–277, 291–292  
 floor beams, 274  
 joints, 273–274, 276–281  
 method of joints, 276–284, 301, 335  
 method of sections, 291–296, 301, 336  
 planar, 273  
 procedures for analysis of, 277, 293, 301  
 purlins, 273  
 roof, 273–274, 335  
 simple, 273–275  
 space trusses, 301–302, 337  
 stringers, 274  
 structural analysis for, 273–304, 335–337  
 tensile force ( $T$ ) and, 275–277, 291–292  
 zero-force members, 282–284
- Two-dimensional systems, 33–38, 82, 91–95, 208–244. *See also* Coplanar forces  
 force vectors, 33–38, 82  
 particle equilibrium, 91–95  
 rigid-body equilibrium, 208–244
- Two-force member equilibrium, 230–231
- U.S. Customary (FPS) system of units, 8
- Uniform distributed load, 370, 525
- Uniform rigid bodies, 212
- Unit vectors, 44, 59, 82
- Units of measurement, 7–10  
 base, 7  
 conversion of, 9  
 derived, 7–8  
 International System (SI) of, 8, 9–10  
 prefixes, 9  
 rules for use, 10  
 U.S. Customary (FPS) system of, 8
- Unknown member forces, 291–292, 336
- Unstable equilibrium, 601–602
- Varignon's theorem, 132–134
- Vectors, 16–85, 125–131, 146–147, 154, 201, 250, 269  
 addition of, 18–19, 47  
 addition of forces, 20–26, 33–38  
 Cartesian coordinate system, 44–49, 56–58, 69, 125–131, 201  
 Cartesian notation for, 34  
 components of a force, 18, 20–22, 81  
 concurrent forces, 47–49, 83  
 coplanar force systems, 33–38  
 cross product method of multiplication, 125–127  
 collinear, 19, 81  
 couple moments, formulation by, 154  
 direction and, 17, 33, 34, 45–47  
 division by scalars, 18  
 dot product, 69–73, 83  
 equations of equilibrium, 250, 269  
 force directed along a line, 59–62  
 forces and, 16–85  
 free, 154  
 line of action, 17, 59–62, 83  
 magnitude and, 17, 33, 34, 45  
 moment of a force about an axis, 146–147  
 moments of a force, formulation by, 128–131, 201  
 multiplication by scalars, 18  
 operations, 18–19  
 parallelogram law for, 18, 20–22, 81  
 physical quantity requirements, 17  
 position ( $\mathbf{r}$ ), 56–58, 83  
 procedure for analysis of, 22  
 rectangular components, 33–38, 44, 82  
 resultant of a force, 18, 20–22, 81  
 scalar notation for, 33  
 scalars and, 17, 18, 69, 81  
 sliding, 128, 166  
 subtraction, 19  
 systems of coplanar forces, 33–38  
 three-dimensional systems, 44–49, 82–83  
 triangle rule for, 18–19, 81  
 two-dimensional systems, 33–38, 82  
 unit, 44, 59, 82
- Virtual work ( $U$ ), 580–615  
 conservative forces and, 597–599  
 couple moment, work of, 582–583  
 displacement ( $\delta$ ) and, 583–590, 600, 612  
 equations for, 583  
 equilibrium and, 600–606, 613  
 force ( $\mathbf{F}$ ) and, 581–583, 585–590, 597–598, 612  
 friction and, 598  
 frictionless systems, 600  
 movement as, 583  
 position coordinates for, 585–586, 600, 612  
 potential energy ( $V$ ) and, 598–606, 613  
 principle of, 581, 583–590, 612  
 procedures for analysis using, 586, 603  
 rigid-bodies, connected systems of, 585–590  
 single degree-of-freedom systems, 599, 601  
 spring force ( $\mathbf{F}_s$ ) and, 597  
 stability of a system, 601–606, 613  
 weight ( $\mathbf{W}$ ) and, 597  
 work ( $\mathbf{W}$ ) of a force, 581–583
- Volume ( $V$ ), 467, 470, 503–505, 523–524  
 axial rotation and symmetry, 503–505, 524  
 centroid of ( $C$ ), 467, 470, 503–505, 523–524  
 integration of, 467, 523  
 Pappus and Guldinus, theorems of, 503–505, 524  
 procedure for analysis of, 470
- Wedges, 430–431, 460
- Weight ( $W$ ), 7, 212, 388–391, 398, 465–466, 488, 523–524, 597  
 cables subjected to own, 388–391, 398  
 center of gravity ( $G$ ) and, 212, 465–466, 523–524  
 composite bodies, 488, 524  
 conservative force of, 597  
 gravitational attraction and, 7  
 internal force of, 388–391, 398  
 rigid-body equilibrium and, 212  
 virtual work ( $U$ ) and, 597
- Work ( $\mathbf{W}$ ) of a force, 581–583. *See also* Virtual work
- Wrench, reduction of force and moment to, 179
- $x$ ,  $y$ ,  $z$  position coordinates, 56, 83
- Zero condition of equilibrium, 87, 117, 208
- Zero-force members, method of joints and, 282–284